11.1 GIVEN: X=42 - 623+22-1 X-m. 1-5 FIND: X, N, AND Q AT to 25 x= 4t - 6t - 2t-1 HAVE .. N = dx = 16t3 - 18t2 + 2 Q = dx = 48t2 - 36t THEN AND AT t= 25: X= 4(2) -6(2) -2(2)-1 OR X= 19 m 1 = 16(5) -18(5) +Z N= 58 3 OR 0= 150 35 0= 4B(2)2-36(2) OR 11.2 GIVEN: X=3t4+4t3-7t2-51.8 X~mm.t-5 FIND: X, N, AND Q AT 1.35 X= 314+413-72-51+B HAVE -N= dx = 12t3+12t2-14t-5 THEN AND a = 2 = 36t2 + 24t - 14 AT t=35: X=3(3)+4(5)-7(3)2-5(3)+8 OR X=281 mm OR N= 385 5 N=12(3)3+12(3)2-14(3)-5 OR 0:382 mm a=36(3)2+24(3)-14 11.3 GIVEN: X=6t2-8+40 cos Tit X~IN., t~5 FIND: X, N, AND a AT 1:65 X= 62-8+40 cosmt HAVE -N= SE = 12t-4017 SINTE THEN a = # = 12 - 4012 cosnt AND AT 1=65: X=6(6) -8+40 cos 67 OR X= 24B IN. OR Nº 15 3. N-12(6) - 4017 SINGT OR Q= -383 N Q= 12 - 40 172 COS617 11.4 GIVEN: X= \$t3 - 2t2-30t+B X- 1t. t-5 FIND: t, X, AND Q WHEN NO X= 3t3- 2t2-30t+B HAVE _ N= dx = st2. St-30 THEN a = du = 10t-5 CUA WHEN N=0: St2-St-30=S(t2-1-6)=0 OR 1=35 AND 1=-25 (REJECT) : 1:35 AT t=35: X= 3(3)= 2(3)=2(3)=30(3)+B OR X=-59.5 ft OR 0; 25 12 a; 10(3)-5 11.5 GIVEN: X=6t - 2t3-12t2+3t+3 x~m t~5 FIND: t, x, AND IS WHEN Q = 0 HAVE .. X= 624-223-1222-32+3 U= dx = 24t - 6t2 - 24t + 3 THEN a= 32 = 72t -12t - 24 AND WHEN a=0: 7222-12t-24=12(622-t-2)-0 OR (31-2)(2++1)=0 OR t= \$5 AND t=-25 (REJECT) 1. t=0.6675 AT t= 35: X=6(3) -2(5) -12(3)2+5(3)2+5(3)+3 OR X=0.259m N= 24(3) -6(3)-24(3)-3 OR N=-8.56

11.6 GIVEN: X=3t3-6t2-12t+5 x-m, t-s FIND: (a) t WHEN 5=0 (b) X, Q, TOTAL DISTANCES TRAVELED WHEN L= 45 HAVE .. X = 3t3-6t2-12t+5 ル=報=9だ-12ド-12 MA 0= 3= 18F-15 (a) WHEN 15=0: 9t2-12t-12 = 3(3t2-4t-4)=0 OR (31+2)(1-2)=0 OR 1=25 AND 1=-35 (REJECT) :: 1=25 (b) AT L=45: X=3(4)3-6(4)2-12(4) +5 OR X=53 m a=18(4)-18 or a=60 52 FIRST OBSERVE THAT .. OSE CZS: NOCO 4>52. R>D Now. AT L=0. X= 5 m f= 23: X= 3(5)3-6(5)2-15(5)+2=-19m THEN 1x2-x01=1-19-51=24 m 14- X2 = 53-(-14) = 72 M

11.7 GIVEN: $X = t^3 - 9t^2 + 24t - 8$ $X \sim 1N$, $t \sim 9$ FIND: (a) t WHEN N = 0(b) X AND TOTAL DISTANCE TRAVELED

WHEN $\alpha = 0$

.. TOTAL DISTANCE TRAVELED = (Z4+72) m = 96 m

HAVE.. X=t3-9t2+24t-B THEN N= dx = 3t2-18t+24 AND a= dx = 6t-18

(a) WHEN N=0: $3t^2-18t^424=3(t^2-6t+8)=0$ or (t-2)(t-4)=0or t=2 and t=4

(b) WHEN Q=0: 6t-18=0 OR t=35
AT t=35: X=(3)=9(3)=24(3)-8 OR X=10 IN.
FIRST OBSERVE THAT.. 0 \(\) 0 \(\) 25 \(\) \(\

Now.. At t=0: $X_0=B$ in. At t=25: $X_2=(2)^3-9(2)^2+24(2)-B=12$ in.



THEN X2-X0 = 12-(-8) = 20 IN.

| X3-X2|= | 10-12| = 2 IN.

TOTAL DISTANCE TRAVELED = (20+2) IN. = 22 IN.

11.8 11.10 GIVEN: X=+3-6+2-36+-40 GIVEN: Qat; AT t=0, N=16" b; AT t=15, N=15" N. 15, X=20 IN. x-12, t-5 FIND: (a) t WHEN NEO (b) N, Q, AND TOTAL DISTANCE FIND : N. X. AND TOTAL DISTANCE TRAVELED TRAVELED WHEN X=0 AT L=75 X= t3- 6t2-36t-40 HAVE __ HAVE . a=kt k- CONSTANT N= dx - 3t2-12t-36 THEN Now of a a = kt At too, well #: | due lo ktat a= d= Lt-12 and OR 15-16 = 2kt OR 15= 16+ 2 kt2 (15) (a) WHEN N=0: 3t2-12t-36=3(t2-4t-12)=0 AT fols, No153: 1550 1650 2k(15)2 OR (+2)(+-6)=0 Yrn of = n = 10-fs or F = -5 20 ms no 10-fs OR t=-25 (REJECT) AND t=65 : 2.65 (b) WHEN X=0: +3-6+2-34-40=0 AT t=15, X= 20 m. : | 20 dx = 1 (16-t2)dt FACTORNG.. (t-10)(t+2)(t+2) =0 OR t=105 OR $X-20 = [16t-\frac{1}{3}t^3]^{\frac{1}{4}}$ OR $X=-\frac{1}{3}t^3+16t^{-\frac{1}{3}}$ (114.) NOW OBSERVE THAT .. O: 1 -65: N.O 654 t 5105: N>0 AND AT 1=0: X=-40 ft THEN .. AT 1-75: M=16-1712 OR N=-33 % X7 = -3(7)3+16(7)+3 OR X7=2.00 IN t=65: X= (6)3-6(6)2-34(6)-40=-256 ft WHEN N=0: 16-13 0 OR 2=45 £=105: 1,0=3(10)2-12(10)-36 OR 1,5144 5 OR 0,548 5 C t=45: x =- 3(4) = 16(4) + 3 = 47 in. NOW OBSERVE THAT .. DS & LAS: NO >0 -4011 0 X -256 \$\$ 4542575: 1540 (0) (105) (65) THEN 1x_-x_1=1-256-(-40)1= 216 ft (75) 3 IN. X10-X6 = D-(-256) = 256 St .. TOTAL DISTANCE TRAVELED = (216+256) 12 = 472 ft THEN X4-X3 = 47- 3 = 42.67 IN. 11.9 GIVEN: Q = 6 5t2; AT t=0, X=-32 St;
AT t=25, N=-6 t5 1x7 - x41 = 12-471 = 45 IN. : TOTAL DISTANCE TRAVELED = (4217+45)IN. = BT.7 IN. FIND: N, X, AND TOTAL DISTANCE TRAVELED 11.11 AT Lass GIVEN: Q = A- 62; AT t=0, X=8 m, N=0; AT 1=15, N=30 5 HAVE .. 32 - a . 6 52 FIND: (a) t WHEN Nº 0 AT 2=25, N=-6 5: [du= 1, 6dt (b) TOTAL DISTANCE TRAVELED WHEN OR N-(-6) = 6(t-2) t=55 OR N= 62-18 (3) ALSO. de = N = 61-18 HAVE _ a= A- 6t2 A - CONSTANT AT t=0, X=-32 ft: 1,2 dx = 6(6t-18)dt gr = a . A - 6 /s Now NT t=0, N=01 1, du = 1 (A-62) dt OR X-(-32) = 3t2-18t OR X=322-18t-32 (ft) OR N = At - Zt3 (2) ht t=15, N=30 3: 30 = A(1)-2(1)3 AT 2=55: NS=6(5)-18 OR NS=12 15 X=3(5)2-18(5)-32 OR X=-47 9t OR A = 32 % AND N= 32t-2t3 A1.50 dx = N = 32t-2t3 WHEN N=0: 61-18=0 OR t=35 AT t=0, x=8m: | x dx = 12 (321-213)dt At t=35: X3=3(3)-18(3)-32 =-59 ft NOW OBSERVE THAT DS & 435: N < 0 OR X = 8+16+2-2+4 (m) (a) WHEN N=0: 321-213 = 21(16-12)=0 354 £ 55: N>0 OR t=0 AND t=45 (b) AT t=4 5: X4=8+16(4)2- = (4)4 = 136 m -59te -47te -32te t=55: X5: 8+16(5)2-2(5)4:95.5 m NOW OBSERVE THAT OCELAS: 500 THEN 1x3-X01=1-59-(-32)1 = 27 ft 45 6 2 55: 50 1x5-x31=1-47-(-591)=12 st .. TOTAL DISTANCE TRAVELED = (27+12) ft = 39 ft X4-X0= 136-B = 128 m THEN 1x5-x41=195.5-1361-40.5 m .. TOTAL DISTANCE TRAVELED : (128+40.5)m = 168.5 m

11.12 GIVEN: aat ; AT &= D, X= 24 m; AT 2.65, X=96 m, N=18 m/s FIND: X(t) AND IT(t) HAVE .. a=kt2 R~ CONSTANT gr = 0 = kts Now AT t=65, N=18 = : 18 do = 1 kt2dt

OR N-18 = 3k(12-216) OR N=18+3k(t3 216) (m) #= N= 18+ 3k(t3-216) ALSO AT t=0, X= 24 m: | X dx = | t[18+3k(t2-216)]dt OR X-24 = 18t+3k(4t4-216t) Now.. AT t= 65, x=96 m: 96-24 = 18(6)+3 k[4(6) -24(6)] OR RE & W THEN. X-24 = 18t+3(4)(4t+-216+) OR XIL) = 100 + 101 + 24 NO N=18+3(4)(2-216) OR IS(t) = 27t + 10 11.13 GIVEN: FOR 25= t= 105, QQ to; AT t=25 5=-15 Mb; AT 1=10 5, N=0.36 Mb; 1X21=21X101 FIND: (a) X AT 1=25 AND AT 1=105 (b) TOTAL DISTANCE TRAVELED FROM 1:25 TO 1:105 HAVE .. Q. Es R ~ CONSTANT Now # = a = } At 7=52 N=-12 2: [292 = [5 fody (a) Have $\frac{dx}{dx} = \sqrt{1 - \frac{d^2}{2}}$ Then $|dx| = \sqrt{1 - \frac{d^2}{2}}$ Now X2 = 5 X10 : 24 2 4 (= 2 (10+ 10+C) AND X= t + + + 1.2 (m) 1. AT 1=25: X2=2+ 2+1.2 OR X2=35.2 m 1=105: X10=10+10+1.2 OR X10=17.6m NOTE: A SECOND SOLUTION EXISTS FOR THE CASE X2 >0, X10 <0. FOR THIS CASE C=-22 is m (b) WHEN 5:0: 1- \$2:0 OR to 85 AT 1=85: X8=8+8+1.2=17.2 m NOW OBSERVE THAT 25= £<85: 15<0 8544(105: 1500 35.2m X 17.2m 17.6m (85) (105) THEN 1X8-X21=117.2-35.21=18 m X10- XB = 17.6-17.2 = 0.4m . TOTAL DISTANCE TRAVELED: (18+0.4)m: 18.4 m

NOTE: THE TOTAL DISTANCE TRAVELED IS THE SAME

FOR BOTH CASES.

11.14 GIVEN: Q = - 8 "/5"; AT t = 45, X=20 m; WHEN N= 16 Ws. X= 4 m FIND: (a) t WHEN NO (b) N' AND TOTAL DISTANCE TRAVELED AT t= 11 5 off = a = - B 32 HAVE THEN Idw = J-Bdt + C C- CONSTANT OR 15:-8++ ((%) ALSO 2 = N = - BE+C AT t= 45, X=20 m: 120 dx = 14 (-8+4)dt OR X-ZO = [-4t3+Ct] 4 OR X=-4t2+C(t-4)+84 WHEN J=16 5, X=4 m: 16=-8++ C = C=16+8+ 4=-42+012-4)+84 COMBINING .. 0 = - 422+ (16+8+)(+-4)+80 SIMPLIFYING .. t2-4++4=0 OR 2 = 25 AND C = 32 3 N= -81+32 (E) x=-4t2+32t-44 (m) (a) WHEN N=0: -8t +32=0 OR 1:45 (b) AT t=0: X = - 44 m t=45: X4 = 20 m 1=115: X11=-4(11)2+32(11)-44=-176 m NOW OBSERVE THAT 05 LC45: 150 4562 \$115: NGO O SOM X (115) x4-x0=20-(-44)=64m 1x11- X41= 1-176-201= 196m ". TOTAL DISTANCE TRAVELED = (64+196) m = 260 m 11.15 GIVEN: Q= K(100-X), R - CONSTRUT; N=0 AT X= 40 mm, X= 160 mm; WHEN X=100 mm, N= 18 mm/s FIND: (a) R (b) N WHEN X=120 MM (a) HAVE NOX = a = k(100-x) WHEN X=40 mm, 5=0: 10 Ndor = 10 R(100-x) dx OR 250 = K(100x-2x2-3200) MHEN X:100 MM, N: 18 MM: 2(18)2-k[100(100)-2(100)-3200] OR k=00932 (P) MHEN X=150 mm: \$102=0.00[1001150]-\$(150) 3500] = 144 OR 15==16.97 5

11.16 GIVEN: Q = K/(X+4)2, K- CONSTANT; WHEN X=0, N=0; WHEN X=BM. 5=4 m/s FIND: (a) k (b) X WHEN No 4.5 M/s (C) ISMAX N dx = a = (x+4)2 (a) HAVE MHEN X50 200: Paga: Px (X+4)5 gx OR 2 50 - k (x+4 - 4) WHEN X=8m, N-4 = : 1(4)2=-k(104-1)3 11.19 OR R= 40 52 (b) WHEN N= 4.5 %: 2(4.5)2 = -48 (x+4 - 4) OR X=21.6 m (C) NOTE THAT WHEN U: DMAR, Q:O. NOW .. TAHT OC 0+0 AS X+00 (4)8+ (4+x + +) copy 8+ 2 2011 5 Nmax . 4.90 5 (a) HAVE 11.17 GIVEN: Q= 6x-14, Q- 4452 x- 82; WHEN X= O. N= 481/s FIND: (a) XMAX (b) IJ WHEN TOTAL DISTANCE TRAVELED = 1 St N # 0 = 6x-14 HAVE WHEN X=0, 5=4\$: [" 15 ds = [(6x-14) dx 11.20 OR [252] 4 = [3x2-14x] OR 2N2 = 3x2-14x+8 (a) FIRST DETERMINE WHERE N=0. 3x2-14x+8 = (3x-2)(x-4) =0 17 A = 3 26 AND X=4 ft NOW OBSERVE THAT AS THE PARTICLE PASSES HAVE THROUGH X=0, 500 AMB QCD AND THAT AT X= 3ft, 5=0 AND Q<0. THUS, THE PARTICLE WILL NEVER REACH X= 4 St AND, THEREFORE, XMAX = 0.667 ft (b) THE PARTICLE WILL HAVE TRAVELED A TOTAL DISTANCE OF I ST WHEN IT PHISES THROUGH X = 3 ft FOR THE SECOND TIME AND IS MOVING TO THE LEFT. THEN ... AT X= 3 ft: 202=3(3)2-14(3)+8=3 5=2.71 ft -11.18 GIVEN: Q=k(X-X), & AND A ARE CONSTANTS; AT t=0, X=1ft, W=0; WHEN X=16 ft, 15=29 ft/5; [(42 5 = x) 2/5 = (13 8 = x) 2 FIND: A AND R HAVE JAX = a = k(x-X) WHEN X=1ft, J=0: [5 JdJ =] k(x-X) X OR 202 = k[2x2-NLNX], = k(2x2- A LNX - 2) AT X= 2 ft: 2 U2 = K(2 (2)2-ALNZ-2)= K(2-ALNZ)

(BUT Y = 8/2 = 8/5/8) = 12/2 (B) = 12/2 = 13/2 = 118 = X

(CONTINUED)

11.18 CONTINUED NOW. 32 = 2: \frac{1}{2}Not = (2) = \frac{k(315-ALNB)}{k(\frac{3}{2}-ALNB)} 6-4ALN2=31.5-ALN8 OR 25.5 = A(LNB-4LNZ) = A(LNB-LNZ) = ALN(2) OR A = - 36.8 fl2 WHEN X=16 ft, N=29 5: 2(29)2= k[2(16)2-25.5) LN(16)-2] NOTING THAT LUCIS) = 4 LNZ AUD LNIZT=-LNIZT HINE .. BAI . K[256 - ZIZS. 5 4 WIZ) - 1] OR K=1.832 32 GIVEN: Q= k(1-1), k - CONSTANT; WHEN X = - 2 M, IT . 6 %; WHEN X=0, 5=0 FIND: (a) k (b) N WHEN X=-1 M Note = a = k(1-ex) WHEN X = - 2 m, No 6 3: | Indu = [R (1 - ex) dx CR 2(12-36) = 12(x+2) = 2)+18 WHEN X=0, 5=0: 0= k(1+2-2)+18 OR R= 4.1011 52 R.4.10 32 (b) WHEN X==1M: 20 = 4.1011(-1+2'+2-2)+18 OR N. 2.43 3 GIVEN: Q =- (0.1+ DIN 6), Q- 1/2, X-m; b= 0.8 m; WHEN X=0, 5=1 m/s FIND: (a) N WHEN X=-1 M (b) X WHERE J= NMX (C) Space 12 2 = 0 = - (01+ SINGB) WHEN K=0, N= 1 3: ["rdu . [- (0.1+ SIN 08) dx OR \$ (152-1) = - [0.1x-0.8 cos \$ 8] E.C - 8.000 8.0 + 11.0 - 2013 (a) WHEN X=-1m: 25=-0.1(-1)+0.8cos 08 -0.3 OR 15 = 10,323 \$ (b) WHEN IS = JMAX, Q=0: -(0.1+5IN 0.8)=0 OR X=-0.080 134 m X=-0.0801m ◀ (C) WHEN X=-0.080 134 m: \[\frac{1}{2} \sqrt{172} = -0.1 (-0.080 134 m) + 0.8 cos \frac{-0.080 134}{0.8} - 0.3 OR Nome: 1.004 5

11.21 GIVEN: a = 0.8 15 +49 , a - 1/2 5 - 1/2 WHEN X=0, 5=0 FIND: (a) X WHEN WEZY MIS (b) IT WHEN X= 40 m HAVE NOX = a = 0.8 / 52+49 WHEN X=0, 5=0: 5 / 544 = 50.8dx OR [15+49] = 0.8x OR 15+49-7 . D.8x (a) WHEN N= 24 3: 124 + 49 -7 = 0.8x (b) WHEN X=40 m: 150+49-7=0.8(40) 11.22 GIVEN: Q=-KINT, K-CONSTANT; AT 1=0. X=0, 5=81 M/s; WHEN X=18 m, 15= 36 m/s FIND: (a) IJ WHEN X=20 M (b) t WHEN 5 = 0 (a) HAVE N' dix = a =- k Tis WHEN X=0, N=BI =: | "Todu = | - kdx OR 3 [N 1/2] = - KX WHEN X=18 m, 5=36 = : \$(362-729) =- kx

WHEN X=18 m, 5=36 = : \$(362-729) =- k(18)

OR k=19 (356)

FINALLY.. WHEN X=20 m: \$(536-729) =- 19(20) OR 13 1 59

(b) Have de = 0 - 19 10 1

AT t=0, 10-81 7: 181 4 5 - 19 dt 50 WHEN 15=0: 2(-9)=-19t 11.23

2((5) = -19t \$ 101 = (10 - 27) 5

ce t=0.9475 €

(CONTINUED)

15= 29.3 m

OR X = 22.5 m

N= 38.4 =

GIVEN: Q = - RU . R - CONSTANT; AT too. X=0, N=16 in 15; WHEN X=6 IN. N= 4 14.15 FIND: (a) I WHEN X=5 IN. (b) t WHEN 5=9 IN/S

(a) HAVE IS dis a = - RIS WHEN X=0, 5=16 5: 1 5-kdx OR - 2 [N =] " = - EX OR 2(赤-4)= kx WHEN X=6 W, 15=4" 15: 2(14 - 4) = R(6) FINALLY .. WHEN X=5 IN .: 2(\$ -4)= 12(5) N= 4.76 T OR = 24

11.23 CONTINUED

(b) HAVE dt = 0 = - 12 N 2.5
AT teo, 5.6 (5): | " = 2.5 ds = | 1 - 12 dt OR - 3[15-3/2] 15 = - 12 t OR 3 (15/2 - 64) = 12 WHEN 12=9 3: 3(93/2 - 14)= 15 OR 1=0.17135

11.24 GIVEN: Q=-5/(25-15) Q-16/5, 15-16/5; AT t.D. X=0, W=N5; AT t=25. 5 = 0.5 Jo FIND: (a) Us (b) t WHEN 520 (C) X WHEN WEI 98/5

dr = a = - 2 No-N (a) HAVE AT 2=0, N= 45: 10 (245-47) du = 10-5 dt OR (2N5-N)2-N2 - 12 ((2N5-N)2) N .- St AT 1:25, 5=0503: (25-055)2-52=10(2) OR \$ 50 = 20 5 = 4 3t (P) HAVE (8-2)3-16=101 (C) HAVE $u \frac{du}{dx} = a = -\frac{s}{2u_5 \cdot u} = \frac{s}{8 \cdot u}$ WHEN X=0, U=15=45: |"16-17du = 15-5dx OR $[4u^2 - \frac{1}{3}u^3]_4^0 = -5x$ OR $(4u^2 - \frac{1}{3}u^3) - [4(4)^2 - \frac{1}{3}(4)^3] = -5x$ OR $(4u^2 - \frac{1}{3}u^3) - \frac{128}{3} = -5x$ WHEN J=1 & [4(1)2-3(1)3]-138=-SX JR X=780 St

11.25 GIVEN: Q= 0.4(1-RU) , R-CONSTANT; AT t=0, X=4 m, N=0; AT t=155, 15 = 4 m/s FIND: La) k (b) X WHEN W= 6 M/S (C) JMHX

(a) HAVE of a . 0.4 (1- RIS) AT t=0, 5=0: 10 du = 10.4dt OR - 1 [LN (1- KV)] = 0.4 E OR LN(1- RV) = -0.4 kt (1) AT t= 155, J=4": LN(1-4k) = -0.4k(15) SOLVING YIELDS R = 0.145703 m OR R=0.1457 m (b) HAVE U du = Q = 0.4(1- KU) NOW - 1- Ex = - 1 + 1- Ex THEN [[- k + K(1-EN)] du = [0.4 dx

11.25 CONTINUED

OR - [= LN(1-KN)] = 0.4(x-4) WHEN N= 6 3:

- [- (0.145 703 + (0.145 703) LN(1-0.145 703 = 6)] = 0.4 (x-4)

OR 0.4 (x-4)= 56.4778

OR X= 145.2 m

(C) THE MAUMUM VELOCITY OCCURS WHEN Q=0. 1. Q=0: 0.4(1-komm)=0

OR 15 MAX = 0.145703

OR Now = 6.86 5

AN ALTERNATIVE SOLUTION IS TO BEGIN WITH EQ.(1). W(1-ks) = -0.4kt

N= 12 (1-0.4Kt)

THUS, UMAL IS ATTAINED AS \$ - 00 ... NMAX = K .. AS ABOVE

11.26 GIVEN: a = - 0.65th a- 7/5: 5- 1/5; AT t=0, x=0, 15=9 m/s

FIND: (a) X WHEN No. 4 M/s

(b) + WHEN N= 1 m/s (C) t WHEN X=6 m

(a) HAVE 20 = 0 = 0 = 1 = 1 = 0 = 0 dx

OR 2[Nt] = -0.6X

WHEN N=4 %: X= 0.3 (3-N'E) (1)

(b) HAVE &= 0=-0.65% OR X= 3.33m \

OR -2[4"] = -0.6t

WHEN 5=1 \$: \frac{1}{17} - \frac{1}{3} = 0.3t

OR t= 2.22 5 .

(C) HAVE 10-3=0.3t = 9 OR 15= (1+0.9t)2 = (1+0.9t)2

Now .. dx = 5 = (1-09E)

AT t=0, x=0: 1 dx . 1 1 19912 dt

OR X= 9[- 49 1109E]

= 10 (1- 1109+)

WHEN X=6 m: 6 = 9+

OR t=1,6675

AN ALTERNATIVE SOLUTION IS TO BEGIN WITH EQ. (1)

X= = 3 (3-15'12)

THEN dx = (3-0.3x)2 Now... AT t=0, x=0: [3 dx dx =] dt

OR t = 03 (3-0.3x) = 3-0.9x

WHICH LEADS TO THE SAME EQUATION AS ABOVE.

11.27



GIVEN: N-7.5(1-0.04x) 5- 1/h x- mi : AT 1=0, x=0 FIND: (a) X AT tolh

(b) a (44/52) AT t.0 (C) t WHEN X = 6 mi

(a) HAVE & . 5 = 7.5(1-0.04x) 03
AT t=0, x = 0: \(\frac{1}{(1-0.04x)^{5/3}} = \frac{1}{5} \frac{1}{1.5} \dt

or $\frac{1}{\sqrt{3}}(-\frac{1}{\sqrt{3}})[(1-\sqrt{3})^{3/3}]_{0}^{3/3} \cdot 7.5t$ or $1-(1-\sqrt{3})^{3/3}=0.21t$ or $1-(1-\sqrt{3})^{3/3}=0.21t$

AT L=1 h: X=00+[1-[1-0.21(1)] 10.7]

OR X= 215 mi

(P) HAVE 0= 2 da (b) HAVE Q = 1/3/X = 7.5(1-0.04x)0.3 dx [7.5(1-0.04x)0.3] = 7.5 (1-0.04x)0.3 [(0.3)(-0.04)(1-0.04x)0.3] = -0.675 (1-0.04x)-0.4 AT t=0, x=0: Q = -0.615 mt x = 1 mt e (36005)2

OR 00= -275-10 15

(c) From EQ (1). t = 021 [1-(1-0.04x)0.7] WHEN X=6 mi: t= 5.21 {1-[1-0.04(6)]3.7}

= 0.832 29 h

DE 1 - 49.9 MM

11.28



GIVEN: N= DIBUE N-5, X-M 50=3.6 m/s FIND: (a) a WHEN X=2 m (b) TIME FOR AIR TO

FLOW FROM XEIM TO X = 3 m

a = 12 dx dx (0.184) (a) HAVE

WHEN X=2 m: Q = - 0.0324 (3.6)2

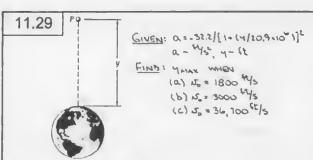
(P) HAVE of " " 0.185.

OR Q=-0.0525 7

FROM X=1 m TO X=3m: 1, Xdx = 1,0,185, dt OR (1x2) = 0.18 No (t3-t1)

OR (t3-t1) = 2(9-1) D.18(36)

OR 13-1, = 6.175



HAVE
$$N \frac{dN}{dt} = \alpha = -\frac{32.2}{(1 + \frac{N}{20.9 \times 10^6})^2}$$

WHEN $Y = 0$, $N = N = 0$

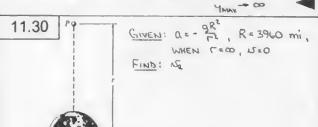
THEN... $\int_{N_0}^{\infty} N dx = \int_{N_0}^{\infty} \frac{1}{(1 + \frac{N}{20.9 \times 10^6})^2} dy$

OR $-\frac{1}{2}N_0^2 = -32.2 \left[-20.9 \times 10^6 \right]_{1 + \frac{N}{20.9 \times 10^6}}^{1}$

OR $M_{MAX} = \frac{N^2}{64.4 - \frac{N^2}{20.9 \times 10^6}}$

OR $M_{MAX} = \frac{N^2}{64.4 - \frac{N^2}{20.9 \times 10^6}}$
 $M_{MAX} = \frac{(1800)^2}{(41.4 - \frac{11800)^2}{20.9 \times 10^6}}$

OR 4MAX = -3.0340 FT THE VELOCITY 36,700 \$ 15 APPROXIMATELY THE ESCAPE VELOCITY IS FROM THE EARTH. FOR IS



HAVE
$$\sqrt{dv} = Q = \frac{2R^2}{\Gamma^2}$$

WHEN $\Gamma = R$, $15 \circ N_0$
 $\Gamma = \omega$, $15 \circ N_0$

THEN $\int_{-\infty}^{\infty} v dv = \int_{R^2}^{\infty} \frac{2R^2}{\Gamma^2} d\Gamma$

(CONTINNED)

11.30 CONTINUED OR - 2 Ne = 98 [+] NE = 129R = (2=32.2 5 = 3960mi = 5280 11 1/2

OR NE = 36.7 NO 5

11.31 GIVEN: N= 55[1- SIN(T)]; AT t=0, X=0, N- No FIND: (a) X AND a AT L= 3T (b) NAVE DURING tOD TO LOT

(Q) HAVE dx = N = N = [1- SIN(T)]
AT t=0, x=0: \[\delta \dx = \langle n = [1- SIN(T)] dt OR X = No[t + T cos(T)] = 5/t+ Tas(Tt)- T] (1) AT L=3T: X3T = No[ST+ Troo(Trost)-T] = No (3T - 2T)

ALSO. a = dt = dt { 5 [1- sin(= 1)]} x = 2.365T AT (=3T: 037 = -45 T cos Th

(b) USING EQ. (1) .. AT t. 0: X = 15[0+ Tasio] - 7]=0 At to T: $X_1 = U_0 \left[T \wedge \frac{T}{\pi} \cos \left(\frac{T^T}{T} \right) - \frac{T}{\pi} \right]$ = $U_0 \left(T - \frac{T}{\pi} \right)$ = 0.363 NoT

Now .. NAVE = XT - XO 0.36355T-0

11.32 GIVEN: N=N' SIN(UNT+ P); AT too, X=Xo.

OR JAME = 0,363No

SHOW: (a) N' = (No2+X2W2)/2X0WN (b) WMAX OCCURS WHEN X=X2[3-(50|X20) 12]/2

(a) At 1:0, 5=50: 50 - 5' sin (0+6) - 5' sin \$ THEN COSO = 1512-151/15' Now of 0 = 5 = 5 = 10 (wit + 4)
AT t=0, X=15: | X dx = | 5 5 5 = 10 (wht + 4) dt OR X-20 = 12 [- 1, cos(w, +4)]

X= x0+ 2 [cosp - cos(w,t+0)]

NOW OBSERVE THAT XMAX OCCURS WHEN cos(wnt+\$) = -1. THEN ...

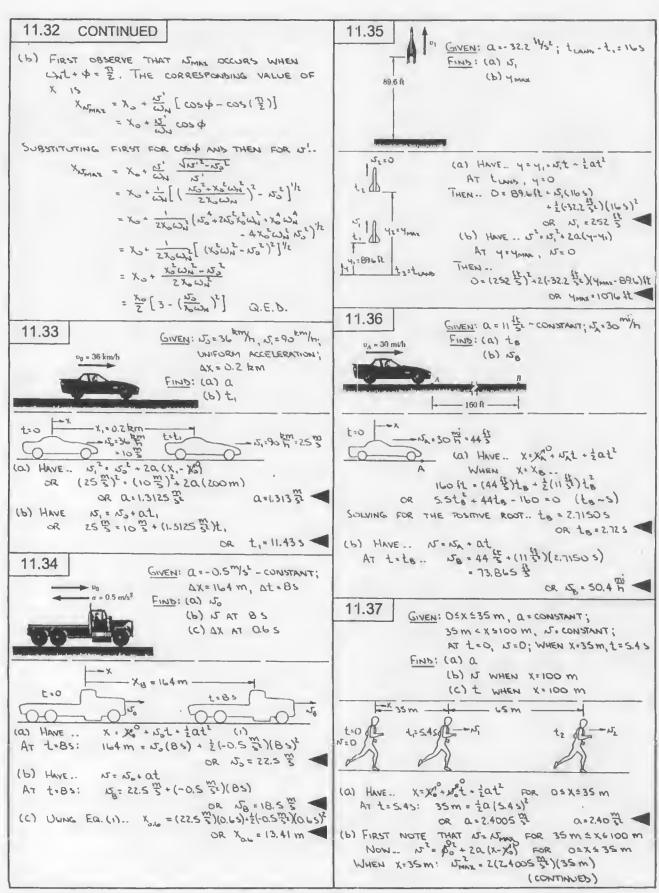
Y-ma = 2x0 = x0 + 15 [cosp - (-1)]

SUBSTITUTING FOR COS O ... X = W/ (VUTT-UDE +1) OR YOUN - 15' = V15' - 152

SAUNTUNG BUTH SIDES OF THIS EMATION...

X2 W2 - 2X0WN + N12 = 112 - 112

OR N1 = 152 + X3 W4 Q.E. Q.E.D. (CONTINUES)



11.37 CONTINUED

OR JMAX = 12.9628 5 JMAX = 12.96 5 (C) HAVE.. X = X, + N, (t - t,) FOR 35 m < X = 100 m

WHEN X=100 m: 100 m = 35 m + (12.9628 5 Xt; 5.4) 5

OR t2 = 10.41 5

11.38

GIVEN: QAB = QCD = 4.8 \$\frac{1}{3}\$; \$\int_{Bc} = CONSTANT\$; \$\frac{1}{2} \times \frac{1}{3} \times

(a) FOR A -B AND (-D HAVE N'= N5+2a(X-X) THEN .. AT B .. NEX = NA - 2(4.8 5-)(3-0)m = 28.8 mg (No. - 5.3666 5) AT D. Sp = Not + 20co (x0 - xc) d= x0 - xc OR (7.2 m) = (28.8 m) + 2(48 m) d AND AT D .. OR d=2.40 m (b) FOR A -B AND (-D) HAVE .. No No+at THEN .. A-B .. 5.3666 = 0+ (4.8 52) the OR LAB = 1.118 04 5 7.2 3 · 5.3666 3 + (4.8 32) /co AND C D .. OR tcb = 0.381965 Now. FOR B-C HAVE Xc = XB = Noctoc 3m = (5.3666 3) tec OR toc : 0.559015 FINALLY, to the + Les (11804 + 055901+038196)5

11.39

GIVEN: AT t=0, XM = Xp =0; AT t= 425, XM=Xp;

NM = CONSTANT; FOR 0 = 1 = 185, ND=0;

FOR 18 s < t s 265, Ap = CONSTANT;

AT t = 265, ND=90 = 5;

FOR 265 < t s 425, ND=90 = 90 = 6

FIND: (a) Xp AT t = 425

(b) NDM

OR 10 = 2.065

- (XP)42 (Np)42 (P O (P) 26 50 t=185 1=265 1=425 (150) 26 = 30 th 25 th (150) 15 = 30 th = 55 th (Np)10=0 (a) PATROL CAR: FOR 185<25265: 15p=(15p/8+0p(t-18) AT 1=265: 25 3 = ap(26-18)5 OR ap= 3.125 72 ALSO, Xp = (XP/18 + (15/18) (1-18) + 20p (1-18)2 AT t=265: (XP)26 = 2(3.125 52)(26-18)2 = 100 m FOR 2654 E \$ 425: Xp = (Xp)26 + (Np)26 (1-26) AT 1:425: (Xp)42 = 100 m + (25 3)(42-26)5 (X=)42=0.5km (b) FOR THE MOTORIST'S CAR. Xm = (XM) + Nmt AT 1=425, Xm=Xp: SOOM = Nm (425)

OR Ny = 11.9048 3

OR 15m = 42.9 m

(v_A)₀ = 129 m/s GIVEN: A

11.40

GIVEN: AT t=0, XA=XB=0;
AT t=1.825, XA=XE=20m,
NA=NB
FIND: (a) an AND as KNOWING

an and ab knowing that both are uniform (b) to when runner B starts to run

(a) FOR RUNNER A: XA = (XX) + (VA) t + \frac{1}{2} a a t^2

AT t = 1.825: 20 m = (12.9 \frac{12}{2})(1.825) + \frac{1}{2} a a (1.825)^2

OR Qa = - 2 10 \frac{12}{2}2

ALSO... 15x = (15x) + Qxt

AT t=1.825: (15x)1.82 = (12.9 =) + (-2.10 =)(1825)

= 9.078 =

FOR RUNNER B: Ng2 = (NG) + 208 (x8-(X6)) WHEN X8=20 M, N8 = Nx : (9.078 =) 12 = 208 (20 M) OR OB = 2.06 03 = 1

(b) FOR RUNNER B: No = (No) + ag(t-to)

WHERE to IS THE TIME AT WHICH HE BEGINS
TO RUN.

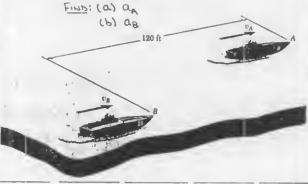
OR to - 2.59 5

.. RUNNER B SHOULD START TO RUN 2.59 S BEFORE A REACHES THE EXCHANGE BONE.

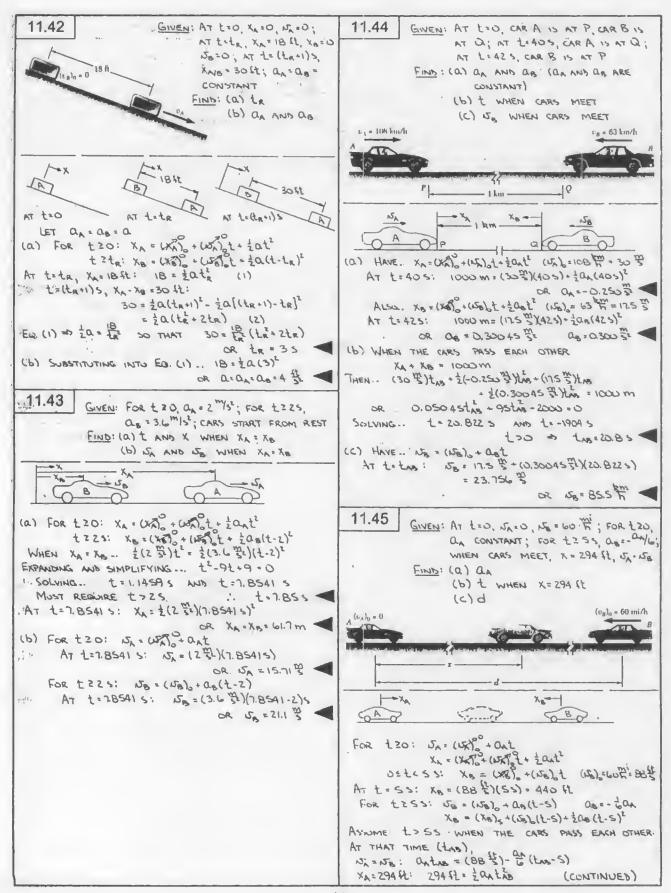
11.41 GIVEN: AT t=0, XNB = 120 ft, NA =NR = 105 H,

a, ab = constants; AT t=85, XA=XB,

NA = 135 H



(a) Have... $N_{R} = (N_{R})_{0} + \Omega_{R}t$ $(N_{R})_{0} = 105 \frac{M^{2}}{5} = 154 \frac{4}{5}$ At t = 85: $N_{R} = 135 \frac{M^{2}}{5} = 198 \frac{45}{5}$ Then $198 \frac{45}{5} = 154 \frac{45}{5} + \Omega_{R}(85)$ OR $\Omega_{R} = 5.50 \frac{4}{5}t^{2}$ (b) Have.. $N_{R} = (N_{R})_{0} + (N_{R})_{0}t + \frac{1}{2}\Omega_{R}t^{2}$ $(N_{R})_{0} = 120 \frac{4}{5}t^{2}$ And $N_{R} = (N_{R})_{0} + (N_{R})_{0}t + \frac{1}{2}\Omega_{R}t^{2}$ $(N_{R})_{0} = 154 \frac{45}{5}t^{2}$ At t = 85: $N_{R} = N_{R}$... $120 \frac{4}{5} + (154 \frac{45}{5})(85) + \frac{1}{2}(5.50 \frac{45}{5})(85)^{2} = (154 \frac{45}{5})(85)^{2}$ OR $\Omega_{R} = 9.25 \frac{45}{5}$



11.45 CONTINUED

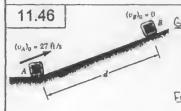
an (3+10-2) 88 THEN ¿antas 44 th - 343 th + 245 = 0 OR

SOLVING .. LAB = 0.795 5 AND \$:7.005 (a) WITH tas> 55, 294 ft = 2ax(7.005)

08 Qx . 1200 32

LAB = 7.00 5 (b) FROM ABOVE NOTE: AN ACCEPTABLE SOLUTION CANNOT BE FOUND IF IT IS ASSUMED THAT LAB & SS.

(C) HAVE.. d = X + (X =) + (88 =) (7.00 = 5) s + = (- 12.00 12 × 17.00-5)25 OR d=906 \$1



GIVEN: (U) AND (NE); AT tols BLOCKS PASS EACH DIHER. AT 1: 245, Xa=d; (XA) MAX = 21 51; an, as ARE CONSTANT AND FIND: (a) an AND as (b) d

(C) NA AT L=15 (a) HAVE . J = (U) + 20, (x - UX)

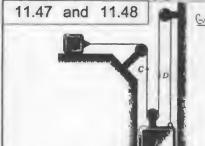
WHEN XA = (XA)MAX, UZ=0 THEN. O=(27 8) +20A(21 ft) OR 04 =- 17.3571 \$\$

NOW .. Xx = (Xx) = (15) t = {axt AND X5 = USE + WEET = 20 st2 AT 1=1 5, THE BLOCKS PAST EACH OTHER :. (XA), + (Xb), = d AT 1=3.45, XB = d THUS .. (XA), + (XB), = (XB)24

OR [(2) \$)(1) + {(-17.35) 1 (x (1)) } - [{200(1)} = 200(3.45)2

(b) AT t=3.45, X3 d: d= 2(3.4700 \$)(3.45) OR d=20.1 ft -

(C) HAVE .. NA = (NA) + QAT AT 1=15: NA = 27 5 + (-17.3571 5)(15) OR NA . 9.64 \$



GIVEN: BLOCKS A AND B AND THE PULLEY! CABLE SYSTEM MWCHZ

FROM THE DIAGRAM (NEXT COLUMN) HAVE ..

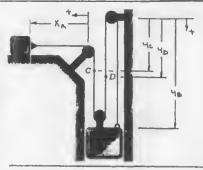
X+348 = CONSTANT THEN 54 35 = 0 (1)

NND

ax + 3ax = 0 (2)

(CONTINUED)

11.47 and 11.48 CONTINUED



GIVEN: UZ . 6 5 -11.47 FIND: (a) NB

(P) No (د) یحوال

(a) SUBSTITUTING INTO EQ. (1) .. 6 5 + 35 = 0 OR NB=5 3

(b) FROM THE DIAGRAM. YE + YO . CONSTANT THEN NB+ ND = 0

.. is= 2 31 (C) FROM THE DIAGRAM .. X4+ 46 = CONSTANT 15x+15c=0 : 15e=-6 1 THEN

NOW .. JEB = NE-NO = (-12)-(52)--83

: 120 - 8 m

11.48 GIVEN: AT t=0, No =0; QB = CONSTANT 1; WHEN I DXAI = 0.4 m, Wal= 4 3

FIND: (a) QA AND QB

25 = TA [(e/A)-64) DNA BU (d)

(a) EQ. (2): QA + 3QB = O AND QB IS CONSTANT AND POSITIVE => QA IS CONSTANT AND NEGATIVE

ALSO, EQ.(1) AND (NE)=0 => (NA)=0

THEN NA = (NA)= + 20A(XA-(XA)=)

WHEN 10XA = 0.4 m: (4 x) = 20A(0.4 m)

OR Q4 - 20 32 -THEID.. SUBSTITUTING INTO EQ (2).. - 20 52+300=0

OR QB = 3 52

(b) HAVE.. UB = (NB) + QB t

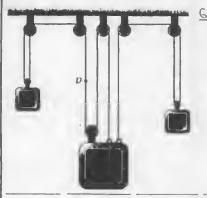
AT t = 25: UB = (3 52)(25) Q8=6.475

OR NE = 13.33 % ALSO - 48 = (48) + (158) + 200 t2

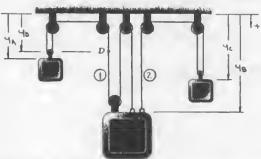
AT t=25: 40-(40)0= 2 (30 m)(25)2

OR 48- (48) = 13.33 m/

11.49 and 11.50



GIVEN: BLOCKS A, B, AND C AND THE PULLEY / CABLE SYSTEM SHOWN



FROM THE DIAGRAM ..

CABLE 1: 244+348 = CONSTANT

THEN - 2NA+3NB =0 (1)

(E) O= JUSHBU - NEHT 204+3000 (2)

AND 03+20(=0 (4)

CABLE 2: 48+24C = CONSTANT

GIVEN: No = 24 31 11.49

FIND: (a) U

(b) 5c

(C) No

(d) JNB

(a) Substituting into Ea. (1). 25/4 + 3(24 5)=0 OR NA: 36 51 4

(b) SUBSTITUTING INTO EQ. (3). (24 5)+245 0 DR UE = 12 1N. 1

(C) FROM THE WAGRAM .. 24A + 46 = CONSTANT

THEN .. 25 + No = 0 SUBSTITUTING FOR UT .. 2(-36 5)+N5=0

OR 15 = 72 31

(d) HAVE -. NOID = NO-NO = 72 12 - 24 12

OR WHS=4851

11.50 GIVEN: (US) =0; Q = CONSTANT 1; AT 1=125, 5 = 18 IN./S

FIND: (a) QA, QB, AND QC

(b) NB AND [40-(48)] AT t=85

(a) Eas. (3) AND (1) AND (UC) =0 => (UA) = (UB) =0 ALSO, EQS. (4) AND (2) AND QC IS CONSTANT AND POSITIVE => Q IS CONSTANT AND NEGATIVE QA IS CONSTANT AND POSITIVE

THEN .. IS = (ISK) + ant 18 = ax (125) OR Qx=1531 AT L=125:

(CONTINUED)

11.50 CONTINUED

SURSTITUTING INTO EQ. (2). 2(1.5 12)+300 =0 OR Q = 1.0 = 1

SUBSTITUTING INTO Ed. (4) .. (-1.0 32) + 20 = 0

OR ac = 0.5 551

(b) HAVE .. No = (158) + ast AT t. BS: NB = (-1.0 18:) (BS)

OR 58.80 11 -

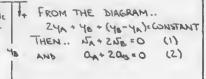
ALSO. 40 = (40) + 650 + 200t AT t.85: 40-(40) = 2(-10 1/2)(85)2

OR 40-140) = 32.0 IN. 1

11.51 and 11.52



GIVEN: COLLARS A AND B AND THE PULLEY CABLE SYSTEM SHOWN



11.51 GIVEN: WAY = 0; QA = CONSTANT ; AT L=85, 158/A1 = 24 14/5

FIND: (a) QA AND QB

(P) To AND AB- (AU) " VI T: P?

(a) Ea. (1) AND (NA)=0 => (NE)0 ALSO, EQ. (2) AND QA IS CONSTANT AND NECATIVE

EN. No = WAT + CAL No POSITIVE

THEN .. NA = (NATO + OAL NOW - UBIA = UB - UA = (as-as)t

FROM Ea.(2). as=- Las so THAT ISUR - ZOAT AT L= BS: 24 3 = - 20x (BS)

OR Q4 = 2 31 AND THEN -- ap =- 2 (-2 52) OR Q == 1 51 (b) AT t= 65: NB = (1 5) (65)

OR 15 = 6 1 Now .. 40 = (40) + (18) + + 2aot2

AT t=65: 48-(48) = 2(1 52)(65)2 OR 40-(40) = 18 IN.

(CONTINUED)

11.52 CONTINUED GIVEN: No - 12 7 11.52 FIND: (a) JA (6) 2 (C) 548 (a) SUBSTITUTING INTO EQ. (1). NA+2(12 11/2)=0 OR N. 2431 (b) FROM THE DIAGRAM .. 24x+ 4c = CONSTANT THEN .. ZUA + NC = U SUBSTITUTING .. 2 (-24 14)) + 150 = 0 OR 5 = 48 51 (C) HAVE .. LEIB - WC-WB = (48 3)-(12 3) OR LEB- 36 51 11.53 and 11.54 GIVEN: BLOCKS A AND B AND THE PULLEY/CABLE SYSTEM SHOWN Ď -+ XD-XA-FROM THE DIAGRAM .. XB+ (XB-XA)- 2XA = CONSTANT THEN .. 255 - 354 = 0 (1) 208-30A=0 (2) AND GIVEN: NB = 300 mm/s-11.53 FIND: (a) 5 (P) Nº (c) 50 (d) 50/A (a) SUBSTITUTING INTO EQ. (1). 2(300 5)-354=0 OR JA = 200 5 (b) From THE DIAGRAM. XB+(XB-Xc) = CONSTANT THEN - 250 - 50 = 0 SUBSTITUTING -. 2(300 5)-4=0 OR 15=600 5 (C) FROM THE DIAGRAM .. (XC-XA)+ (XB-XA)= CONSTANT THEN - NE - 2NA + NO = 0 SUBSTITUTING .. 600 "5" - 2(200 "5") + 45 = 0 DR NS = 200 mm

(d) HAVE -. NEL - NE - NA

* ALSO HAVE .. - X6 - X4 = CONSTANT

THEN .. UD . UA = 0 (3)

= 600 mm - 200 5

OR 54 = 400 5-

(CONTINUED)

11.54 CONTINUED GIVEN: (US) = 150 5; as = CONSTANT; WHEN 11.54 3- (xx) = 240 mm - , 15 = 60 5 FIND: (a) QA AND QB (b) as (C) 50 AND X8-(X8) AT 1:45 ID) FIRST OBSERVE THAT IF BLOCK A MOVES TO THE RIGHT, IN AND EQ. (1) => IS -. THEN, USING 2(150 mm) - 3(NA) = 0 EQ. (1) AT 1:0 .. OR (N) : 100 5 ALSO, EQ. (2) AND QB = CONSTANT = QA = CONSTANT THEN .. No = (No) = 20 = (Xa - (Xa)) = 20 = (Xa) = OR Q4 = - 40 mm OR Qx=13.33 50-THEN SUBSTITUTING INTO EQ. (2)__ 208-3(-40 mm)=0 as=20.0 52 OR a= - 20 52 (b) FROM THE SOLUTION TO PROBLEM 11.53 ... ND+ NA = 0 THEN .. as + an = 0 SUBSTITUTING .. ap + (- \$ 50 mm) = 0 OR ab = 13.33 50 -(c) HAVE .. No * (No) + ast AT 1=45: No 150 5 + (-200 5) (45) OR UB = 70.0 5 ALSO - 48 = (40) + (5) = + 200t2 AT = 45: 48-(48) = (150 5)(45) + 2(-200 52)(45)2 OR 48- (40)= 440 mm -11.55 and 11.56 GIVEN: BLOCKS A, B, AND C AND THE PULLEY/CABLE SYSTEM SHOWN 44 FROM THE DIAGRAM .. 34+ 448 + X = CONSTANT

(CONTINUED)

THEN -- 35x + 458 + NC=0 (1)

30x+400+0c=0 (2)

11.55 and 11.56 CONTINUED

GIVEN: NE = 20 31; (NE) - 30 51; 11.55 QA = CONSTANT; AT 1-35, Xc - (Xc) = 57 mm --

FIND: (a) (NE)

(P) ON AND OF

(C) 44- (4A) AT t=55

(a) SUBSTITUTING INTO EQ. (1) AT t=0.. 3 (-30 5) + 4 (20 5) + (UE) = 0

OR (55) = 10 mm

(b) HAVE .. Xc = (X) + (NE) 2 + Eact AT t=35: 57 mm = (10 5)(35) + 20c(35)2

OR acologi-

NOW .. IS = CONSTANT => Q8 = 0 THEN .. SUBSTITUTING INTO Ed. (2) .. 30,+4(0)+(6 0)00

OR Q1= 2 32

(C) HAVE - YA = (YA) - + (UT)) + { QAt AT t= 55: 4-(4A)0 = (-30 5)(55)+2(2 5)(55)2 OR 44-4410=175 mm!

GIVEN: (NB) = 0, Qx = CONSTANT, (Qc) = 75 mm; AT t= 25, 11.56 NE = 480 mm 1 NE = 280 mm -FIND: (a) QA AND QB

(P) (DA) AND (DE). (C) xc-(xc) at t=35

(a) Eq.(Z) AND Q = CONSTANT AND ac = constant = ab = constant THEN - US = (150) + ast

480 mm = aB(25) AT t= 2 5:

OR Q8 = 240 52

SUBSTITUTING INTO EQ. (2)... 30x+4(240 52)+(75 52)=0

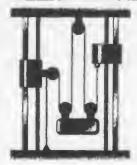
OR Q4 = 345 52 1 (b) HAVE .. JE = (NE) = Oct

AT 1=25: 280 5 = (2) + (75 5)(25)

THEN, SUBSTITUTING INTO EQ. (1) AT t=0 .. 3(UA) + 4(0) + (130 mm) = 0 OR (NA) = 43.3 5

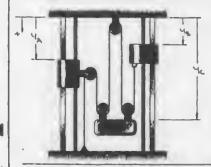
(C) HAVE -- Xc = (Xc) + (15) t + 2 act2 AT t= 3 5: X- (Xe) = (130 mm)(35)+2(15 mm)(35) OR Xc-(xc)o=728 mm-

11.57 and 11.58



GIVEN: COLLARS A AND B. BLOCK C. AND THE PULLEY/CABLE SYSTEM SHOWN

11.57 and 11.58 CONTINUED



FROM THE DIAGRAM .. -4A + (4c-4A)+24c + (4c-4B)= CONSTANT

(1) G-314+ B21- AUS-AND

-2a, -a, +4a, =0 (2)

GIVEN: (NT) =0, (Qx)=7 14: (NE)=8 51 11.57 QB = CONSTANTI; AT 2:25, 48-148 = 20 IN.

FIND: (a) QB AND QL

(b) t WHEN NE = 0 (C) 41-(76)0 WHEN NEO

(a) HAVE .. Yo = (40) + (No) t + 2001 AT t=25: -20 IN. = (-8 1)(25) + 206(25) OR ag= 2 50 1

THEN .. SUBSTITUTING INTO EQ. (2).

-2(7 52)-(-2 182)+4Qc=0

DR QC = 3 321

(b) SUBSTITUTING INTO EQ (1) AT 1.0 .. -2(0)-(-8 1/5)+4(1/2)00 OR (1/2)=-2 5

Now .. WE = (UE) + act WHEN NEO: 0= (-2 13)+ (3 5)) OR 1 = 35

teaulis (C) HAVE - 4 = 1400 + (UE) 2+ 20022 AT { = 35: 4c-(4c) = (-2 5)(35) + 2(36)(25)

OR -16-1761 = 0.667 IN.

11.58 GIVEN: (UA)0=0, (UE) =0; QA = 3t 51; QB = CONSTANT! WHEN YO - (YE) =3ZIN! N= 8 3

FIND: (a) ac

(b) DISTANCE TRAVELED BY CAT 1.35

OR (NE) = 130 5 (a) HAVE .. No. = (NE) + 200 (40- (40)) WHEN 48-(48)= 32 IN.: (8 5)2 = 200 (32 IN.)

> THEN, SUBSTITUTING INTO EQ.(2). -2(-3世岁)-(1岁)+40=0 OR ac= 4(1-6t2) 1/4

(b) SUBSTITUTING INTO EQ. (1) AT t=0 .. -2(0)-(0)+4(NE)0=0 OR (NE)0=0

Now. de ac = \$ (1-62) AT t=0, 5=0: | bedie = 1 \$ \$ (1-62)dt DR No = \$ (t-2+3)

THUS, JE = 0 AT \$2(1-22)=0 OR t=0, t= 125 THEIREFORE, BLOCK (INITIALLY MOVES DOWNWARDS (5,00) AND THEN MOVES UPWARD (VCCO). Now -- Etc = 4(1-213)

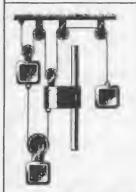
AT t=0, 40 (40): 140 dye = 10 \$ (t-2t) dt

02 40-(40) = 8(t2-t4)

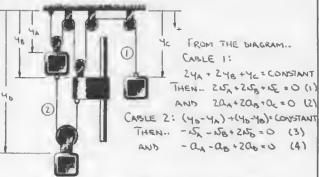
AT t= (\$ >: Ye-(4c)= \$[(1/2)2-(1/2)4]= 1/2 IN. AT t=35: 4c-(4c) = \$[(3) - (3)4] = - 9 IN. ". TOTAL DISTANCE TRAVELED = (32)+1-9-321=91614.

= 9.06 IN.

11.59 and * 11.60



GIVEN: BLOCKS A. C. AND D. COLLAR B. AND THE PULLEY CABLE SYSTEM SHOWN



11.59 GIVEN: AT 1=0, N=0; ALL ACCEVERATIONS CONSTANT; Qc/s = 60 mm/,
Qc/s = 110 mm/,
FIND: (a) WE AT L= 35

(b) 40- (40) AT t=55

(a) HAVE - ac/8 = ac-as = -60 or as = ac+60 ao/4 = ao-ax = 110 or ax = ao-110 SUBSTITUTING INTO EQS (2) AND (4) ..

EQ (2): 2(00-110)+2(0c+60)+0c=0

OR 300+200 = 100 (5) EQ. (4): - (ap-110)- (ac+60)+2ab=0

OR - ac + as = - 50

SOLVING EQS. (5) AND (6) FOR QC AND QD ... ac = 40 mm ap = -10 mm

Now .. No = (Not) act AT t= 35: 15 . (40 52)(35)

(b) HAVE.. 40 = (40) + (45) 01 + 200 12 | A7 1 = 55: AT t=55: 40-(40) = 2(-10 = 1)(55)2

OR 40 - (40) = 125 mm

11.60 GIVEN: AT L=0, 5=0, (4A)= (46)= (46)=; ALL ACCELERATIONS CONSTANT; AT t=25 YCIN = 280 mm !; WHEN 158/ = 80 51 4/4-(4/4)= 160 mm1, 48-(40)= 320 mm1; as > 10 5

FIND: (a) QA AND QB

(a) HAVE - 4 = (4) + (4) + 24 + 24 +

4c = (4c) + (+50) t+ 2ac t2 (CONTINUED)

11.60 CONTINUED

THEN -- Yen = ye - YA = 2 (ac-an) 2 AT t= 25, 44x=-280 mm: -280 mm = 2 (ac-an)(25)2 OR ac = an - 140 (5) SUBSTITUTING INTO EQ. (2) .. 204 +208 + (04-140) = 0

OR an = \$(140-200) (6) Now. No - (Not + ast NA = (WKTO - ant

: Noin = No - No = (as - an) t ALSO - 45 = (46) + Word + 2 ast WHEN SON = 80 mm : 80 = (as - an) t

44.0 160 mm/: 160 = 20222 440 = 320 mml: 320 = 2 ast

THEN 100 = 1 (a6-an) t2 Umic Ea. 17) .. 320 = (80) t OR 1 = 45 160 = 2 an (4)2 DR a= 20 521 THEN 320 = 2 ag (4)2 OR Q8 : 40 37 NOTE THAT EQ. (6) IS NOT USED; THUS, THE PROBLEM IS OVER DETERMINES.

ALTERNATIVE SOLUTION: HAVE., 52 : (1/2) + 202 [42-(42)] 50 : (45) + 208 [40-(40)] THEN. SAY = No- 1/2 = 1200[40-(40)] - 120/ [40-(40)] WHEN NO 80 5 1: 80 5 = 12[108(320 mm) - 102 (160 mm)]

DR 20 = 12 (1200g - 1100g)

SOLVING ENS (6) NUS (8) YIELDS ON AND QUE. (b) SUBSTITUTING INTO EQ. (5).

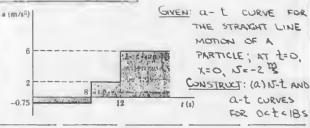
ac = 20- 140 = - 120 3 AND INTO EQ. (4) .. - (20 52) - (40 52) + 200 = 0 OR an = 30 50

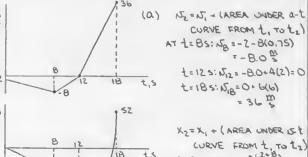
Now .. we . (wet a - act Miner of 2-600 5: -600 000 - (-1200 000)) [

on toss ALSO - 40 = (40)0 + Wolft + 2 ast2 AT 1:55: 40-(46) = 2 (30 57) (55)2

OR 40-(40)=375 mm

11.61 and 11.62





LS AT 1:85: Xp=0-8(208) = - 40 m (CONTINUED)

11.61 and 11.62 CONTINUED AT t= 125: X12=-40- 2(4)(8)=-56m 11.63 t=185: X18=-56 + 2(6)(36) = 52m 11.61 FIND: (b) X, N, AND TOTAL DISTANCE TRAVELED AT 1=185 (b) READING FROM THE CURVES. XIB = 52m NB= 36 3 -FROM t=0 TO t=125: DISTANCE TRAVELED = 56m t=125 TO t=185: DISTANCE TRAVELED= 52-(-56) ". TOTAL DISTANCE TRAVELED = (S6+108)m = 164 m 11.62 FIND: (b) Nomin (C) Xmin (b) READING FROM THE N-L CURVE .. ISMIN = - 85 (C) READING FROM THE X-2 CURVE. XMIN =- 56m 11.63 and 11.64 GIVEN: IS' CURVE FOR THE YTRAIGHT LINE MOTION OF A PARTICLE; AT too, X=- 540 St p (f(A) CONSTRUCT: (a) Q-t AND X-t 60 CURVES FOR octusos

(a) Q_{ξ}^{2} SLOPE OF N-t CURVE AT TIME t FROM t=0 to t=105: N=CONSTANT => Q=0 t=105 to t=265: $Q_{\xi}^{2} = \frac{-20-60}{26-10} = -5\frac{11}{5}$ t=265 to t=415: N=CONSTANT => Q=0 t=415 to t=465: $Q_{\xi}^{2} = \frac{-5-(-20)}{46-41} = 3\frac{11}{52}$ t>465: N=CONSTANT => Q=0

 $X_{2} = X_{1} + (AREA UNBER 15-t CORVE FROM t, To t_{2})$ AT t = 105: $X_{10} = -540 + 10(60) = 60$ ft

NEXT FIND TIME AT WHICH N=0. USING SIMILAR

TRIANGLES... $t_{100} = \frac{2k-10}{60}$ OR $t_{100} = 225$ AT t = 225: $X_{22} = 60 + \frac{1}{2}(12)(60) = 420$ ft t = 245: $X_{11} = 380 - 15(20) = 380$ ft t = 465: $X_{41} = 380 - 15(20) = 80$ ft t = 465: $X_{40} = 80 - 5(\frac{2045}{2}) = 17.5$ ft t = 505: $X_{50} = 17.5 - 4(5) = -2.5$ ft X_{5} X_{5}

-540

11.63 and 11.64 CONTINUED

II.63 FIND: (b) TOTAL DISTANCE TRAVELED AT t=505

(c) t WHEN X=0

(b) FROM t=0 TO t=225: DISTANCE TRAVELED = 420-(-540)

= 960 H

t=225 TO t=505: DISTANCE TRAVELED = 1-2.5-4201

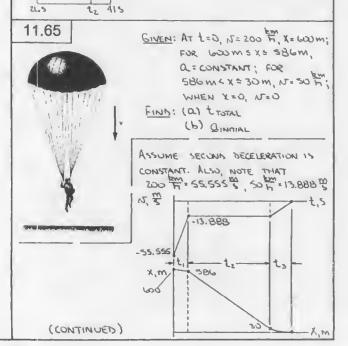
= 422.5 ft

.. TOTAL DISTANCE TRAVELED = (960 + 422.5) ft = 1382.5 ft = 1383 ft (C) Using similar triangles..

Between 0 and 10s: $(t_{x+3}) = 0$ = $\frac{10}{600}$ OR $(t_{x+3}) = 95$ Between 465 and 50s: $(t_{x+3}) = 4$

OR (1x=0)=49.55

11. GA FIND: (b) Xmax (c) t WHEN X = 100 St (b) READING FROM THE X-t WAVE .. Xmm = 420 ft (C) BETWEEN IDS AND ZZS ... 100 ft = 420 ft - (AREA UNITER LET CURVE FROM t, TO 225) HT OR 100=420-2(22-2,)(5) OR (22-t,)(5,) = 640 USING SIMILAR TRIANGLES ... = 60 OR 5, = 5(22-t.) 105 ty THEN _ (22-t,)[5(22-t,)]=640 OR 1,=10.695 AND 1,= 33.3 5 HAVE 105 42, 4225 => t,=10.69 5 BETWEEN 265 AND 415 USING SIMILAR TRIANGLES ... 41-t2 : 15 380 K th 001 H08 F-1or t2=405



(CONTINUED)

11.65 CONTINUED

(a) NOW. AX = AREA UNDER J-L CURVE FOR GIVEN TIME INTERVAL

THEN. (586-600) m = -t, (55.555+13 888) M OR t, = 0.4032 s (30-586)m=-t2(13.888 m) OR t2 = 40.0346 5 (0-30)m==2(ta)(13.888 m)

OR t3 = 4.3203 5 troTAL = (0.4032 + 40.0346 + 4.3203)5

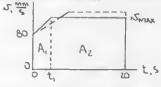
OR LTOTAL : 44.85 (b) HAVE .. QINITIAL = AVINITIAL [-13.888-455.555]] T 0.4032 5 = 103.3 m OR QINITIAL = 103.3 521

11.66

GIVEN: AT to 205, X=4m; N5=80 5, amax = 60 " town = 155,

SPAINT & CONSTANT FIND: (JMW) MIN

FIRST NOTE THAT (80 5)(205) < 4000 mm, so THAT THE SPEED OF THE PALLET MUST BE INCREASED. SINCE NEAINT = CONSTANT IT FOLLOWS



THAT NPAINT = NMAN AND THEN 1, = 5 5. FROM THE N-1 CURVE, A, + A2 = 4000 mm AND IT IS SEEN THAT (WMAX) MIN OCCURS WHEN and (= 15max - 80) is maxim.
(15max - 80) mm = 605 -80) IS MAXIMUM. THUS

1, = (5mm - 80)/60 AND 1, (80+15max) + (20-1,)(15maz)= 4000 SUBSTITUTING FOR t...

(15max - 80) (80+ AMAX) + (20 - 15max - 80) / 15max = 4000

SIMPLIFYING - 5 MAX - 25605 MAX + 486 400 = 0 FOR NAME = 207 THE AND NAME = 2353 TO NTMAX = 2353 MIN 1, 255

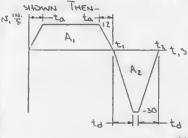
:. (15MAX) MIN = 207 MM



GIVEN: (Smax) ROM = 12 5 (UMAX) LEFT : 30 5; ariam = + 6 1N. alery = = 20 Si FIND: (a) tercLE CONSTRUCT (b) Not AND X-1 CURVES

11.67 CONTINUED

(a) AND (b) THE N-t CURVE IS FIRST DRAWN AS

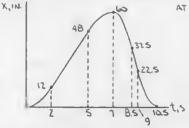


ta = arch = 12 5 = 25 ty: Our: 30 % Z 3 '\$' = 1.55

Now .. A, = 60 IN. OR [(t,-2)) (125)= WIN OR 1,= 75 A = 60 IN. AND

OR { ((t2-7)-1.5) \$ } (30 5) = 60 IN OR t2: 10.55

Now .. torcle = 22 : tereis=10.55 HAVE .. X ;; = X; + (AREA UNDER J-1 CURVE FROM L; TO L;;)



AT t= 2 5: X2 = 2(2)(12) = 12 14. t=55: X5=12+(5-2)(12) = 48 IN. 1=75: X7=60 IN. f=8.55: x =60-2(1.5)(30) = 37.5 IN. t=95: 10=37.5-(0.5)(30) = 22.5 IN. t=10.55: X10.5=0

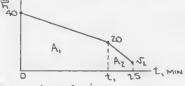
11.68



GIVEN: AT 1=0, J=40#, X=0; WHEN X=2.5 mi, IT = CO TO ; AT 1 = 7.5 MIN, X=3 mi; CONSTANT DECELERATIONS

FIND: (a) t WHEN X= 2.5 Mi (b) AT WHEN X:3 mi (C) afinal

THE IT- & CURVE IS FIRST DRAWN AS SHOWN. 1. Mi

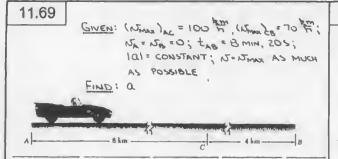


(a) HAVE .. AVE .. A = 25 mi OR (t, min) (+0+20) mix 1 h 60 min 2.5 mi OR ti - 5 MIN

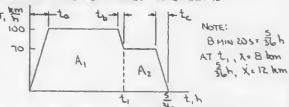
(b) HAVE. Az = 0.5 Mi 20+152 Mi 1 16 MIN : 0.5 Mi OR 52 = 4 TA

(C) HAVE OFINAL = Q12 = (4-20) h SZBUST 1 MIN 1 H (7.5-5) Mil mi

OR OFINAL - O. ISLA SE



THE J-T CURVE IS FIRST DRAWN AS SHOWN, WHERE THE MAGNITUDES OF THE SLOPES (ACCELERATIONS) OF THE THREE INCLINES LINES ARE EWAL.

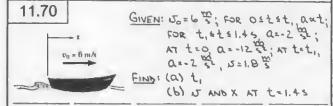


DENOTING THE MAGNITUSE OF THE ACCELERATIONS

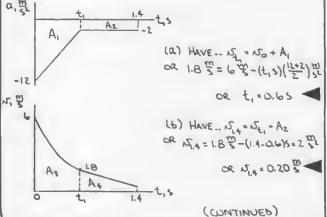
BY a 100 HAVE .. WHERE Q IS IN AWY THE TIMES ARE IN h. Now. A,=8km: (4,)(00)-= (ta)(100)-= (tb)(30)=8 SUBSTRUTING.. 100t, - 2 (100 /100) - 2 (30) (30) = 8 OR t, = 0.08+ 54.5

ALSO. Az=4 km: (36-t, 10)-2(tc)(70)=4 SUBSTITUTING .. (3t-1,)(10) - 2(t)(10) = 4

OR t, 103 - 3 OR a= SI 259 km 1000m 1 (36005)2 OR a=3,96 1



THE Q-t AND IS-T CURVES ARE FIRST DRAWN AS SHOWN.



11.70 CONTINUED

Now .. X 1.4 . A 5+ A4 , WHERE AS IS MOST EASILY DETERMINES USING INTEGRATING. THUS. Now .. of a a = 52 - 12 Ar t=0, 00 0 7: 1 3 du = 1 (50 t-12 bt OR N= 6 4 25 t - 12 t

HAVE.. $\frac{dx}{dt} = L = L - 12t + \frac{25}{5}t^{2}$ THEN.. $A_{3} = \int_{3}^{2} L dx = \int_{3}^{2} (b - 12t + \frac{25}{5}t^{2}) dt$ $= \int_{3}^{2} L dx = \int_{3}^{2} (b - 12t + \frac{25}{5}t^{2}) dt$ $= \int_{3}^{2} L dx = \int_{3}^{2} (b - 12t + \frac{25}{5}t^{2}) dt$ $= \int_{3}^{2} L dx = \int_{3}^{2} (b - 12t + \frac{25}{5}t^{2}) dt$ ALSO.. $A_{4} = (1.4 - 0.6)(\frac{100}{2} + 0.6)$ THEN ..

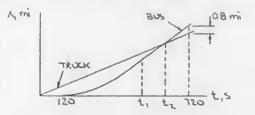
m(8.0 + 40.5) = 1.1X THEN .. N A8.5 = 1, 1 × × ×

11.71 GIVEN: NT = 45 Th; AT 1=0, XT =0, XB =0; · FOR O= 2 = 2 MIN, NO=0; FOR LOS MIN, QB = CONST UNTIL No = 60 € THEN QB=0; AT 2=12 MIN, XB- XT = 0.8 mi

FIND: (a) & AND X WHEN XB " XT (b) as

FIRST NOTE .. 45 Th . 66 \$ PO # 88 & (a) Assuming that the Bus reaches 60 th (at TIME LI) BEFORE IT PASSES THE TRUCK (AT TIME tz), THE WE'L AND X-T CURVES CAN THEN BE DRAWN AS SHOWN.





AT t=7205 (12 MIN): XB-XT = 0.8 mi or [\$(t.-120)5=(88\$)+(720-t.)5=(88\$)]-[720 \$(66\$)]

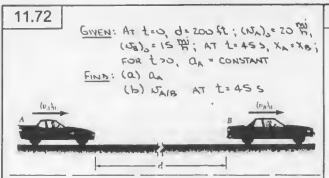
OR t, = 144 S

AT t=t2: XB = XT OR (144-120) > (88 \$)+ (12-144) > (88 \$) = (+2 5) (66 \$) OR t2 = 5285

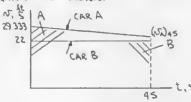
THEN to st, & ASSUMPTION CORRECT : t7 = 528 5 OR t2-8 MIN 485 AT 1= 12: XB = XT = (528 5)(66 \$) = 34 BAB St

(b) HAVE .. a= (50)4-0 - (144-120)5

OR 00 = 3.67 5



(a) FIRST NOTE.. 20 = 29.333 \$\frac{15}{2}\$ 15 \frac{15}{2}\$ 22 \frac{15}{2}\$
THE N-1 CURVES FOR THE TWO CARS ARE THEN DRAWN AS SHOWN.



AT t=45.5, $K_A = X_B$: (AREA), = (AREA), +200 ft OR $(45.5)(\frac{29.5333+15}{2})\frac{45}{5} = (45.5)(22.\frac{15}{5})+200$ ft OR $(N_A)_S = 23.555\frac{15}{5}$ THEN $Q_A = \frac{(N_A)_{45} - (N_A)_{0}}{t_{45}} = \frac{(23.555-29.333)\frac{15}{5}}{45.5}$

(b) HAVE -- SAIB = JA - JB = (23.555-22) 13.
= 1.555 35

OR SAIB = 1.060 TH

11.73 GIVEN: (15) = 56 km (15) = 27 km;

Q =-0.042 52; CAR A JUST AVOIDS

COLLIDING WITH CAR B

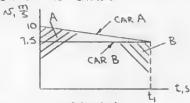


FIRST NOTE.. 36 \$\frac{Fm}{m}: 10 \frac{Fd}{d}\$ 27 \frac{Fm}{m}: 7.5 \frac{Fd}{d}\$

NOW ASSUME THAT UA: NB WHEN Xa: XB; THE

U-1 CURVES FOR THE TWO CARS ARE THEN

BRAWN AS SHOWN.



Now.. Qa = (Na) (-(Na)) or -0.042 m = (7.5-10) m/s

OR -0.042 m = (7.5-10) m/s

OR 1, 54, 524 S

AT 1=1, XA = XB: (AREA)A = (AREA)B + d

OR (59.524 5)(\frac{10+7.5}{2})\frac{m}{5} = (59.524 5)(7.5\frac{m}{5}) + d

OR d=74.4 m

12 m

11.74

GIVEN: AT t=0, SE=0; FOR

0 \$ NE = 1.8 %,

QE = 1.2 %, FOR

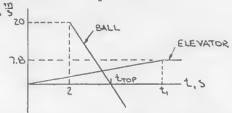
NE = 1.8 %, QE=0;

AT t= 2 5, NE = 20 %,

FIND: T WHEN THE BALL

HITS: THE ELEVATOR

THE N-2 CURVES OF THE BALL AND THE ELEVATOR ARE FIRST DRAWN AS SHOWN. NOTE THAT THE INITIAL SLOPE OF THE CURVE FOR THE ELEVATOR IS 1.2 32, WHILE THE SLOPE OF THE CURVE FOR THE BALL IS -9 (-9.81 32).



THE TIME to IS THE TIME WHEN UP REACHES

7.8 %. THUS... NE = (UP) + Qet

OR 7.8 % = (1.2 %) to or to 6.5 5

THE TIME LTOP IS THE TIME AT WHICH THE BALL REACHES THE TOP OF ITS TRAJECTORY.

THUS. Us = (Us) - 9(t-2)

OR 0 = 20 3 - (9.81 76)(trop - 2)5

OR trop = 4.0387 5

Using the coordinate system shown, have..

O $\leq t \leq t$, : $Y_E = -12m + (\frac{1}{2}Q_E t^2)m$ ($(\sqrt{2}B)_0$) At $t = t_{100}$: $Y_B = \frac{1}{2}(4.0387-2)s = (20\frac{\pi}{3})$ = 20.387 m

AND $Y_E = -12m + \frac{1}{2}(1.2\frac{\pi}{3})(4.0387s)^2$

AT t=[2+2(4.0387-2)]5:6.07745, 48:0

AND AT t:t, 4:-12m+2(6.55)(7.85)=13.35m

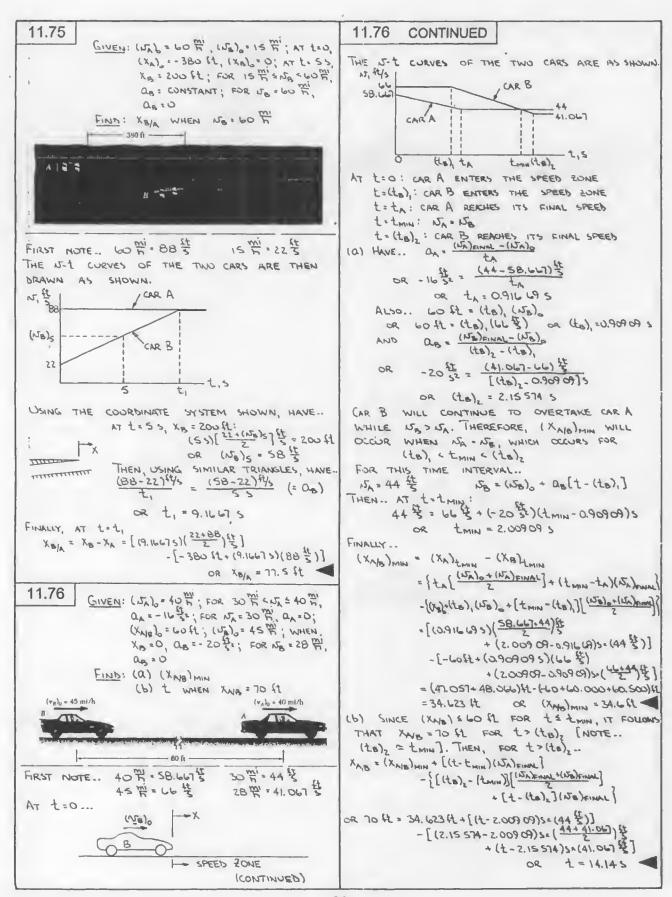
THE BALL HITS THE ELEVATOR (46:46) WHEN

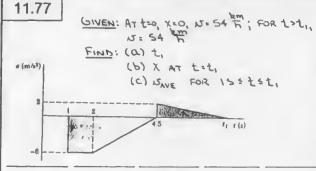
trop < t < t.

FOR { 2 trop : 48 = 20.387 m - [= 29(t-trop)] m
THEN. WHEN 48 = 4 ...

20.387 ×1 - ½(9.81 5/2)(t-4.0387)2 =-12 m+ ½(1.2 5/2)(t 5)2 08 5.50512-39.696t 647.619 =0

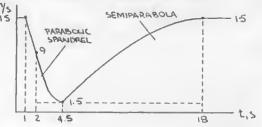
SOLVING. - t=1.525 & AND t=5.675 Now. trop < t < t, => t=5.675





FIRST NOTE .. S+ = 15 3 (a) HAVE. If . Ja + (AREA UNDER a-t curve FROM ta to tb) THEN .. AT L= 25: N= 15 - (1)(6)=9 5 t=4.55: 15=9- 2(2.5)(6) . 1.5 5 1=1: 15=1.5+2(1,-4.5)(2) OR t1=185

(b) USING THE ABOVE VALUES OF THE VELOCITIES THE N-L CURVE IS BRAWN AS SHOWN. N, M/S



Now .. X AT L = 18 S .. XIB . X + E CAREA UNDER THE U-1 CURVE = (15)(15 m)+(15)(15+9) to t=185) +[(2.55)(1.53)+ 3(2.55)(7.53)] +[(13.55)(1.5 3)+ 3(13.55)(13.5 3)] = [15 + 12 + (3.75 + 6.25) + (20.25 + 121.50)] m = 178,75 m OR X18 = 178.8m (C) FIRST NOTE - X, = 15 m X18 = 178.75 m NOW. NAVE = AX = (18-1)5 = 9.6324 mg

(18-1)5

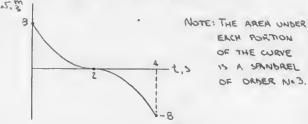
ON WAVE - 34.7 P 11.78 GIVEN: AT t=0, X=0, N=B& a (m/s²) j CONSTRUCT: (Q) N-t AND f (s) X-1 CURVES FOR 042445 FIND: (b) X AT 1=35 = -3(f - 2)2 m/s2

(a) HAVE. IS . IS + (AREA UNDER a-t curve from t, To tz) AND .. X2 = X, + (AREA UNDER U- t CURVE FROM t, TO tz) (CONTINUED)

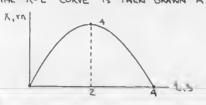
11.78 CONTINUED

THEN, USING THE FORMULA FOR THE AREA OF A PAREBOLIC SPANDREL HAVE ... 0 : (51)(5) = - 8 - 5 (2)(12) = O t= 45: N=0-3(2)(12)=-B3

THE J-t CURVE IS THEN BRAWN AS SHOWN.



Now .. AT t=25: X=0+ (2)(B) = 4 m t=45: X=4- (2)(8)=0 THE X-L CURVE IS THEN DRAWN AS SHOWN.

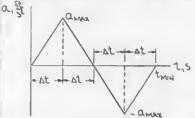


(b) HAVE .. AT t=35: Q=-3(3-2)2. -3 52 5=0-3(1)(3)=-1 = X= 4 - (1)(1)

OR X3 = 375 m

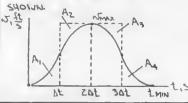
11.79 GIVEN: AT 1=0, X=0, J=0; Xmx=1.2 ft; WHEN X= XMAL, N=0, 1(3) MAK = 4.83 FIND: (a) Emin FOR XMAX = 1.2 St (b) JMAX AND JAVE FOR OSESEMIN

(a) DESERVING THAT I'MAX MUST OCCUR AT LEZTHIN, THE Q-T CURVE MUST HAVE THE SHAPE SHOWN. NOTE THAT THE MAGNITUDE OF THE SLOPE OF EACH PORTION OF THE CURVE IS 4.8 TUS/S.



HAVE .. AT 1= A1: N= 0+2(A1)(amax) = 2 amax st 1=201: Nmax = 2 amax 61 + 2 (61) (amax) = amou st

USING SYMMETRY, THE U-1 IS THEN DRAWN AS



(CONTINUED)

11.79 CONTINUED

NOTING THAT A, : A : A : A AND THAT THE AREA UNDER THE 15-2 CURVE IS EQUAL TO XMIN, HAVE ..

(2 DE) (UMAX) = XMAX

NMAX = amax at => 2 amax at = xmax
NOW ... at = 4.8 fus/s so that

12 S.1 = 540 (et 40 8.4) S

THEN tmin = 4 Dt OR tmin = 2.005 (b) HAVE .. Umax = amax at = (4.8 44.5 * 10t) at = 4.8 44.5 (0.5 5)2

NMX = 1.2 15 OR NAVE : O.6 \$

11.80

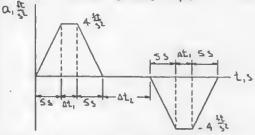
GIVEN: XMAX = 1.6 mi; 10 max 1 = 4 52, 1(2) max 1 = 0.8 84 %; NMAX = 20 K

FIND: (a) then FOR XMAX: 1.6 mi (b) NAVE

FIRST NOTE _ 20 T = 29.333 \$ 1.6 mi = 8448 ft (a) To OBTAIN thin, THE TRAIN MUST ACCELERATION AND DECELERATE AT THE MAXIMUM RATE TO MAXIMIZE THE TIME FOR WHICH IS I I MAX. THE TIME OF REQUIRED FOR THE TRAIN TO HAVE AN ACCELERATION OF 4 4752 IS FOUND (SE) MAX = CMAX DR At = 4 +432 OR at: 55

NOW. AFTER 5 5 THE SPEED OF THE da TRAIN 15.. $N_5 = \frac{1}{2}(\Omega t)(\Omega_{MRN})$ (SINCE $\frac{d\Omega}{dt}$ OR $N_5 = \frac{1}{2}(SS)(4\frac{15}{3}t) = 10\frac{55}{3}$ = CONSTA = CONSTANT)

THEN, SINCE US & DMM, THE TRAIN WILL CONTINUE TO ACCELERATE AT 4 4/52 UNTIL IS - WMAX. THE Q-t CURVE MUST THEN HAVE THE SHAPE SHOWN. NOTE THAT THE MAGNITUSE OF THE SLOPE OF EACH INCLINED PORTION OF THE CURVE IS 0.8 PUSY'S.



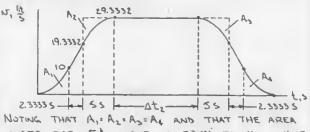
Now. AT t= (10+ at,) s, w= Nmax: : 2[2(5 s)(4 5) + (at,)(4 5) = 29.333 5

OR At, = 2.3333 5 THEN .. AT t= SS: N= 0+2(5)(4)=10 \$ t=7.33335: 15= 10+(2.3333)(4)=19.3332 5 t=12.3333 3: 5:19.3332+2(5)(4)=29.3332 }

USING SYMMETRY, THE U- & CURVE IS THEN DRAWN AS SHOWN.

(CONTINUES)

11.80 CONTINUED



UNDER THE Nº L CURVE IS EQUAL TO KMAX, HAVE .. 2 (2.33335) (10+19.3332) 127

+ (10+ At2)5 . (29.3332 15) = 844B ft

OR Atz = 275.67 5

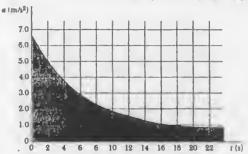
THEN .. & MIN = 4(55) + 2(2.33335) + 275.675

= 300.345 OR 2 MIN = 5.01 MIN (b) HAVE .. NAVE = At = 1.6 mi = 3600 5

> JAVE : 19.18 1 OR

11.81

GIVEN: Q-t CURVE; AT t=0, X=0, N=0 FIND: (a) IS AT tOBS BY APPROXIMATE MENS (b) X AT t=205 BY APPROXIMATE MEANS



SOLUTION PROCESURE

1. THE at CURVE IS FIRST APPROXIMATED WITH A SERIES OF RECTANGLES, EACH OF WISTH At . 25. THE AREA (At)(QAVE) OF EACH RECTANGLE IS PPPROXIMATELY EQUAL TO THE CHANGE IN VELOCITY AU FOR THE SPECIFIED INTERVAL OF TIME, THUS DN= DAVE ST

WHERE THE VALUES OF DAVE AND DIS ARE GIVEN IN COLUMNS I AND 2, RESPECTIVELY, OF THE FOLLOWING TABLE.

TAHT DUA O = 2 TAHT DUITON .S Sz = 15, + A512

WHERE DUTY IS THE CHANGE IN VELOCITY BETWEEN TIMES I, AND IL, THE VELOCITY AT THE END OF EACH 25 INTERVAL CAN BE COMPUTED; SEE COLUMN 3 OF THE TABLE AND THE 5-1 CURVE.

3. THE J-t CURVE IS NEXT APPROXIMATED WITH A SERVES OF RECTANGLES, EACH OF WINTH IT = 25. THE AREA (At) (NIME) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN POSITION AX FOR THE SPECIFIED INTERVAL OF TIME. DX = JAYE At

(CONTINUES)

11.81 CONTINUED

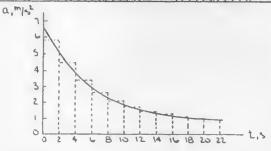
WHERE NAVE AND DX ARE GIVEN IN COLUMNS 4 AND S, RESPECTIVELY, OF THE TABLE.

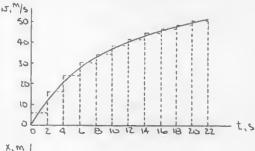
4. WITH NOTING THAT

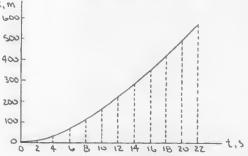
12 = X, + DX12

WHERE DXIZ IS THE CHANGE IN POSITION BETWEEN TIMES t, AND to THE POSITION AT THE END OF EACH 25 INTERVAL CAN BE COMPUTED; SEE COLUMN 6 OF THE THBLE AND THE X-L CURVE.

		- 1	4	3	4	2	6
2.5	a, mist	GAVE, M/3		5, 1/3	SAVE MYS	AK,m	X, W
0	6.63	77777	77777	0	7777	0////	0
2	5.08	5.86	11.75	11.72	5.86	11.72	11.72
		4.47	8.94		16.19	32.38	
4	386	3.38	6.76	20.66	24.04	48.08	44.10
6	2.90	2.58	5.16	27.42	30.00	L0.00	92.18
8	2.25			32.58			152.18
10	1.87	2.06	412	36.70	3464	69.28	221.46
		1.71	3.42		38.41	76.82	
12	1.54	1.42	2.84	43.12	41.54	83.08	298.28
14	1.29	1.23	2.46	42.96	44.19	86.38	381.36
16	1.16			45.42			449.74
18	1.03	1.10	2.20	47.62	46.52	93.04	SL2.7B
		1.00	2.00		48.62	97.24	
20	0.97	0.94	1.88	49.62	50.56	101.12	20.02
22	0.90	7777	7777	51.50	11/1/1	77777	761.14

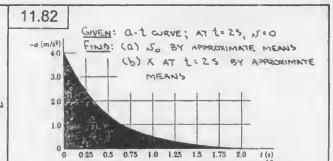






(a) AT 1=85, 5=32.58 5 OR 5=17.3 Th (b) AT 1=205.

X=660 m



SOLUTION PROCEDURE

1. THE Q. I CURVE IS FIRST APPROXIMATED WITH A SERIES OF RECTANGLES, EACH OF WISTH At = 0.25 S. THE AREA (At)(QAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN YELDCITY AN FOR THE SPECIFIED INTERVAL OF TIME. THUS IN BANK STA

WHERE THE VALUES OF DAVE AND AN ARE GIVEN IN COLUMN'S I AND 2, RESPECTIVELY, OF THE FOLLOWING TABLE.

2. Now .. N(2) = No+ 1 adt = 0 AND APPROXIMATING THE AREA GOLD UNDER THE a-1 CURVE BY EQUE OF . EDS. THE INITIAL VELOCITY IS THEN EQUAL TO No=- E AN

FINALLY, USING

N, = N, + DN,

WHERE DUTY IS THE CHANGE IN VELOCITY BETWEEN TIMES t, AND tz, THE VELOCITY AT THE END OF EACH 0.25 INTERVAL CAN BE COMPUTED; SEE COLUMN 3 OF THE THBLE AND THE N- L CURVE.

3. THE U-t CURVE IS THEN APPROXIMIATED WITH A SERVES OF RECTANGLES, EACH OF WIDTH 0.25 S. THE AREA (Ot)(NAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN POSITION AX FOR THE SPECIFIED INTERVAL OF AX = Nave At TIME. THUS ... WHERE JAVE AND DX ARE GIVEN IN COLUMNS

4 INS S, RESPECTIVELY, OF THE TABLE. 4. WITH X0 = O AND NOTING THAT

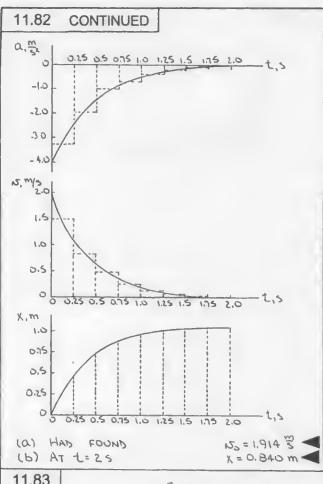
XZ = X1 + DX12

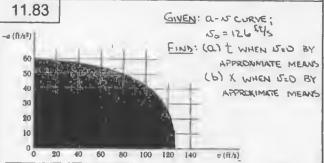
WHERE AXIZ IS THE CHANGE IN POSITION BETWEEN TIMES t, AND t2, THE POSITION AT THE END OF EACH 0,25 5 INTERVAL CAN BE COMPUTED: SEE COLUMN 6 OF THE THBLE AND THE X-2 CURVE.

		- 1	Z	_3	4	5	L.
t, s.	a, m/s2	DAVE, TYS	DU, m/s	5, m/s	SAVE, MY	DX.M	X, m
0	- 4.00	-3.215	77777	1.914	11111	7////	0
0.25	-2.43		-0.804	1.110	1,512	0.378	0.37B
0.50	-1.40	-1.915	-0.479	0.631	0.871	815.0	0.596
0.75	-0.85	-1.125	-0.281	0.350	0.491	0.123	0.719
1.00	-0.50	-0.675	-0.169	0.181		0.067	0.786
1.25	-0.28	-0.390	-0.0AB	0.083	0.132	0.033	0.819
1.50	- 0.13	-0.205	-0.051	0.032	0.058	0.015	0.834
1.75	- 0.06		-0.024	800.0	0.050	0.005	0.839
2.00	0	-0.030	-0.008	0	0.004	3.331	0.840

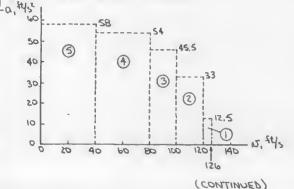
EAN = - 1.914 M/S

(CONTINUES)





THE GIVEN CURVE IS APPROXIMATED BY A SERIES OF UNIFORMLY ACCELERATED MOTIONS (THE HORIZONTAL BASHED LINES ON THE FIGURE).

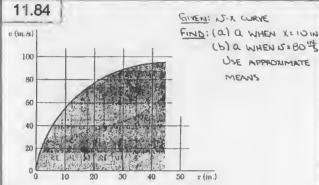


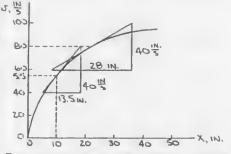
11.83 CONTINUED

FOR UNIFORMLY ACCELERATED MOTION.. $S_z^2 = S_1^2 + 2\Omega(X_2 - X_1)$ $S_z = S_1 + \Omega(t_2 - t_1)$ $S_z = S_1 + \Omega(t_2 - t_1)$ $S_z = S_1 + \Omega(t_2 - t_1)$ $S_z = S_1 + \Omega(t_2 - t_1)$

FOR THE FIVE REGIONS SHOWN ABOVE, HAVE .. a, 4432 | 0x, 42 | 0t, 5 REGION 15, 145 52, 145 126 120 -12.5 59.0 0.480 120 100 2, -33 66.7 0.606 3 100 08 -45.5 39.6 80 - 54 4 40 44.4 0.741 -58 5 13.8 0.690 223.5 2.957

(a) From the table, when N=0 t=2.965
(b) From the table and assuming X=0, when





FIRST NOTE THAT THE SLOPE OF THE ABOVE CURVE 15 3x. Now... du Q = N 3x

(a) WHEN X=10 IN., $D = SS \frac{1N}{S}$. THEN.. $Q = SS \frac{1N}{S} \cdot \left(\frac{40^{1N}/S}{13.5 \text{ NN.}} \right)$

(b) WHEN 5=80 10; HAVE

Q=80 5 (40111/5)

NOTE: TO USE THE METHOD OF MEASURING THE

SUBNORMAL OUTLINED AT THE END OF SECTION

11.8, IT IS NECESSARY THAT THE SAME SCALE

BE USED FOR THE X AND N AXES (0.9., 1 IN.: 50 IN,

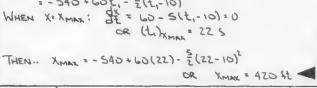
1 IN. = 50 IN./s). IN THE ABOVE SOLUTION, AN ANN

AX WERE MEASURED DIRECTLY, SO DIFFERENT

SCALES COULD BE USED.

11.87 11.85 CONTINUED GIVEN: MOMENT-AREA METHOD OF 12.8432 SECTION 11.8 60 ta-DERIVE: X= X + Not + Late FOR A THE AREA WOER THE CLAVE IS BIVISED INTO PARTICLE IN UNIFORMLY ACCELERATED RECTILINEAR THREE REGIONS AS SHOWN MOTION THE a-t wave for uniformly accelerated MOTION IS AS SHOWN. (a) FIRST NOTE .. Now .. N= No+ 1 adt WHERE THE INTEGRAL IS EWAL TO THE AREA UNDER THE Q-2 CURVE, THEN, WITH No = 25 5, No =0 HAYE.. 0 = 7.5 5 + [2(0.65)(605) - 2(0.155)(15 5) - (2,-0.75)5-(15 5)] USING EQ. (11.13), HAVE .. X = Xo + Not + (AREA UNDER a-t curve)(t-t) OR t = 2.375 5 ₹,=2.38 5 (b) FROM THE BISCUSSION FOLLOWING EQ. (11.13) AND = X = + (+ a)(+ - 2t) = X3 + N5t + 2ate 3VAH, C=CX DAIMCECA $(\bar{t}_-,\bar{t})A$ + CX+CX=XQ.E.D. 11.86 WHERE A IS THE AREA OF A REGION AND & IS GIVEN: a-t CURVE: No = - 23 FIND: X AT t=165 USING THE METHOD THE DISTANCE TO ITS CENTROIS. THEN FOR a (m/s2) [OF SECTION 11.86 1 = 2.375 5 ... X=(7.5 \$)(2.375 5)+ [[{(0.65)(60 })](2.375-0.2)5 e (01.0 - 288.5) [12 21/c 21.6) }]-142 24 12 6 541 -[(1.6255)(153)[2.375-(275-2-1.625)]5] =[17.8125+(39.1500-1.8844-19.8047)]{t OR X=35.3 ft -0.75 11.88 GIVEN: IS- & CURVE FOR THE STRAIGHT LINE 0, 1/5 MOTION OF A PARTICLE; AT 1=0, 6 x = - 540 ft THE AREA UNDER CONSTRUCT: a-t CURVE THE CURVE IS FIRST 0 (RA) FIND: USING THE METHOD OF SECTION 11.8 DIVIDED INTO THREE (a) X AT 1 = 52 5 REGIONS AS SHOWN. (b) XMAX FROM THE DISCUSSION FOLLOWING EQ. (1113) AND ASSUMING Xo = D. HAVE .. X= X=0 + Not, + [A(t,-t) a = de where de is THE SLOPE OF THE WHERE A IS THE AREA OF A REGION AND & IS THE HAVE ... DISTANCE TO ITS CENTROID. THEN FOR tie 165 ... J-t CURVE. THEN .. X=(-2 3)(165)+[[18516-0.75 32]](16-4)5 FROM t=0 TO t=105: 5= CONSTANT => Q=0 t=10 5 TO t=26 5: Q = -20-60 =- 5 9 1/52 1 (4 5)(2 3) (16-10)5 = ((4 5)(6 32))(16-14)5) =[-32+(-72+48+48)]m t=265 TO t= 415: N= CONSTANT => Q=0 t= 415 to t=465: a= -5-(-20) = 3 st/s2 OR X = -8.00 m t >465' N = CONSTANT => Q=0 11.87 GIVEN: a-t CURVE; AT t=0, N=7.5 FT/s; THE a-t CURVE IS THEN DRAWN AS SHOWN. AT t=t .. 5=0 n (11/s2)] FIND: USING THE METHOD OF SECTION 11.8 (a) ti (b) X AT t=t, (a) FROM THE DISCUSSION FOLLOWING EQ. (11.13), X= X0+ 50t, + EA(t,- 1) HAVE .. - 0.75 s --(CONTINUED) (CONTINUED)

11.88 CONTINUED WHERE A IS THE AREA OF A REGION AND ? IS THE DISTANCE TO ITS CENTROID, THEN, FOR 1,= 525... X =- SAO St + (60 \$ X 52 5)+ /- (165)(5 32)(52-18)5 +[(5 5)(3 \$)(52-43.5)5} = [-540+(3120)+(-2720+127.5)] 12 OR X=-12.50 St (b) NOTING THAT XMAX OCCURS WHEN IS = 0 (# = 0), IT IS SEEN FROM THE IS- CURVE THAT YMAX OCCURS FOR 10 5 < 2 < 26 5. ALTHOUGH SIMILAR TRIANGLES WULD BE USED TO DETERMINE THE TIME AT WHICH X. YMAX (SEE THE SOLUTION TO PROBLEM 11.63), THE FOLLOWING METHOD WILL BE USED. FOR 10 S < 1, < 26 S, HAVE $X = -540 + 601, -[(t, -10)(5)][\frac{1}{2}(t, -10)]$ (41) = -540 + 60t, - 2(t,-10)2 WHEN X: XMAX: 3= 60-5(2,-10)=0 OR (t) XMAS = 225

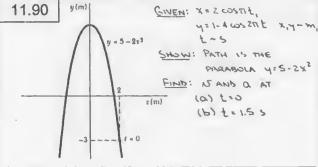


11.89 GOVEN: X=4t4- bt, y=6t3-2t2 x,y-mm, 1-5 FIND: IT AND Q AT (a) t=15 (b) t=25 (C) 1=45

4=6+3-2+2

HAVE. X=4t4-6t THEN SX=dX=16t3-6 5 = 2 = 18t - 4t a, - dis - 48t2 ay = dis = 36t - 4 OUR (a) AT 1=15: Sx =16(1)=6=10 5 54=18(1)=4(1)=14 5 OR 15 = 17.20 5 54.50 =48 mm (2 = 361)-4.32 mm 0x = 48 (1) = 48 mm (b) AT t=23: Nx = 16(2)3-6 = 122 mm Sy = 18(2)2-4(2)=64 mm OR 5 = 137.8 5 4 27.7° 04=36(2)-4=68 mm az + 48(2)2 = 192 5

OR Q = 204 MIR 4 19.50 (C) AT t=45: Nx=16(4) =6=1018 = Nx=18(4) -4(4)=272 = 18(4) = 0.00 0, = 48(4) = 768 3 04:36(4)-4=140 35 OR Q= 781 5 10.33



HAVE .. X = 2 COSTIT 4=1-4 cos 2712 4=1-4(2cos nt-1) THEN .. = 5-8(3)2 OR 4=5-2X2 Q.E.D. Now. L. F. = F. - STIMINE TIS - F. J. BTISINETIE AND Qx = TF = -272 cosnit Qy= TF = 1672 cossnit : 12:0 (a) AT t=0: 5x=0 00 22 0x = -217 2 5 04 = 1677 52 OR Q = 159.1 52 182.9 (b) AT {=1.55: 15, =-27 SIN(1.571) 12 = BT 114 (SU-12) = 54 Z 2 Q OR N. - 6.28 7 - $Q_{x} = 2\pi^{2} \cos(1.5\pi)$ $Q_{y} = 16\pi^{2} \cos(2\pi \cdot 1.5)$ 2-16-M2 OR Q = 157.9 321

11.91 GIVEN: X = 12(t-2) + 22 4= 12 - 1(t-1)2 X,4-12, 2-5 FIND: (a) Nomin (b) t, x, y, AND DIRECTION OF I

WHEN U. I'MIN (a) Have = X= 12(1-2)3+22 4= 12-2(1-1)2 THEN .. Nx = 3K = \$ (t-2)2+ 2t ふっぱいはとしてい = 12-4+1 = ませ~+七+1 = \$(t+2)2 = = (f-s)2 Now. 152 = 15x2 + 5x2 = 16[(++2)4 + (+-2)4] NOTING THAT IS MINIMUM WHEN IS IS MINIMUM, HAVE .. dur # [(++2)3 + (+-2)3] =0

EXPANSING .. (+2+6++12++8)+(+3-6+2+12+-8)=0 OR 2(t3+12t)=0 THE ONLY REAL ROUT OF THIS EQUATION IS LO.

5 = [(0-5) + (12+0)] = 5 OR Nomin = 1.414 5 (b) WHEN IS= NomiN t=0

X= 15 (0-5) 2+(0); OR X=-0.667 ft. OR 4 - - 0.500 ft 4= 12(0)3- 2(0-1)2 AND Nx = \$ (0+2)2=1 5 五=年(0-2)-1至 THEN TAND = 12 = 1 OR PLYNN 45 &

11.92

GIVEN: X=4t-2 SINT, Y=4-2 COST X,4-1N. 2-5 SKETCH: PATH OF THE PARTICLE FIND (a) NIMIN AND SMAX (b) t, X, Y, MND DIRECTION OF NT WHEN J. JMIN AND J. JMAN

HAVE - X= 4t- 2 SINT 4= 4-2 cost 2,5 x,100, 4,100, 4,100 0.5 0 4.28 40 12.57 6.0 20.8 4.0 25.1 2.0

(a) HAVE .. X = 4t-2 SINT 4= 4-2 cost THEN .. Nz : 32 = 4 - 2 cost J. = 37 = 2 SINE

Now. 52 = 5x2 + 5x2 = (4-2 cost)2 + (2 sint)2 = 20 - 16 cost

BY OBSERVATION .. FOR WMIN, COST = 1 SO THAT NMIN = 4 OR Nomin = 2 5

FOR JMAK, COST:-1 SO THAT SMIAX = 36 OR

NMAX = 6 5 (b) WHEN 5= 5min: cost = 1

OR t= 2NT S

WHERE N: 0, 1, 2, ... THEN .. X=4(2NT)-25IN(2NT) OR X.BNT IN. 4=4-2(1) OR 4.2 IN.

15x = 4-2(1)=2 13 D= (THS) NIC S = +2 ALSO ... : Bu = 0

WHEN J= JMAX: cost = -1

OR 1-(2N+1)TT 5

WHERE N=0,1,2, ...

THEN X=4(2N+1)n-ZSIN(2N+1)TI OR X=4(2N+1)TI IN 4=4-2(-1) OR 4=6 IN.

ALSO. Nx = 4-2(-1) = 6 3 Jy: 2 SIN (2N+1) 11 = 0 : 0 = 0 -

11.93 GIVEN: [= A(cost + t smt);

+A(swt-tcost); FIND: (a) t SO THAT I AND Q ARE PERPENINCULAR (b) & SO THAT I AND a ARE PARALLEL

HAVE .. [= A(cost + t sint) i + A(sint - t cost) i 5 = d= A(- sint + sint + t cost) = + A(cost-cost + t sint); = A (t cost) i + A(t sint) j a = ds = Al cost - t sint) i + A(sint + tcost) i

(a) WHEN [AND & ARE PERPENDICULAR, [.a. = 0 : A[(cost+t sint) + (sint-t cost)] · Allcost -t sint) i + (sint + t cost) i] = 0

OR (cost+tsint)(cost-zsint)+(sint-tcost)(sint-tcost)=0 (CONTINUED)

11.93 CONTINUED

OR (cos2 t-t2 sm2t)+(sm2t-t2 cos2)=0 1-1220 OR 1=15 OR

(b) WHEN I AND Q ARE PARALLEL, IXQ =0 :. A[(cost +t sint) = (sint -t cost))]

* A[(cost-tsint)i+(sint+tcost)j]=0

or [(cost+tsint)/sint+tcost)-(sint-tcost)(cost-tsint)]k=0 Expansing (sint cost +t +t' sint cost)

- (sint cost - t + t2 sint cost) . 0

0=5

21:0

GIVEN: [= X,(1-+1)]+(4,2 2 cos 2nt); 11.94 2~5; x,=30 mm, 4,= 20 mm 4/41 FIND: (a) I, I, AND Q AT 600 (b) [, 5 , ND Q AT t = 1.5 5 1.0 0.5 -0.5 -10-

HAVE. = = 30(1- 111 + 20(= 2 cos 271t); THEN .. I = de = 30 (1+1) 1 + 20 (- 20 200 201) - Pt

-30 (++1)2 j-2017 [= 71/2 (2 cosent+2 sun 21/2)] 0 = 9 = -30 (fol) = - 500 (\$500 suf . 5 7 in suf) AND

+ LE(-TISINETT + 4 cosent)]; = - (1+1) = 1 + 107 = 2 + 101 + 1 + (1+1)

(a) 15 t=0: [=30(1-1)1+20(1)]

OR C = 20 ram 2 = 30(+)j-20π[(1)(\$+0)]j

OR N = 43.4 MY 7 46.3 Q = - (1) 1 + 10 TO (1) (0-7.5)

OR Q=743 5 7854°

(b) AT t=155: [=30(1-25)]+20(-0.757(cos37)) = (18 mm) + (-1.8956 mm) j

= (4.80 5) + (2.9778);

CR N. S. 65 5 1318 a = - (2.5) = 1 + 10 m2 = 0.75 m (0 - 7.5 005 3 m) = (-3.84 3) 1 + (70,1582 3))

OR Q = 70.3 50 286.9

11.95 11.96 CONTINUED GIVEN: [= (Rt cos w,t) + ct; + (Rt sin w,t) + AND Q = -3(0) + 3(1) + (2-0) k FIND: NE AND Q THEN 02 (0)2+ (3)2+ 1212 = 13 OR Q= 3,61 58 HAVE .. [= (Rt cosunt) + ctj + (Rt sinunt) & (P) IT I MAD IT ARE PERPENDICULAR, I.V.O [\$(fine f) + ((1++) E) + ((teas til)): THEN .. IT = AT = R(cosunt - wit SINWAT) = + C) 0=[2(cost-toint)i+(3(tin));-(sint+tcost)]=0 +R(sin wat + wat cos wat) k or (3t coot)[3(cost-tsint)]+(3(tin)(3)(tin) a = # = R (-wasinout-wasinout-out coswat) AND + (t sint)(sint+tcost) = 0 +R(un cosunt + un cosunt-uit sin unt) E EXPANDING .. (9t cost - 9te sint cost) + (9t) = R(-zw, sinult - wit cos wit)i o = (ton't+t'sint cost) + OR (WITH t+0) 10+8cost-Bt sint cost =0 + R(ZWN cos wht - Wht sin wht) E Now .. 15 = 15x = 15x = 152 7+2 cos 2t - 2t sin 2t = 0 = [R(cosunt-unt sinunt)] + (c)2 USING TRIAL AND ERROR OR HUMERKAL METHODS, THE +[R(SINWAL+WIT COSWAL)] SMALLEST reogn 15 INCOSE THEN HIS THAT - SUNT SUNT NOTE: THE NEXT ROOT IS \$ = 4.38 S. + Wit sinicht) + (sinicht 11.97 1 (the sure thanks cosunt out to a with GIVEN: No = 315 h; h= 80 m FIND : d = R2(1+ W2 t2) + C2 N= 182 (1+W2+5)+C2 OR ALSO .. Q = Qx + Qx + Q2+ Q2 = [R(-204 SINULT - WAT COSWAT)] + (0)2 4[R(zwa coswat-wat sincat)] = Rylaw sment + 4 - 12 + sment cohont · White cosunt) + (4 wh cosunt FIRST NOTE .. No : 315 F - Aunt sinunt cosunt + unt to sinunt)] = 87.5 m = R2 (4W2 + W4 +2) VERTICAL MOTION OR a= RWH V4+WIZE (UNIFORMLY ACCEL MOTION) 80 m 4=43+(12782-29th 11.96 GIVEN: [. (At cost); AT B. - 80m = - 2 (9.81 32 12 + (A (ta))jo(Bt sint)k OR to = 4.038 555 ~~ ft, t~ s; A=3, HORIZONTILL MOTION LUNIFORM) ナイベルトをメーン B=1 SHOW: (7)2-(7)2-(8)2=1 AT B .. d = (87.5 %)(4.038 555) FIND: (a) IS AND a AT to d=353 m = (b) tmin (t+0) so THAT 11.98 I AND I ARE GIVEN: 5 IS HURRONTAL; PATH OF SHOWBALL FIND: (a) 5 PERPENDICULAR (b) d HAVE [: (At cost)] + (A 12+1)] + (Bt sint) & X=At cost 4=AVELLI Z=BtSINT Fs - (5)5-1 2 m THEN cost = At SINT = BE cost + sint = 1 = (At) = (BE) = 1 (A)2-1=(A)2+(B)2+(B)2+(B)2 OR (A)2-(A)2-(B)2=1 Q.E.D. (a) WITH A=3 AND B=1, HAVE .. (a) VERTICAL MOTION 1 = 2 = 3 (cost - t sint)] + 3 (till) + 1 (sint + t cost) } (UNIFORMLY ACCEL MOTION) 4= 484 (1578t - \$942 WAND U= 9£ = 3(-21114-2114-4 cost) =+3 (from - 4(from)) AT B .. - 1 m = - 1 (9. B) 5) t2 OR to = 0.451 524 5 + (cost + cost - t sut) to HORIZONTAL MOTION (UNIFORM) = -3(251Nt+tcost) = +3 (1241) 1/2) 1= x20+ (5,1) t AT B. 7 m = No (0.451 524 5) + (2 cost - t sint) & No = 15.503 OR No : 15.5031 3 (b) VERTICAL MOTION: AT C .. - 3 m = - \$ (9.81 50) } AT t=0: 15 = 3(1-0) = (0) + (0) } OR 15=3 th OR tc= 27820625 (CONTINUES) (CONTINUED)

11.98 CONTINUED HORIZONTAL MOTION AT C .. (7+d)m=(15.5031 3)(0.782 062 5) OR d= 5.12 m 11.99 GIVEN: No IS HORIZONTAL FIND: RANGE OF VALUES OF S. IF NEWSPAPER LANDS BETWEEN B AND C 215 3 14 VERTICAL MOTION (UNIF. ACCEL. MOTION) HORIZONTAL MOTION (UNIFORM) X= x3+ (5x) = 5= t AT B: 4: - 33 ft = - 2 (32.2 \$) t2 or to = 0. 455 0165 THEN .. X: 7 1 = (15) (0.455 OIL 5) OR (5.) = 15.38 12 ATC: 4: -24:- 2(32.2 5) 12 OR to = 0.352454 5 THEN .. X: 123 Lt = (15) (0,352 454 5) OR (No) = 35.0 \$: 15.38 \$ \$ N \$ 35.0 \$ 11,100 GIVEN: NJ IS HORIZONTAL; 31 IN. 5 / 5 42 IN. FINA: (a) RANGE OF VALUES OF & (b) of WHEN h= 31 IN. AND h= 42 IN. -40 R-1.5 St 2清红 (m) 8 1 28 - 40 91 -(a) VERTICAL MOTION (UNIF. ACCEL MOTION) 4= x3+ (27) 1- 1912 HORIZONTAL MOTION (UNIFORM) WHEN h=31 in., y=-212tt: -212tt =- 1(32.2 1/2)+2

11.100 CONTINUED

THEN.. 40 ft = $(N_0)_{31}(0.3874325)$ OR $(N_0)_{31}=103.244\frac{5}{5}=70.4\frac{mi}{5}$ WHEN h=421N., $y=-1.54t:-1.54t=-\frac{1}{2}(32.2\frac{5}{2})t^2$ OR $t_{42}=0.3052345$ THEN.. 40 ft = $(N_0)_{42}(0.3052345)$

THEN. 40 ft = (15) /2 (0.305 234 5)

OR (15) /2 = 131.047 5 = 89.4 77

... 70.4 77 = 15 = 89.4 77

(b) FOR THE VERTICAL MOTION

Ny = (27) - 9t

Now TAN a = $\frac{(N_1)_0}{(N_2)_0} = \frac{9t}{N_0}$

WHEN h= 31 In .: TANA = (32.2 52)(0.387 432 5) = 0.120833

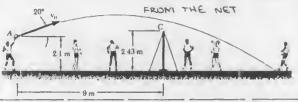
WHEN H= 42 IN.: TANK = (32.2 44)(0.305 2345) = 0.075 000

OR A42 \$ 4.29°

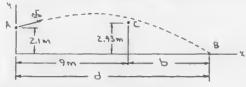
11.101 GIVEN: No 13.40 3

FINE: (a) IF BALL CLEARS THE NET

(1.) DISTANCE THE BALL LANDS



FIRST NOTE. (NI)= (13.40 =) cos 20 = 12,5919 = (NI)= (13.40 =) sin 20 = 4.5831 =



(a) HORIZONTAL MOTION (UNIFORM)

X=XS+(N_x)-{

AT (... 9 m = (12.5919 %)t or to=0.71475 5 <u>VERTICAL MOTION</u> (UNIF. ACCEL, MOTION)

4=40+(54)5t- 29t2
AT C .. 46 = 2.1 m + (4.5831 =)(0.714755)

- 2(9.81 %2)(0.71475 5)2

= 2.87 m

... Ye > 2.43 m (HEIGHT OF NET) \Rightarrow BALL CLEARS HET

(b) AT B, y=0: 0=2.1m+(4.5831 $\frac{1}{5}$)t- $\frac{1}{2}$ (9.81 $\frac{1}{5}$)t²

Solving... t_B=1.271175 \ (THE OTHER ROOT IS

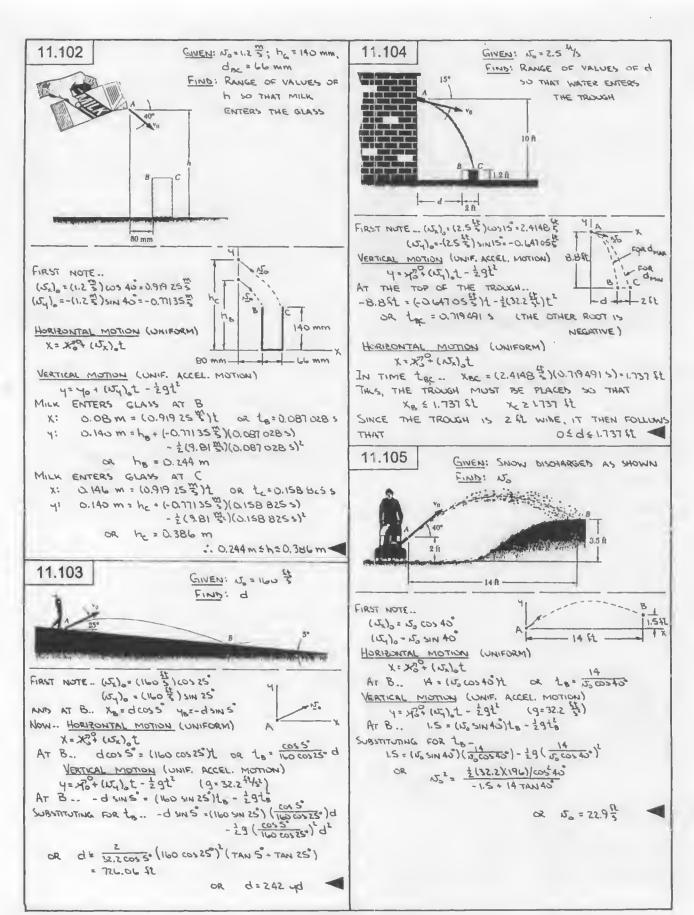
NEGATIVE)

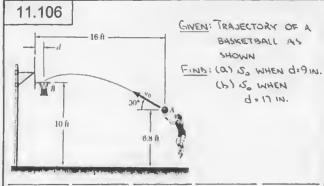
THEN. d=(12,5) to = (12,5)19 \$)(1.271 1755)

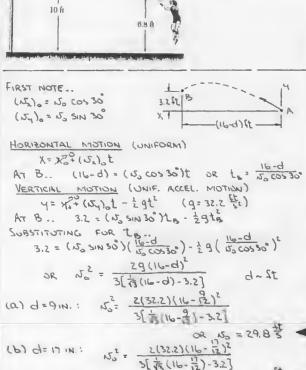
.. THE BALL LANDS b=(16.01-9.00)m=7.01 m FROM THE NET

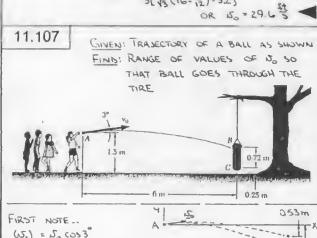
02 t31=0.387 432 5

(CONTINUED)









11.107 CONTINUED

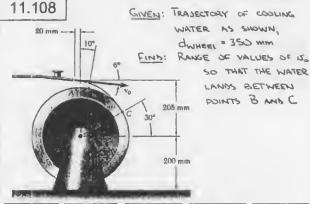
OR 50 = 177.065

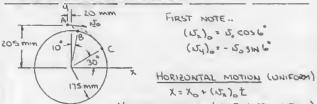
AT B, 4=-0.53 m; 50 = 0.514447-(-0.53)

OR (No) = 14.48 m

AT C, 4=-1.25 m; 50 = 0.514447-(-1.25)

OR (No) = 10.64 m





VERTICAL MOTION (UNIF. ACCEL. MOTION)

Y=Y=+ (LT) t - 29t (3=9.813)

AT POINT B: X= (0.175 m) SIN 10, Y= (0.175 m) COS 10

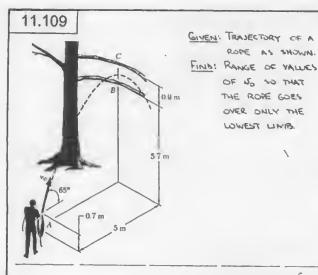
x: 0.175 4110 = -0.020 + (15 cost) t

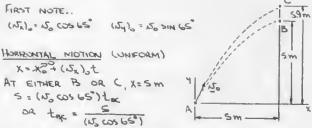
or to = 0.050 188

AT POINT C: X= (0.175 m) cos 30°, 4= (0.175 m) sin 30° X: 0.175 cos 30° = -0.020 + (15, cos 6°) t CR tc = 0.171 554

4: 0.175 SIN 30 = 0.205 * (- $\sqrt{5}_0$ SIN $\sqrt{6}$) $\frac{1}{6}$ - $\frac{1}{2}$ 9 $\frac{1}{6}$ SUBSTITUTING FOR $\frac{1}{6}$ - $\frac{1}{2}$ 91 $\frac{1}{2}$ 9.81) $\frac{0.171554}{3.5056}$ $\frac{1}{2}$ 9.81) $\frac{0.171554}{3.5056}$ $\frac{1}{2}$ 9.81) $\frac{0.171554}{3.5056}$ $\frac{1}{2}$ 9.81) $\frac{1}{2}$ 9.81 $\frac{1}{2}$ 9.81) $\frac{1}{2}$ 9.81 $\frac{1}{2}$ 9.81) $\frac{1}{2}$ 9.81 $\frac{1}{2}$ 9

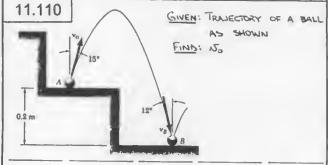
: 0.678 # 22 = 1.511 # .





AT POINT B: 50 - 5 TAN 65 - 5 OR (US) = 10.95 5 AT POINT C: 50 - 5 TAN 65 - 5 OR (US) = 11.93 5

: 10.95 \$ £ 15, 511.93 \$ <



FIRST NOTE .. (NX) = NO SIN IS

(NY) = NO COS IS

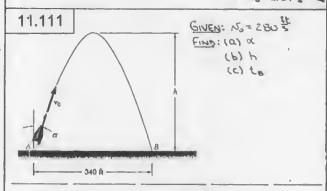
HORIZONTAL MOTION (UNIFORM)

NX = (NX) = NO SIN IS

(CONTINUED)

11.110 CONTINUED

VERTICAL MOTION (UNIF. ACCEL. MOTION) $N_{1} = (N_{1})_{0} - 9t$ $= N_{2} \cos 15^{2} - 9t$ $= (N_{3} \cos 15^{2})_{1}^{2} - \frac{1}{2}9t^{2}$ AT ROINT B, $N_{1} < 0$ THEN.. TAN $12^{\circ} = \frac{(N_{1})_{0}}{(10N_{1})_{0}} = \frac{N_{2} \times N_{1}15^{\circ}}{9t_{0}^{2} - N_{2} \cos 15^{\circ}}$ OR $t_{0} = \frac{N_{2}}{9}(\frac{5N_{1}15^{\circ}}{14N_{1}12^{\circ}} - \cos 15^{\circ})$ $= 0.22259N_{0}$ Noting that $q_{0} = -0.2m$, have.. $-0.2 = (N_{2} \cos 15^{\circ})(0.22259N_{2})^{2}$ $= \frac{1}{2}(9.81)(0.22259N_{2})^{2}$ OR $N_{2} = 2.67\frac{m}{1}$



FIRST MOTE. (LX) = LO SINK = (280 \$) SINK

(LY) = LO COSK = (280 \$) COSK

(A) HORIZONTAL MOTION (UNIFORM)

X = X3 + (LX) = (280 SINK) \(\)

AT POINT B: 340 = (280 SINK) \(\)

OR \(\)

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(b) HAVE -- 154 = (154)0-9t = 280 cos x -9t

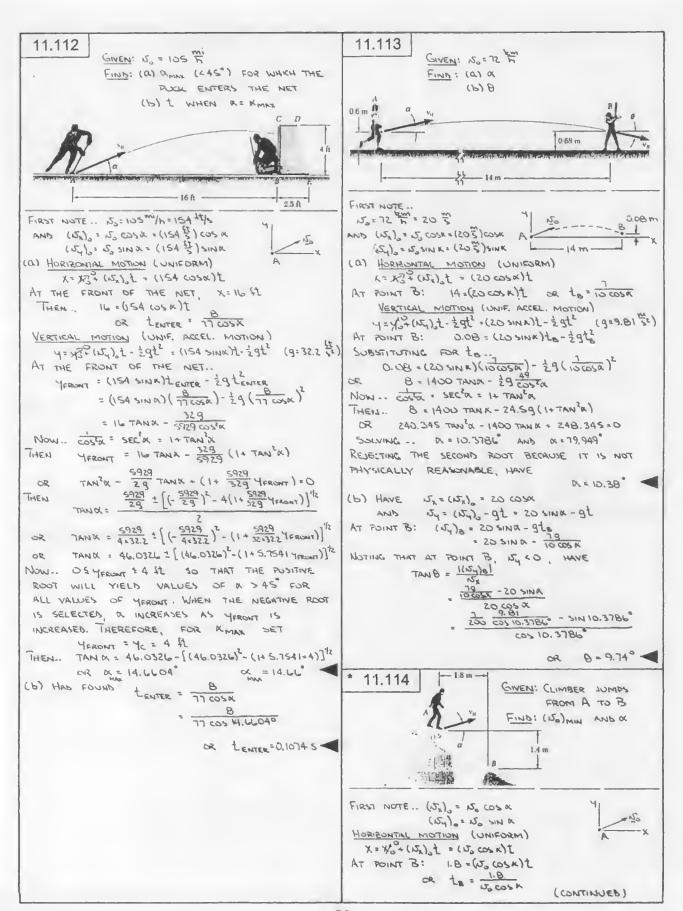
WHEN YEYMAX = h, 154 = 0: 0 = 280 cos x - 9t

OR th = 280 cos x - 013 59° = 8.67433 S

THEN.. $h = (280 \cos \alpha) t_h - \frac{1}{2} 9 t_h^2$ = $(280 \cos 4.01359^{\circ})(8.67433) - \frac{1}{2}(32.2)(8.67433)^2$ or h = 1211 ft

(C) HAD FOUND .. to = 17 SINK

17 14 SIN 4. 013 59° OR to +17.35 5



11.114 CONTINUED

VERTICAL MOTION (UNIF. ACCEL. MOTION) $y = y_0^{0+} (u_1)_0^{1} + \frac{1}{2} gt^2 \cdot (u_2 \sin n)_1^{1} + \frac{1}{2} gt^2 \quad (g = 9.81 \frac{m}{6})_{1}^{1}$ AT POINT B: $-1.4 = (u_2 \sin n)_1^{1} + \frac{1}{2} gt^2_{1}^{1}$ $SUBSTITUTING FOR tB ...
-1.4 = (u_2 \sin n)_1^{1} (\frac{1.8}{15 \cos n}) - \frac{1}{2} g(\frac{1.8}{15 \cos n})^2$ OR $S_0^2 = \frac{1.62 g}{\cos n(1.8 \tan n + 1A)}$ $= \frac{1.62 g}{0.9 \sin 2n + 1.4 \cos^2 n}$

Now minimize S_{c}^{2} with RESPECT to X. HAVE.- $\frac{dS_{c}^{2}}{dA} = 1.629 \frac{-(1.8 \cos 2\alpha - 2.8 \cos 3\alpha \sin \alpha)}{(0.9 \sin 2\alpha + 1.4 \cos^{2} x)^{2}} = 0$ or $1.8 \cos 2\alpha - 1.4 \sin 2\alpha = 0$

OR TAN 2K = 17 OR K = 26.0625° AND K = 206.06° REJECTING THE SECOND VALUE BECAUSE IT IS NOT PHYSICALLY POSSIBLE, HAVE...

FINALLY, 50 = 1.62 + 9.81

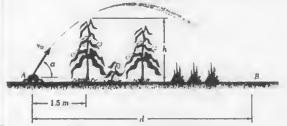
COSE 26.0625 (1.8 TAN 26.0625 - 1.4)

OR (NS)min = 2.945

11.115

GIVEN: Jo = 8 5

FIND: (a) dmax AND & WHEN h=0 (b) dmax AND X WHEN h=1.8 m



FIRST NOTE.. $(N_X)_0 = U_0 \cos n = (8\frac{\pi}{6})\cos n$ $(N_X)_0 = U_0 \sin n = (8\frac{\pi}{6})\sin n$ $(N_X)_0 = U_0 \sin n = (8\frac{\pi}{6})\sin n$ HORROWTH MOTION (UNIFORM)

AT POINT B, X = d: $d = (8\cos n)t$ or $t_0 = 8\cos n$ VERTICAL MOTION (UNIF. ACCEL. MOTION) $Y = y/6 + (N_X)_0 t - 29t^2 = (9\sin n)t - 29t^2$ AT POINT B: $0 = (8\sin n)t_0 - 29t^2$

SIMPLEYING AND SUBSTITUTING FOR $\frac{1}{8}$... $0 = 8 \sin \alpha - \frac{1}{2} 9 \left(\frac{d}{8 \cos \alpha} \right)$

OR d = 9 SINZX (1)

(a) WHEN h=0, THE WATER CAN FOLLOW MAY PHYSICALLY POSSIBLE TRAJECTORY. IT THEN FOLLOWS FROM EQ. (1) THAT d IS MAXIMUM WHEN 2K=90

THEN d= 64 SIN(2x45°)

OR dmax=6.52m

(b) Based on EQ. (1) AND THE RESULTS OF PART

Q, IT CAN BE CONCLUDED THAT I INCREASES

IN VALUE AS & INCREASES IN VALUE FROM

(CONTINUES)

11.115 CONTINUED

O TO 45° AND THEN & BECREASES AS K IS
FURTHER INCREASED. THUS, dmax occurs for the
VALUE OF K CLOSEST TO 45° AND FOR WHICH THE
WRITER JUST PHYSE'S OVER THE FIRST ROW OF CORN
PLINTS, AT THIS ROW XCORN = 1.5 M
50 THAT LEARN = 1.500K

ALSO, WITH YOURN = h, HAVE $h = (B \sin \alpha) t_{conn} - \frac{1}{2} g t_{conn}^2$ Substituting for team and nuting hall BW, $1.8 = (B \sin \kappa) (\frac{1.5}{B \cos \kappa}) - \frac{1}{2} g (\frac{1.5}{B \cos \alpha})^2$

2.259 2.8

NOW... COSTR = SEC²R = 1 + TAN²R

THEN 1.8 = 1.5 TANK - 2.25(9.81) (1+ TAN²R)

OR 0.172441 TAN²R - 1.5 TANR + 1.972441 = 0

SOLVING -- α = 58.229° AND R = 81.965°

FROM THE. ABOVE DISCUSSION, IT FOLLOWS THAT

d = dmax WHEN

FINALLY, USING EQ (1)

d= 64 SIN (2x 58.229°)

02 dmax = 5.84 m

11.116 GIVEN: No = 11.5 B

FIND: (a) dmax

(b) a when dadmax

d

IIIm

FIRST NOTE.. (NE) = NO CON = (11.5 %) CON TO (NE) = NO CON TO (11.5 %) SIND TO (NE) = NO CON TO (11.5 %) SIND TO (NE) = NO CON TO (NE) = NO CO

 $= \frac{1}{2q} (11.5)^2 51N^2 \alpha$ OR $51N^2 \kappa = \frac{2.2 \times 9.81}{11.5^2} \qquad \alpha = 23.8265^{\circ}$

(a) Horizontal motion (UNIFORM) $X = X_0^2 + (U_n)^2 = (11.5 \cos x)^2$

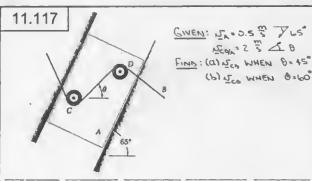
AT POINT B, X=dmax AND L=LB WHERE LB = 115 SIN 23.8265 = 0.473 S6 S THEN ... dmax = (11.5)(cos 23.8265)(0.473 S6)

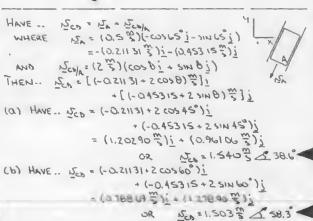
THEN - dmax = (11.5)(cos 23. B265°)(0.473 56)

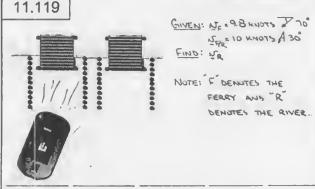
OR dmax = 4.98 m

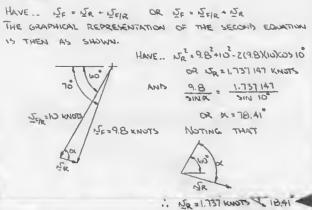
(b) FROM ABOVE

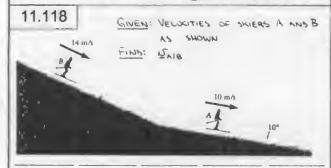
X = Z3.8°











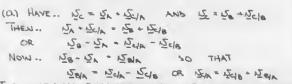
HAVE .. LA . NB + LNB
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



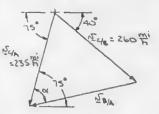
THEN.. $J_{A|B}^{2} \circ 10^{2} + 14^{2} - 2(10)(14) \cos 15^{\circ}$ OR $J_{A|B} \circ 5.05379^{\frac{10}{3}}$ AND $\frac{10}{51N0} \circ \frac{5.05379}{51N15^{\circ}}$ OR $N = 30.8^{\circ}$

: JA16: 5.05 \$ 155.8°





THE GRAPHICAL REPRESENTATION OF THE LAST EQUATION IS THEN AS SHOWN.



HAVE...

150/2 = 2352 + 2606

-2(235)(26)(20)(20)(50)(5)

OR 150/2 = 266.198 PM

AND

260 = 266.798

260 = 266.798 SING = 210.03° OR X = 62.03° .: NSBA = 267 M = 12.97°

(b) HAVE.. LE . LA M) - (235 M) (-COSTS 1- SINTS 1)

(CONTINUES)

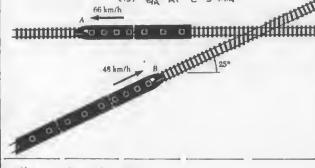


FOR At = 15 MIN: A Cops = (260 m) (4 h) = 65 mi

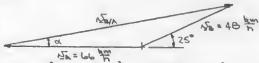
11.121

GIVEN: CONSTANT VELOCITIES OF TRAINS A AND
B; AT t=0, A is AT THE CROSSING;
AT t=10 MIN, B is AT THE CROSSING

(b) THA AT 1=3 MIN



(O) HAVE.. IS " IN + ISUA
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



THEN .. USA = 66 \$ THE - 2(66) COS 155°

OR USA = 111.366 RM/h

AND 48 111.366

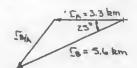
OR A = 10.50 .. Noh = 111.4 km 10.50

(b) FIRST NOTE THAT

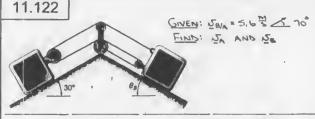
AT 1=3 MIN, A 15 (66 \$ (60) = 3.3 km WEST OF THE CROSSING.

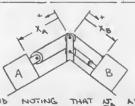
At t=3 min, B is $(48\frac{km}{h})(\frac{1}{160})=5.6$ km southwest of the crossing.

NOW .. IS = IA + IS/A
THEN AT \$2.3 MIN HAVE ..



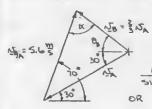
 $r_{\text{WA}}^2 : 3.3^2 + 5.6^2$ - 2(3.3)(5.6)(0525° OR $r_{\text{WA}} = 2.96 \text{ km}$





FROM THE DIAGRAM.. $2X_A + 3X_B = CONSTANT$ THEM.. $2U_A + 3U_B = O$ OR $|U_B| = \frac{3}{2}U_A$

AND NOTING THAT IT AND ITS MUST BE PARALLEL TO SURFACES A AND B. RESPECTIVELY, THE GRAPHKAL REPRESENTATION OF THIS EDUCATION IS THEN AS SHOWN. NOTE: ASSUMING THAT IT IS DIRECTED OF THE INCLINE LEADS TO A VELOCITY BAGRAM THAT DOES NOT CLOSE.



FIRST NOTE ..

0 = 180 - (40 + 30 + 80)

= 110 - 80

DA SIN(110-θ₀) = 3 DA SIN(30+θ₀)

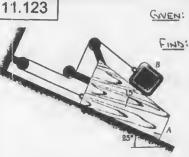
OR DA SIN(110-θ₀) = 0,964 18

OR B = 35.3817 AND B = 4.6183°
FOR B = 35.3817:

No = \frac{2}{5} NA = \frac{5.65111.40}{5111.(30+35.3817°)}

CR NT = 5.94 \$ NE = 3.96 \$ 730 \$... NT = 3.96 \$ 7354

FOR Do = 4.6183: No = 3Nx = 5.6 LIN 40° OR Nx = 9.50 \$ No . 6.34 \$ Nx = 9.50 \$ Nx = 9.50 \$ 730° Nx = 6.34 \$ 1.462°



CNEN: 12 = 8 14. 25.

FROM THE BIAGRAM..

2 XA + XB/A = CONSTANT

THEN.. 2 NA + NB/A = 0

OR | NB/A | = 16 5

AND 2 QA = QB/A = 0

OR | QB/A | = 12 18

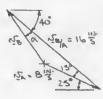
(CONTINUES)

11.123 CONTINUED

Note that her and by must be parallel to the top surface of block A.

(a) HAVE .. US = NA + NOWA

THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS THEN AS SHOWN. NOTE THAT BECAUSE A 13 MOVING DOWNWARD, B MUST BE MOVING UPWARD RELATIVE TO A.



HAVE..

Ly 2 8 + 16 - 2(8)(16)(05) 15

OR LY = 8.5278 18.

AND 8 8.5278

SINK 51N 15

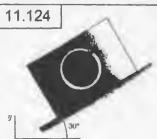
OR 0. 14.05

. NY = 8.53 18. \ 54.1

(b) THE SAME TECHNIQUE THAT WAS USED TO DETERMINE US CAN BE USED TO DETERMINE QB.
AN ALTERNATIVE METHOD IS AS FOLLOWS.

HAVE .. Q8 = Q4 + Q8/A = (6 i) + 12(-cosisi+sinisi); = -(5.5911 182) i + (3.1058 182); OR Q8 = 6.40 182 54.1

* NOTE THE ORIENTATION OF THE COORDINATE AXES ON THE SKETCH OF THE SYSTEM

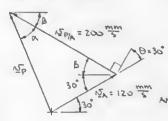


(P) \$\tilde{Q}^2 \tilde{M} \tilde{M}

NOTE: RATHER THAN APPLY THE SAME METHOD OF SULUTION TWICE, TWO EQUALLY APPLICABLE TECHNIQUES WILL BE USED.

(a) METHON 1.

HAVE .. LED = LEA + LEMA
THE GRAPHICAL REPRESENTATION OF THIS ENWATION
IS THEN AS SHOWN.



FIRST NOTE..

\$=90-(30+30)=30

THEN..

Jp2=120+200

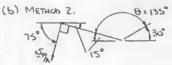
-2(120)(20)(20)(20)(20)

CR Jp=174.35L

TO 120

SING 510 600

OR K = 36.6° ... Np = 174.4 5 166.6° €



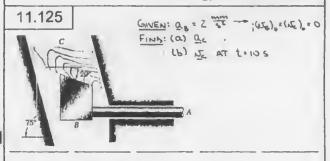
(CONTINUED)

11.124 CONTINUED

HAVE .. No = NO + NO 1 + NO 1 + 200 (- COSTS 1 - SINTS 1)

= (52.159 mm) 1 - (133.185 mm) 1 - SINTS 1)

OR NO = 143.0 mm \ CAC



(a) Have.. Qc = Qx + Qc/8
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.

(b) FOR UNIFORMLY ACCELERATED MOTION ..

NE = (12)0 + act
AT 1=10 5: NE = (0.83506 5 1(105) = 8.3506 5 1 105) = 8.3506 5 1 105 1 = 8.3506 5 1 105 1 = 8.3506 1 105 1 = 8.3506 1 105 1 10



GIVEN: QA = 1,2 = 7; (SB) = 0 QNA = 0.5 = 7 SS FIND: (Q) QB (b) SB AT t= 25

(a) Have.. QB = QA + QWA

THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



HAVE ... Q = 1.22 + 0.5 - 2(1.2)(0.5) cos 50°

OR Q = 0.958 46 34

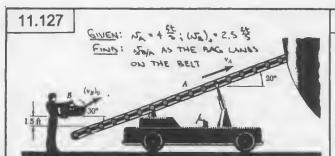
ANC. 0.5 - 0.958 46

SINK SIN 50°

LIET FE BSP.O = BD ...

(b) FOR UNIFORMLY ACCELERATED MOTION ...

AT t= 25: Up = (0.95846 52)(25)=1,91692 5 OR NB=1.917 5 7 23.6



FIRST DETERMINE THE VELOCITY

OF THE BAG AS IT LANDS

ON THE BELT. NOW...

[[U5],] ... (U5), COS 30 ... (2.5 \(\frac{5}{5} \)) COS 30

(U5),] ... (U5), SIN 30 ... (2.5 \(\frac{5}{5} \)) SIN 30

X

HORIZONIAL MOTION (THIERDAY)

(= X2+[(150)x], t (150)x = [(150)x]. = (2.5 cos 30) t = 2.5 cos 30

VERTICAL MOTION (UNIF. ACCEL MOTION) Y=Y_+ + ((NB)/1)t-29t2 (NB)/1= (NB)/1-9t =1.5+(2.5)1N30)t-29t2 = 2.551N30-9t

THE EQUATION OF THE LINE COLUNEAR WITH THE TOP SURFACE OF THE BELT IS

OS NAT X = P

Thus, WHEN THE BAG REACHES THE BELT.. 1.5+(2.5 SIN 30) $t-\frac{1}{2}9t^2$ = [(2.5 COS 30) t] TAN 20 OR $\frac{1}{2}(32.2)t^2+2.5(\cos 30)$ TAN 20-SIN 30) t-1.5=0 OR 16.1 $t^2-0.46198t-1.5=0$

SOLVING .. t . 0.31992 5 AND t = -0.29122 5 (REJECT)
THE VELOCITY US OF THE BAG AS IT LANDS ON
THE BELT IS THEN ..

\$ = (2.5 co + 30)] + { 2.5 5 10 30 - 32.2 (0.31992)] }

= (2.16 51 5)] - (9.0514 5) j

FINALLY - US = US + US/A

CR US/A = (2.1651) - 9.0514) - 4(cos 20) + 51N 20)

=-(1.59367 5) - (10.4195 5)

TO 13

OR NEW = 10.54 3 7813



GNEN: (UZ) = 6 \$ 7 50°

FIND: (Q) US IF US = 15

VERTICAL

(b) US IF US = 15

FIRST DETERMINE THE VELOCITY OF
THE COAL AS IT LANDS ON THE
BELT. NOW..

[(15c)_x]₀ = (15c)₀ cos 50 = -(15c) cos 50
[(15c)_x]₀ = -(15c)₀ sin 50 = -(15c) sin 50

HORIZONTAL MOTION (UNIFORM)

(15c)_x = [(15c)_x]₀ = -6 cos 50
= -3.8567

VERTICAL MOTION (UNIF. ACCEL. MOTION)

(15) = (15) - 29 (4-46) (9:32.2 5)

AT THE BELT: (15) = (-6 SIN 50) - 2(32.2)(-5)

OR (15) = -18.5237 5

(CONTINUED)

11.128 CONTINUED

THEN IS = 13.8567 \$) = (18.5257 \$);

(a) HAYE.. I'C = I'S + I'C IS

IF I'E IS TO VERTICAL, THEN (ISOIN = 0 WHICH

IMPLIES (ISO)X = (ISO)X

... - I'S COSIO = -3.8567

OR 15 = 3.92 \$ 100

(b) HAVE... LE . LE . LIB ... THE GRAPHICAL REPRESENTATION OF THIS EGULATION IS THEN AS SHOWN.



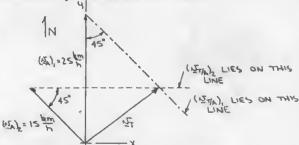
FOR \underline{U}_{0} TO BE MINIMUM, \underline{U}_{0} MUST BE PERPENDICULAR TO \underline{U}_{0} . $\therefore U_{0} = 10.9209 \cos 30.839^{\circ}$

OR NB = 0.581 \$ 210

11.129

FIND: MT. WHERE IT IS CONSTANT (MA) = 12 PM (MIN) (A) = 52 PM (MIN) (MIN) = 52 PM (MIN) (MIN) = 52 PM (MIN) (MIN)

HAVE. IT = ISA + ISTA USING THIS EQUATION, THE TWO CASES ARE THEN GRAPHCALLY REPRESENTED AS SHOWN.



FROM THE DIAGRAM .. (UT) x = 25-15 SIN45° = 14.3934 TO (UT) x = 15 SIN45° = 10.6066 TO

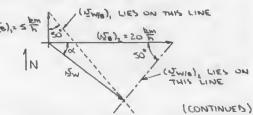
: UT . 17.88 H 436.4°

11.130



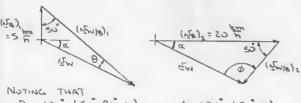
(12mb) 5 200, (12mb) 5 50 22 - 1 (12mb) 5 50 22 - 1 (12mb) 4 20,

HAVE. NEW & NEW LOUDING THE TWO CASES ARE THEN GRAPHICALLY REPRESENTED AS SHOWN.



11.130 CONTINUED

WITH IN NOW DEFINED, THE ABOVE DIAGRAM IS REDRAWN FOR THE TIND CASES FOR CLARITY.



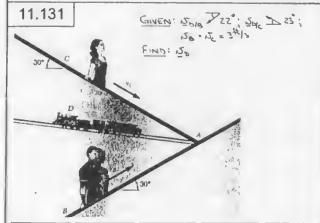
8= 180- (50+90+K) = 40- K

THEREFORE SIN (40°-01)

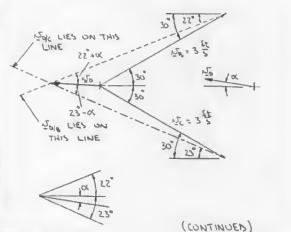
THEN

SIN(130-A) OR SIN 130 COSA - COS 130 SINA = 4(SIN 40 COSX - COS 40 SINA) TANA = SIN130 - 4 SIN40 cos 130 - 4 cos 40

DR K = 25.964° NW = 551N 50° = 15.79 km



HAYE .. No = No + No 13 12" = "E + 120F THE GRAPHICAL REPRESENTATIONS OF THESE EQUATIONS ARE THEN AS SHOWN.



11.131 CONTINUED

THEM - NO 3 SIN (22 - A) 51N7° = 3 EQUATING THE EXPRESSIONS FOR 3.

SINB (SIN 23 COSA - COS 25 SINA)

$$= \frac{\sin 7^{\circ} (\sin 23^{\circ} \cos x + \cos 22^{\circ} \sin x)}{\sin 23^{\circ} - \sin 7^{\circ} \sin 22^{\circ}}$$

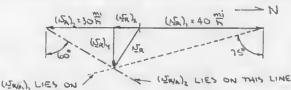
$$= \frac{\sin 8^{\circ} \cos 23^{\circ} + \sin 7^{\circ} \cos 22^{\circ}}{\sin 23^{\circ} + \cos 7^{\circ} \cos 22^{\circ}}$$

THEN
$$N_0 = \frac{35108}{510(22+2.0728)} = 1.024 \frac{52}{5}$$

11.132

FIND: Na

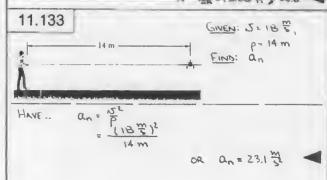
HAVE .. JR = (JA), - (JA/A), でも。(でり)・(でかり) THE GRAPHICAL REPRESENTATIONS OF THELE EDUCTIONS ARE THEN AS SHOWN. NOTE THAT THE LINE OF ACTION OF (INN) MUST BE DIRECTED AS SHOWN SO THAT THE SECONS VELOCITY DIAGRAM CLUSES.



THIS LINE FROM THE DIAGRAM. (No.) = [40+(No.) THUIS" EQUATING THE EXPRESSIONS FOR (NR) -.

[40+(NR),] TANIS = [30-(NR),] TAN 30 OR (NR) = 7.0109 Th

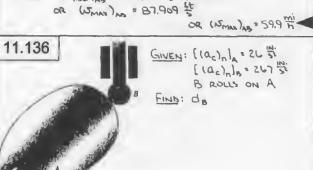
(NR)y = (40+7, 8109) TANIS = 12, 8109 T THEN OF -15.00 PT 7586 5



11.134

GIVEN: CIRCULAR TRACK OF DIAMETER d

FIND: (a) d WHEN $J = 72 \frac{1}{2} \frac{1}{2}$



FIRST NOTE THAT ROLLING WITHOUT SLIPPING

IMPLIES $(N_c)_A = (N_c)_B = N_c$ NOW... $[(Q_c)_n]_a = \frac{N_c^2}{P_A}$ AND $[Q_c)_n]_B = \frac{N_c^2}{P_B}$ WHERE $P_b = \frac{d_b}{Z}$ THEN... $P_A[(Q_c)_n]_A = [(Q_c)_n]_B (\frac{d_b}{Z})$ Substituting... $(2.6 \text{ in.})(26 \frac{N_c^2}{S^2}) = (267 \frac{N_c}{S^2})(\frac{d_b}{Z})$ OR $d_b = 0.506 \text{ in.}$

C 80

11.137

(0a) 2 = 20 mm (0a) 2 = 20 mm (10) 0a AT 2 = 0 (6) 0a AT 2 = 0

(a) At t = 0, $J_{x} = 0$ which implies $(A_{x})_{x} = 0$ i. $a_{x} = (a_{x})_{t}$ or $a_{x} = 70 \text{ st}$ (b) Have uniformly accelerates motion...

i. $J_{x} = (J_{x})_{0}^{2} + (a_{x})_{t}^{2}$ At t = 25: $J_{x} = (20 \text{ mm/s}^{2})(25) = 40 \text{ mm/s}^{2}$ How. $(a_{x})_{0}^{2} + (a_{x})_{0}^{2} + (a_{x})_{0}^{2}$ Finally, $a_{x}^{2} = (a_{x})_{0}^{2} + (a_{x})_{0}^{2}$ $= (20)^{2} + (17.7778)^{2}$ or $a_{x} = 26.8 \text{ st}^{2}$

11.138 GIVEN: d=250 mm; N= 45 m; a= constant;
AT t=95, N=0
FIND: t WHEN Q=40 52

HAVE UNIFORMLY DECELERATED MOTION...

1. 5 = 5 + 0,t

AT \(\frac{1}{2} \text{9} \); 0 = 45 \(\frac{1}{3} \text{4} \text{4} \text{6} \text{9} \);

CR \(\alpha_2 = -5 \)

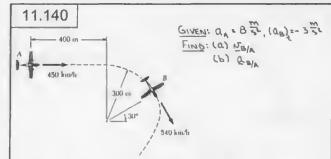
NOW... \(\alpha^2 \text{4} \text{6} \text{7} \text{4} \text{6} \text{7} \text{8} \text{7} \text{8} \text{7} \text{8} \text{7} \text{8} \text{7} \text{8} \text{7} \text{8} \text{8} \text{8} \text{7} \text{8} \text{8} \text{7} \text{8} \text{8} \text{7} \text{8} \text{8} \text{8} \text{7} \text{8} \

11.139 GIVEN: d=420 R; $a_1=constant$; $a_2=14 \frac{ct}{5}$, $a_3=25 \frac{ct}{5}$, $a_4=constant$; $a_2=14 \frac{ct}{5}$, $a_4=constant$; $a_4=14 \frac{ct}{5}$.

Have uniformly accelerated motion... $a_2=a_1^2+2a_1a_2^2+2a_1a_3a_2^2+2a_1a_3a_1a$

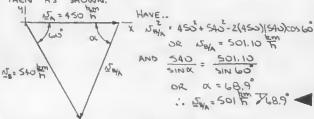
ALSO.. $N = N_1 + \alpha_1 t_1$ AT t = 25: $N = 14 \frac{t}{5} + (2 \frac{5t}{5})(2 \frac{5}{5}) = 18 \frac{5t}{5}$ NOW.. $\alpha_n = \frac{N_2}{6} \frac{(18 \frac{1}{5})^2}{210 \frac{5}{11}} = 1.54286 \frac{5t}{5}$

FINALLY.. $a^2 = a_t^2 + a_n^2$ AT t = 25: $a^2 = 2^2 + 1.54286^2$



FIRST NOTE.. No = 450 h No 540 h = 150 h

(a) Have.. No = No + No No THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS THEN AS SHOWN.



(b) First NOTE .. $Q_{B} = 8\frac{m}{3^{2}} \rightarrow (Q_{B})_{z} = 3\frac{m}{5^{2}} \Delta 60^{\circ}$ NOW .. $(Q_{B})_{n} = \frac{\sqrt{5}}{P_{B}} = \frac{(150^{m}/5)^{2}}{300^{m}} OR (Q_{B})_{n} = 75\frac{m}{5^{2}} \sqrt{30^{\circ}}$

THEN. QB = (QB) + (QB) = 3(-cos60 1+ SIN 60 1) + 75(-cos301-SIN 301) = -(66.452 32) - (34.902 32)

FINALLY.. QB = QA + QB/A

OR QB/A = (-66.4521-34.902)-(81)

= - (74.452 3)1-(34.902 3)

OR QB/A = 82.2 37 7251

11.141

CIVEN: astraight = at = constant,
AT t=0, CAR ENTERS

EXIT RAMP; FOR t>105,
U=20 71, a=4 ast.



FIRST MOTE .. NO : 20 TH BB IT TORTION OF THE HIGHWAY

By observation, amax occurs when is is maximum, which is at t=0 when the car first enters the ramp. For uniformly becelerates motion $S=S_0+Q_0$

(CONTINUES)

11.141 CONTINUED

AND AT E 105: No constant of $\alpha \cdot \alpha_n = \frac{C_n^2}{P}$ THEN $\alpha_{st} = \alpha_e$ of $\frac{1}{4}\alpha_e = \frac{C_n^2}{P} = \frac{C_n^2}{5 + 0}$

THEN AT t = 10.5:

Que = -6.1460 \$\frac{5}{5}\$

THEN AT t = 10.5:

Que = -6.1460 \$\frac{5}{5}\$ \((10.5) \)

OR $u_0^2 = 90.793 \frac{5}{5}$

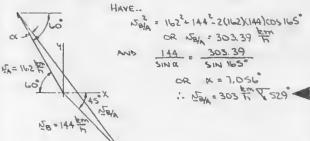
THEM AT \$= 0: amm = \(\alpha_{\text{t}}^2 + \left(\frac{\sighta}{5}^2 \right)^2 \\
2 \left\{ (-6.1460 \frac{\sighta}{5}^2 + \left(\frac{(90.793 \frac{\sighta}{5}^2 \right)^2}{5666} \right)^2 \right\}^2

OR amax = 15.95 \frac{\sighta}{6}

11.142

| Som | Caple = 7 | Som | Caple = 7 | Som | Caple = 2 | Som | Caple = 300 m | Caple = 30

FIRST NOTE.. No 162 h = 45 5 Us = 144 h = 40 5 (a) Have.. No 5 4 4 2010 OF THIS EQUATION IS THEN AS SHOWN.



(b) FIRST NOTE .. (QA) = 750 760° (QB) = 250 745°

THEI. $(a_A)_n = \frac{(45 \frac{m}{5})^2}{300m}$ $(a_B)_n \cdot \frac{(40 \frac{m}{5})^2}{250m}$

OR (QA)n = 6.75 \$2730 (Qa)n = 6.40 \$2 \$4.45

NOTING THAT $Q = Q_{E} + Q_{D}$ HAYE.. $Q_{A} = 7(\cos 60) - \sin 60) + 6.75(-\cos 30) - \sin 30)$ = -(2.3457 %) - (9.4372 %)

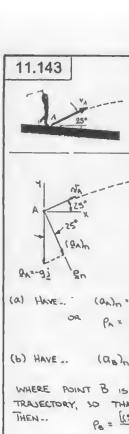
AND Q8 = 2(0045 1 - 51445 1) + 6.40(0045 1 + 51445 1)
= (5.9397 1/2) 1 + (3.1113 1/2)

FINALLY.. QB = QA + QB/A

OR QB/A = (5.9397 j + 3.1113 j) - (-2.3457 j - 9.4372 j)

= (8.2854 %) j + (12.5485 %) j

OR QB/A = (5.04 % 56.6



GIVEN: JA = 50 3 FIND: (a) P AT POINT A (b) P AT THE HIGHEST POINT OF THE TRAJECTORY

(20) = 20 = -95

(a) HAVE.. (an) no PA 3) 2

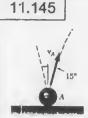
OR PA = (50 3) 2

(9.81 32) cos 25

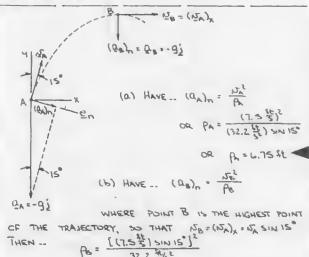
(08)" " " " OS by = 581 m

WHERE POINT B IS THE HIGHEST POINT OF THE TRAJECTORY, SO THAT US = (UT) x . UT COS 25"
THEN ... PS = ((50 P) cos 25")

DR PB = 209 m

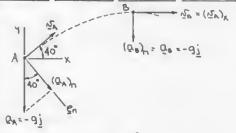


GIVEN: NT = 7.5 5 FIND: (a) P AT POINT A (b) P AT THE HIGHEST POINT OF THE TRAJECTORY



11.144

GIVEN: A. B.Sm FIND: (a) NA (b) P AT THE HIGHEST POINT OF THE TRAJECTORY



(QA) = 24 (a) HAVE .. OR 15,2 = (9.81 cos 40)(8.5 m)

(b) HAVE ..

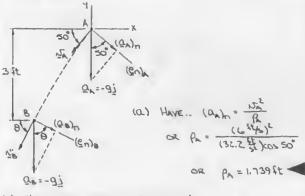
WHERE POINT B IS THE HIGHEST POINT OF THE TRAJECTORY, SO THAT 5 = (5) = 5 cos40 Ps = (63.8766 3/2) cos240

OR P = 3.82 m

11,146

GIVEN: NT . 6 5 FIND: LOT P AT POINT A (b) P AT THE FOINT ON THE TRAJECTORY 3 St BELOW A

OR P6 = 0.1170 St



(b) HORIZONTAL MOTION (UNIFORM) (JB) = (JA) = (6 5) cos so = 3.8567 5 VERTICAL MOTION (WIFE, ACCEL, MOTION)

HAVE... $S_{4}^{2} = (J_{5})_{4}^{2} - 2g(y-f_{5})$ WHERE $(J_{5})_{4} = (J_{5})_{5}^{2} + 4.5963 \frac{4}{5}^{2}$ AT POINT B, y = -34: $(J_{5})_{4}^{2} = (4913 \frac{4}{5})_{-2}^{2}(32.2 \frac{4}{5})(-34)$ THEN... No = (1/2) + (1/2) = (3.85) + (14.6393) = 15.1394 \$\frac{11}{5}\$

AND TRUB = \(\frac{14}{5} \frac{1}{5} + (1/2) \frac{1}{5} \\
\frac{14}{5} \frac{14}{5} \\

11.146 CONTINUED

(as) = 1502 Now .. (15.1394 ft)2 Pe = (52.2 1/2) cos 75.24°

OR PG = 27.9 ST

11.147





HAVE .. (as) = R

WHERE (as) = as ws 8 = 9 cos 8

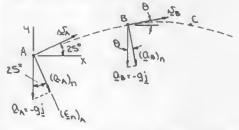
NOTING THAT THE HORIZONTAL MOTION IS UNIFORM.

(UB)x = NA HAVE .. (UB) , NB COSB - Beas :

11,148



GIVEN: 5 = 20 5 FIND: IS AT THOSE POINTS WHERE P= 7PA



ASSUME THAT POINTS B AND C ARE THE POINTS OF INTEREST, WHERE YE = YC AND No. = No. Now...

PA= Q coses

Po = + Ph = + gcos 250 THEN

(WATHUED)

11.148 CONTINUED

HAVE (as) = 52 WHERE (a.) = 9 cos 8 3 NE2 9 WS 250 TAHT OC OR 50 = \$ (058 NA

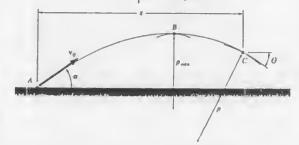
NOTING THAT THE HORIZONTIAL MOTION IS UNIFORM, (NA)x = (NB)x HAVE __ (NA) = NA COS 25° (NB) = NB COS B WHERE Ny cos 25 = 15 cos 8 DR WIDE BE COS 25°

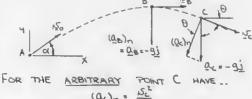
SUBSTITUTING FOR COST IN EQ. (1) HAVE .. NB = 4 (NA COS 25°) NA NB = 3 NA = 3 (20 5) OR NB = NE = 18.17 5

11.149

GIVEN: THE INITIAL VELOCITY IS AND THE TRAJECTORY OF THE PROJECTILE AS SHOWN

SHOW: (a) PB = PMIN, WHERE YB = YMAX (b) Pc = Pmin/cos 30





OR P= NE

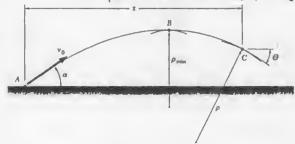
NOTING THAT THE HORIZONTAL MOTION IS UNIFORM, HAVE -(Ja) = (Jc) WHERE (NA) = NO COSA (NE / = NE COSB THEN Becogn = NE COSB NE = COSK No

SO THAT R= 90000 (000 No) = 30 costa (a) IN THE EXPRESSION FOR P. 15, K. AND 9 ARE CONSTANTS, SO THAT A IS MINIMUM WHERE COS B IS MAXIMUM. BY OBSERVATION, THIS OCCURS AT POINT B WHERE 0=0. PMIN : PB = No COSTA

(P) 65 = (000 (120 (00)) Pc = cos & Q.E.D. 11.150

GIVEN: THE INITIAL VELOCITY IS AND THE TRAJECTORY OF THE PROJECTILE AS

FIND: PE IN TERMS OF X, J. a. AND Q







OR R 9 COSB

NOTING THAT THE HORIZONTAL MOTION IS UNIFORM, (N/) = (NE) X= X0+(N5) x = W5 cosa) t HAVE RECO ON = XAZI) GREY = NE COSB 15 COSK = 4 COSO AND (15) . So COSK (1) THEN x eco 2 = 0 eco

PE : Q NO WOOK SO THAT

FOR THE UNIFORMLY ACCELERATED VERTICAL MOTION HAVE (NEX = W= 1 - 9t = N= SINA - 9t

FROM ABOVE -. X= (No COSK) t OR t = No COSK (JE) = NJ SINA - 9 NJ COSA (2)

15 = (15) = + (15) =

SUBSTITUTING FOR (NE), [EQ. (1)] AND (NE), [EQ.(2)] E= (No cosk)2+ (No SINX - 9 No cosx)

= Not (1- 29x TAND + 21x2 Note)

CR 5 = 50 (1- 29x TANK + 32x2)/2

FINALLY, SUBSTITUTING INTO THE EXPRESSION FOR R, OBTAIN --

P = \frac{1000000 (1- 29XTANK) \frac{92x2 3/2} 1000000

* 11.151

GIVEN: C = (Rt cos wit) + (tj+(Rt SINWHT)) FIND: P AT 1=0

HAVE. IT = of = R (coswit - wit sinunt) i + c) + R(sIN WHT + WHI COS WHIT) & NAS - Q = de = R(-Un sinuat-un sinuat-unit cosunt) +R(w) cos whit + wn cos whit - wit & sin whit) he

(CONTINUED)

11.151 CONTINUED

or a = whele(s since + who concent) i (Z Coscont-watsin wat) +

15 = R2(cosant-untsinunt)2 + C2 + R2 (SHUNT+WAT COS WAT)2

= R2(1+ m2 + 1) + c2

N= [R2 (10 white) + C2] 1/16 dr = R2 white 9E = [Bs(1+04fs)+Cs)1/5

02 - 05 + 00 = (3/2) 5 + (1/2) 5

AT t=0: 2 =0

a = WAR(2B) OR a= ZWAR 152 = R2+ C2

THEN, WITH SEO, HAVE .. a. p

* 11.152

GIVEN: [= (At cost) = (A Hi+1); + (Bt swt)k, 1-12, 1-5;

A=3, B=1 FIND: P AT LO

WITH A=3 B=1 HAVE ..

c - (3t cost) + (3 (t2+1)) + (t sint) }

Now. 15 = of = x cost - t smt) = + (1227)]

+ (sixt + t cost) & MD. B = # = 3(- sint - sint - 2 cost) = 3 (#201 - 1 (#201)) +(cost+cost-t sint)k

=-3(251Nt+2cost) 1+3 (+1) 1782

+ (2cost - t sint) ! + (sint + t cost) ?
THEW - N° = 9(cost - t sint) + 9 + 12+1 + (sint + t cost) ?

Exprissing and SIMPLIFYING MELOS.

THEN 5=[t4+19221+B(cos2++t4 sin2+)-B(t2+2) sin2+]12

dis 4t3+38t+81-2cost sint+423 in t+2t2 in cost)-8(19t2-1) smatally typost Z[t+19t2+1+8(cos2++++sin2+)-8/23++) sin2t]/2

Now .. a= a+ an = (dx)+ ()

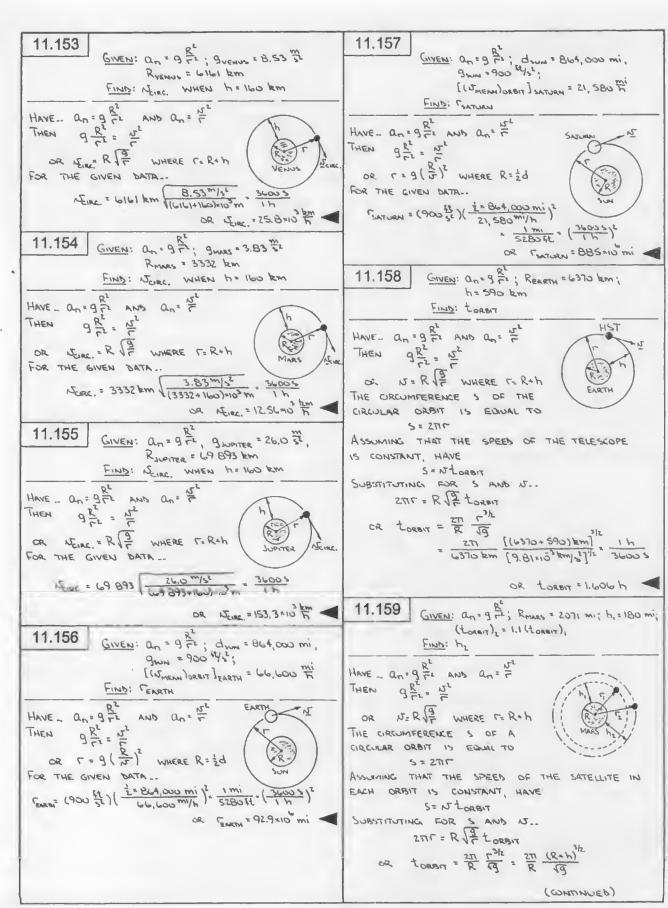
AT 2=0: Q=31+28 08 Q= 113 52

25 0 (2)

THEN, WITH SE = 0 HAVE .. a = P

6. 13 th

or p= 2.50 ft



11.159 CONTINUED

Now.. $(t \cos \alpha)_2 = 1.1 (t \cos \alpha)_1$ or $\frac{2\pi}{R} \frac{(R + h_2)^{3/2}}{\sqrt{9}} = 1.1 \frac{2\pi}{R} \frac{(R + h_1)^{3/2}}{\sqrt{9}}$ or $h_2 = (1.1)^{2/3} (R + h_1) - R$ $= (1.1)^{2/3} (2071 + 180) mi - (2071 mi)$ or $h_2 = 328 mi$





GIVEN: an = 9 to; ha = 120 mi;
ha = 200 mi; REARTH = 3960 mi;
at t=0, A and B
ALIGNED AS SHOWN
FIND: t WHEN A AND B ARE
NEXT RAWALLY ALIGNED

THEN 3 ET = TO OR WERE TORAL

THE CIRCOMFERENCE S OF A CIRCULAR ORBIT IS

EQUAL TO S= 2717

ASSUMING THAT THE SPEEDS OF THE SATELLITES

ARE CONSTANT, HAVE

SUBSTITUTING FOR S AND J.

or token or
$$\frac{2\pi}{R} \frac{r^3k}{49} = \frac{2\pi}{R} \frac{(R+h)^{3k}}{49}$$

NOW he sha => (torent) &> (torent) a

NEXT LET TIME TOTAL BE THE TIME AT WHICH THE

SATELLITES ARE NEXT RADIALLY ALKENED. THEN, IF

IN TIME TOTAL SATELLITE B COMPLETES N

ORBITS, SATELLITE A MUST COMPLETE (N+1) ORBITS.

THUS,

Hus,

$$t_{TOTAL} = N(t_{ORBIT})_B = (N+1)(t_{ORBIT})_A$$
 $N\left[\frac{2\pi}{R}\frac{(R+h_B)^{3/2}}{(Q}\right] = (N+1)\left[\frac{2\pi}{R}\frac{(R+h_A)^{3/2}}{Q}\right]$
 $N\left[\frac{2\pi}{R}\frac{(R+h_B)^{3/2}}{(R+h_A)^{3/2}}\right] = \frac{(R+h_B)^{3/2}}{(R+h_A)^{3/2}} = \frac{(R+h_B)^{3/2}}{(R+h_A)^{3/2}} = \frac{1}{(R+h_B)^{3/2}}$
 $= \frac{1}{(39k_B+200)^{3/2}} = 33.835$ ORBI

$$= \frac{(3960+200)^{3/2}}{(3960+120)^{3/2}} = 33.035 \text{ ORBITS}$$

$$= 27 (R+h_0)^{3/2}$$

$$= 27 (R+h_0)^{3/2}$$

THEN -
$$t_{TOTAL} = N (t_{ORBIT})_8 = N \frac{27!}{R} \frac{(R+h_0)^{3/2}}{\sqrt{9}}$$

$$= 33.83S \frac{27!}{3960 \text{ mi}} \frac{[(3960+200)\text{mi}]^{3/2}}{(32.2) \frac{(t_1 \text{ mi})^{1/2}}{52.80 \text{ ft}})^{1/2}}$$

$$\times \frac{1}{3400} \frac{h}{5}$$

OR trongs 512h

ALTERNATIVE SOLUTION
FROM ABOVE HAVE

FROM ABOVE HAVE (LOBOT) > (LOROT) A
THUS, WHEN THE SATELLITES ARE WEST RADIALLY
ALIGNED, THE ANGLES DA AND DO SWEDT OUT
(CONTINUED)

11.160 CONTINUED

BY RADIAL LINES DRAWN TO THE SATELLITES MUST DIFFER BY 271. THAT IS, $\theta_{A}=\theta_{B}+2\pi$

FOR A CIRCULAR ORBIT .. S= TO
FROM ABOVE .. S= NE AND No R 17

THEN B= = 1 (R) + - RG + RG (R+h) > E

AT TIME FROM : (R+ha) Petrone . (R+ha) 1/2 trone + 211

" 1 h (3960+120) mi] 42 (3960+200) mi] 42

OR tromy . SI. 2 h

11.161

GIVEN: $C=3(2-\frac{1}{2})$, $\theta=4(t+2e^{\frac{1}{2}})$ C=M, t=9, $\theta=RAS$ FIND: (a) I AND Q AT t=0(b) I AND Q AS t=0; THE

FINAL PATH OF THE PARTICLE

HAVE.. $r = 3(2 - e^{-t})$ $\theta = 4(t \cdot 2e^{-t})$ THEN $r = 3e^{-t}$ $\theta = 4(1 - 2e^{-t})$ AND $r = -3e^{-t}$ $\theta = 8e^{-t}$

Nova.. N= rerribes = 32 2-+12(2-2+x1-20+)es

ANI) Q = (r-rb)2-+(rb-2+b)20

-[-30-4812-2-x1/-20-1]2-

+[24(2-et)e-t - 24e-t (1-2et)]e

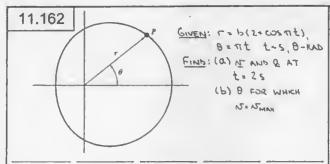
(a) AT t=0: \$\vec{a} = 3\vec{e}_{\tau} + 12(2-1)(1-2)^2\vec{e}_{\tau}\$

\[
\text{Q} = \left[-3-48(2-1)(1-2)^2\vec{e}_{\tau} = \left[-3\vec{e}_{\tau})\vec{e}_{\tau} = \left[-3-48(2-1)(1-2)^2\vec{e}_{\tau}\vec{e}_{\tau}\)

(P) y? f →00: N= (0) 6+ 15(5-0)(1-0) 60 OK 0= (21 25) 64 OK 0= (21 25) 64

Ö = [0-48(2-0)(1-0)] € + (0+0) € 8

AS LOO, TO 6 M. A CONSTANT. THUS, THE FINAL PATH IS A CIRCLE OF RABIUS 6 M. NOTE THAT THE SPEED OF THE PARTICLE IS CONSTANT (24 %); THUS, THE TRANSVERSE (TANGENTIAL) COMPONENT OF THE ACCELERATION IS REW.



HAVE... $\Gamma = b \cdot (2 + \cos \pi t)$ $\theta = \pi$ THEN $\dot{\Gamma} = -\pi^2 b \cos \pi t$ $\dot{\theta} = 0$ NOW... $\dot{\Delta} = \dot{\Gamma} = -\pi^2 b \cos \pi t$ $\dot{\theta} = 0$ AND $\dot{\alpha} = (\ddot{\Gamma} - \dot{\Gamma} \dot{\theta} = 0) = -(\ddot{\Gamma} \dot{\theta} + 1) = 0$ $= (\ddot{\Gamma} - \dot{\Gamma} \dot{\theta} = 0) = -(\ddot{\Gamma} \dot{\theta} + 1) = 0$ $= (\ddot{\Gamma} - \dot{\Gamma} \dot{\theta} = 0) = -(\ddot{\Gamma} \dot{\theta} + 1) = 0$ $= (\ddot{\Gamma} - \dot{\Gamma} \dot{\theta} = 0) = (\ddot{\Gamma} \dot{\theta} + 1) = 0$ $= -2\pi^2 b \cdot (1 + \cos \pi t) = -(\sin \pi t) = 0$ $= -2\pi^2 b \cdot (1 + \cos \pi t) = -(\sin \pi t) = 0$

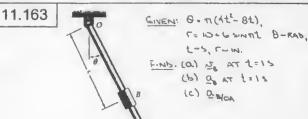
(a) At t=25: 2=-(0)e+76(2+1)e0

OR 1 = 371600

Q=-2726[(1+1)er+(0)eg]

(b) HAVE... $N = \pi b \sqrt{(-\sin \pi t)^2 + (2 + \cos \pi t)^2}$ = $\pi b \sqrt{5 + 4\cos \pi t}$ $\theta = \pi t$ = $\pi b \sqrt{5 + 4\cos \theta}$

By observation, it is smax when $\cos\theta = 1$ or $\theta = 2\pi\pi$, n = 0, 1, 2, ...



B=71(422 BE) HAVE. T: 10.6 SINTIL + = 67 cosni 8 = BTI (1-1) THEN ל ב- בח אמחל 118 + B AT 1=15: T=10 IN. 8 = - 4TT RAD 0 = 0 B= BT EAD/SE F = 0 (a) HAVE. 5- FER + TOEB 15 = - (PL) = 15-SO THAT (b) HAVE .. Q = (= - 02) e + (= + 2 + 0) e \$ (10)(BT) ED or a = (800 52)en 9 = - 4TI RAB (C) HAVE .. aspa= " SO THAT ab/04 = 0 4 11.164

GIVEN: To tot, B = T SINTE

B - RAD, tot, Tolk

FIND: (a) Us AT tolk

(b) Q a AT tolk

(c) Q alia AT tolk

A

HAVE.. $C = \frac{25}{2.4}$ THEN $C = \frac{25}{(2.44)^2}$ AND $C = \frac{50}{(2.41)^2}$

AT t=15: T=5 IN. B=0

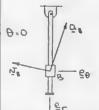
i=-1 IN 15

i=04 IN 152

i=0

(a) HAVE .. N= 18-+1968 - (-1)6-+ (>)+2)68

(b) HAVE .. Q= ("- - P2) = - (- P+ z+ P) = 0 = [0.4 - (5)(2)] = - + [0+2(-1)(2)] = 0 OR Q= - (19.6) = 0 | (4) | 100



(C) HAVE abox = "
SO THAT abox = (04 12)er

11.165

CHIVEN: $\Gamma = \frac{2}{2 - \omega_{STR}}, \ \Theta = \pi t$ $\Gamma = m, \ t = s, \ \Theta - RAB$ $\Gamma = m, \ t = s, \ t = s, \ \Theta - RAB$ $\Gamma = m, \ t = s, \ t = s,$

HAVE.. $\Gamma = \frac{2}{2 - \cos \pi t}$ $\theta = \pi t$ THEN $\dot{\Gamma} = \frac{2\pi \sin \pi t}{(2 - \cos \pi t)^2}$ $\dot{\theta} = \pi$

AND $i' = -2\pi \frac{\pi \cos \pi t}{(z - \cos \pi t)^3}$ $\ddot{\theta} = 0$

(2-cosnt) = -272 2 cosnt -1 - 512 nt

(a) ht t=0: r=2 m B=0

i=0

i=0

i=-272

i=0

Now.. J=ie+rbea = (2)(17)ea

NOW. \$\bar{U} = (\bar{C} + (\bar

= [-sug-(s)(u)]=-

OR Q = -(4T/2 T/2) Pr

0=0

11.165 CONTINUED

(b) AT \$\frac{1}{2} = \frac{1}{2} \frac{1}

C = 0 = 0 = 0 = 0

NOW. 12 = 12- + 19EB = (-1)E+ (1/1/1)ED

OR 12 = - (2 m/) E - + (11 m/) E8

AND.. $\underline{\alpha} = (\ddot{r} - r\dot{\theta}^z)\underline{e}_r + (r\ddot{\theta} + 2\dot{r}\dot{\theta})\underline{e}_{\theta}$ $= [\frac{m^2}{2} - (1)(\pi)^z]\underline{e}_r + [2(-\frac{7}{2})(\pi)]\underline{e}_{\theta}$

80 E S RAD EL DE E

11.166

GIVEN: (= 20 COSO, O= 2 bt2

(b) p; PATH OF THE PARTICLE

(a) Have. $\Gamma = 2a \cos \theta$ $\theta = \frac{1}{2}bt^2$ Then $\dot{\Gamma} = -2a\dot{\theta} \sin \theta$ $\dot{\theta} = bt$ AND $\dot{\Gamma} = -2a(\dot{\theta} \sin \theta + \dot{\theta}^2 \cos \theta)$ $\ddot{\theta} = b$ Substituting for $\dot{\theta}$ AND $\ddot{\theta}$

i = - 2abt sin 0 i = - 2ab(sin 0 = bt2 ws0)

Now.. $S_r = \dot{r}$ $S_r = \dot{r}$

THEN .. W = \ 152 + 152 = 2abt[(-sin B)2 + (cos B)2]12

ALSO.. $Q_r = \ddot{r} - r\dot{\theta}^2 = -2ab(\sin\theta + bt^2\cos\theta) - 2ab^2t^2\cos\theta$ = -2ab(\sin\theta + 2bt^2\cos\theta)

AND $a_{\theta} = r\theta + 2r\theta = 2abcos\theta - 4ab^{\dagger}t^{\dagger}sin\theta$

= Sap (cord-spt sing)

THEN. a = 1 at + a = 2 cos 0) 2 (cos 0 - 2 bt 2 sin 0) 2) 1/2 cos 0 - 2 bt 2 sin 0) 2) 1/2 cos 0 - 2 bt 2 sin 0) 2) 1/2 cos 0 - 2 bt 2 sin 0) 2) 1/2 cos 0 c

(b) Now. $a^2 = a_1^2 + a_n^2 = (\frac{4\pi}{4})^2 + (\frac{\pi^2}{p})^2$

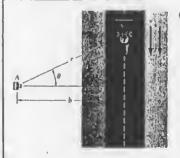
THEN -. $\frac{dy}{dt} = \frac{dt}{dt}(2abt) = 2ab$ SO THAT $(2ab\sqrt{1+4b^2t^4})^2 = (2ab)^2 + a^2$ OR $4a^2b^2(1+4b^2t^4) = 4a^2b^2 + a^2$ OR $a_n = 4ab^2t^2$

NALLY -- an = 52 => p = (2abt)2 + abtt

SINCE THE RADIUS OF CURVATURE IS A CONSTANT, THE PATH IS A CIRCLE OF RADIUS Q.

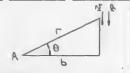


11.167 and 11.168



GIVEN: THE RECTILINEAR
MOTION OF A RACE
CAR AS SHOWN

HAVE.. $\Gamma = \frac{b}{\cos \theta}$ THEN $\dot{\Gamma} = \frac{b}{\cos^2 \theta}$

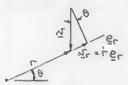


1.167 FIND: IS IN TERMS OF b, B, AND B

HAVE. $U^2 = U_1^2 + U_2^2 = (\dot{r})^2 + (\dot{r}\dot{\theta})^2$ $= \left(\frac{b\dot{\theta} \sin \theta}{cos^2\theta}\right)^2 + \left(\frac{b\dot{\theta}}{cos\theta}\right)^2$ $= \frac{cos^2\theta}{cos^2\theta} \left(\frac{\sin b^2\theta}{cos^2\theta}\right)^2 + \frac{b\dot{\theta}}{cos^2\theta}$ OR $U = \frac{b\theta}{cos^2\theta}$

FOR THE POSITION OF THE CAR SHOWN, 8 IS DECREASING. THUS, THE NEGATIVE ROST IS CHOSEN. 6

ALTERNATIVE SOLUTION



FROM THE DIAGRAM.. $\vec{r} = -15 \sin \theta$ OR $\frac{69 \sin \theta}{\cos^2 \theta} = -15 \sin \theta$ OR $\frac{69}{\cos^2 \theta} = \frac{69}{\cos^2 \theta}$

11.168 FIND: Q IN TERMS OF D, D, D, AND B

FOR RECTILINEAR MOTION Q = dt FROM THE SOLUTION TO PROBLEM 11.167

THEN $\Omega = \frac{d}{dt} \left(-\frac{\partial^2 C}{\partial t^2} \right) = -\frac{\partial^2 C}{\partial t^2} - \frac{\partial^2 C}{\partial t^2} - \frac{$

OR $Q = -\frac{b}{\cos^2\theta} (\ddot{\theta} + 2\dot{\theta}^2 \tan \theta)$

FROM ABOVE -- To COSO TO BOSIND

THEN. = 6 = 6 (BSIND+ 6 COSD) (- 10 SIND) (-26 COSD + 100)

MOINT $Q_1 = Q_2 + Q_3$ $= \frac{cos_3\theta}{cos_3\theta} \left(\frac{1}{2} e^{iin}\theta + \frac{1}{2} \frac{cos_3\theta}{cos_3\theta} \right) - \frac{1}{2} \frac{1$

11.168 CONTINUED

$$\alpha_{\Gamma} = \frac{1}{\cos^{2}\theta} \left(\ddot{\theta} + 2\dot{\theta}^{2} + 2n\theta \right)$$

$$\alpha_{N} = \frac{1}{\cos^{2}\theta} \left(\ddot{\theta} + 2\dot{\theta}^{2} + 2 \frac{b\dot{\theta}^{2} + 2n\theta}{\cos^{2}\theta} \right)$$

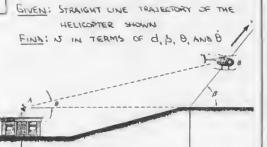
$$= \frac{1}{\cos^{2}\theta} \left(\ddot{\theta} + 2\dot{\theta}^{2} + 2n\theta \right)$$

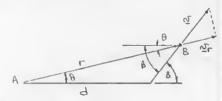
$$= \frac{1}{\cos^{2}\theta} \left(\ddot{\theta} + 2\dot{\theta}^{2} + 2n\theta \right) \left[(\sin\theta)^{2} + (\cos\theta)^{2} \right]^{1/2}$$
THEN $\alpha = \frac{1}{2} \frac{1}{\cos^{2}\theta} \left(\ddot{\theta} + 2\dot{\theta}^{2} + 2n\theta \right) \left[(\sin\theta)^{2} + (\cos\theta)^{2} \right]^{1/2}$

FOR THE POSITION OF THE CITY SHOWN, B IS NECATIVE; FOR Q TO BE POLITIVE, THE NEGATIVE ROOT IS CHOSEN.

(a=- b (6+262 mul)

11.169





FROM THE DIAGRAM ..

OR dsinB = (sinBcos0 - cospsinB)

r = dtang - (-TANBSINB-COSB) THEN (THUBCOSB - SINB)EB = dotang TANBSIND+cosb

FROM THE DIAGRAM

THEN $d\dot{\theta} \tanh \beta \frac{T + n \beta \sin \theta + \cos \theta}{(T + n \beta \cos \theta - \sin \theta)^2} = 15(\cos \beta \cos \theta + \sin \theta)$

> OR N= do TANB SECB (TANBCOSD-SIND)2

= NCOSB (TANBSIND+COSD)

ALTERNATIVE SOLUTION

HAVE .. N2 = N2 + NB = (1)2 + (18)2

USING THE EXPRESSIONS FOR F AND ' FROM ABOVE ..

NZ= [dBTANB TANBSIND+COSB]2

+ (do TANB CON)2

(CONTINUES)

11,169 CONTINUED

NOTE. THAT AS B INCREASES, THE HELICOPTER MOVES IN THE INDICATED DIRECTION. THUS, THE POSITIVE ROUT IS CHONEN.

: N= do TANB SECB

* 11.170



GIVEN: No = CONSTANT FINS: B IN TERMS OF NJ, h. B. AND B

FROM THE DIAGRAM .. 51N(70-B) = 51N(A-B)

OR (SINB COS B + COSASINB) = housB

OR T= THUB COS O + SIN O

ALSO --No = No 51N (B+B) WHERE IS = FB

THEN TRUBCOSO + SIND = No (SINFCOSO + COSP SIND)

DR 0 = 15 COSB (TAMBCOSB + SINE)

ALTERNATIVE SOLUTION

FROM AROVE .. TO TAN BOOSE + SIND

6 = h TTAN 2 SUB- COSO 2 6

Now .. It = 15 + 15 = (1) + (18)

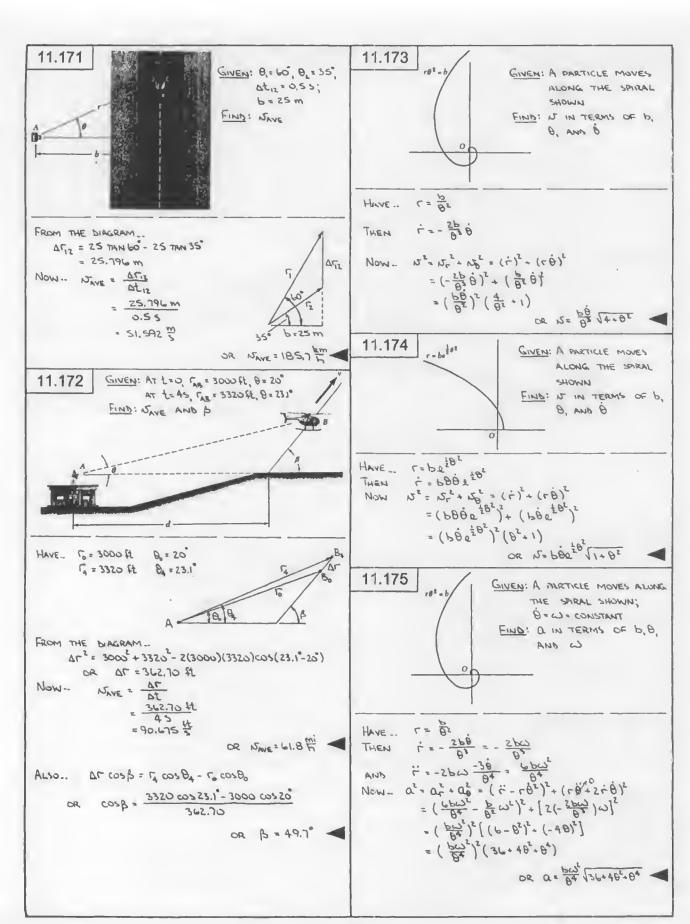
No = [ho TANBEND - COSO) + (TANBEDS + SIND)

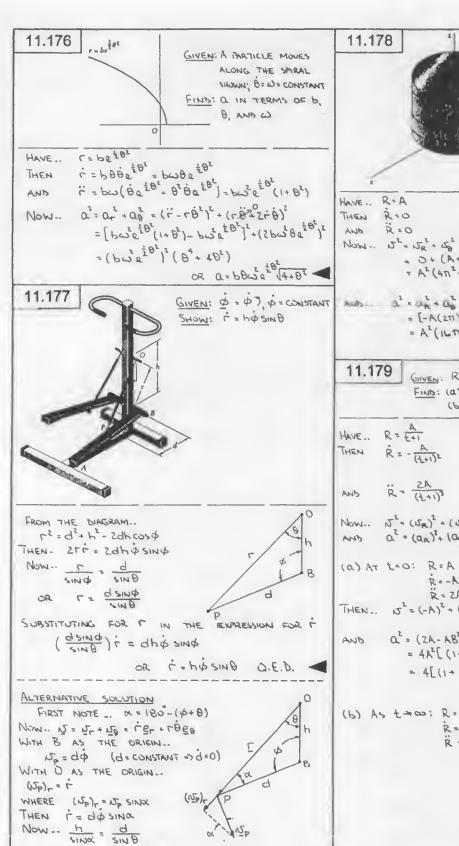
OR No = + TANBODO 8 + SAND (TANBOND + COSD) + 1] 1/2

= + TANBOOD (TANBOD)

NOTE THAT AS & INCREASES, MEMBER BC MOVES IN THE INDICATED DIRECTION. THUS, THE POSTIVE ROO'T IS CHOSEN.

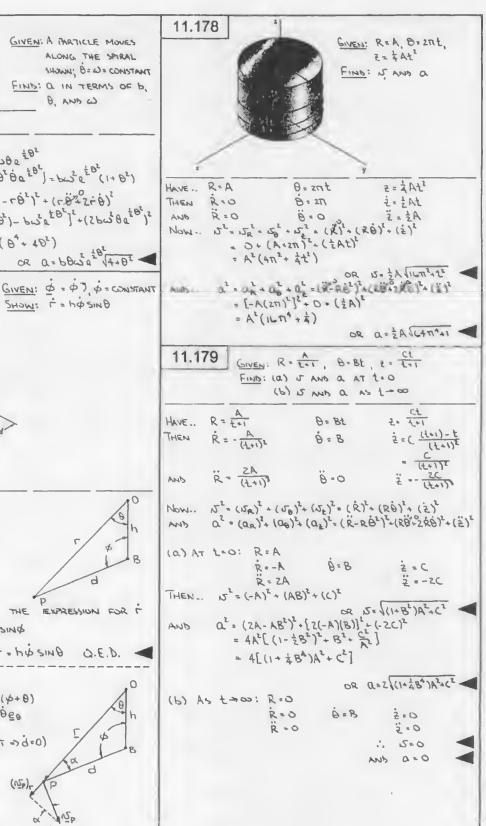
: 0 = 10 cosB (THIS COSO + SIND)2





Q.E.D.

DR dSINX = hSINB THEN .. F = HOSIND



11.180

GIVEN: [= (Rt cosult) + ct ; + (Rt sing t) } FIND: THE ANGLE THAT THE OSCULATING PLANE FURMS WITH THE Y AXIS

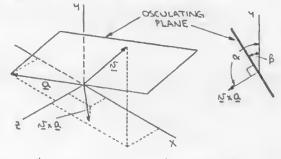
FIRST NOTE THAT THE VECTORS IS AND Q LIE IN THE OSCULATING PLANE.

Now. I = (Rt coscint) i + cti + (Rt sincht) & 5= dr = R(cosunt - unt sinult) i + ci +R(smant = witcosunt) &

 $Q = \frac{d\vec{x}}{dt} = R(-\omega_n \sin \omega_n t - \omega_n \sin \omega_n t - \omega_n^2 t \cos \omega_n t) \underline{i}$ +R(ch coscht + whosent-wat smult)k

= WAR[-(2514 WILL COSWAT)i (# (z cos wat - wat sincut) + [

IT THEN FOLLOWS THAT THE VECTOR (N=0) 15 PERPENDICULAR TO THE OSCULATING PLANE.



(Uza) = WIR R(cosunt-cut sinult) (R(sinult-cut cosult) (thousethortosops) o (thousethortones)

= WRIC(2005Wit-WitsINWIt)i +R[-(sincut+watcoswit)(2 sincut+watcoswit) - (cos unt - white must) (zeosont-unternount)] +C(ZSHWAT+WATCOSWAT)E)

= WR[c (2005 wat - out sindut) = R(2+ wit2); + C(2514 WAT + WAT COS WAT) &]

THE ANGLE & FORMED BY THE VECTOR (U.Q.) AND THE 4 AXIS IS FOUND FROM ..

WHERE 121=1 (15,0). i = - W/R2(2+W/t2) 1(2,01) = 018(c,(5000+-0420004),+ 65(5+07,5),

+ c2(25, 12, 12 + 22,

- WNR((2+ Wh t2) WNR((2+ Wh t2)+R2(2+ Wh t2)2)"/2 THEN

-R(2+6,2t2) [c2(4+6,2t2)+R2(2+6,2t2)2]12

(CONTINUED)

11.180 CONTINUED

THE ANGLE B THAT THE OSCULATING PLANE FORMS WITH THE Y AXIS (SEE THE ABOVE DIAGRAM) IS EGNAL TO

B= 04-90 THEN COSK = COS (B+90) = - SIND

- R(2+W2+2) : - SINB = [(2(4+4)22)+R2(2+4)222) 1/3

R(ZOLDET) THEN TAND = R(ZOLDETE) OR BETAN' (RIZHUNT)

* 11.181

GIVEN: [= (At cost) +(A (tial) + (Bt sint) } 5- ft, t-s; A=3, B=1

FIND: (a) DIRECTION OF EL AT t=0 (b) DIRECTION OF CO AT to 25

FIRST NOTE THAT QL IS GIVEN BY

Now .. [= (3t cost) + (3\t2+1) 1 + (t sint) }

THEN is = dr = 3(cost - t sint) = 30

+ (sint+tcost) +

Q = AF = 3(-SINT-SINT-tcost) 143 (teat-t(AFE))

+ (cost + cost - t sint) k =-3(2 sint + toost) + (++1) > 1

+ (zcost-t sint) k

(a) AT 1=0: 5= (33) Q= (35) + (25) }

THEN 15,0 = 31 x (31+2k) = 3 (-2) +3 5)

15=01=3 1(-2)2+(3)2 = 3113

THEN Q = 3(-2)+3 = 15(-2)+3 B)

 $O \qquad cos \theta_1 = -\frac{2}{\sqrt{3}} \qquad cos \theta_2 = \frac{3}{\sqrt{3}}$ $\theta_3 = 90 \qquad \theta_4 = 123.7 \qquad \theta_2 = 33.7 \qquad \theta_3 = 33.7 \qquad \theta_4 = 33.7 \qquad \theta_5 = 33.7 \qquad \theta_6 = 33.7 \qquad \theta_7 = 33.7 \qquad \theta_8 = 33.7 \qquad \theta_8 = 33.7 \qquad \theta_8 = 33.7 \qquad \theta_9 =$ OR

(b) AT t= 25: N= - (3 5) + (3 5) + (15 5) + (15) = Q = - (6 5) = + [24 (7 6 4) 3/6 5/6] j - (7 5/6)]

THEN .. IZ a = - 2TT

 $= -\left[\frac{3\pi^2}{2(\pi^2+4)^{4/2}} + \frac{24}{(\pi^2+4)^{6/2}} \right] \frac{1}{2} - \left(6 + \frac{3\pi^2}{4} \right) \frac{1}{2}$ +[- 367 + 187 1/2] k

= -4.4398+1 - 13.40220+ 12.99459 k

0x = 103.4° 0y=134.3° 0= 47.4°

NO 15 = [(-4.4398+)2+(-13.40220)2 (12.994 59)2]12 = 19.188 29

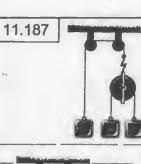
THEN -- Gb = 19.18829 (-4.439841-13.40220)+12.994594)

: cosbx = - 4.439 84 cosbx = - 13.402 20 cosb = 12.994 59

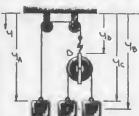
52

OR

11.182 11,184 CONTINUED GIVEN: X = 2t3-15t2+24t+4 X-m. t-5 FIND: (a) t WHEN N=0 Q=-k(5-kx) OR AT 1:0: Q = - 2700 5 (900 5 -0) (b) X AND TOTAL DISTANCE OR 0 = - 2.43 = 10 11 TRAVELED WHEN Q = 0 (b) HAVE # # N = N = KX AT t=0, x=0: 1 dx = 1 dt HAVE .. X= 2t3-15t2-24t+4 N= dx = 6t2-30t+24 a= dx = 12t-30 THEN or - | [LNIN- kxi] = t MIN OR t= 1 LN (No-PX) = 1 LN (1- FX) (a) WHEN N=0: 62-301+24=0 WHEN X= 3.9 IN .: t = 2100 15 LN 1 - 2100 15 OR (1-1)(1-4)=0 OR t=15 AND t=45 (b) WHEN Q=0: 122-30=0 OR t=2,55 OR 1=1.36 mis 5 AT t = 2.55: X2.5 = 2(2.5)3-15(2.5)2+24(2.5) +4 11.185 GIVEN: N= = 6 31; AT 1=0, 4, -4, =0; FOR OR X25=1.5 m FIRST OBSERVE THAT OSECIS NOO t = 45, Sp =0; Qp = 2,45/521 15< £ \$ 2.5 \$ 5 < 0 FIND: (a) t AND Y WHEN YE = YO Now. AT t=0: X0 = 4 m t=15: X1 = 2(1)3-15(1)2+24(1)+4=15 m (b) No WHEN YE = YP (a) For t20: 4= (47) + 5= t 1245: 4= (47) + (47) (1-4) 4 2 ap (1-4) 1.5m WHEN YF = 40 - - 50 (1-4)2 (2.55) (0) THEN .. X1-X0 = 15-4 = 11m EXPANSING AND SIMPLIFYING .. 1x25-X1= 1.5-15 = 13.5 M t'-13++16=0 ". TOTAL DISTANCE TRAVELED= (11+13.5)m=24.5m SOLVING .. t = 1.3765 5 AND t=11.6235 5 MOST REDURE to 45 1. t=11.625 11.183 AT t = 11.6235 5: 4 = (6 5)(11 6235 5) GIVEN: Q=-60x 1.5 Q- 53 X-M; AT 1=0 OR 4=4p=69.7 St 5=0, X=4m (b) FOR t=45: No = (NF) - ap(t-4) FIND: (a) N WHEN X= 2 M (b) N WHEN X=1 m AT t=11.6235 >: No= (2.4 5)(11.6235-4)5 (C) I WHEN X= 0.1 m OR No = 18.305 N dx = Q = - 60x1.5 11.186 HAVE .. GIVEN: UZ = 150 MM WHEN X=4m, 15=0: Jourdus = Jx (-60x-1.5)dx FIND: (Q) JA (P) N.C OR 152 = 120[x-0.3]X (C) 200 OR N3 = 240 (1 - 1) (a) WHEN X=2 m: 152= 240(2-2) OR 15:-7.05 % (a) FROM THE DIAGRAM S= 240(1- 2) (b) WHEN X= 1 m: HAVE .. UR N=-10.95 \$ (XA-XB)+(-XB)+2(-XA) (C) WHEN X=0.1 m: 52=240(101-1) · CONSTANT OR N=-25.3 5 THEN .. NA + 2NB = 0 SUBSTITUTING .. 11.184 GIVEN: No No - KX No 5 X- St; AT 1=0 NA+2(-150 MM)=0 X=0. 5. 900 5; WHEN X=4 IN. -Xc-OR 1 = 300 5 5=0 (b) FROM THE DIAGRAM HAVE .. (XA-XB)+(XC-XB)= CONSTANT FIND: (a) a AT t.O (b) & WHEN X = 3.9 IN. THEN .. NA - 258 + 5 = 0 SUBSTITUTING .. 300 7 - 2(-150 7)+ 4 = 0 FIRST NOTE .. WHEN X= 12 St. N=0.0=(900 \$)-k(2 12) OR NE = 600 5 DR & = 2700 \$ (C) HAYE -- UC/B = LE - NB - (-150 5) (a) HAVE -- N=NJ-KX a= 4= 4 (2- xx)=- kv OR NE = 450 5 (CONTINUED)



CIVEN: NA, NB, NZ CONSTANTS; FIND: IJA, NB, AND NE



FROM THE DIAGRAM ... CABLE 1: 4A+45 = CONSTANT THEN. UA+Ub=0 (1) (ABLE 2: (48-40)+ (4c-40) THATZHOD =

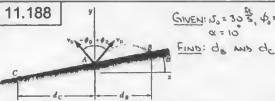
THEN. NB+NC-2ND=0 (2) COMBINING EQS. (1) AND (2) TO ELIMINATE No .. (3)

Now - NAIC = NA - NE = - 300 mm (4) NB/A = NB - NA = 200 mm AND

THEN (3)+(4)-(5)=> (25x+158+15c)+(5x-15c)-(18-5x)=(-300)-(200) OR NA = 125 5

AND USING EQ(5). No - (-125) = 200 OR UB = 75 5

Ea. (4) -- 125 - 15 -- 300 OR 15 = 175 5



GIVEN: 15 = 30 5, \$ = 40. CI = 10

FIRST NOTE .. (50) x = 2 sind = (30 5) sind (5) = 10 cosp = (30 \$) cosp

ALIO, ALONG INCLINE CAB. 9 = X TAN 10

HORIZONTAL MOTION (UNIFORM) X= X3+ (50)xt = (30 SIND) t OR t= 30 SIND

VERTICAL MOTION (UNIF. ACCEL. MOTION) 4= 432 (no/t - 29t2 = (30 coxp)t- 29t2 SUBSTITUTING FOR E --

4= (30 cosp) (x 30 sing) - 29 (30 sing) (9=32.2 些) TAN \$ 1800 SINT \$

AT B: \$ = 40, X=d8: d8 TANIO - TANAO 1800 51140

OR db = 23.5 ft

AT C: $\phi = -40$, $\chi = -d_c$: $-d_c$ TANIO = $\frac{-d_c}{TAN(-40)} - \frac{32.2}{1800} \frac{(-d_c)^2}{510^2(-40)}$

OR dc = 31.6 ft

11.189

GIVEN: (15) - 8 # - (1511); (5) - 8 m 1 (5 m/s) 2 45 FIND: UN WHERE UN IS CONSTANT

HAVE - IN = IS + INIS USING THIS EQUATION, THE TWO CASES ARE THEN GRAPHICALLY REPRESENTED AS SHOWN.

Nuys LIES LINE (25) = 8 pm (n2)=8 Fm

(TMS) LIES ON THIS LINE

FROM THE DIAGRAM .. 12 = (8)2+ (8+8)2 DR NW = 17.89 F AND TANK = 16 OR 01 = 63.4°

: 5W=17.89 FM 634°

=1500 fl

11.190

GIVEN: P=1500 St; N=45 H, N=30 H, 4 512 = 750 St ; Q = CONSTANT

FIND: a WHEN as = 500 ft

FIRST NOTE .. J = 45 # = 66 = N = 30 F1 . 44 St HAVE UNIFORMLY DELELERATED MOTION ..

1. 15 = 15, + 20+ (5-5,) WHEN 5= 52: (44 \$)2= (66 \$)2+202 (750 \$2) OR az = -1,61333 1/32

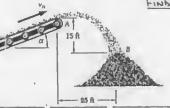
THEN WHEN AS = SOU ST : 125 = (PP #)5+5(-1.P1333 2)(200 tf)

an = 52 = 2742.67 52 2742.67 11/52 - 1.828 45 52

FINALLY. Q2 = Q2 + Q2 = (-1.61333 32)2 (1.82845 32)2 OR a = 2.44 1/2

11.191

GIVEN: No = 24 5 FIND: OL



FIRST NOTE .. (UL) = 5, COSK = (24 5) COSK (Jy)= Jo sne K = (24 5)514 K

HOISIZONTAL MOTTON (UNIFORM) X= x3+ (15x) = (24 cos a)t AT POINT B: 25 = (24 COSA) t OR LB = 24 COSOL

VERTICAL MOTION (UNIF. ACCEL. MOTION) (9=32.2 5) 4= 43+ (M) & - 29t2 = (24 SINK) t - 29t2 AT POINT B: -15 - (24 SINK)tg - 29tg

SUBSTITUTING FOR to .. (CONTINUED)

11.191 CONTINUED

-15 = (24 sina) (25) - 29 (25 cosk)2 1259 -3 = 5 TANK - 1152 COSTA

Now .. COSTA = SECTA = 1+ TANTA

THEN .. -3 = STANX - 1152 (1+ TANZ)

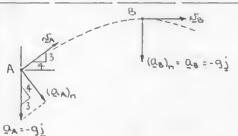
OR 3.4939 TANEA - 5 TANA + 0.49392 = 0

SOLVING -. TANO . 0.106746 AND TANO : 1.32432 THEN .. \alpha = 6.09 AND \alpha . SZ.9°

11.192

GIVEN. PA= 25 m FIND: (a) JA

(b) PB, WHERE 48 = 4MAX



(a) Have .. (an) = 152

OR NA = [4(9.81 %)](25 m)
OR NA = 14.0071 % :. 5 14.01 5 1369

(a8) = 502 (b) HAVE ..

WHERE No = (NA) = 3 NA

Po = (\$ < 14.0071 3)2 THEN ..

OR B = 12.80 m

11,193



GIVEN: No = No -- , No = CONSTANT

FIND: B AND B IN TERMS OF US, h. AND B

FROM THE DIAGRAM F. h No So = Jo SINB

Now .. No = 18 SUBSTITUTING FOR US AND T .. O (Guis) = Guis Qu

(CONTINUED)

11.193 CONTINUED

OR B= TO SINZO

HAVE O = FSINZO

THEN. B = 4 (28 SIND COS B)

SUBSTITUTING FOR B ... 0 = No (2 SINB COSB) (50 SINEB)

OR B=ZHESWBESD

ALTERNATIVE SOLUTIONS

HAVE .. C = SINB

THEN i = - hast b

Now. 12. 152 = (+)2+(+0)2

 $DR \qquad D_{S}^{2} = \left(-\frac{2!N_{F}\theta}{\mu \Theta}\right)^{2} \left(-\frac{2!N_{F}\theta}{\mu \Theta}\right)^{2} \left(-\frac{2!N_{F}\theta}{\mu \Theta}\right)^{2} \left(-\frac{2!N_{F}\theta}{\mu \Theta}\right)^{2}$

OR B= = To SINZA

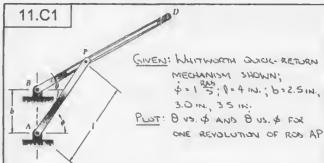
NOTE THAT AS B INCREASES, THE AIRPLANE MOVES IN THE INDICATES DIRECTION. THUS, THE FUSITIVE ROOT IS CHOSEN. B= JasIN2B

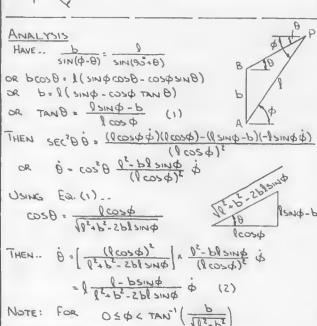
HAVE _ Q = Qr + Q8

NOW NO = CONSTANT => Q=0

: ap = r 8+2 + 0 = 0

OS B = -5 (- 1020 " " 20050) (20 21050).





Eq. (1) => -90:8 < 0 THUS, FOR THESE VALUES OF ϕ MUST USE $\theta = TAN^{-1} \left(\frac{1 \sin \phi - b}{1 \cos \phi} \right) + 360^{\circ}$ WHEN PLOTTING THE GRAPH.

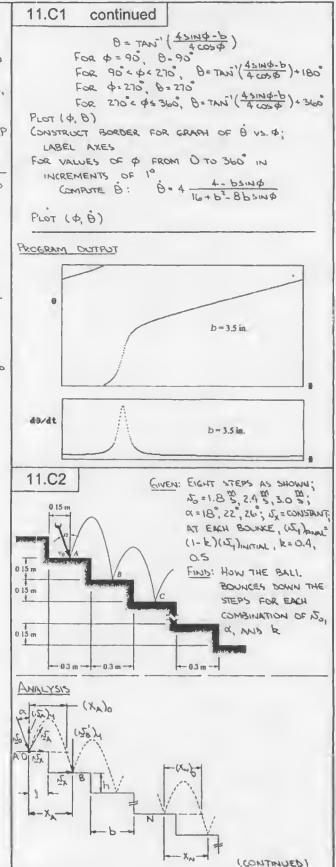
WHEN PLOTTING THE GRAPH.

SIMILARLY,

FOR 90° $< \phi < 270^\circ$, Eq. (1) => -90 $< \theta < 90^\circ$ $\therefore \theta = TAN^{-1} \left(\frac{l \sin \phi - b}{l \cos \phi} \right) + 180^\circ$

FOR 270° < \$ = 360°, EQ. (1) => -90° < 8 = 0 : 8 = TAN- (\frac{1 \sin \psi - b}{1 \cos \phi}) + 360°

OUTLINE OF PROGRAM INPUT VALUE OF bCONSTRUCT BORDER FOR GRAPH OF $0 \le 0 \le 0$; LABEL AXES FOR VALUES OF ϕ FROM $0 = 0 \le 0 \le 0$ INCREMIENTS OF 1COMPUTE $0 : 0 \le \phi < tan'(\frac{b}{11b-b^2})$, $0 = tan'(\frac{4 \le 1 \times 4 - b}{1 \le 0 \le 0}) + 3bo$ FOR $tan'(\frac{b}{11b-b^2}) \le \phi < 9o$



11.C2 continued

FIRST NOTE... $\sqrt{x} \cdot \sqrt{x} \sin \alpha$ $(\sqrt{x})_{q} = (1-k)\sqrt{x} \cos \alpha$ WITH THE ORIGIN OF A RECTANGULAR

WORDINATE SYSTEM AT POWT O..

HORZONTAL MOTION (UNIFORM)

Y= $\chi_{x}^{2} + \chi_{x}^{2} + \chi_{$

CONSIDER THE MOTION OF THE BALL AFTER IT LANDS ON A GIVEN STEP

I. DETERMINE IF THE BALL BOUNCES TWICE ON STEP A: $On STEP A, y=0: O = \frac{(\sqrt{N_A})^4}{N_A^2}(\chi_A)_0 - \frac{1}{2}q \frac{(\chi_A)^4}{N_A^2}$ $OR (\chi_A)_0 = \frac{2}{q} N_A (N_A)_4$

.. IF (XA) < P, THE BALL BOUNCES TWICE ON STEP A.

IN GENERAL, THE BALL BOUNCES TWICE ON STEP N (N=A,B,C,...,H) IF

(XN) = D + (N-1) B - X-1 X)

WHERE $(X_n)_0 = \frac{2}{9} I_{X_n} (N_n)_q$ AND X_n AND $(N_n)_q$ ARE GIVEN BELOW.

2. DETERMINE IF THE LANDS ON STEP B:

ON STEP B, q=-h: $-h = \frac{(N_n)_q}{N_n} X_n - \frac{1}{2} q \frac{X_n^2}{N_n^2}$

SOLVING FOR X_A AND TAKING THE POSITIVE

ROOT $(X_A > 0)$ HAVE.. $X_b := \frac{(U_A)_A}{U_A} + \left[\left[-\frac{(U_A)_A}{V_{A_A}} \right]_c^2 - 4 \left(\frac{9}{2} U_{A_A}^2 \right) (-h) \right]^{1/2}$ $= \frac{(U_A)_A}{U_A} + \left[\left[-\frac{(U_A)_A}{V_{A_A}} \right]_c^2 + 29h \right]$

: IF X = l+b, THE BALL BOUNCES ON STEP B.

IN GENERAL, AFTER THE BALL BOUNCES ON STEP; IF $\sum_{i=1}^{N} X_{i} \leq x_{i} + (i-1)b$

WHERE XM = \frac{n_x}{2} \((n_y)_4 + \left[(n_y)_1 \right]_x + 50 \((i-n) \right) \right]

FINALLY, IF THE BALL BOUNCES ON STEP B, HAVE USING THE EXPRESSION DERIVED ABOVE FOR NY. $(N_B^2)_Y \cdot (N_A)_Y - 9 \frac{X_A}{N_X}$

NOTING THAT $(\sqrt{s})_{\gamma} < 0$ AND THAT THE MAGNITUSE OF THE VERTICAL COMPONENT $(\sqrt{s})_{\gamma}$ OF THE VELOCITY AFTER THE BOUNCE IS $(\sqrt{s})_{\gamma} = (1-k)[q]_{\sqrt{s}}^{N} - (\sqrt{s})_{\gamma}]$

HAYE IN GENERAL - (1-12)[9 xu-1 - (15n-1)4]

11.C2 continued

OUTLINE OF PROGRAM

FOR INITIAL ANGLES K: R=18, Z2, 26

FOR VALUES OF K: R=0.4,0.5

FOR INITIAL VELOCITIES No: No=1.8 \$, 2.4 \$, 3.0 \$

FOR EACH COMBINATION OF K, K, AND No

(DMFUTE No AND (No):

SET INITIAL CONDITIONS: N=1, i=2, MOTAL = 0

WHERE 1,2,3,..., B CORRESPOND TO STEPS

A,B,C,..., H AND MOTAL IS THE SUM OF

THE HOWSONTAL DISTANCES BETWEEN

SUCCESSIVE POINTS OF IMPACT.

DETERMINE IF THE BALL BOUNCES TWICE ON

IF \(\frac{1}{9}\ln X_1(\sqrt{1}\sqrt{1}\r)_1 \leq 0.75 + (N-1)(0.3) - XTOTAL PRINT: BALL FIRST BOUNCES TWICE ON STEP N.

CONSIDER THE NEXT COMBINATION OF K, k, AND NO.

DETERMINE THE NEXT STEP ON WHICH THE BALL BOUNCES

WHERE $X_{N} = \frac{\sigma_{x}}{g} \left\{ (\sigma_{x})_{y} + \left[(\sigma_{x})_{y}^{2} \cdot O^{2}g(i-u) \right] \right\}$

DETERMINE IF THE BALL BOUNCES ON CONSECUTIVE STEPS

IF XTOTAL > 0.15+(1-1)(0.3) AND
1:8 PRINT: BALL MISSES STEP!.

RESET XTOTAL: XTOTAL = XTOTAL - XN

UPDATE !: !=!+!

IF i < B, COMPUTE NEW XN AND XTOTAL AND REPEAT CHECK IF i < B, CONSIDER THE NEXT

COMBINATION OF M, K, AND US DETERMINE HOW THE BALL BOUNCES DOWN THE REMAINING STEPS IF N ? B PRINT: "BALL CONTINUES

TO BOUNCE DOWN THE STEPS.

IF N < 8, UPDATE VALUES FOR

THE NEXT STEP:

24: (21)4=(1-4)[3 2x -(2")]

N: N= i

PROGRAM OUTPUT

α k vo

18° 40% 1.8 m/s Ball first bounces twice on step A
2.4 m/s Ball first bounces twice on step C
3.0 m/s Ball misses step D
Ball continues to bounce down the
steps

50% 1.8 m/s Ball first bounces twice on step A
2.4 m/s Ball first bounces twice on step Ball continues to bounce down the steps

11.C2 continued 3.0 m/s Bali misses step B Ball misses step E Bali misses step G 50% 1.8 m/s Bsli first bounces twice on step A 2.4 m/s Bsll first bounces twice on step C 3.0 m/s Bsll mlsses step C Bsil misses step H 26° 40% 1.8 m/s Bsll first bounces twice on step B 2.4 m/s Bsll misses step D Bs11 mlsses step G 3.0 m/s Ball mlsses step B Bali mlases step D Ball misses step F Bs11 misses step H 50% i.8 m/s Bsli first bounces twice on step A 2.4 m/s Bsil continues to bounce down the steps 3.0 m/s Ball misses step B Ball misses step E Bsii mlsses step G

11.C3

GIVEN: LOB = 10 m; ADRAG = - KUZ, R = 0, ZNOT m, 4 NOZ m; 0 = 70, 100, 130

FIND: JMAX AND THE FIRST
TWO VALUES OF B
FOR WHICH J=0 FOR
EACH COMBINATION OF
BO AND R

ANALYSIS

IN THE TANGENTIAL

DIRECTION, THE TANGENTIAL

COMPONENT OF THE

ACCELERATION OF THE

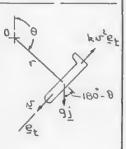
AIRPLANE IS

a g sin(180-8)-ks²

a g sin 8-ks²

RECALLING THAT Q = dE

HAVE du = g sin 8-ks²



Now, since C: constant, have 5: CO THEREFORE, THE DIFFERENTIAL EQUATIONS

PE = dring - Kas

第= 千万

DEFINE THE MOTION OF THE AIRPLANE.

OUTLINE OF PROGRAM

INPUT VALUE OF R

INPUT VALUE OF 80

CASE 1: DETERMINE THE VALUE OF THE VELOCITY AT THE SPECIFIED ANGLE OF THE INPUT OF USE, FOR EXAMPLE, THE MODIFIED EULER

Use, for example, the modified Euler METHOD (SECOND-ORDER RUNGE-KUTTA METHOD - SEE CHAPRA AND CANALE, NUMERICAL METHODS FOR ENGINEERS, 2d (CONTINUES)

11.C3 continued

ED, McGRAW-HILL, 1988.) WITH A STEP SIZE AZ: 0.008 5 TO NOMERICALLY INTEGRATE THE EQUATIONS du = (951NB-KUZ 8 2180° dt = 2-951NB-KUZ 8 2180° de Z. L.N

UNTIL $\theta_i' \leq \theta_i \leq \theta_z$, WHERE θ_i' AID θ_z ARE THE VALUES OF θ AT THE MIDPOINT AND END, RESPECTIVELY, OF THE FINAL TIME INTERVAL.

USE LINEAR INTERPOLATION TO DETERMINE THE FINAL VELOCITY SE:

We simply hercomy of:

PRINT THE VALUES OF R. BO, B, AND J, CASE 2: DETERMINE THE VALUE OF B FOR WHICH THE VELOCITY IS FIRST ZERO USE THE MODIFIED EULER METHOD WITH A

USE THE MODIFIED EULER METHOD WITH A STEP SIZE AT: 0.008 5 TO NUMERICALLY INTEGRATE THE EQUATIONS

1 = 72 90 = 72

WHERE θ_i is the value of θ at the Beginning of a time interval, until $u_2 < 0$, where $u_2 > 0$ the velocity at the end of a time interval. Use linear interpolation to determine the final angle θ_i :

THE FINAL ANGLE β_{ξ} : $\theta_{\xi} = \theta_{1}' + \frac{O - \Sigma_{1}'}{\Delta \Sigma_{2} - \Sigma_{1}'} (\theta_{2} - \theta_{1}')$

PRINT THE VALUES OF k, Bo, AND BE

SUMMARY OF PROGRAM OUTPUT

Maximum velocity sttsined for s relesse sngle $heta_0$

в	V _{men} 01/8						
	k = 0	$k = 2 \times 10^{-1} \text{ m}^{-1}$	$k = 4 \times 10^{-2} \text{ m}^{-1}$	k = 0, theory			
70*	16.23	16.19	11.67	16.23			
100°	12.73	12.71	9.78	12.73			
130°	8.37	8.36	6.97	8.37			

First $[(\theta_0)_1]$ and second $[(\theta_0)_2]$ rest positions for a release angle (θ_0)

	k = 0		$k = 2 \times 10^{-4} \text{ m}^{-1}$		$k = 4 \times 10^{-4} \text{ m}^{-1}$	
66	(8)	(80),	(80)	(80),	(80),	(80),
70°	290.0°	70.0°	289.2°	71.6	229.4"	146.70
100	260.0°	100.0	259.7°	100.6°	223.7°	149.3
130°	230.0	130.0°	229.9	130.2	213.6	154.6

GNEN: (AR TRAVELING ON AN EXIT RAMP; N5=60 T, NEINAL = 0; lamax 1 = 10 4/5; RAMP 15 EITHER STRAIGHT OR CURVED (P = 800 ft); de 15 EITHER CONSTANT OR VARIES LINEARLY DURING TIME INTERVALS OF 15

FIND: LOTOP AND DISTANCE TRAVELED ON THE RAMP FOR EACH COMBINATION OF RAMP TYPE

ANALYSIS CASE 1: STRAIGHT RAMP, JE = CONSTANT

FOR THIS UNIFORMLY DECELERATED RECTILINEAR MOTION HAVE ... St = 0 = -10 3

5= 5= + (-10)t N= N2+2(-10)(x-X)

NOTING THAT Q IS CONSTANT AND WEINN =0,

 $t_{5707} = \frac{N_3}{10}$ (5) $X_{707AL} = \frac{N_3}{20}$ (41)

WHERE ISTOP AND XTOTAL ARE THE TIME FOR THE CAR TO COME TO REST AND THE TOTAL DISTANCE TRAVELED BY THE CAR ON THE RAMP, RESPECTIVELY. ALSO, US = 60 m/h = 88 ft/s

CASE 2: STRAGUT RAMP, OF LINEARLY VARYING

HAVE a= dr AND ASSUMING THAT FOR

ANY TIME INTERVAL a1=0 a2=-104/3

HAVE $\frac{ds}{dt} = \alpha = -\frac{10}{\Delta t}(t-t_1)$ $(\frac{ct}{s^2})$ At $t = t_1$, $s = s_1$: $s = \frac{t}{t_1} = \frac{10}{\Delta t}(t-t_1)dt$

Now -- $\frac{dx}{dt} = \sqrt{x}$ $\frac{5}{x} \left(\frac{1}{x} - \frac{1}{x}\right)^2$ (1) AT t=t,, X=X,: [x, dx = [t, J, - \frac{5}{4}(t-t,)]dt

OR X=X, + 15, (t-t,)- 30t (t-t,)3 (2) FOR At=15 AND WHEN t=tz, HAVE .. (1) => Nz = N, -S (ft)

(2)=> X2= X1+151- = (H) FOR THE FINAL TIME INTERVAL (STEINAL (15) 5=0 AT t=triNA. THEN, ASSUMING t=0

(FOR CONVENIENCE) HAVE .. (1)=> 0= 5, - 1/2 (triNAL) 21 = 12 OR LEINA = IN (5)

AND (2) => XFINAL = XI + SIZEINAL - 3ZEINAL (HZ) WHERE XFINAL IS THE TOTAL DISTANCE, TEINAL IS THE TIME BURATION OF THE FINAL TIME (CONTINUES)

11.C4 continued

INTERVAL, AND IS AND X, ARE THE VELOCITY AND DISTANCE RESPECTIVELY AT THE BEGINNING OF THE FINAL TIME INTERVAL.

(ASE 3: CURVED RAMP, OF = CONSTANT

HAVE .- QL = du = CONSTANT

Now .. Q = Q + Q 1 (1)2

WHERE P= BOD FR AND lamal= 10 4/3 FOR EACH TIME INTERVAL, Q IS CONSTANT AND an is maximum at time t, since the VELOCITY DECREASES FROM t, TO tz.

$$\therefore \ \, \Omega_{\text{MAX}}^{2} = \Omega_{\xi}^{2} + \left(\frac{15^{2}}{p^{2}}\right)^{2}$$

$$\text{OR} \ \, \Omega_{\xi} = -\sqrt{\Omega_{\text{MAX}}^{2} - \frac{15^{2}}{p^{2}}} \quad \left(\frac{57}{57}\right)$$

FOR EACH TIME INTERVAL.

NOW .. Q = CONSTANT (UNIF. ACCEL. MOTION)

THEN .. N= N, + Q (1-1,) (3) AND X= X, + W, (1-t,)+ 2 ax (1-t,)2

FOR At=15 AND WHEN t=t2, HAVE ...

(3) => Wz= W, + Ot

(4) => X2=X,+15,+20+ (4) FOR THE FINAL TIME INTERVAL, 5=0 AT total THEN, ASSUMING t,= 0 HAVE --

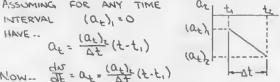
at = - \ans - 15,4 (50)

(3) => 0=5,+ a,(trun) or trun = a,

(4) => XEINN = XI + NIZEINN + 2 Octrime WHERE IS, AND X, ARE THE VELOCITY AND DISTANCE RESPECTIVELY, AT THE BEGINNING OF THE FINAL TIME INTERVAL.

(ASE 4: CURVED RAMP, SE LINEARLY VARYING ASSUMING FOR ANY TIME

INTERVAL (Qt) = 0



Now. $dN = \alpha_{\xi} = \frac{(\alpha_{\xi})_{\xi}}{\Delta t} (t \cdot t_{i})$ $\rightarrow \Delta t \cdot \Delta t$ At $t = t_{i}$, $N = N_{i}$: $\int_{N_{i}}^{\infty} dN = \int_{t_{i}}^{t} \frac{(\alpha_{\xi})_{\xi}}{\Delta t} (t \cdot t_{i}) dt$ OR 5=5,+ (at)2 (t-t,)2 (5)

Ar Let, $x \in X$; $\begin{cases} x \\ dx \end{cases} = \begin{cases} \frac{1}{2} \left[x \right] + \frac{(a_t)_t}{2a_t} (t - t_t)^2 \right] dt$

Now. $Q_{5} = Q_{5}^{5} + Q_{5}^{5}$ $= Q_{5}^{5} + Q_{5}^{5}$ $= Q_{5}^{5} + Q_{5}^{5}$. (P)

11.C4 continued

WHERE P= 800 ft AND 10mmx1 = 10 ft/s.

NOW, FOR ANY TIME INTERVAL,

(an) MAX OCCURS AT t=t, (WHEN THE

VELOCITY IS MAYMUM)

(at) MAX OCCURS AT t=t;

(an) MAX OCCURS AT t=t;

(an) MAX OCCURS AT ALL TIMES (NOTE...

.. Assume a= amax at t=t2. THEN... $Q_{\text{max}}^{z} = (Q_{z})_{z}^{2} + (\frac{S_{z}^{z}}{P})^{2} \qquad (7)$

FOR $\Delta t = 1.5$ AND WHEN $t = t_2$, HAYE

(S) $\Rightarrow N_2 = N_1 + \frac{1}{2}(\Omega_t)_2$ OR $(\Omega_t)_1 = 2(N_2 - N_1)$ (8)

(b) => $\chi_2 = \chi_1 + \zeta_1 + \zeta_1 (Q_2)_2$ (ft)

(DMBINING EUS. LT) AND (B) TO ELIMINATE $(Q_2)_2$. $Q_{MAX}^2 = \left[2(\zeta_2 - \zeta_1)\right]^2 + \frac{N_2^4}{P^2}$ OR $\frac{\zeta_2^4}{P^2} + 4U_2^2 - 8\zeta_1U_2 + (4U_1^2 - Q_{MAX}^2) = 0$ - A GNARTIC EQUATION WHICH DEFINES U_2 .

FOR THE FINAL TIME INTERVAL, $U_2 = 0$ AT $t = t_{FINAL}$. Then, Assuming $t_1 = 0$ have..

EQ. (8): $(a_t)_2 = 2(D_2 - D_1)$ WHERE $D_2 < 0$ (5) $\Rightarrow D = D_1 + \frac{1}{2}(a_t)_2 + \frac{1}{2}D_1$ OR $t_{FINAL} = \sqrt{-\frac{2D_1}{(a_t)_2}}$ (5)

(6) > XFINK = X, + 15, tring + & (QE), tring (SE)
WHERE IS AND X, ARE THE VELOCITY AND
DISTANCE, RESPECTIVELY, AT THE BEGINNING OF
THE FINAL TIME INTERVAL.

OUTLINE OF PROGRAM

INPUT INITIAL VELOCITY No.

CONSIDER EACH CASE:

CASE 1: STRAGHT RAMP, JF = CONSTANT

COMPUTE TIME LSTOP: LSTOP = 10

COMPUTE DISTANCE XTOTAL: XTOTAL " \\\
\text{ZO}
\text{PRINT THE VALUES OF LISTOP AND XTOTAL CASE 2: STRAIGHT RAMP, OF LINEARLY VARYING FOR EACH SUCCESSIVE TIME INTERVAL COMPUTE NIZ: NZ = UZ - S

WHILE 152 > 0

UPDATE DISTANCE X; X=X+5,-3

UPDATE TIME AND SPEED:

L=L+1; 15=5

FOR THE FINAL TIME INTERVAL
COMPUTE TRINAL: TRINAL: 150

COMPUTE TIME \$5TOP: \$5TOP = \$1 + \$ FINAL

COMPUTE DISTANCE XTOTAL:

XTOTAL = X1 + U1 + FINAL - 3 + \$ FINAL

PRINT THE VALUES OF \$5TOP AND XTOTAL

CASE 3: CURVED RAMP, St = CONSTANT

FOR EACH SUCCESSIVE TIME INTERVAL

(CONTINUES)

11.C4 continued

COMPUTE $a_{\xi}: A_{\xi} = -(100 - \frac{U_1^2}{64 \times 10^4})^{1/2}$ COMPUTE $U_2: U_2 = U_1 + Q_{\xi}$ WHILE $U_2 > 0$ UPDATE DISTANCE $X_i: X_i = X_1 + U_1 + \frac{1}{2}Q_{\xi}$ UPDATE TIME AND SPEED: $t = t + 1 : U_1 = U_2$

FOR THE FINAL TIME INTERVAL COMPUTE Q: Q== (100 - 51/4)1/2

COMPUTE TENNE: TENNE - OT COMPUTE TIME TSTOP: TSTOP = T+ TENNE COMPUTE DISTANCE XTOTAL!

PRILIT THE VALUES OF ESTOP AND KTOTAL CASE 4: CURVED RAMP, OF LINEARLY VARYING FOR EACH SUCCESSIVE TIME INTERVAL

SOLVE THE EQUATION $\frac{S_2^4}{L4 \times 10^3} + 4 N_2^2 - 8 S_1 N_2 + (4 N_1^2 - 100) = 0$

FOR No USING NEWTON'S METHOD
(SEE, FOR EXAMPLE, CHAPRA AND
CANALE, NUMERICAL METHODS FOR
ENGINEERS, 2d ED., McGRAW-HILL,
1988.)

WHILE NZ > 0

COMPUTE (QDZ: (QE)Z = Z(NZ-NZ)

UPDATE DISTANCE X; X; = X; + NZ+ to (QE)Z

UPDATE TIME AND SPEED:

FOR THE FINAL TIME INTERVAL

COMPATE $(a_{\ell})_2$: $(a_{\ell})_2 = 2(b_2 - b_1)$ COMPATE t_{FINAL} : $t_{FINAL} = [-2, \frac{b_1}{(a_{\ell})_2}]^{1/2}$

COMPUTE TIME tSTOP: tSTOP = t+teiner

COMPUTE DISTANCE XTOTAL:

XTOTAL = X1 + N, TEINAL + L(QE) 2 TEINAL
PRINT THE VALUES OF TSTOP AND XTOTAL

PRISGRAM OUTRUT

For a atraight highway and a constant rate of change of the speed,

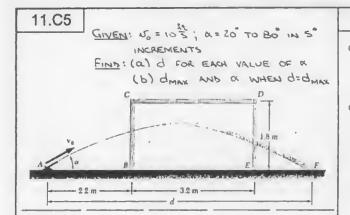
time to stop = 8.80 s

diatance traveled = 387.2 ft

For a straight highway and a uniformly varying rate of change of the apeed, time to stop = 17.77 a distance traveled = 789.2 ft

For a curved highway and a constant rate of change of the speed, time to stop = 11.29 a distance traveled = 581.4 ft

For a curved highway and a uniformly varying rate of change of the apeed,
time to atop = 20.71 a
distance traveled = 1015.3 ft



ANALYSIS

HURISOUTAL MOTION (UNIFORM) 4

X=X0+ (N=cosx)t

OR t= N=cosx

A

VERTICAL MOTION (UNF. ACCEL. MOTION) $y = \chi_{s}^{2} + (J_{s} \sin x)t - \frac{1}{2}gt^{2}$ $J_{s}^{2} = J_{s} \sin x - gt$ SUBSTITUTING FOR t. $y = (TANN)x - \frac{1}{2}g\frac{x^{2}}{J_{s}^{2}\cos^{2}x}$

AT POINT F, X=d AND 4=0: 0= (TANA)d - 29 52 005ta

OR d= \$ SINZX

AT THE MAXIMUM THEORETICAL HEIGHT YMAX OF
THE WATER, Ng = D. THEN.

D = SSINK - 9tymax OR tymax = 3 SINK
THEN U = (15 SINK) - 10 (150 SINK) 2

THEN YME = (20 SINX) (20 SINX) - 29 (20 SINX) 2

= 2 2 3 SINZX

AND $X_{\text{max}} = (x_0^2 \cos x)(\frac{x_0^2}{9} \sin x)$

IF THE WATER HITS THE ARBOR, Y=1.8 W AT THE POINT OF IMPACT. THE CORRESPONDING VALUE OF X IS THEN. X2

WHERE THE (+) AND (-) SIGNS CORRESPOND TO THE WATER HITTING THE ARBOR FROM ABOVE AND FROM BELOW, RESPECTIVELY.

OUTLINE OF PROGRAM

INPUT MINIMUM AND MAXIMUM VALUES OF A INPUT SIZE OF INCREMENT OF A FOR EACH VALUE OF A COMPUTE Y AT X=2.2 m:

Y_{2.2} = 2.2 TANX - 2.5 X X

COMPUTE 4 AT X= S.4 M:
45.4 = S.4 TANN - 0.14589

(CONTINUED)

11.C5 continued

- (1) IF Y2271.8m AND YS471.8m

 COMPUTE d: d= 100 51N 2R

 PRINT THE VALUES OF R AND d

 NEXT VALUE OF R
- (2) IF Y2.271.8m AND Y5.45 m

 COMPUTE (XARBOR)ARDYE:

(XARBOX) MONE = 100000X (SINN +

PRINT THE VALUES OF K AND

(XARBOX)ABOVE

NEXT VALUE OF A

COMPUTE YMAX: YMAX: 90 SINZA

COMPUTE XYMAX: XYMAX: 97 SINZA

- (3) IF YMAX < 1.8 M

 COMPATE d: d = 100 SIN 2K

 PENT THE VALUES OF A AND C

 NEXT VALUE OF A
- (4) IF 2.2 $m = x_{\text{MMax}} \leq 5.4 \text{ m}$ COMPUTE (XARBUX) BELOW: $(x_{\text{MABUX}})_{\text{BELOW}} = \frac{100 \cos x}{9} (\sin x - \frac{1}{3} \cos x)$

PRINT THE VALUES OF K AND (XARBOR) BELOW NEXT VALUE OF &

- (5) IF Y22 = 1.8 m
 PRINT "THE WATER HITS THE ARBOR
 AT CORNER C."
 NEXT VALUE OF M
- (6),(7) IF Xymax < 2.2 m OR IF Y5.4 < 1.8 m

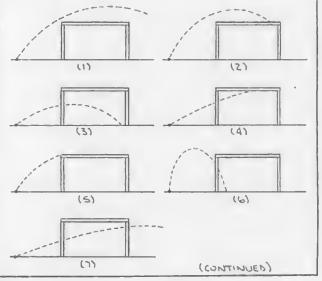
 AND XYMINX > 5.4 m

 COMPUTE d: d= 100 SIN 200

 PRINT THE VALUES OF A AND O

 NEXT VALUE OF OX

THE SEVEN POSSIBLE TRAJECTORIES TESTED FOR IN THE PROGRAM ARE ILLUSTRATED BELOW.



11.C5 continued

PROGRAM OUTPUT

(a)

For α = 20.00°, the water hits the ground at d = 6.552 m For α = 25.00°, the water hits the ground at d = 7.809 m For α = 30.00°, the water hits the ground at d = 8.828 m For α = 35.00°, the water hits the ground at d = 9.579 m For α = 40.00°, the water hits the top of the arbor from below at x = 3.106 m For α = 45.00°, the water hits the top of the arbor from below at x = 2.335 m For α = 50.00°, the water hits the ground at d = \$10.039 m For α = 50.00°, the water hits the ground at d = \$8.288 m For α = 60.00°, the water hits the ground at d = 8.828 m For α = 60.00°, the water hits the ground at d = 7.809 m For α = 70.00°, the water hits the ground at d = 6.552 m For α = 80.00°, the water hits the ground at d = 6.552 m For α = 80.00°, the water hits the top of the arbor from above at x = 4.557 m For α = 80.00°, the water hits the top of the arbor from above at x = 3.133 m above at x = 3.133 m

(b)

For α = 46.20°, the water hits the top of the arbor from below at x = 2.202 m

For α = 46.21°, the water hits the top of the arbor from below at x = 2.201 m

For α = 46.22°, the water hits the top of the arbor from below at x = 2.200 m

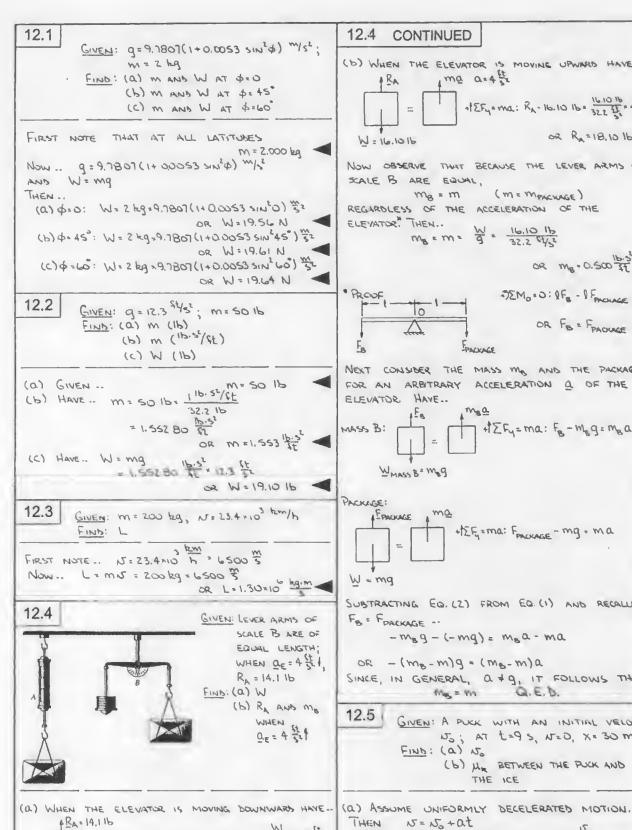
For α = 46.23°, the water hits the ground at d = \$10.184 m

For α = 46.24°, the water hits the ground at d = \$10.184 m

For α = 46.25°, the water hits the ground at d = \$10.184 m

For α = 46.25°, the water hits the ground at d = \$10.184 m

For α = 46.26°, the water hits the ground at d = \$10.184 m For $\alpha=46.27^\circ$, the water hits the ground at d=10.184 m for $\alpha=46.28^\circ$, the water hits the ground at d=10.184 m for $\alpha=46.28^\circ$, the water hits the ground at d=10.183 m for $\alpha=46.30^\circ$, the water hits the ground at d=10.183 m



+18 Fy= ma: W- 14.11b = 32.2 FE = 4 52

(CONTINUED)

ma a=4 ==

OR W=16.10 16

(b) WHEN THE ELEVATOR IS MOVING UPWARD HAVE .. ma a=4 ft of EFy= ma: Rx-16.10 16= 322 FF= 4 52 02 R=18.10 16 NOW OBSERVE THAT BECAUSE THE LEVER ARMS OF SCALE B ARE EQUAL. m8 = m (m=mpackage) REGARDLESS OF THE ACCELERATION OF THE M8 = M = 9 = 16.10 16 OR MB - 0.500 IF. 3 JEMO = D: 1FB - 1 FANCINGE = O OR FB = FPACE PACKAGE NEXT CONSIDER THE MASS WE AND THE PACKAGE FOR AN ARBTRARY ACCELERATION Q OF THE +1 EF4=ma: Fx-Mg=mga (1) + EF= ma: FACKAGE - mg = ma SUBTRACTING EQ. (2) FROM EQ. (1) AND RECALLING -mgg-(-mg)= mga-ma OR - (mp-m)9 = (mp-m)a SINCE, IN GENERAL, Q + 9, IT FOLLOWS THAT Q.E.D. May = WY GIVEN: A PUCK WITH AN INITIAL VELOCITY 5; AT 1=95, N=0, X=30 m FIND: (a) No (b) My BETWEEN THE PUCK AND THE ICE THEN N= N5+at Oz- 0 AT t=95: 0 = N3+Q(9) OR ALW. 12 = 15 + 20 (x-X) Ar t=95: 0= No2 + 20 (30) (CONTINUED)

12.5 CONTINUED

SUBSTITUTING FOR Q. 0= 52 + 2(- 5)(30)=0 OR 50 = 6. 4667 \$

OR 15:6.67 5. AND Q = - 6,6667 = -0.74074 TE (b)

HAVE - + 12F = 0: N-W= 0

OR N=W=ma

SLIDING: F= MRN

" Mx mg == EFx = ma: - F = ma

OR - MR mg = ma

OR MR = - 3 = - -0.740.74 m/s2

OR 11 = 0.0755

12.6

GIVEN: AN AUTOMOBILE INITIALLY AT REST; US: 0.80 BETWEEN THE TIRES AND

THE PAVEMENT FIND: (a) JMAX WHEN X= 400 M FOR FRONT-INHEEL BRIVE, WERONT/W = 0.62 (b) Nome WHEN X=400 M FOR REAR-

WHEEL DRIVE WREAR/W = 0.43

(a)

FOR MAXIMUM ACCELERATION .. Fr = FMAX = US NF = 0.8 (0.62W)

= 0.496 W = 0.496 mg Now .. = EFx = ma: FE = ma

or 0.496 mg = ma

THEN Q = 0.496 (9.81 32) = 4.86576 32

SINCE a is constant, HAVE ...

WHEN X= 400 m: 15 2 (4.86576 32) (400 m)

OR JMAX - 62.391 5 OR JMAN = 225 Fm

FOR MAXIMUM ACCELERATION .. FR = FMAX = USNR = 0.8(0.43W) NAB

= 0.344 W = 0.344 mg

Now -- EFx = ma: Fx = ma

OR 0.344 mg = ma

THEN Q = 0.344 (9.81 52) . 3.37464 5

SINCE Q IS CONSTANT, HAVE --

WHEN X = 400 m: 52 = 2(3.37464 =)(400 m)

OR JMN = 51.959 5

OR Nomax = 187.1 FM

12.7

GIVEN: (a) LEVEL = 3 1/51; BURGRADE " 7",
(No) URGRADE = 60 mi/h; P= CONSTANT FIND: XUPCRADE WHEN No 50 mi/h

FIRST CONSIDER WHEN THE BUS IS ON THE LEVEL SECTION OF THE HIGHWAY.

HAVE .. - EFx = ma: P = q a cever NOW CONSIDER WHEN THE BUS IS ON THE URGRADE

HAVE .. +- EF = ma: P. WSINT . Qa SUBSTITUTING FOR P. GALEVEL -WSIN7 = ga OR a = aLEVEL - 9 SINT = (3-32,2 SINT) } = - 0.924 19

FOR THE UNIFORMLY DECELERATED MOTION .. LZ = (No) DECEMBE + SQ (XJECOME - KS)

NOTING THAT 60 THE SHEN WHEN N=50 mi (= & N), HAVE ... (= 88 5) = (88 \$)2+2(-0.924 9 \$) NAGRASE OR XJRGRADE = 1280.16 St

OR XURGRADE = 0.242 mi

GIVEN: QAB = 18 M/5 1; 12.8 (he) no = (he) ec . He FIND: Que

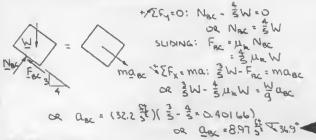
FIRST CONSIDER THE MOTION OF THE PACKAGE ON SECTION AB.

* EF4:0: NAB - 3W:0 OR NABO 3W SLIDING: FAB = ALE NAB WALLS -VEFx=ma: 3W-Fre = mans MR: 3 (= 1845) THEN

= 0.401 66

NOW CONSIDER SECTION BC.

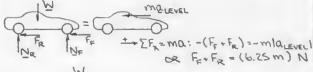


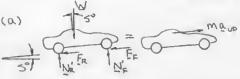


12.9 GIVEN: AN AUTOMOBILE'S BRAKING DISTANCE,

XBR, FROM 90 tem/h ON LEVEL FIND: (a) XBR FROM 90 EM/h FOR A 5° INCLINE - UP bm/h FOR A 3% INCLINE - DOWN

FIRST CONSIDER BRAWING ON LEVEL PAVEMENT. ASSUMING UNIFORMLY DECELERATED MOTION HAVE. 12= (No) + 2 OLEVER (X- X) NOTING THAT 90 km/h = 25 m/s 0 = (25 m) + 20 LEVEL (50 m) HAVE .. OR QLEVEL = - 6.25 32





ASSUMING THAT THE BRAKING FORCE (FF+FR) IS INDEPENDENT OF THE GRADE, HAVE .. == EFx = ma: - (FE+FR) - WSIN 5° = Maup OR - 6.25 m - ma sins = maux THEN QUP = - (6.25+9.81 SINS") =-71050 5 FINALLY - 152 = (No)2 + 2 Que (XBR - 40) SUBSTITUTING .. 0= (25) 2+ 2(-7.1050 52) YER OR XBR = 44.0 m

(b)

EFx = ma: WSIND-(FF+FR) = Manown NOW -. TAND = 0.03 => O SMIALL => SIND = TAND THEN. mg TANB - 6.25 m = masown OR abown= 9.81(0.03)-6.25 =-5.9557 52 FINALLY - 100 - (No)2 + 2 aboun (XBR - XS) SUBSTITUTING - 0= (25 x)2+2(-5.9557 x2)XBR OR XBR = 52.5 M

12.10



GIVEN: M = 20 kg; 15=0; AT 1 = 105, AX= 5 m; MS=0.4, MR=0.3

123

FIRST OBSERVE THAT THE PACKAGE IS UNIFORMLY ACCELERATED SINCE ALL OF THE FORCES ARE

CONSTANT. THEN. Sm= 20(105) AT 12 105: OR a = 0.10 \$

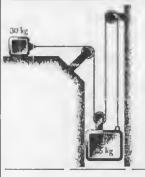
NOW .. +1 EFy=0: N-Wasso-Psinso.0 OR N = mq cos 20 - PSIN SO

SLIDING: F= MEN · Mr (mg cos 20+ P sin 50°) = EFx = ma: Pcosso - Wsin 20 - F = ma THEN .. PCOS SO - mg SINZO - MR (mg cos 20+ PSINSO) - ma P= m[a+9(sin 20+ MR cos 20)] COS SU - MR SIN SU 20/28 [0.10 3 +9.81 32 (5120+0300520)]

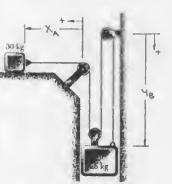
COS 50° - 0.3 SIN 50°

OR P = 301 N

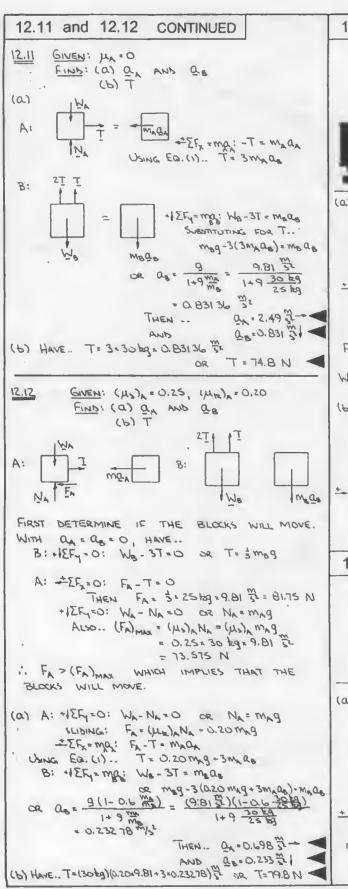
12.11 and 12.12

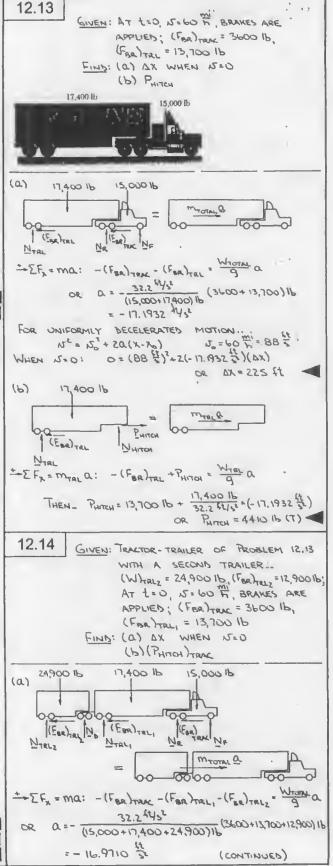


GIVEN: BLOCKS A AND B AND THE PULLEY! CABLE SYSTEM WHICH IS OF NEGLIGIBLE MASS, SHOWN; (NE) = (NE) = 0

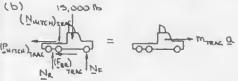


FROM THE DIAGRAM .. THATEHOD = BYE + X THEN .. NA + 3NB = 0 AND Q4+308=0 OR ax=-300 (1)





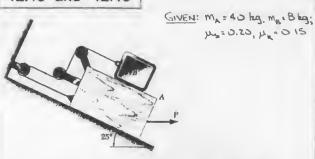
12.14 CONTINUED

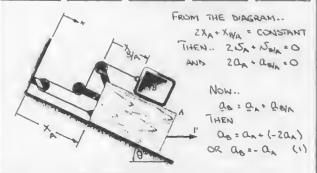


THEN -. (PHITCH) TRAC = -3600 16 - 15,000 (-16,9710 52)

OR (PHITCH) TRAC = 4310 16 (T)

12.15 and 12.16





12.15 GIVEN: P=0, B= 25° FIND: (a) QB (b) T

FIRST BETERMINE IF THE BLOCK'S WILL MOVE FOR THE GIVEN VALUE OF B. THUS, SEEK THE VALUE OF B FOR WHICH THE BLOCK'S ARE IN IMPENSING MOTION OF A DOWN THE INCLINE.



+/ΣFy = 0: NAB - We COS θ = 0

OR NAB = me g COS θ

NOW.. FAB = με NAB

= 0.2 meg COS θ

ΣFx = 0: - T + FAB + We SIN θ = 0

OR T = meg (0.2 COS θ + SIN θ)

(CONTINUED)

12.15 and 12.16 CONTINUED

= $9[m_k \sin \theta - 0.2(m_k + 2m_e)\cos \theta]$ EQUATING THE TWO EXPRESSIONS FOR T... $m_k q(0.2\cos \theta + \sin \theta) = 9[m_k \sin \theta - 0.2(m_k + 2m_e)\cos \theta]$

 $m_B g(\Delta z \cos \theta + \sin \theta) = g[m_A \sin \theta - \Delta z (m_A + 2 m_B) \cos \theta]$ OR $B(\Delta z + 7AN\theta) = [40 7AN\theta - \Delta z (40 + 2 + 8)]$

OR TAND: 0.4 OR θ * 21.8° FOR IMPENDING MOTION. SINCE θ < 25°, THE BLOCKS WILL MOVE. NOW CONSIDER THE MOTION OF THE BLOCKS.

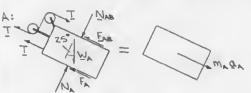
(a) 25°
B: I WB = +12Fy=0: NAB - WB COLZS = 0

FAB. NAB = MB COLZS = 0

SLIDING: FAB = MB MAB.

= 0.15 mg gaszs

T= MB (B) (0.15 COS 25° + SIN 25°) - QB = 8[9.81(0.15 COS 25° + SIN 25°) - QB] = 8(5.87957 - QB) (N)



+/EFy=0: NA-NAB-WA COSES=0

OR NA= (MA+MB) gCOSES

SLIDING: FA= HKNA=0.15(MA+MB) gCOSES

-EFX=MAA: -T-FA-FAB+WASIN 25=MAAA

SUBSTITUTING AND USING EQ. (1).
T=MAGSIN 25-0.15(MA+MB) gCOSES-0.15MB gCOSES

-MA(-QB)

= 9[masin 25-0.15(ma+2mb)cos25]+ maab = 9.81(40 sin 25-0.15(40+2x8) cos25]+40 ab = 91.15202+40 ab (N)

EQUATING THE TWO EXPRESSIONS FOR T.. $8(5.47952-Q_B)=91.152.02+40.Q_B$ $0R Q_B=-0.985.75 \frac{W}{2}^2$

:. Q8 : 0.986 3 - 125 < (b) HAVE.. T= B [5.47952 - (-0.98575)]
OR T= 51.7 N

12.16 CONTINUED

12.16 GIVEN: P = 40 N -, 8.25°
FIND: (a) as
(b) T

FIRST DETERMINE IF THE BLOCKS WILL MOVE FOR THE GIVEN VALUE OF P. THUS, SEEK THE VALUE OF P FOR WHICH THE BLOCKS MRE IN IMPENDING MOTION OF A DOWN THE INCLINE.

B: T WB

+/ΣFy=0: NAB - WB COS 25° 0 OR NAB = MB 9 COS 25° NOW - FAB = μ NAB = 0.2 MB 9 COS 25° -ΣFx=0: -T+FAB+ WB SIN 25° 0

OR T= 0.2 mag cos25+ mag sin25" = (Bkg)(9.81 7)(02 cos25+ sin25) = 47.39249 N

A:

T

ZS°

EAB

A | EFy = 0: NA - NAS - WA COS 25°

T

ZS°

OR NA = (MA+MB) 9 COS 25°

NOW - FA = MS NA

P OR FA = 0: 2 [(MA+MB) 9 COS 25°

-P SIN 25°]

~ EFx = 0: -T - FA - FAB + WA SIN 25 - P COS25 = 0 OR -T - 0.2 [(MA+MB) g COS 25 - P SIN 25] - 0.2 MB g COS25 + MB g SIN 25 + P COS 25 = 0

OR P(U.2 SIN 25°+ COS 25°) . T+U.2[(MA+2MB)g COS 25°] - MAG SIN 25°

THEN $P(0.2 \sin 25 + \cos 25) = 47.39249 N$ $+9.81 \% \{0.2\{(40 + 2 - 8)\cos 25 - 40\sin 25\} | y\}$ OR P = -19.04 N FOR IMPENDING MOTION.
SINCE P < 40 N, THE BLOCKS WILL MOVE. NOW
CONSIDER THE MOTION OF THE BLOCKS.

(a)
B: T We
EAGNAR = Jame Qe

+/Efy=0: NAS-Wacos 25 = 0 OR NAS = Mag cos 25

SUBING: FAB = MR NAB
= D.15 Mag cos 25°

~ [Fx = mb ab: -T + Fxb + Wb sin 25 = mb ab

OR T = mb [9(0.15 cos 25+5in 25) - ab]

= 8[9.81(0.15 cos 25 + sin 25) - ab]

= 8(5.47952-ab) (N)

T ZS° NAR

T ZS° NA ZS° P (CONTINUED)

12.16 CONTINUED

+/EFy=0: NA -NAS -WA COS25+PSIN 25 = 0

CR NA = (MA+MA)9 COS25-PSIN 25

SUDING: FA=14RNA

= 0.15 [(m, o ma)g coo 25- P oin 25)

- EF = maax: -T-FA-FAS + Wa sin 25 + P coo 25 = maaa
Substituting AND USING ED. (1).

T: mag sin 25 - 0.15 [(ma + ma)g cos25 - Psin 25]
- 0.15 mag cos 25 + Pcos25 - ma (-0a)
= g[masin 25 - 0.15 (ma + 2 ma) cos 25]

+ P(0.15 5422 + 0525) + M, Q, + P(0.15 5422 + 0525) + M, Q, + P(0.15 5422 + 0525) + 0525 + 0525 + 000

= 129,940 04 + 40 00 (N)

EDUATING THE TWO EXPRESSIONS FOR T .- 8(5.47952- QB) = 129.940 04 + 40 QB

OR QB =-1.793 83 54

(b) HAVE .. T= 8[5.47952-(-1.79383)] CR T=58.2 N

12.17

GIVEN: AT \$=0, UZ. NZ =0, BELT BEGINS
TO MOVE - SO THAT SLIPPING OF
BOTH BOXES OCCURS; (ME)A = 0.30,
(ME)B = 0.32
FIND: QA AND QB



ASSUME THAT Q8 > QA SO THAT THE NORMAL FORCE NAS BETWEEN THE BOXES IS ZERO.

A: $\frac{1}{15}$ = \frac

B: +1 EFy =0: Ng - We cos 15 = 0

OR Ng = We cos 15°

OR Ng = We cos 15°

OR Ng = We cos 15°

EN Ng = C32 We cos 15°

EN Ng = EFx = mgae: Fe - We SIN 15° = mgae

OR OR Ng = We as 15° - We SIN 15° = We as

OR 08 = (32.2 (2) (0.32 cos 15° - 514 15°) = 1,619 52

Q₈ > Q_A → ASSUMPTION IS CORRECT

.: Q_A = 0.997 5 < 15

Q_B = 1.619 5 < 15

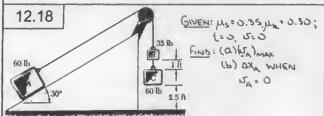
NOTE: IF IT IS, ASSUMED THAT THE BOXES

REMAIN IN CONTACT (NAS + O), THEN

(CONTINUED)

12.17 CONTINUED

a= a AND FIND (EF= ma) A: 0.3 WA cosis - NA SINIS - NAS . Tha B: 0.32 Wg cos 15 - Wa SNIS - Nos . We a SOLVING YIELDS Q=1.273 4432 AND NAS =- 0.859 16. WHICH CONTRADICTS THE ASSUMPTION.

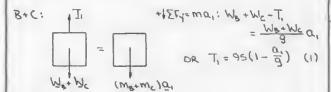


FIRST DETERMINE THE COMBINED MINIMUM WEIGHT OF BLOCKS B AND C: FOR IMPENDING MOTION OF PHOKAGE A UP THE INCLINE.

301WA _ T + 1/2 Fy = 0: NA - WA COS 30 = 0 02 NA = WA COS 30 Now .. FA = Ms NA = 0.35 WA COS 30 2, EFx = 0: T-FA-WA SIN30 = 0 OR T. WA (0.35 CO 530 + YM 30)

W = 60 16 0 T = 482 16 THEREFORE SINCE THIN IS LESS THAN THE (9516) PACKAGE A WILL MOVE UP THE INCLINE WHEN BLOCKS B AND C ARE RELEASED.

(a) "MOTION I .. A, B, AND C MOVE TOGETHER THROUGH 2.5 St.



. 0.3 Wa cos30 = EFx = MA Q : T, - FA - WA SIN 30 = ga a, OR T, = 60 (0.3 cas 30 + 51 N 30 + 0) = 60(0.759 808 + 2)

EQUATING THE TIMO EXPRESSIONS FOR Tz ... 95(1- 317) = 60 (0.759 808 + 217)

a, = 10. 26 48 52

MOTION 2".. C IS AT REST. A AND B MOVE TOGETHER THROUGH I St. FOR THIS CASE, Eas. (1) AND (2) BECOME.. $T_2:35(1-\frac{\alpha}{3})$ (1') (CONTINUED)

12.18 continued

72 = 60(3.759 808 + 3) (2') 35 (1- 22) = 60(3.759808 + 21) THEN

OR 02 = - 3.5889 52 " SINCE Q, CO. A BEGINS TO DECELERATE AFTER BLOCK C REACHES THE GROWD; THUS, (UTA) MAN OCCURS AT THE END OF "MOTION I." UNIFORMLY ACCELERATION OF "MOTION I, HAVE .. 15x2 = (5x3)2 + 2a, (x-X)

WHEN AX+2.5 ft: (5) max = 2 (10.2648 \$)(2.5 ft) OR (4) max = 7.16 \$ 623

(b) FIRST NOTE THAT AT THE END OF MOTION 2,

THE SPEED OF PACKAGE A 15..

(NOTE: (NA) max + 202 6x2

= (7.1641 52) 2+ 2(-3.5889 52)(1 ft)

OR (NA) 2 6.6443 35

"MOTION 3 .. B AND (ARE AT REST, A CONTINUES UP THE INCLINE AND FINALLY COMES TO REST. FOR THIS CASE, T=0 30 THAT EQ (2) BECOMES $(60(0.759808+\frac{0.3}{3})=0$ (2"),

THEN- 03= -0.759808 (32.2) = -24.466 SE

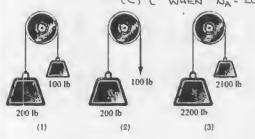
THEN .. Unic = (Unic + 20,5(X-X))3 WHEN Unic = 0: 0= (6.6443 =)+2(-24.446 =)0x3 OR AX = 0.9022 ft

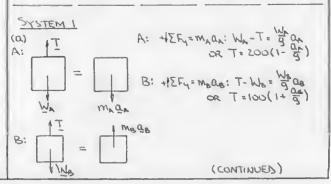
THE TOTAL DISTANCE DXA TRAVELED BY A UP THE INCLINE BEFORE COMING TO REST IS

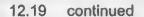
11/250P.0 + 1+2.5 1 = 2XA+2XB+1XB = AXA OR DXA = 4.40 ft

12.19 GIVEN: THE THREE SYSTEMS SHOWN; WS = 0 FIND (FOR EACH SYSTEM): (a) an

(b) NA WHEN BYA = 10 ft (C) + WHEN NA : 20 5







EQUATING THE TWO EXPRESSIONS FOR T AND NOTING THAT IQAI = 1 asl ... $ZOO(1-\frac{QA}{9}) = 2OO(1+\frac{QA}{9})$ or $Q_A = \frac{1}{9}9 = \frac{1}{3}(32.2 + \frac{1}{5}e) = 10.7333 + \frac{1}{3}$ an=10.73 52 (b) HAVE - UT = (NA) + 20, (4.40)

WHEN DY = 10 ft: 12 = 2(10.7333 \$)(10 ft) OR NA = 14.65 15

(C) HAVE .. No (15) - Out WHEN 15 = 20 15: 20 1/3 . (10,7533 50)} OR t = 1.863 5

SYSTEM 2 + /EFy=maa: Wa-T. Taan (0) J =1001b OR .. On = (38.2) (1- 100) OR Q4 = 16131 MAGA

(b) HAVE -. 152 = (NO) = 20x(4-40) OR 5 = 17.94 5

(C) HAVE .. Up = (ph) + ant by the NHEN Up = 20 = 20 17 = (16.1 52)t OR t - 1.242 5

SYSTEM 3

(0) A: +1 EF = maa: Wa-T - Gas AI OR T= 2200(1- 0) B: +1 E Ky = M& as: T- Wo = g as OR T: 2100 (1+ 2)

+ ma as EQUATING THE TIMO EXPRESSIONS FOR T AND NOTING THAT | Q_ |= |Q_ | = ...

Z200(1- \(\frac{a_0}{4} \)) = Z100(1+ \(\frac{a_0}{4} \))

OR \(Q_0 = \frac{43}{3} \) = \(\frac{43}{3} \)(\$2.2 \(\frac{a_0}{3} \) = 0.748 84 \(\frac{43}{3} \)

1. ax = 0.749 51

(b) HAVE .. WE = (PA) = 202 (4-40) WHEN DY = 10 ft: " 5 = 2(0.748 BA (t))(10 ft) OR NA = 3.87 5

(C) HAVE .. 5 = (U) + ant WHEN JE 20 \$: 20 \$ = (0.748 84 5) } OR t = 26.7 5 12.20

GIVEN: Q . CONSTANT; ma = 3 kg; MOTION OF B IS IMPENDING; My = 0.30, M = 0.25 FIND: (Q) QEL WHEN GEL! MAD NA= Nec= 2 No (b) NAS AND NOC WHEN QELESON

FIRST OBSERVE THAT BECAUSE B IS NOT MOVING RELATIVE TO A MIS TO C THAT QB . QEL.

(2)

HAVE .. F= MSN = 0.30 (ZWa) = 0.6 WB = 0.6 MBQ FOR QEL TO BE! THE NET YEXTICAL FORCE MUST

BE ! WHICH REDURES THAT THE FRICTIONAL FORCES BE ACTING AS SHOWN. IT THEN FOLLOWS THAT THE IMPENSING MOTION OF B RELATIVE TO A AUG C IS DOWNWARD. THEN ..

+ Ety = MB aEL: 2F - WB = MBAEL 02 2(0.6 mgg) - mgg = mgaEL azl = 0.2 = 9.81 002

OR QEL = 1,962 52

(b) ?E ,

HAVE -- F= MSN : 0.30 N

NOW OBSERVE THAT BECAUSE THE DIRECTION

OF THE IMPENDING MOTION IS UNKNOWN, THE DIRECTIONS OF THE FRICTIONAL FORCES IS ALSO UNKNOWN (ALTHOUGH FIRET MIUST BE DOWNWARD) + IEFy = MBaEL: + ZF - WB =-MBI agel = 2F = m8 (9-1acc1)

= 3 kgx (9.81-2) 52

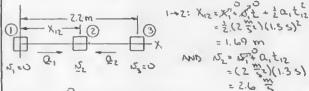
SINCE THE MAGNITUDE OF F MUST BE POSITIVE IT THEN FOLLOWS THAT F ! AND THAT THE IMPENDING MOTION OF B RELATIVE TO A AUD C IS DOWNWARD. FINALLY .. 2 (0.30 N) = 3 kg x (9.81-2) 5

OR NAB = Nac = 39.1 N



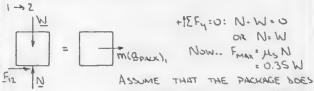
(P) JUNCHURE BELT WHEN

(a) FOR THE UNIFORMLY ACCELERATED MOTION OF A POINT ON THE BELT HAVE ..



 $2 \rightarrow 3!$ $0 = (2.6 \frac{3}{5})^{2} + 20_{2}(x_{3} - x_{1})$ $0 = (2.6 \frac{3}{5})^{2} + 20_{2}(2.2 - 1.69) \text{ M}$ $0 = 0 = 0.62745 \frac{3}{5}^{2}$

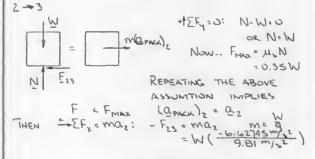
(b) Now consider the process for each portion of the motion



NOT SLIP NOR IS IN IMPENDING MOTION RELATIVE TO THE BELT.

THEN FIZEF MAI: (amck) = a, AND _ = EFx = Mai: FIZ = Mai 2 m/s = 3 = W (7.81 m/s) = 0.204 W

.. F_{12} (0.204 W) \leftarrow F_{MAX} (0.35 W) \Rightarrow ASSUMPTION IS CORRECT (NO SUPPING) SO THAT \leftarrow CXPACUMCE/BELT)12 \approx O



OR F23 = 0.676 W

THEN ... 0.35 W AND THE BELT COMES TO REST.

OR N2 = MA(Q)

OR N2 = MA(Q)

N2 F2 -> EFx = MA(X): -F2 = - MA(X)

N2 F2 -> EFx = MA(X): -F2 = - MA(X)

THEN ... 0.3[MA(Q - Q2511165)] = MA(X) COSES

OR Q2 = 0.3(9.81 M/S)

(CONTINUED)

12.21 continued

THEN. SUPPNIC: $F_{23} = \mu_e N$ = 0.25 mg $\xrightarrow{}$ $\sum F_x = m(\alpha_{pack})_2 : - F_{23} = m(\alpha_{pack})_2$ OR $-0.25 mg = m(\alpha_{pack})_2$ OR $(\alpha_{pack})_2 = -0.25 (9.81 m/s^2) = -2.4525 m/s^2$ NOW. $(\alpha_{pack})_2 = \Omega_2 + (\alpha_{pack})_{20} = -2.4525 m/s^2$ OR $(\alpha_{pack})_2 = -2.4525 m/s^2 - (-6.62745 m/s^2)$ $= 4.17495 m/s^2$ FOR THE BELT. $M_3 = M_2 + \alpha_2 t_{23}$ OR $0 = 2.6 m/s + (-6.62745 m/s^2) t_{23}$ OR $t_{23} = 0.392315$

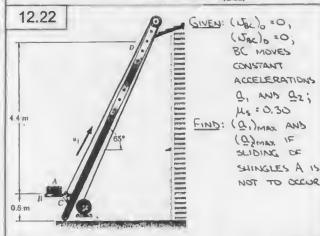
THEN. (XPACKIBELT) 23 = 23 + (-6.62745 52) t23

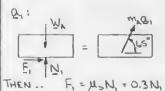
THEN. (XPACKIBELT) 23 = 23 + 26 23 + 2(0 million 1) 2 t23

= 2 (4.17495 52) (0.392315)

= 0.321 m

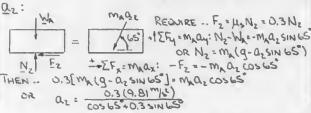
FINALLY.. XPACUAGE BELT (XPACUBELL) 12 (MINUSCHI) 23
OR SPACUAGE BELT = 0.321 M ->





NOTE THAT THE DIRECTION OF \underline{G}_1 FIXES THE DIRECTION OF \underline{F}_1 AND THAT FOR $(\underline{G}_1)_{max}$, $\underline{F}_1 = (\underline{F}_1)_{max}$

OR (Q1)max 19.53 \$2 165°



CR (Q2)MAX: 4.24 5- 765



GIVEN: 1=0, 15=0; 16=0.40, -HR 20,30

FIND: (a) (QTRIER) MIN SO THAT PLYWOOD SLIDES

> (b) arever so THAT M S : MYSTIROUNIAKE AT 1 = 0.95

(a) SEEK THE VALUE OF QTRUCK SO THAT RELATIVE MUTION OF THE PLYWOUS WITH RESPECT TO THE TRUCK IS IMPENDING. NOTE .. Q PLYWOOD = Q TRUER

+ 12 Fy = mply ay: N- Woly cos 20 = - Mply ate sinzo OR Nompey (9 cos 20 - atr SIN 20)

+offx = MALY ax: F-WALY SIN 20 = MALY aTR COS 20 OR F = MPLY (9 SIN 20 + QTR COS 20) SUBSTITUTING INTO EQ. LI) ..

MPLY (9 SINZU+ QTR COSZO) = 0.4 MPLY (9 COS 20- QTR SINZU) OR OTR = 9(04 65 20 - 51N 20) = (9.81 32) 0.4 - TANZO
1+0.4 TANZO

OR (QTRUCK) MIN = 0.309 50 -

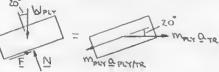
(b) FIRST NOTE THAT BECAUSE ALL OF THE FORCE'S ARE CONSTANT, THE ACCELERATIONS ARE ALSO CONSTANT. THEY-.

XPLYITE = (XPLYITE) + 45t + 2 aprilie t2 AT t= 0.9 5: 2m = 2 apryTR (0.9 5)2

amytra = 4.93827 5 720 DR.

NOW -apry = are + aprylin

THEN



HAVE .. F = MEN = 0.3 N

== EFX = MPLY ax: F- WPLY SIN ZO = MPLY (aTR COSZO-anyTER) OR F = mary (9 SIN 20 + are cos 20 - aprilia)

+12 Fy = May ay: N- Wary cos 20 = - MARY OTR SIN 20 OR N= mpy (9 cos 20 - QTR SIN 20)

SUBSTITUTING INTO ÉQ. (1) .. MPLY (9 SIN 20 + QTR COS 20 - QPLY/TR)

OR are = 3(0.3 cos 20 - SIN 20) + anxite COS 20 + 0.3 51N 20

(9.815) (0.3cos 20 - SIN 20) + 4.938 27 52 COS 25 4 0.3 SIN 20°

OR arriver = 4.17 52 -

12.24

GIVEN: SHIP OF WEIGHT WI HAVING A PROPULSIVE FORCE Fo; AT t=0, 15. No (" Noman), FORWARD, ENGINES ARE REVERSED; FWATER A 152

FIND: X WHEN 5=0

FIRST CONSIDER WHEN THE SHIP IS MOVING FURWARD.

WHERE & IS A CONSTANT Fo - kut = 0, or k = For

NOW CONSIDER WHEN THE SHIP IS DECELERATING.

N EWATER == [F = ma: - F - Fwares = qa OR a = - \frac{9}{W} (F_3 + \frac{F_0}{45} U^2) = - \frac{9}{15} \frac{F_0}{15} (U^2 + U_0^2) Now .. 12 du = 0 = - 2 10 (2+ 2) AT too, X=0, 5=50: $X = -\frac{5}{22} \frac{E}{M} \left[\frac{5}{7} \Gamma M (\Omega_5 + \Omega_5^2) \right]_2^2 = -\frac{5}{7} \frac{3}{22} \frac{E}{M} \Gamma M$

12.25



GIVEN: CONSTANT FORCE P; PISTON AND ROD OF MASS M; FOIL = EU; AT 1:0, X=0, 5:0 SHOW: F(X,N,t)=0 IS LINEAR

IN X, W, AND t

== EFx = ma: P-Foil = ma OR a=m(P-R5)

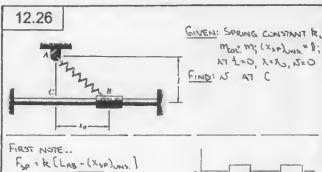
Now -: 2= = a = 4 (P- 12) AT t=0, 5=0: [dt = m] P-RS

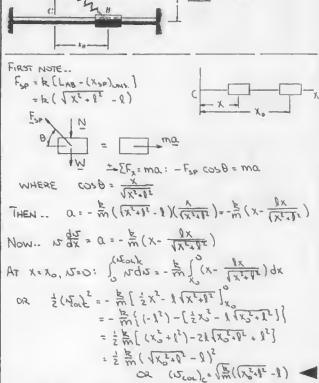
OR t=m[- + LN(P-ku)] OR t=- EW P- EU ALSO. 10 du = a = 1 (P- 20) AT X=0, 500: [Xdx = m] P-RV

OR X= M [[- k + k(P-ku)]du] = m[- = - [rn(b- pa)], =- W(F + ES (N B- FR)

USING EQ. (1) - X = - TE + Et XR+MN-Pt = 0

WHICH IS UNEAR IN X, N, AND 2.





12.27 GIVEN: AUTOMOBILE WEIGHING 2700 lb, FRONT-WHEEL DRIVE, WER = 0.62W; 4= 0.70, D=0.0122 D-16, 5-24; AT 20, X=0, 5=0 FIND: Nome WHEN X= 0.25 mi

F = Fmax FOR 5 = Smax : F = Ms NF = 0.70 (0.62 W) NF = 0.62W

= 0.002 \(\) (217 W - 622)

Now.. 10 dx = a = 0.002 & (217W-652)
AT X=0, 5=0: 0.002 & 3 dx = 10 217W-652

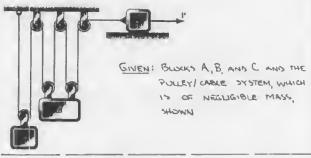
OR 0.002 W X = - 12 [LN(217 W-642)]=-12 LN(217 W-642)

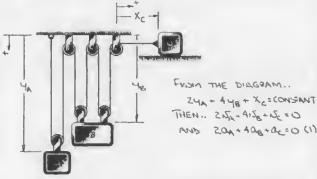
217W-652 -0.024 WX OR N=[21) W (1- 0.02+ 0 x)]1/2

WHEN X=0.25 mi = 1320 ft: NMAX = \left(\frac{217}{6} (2700)(1-\frac{2}{6} 0.024 \frac{32.2}{2700} \cdot 1320)\right)^{1/2} = 175.285 \frac{57}{5}

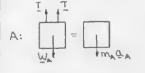
OR JMAX = 119.5 Th

12.28 and 12.29





12.28 GIVEN: mx = 4 kg, mx = 10 kg, mx = 2 kg; FIND. (a) QA, QB, AND QC



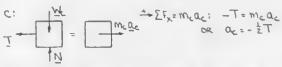
· IEFy= maan: Wa - 2T = Maan OR an = mx (mxg-2T)

3:
$$\frac{2T}{4} = m_B a_B$$
: $W_B - 4T = m_B a_B$

$$\approx a_B = \frac{4\pi}{m_B} (m_B q - 4T)$$

$$= q - 75T$$

$$= q - \frac{1}{5}T$$



SUBSTITUTING THE EXPRESSIONS FOR QA, QB, AND QC INTO Ea. (1) ..

2(9-2T)+4(9-3T)+(-2T)=0 OR T= 309 = 51 (9.81) = 18.9871 N (a) THEN.. ax=9.81-2(18.9871)

OR an = 0.316 521 as = 9.81 - \$ (18.9871)

OR QB = 2.22 521 ac = - = (18.9871)

OR ac=9.49 52-(b) HAVE ... T=18,99 N

(WINTINUED)

12.29. GIVEN: MA = B kg, MB = 16 kg, MC = 10 kg;

MA = 0.30, MA = 0.20; AT t=0, N=0;

AT t=0.85, Ayb = 2 ml

FIND: (a) QA, QB, AND QC

(b) T

(c) P

(a) First note that because all of the forces are constant, all of the accelerations are constant. Then...

40 = (40) + (40) t + 200t2 AT t=0.85: 2m=200(0.85)2 OR Q0 = 6.25 52

: as 26,25 52 1

A:
$$T = T = m_A \alpha_A$$
: $W_A - 2T = m_A \alpha_A$

or $m_A g - 2T = m_A \alpha_A$

or $g - 2T = g \alpha_A$
 $g = g - 2T$

Comparing Eqs (2) and (3), IT FOLLOWS THAT $Q_A = Q_8$ $\therefore Q_A = 6.25 \frac{m}{2}$

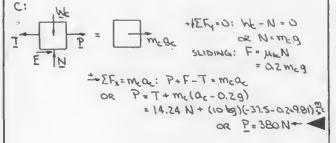
Substituting into Eq. (1).. 2(6.25) + 4(6.25) + 4 = 0 $4 = -37.5 \frac{m}{5}$

. ` ac=37.5 ₹ --

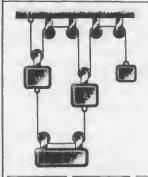
(b) from Ea (2) -- T= 4(9-02) = 4(9.81-6.25)

. OR T = 14.24 N

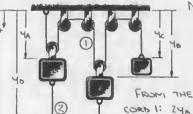
(0)



12.30 and 12.31



GIVEN: BLOCKS A.B.C., AND B AND THE PULLEY/CABLE SYSTEM, WHICH IS OF NEGLICIBLE WEIGHT, SHOWN; WA = WB = 20 16, WC = 14 16, WB = 16 16



NOTE: AS SHOWN, THE SYSTEM IS IN EDUILIBRIUM.

FROM THE DIAGRAM..

CORD 1: 24x + 24x + 4c = CONSTANT

THEN.. 2Nx + 2Nx + Nc = 0

ANX 20x + 20x + 0c = 0 (1)

CORD 2: (4y-4x) + (4y-4x) = CONSTANT

THEN.. 2Nx - Nx - Nx = 0

NO 200-04-08=0 (2)

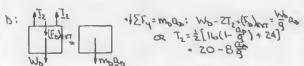
12.30 GIVEN: (FD) ET = 24 16 +

FIND: (Q) QA, QB, QC, AND QB

(b) T, (= TABC)

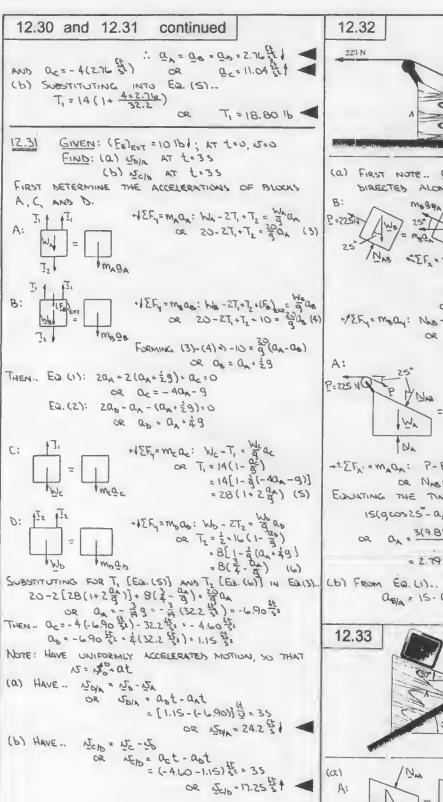
(a)
$$T_1$$

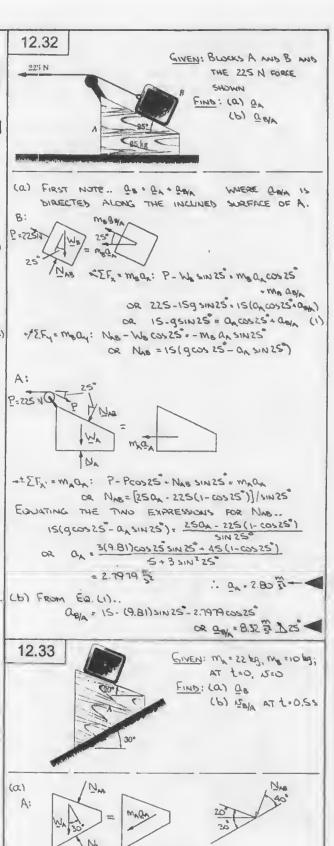
A: $W_1 = W_2$
 $W_2 = W_3$
 $W_4 = W_4 = W_4$
 $W_4 = W_4 = W_4$
 $W_5 = W_6$
 $W_6 = W_6 = W_6$
 $W_7 = W_8$
 $W_8 = W_8 = W_8$
 $W_8 = W_8$



SUBSTITUTING FOR T, [ED. (S)] AND T_2 [ED. (b)] IN EQ. (3)...

20-2-14(1+ $\frac{40}{9}$)+ (20-8 $\frac{3}{9}$) = $\frac{20}{9}$ $\frac{2}{9}$ $\frac{2}{$





WASIN 30 + NAB COSAO - MY ON

(CONTINUED)

OR NAR = 22 (OA - 19)

12.33 continued

NOW NOTE: QB = QA + QBIA WHERE QBIA IS DIRECTED ALUNG THE TOP SURFACE OF A.

B:



~ ΣFx, = Mg ax: Wg SINZO = Mg agg - Mg ag cos So OR agg = 9 SIN 20 + 0, cos So = (9.81 SIN 20 + 6, 4061 cos So) gz = 7.4730 π

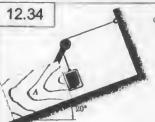
QA - 201 20 QB AND

HAVE $Q_{0}^{2} = 6.40ca^{2} + 7.4730^{2}$ $-2(6.40ca)X7.4730)cos 50^{2}$ OR $Q_{0} = 5.9447 \frac{M}{2}$ AND $\frac{7.4730}{51N} = \frac{5.9447}{51N} \frac{50}{50}$ OR $Q_{0} = 74.4^{2}$ $Q_{0} = 74.4^{2}$ $Q_{0} = 5.94 \frac{M}{2} \sqrt{7.15.6^{2}}$

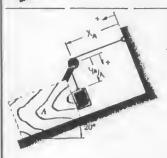
(b) Note: Have uniformly accelerated motion,

50 THAT $S = 55^{\circ}$ Qt

Now... $S_{MA} = S_{A} - S_{A} = S_{A} + S_{A}$



GIVEN: WA = 50 16, Wo = 30 16 FIND: QA AND T



From the diagram.. $X_A + Y_{BA} = CONSTANT$ THEN.. $N_A + N_{BA} = 0$ AND $Q_A + Q_{BA} = 0$ $CR Q_{BA} = -Q_A$ (1)

FIRST NOTE: QB = QA + QBA WHERE QBA IS DIRECTED ALDING THE SIDE OF A

(CONTINUED)

12.34 continued

B: AT

No Mo QA

- ΣFx = mgax; Wg sin 20 - NAB = mgax CR NAS = Wg (sin 20 - 3) - ΣFy = mgay; Wg cos 20 - T = mg agy 2 agy T = Wg (cos 20 + ag)

A:

EFx=maax: Wasinzo-Nas-T

Wasinzo-T

Wasinzo-Nas-T

Wasinzo-Nas

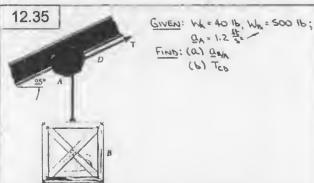
Now substitute THE EXPRESSIONS FOR No AND T INTO ED. (2)...

SO SIN 20 + 30(SIN 20 - $\frac{a_1}{3}$) - 30(cos 20 + $\frac{a_1}{3}$) - 50 $\frac{a_2}{3}$ OR $a_1 = \frac{1}{11}(32.2 \frac{54}{51})(8 \sin 20 - 3 \cos 20)$ = -0.24272 $\frac{54}{51}$

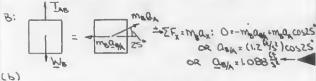
: Q_A=0.243 \$\frac{4}{5^2} \rightarrow 200 \\

\[
\text{T=(30 lb)(cos 20 + \frac{-0.24272 \frac{4}{5^2}}{32.2 \frac{4}{5^2}})}
\]

OR T = 28.0 16



(a) FIRST NOTE: QB · QA · QBA WHERE QBA IS DIRECTED PERPENDICULAR TO CABLE AB



A: N. N. Jes

FOR CRATE B..

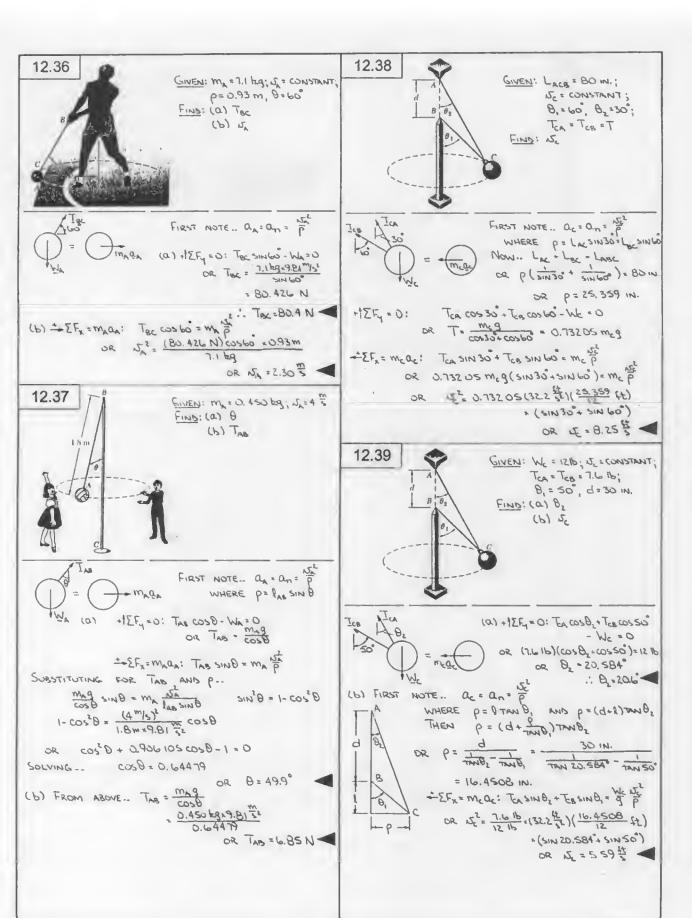
* [EFy = Mody: Two - Wo = Fact on 250

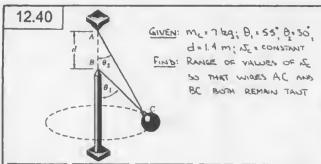
OR This = (500 Ib) [1+ (1.2 4/2) 5w25]

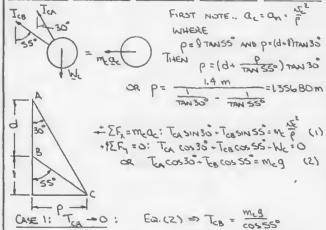
= 507.87 Ib

OR Tob = (507.87 16) SIN 25 + 1.2 Hys. + (4016)(51N 25+ 22.24/52)

OR Tcb = 233 16







SUBSTITUTING INTO ED. (1) ... \(\frac{meg}{cos 55° 5 \in 55° 2 \text{ Me \frac{NE^2}{P}}}{cos 55° 5 \text{ Me \frac{NE^2}{P}}}\)

OR (\(\frac{NE}{E}\)\)\(\text{Text = 0} = 4.36 \frac{50}{5}\)

NOW FORM (COSSO)(1) - (SIN30)(2)...

TO SINSS COSSO - TO COSSO SINSO = M. JE COSSO - M. JENSO

OR TOB SIN2S = M. JE COSSO - M. J. SIN30

.'. (NE) MAX OCCURS WHEN TOB = (TOB) MAX, WHICH

OCCURS WHEN TON = 0.

(IE) MAX = 4.36 \$ AND WIRE AC WILL BE TAUT IF No 4.36 \$.

CASE 2: Teg =0: Eq. (2) => Tex = meg.

SUBSTITUTING INTO ED. (1) ... " COX 30" = MC P2"

OR (NC2) = (1.35680 m) (9.81 52) THU 30"

OR (NC) = 277 5

NOW FORM (cos 55)(1)-(sin 55)(2).

En sin 30 cos 55-Ten cos 30 sin 55 = me = cos 55-megsin 55

OR - Ten sin 25 = me = cos 55-meg sin 55

... (LE) min occurs WHEN Ten = (Ten) max, WHICH

OCCURS WHEN TEB = 0.

.. (JE) min = Z. ?? " MAD WIRE BC WILL BE TANT IF NE > Z. ?? " ...

:. BOTH WIRES ARE THUT WHEN 2.77 \$ < 4.36 \$

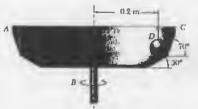
12.41

GIVEN: Mo = 0.1 kg; NS = CONSTANT

FIND: RANGE OF VALUES OF NS SO THAT

NEITHER OF THE MORMAL FORCES

EXCEEDS 1.1 N



No No mo Qu

FIRST NUTE - Q = Q , 2)

 $-\frac{1}{2} \sum_{k} \sum$

CASE 1: N. IS MAXIMUM LET N. = 1.1 N

EQ. (2). (1.1 N) SULO + NZ SINZO = (0.1 kg)(9.81 52)

OR NZ = 0.082954 N

: (NE WI) MAX < 1.1 N -. O.K.

EQ. (1) ... = 3.2 m (1.1 cos 60+0.082954 cos 20) N

2 (No) N. MAX = 1.121 5

: (5) min occurs when N, = (N.) max : (S) min = 1.121 3

CASE 2: No IS MAXIMUM

Ea. (2) ... N, sinka + (1.1 N) sinza = (0.1 kg)(9.81 x2)

OR N, = 0.698 34 N

: (NI)(NE)MAX < 1.1 N .. O.K.

Ea. (1) ... (50) Nelman 2.2 m (0.69834 cos60+1.1 cos20) N

DR (156)(NE)/ME = 1.663 5

NOW FORM (\$1060)(1) - (cos60)(2)...

Now FORM (\$1060)(1) - (cos60)(2)...

No cos60 - No sin 60 - No g cos60

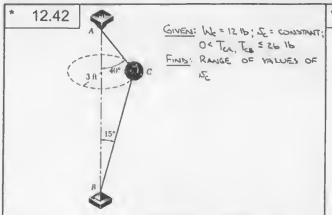
Cos No cos40 = No sin 60 - No g cos60

Cos60 - No max occurs WHEN No = (No) max

Cos60 - No max occurs WHEN No = (No) max

: FOR N, N2 5 1.1 N

1.121 3 5 5 5 5 1.663 5



FIRST NOTE.. Qc=Qn= DE

AO (A) = mcQc

INHERE p= 3 ft

TCB + EFx=mcQc: Ensin40+TcBsin15= Ne NE

TCB + EFx=mcQc: TcA cos 40-TcBcos15-Nc=0 (2)

NOTE THAT EQ. (2) IMPLIES THAT

(3) WHEN TCB= (TcB)max, Tch= (TcA)max

(b) WHEN TCB= (TcB)min, Tch= (TcA)min

(ASE 1: TCA IS MAXIMUM

LET TCA = 26 1b

ED. (2)... (26 1b) cos 40 - Te cos 15 - (12 1b) = 0

OR TCB = 8.1964 1b

... (TcB)(TCA)(MAX = 26 1b -.. OK ((TCB)(MAX = 8.1964 1b))

ED (1)... (152)(TCA)(MAX = 12.51) (26 SIN 40 + 8.1964 SIN 15°) 1b

OR (NE)(TCA)(MAX = 12.51)

OR (NE)(TCA)(MAX = 12.51)

NOW FORM (COSIS)(1) + (SINIS)(2)...

TEM SIN 40 COSIS * TEM COSAD SINIS = ME IS COSIS * WE SINIS

OR TEM SINSS = ME IS COSIS * WE SIN IS (3)

... (NE) MAX CCCURS WHEN TEM = (TEM) MAX

... (NE) MAX = 12.31 \$\frac{1}{2}\$

CASE 2: TEA IS MINIMUM

BECAUSE (TCA) MIN OCCURS WHEN TCB = (TCB) MIN,

LET TCB = 0 (NOTE THAT WIRE BC WILL NOT BE TAUT).

EO. (2)... TCA CUS 40 - (12 lb) = 0

OR TCA = 15.6649 lb < 26 lb... OX

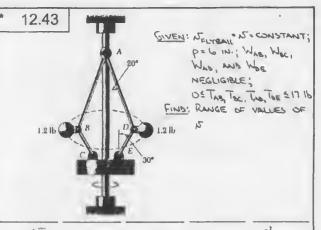
NOTE: EQ (3) IMPLIES THAT WHEN TCA = (TCA) MIN,

NE = (UC) MIN. THEN...

EQ. (1)-- (NC) MIN = (322 TC) (3 St) (15.6649 lb) SIN 40

OR (NC) MIN = 9.00 ST

9.00 \$ < DE \$ 12.31 \$\$



FIRST NOTE .. a.a. p 20 WHERE P: 0.5 ST 42Fy=0: Toy cos 20-ToE cos 30-WOO NOTE THAT ED. (2) IMPLIES THAT (a) WHEN TOE: (TOE) MAX, TOA: (TOA) MAX (b) WHEN TOE: (TOE) MIN, TEA: (TEA) MIN CASE 1: Top IS MAXIMUM LET TOA = 17 16 EQ (2) .. (17 16) cos 20 - TOE COS30 - (1.2 16) = 0 OR THE = 17.06 16 .. UNACCEPTABLE (>17 16) Now LET TOE = 17 16 EQ (2) .. Ton cos 20 - (17 16) cos 30 - (1.2 16)=0 OR TOA = 16,9443 1b .. OK (KI7 1b) : (TOA) MAX = 16.9443 16 (TDE) MAX = 17 16 Ea. (1) ... (16.9443 SIN 20 + 17 SIN 30) 16 OR NITON 13.85 5

Now FORM (COS30)(1) + (SIN30)(2).
TOA SIN20 COS30 + TOA COS20 SIN30 = MUT COS30 + WSIN30

OR TOA SINSO = M TO COS30 + WSIN30

(3)

MMAN OCCURS WHEN TOA = (TOA)MAN

: DMAX = 13.85 5

CASE 2: Top IS MINIMUM

BECAUSE (TOA)MIN OCCURS WHEN TOE: (TAE)MIN,
LET TOE = 0.

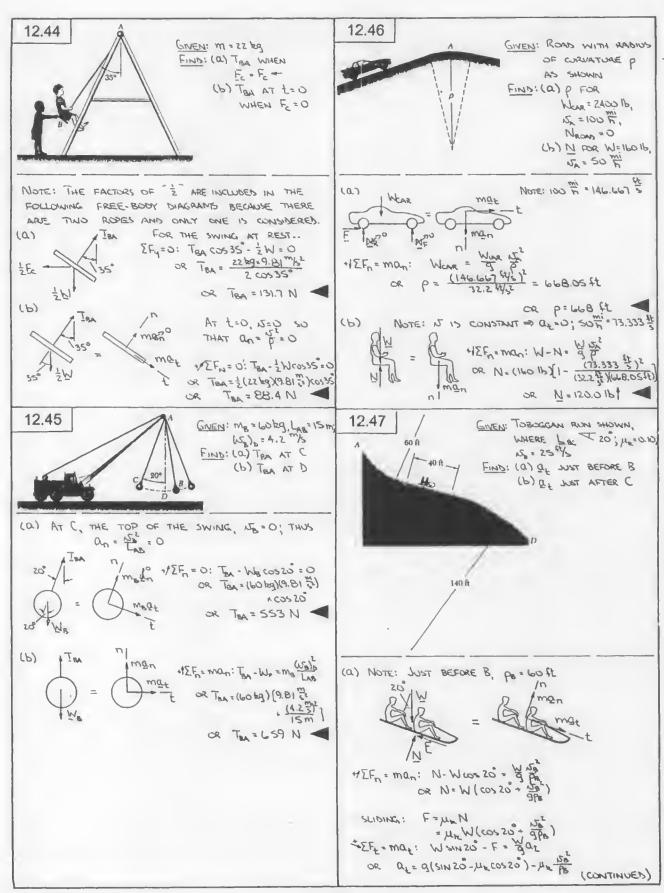
E.Q. (2) .. TOA COS 20 - (1.2 16) = 0

OR TOA = 1.27701 16 < 17 16... OK

NOTE: EQ. (3) IMPLIES THAT WHEN TOA: (TOA)MIN,
N = NOMIN. THEN..
E.Q. (1) (N²)MIN = (52.2 (1) (0.5 (1)) (1.27701 16) SIN 20

OR NOMIN = 2.42 5

:. 05 TAB, TBC, TAD, TOE 5 17 16 WHEN 2.42 EL 5 N 5 13.85 Et





THEN .. Q = (32.2 5) (SIN 20 - 0.1 COS20) - 0.1 (25 5)2

(b) IT IS FIRST NECESSARY TO DETERMINE A.



*/EFy = 0: Noc - Wcos 20° = 0 or Noc = Wcos 20°

SLIDING: FBC = μR NBC = μR WCOS 20° = ΣFx = marc: WSIN20 - FBC = 9 ABC OR OBC = 9(SIN20 - μR COS 20°) = (32.2 ⁶/₂ ⁶) (SIN20 - 0.1 COS 20°) = 7.9B72 ⁶⁴/₂ ⁶

FOR THIS CHIFORMLY ACCELERATED MOTION HAVE ...

NE - No + 20BC AXEC ...

NE = NS + 20BC AXEC = (25 1/5) 2 + 2(1.9872 1/52)(40 ft) OR NE = 35.552 1/5

NOW .. JUST AFTER C, R = 140 ft



*/ΣΕη = Man: Wcos 20 - N - 9 ρ ρ)

SLIDING: F=μ_EN (cos20 - gp.) -Σξ=ma_ε: WsiN20 - F= ga_ε

OR Q1 = 9(SINZO - 12 COS 20) + 12 1/22

NOTE: 9(51N20- MECOS 20) = Que THEN-- QL = 7.9872 5 + 0.1 (35.552 9/6)2

oe āt = 8.89 \$ 1 150€



Й→0 П=Д: ННЕИ д=30'

FIND: (a) No.
(b) FORCE EXERTED ON
THE SURFACE BY
THE BLOCK WHEN

(a) WHEN 8=30...

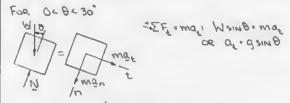
WEFn=man: Wcoc30=m 530

mal

mal

(CONTINUED)

12.48 continued



Now- Qt = N dis AND ds = PdB

THEN Dis = PQ (20030 - 2)

THEN Dis = PQ (1-0050)

THEN Dis = PQ (30030 - 2)

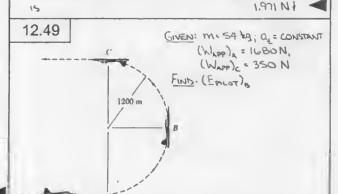
THEN Dis = PQ (30030 - 2)

THEN DIS = PQ (30030 - 2)

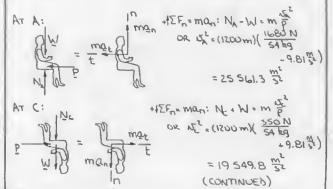
= 8.8006 m/52

(b) WHEN $\theta = 0$.. $V = V_0 = 291 \frac{\pi}{5}$ $V = V_0 = 291 \frac{\pi}{5}$

IN THE FORCE EXERTED ON THE SWEFACE BY THE BLUCK



FIRST NOTE THAT THE PILOT'S APPARENT WEIGHT IS EQUAL TO THE VEXTICAL FORCE THAT SHE EXERTS ON THE SEAT OF THE SET TRAINER.



12.49 continued

SINCE Q: CONSTANT, HAVE FROM A TO C .. UR 19 549.8 m/s = 25 561.3 m/s + 20/ (TI = 1200 m)

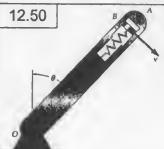
THEN FROM A TO B ... NB = NA + 201 USAB = 25 561.3 me +2 (-0.797 30 m) (2x1200 m)

AT B: OR NB=1014.98 N=

+1EF = may: W+ ? = m1a,1 or P= (54 kg) (0.79730-9.81) 52 or P8= 486.69 N1

FINALLY .- (FPILOT) = (No + Po = (1014.98)2+(486.69)2 = 1126 N

OR (Epiral) = 115P N 7 SZP



GIVEN: WR = 0.5 16; B . CONSTANT; WHEN 8.180, NEWS : 0816 FIND: RANGE OF VALUES OF 8 SO THAT NEACE = 0

FIRST NOTE THAT B CONSTANT => No = CONSTANT => at = 0 WHEN 8=180:

M (E26)100.

+12Fn = man: (NFACE) 180° + (FSP) 180° - W = m 150°

FOR AN ARBITRARY VALUE OF 0:

FF = man: NERCE + For + WOOSE = m p

NOW .. AS BLOCK B LOSES CONTACT WITH THE CAVITY AT A. NEACE -O, FOD = (YEAR) 180° + (FO) 180° - W (= M AMM)

THEN ... (FSP) 180° + WCOSD = (NEACE) 180° + (FO) 180° - W (= M AMM) NEACE -O, FED = (FEP) BS, P= PMAN cos0 = (NEACE) 180 -1 = 0.816 -1 = 0.6

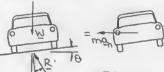
: BLOCK B IS NOT IN CONTACT WITH THE FACE OF THE CAVITY AT END A WHEN -53.1 5 8 5 53.1°

12.51

GIVEN: CAR TRAVELING AT A CONSTANT SPEED NO ON A ROAD BANKED AT AN ANGLE

FIND: RANGE OF VALUES OF IT SO THAT THE CHR DOES NOT SKID; W= f(r, 0, 45)

CASE 1: N= NMAX

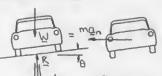


NOTE: R . F . N

== EF_= man: RSIN(0+A) = m - (1) +12F4 = 0: RCOS(8+45) - W= 0

RSIN(0+42) = My 15mm (2)
RCOS(0+42) = My 15mm (2)
RCOS(0+42) = My 15mm FORMING (1) .. OR NMAX = 1 gr TAN (8+ 45)

CASE 2: N= SMIN



NOTE: R = F+N

+ ΣFn= man: R SIN (0- φs)= m - (3)

+1EFy=0: RCOS(B-45)-W=0 OR RCOS(B-A). mg (4) FORMING (3) RSIN(B-4) m Smin CR JMIN: 195 TAN(B-A)

". FOR THE CAR NOT TO SKIB ..

(OF TAN (8-96) & U & 195 TAN (8+96)

12.52

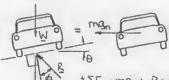
GIVEN: 5=95 th; 1=40 m; 11=0.70 FIND: (a) ON FOR NO SKIDDING WHEN 8=10 (b) DU FOR NO SKIDDING WHEN 9:-5°



FIRST NOTE .. THUPS: 0.70 (-US) OR \$ = 34.992°

ALSO, REDURING THAT THE SPEED OF THE CAR BE DECREASED TO AVOID SKIDDING, IMPLIES THAT IMPENDING SUBBING IS OUTWARD.

(a) 8=10



== EFn = man: RSW(B+ 46) = m = + (EFy = 0: RCOS(0+45)-W=0

(1)

(2)

DR RCOS(8+45) = mg FORMING (1) RSIN(B+4s) mx RCOS(0+95) (CONTINUES)

12.52 continued

OR 12 - 9 (9 + Φs) = (9.81 = 140 m) TAN (10+34.992) 12.53

OR 15 = 19.8063 = 71.302 bm/h

THEN... Δ15 = 15 - 15 = (95 - 71.302) bm/h

OR Δ15 = 25.7 = (0.3 5.2) bm/h

OR Δ15 = 25.7 = (0.3 5.2) bm/h

(P) 0:-2,

2 -2 Fn = man: RSIN(Φ-181)=m - (3) +1 EFn = 0: Rcos(Φ-181)-W=O OR Rcos(Φ-181)=mq (4)

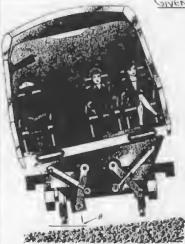
FORMING $\frac{(3)}{(4)} = \frac{R \sin(\phi_2 - 1\theta_1)}{R \cos(\phi_2 - 1\theta_1)} = \frac{m \frac{s_2}{r_1}}{mq}$

DR 5=9 TAN (45-181)= (9.81 =) (40 m) TAN (34.992-5")

DR 5=15,0492 = 54.177 12m/h

THEN - AN = N3-N = (95-54.177) PM/h
OR AN = 408 FM

12.53 and 12.54



SIVEN: B-6; X= 60 TH FOR A CURVE OF RADIUS P; N=100 TH

 $F_{s} = M \left[\begin{array}{c} \frac{1}{2} \sum_{i=1}^{N} \sum_{j=1}^{N} \sum_{j=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_{j=1}^{N} \sum_{i=1}^{N} \sum_{j=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_{i=1}^{N} \sum_$

12.53 and 12.54 continued

12.53 GIVEN: A PASSENGER OF WEIGHT W FIND: (a) Fo WHEN \$00 (b) \$\Phi\$ FOR For \$0

(a) SUBSTITUTING THE KNOWN VALUES INTO EQ. (1).. $F_S = W \left[\frac{(100 \text{ m/h})^2}{(60 \text{ m/h})^2} \text{ TANLG COSLG} - SINLG \right]$ $= W \left(\frac{2}{5} - 1 \right) \text{ SINLG}$

(b) SETTING $F_5 = 0$ IN EQ. (1). $0 = W \left[\frac{(100^{mi}/h)^2}{(60^{mi}/h)^2} \text{ TRNG} \cos(6^4\phi) - \sin(6^4\phi) \right]$ OR $\text{TAN}(6^4\phi) = \frac{25}{9} \text{ TANG}$ OR $6^4\phi = 16.28^{\circ}$

oz \$=10.28° ◀

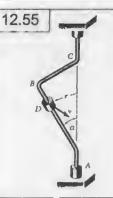
12.54 GIVEN: F = 0.1 W

SUBSTITUTING THE KNOWN VALUES INTO EQ. (1)... $O.1 \, W = W \left[\frac{(100^{mi} | h)^2}{(60^{mi} | h)^2} \, \text{TANLO} \, \cos(6+\phi) - \sin(6+\phi) \right]$

OR $[0.1+\sin((6+\phi))^2 = [\frac{25}{47}\tan(6^2\cos(6+\phi))^2]$ OR $0.01+0.2\sin((6+\phi)+\sin^2(6+\phi))$ $=0.085238[1-\sin^2(6+\phi)]$ OR $1.085238\sin^2(6+\phi)+0.2\sin(6+\phi)-0.075238=0$

SOLVING FOR THE ROSTIVE ROST... $SIN(6^+4) = 0.186816$ OR $6^0+9 = 10.77^0$

oR. 6 - A.TI ◀



GIVEN: Mo = 0.3 kg; a = 40 PARC = 5 TO (CONSTANT) FIND: T IF T = CONSTANT



FIRST NOTE .. No TO ARE

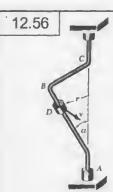
OR N= mg OR N= mg

 $CR = man: N \cos 40 = m \frac{N}{R}$ $CR = \frac{mq}{\sin 40} \cos 40 = m \frac{(R \dot{\theta}_{ABC})^2}{R}$

OR (= 62 TAN400 = 9.81 m/s2 1 (5 RAD), 2 TAN40

=0.468 m

DR 1 = 468 mm



GIVEN: Mo: 0.2 kg; 0:30, 1:0.6 m,

1, : 0.30; Basc: CONSTANT

FIND: RANGE OF VALUES OF U

SO THAT COLLAR D DOES

NOT SUDE ON THE ROD

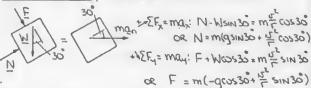
CASE 1: 15 = 5 MIN, IMPENDING MOTION DOWNWARD

 $\frac{301}{N} = \frac{301}{100} + \frac{$

Now .. $F = \mu_s N$ THEN.. $W(g\cos 30^\circ - \frac{\mu^2}{r} \sin 30^\circ) = \mu_s = m(g\sin 30^\circ + \frac{\mu^2}{r} \cos 30^\circ)$ OR $u^2 = gr \frac{1 - \mu_s \tan 30^\circ}{\mu_s + \tan 30^\circ}$ $= (9.81 \frac{m}{s^2})(0.6 m) \frac{1 - 0.3 \tan 30^\circ}{0.3 + \tan 30^\circ}$

OR JMIN = 2.36 5

CASE 2: U= None, IMPENDING MOTION UPWARD



Now.. F= 1/2 N
THEN m(-qcos30+12 sin30) = 11= m(qsin30+12 cos30)

· OR 15 = 90 1+ 11= TAN30 - (9.81 \frac{m}{5}\)(0.6m) 1+ 0.3 TAN30 - (9.81 \frac{m}{5}\)(0.6m) TAN30 - 0.3

FOR THE COLLAR NOT TO SUDE ..

2.36 \$\frac{10}{20} \in 5 \frac{10}{20} \

12.57



GIVEN: Wb=0.61b; T=8 IN.,

B=10 RMS (CONSTANT);

COLLAR D DOES NOT

SLIDE ON THE ROS

FIND: (a) (µs)min WHEN

a=15°

(b) (µs)min WHEN

a=45°

(CONTINUED)

12.57 continued

FIRST NOTE THAT N=(BASC = (BEFE)(10 PAD), 30 BE AND THAT REQUIRING MS=(MS)MIN IMPLIES THAT SUBING OF COLLAR D IS IMPENDING. ALSO, MS = TAN &

NOW CONSIDER THE TWO ROSSIBLE CASES OF IMPENDING MOTION.

CASE 1: IMPENDING MOTION DOWNWARD

Now. F = $\mu_s N$ THEN. $W(\cos x - \frac{y^2}{9r} \sin x) = \mu_s * W(\sin x + \frac{y^2}{9r} \cos x)$ OR $\frac{y^2}{9r} = \frac{1 - \mu_s \tan x}{\tan x + \mu_s} = \frac{1 - \tan \phi_s \tan x}{\tan x + \tan \phi_s}$ $= \frac{1}{\tan(x + \phi_s)}$

CASE 2: IMPENDING MOTION UPWARD

 $\frac{1}{N} = \frac{1}{N^2} \sum_{k=1}^{N} \sum_{k=1}^$

NOW .. F= M=N

THEN .. W(-cosx + gF SINK) = M= W(SINA + gF COSA)

OR of = 1+ Ms TANA = 1+ TANK TANGS
TANA-TANG

Now - 32 (35 the) (8 tf) = 0.483

THEN $\tan(\alpha \pm \phi_s) = 0.483$ OR $\alpha \pm \phi_s = 25.781^\circ$, $\phi_s \ge 0$ AND WHERE THE "+ CORRESPONDS TO IMPENDING MOTION BOWNWARD AND THE "- TO IMPENDING MOTION UPWARD.

(a) α=15°: HAVE 15°± \$=25.781°

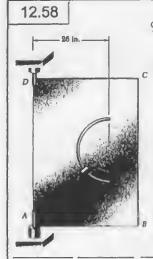
4,20 =>°+° 50 THAT \$=10.781°

THEN (US)MIN = TAN 10.781°

OR (MS) MIN . 0.1904, MOTION IMPENDING DOWNWARDS (b) x=45°: HAVE 45°± \$25.781°

A=45: HAVE 45 ± \$= 25.781" \$=200" = 50 THAT \$=19.219" THEN (Ms)min = TAN 19.219"

OR JUSTIMU & 2349, MOTION IMPENDING UPWARD



GIVEN: T=10 IN, \$ABCD=14 = ; WE=0.816; 45=0.35; ME=0.25

C FIND: (a) F AND IF THE BLOCK

SLIDES IN THE SLOT AT

t=0 WHEN B= BD

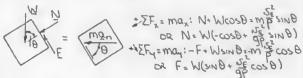
(b) F AND IF THE BLOCK

SLIDES IN THE SLOT

AT t=0 WHEN B=40

FIRST MOTE.. P = 1/2 (26-10 SINB) St THEN an = P= P+ABCD = [1/2 (26-10 SINB) St](14 =)2 = P= P+ABCD = [1/2 (26-10 SINB) St](14 =)2

ASSUME THAT THE BLOCK IS AT REST WITH RESPECT TO THE PLATE.



(a) Have 0.80°... THEN

N = (0.8 lb) [-00080 + 32.24], 2 98 (13-5 51080)] - 51080]

F. (0.8 lb) [51080 + 32.24], 2 98 (13-5 51080)] - 51080]

1.92601 lb

NOW -- FMAX = MSN = 0.35(6.315916) = 2.2106 16

.. THE BLOCK BOES NOT SLIDE IN THE SLOT

(b) HAVE 8=40 .. THEN
N=(0.8 16)[- COSAO + 32.2445 = 3 (13-5 SIN40) \$ = 5.4924 16

F = (0.816)[SIN40 + 32.2 7/32 = 3(13-551N40)] + x (0540)]

NOW - FMAX = USN FROM WHICH IT FOLLOWS
THAT F > FMAX

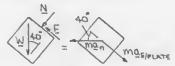
.. BLOCK E WILL SLIDE IN THE SLOT AND QE : Qn + QEIPLATE

= Qn + (QE/PLATE) + (QE/PLATE)

AT t=0, THE BLOCK IS AT REST RELATIVE TO THE PLATE, THUS, (GEIDLATE), =0 AT t=0, SO THAT DEPLATE MUST BE DIRECTED TANGENTIALLY TO THE SLOT.

(CONTINUED)

12.58 continued



* EF = ma : N + Woos 40 = m p sin40

OR N = W (-cos 40 + up sin40) (AS ABOVE)

= 4.4924 16

SLIDING: F= M&N = 0.25 (4.4924 16) = 1.123 16

NOTING THAT F AND QEPRANE MUST BE SIRECTED AS SHOWN (IF THEIR DIRECTIONS ARE REVERVED, THEN ΣF_{\star} IS WHILE MQ, 12 1), HAVE THE BLOCK SUDES DOWNWARD IN THE SLOT AND F_{\star} I. 123 Ib Λ 40°

ALTERNATIVE SOLUTIONS

(Q) ASSUME THAT THE BLOCK IS AT REST WITH RESPECT TO THE PLATE.

THEN.. TAN (\$-10) = man W It = 3 P(\$\frac{1}{2} P(\$\frac{1} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\frac{1}{2} P(\$\fr

OR \$-10 = 6.9588°

Now.. TAN \$50 Ms Ms= 0.35 SO THAT \$5=19.29

. O c \$ c \$ => BLOCK DOES NOT SUBE AND R IS DIRECTED AS SHOWN.

NOW- F = R SIND AND R = W SIN(4-10)

THEN.. F = (0.81b) SIN 16,9588° = 1,926 1b

"THE BLOCK DOES NOT SLIDE IN THE SLOT AND
F = 1,926 1b _ 1 80°

(b) Assume that the block is at rest with respect to the plate.

EF= ma: W.R= man FROM PART a (ABOVE), IT THEN FOLLOWS THAT

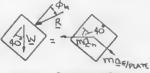
TAN (4-50) = 9 (13-5 SIN 40) 145

OR \$-50 = 5.752°
AND \$=55.752°

NOW $\phi_s = 19.29^\circ$ SO THAT $\phi > \phi_s$." THE BLOCK WILL SUDE IN THE SLOT
AND THEN $\phi = \phi_s$ WHERE TANDE IN $\mu_t = 0.25$ OR $\phi_t = 14.0312^\circ$

12.58 continued

TO DEFERMINE IN WHICH DIRECTION THE BLOCK WILL SLIBE, CONSIDER THE FREE BODY DIAGRAMS FOR THE TWO POSSIBLE CASES.



MARIPLATE

DOWNWARD

NOW - EF= ma: W+R = man + maripute FROM THE DIAGRAMS IT CAN BE CONCLUDED THAT THIS EDUCATION CAN BE SATISFIED ONLY IF THE BLOCK IS SLIDING DOWNWARD. THEN .. * [= max: Wcos 40 + R cosque = m = sin40 Now .. F = R sing THEN .. VICOSTO + THING & W SE SHITO

OR F = M& W (- cos 40 + WE SIN 40) = 1.123 Ib (SEE THE FIRST SOLUTION)

". THE BLOCK SLIDES DOWNWARD IN THE SLOT AND F=1.123 1b 1 40°

12.59

GIVEN: d=0.225 m; 5=0, Q=4 32; m=1.6=0 kg FIND: (a) N AT 1:35 (b) FUET AT 1:35

(a) a = constant => Uniformly Acceleration motion THEN .. IS = 55 + Out

AT 1:35: S= (4 m/5)(35)

OR 15=12 5

(6)

EF = may: Ft = may OR F = (1.6 x10 kg)(4 5) = 6.4 x10" N

Fn = (1.6 x 10 kg) (12 m/s) 2 m/s

(0, 225 m) = 2.048 x10 N

FINALLY - FTUFT = VF2+F2 = {(6.4×10 N)2, (2.048×10 N)2 OR FTUET : 2.05410 N 12.60



GIVEN: 5=0, (00) = 0.24 % OF CHIZZE & HICLBY SUDE AT 1=105 FIND: US

FIRST NOTE THAT (aB) = CONSTANT IMPUES UNIFORMLY ACCELERATED MOTION.

: N= NS+(00) t AT t=103: No = (0.24 52)(105) = 2.4 5

F TOP VIEW m(0,8)2

IN THE PLANE OF THE TURNTABLE ...

EF=maa: F=malash THEN .. F= m ((a)) + (a)?



= m ((a)2+(52)2 +125,=0: N-W=0

DR N=mag AT t= 10 s: F= M. N= M. mag

THEN .. May = ma ((as/2+(15)2 DR MS= 9.81 m/s (0.24 m) 1. (2.4 m/s) 2 7 1/2

OR M3 = 0236

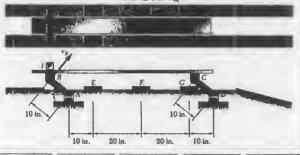
12.61

GIVEN: PARALLEL-LINK MECHANISM ABCD; JB = 2.2 14's

FIND: (a) (45) MIN IF COMPONENTS ARE NOT TO SLINE

(b) B FOR WHICH SLIDING IS

DNIBUNG



== EFx = max: F = & p cos0 + / [Fy = may: N-W=- W 150 51ND OR N= W (1- 48 514 B)

NOW .. FMAX = ILSN = ILSN (1- gp SNB) AND FOR THE COMPONENT NOT TO SLIDE F & FMAX

A 28 cos 8 = M2M (1 - 26 2108)

12.61 continued

.. MUST DETERMINE THE VALUES OF B WHICH MAXIMIZE THE ABOVE EXPRESSION, THUS ..

$$\frac{d}{d\theta}\left(\frac{d\theta}{d\theta} - \sin\theta\right) = \frac{(\frac{d\theta}{d\theta} - \sin\theta)(-\cos\theta)(-\cos\theta)}{(\frac{d\theta}{d\theta} - \sin\theta)^2} = 0$$

winder() = ety SC3 = 9 = 0 MIS SO NOW. SINB = (32.2 tt)(10 tt) = 0.180373

OR 0 = 10.3915 AND 0 = 169.609°

(a) From ABOVE,
(b) MIN =
$$\frac{39}{39} - 3100$$
 WHERE $5100 = \frac{53}{39}$
(LLS) MIN = $\frac{39}{39} - 3100$ WHERE $5100 = \frac{53}{39}$
(LLS) MIN = $\frac{53}{39} - \frac{5}{3} = \frac{5}{1 - 5} = \frac{5}{$

OR (Ms)my = 0.1834

(b) HAVE IMPENDING MIDTION TO THE LEFT FOR 0 = 10.39

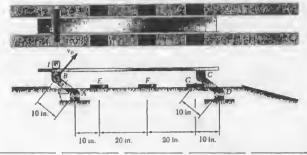
= TAN 10.3915°

TO THE RIGHT FOR 8=169.6

12.62

GIVEN: PARALLEL-LINK MECHANISM ABCD: Ms = 0.35, Mx = 0.25 FIND: (a) (US) MAX IF COMPONENT I IS NOT

TO SUDE ON MEMBER BC (b) B FOR WHICH SUDING 15 IMPENDING



== \frac{1}{2} \fr man . IEFy = may: N. W = - & P SINB OR N=W(1- 40 SMB) NB Now .. Form = MSN = MSW(1- 37 SIND) AND FOR THE COMPONENT NOT TO SLIDE .. F & FMAX OR No COSB = NOW (1 - 43 SINB)

TO ENSURE THAT THIS INEQUALITY IS SATISFIED, US I'M MUST BE LESS THAN OR EQUAL TO THE MINIMUM VALUE OF ME 90/10050+MESIND), WHICH OCCURS WHEN (COSO+ 12, SIND) IS MAXIMUM. THUS .. (CONTINUED)

12.62 continued

6 = 8 = 0 = = (BNIC = 4 + BEW) 38 OR TANB = MS 145=0.35 02 B = 19,2900°

(a) THE MAXIMUM ALLOWED VALUE OF US IS THEN .. Mens = Ms cos Bons sind WHERE TAN BONS Busyle = Bux(Burn) + Beco = (322 \$)(12 4) SIN 19.2900° OR (NE) MAX = 2.98 11

(b) FIRST NOTE THAT FOR 90° < 8 = 180°, EQ. (1) BECOMES No COSA+ MS SING

WHERE A = 180° - 0. IT THEN FOLLOWS THAT THE SECOND VALUE OF B FOR WHICH MOTION IS IMPENDING IS ..

B=180-19.2900 = 1607100

". HAVE IMPENDING MOTION TO THE LEFT FOR 8=19.29" TO THE RIGHT FOR BEILDIT

ALTERNATIVE SOLUTION

EF=ma: W+R=man THEN

FOR IMPENDING MOTION, \$= \$. ALSO, AS SHOWN ABOVE, THE VALUES OF B FOR WHICH MOTION IS IMPENDING MINIMIZE THE VALUE OF S, MID THUS THE VALUE OF an (an = No). FROM THE ABOVE BIAGRAMI IT CAN BE CONCLUDED THAT an is minimum when man and R ARE. PERPENDICULAR. THEREFORE ...

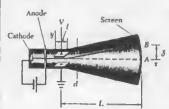
FROM THE DIAGRAM... PEONE) AND Man = WSINDS OR JB = 9P SINB (AS ABOVE)

FOR 90 4 0 4 180 HAVE ..

(man)_{MIN}

FROM THE BIAGRAM .. $\alpha = 180^{\circ} - \theta$ (as above) $\alpha = \phi_s$ AND man . WSINGS OR JB = 9P SIND (AS ABONE)

12.63 and 12.64



GIVEN: Jr . J. (= CONSTANT)

FIRST NOTE THAT THE HURIZONTAL COMPONENT OF THE VELICITY OF AN ELECTRON IS A CONSTANT (NJ) RECARDLESS OF THE VALUE OF THE POTENTIAL V. THEN..

1= X20+50t

THE TIME trate FOR AN ELECTRON TO TRAVEL BETWEEN THE PLATES IS THEN.

OR tours = 15

AND THE TIME TEXTERN TO TRAVEL FROM THE END OF THE PLATES TO THE SCREEN IS.. $(L-\frac{1}{2}\ell)$ * No $(t_{\rm SCREEN})$

OR theren = L- 1

NEXT CONSIDER THE VERTICAL MOTION OF AN ELECTRON AS IT MOVES BETWEEN THE PLATES.

THEN, FOR THE UNIFORMLY ACCELERATED MOTION

IN THE Y DIRECTION HAVE $N_{\gamma} = (N_{\gamma}^{2})^{2} + \Omega_{\gamma}^{2}$ $N_{\gamma} = (N_{\gamma}^{2})^{2} + \Omega_{\gamma}^{2}$

AT THE END OF THE PLATES...

(LT) = (ND) (LD)

= QVI

= QVI

TOTAL STATES...

Z md LD ...

Z md LD ...

12.63 FIND: & IN TERMS OF V, 150, R, M, d. P. L

FIRST NOTE THAT THE VELOCITY OF AN ELECTRON IS CONSTANT AFTER IT LEAVES THE PLATES.



THEN, FROM THE END OF THE PLATES TO THE SCREEN ..

 $A = A^{\delta} + (\mathcal{Q}^{\delta})^{\delta} f = (\frac{\delta \Lambda \delta_{\sigma}}{S M q \Omega_{\sigma}}) + (\frac{\delta \Lambda \delta}{M q \Omega_{\sigma}}) f$

AT THE SCREEN: $\delta = \frac{\varrho V \ell^2}{2md J_0^2} + \left(\frac{\varrho V \ell}{md J_0}\right) \left(\frac{L - \frac{1}{2}\ell}{J_0}\right)$

OR S= RYIL (CONTINUED)

12.64 continued

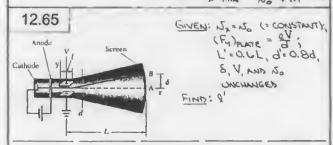
12.64 GIVEN: AT X= (= -4) = 0.05d FIND: (| MIN IN TERMS OF C, M, Jo, V

AT X= 1, HAVE

Z

T= Y1 = 2V02

Z md vot



From the solution to Problem 12.63 HAVE $\delta = \frac{\text{QVIL}}{\text{mdJ}_{c}^{2}}$

THEN, SINCE & IS UNCHANGED, HAVE

(6) MODIFIED = (8) DRIGINAL:

MILL

M

OR 110.66) 11

OR 1'-1.3331

12.66 and 12.67



GIVEN: Mg = 0.2 kg; r = 250+150 sinnt, 8 = T(4t2-8t) r~mm, t~s, 8~RAB

HAVE. $\Gamma = (0.25 + 0.15 \sin \pi t) m$ $\theta = \pi (4t^2 - 8t) an$ THEN $\dot{\Gamma} = (0.15 \pi \cos \pi t) \frac{\pi}{2}$ $\dot{\theta} = \pi (8t - 8) \frac{\alpha n}{2}$ AND $\ddot{\Gamma} = -(0.15 \pi^2 \sin \pi t) \frac{\pi}{2}$ $\dot{\theta} = 8\pi \frac{\alpha n}{2}$

12.66 FIND: (a) Fr AND FO AT t=0
(b) Fr AND FO AT t=0.55

(a) AT t=0: F=0.25m % i=0.15n %

0 = - 817 EXP

12.66 and 12.67 continued

Now. ar=1-182 = 0-10.25 m)(-BTI RAD)2 =-16TI2 5 AND Q= (0+200 = (0.25 m)(81) (20)+2(0.151) (7)(-87) (20) = T(2-2.4T) =

FINALLY .. Fr = mar = (0.2 kg) (-16 TI = m)

OR F==-31.6 N

B= mag = (0.2 kg)[T1 (2-2.4T1) = 8 OR FB = - 3.48 N

(b) AT t=0.55: F=0.40 m

0 = - 4TI RAD 0=7 " =-0.15TZ 32 9 - 8T 3AD

Now .. ar = 1-192=-(01512 1/2)-(0.40 m)(-471 200)2 = -6,55 m2 me

AND Q = TB+2rB = (0.40m)(87 (32) + 0 = 3.2017 52

FINALLY. Fr = mar = (0.2 kg) (-6.55 TI 52)

OR F==- 12.93 N Fo = mao = (0.2 kg)(3.20 m =)

OR FB = 2.01 N

12.67 FIND: FO AND FO AT toliss

AT t=1.55: r=0.10m B = 4TT RAD L = 0.

" = 0,15 TT 52 B = BTT BAB Now- a= "- roe" (0.15 12 m2) - (0.10 m) (4TE RAD)2

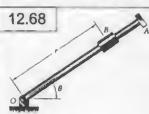
= -1.45 TIZ TZ 0 = 18+2+8 - (0.10 m)(81 PA) +0 = 0811 3

FINALLY .. F= mar = (02 bg)(-1.45 T2 T2)

OR F==- Z.86 N

FB = mab = (0.2 kg)(0.87 52)

FA . U.SO3 N



GIVEN: WB = 516;

C= 10 B= 2 SINTE r- 12 t-s, B-RAD

FIND: (a) Fo AND FO AT t=15 Lb) F AND FO AT 1-65

HAVE. 1 = 10 8 D = (TSINAT) RAD THEN .. 1 = - (+4) = 5 B=(2 cusnt) & B = - (271 SINTE) -= 8

(a) AT t=15: r=2 St r=-0.4 15 8 = -2 RAB ドーロルは O = 0

Now. ar = 1 - 182 = (0.16 5=) - (2 ft) (-2 = 5)2 = -7.84 4/32

an=+2+0=0+2(-0.4 5)(-2 20)=1.6 5

FINALLY .. Fr = MB Or = 52.2 4752 (-7.84 5)

OR F =- 1.217 16 Fo = MB Qo = 32.2 8452 (1.6 452)

CR F8 = 0.248 16 (CONTINUED)

12.68 continued

(b) AT t: 65: F= 1 It

F=- U.1 5 0 = 2 Th " = 0.02 Ex B = 0

Now. ar = "- - + 02 = (0.02 \$ 2) - (1 ft x 2 = 1) 2 = - 3.98 52

00=10+210=0+2(-0.1 =)(2 RAS) 0-0.4 St

FINALLY .. Fr = MBQr = 316 (-3.98 52)

Forma a = 32.2 Mys (-04 1/5 2)

OR 5 = - 0.0621 16

12.69



GNEN: B = Ct C = CONSTANT, r -- k : AT t=0, F= 5

FIND: (a) T IN TERMS OF

m, c, k, 5, t

(b) Q, FORCE

EXERTED ON B BY ARM AA'

KINEMATICS

HAVE. OF = = - k AT t=0, r. T: 15dr= 13-kdt

OR F= Fo- kt ALSO .. " = 0

Now .. ar = r-rb2 = 0- (ro-kt)(ct)2=-c2(ro-kt)t2

ap=rB+2+0=(ro-kt)(c)+2(-kxct) * c(5-3kt)

KINETICS



(a) += EF= mar: -T=m[-c/r-kt){) OR T=mc2(5-kt)t2

(b) \$\(\max_{\begin{cases}
\text{E}_{\begin{cases}
\text{F}_{\begin{cases}
\text{C} & \text{C}_{\begin{cases}
\text{C} & \text{C}_{\begin{cases}
\text{C} & \text{C}_{\begin{cases}
\text{C} & \text{C}_{\begin{cases}
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\text{C} & \text{C}_{\begin{cases}
\te

OR Game(Forshit)

12.70



CIVEN: MB: 3kg; 0=0.75t

1. 0.5 % AT too

FIND: L WHEN TOW. WHERE Q IS THE

FORCE ON B FROM AN

KINEMATICS

F. F. O.S. HAVE

AT t=0, 1=0: \[dr = \] 0.5dt OR 12(0,5t) m

ALSO .. " = 0

0 = (0.75L) RAD 8 : 0.75 RAD

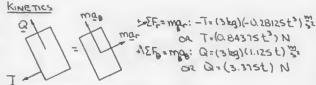
Now .. ar= "- rb" = 0- (10.5t) m) (10.75t) = 02

= - (0.28125 t3) 52

Q= -10+2+0 = [(0.52)m](075 30) AND +2(0.5 % (10.75t) 50)

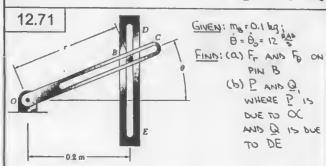
= (1.125 t) Th





NOW REQUIRE THAT. T= Q OR (0.84315t) N: (3.315t) N DR t2 4.000

OR t= 2.00 s



KINEMATICS

FROM THE DRAWING OF THE SYSTEM HAVE.. $C = \frac{0.2}{\cos \theta} \text{ m}$

THEN + = (0.2 SIND 6) M 0 = 12 RAB

"= 0.2 cos0(cos20)-SINO(-2cos0 SINO) 62 = (0.2 (0.5) (0.5) m

SUBSTITUTING FOR B .. F = 0.2 - SINB (12) = (2.4 - SINB) & " = 0.2 (1451N2) (12)2 = (28.8 (1+51N2)) m

Now. $\alpha_r = r - r\dot{\theta}^2 = (28.8 \frac{1 + 51N^2\theta}{\cos^2\theta}) - (\frac{0.2}{\cos^2\theta})(12)^2$

 $Q_{\theta} = r\ddot{\theta} + 2\dot{r}\dot{\theta} = 0 + 2(2.4 \frac{\sin \theta}{\cos^2 \theta})(12)$ $= (57.6 \frac{\sin \theta}{\cos^2 \theta}) \frac{m}{e^2}$

(a) HAVE.. F = maar = (0.1 kg)[(57.6 \frac{\sin^2\theta}{\cos^3\theta})\frac{\sin}{\si}] For mode = (0.1 kg) ((57.6 25/6)) (5) OR FO= (S.76 N) THUB SECO.

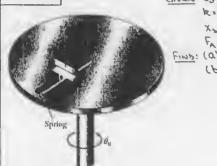
(b) = 18

NOW .. + 1 EFy: Fg cos 8 + Fr sin 8 = P cos 8 OR P = 5.76 TAND SEC 8+(5.76 TAN & SEC 8) TAN & OR P=(5.76N)TMBSECB YB (CONTINUED)

12.71 continued

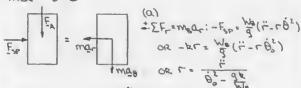
== Fr = Q cos B OR Q = (S.76 TANED SECD) COSB OR Q . (S.76N)TAN'B SEC'B -

12.72



GIVEN: 03 = 15 840; WB = 0.5 16 R = 4 MYCE; WHEN COO, X5000; == 40 84/5 FA = 216 FIND: (a) F (6) 2-

FIRST NOTE ... WHEN TOO KODED TO FORE KT AND 8.8 15 5 THEN BOD



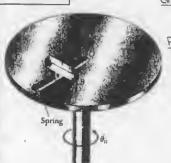
-40 845 THEN .. F = -(15 800)2 - (32.2 4452)(4 (44)

(b) -12F6 = map: Fx = Wa (-8-2-0) Now .. Not = F 5- 35x = (32.2 52)(21b)
2(0.51b)(15 825)

OR N= 4.29 5

C= 1.227 St

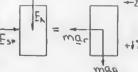
12.73



GIVEN: 8 = 12 5 : W= 8.05 02; WHEN TED, Xxx :O: AT 1.0 + =0, T. IS IN. FIND: (a) T AND FA AT 2=0.15 R = 2.25 1994

(b) T AND FA AT 2-0.15, R=3.25 1488 WHERE FA IS THE HORIZONTAL FORCE EXERTED ON THE SLIDER BY THE DISK

FIRST NOTE ... WHEN (=0, X sp =0 => Fsp = kr 8 . 8 = 12 RAM 5 = 15 IN. = 1.25 f2 aus 0=0 THEN



 $\frac{1}{2}\sum_{r=1}^{n}m_{\theta}\alpha_{r}:-F_{sp}:\frac{W_{\theta}}{9}(\ddot{r}-r\dot{\theta}_{s}^{2})$ or $\ddot{r}+(\frac{k_{\theta}}{W_{\theta}}-\theta_{s}^{2})r=0$ (1) +1 EF = maas: F = q (10 + 2+ B) (2)

12.73 continued

(a) k = 2.25 16/st

SUBSTITUTING THE GIVEN VALUES INTO EN (1)..

THEN .. If = 100 AUD AT t=0, 100:

NOTE: 1-0 IMPLIES THAT THE SLIBER REMANS AT ITS INITIAL RADIAL POSITION.

WITH +=0, EQ. (2) IMPLIES

FH = 0

(b) k= 3.25 16/12

SUBSTITUTING THE GIVEN VALUES INTO EQ (1)...

" + [3.25 15/4.32.24452 - (12 20)2] T = 0

Now .. " = \$\frac{1}{4}(") " = 5\frac{1}{4} = 5\fra

SO THAT No dis +645 = 0

AT t=0, No=0, T=5: 10 No dis = -64/6 rdr

OR No --64(-- -62)

Now .. No = Of = B 1 1 2 - 12 = 1 ddt - 81

THEN | SIN'(E) | COSSAGE = Bt

OR | SIN'(E) - P = 82

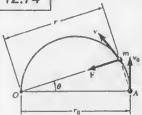
OR T = To SIN (8t+ =)= To cos8t = (1.25 ft) cos8t

FINALLY.. AT t= 0.15: T= (1.25 ft) (05 (8 > 0.1)

EQ(2).. FA = (0.05/16)16 OR F= 0.871 ft 32.2 fys2 2.[-(10/5)51N(8=0.1)](12 35)

OR Fx=-2.69 16

12.74



GIVEN: CENTRAL FORCE F AND
SEMICIRCULAR PATH
SHOWN; AT \$-0,
\$-0, 15-15

SHOW: N= 150020

HAVE - $U = \dot{\Gamma} e_{r} + \dot{\Gamma} \dot{\theta} e_{\theta}$ SO THAT $U' = \dot{\Gamma}^{2} + \dot{\Gamma}^{2} \dot{\theta}^{2}$ (1) FROM THE DIAGRAM... $\Gamma = \Gamma_{0} \cos \theta$ THEN $\dot{\Gamma}^{2} - (\Gamma_{0} \sin \theta)\dot{\theta}$ SUBSTITUTING INTO EQ (1).. $U' = (-\Gamma_{0} \dot{\theta} \sin \theta)^{2} + (\Gamma_{0} \cos \theta)^{2} \dot{\theta}^{2}$ $U' = (-\Gamma_{0} \dot{\theta} \sin \theta)^{2} + (\Gamma_{0} \cos \theta)^{2} \dot{\theta}^{2}$

CR 5=70 B (2)

AT t=0: 50= 70 8 FROM EA. (12.27): 728 - 728

OS 9 = 1023 = (12079) = 1200,0

SUBSTITUTING FOR B IN EQ. (2)...

2 N= COSTO Q.E.D.

12.75

GIVEN: CENTRAL FORCE F AND SEMICIBLULAR PATH
SHOWN; AT \$=0, [-[.

6=0, [-1]
FIND: (a) F WHEN 8=0

(b) F WHEN 8=45°

FROM THE DIAGRAM. F= 6 COS B
THEM F=-(651NB)B
NOW. S= FEF+ FBEB

NOW. \$= re-+rees SO THAT AT \$=0... \$5= 7.00 FROM EQ. (12.27): 720= 60

FROM PROBLEM 12.74: 5= 5050

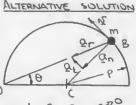
Now. $Q_{\epsilon} = \frac{dE}{dE} = \frac{dE}{dE} \left(\frac{\omega_{\epsilon}}{\omega_{\epsilon}} \right) = v_{\epsilon} \frac{2\cos\theta \sin\theta}{\cos^{2}\theta} \dot{\theta}$ $= 2 v_{\epsilon} \frac{\cos^{2}\theta}{\sin\theta} \cdot \left(\frac{v_{\epsilon}}{v_{\epsilon}} \frac{\cos^{2}\theta}{\cos^{2}\theta} \right)$ $= 2 \frac{v_{\epsilon}}{v_{\epsilon}} \frac{\cos^{2}\theta}{\cos^{2}\theta} \dot{\theta}$

FINALLY.. Ft = mat = 2m = cos = 0

(a) WHEN 8=0 (b) WHEN 8=45°: FE=2M 5 50545 FE=0

OR FE . 8m F

12.75 continued



FIRST NOTE THAT TRIANGLE OBC IS AN ISOSCELES TRIANGLE.

: 4 OBC . B FOR CENTRAL FORCE MOTION ag = O

: Q . Q . + & 0

Now - a = at + an ox at + an = ar FROM THE ABOVE DIAGRAMI ..

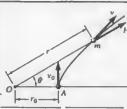
at = an TAND P= 7 WHERE an= y

AND FROM PROBLEM 12.74

2. Cos 28 Q = (10/00/2) = SINB = SINB = SINB

FINALLY - FE = May = 2m To COSED (AS ABOVE)

12.76 and 12.77



GIVEN: CENTRAL FORCE F AND PATH SHOWN; C= 5 (65 20) \ = 7 8=0, N. N. T. T.

(= 10058 = (100,50); HAVE + = 1 [- = (cosze) =] (-251m28) = 10 SIN 20 0

Now .. 5 = 18 + 18 80 SO THAT AT t=0. 5=60.
FROM ED. (12.27): r20 = 520. OR 9: 1016 = (1016) 2 : 50 COS 28

12.76 FIND: No AND NO AS FUNCTIONS OF B

HAVE. Note to To conses " To cos 28 OR 15= No Tros 20 AND 50=10= 100120 = 500120

OR 15 = 5, (cos 20

SHOW: (a) SAF, AND FAF

(a) FROM THE SOLUTION TO PROBLEM 12.76 HAVE NT = No Trossa No = No VCO22B (CONTINUED) 12.77 continued

Now .. Nr = Nr + Nr = Nr 20) + (No 100 50) = Nr 50 = (No (00 500)) + (No 100 50) = Nr 50 Now .. Nr = Nr + Nr 60) + (No 100 50) = Nr 50 Now .. Nr = Nr + Nr 60) + (No 100 50) = Nr 50 Now .. Nr = Nr + Nr 60 Now .. Nr = Nr 60 Now .. Nr = Nr 60 Now .. Nr = Nr 60 Nr OR N = 150520

RECALLING THAT 12 TOURS

IT FOLLOWS THAT IS - 25 DR NORT Q.E.D.

NOW- FENTENS SIN 20 NUS FE TOUS 28

COMBINING .. + = 5 F SIN 28

NOTING THAT THE SIN 28 HAVE.. = 150 1+ cos 20 = 20 (1+ cos 20) c

Now .. ar = 1-18 9 = 5 cos28 (FROM ABOVE) = 5 (1+00,28)1-1(\$ 00,28)2

FINALLY .. F = Fr + FB AND FOR CENTRAL FORCE MOTION, FO = O. THEN.

F=Fr= mar = m 200 r

OR FAT Q.E.b. (b) FIRST NOTE .. N= 50 (PART a)

AND QB = O (CENTRAL FORCE MOTION)

NOM .. of - gr. of (201) = === = 15 (5 C SIN 28) (FROM PART a)

= MEL SINSA HAVE - a = a2 + an = a2 + a0 ar = 52 (PARTA)

HAVE ... $\Omega_{n}^{2} = (\frac{N_{0}^{2}}{\Gamma_{0}^{2}}\Gamma)^{2} - (\frac{N_{0}^{2}}{\Gamma_{0}^{2}}\Gamma \sin 2\theta)$ $= \frac{N_{0}^{2}}{\Gamma_{0}^{2}}\Gamma^{2}\cos^{2}2\theta \qquad \Gamma^{2} = \frac{N_{0}^{2}}{1\cos^{2}2\theta}$

OR an = 50

OR $\frac{\sqrt{2}}{\Gamma} = \frac{\rho}{\rho} \sqrt{2\Gamma} \left(\frac{1}{\rho} \right)^2$ OR FINALLY. an = P.

OR P= 12 -3

OR PARS Q.E.D.

12.78 GIVEN: A PLANET OF RADIUS R AND OF BENSOTY P; MOON HAVING CRBITAL RABIUS C= 2R 5+0M: I= (24716P)12 HAVE. F=G Mm [Ea. (12.28)] AND F=Fn=man=m = CHM = ME OR N2 EM FOR THE PLANET .. M=PV (17 R3) できたりまからりまからだ THE TIME I FOR THE MOON TO COMPLETE ONE FULL REVOLUTION IS I = STIC = SUC (376 P 2) - 5 FOR 1= 2R .. I = 150 (2R)/2 OR 2 = (241) Q.E.D. 12.79 GIVEN: A PLANET OF RADIUS R HAVING AN ACCELERATION OF GRAVITY Q AT ITS MARFACE; T, THE ORBITAL PERIOS OF A MOON SHOW: r=f(R, q, E), WHERE r IS THE ORBITAL RABIUS OF THE MOON FIND: 9 FOR JUPITER; R=71492 km. TEUROPA = 3.551 DAYS, EUROPA = 670.9410 KM HAVE .. F = G MM [Ea. (12.281] MO F= Fn = Man = M F THEN GMM: ME OR 52 GM SO THAT 52 2 822 EQ. (12.30) Now GM = gR2 OR 5 = R (9 FOR ONE ORBIT. $T = \frac{2717}{N} = \frac{2717}{R[\frac{9}{2}]}$ OR T= (922 R2) 113 SOLVING FOR 9. 9=472 TERE AND NOTING THAT [=3.55] DAYS = 306 BOGS. THEN 9 JURITER = 472 TEUR RJUP

12.80 GIVEN: SATELLITE IN A GEDSTNICHEDWOUS EARTH ORBIT : T = 23.934 h FIND: (a) ALTITUDE IN OF THE WITELLITE (b) VELOCITY IS OF THE SATELLITE FIRST NOTE .. C= 23.934 h = 86. WZ4=10 5 AND REARTH = 3960 mi = 20.9088=10 \$\$ REARTH = 6 37 × 10 M (a) FROM THE SULUTION TO PROBLEM L= (35565),12 Now .. h= r-R THEN .. SI: H=[9.81 50 (86.1624 610 5)2 (6 37 610 m)2]13 - 6.37=10 W = (42.145-6.37) 410 m OR h=35,7700 km C) SOLUTIO: H= (32.2 3 (86.1624.1035) 2 (20,988.186 (1)6) - 20,9088x10" 12 = (138.3343-20.9088) = o (t OR h=22,240 mi (b) HAVE .. 5 = 271-THEN. SI: 5=27 42.145 x10 m 02 7=3010 E US. UNITS: 5= 27 138.3343.10 CE OR 15=10 090 \$\$ 12.81 GIVEN: FMOON = 238, 910 mi, TMOON = 27,32 DAYS FIND: MASS M OF THE EARTH HAVE .. F = 6 TE [EQ. (12.28)] AND F=Fn=man=man= Now .. No ZT 6-2 (T) = 5 (T) = 1 TAHT OC NOTING THAT I = 27.32 DAYS = 2.3604 10 5 T = 238,910 mi = 1.26144 = 109 ft HAVE .. M= 1 (2.3604 = 10 5) (1.26144 = 13 (t) OR M= 413x 1021 16.52

= 4T2 (506 806 5)2(71,492 x 106 m)2

SJUPITER = 2.53 9 EARTH

NOTE:

OR 9 NATER = 24.8 5

12.82 12.84 GIVEN: ALTITUDE h= 380 km or SPACECRAFT GIVEN: FOR THE MODILY JULIET AND TITANIA OF IN ORBIT ABOUT MINRS; PMARS 3.94 Mg/m3 URANUS, TJ = 0.4931 DAYS. RMARS = 3397 km TT = 8.706 DAYS, 5 = 49,000 mi FIND: (a) MASS M OF URANUS FIND: (a) TIME I OF ONE ORBIT (b) VEWLTY IS OF THE SPACECRAFT (P) C HAVE. F. G Mm [ED.(12.28)] (a) FROM THE SOLUTION TO AND F= Fn = man = m & PROBLEM 12.78 HAVE THEN 6 Mm = ME I= (E) 1/2 OR M= ENL WHERE C= R+h = (3397+380)km Now 52 = 3777 km SO THAT M. E(2007) 2 = E(2007) 2-3 THEN. [=[(66.7340"12 mgs)(3.94200 mg)] 1/2 (3771 km) 1/2 Now .. Ts = 0. 4931 DAYS = 42, 604 5 AND 5 = 40,000 mi = 211.2 x10 ft (a) Using Eq. (1). = 7019,55 OR T=1 H ST MIN (B) HAVE . J. 2711 OR M = 5.96 110 11 SU (3) 11 210 m) (b) REWRITING EQ (1) .. MG = T2 AND THEN TT2 T2 OR N=3580 € 12.83 GIVEN: ALTITUDE ha = 3400 km of SATELLITE OR IT = (B. 706 DAYS) = (40,000 mi) IN ORBIT ABOUT SATURN; U" = 24.45 km; OR 5 = 271.2 = 17 SO FOR MOON ATLAS, F= 137.64x10 km, TATLAS = 0.6019 DAYS 12.85 FIND: (a) RADIUS R OF SATURN GIVEN: SPACECRAFT OF WEIGHT W: 1200 16; (b) MASS M OF SATURN he = 2800 mi; Mmour = 0,01230MEATH, RMOON = 1080 mi HAVE . F= G FT [Ea. (12.28)] ATLAS M FIND: (a) GRAVITATIONAL FORCE F ON THE SATELLITE AND F=Fn=man=m= SPACECRAFT, EARTH ORBIT THEN GMM = ME (b) Fm. TE = Im (c) gmoon OR GM . CS2 Ea. (12.29): 9= GM FIRST NOTE THAT RE = 3960 mi THEN TE = RE+hE = (3960+2800) mi AND THEN GRE- CUE im 0050 = (a) HAVE .. F = G Mm EQ.(12.28) AUD GM = gR2 EQ.(12.29) he OR N=R() (1) AND B10=212 = 5 (R+h (2) NOW .. [= 277 = 277 [USING EQ. (1)] THEN .. F = QR2 m = W(R)2 OR R/9 = 27 (3) FOR THE EARTH ORBIT .. F = (1200 16) (3960 ml) 2 OR F = 412 16 (a) Using Eas. (2) AND (3) ... RSATURU (SATURU = No VR+hs = ZTITA 42 (b) FROM THE SOLUTION TO PROBLEM 12.81 HAVE M=早(点), L, OR R = (271 (2)2)2 - hs I = ZTIC 1/2 NOTING THAT TA = 0.6019 DAYS = 52.00 +2:10 5 Now.. TE = Em => 2TICE = 2TICME HAVE. R=[\frac{271 (137.64x10 m)3/2}{(24.45x10 m)(52.0042x10 s)}]^2 - 3400x10 m OR IM = (Mm)12 = (0.01230) (6760 mi) = 60. 273×10 m OR R=60.3x10 km OR 1 = 1560 mi (C) HAVE .. GM= gR2 Ea. (12.29) LA) FROM ABOVE _ GM = rus THEN. M = 50 % & 50 (R+ho)

= (24.45×10 m) (60.273×10 13400×10) m

= (6.73×10 2 m) OR M = 570×10 by SUBSTITUTING INTO EQ. (1) (CONTINUED)

12.85 continued

OR 9m = (Re)2 (FM)3 GE = (RE)2 (MM) GE USING THE RESULTS OF PART (b). THEN... 9m = (3960 mi) (0.01230) (32.2 (\$)

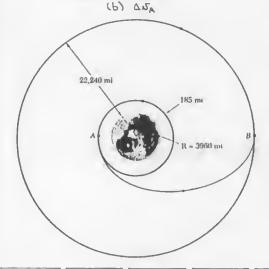
OR 9moon = 5.32 52

NOTE: 9 MOON = 69 EARTH

12.86

GIVEN: CIRCULAR ORBITS AND ELLIPTIC TRANSFER ORBIT AB SHOWN; DUB = 4810 9/s

FIND: (a) (58)TR



FIRST NOTE .. R = 3960 mi = 20.908840 IL T = (3960+185) mi = 4145 m1 = 21.8856 110 ft

rg=(3960+22,240)mi=26,200 mi=138.336=10 ft (b) NOW-. (NE bac= (NE)+A+ ΔVE ULAR ORBIT.. ΣFn=MQn: F=M4Σ OR ONE = (1535.5-1551.3) "S FOR A CIRCULAR ORBIT .. EF = Man: F= m F

EQ. (12.28): F= 6 Mm THEN G FE= M C

OR JE GMI

EQ. (12.29): GM=qR2 SO THAT JE 9R2 FOR A CIRCULAR ORBIT

THEN .. (NA LAC. = 32.2 Ft/s . (20.9088 x10 ft)

OR (JA) & 196 = 25, 362 5

(NB) 2 = 32.2 4/3 x (20,9088,10 ft)2 02 (No) EIRC = 10,088 5

(a) HAVE. (UB LOR = (UB)TR + OUE OR (UB) TR = (10,088-4810) = 5278 5

OR (UB) TR = 5280 5

(b) CONSERVATION OF ANGULAR MOMENTUM REDVIRES CAM(UL)TR = CBM(UB)TR THAT

OR (NA) TR = 26,200 mi x 5278 5 = 33,362 5

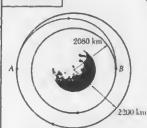
(CONTINUED)

12.86 continued

Now .. (NA)TR = (NA) CIRC + DNA ANT = (33,362-25,362) = CR

OR DUT = 8000 15

12.87



GIVEN: CIRCULAR ORBITS ABOUT THE MOUN AND ELLIPTIC TRANSFER DRBIT AB AS SHOWN: DUS = 26.3 5; mmoon = 73, 49=1021 kg

FIND: (a) (NB)TR

(b) DNB

FOR A CRESCAR ORBIT. EF = Man: F= M TE

THEN. GMm = m & OR NZ GM

THEN .. (NA Like 2 66.73×10 trg.52 × 73.49 ×10 trg

2080 x103 M

OR (NB)circ = 1535,5 5

(a) HAVE -. (JA) TR = (JA) CORC + AUT = (1493.0-26.3) = = 1466.7

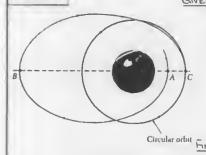
CONSERVATION OF ANGULAR MOMENTUM REGUIRES THAT TIMOSTR - TOM (NE)TR

DR (LE) m = 2200 km = 1466.7 5 = 1551.35

OR (5)/10 = 1551 5

OR AUB:-15.85

12.88



GIVEN: CIRCULAR ORBIT ABOUT VENUS AND ELLIPTIC TRANSFER ORBITS AB AWA BC;

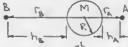
E = 6420 km; 5 = 7420 3 h = 288 bar

DNE = - 264 %; MYENUS = 4.869=10 ha

Ryenus: 6052 km Circular orbit FND: (0) (58) TRAB (b) hB

FIRST NOTE .. TA = R+ ha

= (6052 + 288) km = 6340 km



FOR A CIRCULAR ORBIT .. EF . Man: F= m F

THEN (AE) 200 - 66.73 × 10 10 5 M (C)

12.88 continued

(No) CIRC = 7114.0 5

(No) use = (No) TREE + DNG Now ..

(LE) TR = [7114.0 - (264)] = 7378.0 5

(a) CONSERVATION OF ANGULAR MOMENTUM REDIRES THAT .. AB: FAM(UA) . FBM(US)TRAS

BC: 12 m (NE)tresc = 12 m (NE)tresc (5)

THEN (1) => TB (NB)TRBC TC (NE)TRBC (5) TRAB (NA)

Now .. (NB) TREE = (NB) TRAF + ONB

(NB)TRAS + ONE CONTRAC THEN ... (NE)TRAS

OR (NB)TRAB = TE (NETTRAB-1 24.5 3 6420 km , 73780 7 -1 6340 km 7420 mg (NA) =3557.7 2

OR (NB) TRAB = 3560 5 (b) FROM EQ. (1) ...

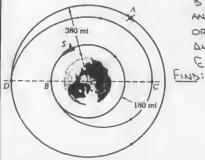
TB = NSA TA = 7420 M 3557.7 % 6340 km = 13 223 km (NE)TRAB

TB=R+ha Now .. OR hg = (13223-6052) km

02 h8=7170 km

12.89

GIVEN: CIRCULAR ORBITS A AND FIRST NOTE. FOR R (T-TA) B ABOUT THE EARTH AND ELLIPTIC TRANSFER ORBITS BC AND CD; DUG = 280 \$ AUE=260 E = 4289 mi FIND: AND



FIRST NOTE .. R= 3960 mi = 20.9088110 ft TA = (3960+380) mi = 4340 mi = 22.9152x10 St 13 (3760+180) mi = 4140 mi = 21.8592=10 ft FOR A CIRCULAR ORBIT .. EF = ma: F=m = Ea. (12.28): F= G MW G TE = M T THEN .. 52 - GM - gR2 USING EQ. (12.29) THEN .. (NA) = 32.2 4/5 4 (20.908110 11)2 22.9152 210 51 OR (UA) CIRC = 24,785 445 (UB) CIRC = 32.2 4452 x (20.9088 x106 ft)2 21.8592 x106 ft AND

OR (NE) CIRC = 25,377 14's HAVE .. (NB) TRAC = (NB) CIRC + ONB = (25,377+280) \$
= 25,657 1/5

(CONTINUED)

12.89 continued

CONSERVATION OF ANGULAR MOMENTUM REWIRES THAT ...

BC: 6 m (Notraco = 6 m (Notraco

FROM EQ (1). (NE) TROE = (15) TROE = 4140 mi 25,657 5

Now. (NE) + a = (NE) + a = + a NE = (24,766+260) \$

FROM EQ. (2). (50) 120 = TE (50) 120 = 4289 mi = 25,026 5 = 24,732 \$

FINALLY. (NA) CIRC = (NO)TRED + AND OR DUS = (24,785-24,732) 445 OR AND = 53 1/3

12.90

GIVEN: 5 = T. O. 16 5 (xy) = 0; k=2'b/4: NEGLECT FRICTION AND

MROS; W= 3 16 FIND: (a) (an), AND (an) - 6 in. ---(b) (acourages) (C) (NB)0

mag

(a) FB = 0 AND AT A, Fr = - FB = 0

(QA)=0 (QA)8=0

(b) 2. [F== mar: - F== m ("- r0")

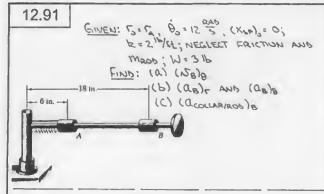
NOTING THAT acouragnos = ", HAVE AT A ...

O = m[acouragnos - (6 in.)(16 E)2]

OR (ACOLLAR/ACO) 1536 5 (C) AFTER THE CORD IS CUT, THE ONLY HORROWTHL FORCE ACTING ON THE COLLAR IS DUE TO THE SPRING. THUS, ANGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED.

.. TA MUTA) = TO M (NO) WHERE (NA) = TA BO THEN .. (NB) = 61N. [(61N.)(16 RAD)]

OR (158) - 32.03



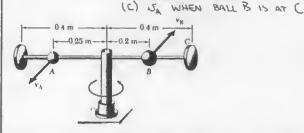
(a) AFTER THE CORD IS CUT, THE DULY HORIZONTAL FORCE ACTING ON THE COLLAR IS DUE TO THE SPRING. THUS, HUGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED.

THEN. (No = 6 III. [(6 III.)(12 Rab)]

(b) Have.. $F_0 = 0$ Now.. $\Rightarrow \xi F_r = ma_r$: $-(F_{50})_{g} = \frac{1}{9}(a_{g})_{r}$ or $(a_{g})_{r} = -\frac{216}{316} = 32.2 \frac{44}{32} = -21.467 \frac{43}{32} = -257.6 \frac{10.4}{32}$

(C) HAVE.. $Q_{7}=7-7\dot{\theta}^{2}$ NOW.. QCOLLARADO ? T AND $\dot{\theta}_{8}=\frac{(N_{0})_{9}}{7}$ THEN.. AT B: $(Q_{COLLARADO})_{8}=-257.6\frac{N}{3}+181N.4\frac{(24.0\frac{N}{3})}{181N}$ OR $(Q_{COLLARADO})_{8}=-226\frac{N}{3}$

12.92 GIVEN: $m_{h^{\pm}}$ 0.2 kg, m_{θ} : 0.4 kg, m_{θ} 0.5; $(S_{h})_{o}$ = 2.5 %; NEGLECT FRICTION; AT t=0, BALL B BEGINS TO MOVE FROM B TO CFIND: (a) $(a_{\theta})_{r}$ AND $(a_{\theta})_{\theta}$ AT t=0 (b) a_{θ}_{ROD} AT t=0



(a) WHEN THE PIN HOLDING BALL B IS REMOVED,

THERE ARE THEN NO HORIZONTAL FORCES

ACTING ON THE BALL. THEREFORE,

AT \$\frac{1}{2}O_1 \quad \text{Fr} = 0 \quad \text{AND} \quad \text{Fg} = 0

(CONTINUED)

12.92 continued

SO THAT

[(a₀)₀]=0

(b) HAVE .. Or : "- (b')

NOW .. QBIRCO = " AND B = TA (SA) =] 2 ...

THEN, AT 1=0.. (QBIRCO) - (TB) = [TA) =] 2 ...

OR (QBIRCO) = 0.2 m - (2.5 m) 2

OR (CORRES) 3: 20.0 M

(C) NOW, Fr = 0 AND FO = 0 WHILE B IS MOVING

FROM ITS INITIAL TO ITS FINAL POSITION. THUS,

ANGULAR MOMENTUM ABOUT THE SHAFT IS

CONSERVED. THUSS.

To ma (NA) + ((c) Me (No) = To ma No + To mo No

WHERE () DENOTES THE STATE WHEN BALL B IS

AT (. NOW...

(NE) = ((E) & = ((E) \(\frac{(NE)}{\text{TA}} \)

AND NE = (E) \(\hat{b} = \text{TE} \(\hat{O} \)

THEN -- $\Gamma_A M_A (L_A^{-1})_a + (\Gamma_B)_a M_B \left[\frac{(\Gamma_B)_a}{\Gamma_A} (L_A)_a \right] : \Gamma_A M_A L_A^{-1} + \Gamma_a^{-1} M_B \left(\frac{\Gamma_a^{-1}}{\Gamma_A^{-1}} L_A^{-1} \right)$ OR $\left\{ 1 + \frac{m_B}{m_A} \left[\frac{(\Gamma_B)_a}{\Gamma_A} \right]^2 \right\} (L_A^{-1})_a = \left\{ 1 + \frac{m_B}{m_A} \left(\frac{\Gamma_B^{-1}}{\Gamma_A} L_A^{-1} \right) L_A^{-1} \right\}$

12.93

GIVEN: INITIAL STATE OF THE BALL

DECINED BY 1, B, FIND

THE FINAL STATE DEFINED

BY 12, B2

Find: (a) RELATION AMONG. $1, \theta, 1_2, \text{ AND } \theta_2$ $(b) \theta_2 \text{ WHEN } 0, = 0.0 \text{ m}, 1_2 = 0.0 \text{ m}, 1_3 = 35°$

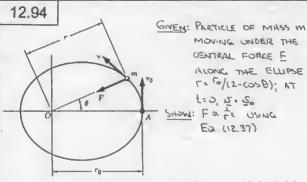
(a) FOR STATE I OR 2... $+1\Sigma F_{\eta} = 0$: $T\cos\theta - W = 0$ OR $T = \frac{mq}{\cos\theta}$ $\frac{1}{2} = \Sigma F_{\eta} = m\alpha_{\eta}$: $T \sin\theta = m\frac{S^2}{2}$ WHERE $T = 1 \sin\theta$ OR $S^2 = 91 \sin\theta$ $T \tan\theta$ $S^2 = \frac{1}{2} \sin\theta \cos\theta$ THEN FOLLOWS THAT $S^2 = \frac{1}{2} \sin\theta \cos\theta$ NOW.. $\Sigma M = 0 \Rightarrow H_{\eta} = \cos\theta$ Then $T = \cos\theta$ $T = \cos\theta$ T =

Now. $\Sigma M_1 = 0 \Rightarrow H_1 = CONSTRUCT$ Thus. $\Gamma_1 M S_1 = \Gamma_2 M S_2$

DR No PENDO

COMBINING EUS. (1) AND (2) - (P, SIND) 2 1, SIND, TAND, TAND,

(b) HAVE... $(0.8 \text{ m})^3 \sin^3 \theta_1 \tan \theta_1 = l_2^3 \sin^3 \theta_2 \tan \theta_2$ OR $\sin^3 \theta_2 \tan \theta_2 = 0.313197$ OR $\theta_1 = 43.6^\circ$



HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2}$ EQ. (12.37)

WHERE $u = \frac{1}{4}$ AND $mh^2 = \omega NSTANT$ NOW. $u = \frac{1}{4} = \frac{1}{4}(2 - \omega S\theta)$ THEN: $\frac{du}{d\theta} = \frac{d}{d\theta} \left[\frac{1}{4}(2 - \omega S\theta)\right] = \frac{1}{4}SN\theta$ AND $\frac{d^2u}{d\theta^2} = \frac{1}{4}\omega S\theta$ THEN $= F n \left(\frac{1}{4}\right)^2 \left[\left(\frac{1}{4}\cos\theta\right) + \frac{1}{4}(2 - \omega S\theta)\right] = \frac{2}{4}\frac{1}{4}$ NOTE: F > 0 IMPLIES THAT F = 1S ATTRACTIVE.

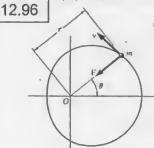
12.95 GIVEN: PARTICLE OF MASS M MOVING
UNDER A CENTRAL FORCE F ALONG
THE PATH (= (5 SIN B)
SHOW: F & FS USING EQ. (12.37)

HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2}$ Eq. (12.37) WHERE $u = \frac{F}{mh^2}$ AND $mh^2 = constant$ $\therefore F \times u^2 \left(\frac{d^2u}{d\theta^2} + u \right)$ Now ... $u = \frac{1}{F} = \frac{1}{65000}$

THEN - $\frac{du}{d\theta} = \frac{1}{d\theta} \left(\frac{1}{\sqrt{cos^2 \theta}} \right) = -\frac{1}{\sqrt{cos^2 \theta}}$ $= \frac{1}{\sqrt{cos^2 \theta}} \frac{1 + \cos^2 \theta}{\sqrt{cos^2 \theta}}$ $= \frac{1}{\sqrt{cos^2 \theta}} \frac{1 + \cos^2 \theta}{\sqrt{cos^2 \theta}}$

THEN .. F \(\langle \(\frac{1}{\rangle} \rangle \left(\frac{1}{\rangle} \rangle \frac{1}{\ran

NOTE: FOO IMPLIES THAT F IS ATTRACTIVE.



GIVEN: PARTICLE OF MASS

M MOVING UNDER

THE CENTRAL FORCE

F. ALCUG THE

CARDIOIS

T= \frac{r_2}{2}(1+\cos\theta)

SHOW: F.C. T+ USING

E.G. (12.97)

HAVE $\frac{d^2u}{db^2} + u = \frac{F}{mh^2u^2}$ EQ. (18.37)

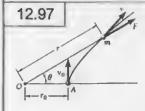
WHERE $u = \frac{1}{f}$ AND $mh^2 = constant$ $f \propto u^2 \left(\frac{d^2u}{db^2} + u\right)$

Now .. U = 1 : 3 1 cosb = 2 sing

AND $\frac{\partial^2 u}{\partial z^2} = \frac{c}{c} \frac{(z^2 - 1)^2}{(z^2 - 1)^2}$ $= \frac{c}{c} \frac{(z^2 - 1)^2}{(z^2 - 1)^2} = \frac{c}{c} \frac{(z^2 - 1)^2}{(z^2 - 1)^2}$ $= \frac{c}{c} \frac{(z^2 - 1)^2}{(z^2 - 1)^2} = \frac{c}{c} \frac{(z^2 - 1)^2}{(z^2 - 1)^2}$

THEN. Fa (+)2[202(3-3)+ +] = 3 - 2

NOTE: FOO IMPLIES THAT E IS ATTRACTIVE.



GIVEN: PARTICLE OF MASS M
MOVING UNDER THE
CENTRAL FORCE F
ALONG THE PATH
T=Tol VCOS20; AT \$=0,
15 = 15
SHOW: FAT USING EQ.(12.37)

HAVE $d^{2}u + u = \frac{F}{mh^{2}u^{2}}$ Eq. (12.37) WHERE $u = \frac{1}{h}$ AND $mh^{2} = constant$:. $F = u^{2} \left(\frac{d^{2}u}{d\theta^{2}} + u \right)$

 $= -\frac{1}{4} \frac{(\cos 50)^{\frac{1}{2}}}{(\cos 50)^{\frac{1}{2}}} = -\frac{1}{4} (\frac{1}{6})^{\frac{1}{2}} [1 + (\frac{1}{6})^{\frac{1}{2}}]$ $= -\frac{1}{4} \frac{36}{4} = -\frac{1}{4} \frac{36}{4} (\frac{1}{4} \cos 50) = -\frac{1}{4} \frac{36}{4} (\frac{1}{6} \cos 50) = -\frac{1}{4} \cos 50) = -\frac{1}{4} \frac{36}{4} (\frac{1}{6} \cos 50) = -\frac{1}{4$

IHEN .. Ex (7) 5 \ - \frac{24}{17} \[1+ (\frac{12}{12})_4 \] + \frac{1}{17} \] = - \frac{24}{17}

HOTE: FCO IMPLIES THAT E IS REPULSIVE.

12.98

GIVEN: PARABOUR TRAJECTORY OF GALILEO SPACE CRAFT ABOUT THE EARTH; MINIMUM ALTITUSE = 960 EM

FIND: NMAX

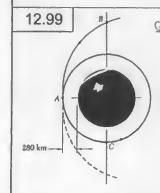
FIRST NOTE .. R= 6.37 110 m SO THAT 6=(6.37-10-960-103)m NOW -- Nomax = No

AND FROM PAGE 709 OF THE TEXT No = [ZGM - [ZgR2 USING EQ. (12.30)

THEN .. ISMAX = [2-9.81 M/52 x (6.37 = 10 m)2]1/2

= 10 421.7 2

OR JMAX : 10.42 8m



GIVEN: TARABULIC APPROACH TRAJECTURY AND CIRCULAR ORBIT ABOUT VENUS; MVENUS= 4.87-10 bg R = 6052 km

FIND: (a) (NA) PAR (b) 105,1

FIRST NOTE .. 1/4 = (6052+280) km . 6332 km (a) FROM PAGE 709 OF THE TEXT, THE VELOCITY AT THE POINT OF CLOSEST APPROACH ON A PARABOLIC TRAJECTORY IS GIVEN BY

THUS, (Na) pag = [Zx66.73710 kg.52 x 4.87x104 kg] 1/2 = 10 131.4 %

OR (Un) PAR = 10.13 5

(b) HAVE .. (NA) CIRC = (NA) PAR + ANA

Now. (Na)ciec = (GM EQ. (12.44) = = ((5/2) 000

THEN. DUA = (Un)ma - (Un)ma = (1/2 -1) (10.1314 km) = - 2.97 Em

" 122 = 1201 :

12,100

GIVEN: TRAJECTORY OF GALILED SPICECRAFT ABOUT THE ENETH; AT THE POINT OF CLOSEST APPROACH, IJ . 46.2+10 \$ ALTITUDE = 188.3 mi FIND: E AT POINT OF CLOSEST APPROACH

FIRST NOTE .. R = 3960 mi=20.9088 40 \$ AND 5 = (3960+188.3) mi . 4148.3 mi = 21 9030×10 H

HAVE - T = GM (14 E cos B) ED. (12.39')

AT POINT O, 1=10, 8=0, h=h=1=56
ALSS.. GM=gR2 ED. (12.30)

THEN - 1 = 9R2 (1+E) OR E = (0.502 -1 = (21.9030=10 ft)(46.2m3 =) 2 ft)2 (32.2 ft)(20.9088=10 ft)2

188.3 mi

12.101

GIVEN: TRAJECTORY OF GALILED SPACECRAFT ABOUT IO: AT THE POINT OF CLOSEST APPROACH T = 1750 mi 5= 49.470 \$; MI= 0.01496 Mas FIND: E AT POINT OF CLOSEST AFFROACH

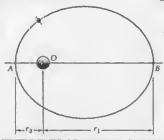
FIRST NOTE .. 6 = 1750 mi = 9.24 110 ft REARTH = 3960 mi . 20.908840 ft HAVE .. ! = GM (1+ E cosb) Ea. (12.39)

AT POINT D, r=6, 0=0, h=h=555 ALSO .. GM TO = G (0.01496 MENERY)

= 0.01496 9 REARTH UUNG ED (12.30) 1 = 0.01496 9 REARTH (1+ E)

6 25 5 (9.24×10 CE)(49.4×10 3 5)2 OR E = 0.014969 RZZ - 1 = (9.24 × 10 ° E) (49.4 × 10 ° E) 12 - 1 OR E = 106,1

12.102



GIVEN: ELLIPTIC ORBIT OF A SATELLITE ABOUT A PLANET OF MASS M DERNE: 1 + 1 = 2GM

HAVE . I = BM (1+ E COSD) EQ. (12.39')

Now. AT A: 5.6, 8=0: 1. = 6M (1+6) (1)

AT B: F= F, B= 180: : + = GM (1-E)

OR 1 = 2GM Q.E.D.

12,103 GNEN: ELLIPTIC AND CIRCULAR DABITS OF THE SPACE SHUTTLE ABOUT THE EARTH: NO PERPENDICULAR 70 F. FIND: (a) 50 R = 3960 mi (b) QUE

FIRST NOTE .. T = (3960+ 40.3) mi + 4000.3 mi 57 OIX 2121.13 = Ta = (3960+336) mi = 4296 mi \$2.6829x106 St R = 3960 mi = 20.90BB NO 4

(a) FROM THE SOLUTION TO PROBLEM 12, 102, HAVE FOR THE ELLIPTIC ORBIT AB ..

Now.. h=h== ~~~ GM=gR2 (EQ.(12.50)) THEN ... 1/A + 1/B = 29R2 (7,15)2 DR NS = R (29)1/2 = 26,272 5

OR No = 26.3=10 st (b) FOR THE ELLIPTIC ORBIT AB HAVEh= ha= ho: [a No = 10 (No) po

THEN .. (No) AB = 4000.3 mi = 26,272 ft = 24, 464 \$

FOR THE CIRCULAR ORBIT, USE EQ. (12.44) (28) PSE = (3/6) = 50 2088 x10, EF (355 4A25 24) = 24,912

FINALLY .. (No) CIRC = (No) AB + DNO OR DNO = (24,912-24,464) (445).

OR DUE : 448 -

12.104 GIVEN: A PLANET OF RADIUS R AND A SPACE PROBE IN A CIRCULAR ORBIT ABOUT THE PLANET AT AN ALTITUDE OR AND HAVING A VELOCITY So: ELLIPTIC ORBIT, WHERE 5= \$55.

FIND: BMIN SO THAT THE PROBE DOES NOT CRASH

FOR THE CIRCULAR ORBIT .. 150 - (GM [EQ.(12.44)] WHERE TA = R+ aR = R(1+x) THEN .. GM = 52 R (1+0x)

(CONTINUES)

12.104 continued

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FUR THE ELLIPTIC ORBIT ..

Now. h= h= (()) AB = (() () () () ()

THEN ..

R(1+a) + To = 250 R(1+a)
[R(1+a) A55]2 = Z B2R(1+X)

NOW .- BMIN CORRESPONDS TO B- R. THEN ..

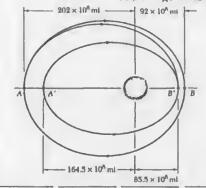
OR BMIN 1 2200

12.105 GIVEN: ELLIPTIC ORBITS AB AND A'B' OF A SPACECRAFT ABOUT THE SUN

AND THE ELLIPTIC TRANSFER DRBIT AB'; MSUN = (332.8 HIS')

FIND: (a) NA (ON AB)

(B) I AVA I AND I DIBI



FIRST NOTE .. REARTH = 3960 mi = 20,9088 x10 ft 1 = 202 x10 mi = 1066.56 x109 \$2 To = 92 = 10 mi = 485.76 = 10 ft

FROM THE SOLUTION TO PROBLEM 12,102, HAVE FOR MY ELLIPTIC CRBIT ABOUT THE SUN ..

(a) FOR THE ELLIPTIC ORBIT AB HAVE ..

1,=14, 12=18, h= ha= 14 1/4 ALSO .. GM SUN = G[(332.8=10)) MENETH]

= 9 REARTH (352.8×103) USING ED. (12.30)

= 52, 431 \$ OR NA = 52.4 x10 \$

12.105 continued

(b) FROM PART (a) HAVE 2GM sun = (= Nai (+ +)

THEN, FOR ANY OTHER ELLIPTIC ORBIT ABOUT

FOR THE ELLIPTIC TRANSFER ORBIT AB HAVE .. 1=1A, 12=1B', h= hTR = [A (JA)TR

$$OR (S_A)_{TR} = N_A \left(\frac{r_A^2 + r_B^2}{r_A^2 + r_B^2} \right)^{1/2} = N_A \left(\frac{1 + \frac{r_A}{r_A}}{1 + \frac{r_A}{r_B^2}} \right)^{1/2}$$

$$= \left(52,431 \frac{\text{Gt}}{5} \right) \left(\frac{1 + \frac{202}{92}}{1 + \frac{202}{85.5}} \right)^{1/2}$$

$$= 51,113 \frac{\text{Gt}}{5}$$

NOW -- hTR = (ha)TR = (he')TR: [(NA)TR = (3'(NB')TR

FOR THE ELLIPTIC ORBIT A'B' HAVE ..

OR
$$S_{6}' = S_{A} \frac{C_{A}}{C_{6}'} \left(\frac{\frac{1}{C_{A}} + \frac{1}{C_{6}}}{\frac{1}{C_{A}} + \frac{1}{C_{6}}} \right)^{1/2}$$

$$= (52,431 \frac{C_{A}}{5}) \frac{202 \times 10^{6} \text{ mi}}{85.5 \times 10^{6} \text{ mi}} \left(\frac{1}{164.5 \times 10^{6}} + \frac{1}{92 \times 10^{6}} \right)^{1/2}$$

FINALLY .. (NA) TR = NA + ANA OR ANT = (51, 113 - 52431) 5

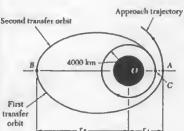
OR 105/1=1318#

NR = (NR) /1R + DNB AND OR DUB = (116, 862-120, 758) } = - 3896 \$

OR 1251=3900 5

12.106

GIVEN: PARABOLIC APPROACH TRAJECTORY. ELLIPTIC TRANSFER ORBITS AB AND BC. AND CIRCULAR DRBIT OF A SPACE PROBE ABOUT MARS! TA = 9x10 km, To = 180x10 km;



Approach trajectory MMARS = 0.1074 MEARTH

FIND: (a) | ANA | (b) 10581 (C) 105/

(CONTINUED)

12,106 continued

(a) FOR THE PARABOLIC APPROACH TRAJECTORY, POINT A IS THE POINT OF CLOSEST APPROACH. THEN, FROM PAGE TOP OF THE TEXT HAVE

NOW. GMMARS: G(0.1074 MENRY) = 0.1074 g REARTH USING ED. (12.30)

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR

ANY ELLIPTIC ORBIT ABOUT MARS...
$$\frac{1}{\Gamma_1} + \frac{1}{\Gamma_2} = \frac{2GM_{MARS}}{h^2} \qquad (1)$$

FROM ABOVE .. 26Mmas: Ta [(UA) ma] THEN .. FOR THE ELLIPTIC TRANSFER ORBIT AB ..

WHERE
$$h_{AB} = (h_{A})_{AB} = (h_{$$

= 3008.0 \$

FINALLY .. (15) AB = (15) PAR + ANA OR ANA = (3008.0-3082.3) S OR 105/1=74.3 5 4

(b) FOR THE ELLIPTIC TRANSFER ORBIT AB ..

= 150.40 m NOW APPLY EQ. (1) TO THE SECOND ELLIPTIC TRANSFER ORBIT BC AND USE

OR
$$(\sqrt{2})^{8}C = \frac{180 \times 10^{3} \text{ pm}}{(\sqrt{2})^{8} \times (\sqrt{2})^{1/2}} \left(\frac{180 \times 10^{3} \text{ pm}}{(\sqrt{2})^{1/2}} + \frac{1}{4 \times 10^{3} \text{ pm}} \right)_{1/2}$$

(C) FOR THE ELLIPTIC TRANSFER ORBIT BC ..

THEN- (NE) = (hc) = (BOXID) km x 101.62 m/5 = 4572.9 7

FOR THE CIRCULAR ORBIT HAVE ..

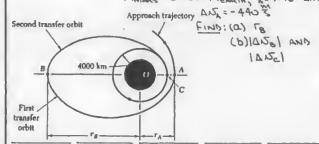
12.106 continued

RECALLING FROM DART (a) THAT $(N_h)_{pag}^2 \sqrt{\frac{2GM_{mass}}{G}}$ HAWE $(N_c)_{circ} = (N_h)_{pag} \left(\frac{r_h}{2r_c}\right)^{1/2}$ $= (3082.3 \frac{m}{5}) \left(\frac{9 \times 10^3 \text{ km}}{2 \times 4 \times 10^3 \text{ km}}\right)^{1/2}$ $= 3269.3 \frac{m}{5}$ FINALLY ... $(N_c)_{circ} = (N_c)_{gc} + \Delta N_c$ $\Delta N_c = (326.3 - 4572.9) \frac{m}{5}$

12.107

GIVEN: PARABOLIC APPROACH TRAJECTURY,
ELLIPTIC TRANSFER ORBITS AB AND
BC, AND CIRCULAR ORBIT OF A
SPACE PROBE ABOUT MARS;
MMARS = 0.1074 MENERY, [=9×10] km,

OR 105= 1304 =



(a) FOR THE PARABOLIC APPROXICH TRAJECTORY, POINT A IS THE POINT OF CLOSEST APPROXICH. THEN, FROM PAGE TOP OF THE TEXT HAVE

(UL) PAR = \[\frac{2GM_{MARS}}{C}

NOW -. GMMAS = G(0.1074 MENATH)
= 0.1074 gREADH USING EQ.(12.30)

THEN. (JA) PAR = REARTH (2x0.10749)12
= (6.57 × 10 m) (0.2148 × 9.81 m/s²)12
= 3082,3 m

NOW - (JA) AR = (JA) PAR + DJA = (3082.3-440) = 2642.3 =

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR ANY ELLIPTIC ORBIT ABOUT MARS...

FROM ABOVE .. ZGMMAN = TELLIFIC TRANSFER ORBIT AB ..

WHERE has = (ha) as = Ta (Na) as

THEN - I 1 To = Ta (Na) pas]2

= [(Na) ma]2 To [(Na) has]2

OR $\frac{1}{16} = \frac{1}{16} \left\{ \left[\frac{(15_{A})_{AB}}{(15_{A})_{AB}} \right]^{2} - 1 \right\} = \frac{1}{9 \times 10^{3} \text{ km}} \left[\left(\frac{3082.3 \frac{\text{m}}{2}}{2042.3 \frac{\text{m}}{2}} \right)^{2} - 1 \right]$

OR To = 24.946 x103 km OR To = 24.9x103 km

(b) FOR THE ELLIPTIC TRANSFER ORBIT AB..

has = (ha)as = (hs)as : [(Ua)as = [(Us)as (CONTINUED)

12.107 continued

THEN -. (NE) AS = $\frac{9210^3 \text{ km}}{24.94610^3 \text{ km}} = 2642.3 \frac{\text{m}}{\text{s}}$ = 953.3 $\frac{\text{m}}{\text{s}}$

NOW APPLY EQ. (1) TO THE SECOND ELLIPTIC TRANSFER DRBIT BC AND USE

THEN ..
$$\frac{1}{r_{8}} + \frac{1}{r_{2}} = \frac{r_{8} (U_{8})_{9C}}{(U_{8})_{9C}}^{2}$$

OR $(U_{8})_{9C} = \frac{(U_{8})_{9C}}{r_{8}} \left(\frac{r_{8}}{r_{8}} \cdot \frac{1}{r_{8}} \right)^{1/2}$

$$= \frac{3082.3 \frac{m}{5}}{24.946 \times 10^{3} \text{ bm}} \left(\frac{9 \times 10^{3} \text{ km}}{4 \times 10^{5} \text{ km}} \right)^{1/2}$$

THEN.. (5) BC = (5) AS + DNB OR DNB = (68) 2-953.3)

Now.. FOR THE ELLIPTIC TRANSFER ORBIT BC..

hac = (ha)ac = (hc)ac: F(Ua)ac + C(Ua)ac

THEN.. (NE)ac = 24.946 no bm = 688.2 m

= 4292.0 m

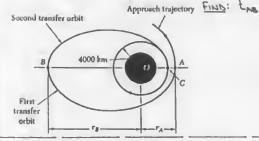
FOR THE CIRCULAR ORBIT HAVE ..

FINALLY -- (UE) circ = (UE) ac + AUE OR AUE = (3269.3-4292.0) \$\frac{1}{2}\$

OR 1A\frac{1}{2}\$ = 1023 \$\frac{1}{2}\$

12.108

GIVEN: ELLIPTIC TRANSFER ORBIT AB OF PROBLEM 12.106; TA = 9×103 km,



FROM THE SOLUTION TO PROBLEM 12.106 HAVE

(UTA)AB = 3008.0 \$

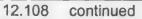
FROM EQ. (12.45) IT FOLLOWS THAT

\$\frac{1}{48} = \frac{1}{2} (\tau_{\text{Eulipse}})_{\text{AB}} = \frac{1}{12} \text{Tights}

WHERE Q = \frac{2}{\Gamma_{A} + G_{B}} = \frac{2}{2}(9x10^{2} + 180x10^{3}) = 94.5x10^{3} km

AND b = \frac{1}{G_{A}G_{B}} = \frac{2}{G_{A}G_{B}}(180x10^{3})^{1/2} = 40.249x10^{3} km

ALSO... has = \Gamma(15A)_{AB} = (9x10^{3}M) \frac{2}{3}(27.072x10^{3})^{1/2} \frac{1}{2} \frac{1}



THEN -- tag = T(94.5 *10 m) (40.249 *10 m)

= 441.384 *10 5

OR ta=122 h 36 MIN 245

12.109

GIVEN: ELLIPTIC ORBIT OF THE CLEMENTINE

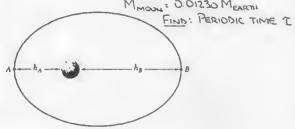
SPACECRAFT ABOUT THE MOON;

ha = 400 km, ha = 2940 km;

RMOON = 1737 km,

MMOON = 0.01230 MEARNI

- a -



FIRST NOTE..

(A = (1737+400) = 2137 km

(B = (1737+2940) = 4677 km

Now. 2 = 271ab Eq. (12.45)

WHERE Q=2(12074 4677) tm =2(21374 4677) tm

AND b= TATE

FROM THE SOLUTION TO PROBLEM 12.102 HAVE ..

NOW -- GMMOON = G(0.01230 MEARTH) = 0.01230 gREARTH USING ED. (12.30)

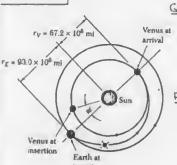
THEU. LZ 2(0.01230 9 REARTH) 0.01230 9 REARTH CA+TE

= 50 (SNO.012309 REACH)

CR $h = bR_{EARTH} \left(\frac{0.012309}{\Omega} \right)^{1/2}$ THEN.. $I = \frac{2\pi ab}{bR_{EARTH} \left(\frac{0.012309}{\Omega} \right)^{1/2}} = \frac{2\pi a^{3/2}}{R_{EARTH} (0.0.12309)^{1/2}}$ $= \frac{2\pi (340) \times 10^3 \text{ m}}{(6.37 \times 10^5 \text{ m})(0.01230 \times 9.81 \text{ m/s}^2)^{1/2}}$ $= 17.8571 \times 10^3 \text{ s}$

OR T. 44 57 MIN 375

12.110



GIVEN: ORBITS OF VENUS

AND THE EARTH AND
THE ELLIPTIC TRANSFES
ORBIT OF A SPACE
PROBE;
MSUN - 332 BNOTH

FIND: \$\phi\$, THE RELATIVE

POSITION OF VENUS

WITH RESPECT TO

THE EARTH AT THE

TIME OF INSERTION

FIRST DETERMINE THE TIME LARDER FOR THE PROBE TO TRAVEL FROM THE EARTH TO VENUS. NOW... LARDER = 2 Tra

WHERE IT IS THE PERIODIC TIME OF THE ELLIPTIC TRANSFER ORBIT. APPLYING KEPLER'S THIRD LAW TO THE ORBITS ABOUT THE SUN OF THE EARTH AND THE PROBE OBTAIN.

TEARTH QEARTH

WHERE are = 2 (FE+FV) = 2 (93×10° + 67.2×10°) mi

WID GEARTH = PE (NOTE: ERARH = 0.0167)

THEN .. \$\frac{1}{\text{PROBE}} = \frac{1}{2} \left(\frac{\alpha_{\text{TE}}}{\text{TE}} \right)^{3/2} \tag{365.25 DAYS}

= \frac{1}{2} \left(\frac{\text{ROBIND}}{\text{P3.0 x10}} \frac{\text{mi}}{\text{mi}} \right)^{3/2} \left(365.25 DAYS)

= 145.977 DAYS

= 12.6124 410 5

IN TIME trade, VENUS TRAVELS THROUGH THE ANGLE θ_V GIVEN BY $\theta_V = \frac{1}{V_V} t_{PROBE} = \frac{1}{V_V} t_{PROBE}$

ASSOMING THAT THE ORBIT OF VENUS IS CIRCULAR (NOTE: EVENUS: 0.0068). THEN, FOR A CIRCULAR ORBIT..

No = ([EQ. (12.44)]

Now. GMsun = G(332.8×10 MEARTH)
= 332.8×10 (9 REARTH) USUG EQ.(1230)
THEN- By = LPROBE [332.8×10 GREARTH)]1/2

= trade REARTH (352.8 4103) 1/2

WHERE REARTH = 3960 mi = 20,9088×106 ft 12005 Ty = 67.2×106 mi = 354.816×109 ft

THEN - By = (12.6124×10°5)(20.9088×10°4) (332.8×10° x32241)/2
(354.816×10°41)/2

= 4.0845 RAD = 234.02°

FINALLY .. \$ = 0, - 180°

= 234.02°-180°

OR \$= 54.0°

12.111

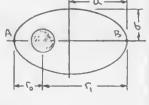
GIVEN: ELLIPTIC ORBIT ABOUT THE SUN OF THE WIMET HYAKUTAKE; E: 0999887. TMIN = 0.230 RE; KE = GEARTH FOR THE EARTH'S ORBIT ABOUT THE

MA FIND: T FOR THE COMET

USING EQ. (12.39') HAVE FOR ANY ELLIPTIC ORBIT ABOUT THE SUN ..

7 = (H200) (1+ E 0018)

1 = GM + (1+ E) (1)



AT B, D=180: = = GNown (1-E) (2)

FORMING (2) => = 100 DR 1= 146 TO

Now. Q = 2 (10+1,) = 2 (10 + 1+6 10) = 10

APPLYING KEPLER'S THIRD LAW TO THE ORBITS ABOUT THE SUN OF THE EARTH AND THE COMET HAVE --

Q COME Q EARTH

FROM ABOVE -. OCOMET = (1- ECOMET " 1- ECOMET" 1- ECOMET 0.230 RE 1- EWMET

CEARTH = RE 0.230 RE 1 = (0.230) = (0.230) TEARTH = (RE) and THEN .. OR [LOMET = (0.230)1/2 (140)

12.112 GIVEN: PARABOLIC TRAJECTORY OF THE GALILED SPACE CRAFT ABOUT THE EARTH; J= 10.42 FIND: tac 7330 km

HAVE .. Ea. (12.39') FOR A PARABOLIC TRAJECTORY, E=1 NOW .. AT A, 8=0: 1= GM (1+1) 12 AT (8=90: = GM OR R= ZGM OR TE = GM

1. rc = 27. AS THE SPACECRAFT TRAVELS FROM B TO C THE AREA SWEPT OUT IS THE PARABOLIC AREA BAC. THUS,

(CONTINUED)

OR TOMET = 91.8×10 yr

12.112 continued

AREA SWEPT OUT = ABAC = \$(T)(T) = \$T2 NOW .. of = th , WHERE h = CONSTANT A= 2ht OR to 2 The h= TAUA THEN 2×3× = 16 CA = 16 7530 km 10.42 km/s = 3751.85

DR Lac = 14 ZMIN 325



GIVEN: PARABOLIC APPROACH TRAJECTORY AND CIRCULAR ORBIT ABOUT VENUS OF A SPACE PROBE: R=6052 km Myens = 4.87 mo2 kg

FIND: tac

FROM THE SOLUTION TO PROBLEM 12.99 HAVE .. (JA) PAR . 10, 131.4 5 AUD (JA) ORC = (2 (JA) MR = 7164.0 5 ALSO, TA = (6052+280) EM = 6332 EM FOR THE PARABOLIC TRAJECTORY BA HAVE = GMV (1+ E COSD) [EQ.(12.391)]

WHERE E=1. NOW-- GMV (1+1)

AT A, $\theta=0$: $\frac{1}{L} = \frac{GMV}{h_{BA}}$ (1+1)

AT B, $\theta=-90$: $\frac{1}{L} = \frac{GMV}{h_{BA}}$ (1+0)

AT B, $\theta=-90$: $\frac{1}{L} = \frac{GMV}{h_{BA}}$ OR $\frac{1}{L} = \frac{h_{BA}}{GMV}$

:. (B = 2/7 AS THE PROBE TRAVELS FROM B TO A, THE AREA SWEDT OUT IS THE SEMIPARABOUC AREA DEFINED BY VERTEX A AND POINT B. THUS,

(AREA SWEDT OUT) BA = ABA = 3(TA)(TE) = 3TA Now - dA = 1h , WHERE h = CONSTANT

Now - It = 2"

THEN A = 2ht OR tex = ZABA

THEN A = 2ht OR Tex = ZABA

THEN A = 2ht OR TEX = B TA . HBA = TANA = 8 6332×103 m 10131.4

FOR THE CIRCULAR TRAJECTORY AC,

the = \frac{1/2 \Gamma_{\text{Line}}}{\text{UA} \text{Line}} = \frac{\text{TI}}{2} \frac{\text{U332-10}^3 m}{7164.0 m/s} = 1388.37 \text{S}

FINALLY .. toc = ton + the = (1666.63+1388.37)5 = 3055.05 OR to = SO MIN SSS

12.114

GIVEN: CIRCULAR ORBIT OF RADIUS TIR OF A SPACE PROBE HAVING VELOCITY IS ABOUT A PLANET OF RADIUS R; AT POINT A, VELOCITY IS REDUCED TO BUS (BCI) SO THAT PROBE IMPACTS AT POINT B FIND: 4AOB IN TERMS OF IT AND B

HAVE FOR THE CIRCULAR ORBIT No= (GM (EQ.(12.44)) OR GM= nRs2 FOR THE ELLIPTIC ORBIT ABC T = GM (1+ (COSB) [EQ.(12.39)] 15 HABE = (HA)ABE = TA (NA) ARE

=(nR)(BND) THEN - T = (nRBS) (1+ E cos B) = TRBZ (I+ Ecos B)

NOTING THAT POINT C IS THE PERIGEE OF THE ELLIPTIC IMPACT TRAJECTURY SO THAT ANGLE 8 IS DEFINED AS SHOWN HAVE ..

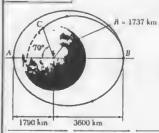
AT A, 8=180: nR = nRBE (1-E) OR E=1-B

AT B: R = 1/2 (1+ E WSB) = 1/2 [1+ (1-B) WSB] FINALLY .. (No = (No) + AND OR AND = (869.43-950.43)]

OR $COS\theta = \frac{nB^2-1}{1-R^2}$ Now - 4 AOB = 180 - 8 SO THAT COS (180- 4 AOB) = mB2-1 OR - COS (4 AOB) = 1082-1-BZ

OR * ADB = cos' 1- nB

12.115



GIVEN: ELLIPTIC ORBIT AND ELLIPTIC IMPACT TRAJECTORY OF LUNAR ORBITTER Z; MMOON : 0.01230 MEARTH FIND: I AND FOR IMPACT

AT POINT C

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC ORBIT AB ..

TA+TB = ZGMMON

WHERE has = (ha) as = (B(NB) AB

AND GMMOUN = G(0.01230 MENRTH) = 0.012309 REARTH USING EQ. (12.30)

THEN .. 1 + 1 = 2(0.01230 9 REARTH)

(CONTINUED)

12.115 continued

OR
$$(N_8)_{AB} = \frac{R_{EAR,TM}}{\Gamma_8} \left(\frac{0.02469}{\frac{1}{\Gamma_8} + \frac{1}{\Gamma_8}} \right)^{1/2}$$

$$= \frac{6.37 \times 10^8 \text{ m}}{3600 \times 10^3 \text{ m}} \left(\frac{0.0246 \times 9.81^{\frac{10}{3}}}{1790 \times 10^5 \text{ m}} + \frac{1}{3600 \times 10^3 \text{ m}} \right)^{1/2}$$

$$= 950.43 \frac{m}{3}$$

FOR THE ELLIPTIC IMPACT TRAJECTORY HAVE. $\frac{1}{h} = \frac{GM_{\text{moon}}}{h_{\text{ec}}^2} + C\cos\theta \qquad [E0. (12.39)]$

NUMBER her = (ha) or = (ba) or THE APOSEE OF THIS TRAJECTORY, HAVE 1 = GMMDA - C

OR (= ammosn - 1

AT (, 8=-70: = (Manage + (cos(-70)) OR C = 1 (1 - GM MOON)

THEN - GMMOON - 1 = 10570 (1 - GMMOON)

OR hac = (Mmoon (1+00>70))

OR (UB) = REARTH (0.012309 (1+0570)) 1/12

(No) BC = 3500 m (3.01230(9.81 32) (1+ 05700) 1/2 (No) BC = 6.37 × 10 m (3.01230(9.81 32) (1+ 05700) 1/2 = 869.43 3

OR 10501 . 81.0 5

12.116



GIVEN: HYPERBOLIC TRAJECTORY OF A PROBE, E=1.031; ALTITUBE AT B = 450 Em. NS & B2.9; FOR JUPITER R=71.492×103 /2m, M=1.9×1027 /2q FIND: (a) & AOB

FIRST NOTE.. $F_8 = (71.492 \times 10^3 + 450) \text{ km} \cdot 71.942 \times 10^3 \text{ km}$ (a) HAVE... $\frac{1}{L} = \frac{GM}{h^2} (1 + \epsilon \cos \theta)$ [Eq. (12.39')]

At A, $\theta = 0$: $\frac{1}{L} = \frac{GM_2}{h^2} (1+\epsilon)$ OR $\frac{1}{GM_3} = \frac{GM_3}{h^2} (1+\epsilon \cos \theta_0)$ At B, $\theta = \theta_0 = 4A08$: $\frac{1}{L} = \frac{GM_3}{h^2} (1+\epsilon \cos \theta_0)$ OR $\frac{1}{GM_3} = \frac{1}{L} (1+\epsilon \cos \theta_0)$

(b) NB

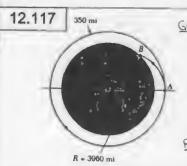
THEN .. TA (I+E) = TO (I+E COSBB)

OR COS DB = £[[(1+ E) -1] = 1 70.8403 km (1+1.031)-1] = 0.968 73 (CONTINUED)



OR OB = 14.3661° : 4 AD8 = 14.37° (b) FROM ABOVE _ h = GM, r, (1+ & cos00) WHERE h= m I Is x miss = 15 No sind \$ 100+82.9") - 97.2661° (18 Ng SIND) 2 = GM, 18 (1+ E cos 80) OR No = 5100 [CM3 (1+ E cos 80) 1/2 = 51497 2661 (66.75=107 m3 = 1.9=107 kg

·[1+(1.031)(0.96873)] OR 15 = 59.8 5



GIVEN: (IRCULAR ORBIT AND THE ELLIPTIC DEXENT TRAJECTORY OF A SPACE SHUTTLE; AJ =- SWHS; ALTITUDE AT B = 75 mi FIND: 4 AOB

FIRST NOTE .. R=3960 mi = 20,9088 x10 ft ra = (3960+350)mi = 4310 mi = 22.7568210 ft 1 = (3960+75)mi = 4035 mi

FOR THE CIRCULAR ORBIT HAVE

NEIRC = (9RE [EQ. (12.44)] = 20.9088 x10 ft (32.2 745) = 24, 871 15

NOW. (UZ) AB = NGIRC + AUZ = (24, 871-500) \$ = 24,371 475

FOR THE ELLIPTIC DESCENT TRAJECTORY HAVE.

T = GM + (cos 8 [EQ. (12.39)]

NOTING THAT POINT A IS AT THE APOGEE OF

NOTING THAT

THIS TRAJECTORY, HAVE...

AT A, 8=180: The GM - C

CR C= THE - TA

CR C= THE - TA AT B, 8 = 80 = 180 - 4AOB: 1 = 6M + Ceos 8

OR C= LOSB (To - GM)

THEN .. GM - TA = COSDA (TB - GM) OR COS By = GM -

NOW. h= (ha) as = (Na) as AND GM = 9RE EQ. (12.30)

FROM ABOVE - GR2 - TA (JCIRC) [EQ. (12.44)]
THEN- GM - TA (JCIRC) - TA (JCIRC) 2

THEN- GM - TA (JCIRC) 2

TA (JCIRC) 2

TA (JCIRC) 2

(CONTINUED)

12.117 continued

(0) 08 = \frac{1}{10} - \frac{1}{10} \left[\frac{100}{100} \right]^2 - \frac{1}{10} \left[\frac{100}{100} \right]^2 - \frac{1}{10} = \frac{1}{100} = \frac{1 SO THAT 4310 mi - (24,871 645) 4035 mi - (43,371 645) (24,871 645) = 0.64411

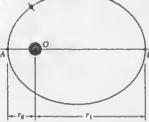
OR B = 49.901° FINALLY .. 4 AOB = 180-49,901

OR XAOB = 130.1°

12.118

GIVEN: ELLIPTIC ORBIT OF A SATELLITE AS SHOWN PHOM: == = = (= + =)

WHERE P = Pa = Pa



FROM THE SOLUTION TO PROBLEM 12.102 HAVE .. TO+ T = 2GM WHERE h= h= F.S.

CONSIDER THE SATELLITE AT POINT A ..

== EFn=man: Fx=m == O= Omlan)n

Now - FA = G Mm

THEN - M P = G TE [(85.(12.28)]

OR GM = p(05, 6) = ph EINALLY .. T+ T = 5(\$ 45) OR == = (10+ 1) Q.E.D.

12.119

GIVEN: ELLIPTIC ORBIT OF A SATELLITE AS SHOWN; FOR COMET

HYAKUTAKE, CO = U. 230RE € . 0.999887: Re = 149.6 =10 km

FIND: (a) E IN TERMS OF TO AND T (b) I, FOR COMET

HYAKUTAKE

(a) HAVE. T= GM (1+ € cos 8) EQ.(12.39')

AT A, 8=0: & = 60 (1+E) 2 = 10 (1+E)

AT B, 8=180: 1 = 5 (1-E) E) h2 OR GM = 1, (1-E)

THEN .. TO (1+ E) = T, (1-E) DR E = 11-10

(b) FROM ABOVE .. 1 = 1+6 5

WHERE TO: 0.230 RE

12.119 continued

THEN. 1 = 1+0.99988] = 0.230 (149.6=109 m)

OR T = 609 =10 m

NOTE: 1, = 4070 RE OR 1,= 0.064 LIGHT YEARS

12.120

GIVEN: ELLIPTIC ORBIT OF SEMIMAJOR AXIS Q AND ECCENTRICITY & OF A SATELLITE ABOUT A PLANET OF MASS M SHOW: H= (GMa (1- E2)

HAVE. == GM (1+ E COSB) EQ. (12.39) AT A, B = 0: = = GM (14E) AT B, B=180: 1 = GM (1-E)

OR 1 = 12 1 - E THEN .. P+1 = M 1+E + M 1-E = M 1-E

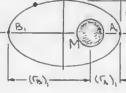
Now - a = 2(5+5) SO THAT 2a = 6M 1-62 h = (GMa (1-E2) Q.E.D.

12.121

GIVEN: TWO ELLIPTIC DRBITS OF SEMIMAJOR AXES Q, AND Q2 ABOUT A BODY OF MASS M; PERIODIC TIMES I, AND I, OF TWO SATELLITES IN THE ELLIPTIC ORBITS

DERVE: KEPLER'S , THIRD LAW (Tr = Qr = Qr =) USING EQS. (12.39) AND LIZ.45)

CONSIDER THE ELLIPTIC DABIT OF SATELLITE I. NOW 1 = GM + (cos B Ea(1239) - (B)



THEN, FOR ORBIT 1... AT A, 8.0: (6) = 6 + C,

AT B, 0-180: (C) = CM - C,

THEN -- (TA), + (TO), = (GM + C,) + (GM - C,) OR (TA) + (TA) - ZGM

(5), (5), Now a = 2(() + (()) b, = 1(5), (58),

20, ZGM THEN ..

OR h = b, (GM)

Ea (12.45) FOR DRBIT 1.. $21 = \frac{2\pi a_1 b_1}{b_1 \sqrt{GM}} = \frac{2\pi}{6M}$

(WNTINUED)

12.121 continued

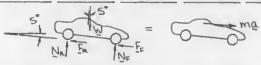
SIMILARLY, FOR THE ORBIT OF SATELLITE 2. I2 = 10 0 22

THEN ..

12.122

GIVEN: AUTOMOBILE OF WEIGHT 3000 15 MOVING DOWN A 5° INCLINE; N = 50 m; AT 1=0, FRANK = 1200 h

IS APPLIED FIND: X WHEN U=0



HAVE. - EF = ma: WSINS- (FE+FE) = Qa

WHERE FE+FR = FRANCE

THEN -- Q = (32.2 5t) (SINS - 1200 16) = -10 0736 5t

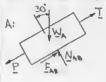
FOR THIS UNIFORMLY DECELERATED MOTION HAVE -N2 = N2 + SO(X-X)

WHERE NJ = 50 TH = 73.333 Th OR X= 267 St

12,123

GIVEN: Mx = 30 kg, Mx = 15 kg; M3=0.15, M4 = 0.10; 8.30, P. 250 N FIND: (a) QA (b) T

FIRST DETERMINE IF THE BLOCKS WILL MOVE FOR THE GIVEN VALUE OF P. THUS, SEEK THE VALUE OF P FOR WHICH THE BLOCKS ARE IN IMPENDING MOTION, WITH THE IMPENDING MOTION OF A DOWN THE INCLINE.



AEFy = 0: NAB - WA COS30 = 0 OR NAB = MAG COS 30 NOW .. FAB = MINAS = 0.15 m, q cos 30

== EFx = 0: T-P+FAR - WASIN30 = 0 OR T = P + mag (sin30 - 0.15cos30)

SUBSTITUTING. T= P+ (30 kg)(9.81 52)(511130-0.15cos30) =(P+108.919) N

T AEFy =0: No - NAS - Wa COS30 = 0 OR No = 9 cos 30 (mx+me) Now - FB = Ms NB

= 0.15 9 cos30 (MA+Ma) == = == == - WB SIN 30 = 0 OR T = mg sin30 + 0.15 magcos30 + 0.15g cos30 (Ma+ma)

12.123 continued

OR T= 9 [mg sin30 + 0.15 (2ma+mg) cos30"] = (9.81 m) [(15 kg) sin 30 + 015 (2-30+15) kg = cos 30] = 169.152 N

THEN -- 169.152 N = (P+108.919) N

OR P = 60.2 N FOR IMPENDING MOTION OF A DOWNWARD. SINCE P < 250 N, THE BLOCKS WILL MOVE, WITH A MOVING DOWNWARD. "NOW CONSIDER THE MOTION OF THE BLOCKS.

(a) A: P=25016

12 Fy = 0: NAS - WA COS30 = 0 OR NAG = MAG COS30 SUDING: FAB = MANNAS = 0.1 m, 9 cos30 2 EFx = ma: -T+P-FAB+WASIN30 = MAQA OR T = P + mag(sin30-01cos30)-mag

T = 250 N+ (30 kg){(9.81 32 X SIN 30-0.1 COS 30)-an] = (371.663-30a,) N

B:

12 Ey = 0: No - Nos - We cos30 = 0 OR No = 9 cus30 (MA+ MB)

SLIDING: Fo = MA No

= 0.1 9 cos 30 (m+ma)

== Fx = mg as: T - FAS - F8 - WB SIN30 = mg as OR T = Mag sin 30+ 0.1 mag cos30+0.19 cos30 (ma+ma)+maga = 9[m851N30+0.1(2mA+MB)cos30]+m8a8 = (9.81 3) ((15 kg) 51130 + 0.1(2=30+15) kg = cos 30) + (15kg)a.

> = (137.293 + 15as) N (5)

EQUATING THE EXPRESSIONS FOR T (Eas.(1) AND (2)] AND NOTING THAT QA = Q8 __

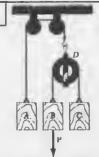
371.663 - 30 QA = 137.293 + 15 QA OR a = 5.2082 52

1. Qx = 5.21 ₹ 730

(b) SUBSTITUTING INTO EQ (1) .. T= (371.663-30: 5.2082) N

OR T = 215 N

12.124



GIVEN: WA . 20 16, Wa = We = 10 1b; AT 2=0, 5=0; AT t=25, ays = 841 FIND: (a) P (b) THO

FROM THE DIAGRAM .. CORD 1: YA+ YO = CONSTANT THEN .. UN+ US = 0 AND antanto coes 2: (40-40)+ (46-40) · CONSTANT THEN _ NB +NC - 2 NB = 0

AND an+ac-200=0 OR .. 20x+ aB+ ac =0 (1) NOW. HAVE UNIFORMLY ACCELERATED MOTION BEENIX

ALL OF THE FORCES ARE CONSTANT. THEN... 43 = (40) + (58) 2 + 200t2 AT t=25, Ays= Bft: Bft = 208 (25)2

+1EK = ORDAD: 2Tec -Tas=0 PULLEY D:

+1EFy=maa: Wa-Tab= qax BLOCK A: · IEFy = mea: We-Toc = qae BLOCK C: OR Ox = 9(1- 2/AD)

SUBSTITUTING THE EXPRESSIONS FOR Q AND Q INTO Eq. (1) - 29(1- TAD) + OB + 9(1- TAL)=0

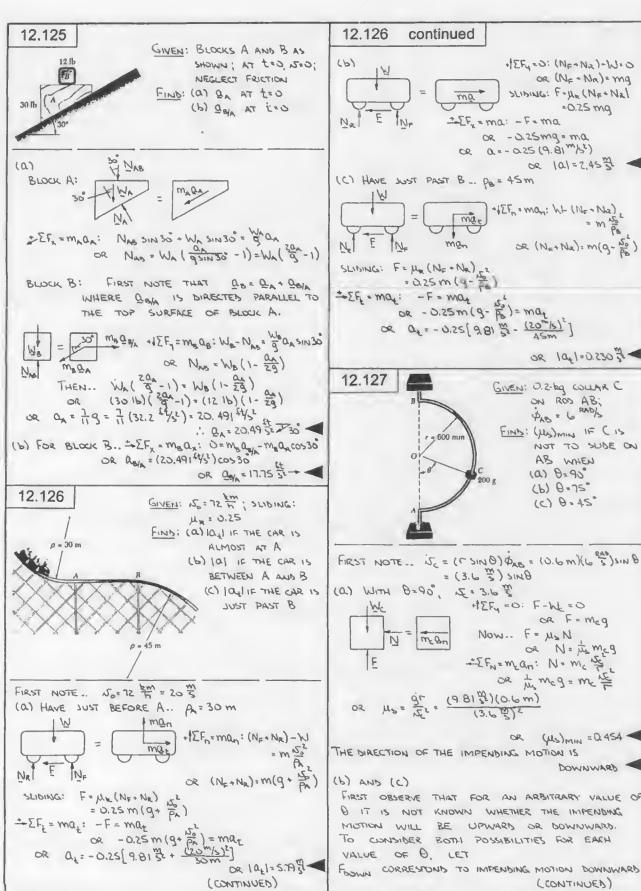
OR (Wh + 2Wc) This = 3+ 3 THEN .. (2016 2=1016) Tay = 3+ 32,2 A/3

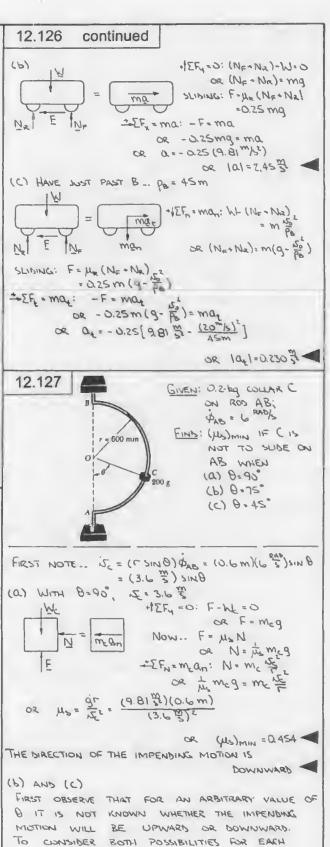
> OR Two 20.828 16 AND THEN TEC = 10.414 16

Tac +1 EFy=maa: P+Wa-Tac= Q= Qs BLOCK B: OR P=TBC+WB(Q=1) P= 10,414 1b+(101b)(4 4/5 -1) SUBSTITUTING --

OR P = 1.656 16

(b) HAVE FROM ABOVE .. TAN = 20.8 16





(CONTINUED)

12.127 continued

THEN, WITH THE TOP SIGN CORRESPONDING TO FORM, HAVE.



m_{eQn}

+ Σ Fy=0: Ncosθ ± F sinθ - Wc = 0 Now... F = μs N THEN Ncosθ = μs Nsinθ - meg = 0 α N = meg cosθ = μs sinθ

Bus E = Usmed

TER = mean: NSIND = FCOSD = MEP P= FSIND SUBSTITUTING FOR N AND F.meg COSD = Me Med COSD = Me Med FSIND

OR TAND + US = SEZ 12 MATERIA = GESTAND

DR US = + TANB - gr sinb TANB

Now - 22 = (13.6 3) SIN 8 = 2.2018 SIN 8

THEN .. HS = + TAMB - 2.2018 SIND BURT BURT BURS 8105.2+1

(b) 0=75° TANTS-2.2018 SINTS = ± 0.1796

THEN. DOWNWARD: US=+0.1796 UPWARD: US<0... NOT POSSIBLE

: (Us)min=0.1796

THE DIRECTION OF THE IMPENDING MOTION IS

DOWNWARD

(C) 8=45° 1+2.2018 SIN 45° = 2(-0.218)

THEN -- DOWNWARD: US CO -- NOT POSSIBLE UPWARD: US = 0.218

815.0 = mm(eq) :

THE DIRECTION OF THE IMPENDING MOTION IS

NOTE: WHEN TAN B - 2.2018 SIN B = 0

OR B = 62.988°

US = 0. THUS, FOR THIS VALUE OF B FRICTION IS

NOT NECESSARY TO PREVENT THE COLLAR FROM
SLIBING ON THE ROD.

12.128 B C

GIVEN: We = \$ 16, 6 = 20 IN.; WHEN 8 = 20, 8 = 15 \$6, 15

B WHEN 8=20°
(b) P AND Q WHEN
B=20°, WHERE P IS
DIE TO CC AND Q
IS DUE TO DE

KINEMATICS

FROM THE BRAWING OF THE SYSTEM HAVE -= 2600 0

BIBNIE 6 - 12 - 13 HT

AND $\ddot{r} = -2b(\ddot{\theta} \leq iN\theta + \dot{\theta}^2\cos\theta)$

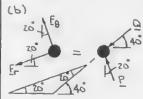
Now. ar= "- (b = - 2 b (0 sup + 0 cos 0) - (2 b cos 0) 62

- 28(85 mg+ 282056) (250 56) (12 56) 20 > 20)

= - 1694.56 3/2

= 2 (72 ft)[(250 = 20) cos20 - 2(15 = 20) sin 20]
- 270.05 145

(a) HAVE .. Fr = mar = \$\frac{1}{52.25\frac{1}{5}}\frac{1}{5}\frac{1}\frac{1}{5}\frac{1}{5}\frac{1}{5}\frac{1}{5}\frac{1}{5}\frac{1}



= 14.00 09 16 EF6: F6 = P-Q SINZO

AEF : F = P-Q SINZO OR P=(Z.O967 +14.0009 SINZO) 16

= 6.89 16 .'. P=6.89 16 & 70 Q=14.00 162740

12.129

GIVEN: CENTRAL FORCE F AND
PATH SHOWN;

T= [-](COSZB; AT 1=0,

I=0, F=0, T=0

FUNCTIONS OF B

HAVE .. T= COSZO

THEN. : = 25 SINZO O

Now. 5 = 12,+ + 000 so THAT AT 1=0... 5 = 50. FROM EQ. (12.27): 120 = 520.

OR B = (-12 - 1-15) ((0) 28) /2 = (5) COS 20

THEN $\dot{c} = \frac{2C_0 \sin 2\theta}{\cos^2 2\theta} \left(\frac{\kappa_0}{C_0} \cos^2 2\theta \right) \cdot \kappa_0 \cos^2 \theta$

12.129 continued

Now No . i

12.130

GIVEN: RASHUS F OF THE MOUN'S OXBIT;

RADIUS R OF THE EARTH; THE

ACCELERATION OF GRAVITY 9 AT

THE EARTH'S SURFACE; THE

PERIODIC TIME TO OF THE MOON

SHOW: F= f(R, 9, T)

FIND: F KNOWING THAT T= 27.3 BAYS

HAVE -- $F = G \frac{Mm}{r^2}$ [EQ. (12.28)] AND $F = F_n = ma_n = m \frac{s^2}{r}$ THEN $G \frac{Mm}{r^2} = m \frac{s^2}{r}$



OR N= T NOW GM= 9R2 EQ. (12.30) SO THAT N2 = 9R2 OR N= R (5)

FOR ONE ORBIT. $T = \frac{2\pi \Gamma}{N} = \frac{2\pi \Gamma}{R \sqrt{\frac{2}{R}}}$ OR $\Gamma = \left(\frac{9T^2R^2}{4T^2}\right)^{1/3}$ Q.E.D.

Now .. T= 27.3 Days = 2.35872 = 0 \$

R = 3960 mi = 20.9088 × 10 \$ \$\$

 $\underline{SI}: \qquad \Gamma = \left[\frac{9.81}{5} \frac{m}{5} \cdot (2.35872 \times 10^5 \text{ s})^2 \cdot (6.37 \times 10^5 \text{ m})^2}{4\pi^2} \right]^{1/3}$ $= 382.81 \times 10^5 \text{ m}$

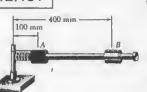
OR T= 383×10 km

U.S. CUSTOMARY UNITS:

T= [32.2 \$ 10 (2.358)2x10 5) 2x(20.9088x10 ft)2] 1/3
= 1256, 52=10 (t

OR (= 238 x10 mi

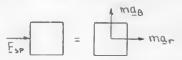
12.131



GIVEN: M=0.25 kg, k=6 m, (Lo)+=0.5 m; AT t=0, Bo=16 & COLLAR IS AT A; NEGLECT ERICTION AND MROS

(b) (QB), AND (QB)B (c) (QCOUNRIROD)B

FIRST NOTE.. FSP = k[(Lo)SP-T] AT B: (FSP)B = 6 m (0.5-0.4) m = 0.6 N



(a) AFTER THE CORD IS CUT, THE ONLY HURIZONTAL FORCE ACTING ON THE CULLAR IS DUE TO THE (CONTINUED)

12.131 continued

SPRING. THUS, ANGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED. ... To m (No.) = & m (No.) & WHERE (No.) = To m (No.) & WHERE (No.) & WHERE (No.) = To m (No.) & WHERE (NO.

THEN - (NE) = 0.1 m [(0.1 m)(16 & 5)]

(b) HAVE $F_0 = 0$... $(a_{2}b_{0} = 0)$ Now. $-2F_{0} = ma_{0}$: $(F_{20})_{0} = m(a_{0})_{0}$ or $(a_{0})_{0} = \frac{0.6 \, \text{N}}{0.25 \, \text{kg}}$

(C) HAVE. Qr= "- TB?

NOW. - Qwinelros = " AND 80 = (156)

THEN. IT B: (QUILLURADO) = 2.40 5 + (0.4m) (0.400 3) 2

OR (QUILLURADO) = 2.80 52

12.132

GIVEN: TRAJECTORY OF THE VOYAGER I SPACECRAFT ABOUT SATURY; AT THE FOILT OF CLOSEST APPROACH,

T= 185×10 km, N= 21.0 km/s; FOR

THE CIRCULAR DRAFT OF THE MOON

TETHYS, T= 295×10 km,

N=11.35×10 km/s

FIND: E AT THE POINT OF CLOSEST APPROACH OF VOYAGER I

FOR A CIRCULAR ORBIT

S= VGM EQ. (12.44)



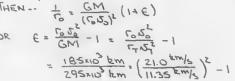
FOR THE DRBIT OF TETHYS ...

FOR NOVAGER'S TRAJECTORY HAVE..

T = GM (1+ E COSB)

WHERE h = G.S.

WHERE h= 65 AT D, r= 6, 8=0 THEN -- to = GM (65)1 (146)



OR E = 1.147



150 mi

40 mi

GIVEN: ELLIPTIC AND CIRCULAR ORBITS OF THE SHUTTLE COLUMBIA ABOUT THE EARTH

FIND: (a) tAB
(b) Terac

FIRST NOTE.. R = 3960 mi = 20.9088 = 10° St

(3960 + 40) mi = 4000 mi = 21.120 = 10° St

(3960 + 150) mi = 4110 mi = 21.7008 = 10° St

(a) THE PERIODIC TIME T OF AN ELLIPTIC ORBIT IS

(EQ. (12.451)

: $t_{AB} = \frac{1}{2}I = \frac{\eta ab}{h_{AB}}$ WHERE $Q = \frac{1}{2}(A + f_B) = \frac{1}{2}(21.120 \times 10^{-4} \times 1.7008 \times 10^{-6})$ ft $b = (\frac{1}{12} + \frac{1}{12} + \frac{1}{12} \times 1.7008 \times 10^{-6})$ ft $b = (\frac{1}{12} + \frac{1}{12} \times 1.120 \times 10^{-6})$ ft $= 21.4084 \times 10^{-6}$ ft

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC ORBIT..

1/A + 1/B = 2GM

Now .. GM = gR2 [EQ. (12.30)]

SO THAT $h_{AB} = \left(\frac{29R^2}{r_A^2 + r_B^2}\right)^{1/2} = \left[\frac{2(32.2\frac{c_L}{32})(20.9080 \times 10^5 ft)^{3/2}}{21.120 \times 10^5 ft} + \frac{1}{21.1008 \times 10^6 ft}\right]^{1/2} 8: W_{8/30}$ $= 548.95 \times 10^9 \frac{ft^2}{5}$

FINALLY -- tab = TT(21.4104x10 ft)(21.4084x10 ft)

= 2623,25

OR tas = 43 MIN 435

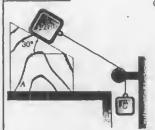
(b) FOR THE CIRCULAR ORBIT

WHERE NEIRC = $\left[\frac{9R^2}{\Gamma_8}\right]$ [EQ.(12.44)]

THEN .- TCIRC = 271 15 12 271 (21.7008=10 ft) 3/2 (20.908) 210 ft) (32.2 ft/s) 1/2 = 5353.55

OR TGIRC = 1 h 29 MIN 135

12.C1



GIVEN: MA = 20 kg, MB = 10 kg, MC = 2 kg; 2=0, J. 0; M = 0

FIND: QA AND QBIA FOR

MED USING QUEDOIL

WHILE QA TO AND

QUE OIL WHILE QBIATO

ANALYSIS KINEMIATIKS

HAVE .. Qg = Qa + QB/A

Q BYA

WHERE QUA IS DIRECTED ALONG THE INCLINED SURFACE OF A. THEN.

QB = QA (- COS 30]' - SIN 30 j') + QBA j'

ALSO, SINCE THE CORD IS OF

CONSTRUT LENGTH $\alpha_{c} = (\alpha_{B})_{X'}$ $= (\alpha_{B})_{X'} = (\alpha_{B$

EUNETICS

* (2 - mc αc: T- Wc α-mc αc

OR T = mc (g- αc)

" mc (g- ασ/k+ ακ cos30)

-- (2)

 $\frac{3}{1} = \frac{30}{m_{\text{B}} \alpha_{\text{A}}} = \frac{30}{m_{\text{B}} \alpha_{\text{B}} \alpha_{\text{A}}}$

Ne ZFx, = maax: T-Fx+ Wa SIN30 = maax-max cos30

OR T-Fx+109 SIN30 = 10 aax - 100x cos30 (3)

+/EFy1 = MBay1: NAB - WB COS30 = - MBOL SIN30
OR NAB = 109 COS30-100L SIN30 (4)

SLIDING: FAB = M NAB

OR FAS = 10M (900530-0,51030) (5)
SUBSTITUTING EQS. (2) AND (5) INTO EQ. (3)...

2 (9-00/A+0,00030)-10/4 (9 c0530-0, 51130)+10951130

OR 9 (1-54 cos30+5 51N30)

= 608/A-OL (5/45/430+600530) (6)

NOTE: BLOCK A WILL NOT MOVE (Q. = 0) BEFORE
BLOCKS B AND (WILL NOT MOVE (Q. = Q. = 0).
THEREFORE, THE SYSTEM WILL REMAIN AT
REST WHEN

g(1-5/2 cos30+5 sin30)=0

OR 120.808 FOR NO MOTION

A: FAD NAB = MAQA

12.C1 continued

* ΣΕη = 0: NA - NAS COSSO - FAS SINSO - WA = 0

OR NA = NAS (COSSO - μ SINSO) + 209 (8)

SLIDING: FA = μNA

OR FA = MNAS (0530+MSIN30)+20Mg (9)

SUBSTITUTING EQ. (9) INTO EQ. (7).

NAS (51430- LOSSO) - INAS (05800+ MS1030) - 20Mg

OR NAB (11-µ2) SIN30-2µCD330] - 20µq = 200A SUBSTITUTING FOR NAS [EQ. (41]... (109 CD330-100A SIN30) [(1-µ2) SIN30-2µCD530]

- 20mg = 200m

LET A= (1-12) SIN30 - 2110530

THEN .. 9 (A COSSO - ZLL) = (2+ A SINSO) OA

OR Q = A cos30 - 2 M q (10)

NOTE: BLOCK A WILL REMIND AT REST WHEN

g(Acos30-2µ)=0

OR [(1-m²) sin 30 - 2 m cos 30] cos 30 - 3 m = 0

OR JUZ 0.12188 FOR BLOUK A TO REMAIN

AT REST

NOW. REWRITE EQ. (6) AS ONE & [9 (1- SM COS 30 + 5 SIN 30")

+ an(SMSIN30+60530)) (11)

WHICH REDUCES TO

agy = 310 (1-5400230+521430) (15)

WHEN ON = 0

OUTLINE OF PROGRAM

INPUT INITIAL VALUE OF M: M=0

(DMPUTE A: A= (1-M2) 51N30-2MCOS30

COMPUTE QA: QA = A COS30-2M 9

WHILE ax >0

COMPLETE ARA:

Oela = 6 (9(1-54 cos30+5 sin30)

PRINT THE VALUES OF M. A., AND DAIL

UPDATE M: M= M+0.01

INCREASE IL TO THE NEXT TENTH:

M= 10 [INTEGER VALUE (10,11)] + 0.1

COMPUTE OBJA: 9 (1-5/4 COS30 + 5 SIN 30)

WHILE aby >0

PRINT THE VALUES OF IL AND QUAN UPDATE IL: IL= IL+ O.1

12.C1 continued

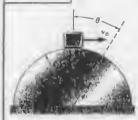
PROGRAMI OUTPUT

28	accel. of A, m/s'	accel. of B wrt A, m/s'
0.00	1,888	7.358
0.01	1.742	7.167
0.02	1.594	6.975
0.03	1.445	6.780
0.04	1.295	6.582
0.05	1.143	6.382
0.06	0.989	6.179
0.07	0.833	5.973
0.08	0.676	5.764
0.09	0.518	5.553
0.10	0.357	5.339
0.11	0.195	5.122
0.12	0.031	4.901

For those values of μ for which the wedge is at rest

μ	accel.	of B wrt	Α,	m/s*
0.20		4.307		
0.30		3.599		
0.40		2.891		
0.50		2,183		
0.60		1.475		
0.70		0.767		
0.80		0.059		

12.C2



GIVEN : W= 11b, N=105;

0= 11b, N=105;

FIND: 8 AT WHICH THE BLOCK

LEAVES THE SURFACE;

11=0,005,010,...,0.4

ANALYSIS

=

D-max

VZFn=man: Wω=0-N=mp or N=m(gco=0-' slibinG: F=μN

* EFE = maz: WSIND-F = maz

Now. a= dt .. dr = g(sINB-HE COSB)+HE & (1)

ALSO.. $S = r\theta$ or $d\theta = r\theta$ (2)

Thus, differential equations (1) and (2)

DEFINE THE MOTION OF THE BLOCK.

As the block leaves the surface, $N \rightarrow 0$.

Thus, $q\cos\theta - r\theta = 0$ DEFINES THE VALUE OF θ at which the blocks leaves the surface.

OUTLINE OF PROGRAM

FOR EACH VALUE OF ILL

DEFINE THE INITIAL VALUES OF IT AND B

USE THE MODERED EVER METHOD (SEE THE

SOLUTION TO PROBLEM 11.C3) WITH A STEP

(CONTINUED)

(CONTINUED)

12.C2 continued

SIZE At = 0.01 & TO NUMERICALLY INTEGRATE THE EQUATIONS

みころ(いいの・かんのとり)ナルが

WHERE P = 5 St. COMPLTE N, AND No: N, = cost, - gp N2 : CO : 02 -

WHERE O, AND U, ARE THE VALUES OF O AND THE VELOCITY, RESPECTIVELY, AT THE BECHNING OF A TIME INTERVAL, AND 02 AND NI HRE THE VALUES AT THE END OF THE TIME INTERVAL.

IF N2 > 0, UPDATE IT AND 8: IT = IZ; B = B2 IF N2 40, USE LINEAR INTERPOLATION TO DETERMINE THE VALUE OF 8 AT WHICH $\theta = \theta_1 + \frac{O - N_1}{N_2 - N_1} (\theta_2 - \theta_1)$

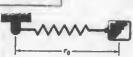
PRINT THE VALUES OF IL AND B

PROGRAM OUTPUT 0.00 0.05 0.10 29.11* 29.61 30.16

0.20 31.33 31.969

0.30 0.35 33.35*

12.C3



GIVEN: BLOCK OF MASS IN AND SPRING OF CONSTANT &: THE CHA COLU, O= + TA SPRING IS HORIZONTAL AND UNSTRETCHED

FIND: (a) TAND & WHEN THE BLOCK PASSES UNDER

> THE PIVOT O (b) Rym WHEN 5=1 m SO THAT J- WHEN THE BLOCK PASSES UNDER O

FIRST NOTE .. F = 1 X2+42 COS0 = 3 SIND - F Fap= k(1-10)

== EFx = max: - Fsp cos8 = max DR ax=- Por (- To) cos B 9/2 = - 1/2 (1/2 and 5 - 12) (1)

+ 1 ERy = may: W-FSF SIND = may OR ay = 9 - m(1-16) SIND OR del = 9 - 1 (1x2+45 - C) (1x+45) (5) (CONTINUES)

12.C3 continued

dy = 15, (4) 能。 ~ (3) ALSO .. THEREFORE, DIFFERENTIAL ENATIONS (1)-(4) DEFINE THE MOTION OF THE MASS. Now .. Is = (I Las I LAN By = TAN IT DEFINE THE MACHITUDE AND DIRECTION, RESPECTIVELY OF THE VELOCITY.

OUTLINE OF PROGRAM

INPUT VALUE OF R/m INPUT UNSTRETCHED LENGTH OF THE SPRING FO INPUT SYSTEM OF UNITS

DEFINE THE INITIAL CONDITIONS:

X1=5, 41=0; (Jx),=0, (Jy),=0 USE THE MUDIFIED EULER METHOD (SEE THE SOLUTION TO PROBLEM 11. (3) WITH A STEP SIZE At = 0.001 5 TO NUMERICALLY INTEGRATE THE EQUATIONS

OF = - 1 (1x=+1=-1)(1x==1) TE = 3- 1/2 (1x5415-12)(1x5415) gx . 2x dy . 54

WHEN X, 70 AND X240 COMPUTE GAND G: G= VX2442 G= VX2442 COMPLE S, MD BG: 5,= {(5) }+ (5); (BS) = TAN (5%)

COMPUTE No AND BUZ: 5, 0 (WX) 2 + Wy)2 (Bu) = TAN" (Wy)=

WHERE (), AND () DENNITE VALUES AT THE BEGINNING AND END, RESPECTIVELY, OF A TIME INTERVAL.

USE LINEAR INTERPOLATION TO DETERMINE THE VALUES OF 1, 5, AND BY AT X=0:

L= L1 + O-X1 (L5-L) $Q^2 = (Q^2)^i + \frac{\chi^{F-\chi^i}}{O-\chi^i} \left((Q^2 - \chi^i) \right)$ $V_i = \chi^i + \frac{\chi^{F-\chi^i}}{O-\chi^i} \left((Q^2 - \chi^i) \right)$

PRINT THE VALUES OF PM, TO, T, J, AND BU

PROGRAM OUTRUT

 $k/m = 15.00 /e^{\circ}$ Unstretched length of the spring = 1 m

X2 = ~.002 m X1 = 0.001 mr = 2.765 mv = 2.740 m/s

Angle v forms with the horizontal = -6.19*

k/m = 20.00 /s2 Unstretched length of the spring - 1 m

X1 = 0.001 mx2 = -.002 mr = 2.372 mv = 2.983 m/s

Angle v forms with the horizontal = 0.93*

(CONTINUED)

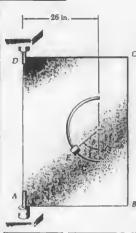
12.C3 continued

 $k/m = 25.00 /s^4$ Unstretched length of the spring = 1 m

(b) $k/m = 19.11 / m^2$ Unstretched length of the spring = 1 m

Angle v forms with the horizontal = -0.00°

12.C4



GIVEN: MS = 0.55; PABED = 14 TS,

2 TS; FONT = 10 IN.;

WE = 0.8 ID

FIND: RANGE OF VALUES OF B

FOR WHICH THE BLOCK

BOES NOT SLIDE

ANALYSIS

FIRST NOTE.. $\rho = iz(26-10 \sin \theta) it = iz(13-5 \sin \theta) it$ THEN $Q_n = \frac{1}{\rho} = \rho \dot{\theta}^2 = \frac{1}{c}(13-5 \sin \theta) \dot{\phi}^2$ ($\frac{it}{3}$)

NOW CONSIDER THE FOLLOWING FOUR CASES.

CASE 1: THE BLOCK IS RESTING ON THE INNER

SURFACE OF THE SLOT; DOWNWARD MIOTION

15 IMPENDING ($0 \le \theta \le 90$)

HAVE. - F=μ, N 2 THEU. W(sinθ + ^{ΔE}/₉ cosθ) = μ, · W(cosθ - ^{ΔE}/₉ sinθ) OR [6qsinθ + φ(12 - 5 sinθ)cosθ]

| θεω(θης 2-51) | θεω(θης 6-61) | θεω(θης 6-61) | θης 6-61) | θης

CASE 2: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLOT; DOWNWARD MOTION IS IMPENDING (050590)

1 ΕΕς=0: FSINB-NOSB-W=0

- ΣΕς=man: FcosB+NSINB= W JE

THEN... F=W(SINB+ JE cosB)

AND .. N=W(-cos0+ 35 SIND)

HAVE _ F= MSN

(CONTINUED)

12.C4 continued

THEN. W(SIND . TE COSD) - MS KW(-COSD + JE SIND)

OR [69 SIND + \$2 (13-5 SIND) COSD]

= 0.35[- $\omega_{\text{GCD}}\theta + \hat{\phi}^{\text{E}}(13-5\sin\theta)\sin\theta$] (2)

CASE 5: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLUT; DOWNWINDS MOTION IS IMPENDING (90°+0+180°)

| IF = 0: Fcosa + N sinα · W = 0 | IF = man: - F sinα · Ncosα = & JE THEN ... F = W(cosα - JE sinα) | F AND . N = W(sinα · SE cosa)

HAVE - F-μ. N (SE SINα)=μ. W(SINα+ 3F COSα)
NOW α=θ-90

SUBUTITUTING.. [COS(B-90") - 25 SIN(B-90")]

OR (SIND+ 25 COSB) - M. (COSB - 25 SIN B)

WHICH IS IDENTICAL TO THE DEFINING EWATION OF CASE ?

CASE 4: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLOT; UPWARD MOTION IS IMPENDING (90°5 0 : 180°)

#ΣΕς=0:- FWSA + NSINA - WE O + ΣΕς= man: FSINA + NCOSA : M JE THEN. F= W(-WSA + JE SINA) AND. N: W(SINA + JE COSA)

HAVE.. F = M.S.N

THEN.. W(-COSM + OF SINK) = M. SINK + OF COSK)

NOW &= 0-90°

SUBSTITUTING.. [-COXO 90) + OF SIN(0-90)]

OR (-SIND-OF COSD) = M. (-COSD OF SIND)

WHICH IS THE SAME AS THE DEFINING EQUATION

OF CASE | AFTER MULTIPLYING BOTH SIDES OF

THE EQUATION BY -1.

It is next necessary to solve Eqs. (1) and (2) for θ . Each of these evaluations can be expressed as $f(\theta)$, and then the values of θ for which $f(\theta) = 0$ can be determined. Substituting for $g(32.2^{\frac{1}{4}}/5^{\frac{1}{4}})$ and then simplifying, find... $f_1(\theta) = (67.62-13\dot{\phi}^2)(05\theta-(4.55\dot{\phi}^2+193.2)\sin\theta$ $f_2(\theta) = -(67.62+13\dot{\phi}^2)(05\theta+(4.55\dot{\phi}^2-193.2)\sin\theta$

 $f_2(\theta) = -(67.62 + 13\dot{\phi}^2)\cos\theta + (4.55\dot{\phi}^2 - 193.2)\sin\theta$ -1.75\dot\dot^2\sin^2\theta + 2.5\dot\dot^2\sin2\theta

NOTE: FOR THOSE VALUES OF B FOR WHICH THE BLOCK
15 AT REST WITH RESPECT TO THE PLATE,

FMAX = \(\mu_N \) \(\text{Z} \) \(\text{F} \)

WHERE N AND F ARE GIVEN ABOVE FOR EACH OF THE CASES. ALSO, f(0) = FMAX - F

(CONTINUES)

12.C4 continued

OUTLINE OF PROGRAM

INPUT VALUE OF &

CONSIDER CASES | AND 4

FOR VALUES OF B FROM O TO 179 IN

INCREMENTS OF 1"

COMPUTE f, (8): f. (0) = (67.62-130) cos0 - (4.550+193.2) sin0

41.75 \$2 SIN2 0 + 2.5 \$2 SIN20

COMPLTE f. (0+1°)

COMPUTE 4,181 + 1,18+1°) TO DETERMINE

IF A ROOT LIES BETWEEN 8 MIS

(0149)

IF 1, (0) + 1, (0+1") = 0, SOLVE 1, (0) FOR B USING NEWTON'S METHOD (SEE THE SOLUTION TO PROBLEM ILC4) FRINT THE VALUE OF BROOT AND WHETHER FMAX - F AT B IS

2 OR 50

CONSIDER CASES 2 AND 3

FOR VALUES OF B FROM O TO 179 IN

INCREMENTS OF 1°

COMPUTE f, (0): f, (B) = - (67.62+13\$ 2)COSB+ (4.55\$ -193.2) SINB

05 MIS & S+ 8 SMIS & SIN 20

COMPUTE & (8+1°)

COMPUTE f2(B) + f2(B+1)

IF \$ (8) . \$ (8+1°) & O, SOLVE \$ 2(8) FOR B USING NEWTON'S METHOS PRINT THE VALUE OF BROOT AND WHETHER FMAX - F AT B 13

2 OR & 0

PROGRAM OUTPUT

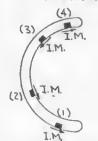
(Q) Rate of rotation = 2 rad/s At θ = 4°, F(max) - F >= 0 $\theta(1)$ = 4.68°

At $\theta = 140^{\circ}$, F(max) - F = < 0 $\theta(3) = 140.57$

(b) Rate of rotation = 14 rad/s At 0 = 115°, F(max) - F >= 0 0(4) = 115.91

At $\theta = 77^{\circ}$, F(max) - F = < 0 $\theta(2) = 77.63^{\circ}$

NOTE: IN THE ABOVE OUTPUT, THE I'M B(1) DENOTES THE CASE FOR WHICH MOTION IS IMPENDING.



12.C5

GIVEN: TWO POINTS ON THE TRAJECTORY OF A SPACECRAFT: B, AND B, OR T. AND THE RADIAL DISTANCE TO AND THE VELOCITY AT THE APOCEE OR

THE PERICE

FIND: TIME I FOR THE SAKECRAFT TO TRAVEL BETWEEN THE POINTS

. (a) B AND C OF PROB. 12.115: 28 = 8684 3

(b) A AND B OF PROB. 12.117;

ANALYSIS

= GM (1+ E cosθ) [Ea. (12.39')] HAVE --

WHERE H= TAPUCKE NAPORER " FRENCHE NPERCEE " TO WAS PAROGEE = 180 PREPAREE = D

GM = G (MEARM) MENTIN " (MENTIN) 9 REARTH

(Meran) 9R2 (1+Ecoson)

E = coster

THUS THE ECCENTRICITY OF THE TRAJECTORY CAN BE DETERMINED.

FROM PAGE 698 OF THE TEXT HAVE ..

= }h

WHERE h is a constant. THEN..

WHERE dA = 2(1)(100)

OR AA = 2 - 2 AB AND L= FETZAB

WHERE (IS GIVEN BY EQ. (12.39').

OUTLINE OF PROGRAM

SET VALUE OF AB: AB = 0.05

INPUT UNITS AND CONSTANTS

THE THE NWORN ARE CALLER PHETHIN TORNI

ARUGEE OR THE PERIGEE

SET VALUE OF PAP: PAP : O (PERIGEE) BAP = 180 (APOGEE)

TUPUT THE DISTANCE TAP TO AND THE VELOCITY IN AT THE ARGEE OR THE PERIGEE

THRUT THE VALUE OF B, FOR THE FIRST POINT

ON THE TRAJECTORY INPUT WHETHER THE SECOND POINT ON THE TRAJECTORY IS DEFINED BY THE VALUE OF BZ (CASE !) OR BY THE VALUE OF THE

RADIAL DISTANCE (CANE 2)

INPUT M/MEARTH

COMPUTE THE ECCENTRICITY E OF THE

TRAJECTORY:

(COMMUED)

12.C5 continued

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CASE 1:
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INPUT THE VALUE OF θ_2 IF $\theta_2 < \theta_1$, SET $\Delta\theta = -\Delta\theta$ FOR VALUES OF θ FROM θ_1 TO $\theta_2 - \Delta\theta$ IN
INCREMENTS OF $\Delta\theta$ UPDATE LARGA A: $A = A + \frac{1\Delta\theta}{2} \left[\frac{(m_{\text{EMEM}})}{(\Gamma_{\text{AP}} J_{\text{AP}})^2} (1 + \varepsilon \cos\theta) \right]^{-2}$ COMPUTE TIME $t : t = \frac{2\Delta}{\Gamma_{\text{AP}} J_{\text{AP}}}$

PRINT THE VALUES OF TAP, JAP, θ_1 , θ_2 , and t (ase 2:

INPUT THE VALUE OF T

INPUT THE VALUE OF TZ

SET THE INITIAL VALUE OF $\theta: \theta = \theta_1$ WHILE $\Gamma > \Gamma_2$ IF $\Gamma_1 > \Gamma_2$ OR WHILE $\Gamma < \Gamma_2$ IF $\Gamma_1 < \Gamma_2$ COMPUTE $\Gamma: \Gamma = \left[\frac{(M_{ENDTH})^2 R_{ENDTH}}{(\Gamma_{AP} \cup \Gamma_{AP})^2}, (1 + \varepsilon \cos \theta)\right]^2$

UPDATE AREA A: $A = A + \frac{1}{2}r^2 \Delta \theta$ UPDATE $\theta: \theta = \theta + \Delta \theta$ COMPUTE TIME $t: t = \frac{2A}{r_{AP}r_{AP}}$ PRINT THE VALUES OF r_{AP} , r_{AP} , r_{AP} , r_{AP} , and t

PROGRAM OUTPUT

(O)
The radial distance to and the velocity at the apogee are, respectively, 3600 km and .8694 km/s
01 = 100° 02 = 290°
Time t = 1 h 10 min 29 s

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The radial distance to and the velocity at the apogee are, respectively, 4310 mi and 24371 ft/s 01 = 180° r2 = 4035 mi Time t = 0 h 33 min 30 s

(6) TAND HONTHE MOON 13.1 continued GIVEN: HASS OF SATELLITE, M= 1500 kg 13.4 SPEEL OF SATELLITE, N= 22,9x103/cm/h HTSAS BATHO (4) PIND: KINETIC ENERGY, T T= 1 m ~= 1 (4kg)(25m/s)= 1250 N.m T= 1250 J N= 22.9 × 10 2 m/4 = 6,36 × 10 m/s W=mg=(4 &g)(9.81 m/52)= 39.240 N T= 1 m 02 = 1 (1500 kg) (6.36×103m/s)2 Ti+U1-2= T2 T,=0 U1-2=Wh T2=39,240 N T= 30.337 x 10 9 N.M h=T2 = (1250 N.M) = 31.855 M NOTE: ACCRLERATION OF STAULTY (39.240 N) HAS NO EFFECT ON THE MASS T= 30.3 GJ h= 31.9 M OF THE SATELLITE. (b) OUTHE HOOD HASS IS UNCHANGED, M= 4 LQ 13.2 GIVEN: WEIGHT OF SATELLITE, W= \$70 16 SPEED OF SATERLITE, V = 12,500 mi/h THUS T IS UNCHANGED FIND: KINGTE ENGREY, T WEIGHT ON THE HOON IS, WIM = MAM=(4 kg)(L62 m/62) Wm = 6.48 N N= (12,500 mi/h) (h/3600 5)(5280 ft/mi) h= T = (1250 wim) = 192,9 m Wm (6.08 W) N= 18,333 ft/5 MASS OF SATELLITE = (87016)(32.2 ft/s2) h=192.9m m = 27.019 16.52/f+ T= 1 mv= 1 (27,09) (18,350) 13.5 GIVEN: DISTANCE d= 120 m T = 4.5405 x109 16.54 US= 0.75, NO SLIPPING NOTE: ACCELERATION OF GRAVITY HAS 60 % OF WEIGHT ON FRONT WHEELS NO EFFECT ON THE HASS OF THE 40% OF WEIGHT ON RETR WHEELS T= 4.54 × 10 16.81 SATELLITE FIND: MAXIMUM THEORETICAL SPEED AT 120 M STARTING FROM REST GIVEN : WEIGHT OF STONE WE 516 (a) FOR FRONT WHEEL DRIVE 13.3 VELOCITY OF STONE, V= 80 ft/s (b) FOR REAR WHEEL DRIVE ACCULERATION OF GRAVITY ON THE (a) FRONT WHEEL DRIVE SINCE GO TO OF WEIGHT IS DISTRIBUTED ON FRONT MOON, 8m= 5.31 ft/52 FIND: (a) KINSTIC SUECEY,T WHEELS, THE HAXIMUM FORCE TO HOVE THE CAR F= 45 N = (0.75) (0.6W) = 0.450 mg HEIGHT h, FROM WHICH STONE FOR 120 m U1-2= (0.450 mg)(120 m)= 54 mg WAS DROPPED (b) TAND h ON THE HOON T,= 0 T, + U1-2=T2 (a) OUTHE BARTH 0+54mg= 1 m 12 T= 1 mv= 1 (516 (80 ft/s) N2= (2)(549)=(108)(9.81 W/52) T= 496.89 lb.ft N2= 1059.5 T= 497 16. 9+ N2= 32.55 W/S T,+U, = T, T,= 0 U, -2= Wh=(516)(h), T2=497 16.4 Nz=-117.2 km/h Wh=T2 h= 516 = 99.4 ft (P) BENE MHEET PEINE n=99.4ft USE SAME SOLUTION AS FOR (A) EXCEPT THAT (b) ON THE HOOL 40% WEIGHT IS DISTRIBUTED ON REAR WHEELS F= 43 N= (0.75)(0.40W)= 0.3 mg MASS IS UNCHANGEL THUS T IS UNCHANGED T= 497 16-84 4 FOR 120 m U1-2= (0,3 mg) (120m)= 36mg WEIGHT ON THE HOON 13: Wm = mg = (516) (5.31 f1/32) T,= 0 T1+41-2= T2 0+36mg=1 mV2 Wm = 0.8245 16 h= Tz = (49716++) = 6027 ft N3= (2)(36)(9)=(72)(9.81M/52) = 706.32 N= 26.58 M/s hm= 603ft 4 N3= 95.7 km/h SIVEN: MASS OF STONE, M= 4 RQ NOTE: THE CAR IS TREATED AS A PARTICLE IN THIS 13.4 VELOCITY OF STONE, N = 25 m/s PROBLEM. THE WEIGHT DISTRIBUTION IS ASSUMED ACCOLURATION OF GENUITY TO BE THE SAME FOR STATIC AND BYNAMIC ON THE MOON, 9m= 1.62 m/32 CONDITIONS, COMPAGE WITH SAMPLE PROBLEM FIND: 16.1 WHERE THE VEHICLE IS TRENTED A & (a) KINETIC ENERGY T A RIGIO BODY. HEIGHT h, FROM WHICH THE STONE

WAS DROPPED

13.6



GINEN: 1320 H DEAG BACE TRACK, CAR STARTS FROM REST CARS FRONT WHEELS OFF THE GROUND FOR FIRST GOST WHEEL'S ROLL WITHOUT SLIPPING FOR REMAINING 1260 ft WITH 60 % OF WEIGHT ON REAR WHEELS 1/4 = 0.60 , 45= 0.85, NO AIR OR ROWING LESISTANCE FIND: (a) SPEED OF THE CAR AT END OF FIRST GOFT (b) HAXINUM THEORETICAL SPEED AT FINISH LINE

(a) EIRST GO ft: REAR WHEELS SHID TO GENERATE THE HAXIMUM FORCE. SINCE ALL THE WEIGHT IS ON THE REAR WHEELS THIS FORCE IS: F=4kN=(0.60)(W)

T=0 T2= 1 4 V2

U1-1=(F)(60ft)=(0.61W)(60)=36W T1+U1-2=T2 36か=子前から NG0= 48.15 ft/s

150=32.8 mi/h

(b) FOR 1320 IT REAR WHERE'S SKID FOR FIRST GOST AND ROLL WITH SCIOING IMPENDING FOR REMAINING 1260 ft WITH 60 % OF THE WEIGHT ON THE REAR (DEIVE) WHEELS, THE HAXIMUM FORCE GENERATED IS:

FIRST GUST 'E = (.6 YW) AS IN (a) BEHAINING 1240 H FZ = 40 N = (0.85)(0.60)(W) = 0.510 W T=0 T2== 1 101320

U1-2= (.6)(W)(60)+(510)(W)(260 = (36+642.6)W = 678.6W 0+678.6W=+ \ Nis20 NI 320 = 43702 V1320 = 209.05 ft/s

N1320 142.5 mulh

SEE NOTE FOR PROB 13.5 FOR DISCUSSION OF WEIGHT DISTRIBUTION

13.7



GIVEN: 1320 FT DRAG RACE TEACH, CAR STARTS FROM REST. CARS' FRONT WHEELS OFF THE GROUND AND REAR WHEELS SKID FOR FIRST GO ft SPEED AT END OF FIRST GOFT IS 36 milh. WHEELS BOLL WITH SLIPPING IMPENDING FOR REHAINING 1260 ft, WITH 75% OF THE WEIGHT ON REAR (DEIVE) WHEELS. 46=0.8045 NO AIR OF POLLING SESISTANCE

13.7 continued

FIND: SPEED OF CAR AT GNO OF RACE

FIRST 60 ft: DINCE ALL THE CARS WEIGHT IS ON THE REAR WHEELS WHICH SEID, THE FORCE HOUING THE CAR 13

F = 4 & N = (4 &) (W) Non= (36 milh)(88 ft/s)/60 milh) 150= 52.8 ft/s T,= 0 T2= 1 m 0 == 1(12)(52.8 ft/s)=(1393.9)(12)

> U1-2=(F)(60ft)=(44)(W)(60ft) T1 + U1-2= T2-0+ 60 4 KW = (1393.9)(W) $4\mu = \frac{(1393.9)}{(60)(32.2)} = 0.72149$

FOR 1320 ft FORCE HOUING THE CAR IS FOR FIRST GOST, F. = (41)(W) = (0.72149) W FUR REMAINING 1260 ST, WITH 75 % OF WEIGHT ON SEYS (DEIDE) MHREC'S AND INDENDING SADIN F2 = (4,5)(0.75)W 45 = 4,1(0.80)=(0.72149)10.80) F2= (0,90186)(75)W= 0.6764 45= 0.90186 TI= 0 T2= 3(4) (V1020)2

U1-2= (F.) (60 ft) + F2 (1760 ft) = (0.72 149)(W)(60ft)+(0.6764)(W)(1260ft) = 43.29 W + 852.3 W = 895.55 W Ti+ U1-2=T2 0+898.55W=== (W)(N, 10) N120= (29)(895.55)=(2)(322 (+/52) (895.55) N₁₃₂₀= 57,673 N₁₃₂₀= 240.2 ft/s
we 13.5 N₁₃₂₀= 163.7 mi/h SEE NOTE POR PROB 13.5

13.8

EIND:



GIVEN: 400 m DEAG RACE TRACK, CAR STARTS FROM REST FRONT WHERE'S OFF THE GROUND AND REAR WHEELS SEID FOR FIRST 20 M. WHEELS ROLL WITH SLIPPING IN PENDING FOR REMAINING 380 M, WITH 80% OF THE WEIGHT ON THE REAR DRIVE WHEELS PEAK SPEED AT BUD OF THE RACE = 270 2m/h 4x= 0.7543

> (a) COEFFICIENT OF STATIC FRICTION, US (b) SPEED ATTHE END OF THE FIRST 20 m

(a) FORCE HOUING THE CAR FOR THE FIRST 20 M, WITH ALL OF THE WEIGHT ON THE REAR DRIVE WHEELS AND THE WHEELS SKIDDING,

F1=4KN=4KW=(0.75)W5)mg 40-0.7545

FORCE HOUING THE CAR POR REHAINING 380 W WITH 80% DE THE WEIGHT ON THE REAL (DRIVE) WHEELS AND SCIPPING IMPENDING (CONTINUED)

13.8 continued

F2=43(0.80)(W)=43(0.80)(W)=43(.80)mg

 $T_1=0$ $U_{400}=(270 \frac{km}{h})(\frac{1000 m}{km})/(3600 \frac{s}{h})$ $U_{400}=75 m/s$ $T_2=\frac{1}{2} m U_{400}^2=\frac{1}{2} m (75)^2=2812.5 m$

 $U_{1-2} = F_1(20m) + F_2(380m)$ $U_{1-2} = (U_{1})(75) + (U_{1})(180) + (U_{1}$

(b) FOR FIRST 20 M $M_{k}=(0.75)(45)=0.6741$ $F_{1}=M_{k}N=(0.6741)(mg)$ $T_{1}=0$ $T_{2}=\frac{1}{2}mN_{20}^{20}$ (1.2=(0.6741)(mg)(20m)=13.481mg $0+(13.481)(mg)=\frac{1}{2}mN_{20}^{20}$ $N_{20}^{2}=(2)(13.481)(4.81)=264.5$

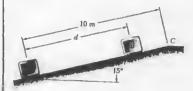
N20= 10.26 m/s

SEE NOTE FOR PIB.5

N20= 58.6 km/h

45= 0.899

13.9



GIVEN: UNTC=0

EIND:
(4) INITIAL T AT A
(b) T AS PACKAGE
RETURNS TO A

(a) 15° 12" F= 44N

UP THE PLANE, FROM A TOC, UE=0

Ta= \(\frac{1}{2} \text{ M N}_{A}^{A} \), Tc=0

U_{A-C}=(-W > 1N 15 - F)(10 m)

\(\text{TF=0} \) N-WCOS (5° = 0)

N= WCOSIS*

F= 4k N= 0.12 WCOSIS*

 $U_{A-C} = -W(sun 15°+0.12 cos 15°)(10 m)$ $T_{A}+U_{A-C}=T_{C} \qquad \frac{1}{2} \frac{w}{9} N_{A}^{2}-W(sin 15°+0.12 cos 15°)(10m)$

 $N_A^2 = (2)(9.51)(5inis^0 + 0.12cosis^0)(10hi)$ $N_A^2 = 73.5$ $N_A = 8.57 m/s /$

(b) DOWN THE PLANE FROM CTO A

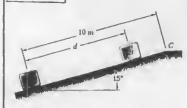
 $T_{c}=0$ $T_{A}=\frac{1}{2}MN_{A}^{2}$ $U_{c-A}=(WSINIS^{2}-F)10$ (FREVERSES DIRECTION) $T_{c}+U_{c-A}=T_{A}$ $O+W(SINIS^{2}-O.12cosis^{2}X10m)=\frac{1}{2}MN_{A}^{2}$

UA2 = (2)(9.81)(51N 150-0.12 cos 150)(10m)

 $N_A^2 = 28.039$

N= 5.30 m/s/

13.10



GIVEN: UAT A= 8 m/s /

FIND:

(a) DISTANCE d

PACKAGE HOUES UP

THE PLANE

(b) YELOCITY VA,

AS PACKAGE RETURNS

TO A.

(A) UP THE PLANE FROM A TO B

TA= 1 M NA = 1 4 (8 M/s) = 32 g TB=0

UA-B=(-WSIN 15°-F) d F= MM M = 0.12 N

EF= 0 M-wcos 15°= 0 N=wcos 15°

UA-B=-W(SIN 15+0.12 cos 15°) d=-wd(0.3747)

TA+UA-B=TB 32 g TB=0

d= (32)/(9.81)(.3747)

d= 8.70 M

(b) DOWN THE PLANE FROM B. TO A (F REUZESSES BIRECTION)

TA = \(\frac{1}{2} \) U_A^2 \) TB = O \(\delta = 8.72 \) M/s

UB-A = (WSIN IS^- F) d = W(SIN IS^- O. 12 COS IS^ \) \(\frac{8}{2},70 \) UB-A = 1.245 W

 $T_{B} + U_{B-A} = T_{A}$ $O + 1.245W = \frac{1}{2} \frac{W}{4} V_{A}^{2}$ $V_{A}^{2} = (2)(9.81)(1.245) = 253.9$ $V_{A} = 4.94 \text{ m/s}$

NA=4,94 W/S /

13.11



GIVEN; AT A U=No

FOR AB, 420,40 AT B, V= 8 ft/s

FIND; No

La Man

TR= 1 m No2 TR= 1 m (2 H6)

U_{A-0} = (w sin is - 4, N) (xoft) ZF=0 N - WCOSIS = 0 N = WCOSIS 0

TA+UA-B=TB

1 mv2-2551 mg= 32 M

Vo2=(2)(32+(2.551)(32.2ft/52))

vo= 228.29

U = 15.11ft/5

13.12



GIVEN; AT A, U=Vo ATB, V=O FOR AB, 41=040

FIND: U.

TA = 3 M NO TB = 0 Un-a= (WSINISO-ULN)(2091) EF# 0 N- COS 150 = 0 N= Wcos 150

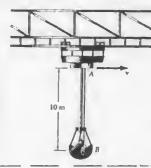
UA-B= W(SINIS - 0.40 cos 150) (20 ft) UA-8= - (2,551)(W) = -2,551 mg

TA+UA-B=TB 1 m No - 2.551 mg = 0

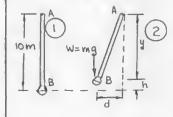
> No= (2)(2.551)(32.2 ft/52) N= 164.28

> > No= 12.82 ft/s DOWN TO THE LEFT

13.13



GIVEN CRANEMOURS AT VELOCITY, V AND STOPS SUBSTILLY BUCKET IS TO SWING NO HORE THAN 4 M HORIZONTALLY FIND: HAXIMUH ALLOWABLE VELOCITY N



U1-2= -mah

d= 4 M

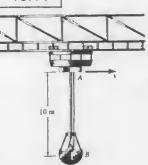
AB2 = (10m)2 = d2+ y2 = (4m)2+ y2 $y^2 = 100 - 16 = 84$ $y = \sqrt{84}$ $n = 10 - y = 10 - \sqrt{84} = 0.8349 \text{ m}$ $U_{1-2} = -m(9.81)(0.8349) = -0.8190 \text{ m}$ T1+U1-2=T2

1 mv2-0.8190m=0

 $v^2 = (2)(0.8190) = 16.38$

U= 4.05 M/S

13.14



GIVEN: CRANE MOVES AT VELOCITY U=3M/5 AND STOPS SUDDENLY FIND: MAXIMUM HORIZONTAL DISTANCE & MOVED BY THE BUCKET

d= 2.99 W.

REFER TO FREE BODY DIAGRAM IN P. 13.13 T= = M W= = = M (3m)= 4.5 M U,=N=3m/S T2=0 U1-2= - mgh T,+U1-2=T2 4.5m-mgh=0 = 0.4587 $\overrightarrow{AB}^2 = (10)^2 = d^2 + y^2 = d^2 + (10 - 0.4587)^2$ $100 = d^2 + 91.04$

13.15



d2= 8.96

GIVEN CAB WEIGHT, WE= 4000 16 TRAILER WEIGHT, WY= 12,000 16,2% GEADE 70% BRAKING FORCE SUPPLIED BY TRAILER 30% BRAKING FORCE SUPPLIED BY CAB

EIND:

(A) AVERAGE BRAZING FORCE TO SLOW DOWN FROM 65 mily TO 40 mily AS SHOWN

(b) AVERAGE FORCE BETWEEN CAB MID THERE

WI CAB-TRAILER SYSTEM



v= 65mc/h= 95.33ft/s Uz= 40 milh= 58.67 ft/s

TSIN= 2/100

Ti = = (MT+mc) 10,2= = (MT+mc) (95.33 ft/s)2 Ti = (4,544) (mit me) Tz = 2 (mr + me) (02) = 2 (mit me) (58,67 (1/6)2 Tz=(1721)(m++ Mc) T1 + U1-2 = T2 U1-2= - 1000 Fo + (W+1W2)(20 ft) 4544 (MAME) - 1000 FB+ (WrtWc) (20 ft) = 1721 (MIN)

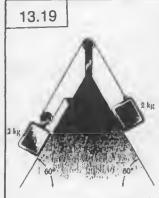
 $F_{B} = [(4544 - 1721)(\frac{16,000}{(32.2)} + (16,000)(20)] \frac{1}{(1000)} = 17227$

FB= 1723 16

(CONTINUED)

13.15 continued 13.17 GIVEN: 2000-kg CAB (b) TRAILER CONSIDERED SEPARATELY 8000- kg TRAILER LEVEL GROUND. - T= 1 (m+) (45,33)= 4544 m+ TRUCK COMES TO Fe T3 = 1 (MT) (58.67) = 1721 MT ASTOP IN 1200 M. 0.70FB 60% OF BRAKING 4544 MT - 1000 (Fct.70FB)+20W=1721MT T1+41-2=T2 FORCE FLOMTERILER From (a) Fo= 1722.7. 90% OFBRAKING 1000 F= (4544-1721)(12,000) - (700)(17227)+(20)(12,000) FORCE FROM CAB FIND: (4) AVERAGE BRAKING Fc= (1052) - 12059+ 240= 86.1 lb SEE NOTE FOR PISS F= 86.1 16(C) WI AUGRAGE FORCE IN THE COUPLING 13.16 TRAILER AND CAB ₩T v= (90 km/h)(1000 km) (1h 3600 s) (a) 65 mi/h 40 mi/h U1= 25 m/s 12=0 T1+U1-2=T2 T= 1 (m++ m2) (v,)=(1) (w,000 kg) (25 m/5) 1000 ft T1=3125×10 N-m T2= 0 GIVEN; CAB WEIGHT, WE= 4000 16 TRAILER WEIGHT, WT = 17,000 16 3125×10-(1200m)(FB)=0 FB= (125×103 N·m) = 2604 N·m 2 % UP GRADE FIND (0) AVERAGE FORCE ON THE WHEELS TO SPEED UP F (1200 m) (b) AVERAGE FORCE IN THE COUPLING FB= 2.60 KN (b) CAB CONSIDERED SEPERATELY CAB-TRAILER SYSTEM I No (0) N= 40 mi/n= 58.67ft/s T= 1 m= (v,)= (1000 X25 m/s)2 1515 2/100 Uz= 65 mi/h= 95.33 ft/s Ti= 625 X103 N.M , T2= 0 0.40 FB = 2609 U.M (FROM (A)) TI= 1 (m++ me) (v) = 1 (M++me) (58.67)=1721 (m+me) T,+U1->= T2 625×103-(0.40)(2604)(1200)+(Fc)(1200)=0 T2= 1 (m+mc)(v2)= 12 (m+mc) (95.33)= 45440m+mc) Fe=(0.40) (2604)-625=1042-521 T, 1 U1-2= T2 U1-2= (1000)(F)-(1000)(2/100)(W+1WL) SEE NOTE FOR PIB. 5 Fc= 521 N (c) 1721 (mytme) + 1000F-20 (WytWe) = 4544 (mytme) 1000 F = (4544-1721) (16,000) +20(16,000) 13.18 GIVEN: 2000 Ag CAB, 8000 kg TRAILER AVERAGE BRAKING FORCE BOOON 90 km/h F= 1403. + 320 = 1723. 1b LEVEL GROUND F=1723 lb FIND: (a) DISTANCE K. TO COHE TO (b) A STOP TRAILER COUSIDERED SEPARATELY (b) FORCE IN COUPLING FO (TRAILER BRAKES FAR) FO Ti= = (M+) (58.67)= 1721 MT T2= 2 (mi) (95.33) = 4544 mr WT (4) . 1721 MT + 1000 Fc - (1000 1/2) WT = 4544 MT U,= 25 M/s 1000 Fc = (4544 - (721) (17,000) + (20)(12000) T1= 3 (m11 mc) (25)=3125x103 J T2=0 U1-2= FBX F= 1052+240=129216 F == 1292 16(T) 3125×103 - (3000) X = 0 X= 1042 m SEE NOTE FOR P 13.5 (b) TRAILER CONSIDERED SEPARATELY WT T1= = Mx(25)2=(4000)(625) Ti = 2500 x103 J T2= 0 Ti+U1-2= T2 2500×102-(Fc)(x)=0 Feom (a) X= 1092 m 2500×100- Fc (1042) = 0 FC= 2500 x 103 = 2399.2 N 1042 Fc= 2,40kN(c) 4

SEE NOTE FOR P13.5

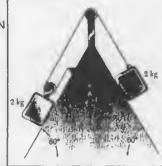


BLOCKS RELEASED FROM REST; NO FRICTION

(a) YELOCITY OF BLOCK B AFTER IT HAS HOUED

(b) TENSION IN THE CABLE.

13.20

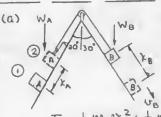


GIVEN:

BLOCKS RELEASED FROM REST; FRICTION 45= 0.30, 46=0.20

FIND:

(a) VELOCITY OF BLOCK B AFTER IT HAS MOVED ZM. (b) TENSION IN THE CABLE.



KINGHATICS X8=ZXA

Na= 2 Va

A AND B ASSURE B MOVES DOWN U,=0 T,=0

T2 = 1 MAVA2+ 1 MOVB= 1 (2kg) (NB+NB) T2= \$ VB

U1-2=-mag(co=30)(x1)+mBg(cos30) RB

KB=2M RA= IM

U1-2=(2)(981)(1)(-1+2)

U1-2 = 16.99 J

SINCE WORK IS POSITIVE BLOCK B DOES MOVE DOWN

T1+U1-2=T2

0+16.99= 5 VB

NB = 13.59

UB= 3.69 m/s THOIS SHT OT HWOOD

(b)

B ALONE

U,=0 T,=0 Uz= 3.69 m/s (FROM (a))

 $T_2 = \frac{1}{2} m_B v_2^2 = \frac{1}{2} (2) (3.69)^2 = 13.59 J$

U1-2 = (mgg)(cos 30°)(xB)-(T)(xB)

U1-2=[(2kg)(9,81m/52)(13)-(T)](2m)

U1-2= 33.98 -ZT

 $T_1 + U_{1-2} = T_2$ 0 + 33.98 - 2T = 13.59 2T= 33.98-13.59= 20.39

T= 10.19 N

(2) UB BRUILIBRIUM BLOCK B

CHECK AT O TO SEE IF BLOCKS MOVE, WITH HOTION IMPRUDING AT 8 DOWDWARD DETERHINE REQUIRED FRICTION FORCE AT A FOR

12 F= No-(MB91(SIN 30°)=0

NB=(29)(1)= 8

> = T-(M89) (CUS30°)+(FB)=0 (FB),=45 NB=(0,30)(g)

T= (2 g)(13/2)-(0.30)9

T= (13-0.30)(9)

BLOCK A

~ ZF= NA-(mag)(SIN 30°)=0

NA= (29)(1)=9

1

1 ZF= 2T-mag1(cos 38)-(FA) = 0

SUBSTITUTE T FROM D INTO D (2)

(FA)= (2)(13-0.30)(1)-139 LEG. FOR EQUIL (Fax = (13-0.60) g = 1.132 g MAX FRICTION THAT CAN BE DEVELOPED AT A= 45 NA = 0.39_

SINCE 0,39 < 1,1329 BLOCKS HOVE

(a) A LUD B

(FA) = 4k NB= (0.20) g (FA) = 4k NA= (0.20) g KINEHATICS X8=2XA UB= 2VA

T = 0 T = 1 MAUA+ 1 MO UB= (2) (2) (4)

U1-2=-MAQ(cos20)(2A)+MQQ(cos30) XB - (FA), (XA) - (FO), (XA)

XB=2 M. XA=1M

U1-2=[(2kg)(13/2)(1m)+(2kg)(13/2)(2m) - (0,20)(1M) - (0,20)(2M)][9.81 M/52]

U1-2=[(1732)-(0.6)][9.81] = 11.105 J

T1+U1-2=T2

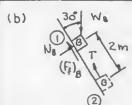
0+11.105=1,25Vg

Na = 8,88

UB= 2-98 m/s

(CONTINUED)

13.20 continued



B ALONE

 $N_1 = 0$ $T_1 = 0$ $V_2 = 2.98 \text{ m/s (FROM (a))}$ $T_2 = \frac{1}{2} M_B \cdot V_B^2 = (\frac{1}{2})(2)(2.48)^2$

 $N_0 = m_0 g \sin 30^\circ = g N$ $T_2 = 8.88 J$ $U_{1-2} = m_0 g (\cos 30^\circ)(2) - (T1(2) - (F_0)_{\xi}(2)$

 $U_{1-2} = (2lg)(9.81 \text{ m/s}^2)(\frac{5}{2})(2m) - 2T - (0.21(9 \text{ N})(2m))$

U1-2 = 213 9-2T-0.69

 $T_1 + U_{1-2} = T_2$ 0 +259 -2T-0,49 = 8.88 2T=(213-0,4)(9)-8.88= 21.179

T= 10.59 N

100 lb 20 lb

13.21

GIVEN:

SYSTEM AT REST WHEN 100 Ib FORCE 13
APPLIED TO NO FRICTION IGNORE PULLEYS HASS

FIND:

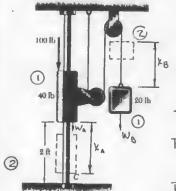
(A) VELOCITY, UN OF A

JUST BEFORE IT HITS C

(b) UN IF COUNTERWEIGHT

B IS REPLACED BY A

20-16 DOWNWARD FORCE



KINEMATICS

XB=2XA

VB=2VA

(a) BLOCKS A AND 8

T2= 1 MOV0+ 1 MAUA

T2=\frac{1}{2} \land 16 \land 32.2 \text{ft/s} \rangle (20) \land \frac{1}{2} \text{(b) \frac{1}{2} \text{BLOCK A}} \\
+ \frac{1}{2} \land 40 \text{ 16} \land 32.2 \text{ft/s} \rangle (42) \text{T} = 0 \text{T}

T2=(60/32.2)(VA)2

 $U_{1-2} = (100)(x_A) + (W_A)(x_A) - (W_B)(x_B)$

U1-2=(1001b)(2ft)+(401b)(2ft)-(201b)(4ft)

U1-2= 200+80-80= 200 lb.ft

(CONTINUED)

13.21 continued

 $T_1 + U_{1-2} = T_2$ $0 + 200 = (60/32.2) V_A^2$ $V_A^2 = 107.33$

UA=10.36ft/5 €

(b) SINCE THE ZOID WEIGHT AT B IS REPLACED

BY A ZOID FORCE THE KINETIC ENERGY

AT(2) 15 T2=1 MAVA=1 (4019)VA2 T1=0

THE WORK DONE IS THE

SAME AS IN PART IQ)

U1-2= 200 lb.ft

Ti+U1-2=T2

0+200 = (20/g) UAZ

NA=17.94 ft/s

13.22



GIVEN:

ma=11kg mb=5kg
h=2m

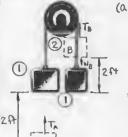
SYSTEM RELEASED FROM
REST

Va=3m/S JUST BEFORE
HITTING THE GROUND

FIND:

IN FRICTION

(b) TENSION IN GACH POETION OF COED



(2) WA

(a) $V_1 = 0$ $T_1 = 0$ ENERGY DISSIPATED $V_2 = V_A = 3 \text{ m/s} = V_B$ $T_2 = \frac{1}{2} (\text{MA+MB}) V_2^2$

 $T_2 = (\frac{16}{2} \text{ kg})(3 \text{ m/s}) = 72 \text{ J}$

U1-2= M, g (2) - Mog (2) - Ep

U1-2=(6hg)(9.81m/s2)(2m)-Ep

U-2= 117.72-EP

 $T_1 + U_{1-2} = T_2$ 0 + 117.72 - Ep = 72

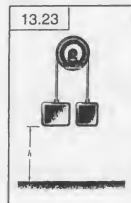
Ep=117.72-72= 45.7 J

b) <u>BLOCK A</u> $T_1 = 0$ $T_2 = \frac{1}{2} m_A U_2^2 = (\frac{11}{2} kg)(3 m/s)^2 = 49.5 J$ $U_{1-2} = (m_A g - T_A)(2) = [(11kg)(9.81 m/s^2) - T_A][2m]$ $U_{1-2} = 215.82 - 2T_A$

Titul-2=T2 0+215.82-2TA=49.5

0-98.1+2TB=22.5

TB= 60.3 N



GIVEN: WA= 2016; WB= 816 H= 1.5 ft SYSTEM RELEASED

FROM REST
BLOCK A HITS THE
GROUND WITHOUT
REBOUND

BLOCK B REACHES A HEIGHT OF 3.5ft

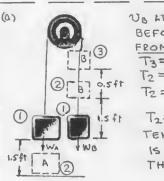
FIND:

(a) Up JUST BEFORE

BLOCK A HITS THE GROUND

(b) ENERGY, Ep, DISSIPATED

BYTHE PULLEY IN FRICTION



 V_{B} LT (2) = V_{A} AT (2) TUST BEFORE IMPACT FROM (2) TO (3); BLOCK B V_{3} = 0 V_{2} = $\frac{1}{2}$ M_B V_{6} ² V_{2} = $\frac{1}{2}$ (81b/32.2 ft/5²) V_{6} ²

 $T_2 = 0.1242 U_B^2$ TENSION IN THE CORD IS ZERO THUS $U_{2-3} = (816)(0.5ft)$

U2-3= 4 16.4+

 $T_2 + U_{2-3} = T_3$ 0.1242 $U_B^2 = 4$ $U_B^2 = 32.2 = U_A^2$

UA=5.68 ft/5

(b) FROM () TO (2)
BLOCKS A AND B

Theo $T_2 = \frac{1}{2} (m_A + m_B) U_2^2$ Just before inpact, $U_2 = U_B = U_A = 5.68 \text{ ft/s}$ $T_2 = \frac{1}{2} (2816/32.2 \text{ ft/s}^2) (5.68)^2$

T2= 14 16.ft

(Ep = ENERGY DISSIPATED BY PULLEY)

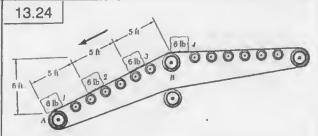
U1-2= (1216)(1.5ft) - Ep = 18-Ep

 $T_1 + U_{1-2} = T_2$

0 + 18-Ep= 14

-Ep= 14-18

Ep=4.00 ft.1b

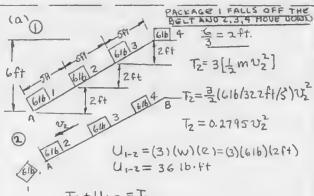


GIVEN:

CONVEYOR IS DISENGAGED, PACKAGES
HELD BY FRICTION AND SYSTEM IS
RELEASED FROM REST. NEGLECT MASS
OF BELT AND ROLLERS. PACKAGE I
LEAUES THE BELT AS PACKAGE 4 CONES
ONTO THE BELT.

FIND:

- (a) VELOCITY OF PACKAGE 2 AS IT LEAVES THE BELT AT A
- (b) VELOCITY OF PACKAGE 3 AS IT LEAVES THE BELT AT A.

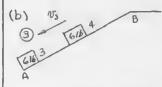


T1 + U1-2 = T2

0+36=0.2795022

N2= 128.8

U2=11.35ft/5 ◀



PACKAGE 2 FALLS
OFF THE BELT AND
ITS ENERGY IS LOST
TO THE SYSTEM
AN 3 AND 4 HOVE
DOWN 2 H.

 $T_2' = (2)[\frac{1}{2} \text{ m N}_2^2]$ $T_2' = (6)[\frac{1}{32.2} + \frac{1}{5^2}](128.8)$

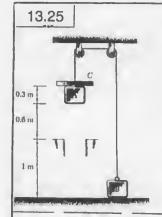
 $T_3 = (2)[\frac{1}{2} m v_3^2]$ $T_2' = 24 \text{ lb.ft}$

 $T_3 = (616/32.2ft/s^2)(03^3)$ $T_3 = 0.1863403^3$ $U_{2-3} = (2)(W)(2) = (2)(616)(2ft)$

 $U_{2-3} = 24 \cdot 16.4t$ $T_2 + U_{2-3} = T_3$

24+24=0.18634N3

V3=16.05 ft/s ◀



GIVEN:

ma= 4 lg

mg= 5 lg

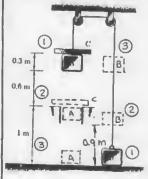
mc= 3 lg

system released

From rest

FIND:

VA, JUST BEFORE IT STRIKES THE GROUND



Position 1 to position 2 $V_1=0$ $T_1=0$ At 2 Before C is
REMOVED FROM THE

SYSTEM $T_2=\frac{1}{2}(M_A+M_B+M_C)V_2^2$ $T_2=\frac{1}{2}(12dg)V_2^2=6V_2^2$ $U_{1-2}=(M_A+M_C-M_B)q(q_m)$

U1-2= (4+3-5)(g)(.9m)=(zkg)(9.81m/f)(Am) U1-2= 17.658 J

T1+41-2=T2

0+17.658 = 6 V2 2 V2 = 2.943

AT POSITION 2, COLLAR C IS REHOUED FROM

POSITION 2 TO POSITION 3 $T_2' = \frac{1}{2} (m_A + m_B) U_2^2 = (9 \text{ kg}) (2.943)$ $T_3' = 13.244 \text{ T}$

T3= 1 (MA+ MB) (V3)= 1 03

U2-3= (MA-MO)(g)(0.7m) = (-1 kg)(9.81m/s2)(0.7m)

Usi-3 = -6.867 J

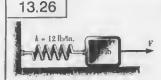
 $T_2' + U_{2-3} = T_3$

13.244-6.867=4.503

U3= 1.417

VA = V3 = 1.190 m/s

UA= 1.190 m/s



GIVEN:

45-0.60, 41-0.40

FORCE F IS SLOWLY

APPLIED UNTIL THE

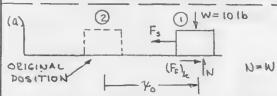
TENSION IN THE

SPRING IS 20 Ib

AND THEN DREASED

FIND:

(A) VELOCITY OF BLOCK AS IT RETURNS TO ITS ORIGINAL POSITION (b) THE MAXIMUM VELOCITY OF THE BLOCK



FIND INITIAL POSITION YO, OF THE BLOCK AT()

AT 1, FS = 2016 FS=12x0 2016=1(4416/ft) x 2016=10/194=0.1389 ft

 $T_1 = 0$, $T_2 = \frac{1}{2} \left(\frac{W}{3} \right) V_2^2 = \left(\frac{1}{2} \right) (10 \text{ lb/32.2ft/s}^2) V_2^2$

 $U_{1-2} = \int_{-F_5} F_5 dx + [F_4]_{\ell_6} (F_4)_{\ell_6} (F_4)_{\ell_6}$

U1-2= 1.389 - 0.5556= 0.8335 16.ft

 $T_1 + U_{1-2} = T_2$ 0 + 0.8335 = 0.1553 U_2^2 $U_2^2 = 5.367$ $U_2 = 2.32$ ft/s

AT ORIGINAL POSITION, U=2.32ft/s

(b) FOR ANY POSITION & AT A DISTANCE

XTO THE RIGHT OF THE ORIGHAL POSITION (2)

TI=O Tz=\frac{1}{2}(\frac{10}{3})(\frac{10}{2}\frac{1}{2})^2=0.1553\frac{10}{2}\frac{2}{3}.

 $U_{1-2} = \int_{-F_3}^{F_5} dx + \int_{-F_5}^{F_5} (F_5)_0 dx$ $\kappa_0 = 0.1389$

U1-2'= [-144x2-] + (Ff) (x-x0) (Ff)= 416

 $T_1 + U_{1-2} = T_2$ $0 + (72 \text{ lb/ft} (0.1389)^2 - 1/2) + (4 \text{ lb})(1/2 - 0.1389)$ = 0.1553 U_2^2

MAX V WHED dx = 0 - 1447+4=0

MAX U, WHEN X = 0.027778 m

0.1553 Umax= (72) ((0.1389)-(0.02718)]+(4) (.02718-0138)

0.1553 $U_{\text{max}}^2 = 1.3336 - 0.4445 = 0.8891$ $N_{\text{max}}^2 = 5.725$

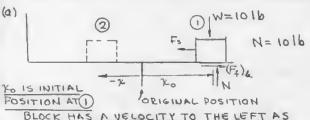
Umax= 2.39 ft/s -



45=0.60, 4k=0.40
FORCE F IS SLOWLY
APPLIED UNTIL THE
TENSION IN THE
SPRING IS ZOID
AND THEN RELEASED

FIND:

(a) DISTANCE THE BLOCK MOVES TO THE LEFT BEFORE COMING TO A STOP
(b) WHETHER THE BLOCK THEN MOVES BACK TO THE RIGHT.



BLOCK HAS A VELOCITY TO THE LEFT AS IT REACHES ITS ORIGINAL POSITION

(SEE P 13.26)

$$K = 12 lb/ln = 144 lb/ft$$

$$T_1 = 0 \quad T_2 = 0 \qquad F_5 = 144 \times$$

$$U_{1-2} = \int_{F_6}^{-F_6} dx + \int_{F_6}^{-F_6} (F_6)_{k} dx \qquad (F_6)_{k} = 4 lb$$

$$T_7 = 0 \quad T_8 = 0.4 lb/ft$$

$$T_8 = 0.4 lb/ft$$

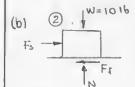
$$U_{1-2} = -\frac{144}{2} \chi^2 \int_{x_0}^{x} + (F_{\epsilon})_{\epsilon} (-x - x_0)$$

TI+U1-2=T2 0-72(x-x0)(x+x0)-4(x+x0)=0 (b

 $-72(x-x_0)-4=0$ AT ① $F_5=201b$ $-72x=4-72x_0$ $F_5=kx_0=144x_0$ $x_0=\frac{20}{144}=0.1389$ y=0.0833ft

TOTAL DISTANCE MOVED TO THE LEFT = Yo+X

 $\chi_{0} + \chi = 0.1389 + 0.0833$ $\chi_{0} + \chi = 0.2225 + 0.0000$



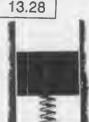
IF FS AT 2 IS LARGER
THAN THE MAXIMUM
POSSIBLE STATIC
FRICTION FORCE, THEN
BLOCK WILL MOVE TO
THE RIGHT

FROM (a) WITH X= 0.0833 ft

 $F_s = (144)(0.0833) = 12 lb$ $(F_s)_s = 1/5 N = (0.60)(10) = 6 lb$

SINCE FS>(F1)S

BLOCK HOURS TO THE RIGHT

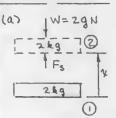


GIVEN:

3hg Block RESTS ON Zhg BLOCK.
WHICH IS NOT ATTACHED TO A
SPRING OF CONSTANT 40 N/M
UPPER BLOCK IS EUROBRILY
REMOVED

EIND:

(a) Nomax OF 2 kg BLOCK
(b) MAXIMUM HEIGHT, h.
RENCHED BY THE 2 kg BLOCK



AT THE INITIAL POSITION ()
THE PORCE IN THE SPRING
EQUALS THE WEIGHT OF
BOTH BLOCKS, I.E. 59 N
THUS AT A DISTANCE X
THE FORCE IN THE SPRING
IS FS= Ig-&X

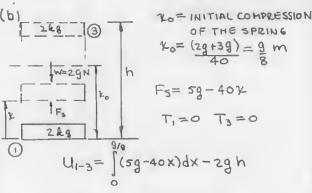
HAX VELOCITY OF THE 2 Lg BLOCK OCCURS WHILE THE SPRING IS IS STILL IN CONTACT WITH THE BLOCK.

 $T_1 = 0$ $T_2 = \frac{1}{2} m v^2 = (\frac{1}{2})(2kg)(v^2) = v^2$ $U_1 - 2 = \int (5g - 40x) dx - 2gx = 3gx - 20x^2$

 $T_1 + U_{1-2} = T_2$ $O + 3gx - 20x^2 = v^2$ (1) MAX V WHEN $\frac{dv}{dx} = 0 = 3g - 46x$ $\frac{dv}{dx} = 0 = 3g - 46x$

SUBSTITUTE IN (1) 7. (MAX V)= 0,7358 M VHAX = (3)(9.81)(0.7358)-(20)(0.7358)2 = 21.65-10.83=10.83

Umax= 3.29 m/5



$$U_{1-3} = \frac{59^2 - 209^2 - 29h}{64}$$

$$T_1 + U_{1-3} = T_3 \qquad 0 + \frac{209^2 - 29h}{64} = 0$$

$$h = \frac{109}{64} = \frac{(10)(9.81)}{64}$$

h= 1.533 m

13.29



GIVEN:

BLOCK WHICH IS ATTACHED TO A SPRING OF 40 N/M WHEN UPPER BLOCK IS SUDDENLY REMOVED

EIND:

(a) Nmax OF 2 kg BLOCK
(b) MAXIMUM HEIGHT h
REACHED BY 2kg BLOCK

(a) SEE SOLUTION TO (a) OF P13.28

Nmax=3.29 m/s

(b) REFER TO FIGURE IN (b) OF P13.28

T1=0 T3=0

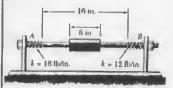
U1-3= \int (5g-40x)dx-2gh

THE Z & BLOCK THE INTEGRATION MUST BE CARRIED OUT THROUGHOUT THE TOTAL DISTANCE N.

 $T_1 + U_{1-3} = T_2$ $0 + 5gh - 20h^2 - 2gh = 0$ $h = \frac{3g}{20} = \frac{(3)(q.81)}{20}$

h=1.472m

13.30



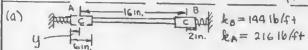
GIVEN:

WC= BIB COLLAR C COMPRESSES SPRING AT . B ZIN. AND. IS RELEASED

FIND:

(a) DISTANCE TRAVELED BY COLLAR WITH NO FRICTION.

(b) SAME AS (Q) WITH FRICTION, 4=0.35



SINCE COLLAR C LEAVES THE SPRING AT B AND THERE IS NO FRICTION IT MUST ENGAGE THE SPRING AT A

TA+UA-B=TB 0+2-1084=0

y= 0.1361 ft=1.633 in.

TOTAL DISTANCE = 2+16-(6-1.633)=13.63 m.

13.30 continued

(b) ASSUME THAT C DOES NOT REACH THE SPRING AT B BECAUSE OF FRICTION $V_1 = V_2 = V_3 = V_4 =$

TA+UA-0=TD 0+2-2804=0

THE COLLAR MUST TRAVEL 16-6+2=12 IN. BEFORE IT ENGAGES THE SPRING AT 8. SINCE y = 8.57 IN.
IT STOPS BEFORE ENGAGING THE SPRING AT B

TOTAL DISTANCE=8.57 IN.

13.31



GIVEN:

WC=61b.
UPPER SPRING IS
COMPRESSED ZIN AND
COLLAR C 15 RELEASED

FINO:

(a) Ym, THE HAVINUM DERECTION OF THE LOWER SPRING

(b) Um, THE HAXIMUM VELOCITY OF THE COLLAR

(a) 18 lb/in. (Fe) 2 2/12

SPEING CONSTANTS 1816/IN = 216 16/ft 1216/IN.= 144 16/ft

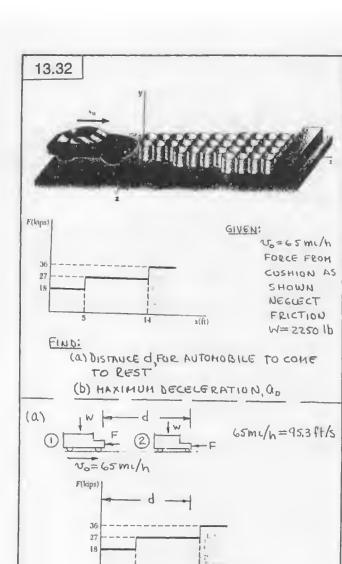
MAXIMUM DEFLECTION AT 2 WHEN VELOCITY OF COLLAR C IS ZERO $\textcircled{1}_{2}=0$ $\textcircled{1}_{2}=0$ $\textcircled{1}_{1}=0$ $\textcircled{1}_{1}=0$

U1-2=Ue+Ug= (Fe),dx- (Fe)2dx+Wc(1+y)

U1-2= (216 16/ft) (=ft) -(144 16/ft) (y2) + 616(1+y)
U1-2= 3-72 y2+6+64=-72 y2+69+6

 $T_1+U_{1-2}=T_2$ 0 - 72 ym +6ym+6= 0 $y_m=\frac{1}{3}$ ft = 4.001n.

(b) Maximum VEWCITY OCCURS AS THE LOWER SPEING IS COMPRESSED A DISTAUCE y' $T_1=0$ $T_2=\frac{1}{2}$ $M_cV^2=\frac{1}{2}(\frac{6}{9})V^2=\frac{316}{32.2}ft/s^2)V^2$ $T_1+U_{1-2}=T_2$ 0 -72 $y'^2+6y'+6=(0.09317)V^2$ $\frac{dV^2}{dy!}=0.041667ft$ $-0.125+0.250+6=0.09317V_M^2; V_M=811ff/s=973m/s^2$



ASSUME AUTO STOPS IN SEDE 14 ft

U1 = 95.33 ft/s T1 = 2 M U1 = 1 (225016) (9534/5)

Ti= 317,530 lb.ft=317.53k.ft

U2=0 T2=0

 $U_{1-2} = (18k)(5ft) + (27k)(d-5)$ = 90 + 27d - 135 = 27d - 45 k. ft $\Gamma_1 + U_{1-2} = \Gamma_2$

317.53 = 27d-45

d= 13.43ft

ASSUMPTION THAT d < 14 ft 15 OK

(b) MAXIMUM DECELERATION OCCURS WHEN

F IS LARGEST. FOR d = 13.43 ft, F= 27k.

THUS $F = Ma_D$ (27,000 lb) = $\left(\frac{2250 \text{ lb}}{32.2 \text{ FH/s}^2}\right) \left(a_D\right)$

ap = 386 ft/s2

13.33

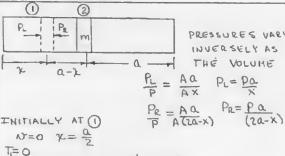
GIVEN:

PISTON AREA A
PISTON MASS M
INITIAL PRESSURE P
PRESSURE VARIES
INVERSELY WITH
VOLUME. PISTON

MOVED YZ AND RELEASED

FIND:

VELOCITY OF THE DISTON AS IT RETURNS



T=0 AT (2), $\chi = \alpha$, $T_2 = \frac{1}{2} m v^2$ $U_{1-2} = \int (p_1 - p_2) A dx = \int p_0 A \left[\frac{1}{x} - \frac{1}{2a-x} \right] dx$ $U_{1-2} = p_0 A \left[ln x + ln(z_0 - x) \right]_{a/2}^{a}$ $U_{1-2} = p_0 A \left[ln a + ln a - ln(a/2) - ln(3a/2) \right]$ $U_{1-2} = p_0 A \left[ln a^2 - ln 3a^2/4 \right] = p_0 A ln(4/3)$ $U_{1-2} = T_2$ $O + p_0 A ln(4/3) = \frac{1}{2} m v^2$ $v^2 = 2p_0 A ln(4/3) = 0.5754 p_0 A$

U=0.759√<u>pan</u>

13.34 P GI

GIVEN:

ACCELERATION OF GRAVITY 90 AT EARTHS SURFACE

ACCECERATION OF GRANITY

GIN AT HEIGHT IN ABOUF

THE EARTHS' SURFACE

INTERMS OF GO, N. P.

AND ERROR IN WEIGHT AT IN IF WEIGHT

AND FEROR IN WEIGHT AT IN IF WEIGHT AT FARTHS' SURFACE IS USED FOR (a) h=1km

$$F = \frac{GM_{em}}{(h+R)^2} = \frac{GM_{em}/R^2}{(\frac{h}{R}+1)^2} = mgh$$
ATEARTHS'SURFACE (h=0) GMEM/R^2=mgo
$$GM_{em}/R^2 = go \quad g_h = \frac{GMe/R^2}{(\frac{h}{R}+1)^2}$$

THUS $g_h = \frac{g_o}{(\frac{h}{R}+1)^2}$

13.34

continued | R = 6370 km

AT ALTITUDE h TRUE WEIGHT F= Mgh=WT ASSUMED WEIGHT WO = mgo - mgh = go-gh
ERROR = E = Wo-WT = mgo-mgh = go-gh
Wo

$$g_{h} = \frac{g_{0}}{(\frac{h}{k}+1)^{2}} = \frac{w_{0}-w_{1}}{W_{0}} = \frac{mg_{0}-mg_{h}}{mg_{0}} = \frac{g_{0}-g_{h}}{g_{0}}$$

$$= g_{0} = \frac{g_{0}}{(\frac{h}{k}+1)^{2}} = \frac{g_{0}-g_{0}}{(\frac{h}{k}+1)^{2}} = \frac{g_{0}-g_{0}}{(\frac{h}{k}+1$$

(b)
$$h = 1000 \text{ km } P = 100 \left[1 - \frac{1}{(1 + \frac{1000}{6370})^2}\right]$$

 $P = 25.3\%$

13.35

GIVEN: VELOCITY AT HOODS SURFACE=U. VGLOCITY AT HEIGHT h= 0 RADIUS OF THE MOON, PM ACCELERATION OF GRAVITY ON THE HOONS' SURFACE, SM

FIND: FORMULA FOR hy/hu. WHERE UN IS FOUND USING NEWTONS LAW OF GRAVITATION AND NU IS FOUND USING A UNIFORH GRAUITATIONAL FIELD

NEWTONS LAW OF GRAVITATION

$$T_{i} = \frac{1}{2} m v^{2}$$

$$U_{i-2} = \int_{C} -F_{i} dr$$

$$F_{in} = \frac{m g m P^{2} m}{r^{2}}$$

$$P_{in} = \frac{m g m P^{2} m}{r^{2}}$$

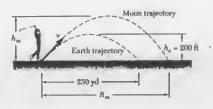
$$\frac{1}{2} m v_0^2 + m g_m (\varrho_m - \frac{\varrho_m}{\varrho_m + h_n}) = \frac{1}{2} m v^2$$

$$h_n = \underbrace{\left(\frac{v_0^2 - v^2}{2g_m}\right) \left[\frac{P_m}{P_m - \left(\frac{v_0^2 - v^2}{2g_m}\right)}\right]}_{2g_m} \tag{1}$$

UNIFORM GRAVITATIONAL FIELD

$$T_1+U_{1-2}=T_2$$
 $\frac{1}{2}mv_0^2$ $\frac{1}{2}mghu = \frac{1}{2}mv^2$
 $hu = (v_0^2-v^2)/2gm$ (2)

13.36

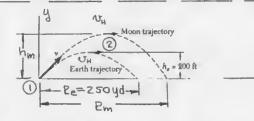


GIVEN:

EARTH TRAJECTORY AS SHOWN MAGNITUDE AND DIRECTION OF U ON THE GARTH IS THE SAME ON THE HOOM TRAJECTURY IS A PARABOLA gm= 0.165 ge

FIND:

RANGE PM OF THE BALL ON THE HOON



SOLVE FOR hm

AT MAXIMUM HEIGHT THE TOTAL DELOCITY IS THE HORIZONTAL COMPONENT OF THE VELOCITY WHICH IS CONSTAUT AND THE SAME IN BOTH CASES T= 1 mv2 T2= 1 mv+2

U1-2 =-mgehe EARTH U1-2 =-mgmhm MOON

EARTH 1 MUZ mache = 1 mun2

MOON 5 MUZ- Mgmhm=5MU,2.

SUBTRACTING -gehetgmhm=0 hm = ge ne gm hm=(200ft) (ge 0.165ge)= 1212ft

(y-he)=-Ce(x-Pe)2 EARTH

(y-hm) = - Cm (x-2m) 2 HOON

AT X=0, U IS THE SAME THUS DE ISTHE SAME

dy = Ce Re = Cm Rm

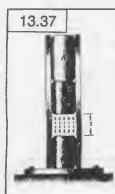
dx = 0

AT X=0, y=0 he = Ce Re hm=Cm Rm

The control of the contr

hm = Cm Em2 = Pm he Ce Re2 Pe hm = ge = 2m Rm = (ge/(0.165 ge)(250yd)

Pm= 1515 yd 4



MA= 300-9 (NON HACUETIC)

MB= 200-9 (MAGNETIC)

R= 4 MM, INITIALLY

REPELLING FORCE

BETWEEN BAND C IS

F = K/X²

BLOCK A IS SUDDENLY

REMOVED NO AIR RESISTANCE

FIND:

(a) HAXIMUM VELOCITY,

Um of B

(b) MAXIMUM ACCELERATION

am of B

13.38

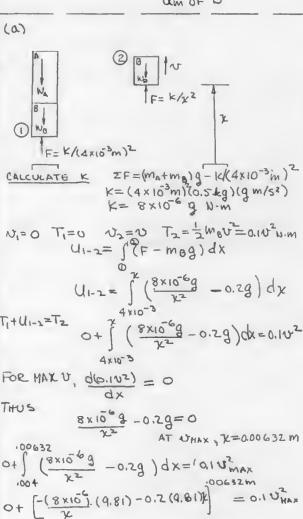
GIVEN:

WB=0.4-1b (MAGNETIC)
WA=0.6-1b(NON-MAGNETIC)
Y=0.15 IN. INITIALLY
REPOLLING FORCE
BETWEER BAND C 15
F= K/L2 HO MR RESISTING
BLOCK A 15 PLACED ON
BLOCK B AND RELEASED

FIND:

(a) MAXIMUM UELOCITY OF A AND B

(b) HAXIMUM DEFLECTION OF



(b) MAXIMUM ACCEUERATION AT X=0.004 M

WHEN EF ARE THE GREATEST

(8×106)(9.81)/(0.004)2-(0.2)(9.81)=(0.2)QM

ZF=K/x2-WB=MBQ

(a) CALCULATE K EQUILIBRIUM AT () X = K/K; - W0 = 0 74=0.15in = 0.0125ft K= (.0125ft) (0.416) K=0.0000625 ft3.16 Uz=V Tz=1 (MA+MB) V V=0 T=0 T2=2 (116 1/52) 12 T2= 0.01553 V2 [F-(WA+WB)]dx [.0000 625 -1]dx= 0.01553 03 FOR MAX U, d (.001553U2) 0.0000025-1=0 X=0.007906ft AT UM ? 0.007906 0.0000625 -1 dx= 0.01553Vm Um = 0.10876 Nm= 0.3298 ft/s Nm= 3.96 m/s (b) MAXIMUM DEFLECTION WHEN V=0

 $T_1=0$ $T_2=0$ $X_1=0$ $X_2=0$ $X_1=0$ $X_2=0$

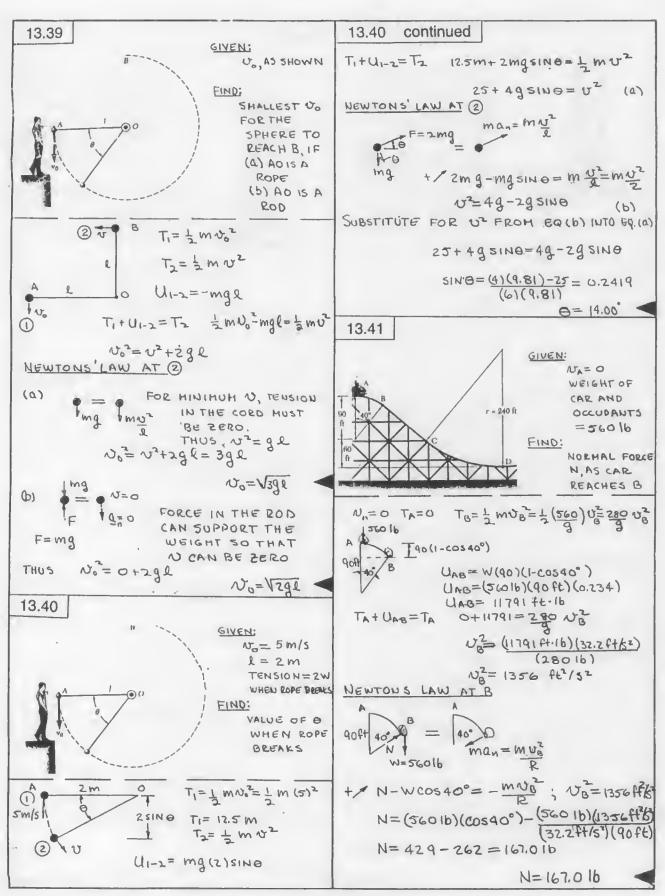
 $-0.0000625 \left[\frac{1}{2} - \frac{1}{0.0125} \right] - 12 + 0.0125 = 0$

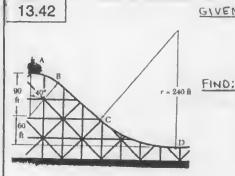
MAXIMUM DEFLECTION = 0.0125-0.005=0.0075ft

= 0.090 in.

2=0.005ft

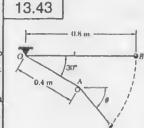
NHAX = 0.1628 m/s NH = 162.8 mm/s





UA= 0, CAR AND OCCUPANTS WEIGH SO 16

MAXIMUM Nu AND MINIMUM, NMIN NORMAL FORCE ON THECAR AS IT GOES FROM A TO D



GIVEN:

SPHERE RELEASED PROH REST AT B. (UB=0)

FIND:

TENSION IN THE CORD, (a) JUST BEFORE IT COMES IN CONTACT WITH THE PEG (b) JUST AFTER CONTACT WITH PEG

NORMAL FORCE AT B

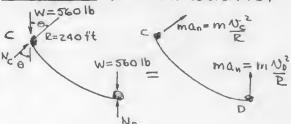
SEE SOLUTION TO PROB. 13.41, No=167016

NEWTONS LAW

FROM B TO C (CAR HOVES IN A STRAIGHT LIVE) +/ No-WCUS40=0 ma No=(560) cos 40°

NB=42916

AT C AND D (CAR IN THE CURUE ATC)



+ 1 Nc - WCOSO = W De Nc= 560 (cose+ Nc2)

AT D

$$N_0 = 560 (1 + \frac{U_0^2}{gR})$$

SINCE VO > VC AND COSO < 1, NO > NC WORK AND ENERGY FROM ATO D

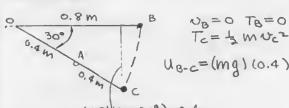
UA-B= W (90+60)=(560 16) (150 ft) UN-B= 84000 lb.ft

0+84000=280 ND2 TA+UA-B=TB N2= 300

No= 560 (1+ \(\vec{v}_{0}^{2}\))= 560 (1+\(\frac{300}{240}\))= 1260 lb

NHIN= NB= 167.010; NHAZNO=126019

VELOCITY OF THE SPHERE AS THE CORD CONTACTS A



(0.8)(SIN 30°)=0.4 TB+UB-=Tc 0+0.4 mg = 1 m Uc2

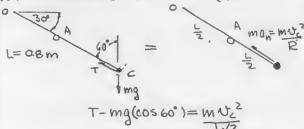
U= = (0.8)(9)

NEWTONS LAW CORD ROTATES ABOUT POINT O (a) 0 300 (R=L) man=mva 1=0.8m

> + T-malcos60 = MNc T= mg(cos.60°)+m(0.8)g

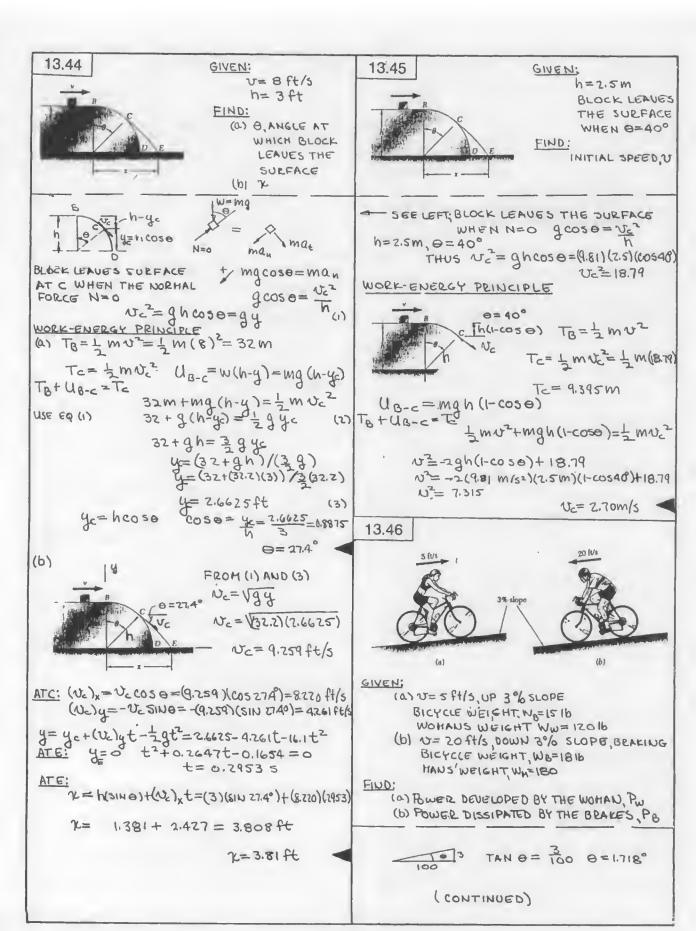
T= 3 mg T= 1.5 mg

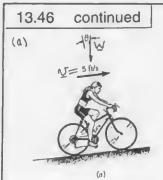
(b) CORD ROTATES ABOUT A (R= 1/2)



T=mg/2+m(0.8)(g)/0.4 T= (+2) mq = = mg

T= 2.5 mg





W=Wg+Ww=15+120 W=13516 Pw=W·Y=(WSINB)(V) Pw=(135)(SIN1718°)(5) Pw= 20.24 ft:16/5

Pw= 20.2 ft.16/s



W= Wot Wm= 181180
W= 19816

BRAKES HUST

DISSIPATE: THE

POWER GENERATED
BY THE BIKE AND THE
MAN GOING DOWN
THE SLOPE AT
20 ft/s

PB=W.V=(WSINB)(U)
PB=(198)(SIN1.718)(20)

Pn= 118,7 ft.16/5

13.47

(b)



GIVEN:

(a) HASS FLOW RATE, M(KG/H), L(M), b(M) (b) MASS FLOW RATE, W(TONS/H), L(ff), b(ft) MOTOR EFFICIENCY, N

FIND:

(b) POWER IN MP

(a) MATERIAL 13 LIFTED TO A HEIGHT & AT

A RATE, (M Kg/H) (g M/S2) = (mg (N/H))

THUS

LU = [mg(N/H)] (bin)) - (mg b) N.M

1000 N.M/S = 1 kW

(3600 S/H) - (3600 N.M/S)

THUS, INCLUDING TO TOR EFFICIENCY of

(3600) (1000 N.M/S) (7)

EW

P(kw) = mg b (N.M/S)

(3600) (1000 N.M/S) (7)

P(kw) = 0.278×10 - 6 mg b

M

(b) Au = [w(tons/H)(2000 1b/Ton) [b(ft)]

(0) Au = (W(TONS/H)(ZOCO 16/TON)(6/FE))
= Wb f1.16/s; 1hp= 550 ft.16/s

WITH M, hP = [W. b (f1.16/5)][1hP = 1.010x10 1/16

13.48



GIVEN:

2000 ID CAR CEAR WHEEL DRIVE,

SKIDS FOR FIRST 60ft WITH FRONT WHEELS

OFF THE GROUND, ME = 0.60

ROLLS WITH SLIDING IMPENDING FOR

REMAINING 1260 Ft WITH 60% OF

ITS WEIGHT ON REAR WHEELS, 45=0.85

FIND:

(a) HP DEVELOPED AT END OF 60ft PORTION OF THE RACE
(b) HP DEVELOPED AT THE END OF THE PACE

(a) FIRST 60 ft (CALCULATE VELOCITY AT 60 ft)

FORCE GENERATED BY REAR WHEELS=4KW

SINCE CAR SKIDS, THUS F=(0.6)(2000 16)

F= 1200 16

WORK AND ENERGY T=0 T= 1 W V2=1000 V20 T+U1-2=T2 U1-2= (F)(60ft)=(12001b)(60ft)=72,000 lbft

 $T_1+U_{1-2}=T_2$ $0+72_1000=\frac{1000}{9}$ V_{60}^2 $V_{60}^2=(72)(322)=2318.4$ $V_{60}=48.15$ ft/s

P= (12001b) (48.15 ft/s)
P= 57780 ft.1b/s

1 hp=550 ft.16/5

hp=(57780ft.1b/s)=105.1 (550ft.1b/s) hp

(b) END OF PACE (CALCULATE VELOCITY AT 1320 Pt)

FOR FIRST GOFT, FORCE GENERATED

BY REAR WHEELS -F5= 1200 16 (SEE &!)

FOR REMAINING 1260 PT WITH 60%

OF WEIGHT ON REAR WHEELS, THE

FORCE GENERATED AT IMPENDING

SLIDING IS 45 (60) (W=6.85) (6.60) (2000)

 $\frac{T_1 + U_{1-2} - T_2}{T_1 + U_{1-2} - T_2} = \frac{T_1 - 0}{2} = \frac{1000}{2} V_{1320}^2 = \frac{1000}{2} V_{1320}^2$

11-2=(F5)(60ft)+(F1)(1260ft)

U1-2=(1200 16) (60 ft)+(1020 16) (1260 ft)
U1-2=1,357,200 16. ft

0+1,32,500=1000 N3

V1320= 209 ft/s

Power = $F_{I} \cdot v_{1320}$ P=(1020'1b)(209 ft/s)=213,230 ft/s

hp = (213,200 ft.16/s) = 388



1000 kg CAR, REAR WHEEL DRIVE

SKIDS FOR FIRST 20 M, WITH FRONT WHEELS

OFF THE GROUND, UK = 0.68

ROLLS WITH SCIDING IMPENDING FOR

REHAINING 380 M WITH 80% OF ITS

WEIGHT ON REAR WHEELS, US = 0.90

FIND:

(a) POWER DEVELOPED ATEND OF 20 m (1) kw sho) (b) Bwer Developed at END OF THE RACE (kw sho)

(a) FIRST ZOM (CALCULATE VELOCITY AT ZOM)

FORCE GENERATED BY REAR WHEELS= MKW

SINCE CAR SKIDS. THUS FS=(0.68)(1000)(9)

FS= (0.68)(1000 kg)(9.81 M/52)=6670.8 M

WORK AND ENERGY T=0, Tz=1 MN2=500020

Ti+U1-z=Tz

U₁₋₂= (20 m)(F₅)= (20 m)(6670.8 N) U₁₋₂= 133420 T 0+133,420= 500 020

 $U_{20}^{-} = 133,420/500 = 266.83$

 $U_{20} = 16.335 \text{ m/s}$

POWER = (F5) (V20) = (6670.8 N) (16,335 m/5)
POWER = 108,970 J/5=108,97 kJ/5

1 RJ/s=1 RW

IMP=0.7457 kw POWER=109.0 kT/5=109.0 kw
POWER=(109.0 kw) = 146.2 Mp
(0.7457 kw/hp)

(b) END OFRACE (CALCULATE VELOCITY AT 400 M)

FOR REMAINING 380 M, WITH

80% OF WEIGHT ON REAR WHEELS

THE FORCE GENERATED AT IMPENSING

SLIDING IS (45)(0.80)(Mg)

 $F_{I} = (0.90)(0.80)(1000 \text{ hg})(9.81 \text{ m/s})$ $F_{I} = 7063.2 \text{ N}$

WORK AND ENERGY, FROM 20 m(2) TO 28 m(3) $V_2 = 16.335 \text{ m/s} (FROM PART (a))$ $T_2 = \frac{1}{2} (1000 \text{ kg}) (16.335 \text{ m/s})^2$ $T_2 = 133420 \text{ T}$ $T_3 = \frac{1}{2} \text{ mN}_{300} = 500 \text{ U}_{30}^2$

 $U_{2-3} = (F_1)(380 \text{ m}) = (7063.2 \text{ N})(380 \text{ m})$ $U_{2-3} = 2,684,000 \text{ T}$ $T_2 + U_{2-3} = T_3$

 $(133,4201)+(2,684,0001)=5000_{30}^{3}$

POWER = $(F_{\Sigma})(v_{30}) = (7063.2 \text{ N})(75.066 \text{ m/s})$ = 530,200 J

hw POWER = 530,200J = 530 kWhp POWER = 530 kW = 711 hp(0.7457 kW/hp) 13.50



GIVEN:

CAR MASS, Mc= 1200 kg LIFTHASS, Mc= 300 kg SYSTEM RISES 2.8 M IN 155.

FIND:

(a) AUERAGE POWER OUT PUT OF PUMP, (PP), (DIAUTE AGE ELECTRIC POWER, (PE), WITH 7 = 82%

(a) $(P_p)_A = (F)(U_A) = (m_c + m_c)(g)(U_A)$ $U_A = 5/t = (28 m) /(55) = 0.18667m/5)^{\frac{1}{5}}$ $(P_p)_A = [(1260 + g) + (300 + eg)](4.81 m/52)(0.18667m/5)^{\frac{1}{5}}$

(Pp) = 2747 kT = 275 kW (D) (Pe) = (Pp) / m = 6.75 kW (0.82)

(PE) = 3.35 LW

13.51



GIVEN:

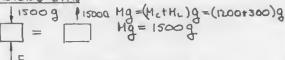
CAR HASS, Mc= 1200 Lg
LIFT MASS, Mc= 500 kg
PEAK VELOCITY AT
HID HEIGHT IN
7.5 S INCREASING
UNIFORMLY. VELOCITY
DECREASES UNIFORMLY
TO O, IN ANOTHER
7.55

FIND:

HANHUM LIFTING

7.55
PEAK DUMP POWER,
P=6 kW, WHEN
VGLOCITY IS HAXIMUM

NEWTONS LAW



+ EF=F-15009=1500Q (1)
SINCE MOTION IS UNIFORMLY ACCEURATED

Q= CONSTANT

THUS, FROM (1), F IS CONSTANT
AND PEAR POWER OCCURS
WHEN THE VELOCITY IS A
STANT MAXIMUM AT 7.55.

O CONSTANT MAT

Q = NMAX

 $P = (6000 \text{ W}) = (F,)(V_{HAX})$ $N_{HAX} = (6000)/F$

THUS a= (6000) /(7.5)(F) (2)

SUBSTITUTE (2) INTO (1)

F-1500g = (1500) (6000) /(75)(F)

F2-(1500kg) (981 m/s) F- (1500kg) (6000 n·m/s) = 0)

F= 14,800'N

F=14.8 &N

13.52 GIVEN:

> W= 100 TONS P = 400 hp U= 50 mi/h CONSTANT

FIND:

(a) F, FORCE NEEDED TO OVERCOME AXLE FRICTION, ROLLING PESISTANCE AND AIR RESISTANCE

(b) AP, ADDITIONAL UP TO HAINTAIN THE SAME SPEED UP A 1-PERCENT GRADE

(1) P= 400 hp = (550 ft.1b/hp)(400 hp)=220,000 ft.1b

N= 50 mi/h= 73.33 ft/s

P= 5. V

$$F_e = P/v = (220,000 \frac{f+1b}{5})/(13.33 \frac{f+}{5})$$

(b) of w _ v=73.33 ft/s

W= (100 TONS) (2000 16/100) W=200,000 lb

F= 3000 lb .

Co = tAU 1 = 0.573° DP=WSIUD.V

AD=(200,000 lb)(SIN.573°)(73.33 ft/s) AP= 146,667 ft. 16/5

DP= 267 hP

13.53 GIVEN:

> W= 600 TONS UNIFORM ACCELERATION FROM OTO SOHILL IN 405 CONSTANT SONULU AFTER 405 HORIZONTAL TRACK FR, FRICTION AND ROLLING RESISTANCE = 3000 lb NEGLECT AIR RESISTANCE

FIND:

P. POWER REQUIRED AS A FUNCTION OFTIME t.

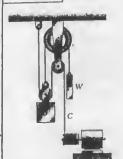
U= 50m1/4=73.33 ft/s W=(600 TOUS)(2000 16/TOU) V W=1200,000 1b U, = 73.33 ft/s FOR UNIFORM HOTION 405. a = V/L=(73.33ftb)/(405) W=GOOTONS a=1.833f+/52 N= 1.833t FR=3000 lb ZF=F-Fe=Ma=Wa F=(3000 16)+(1,200,000 16) (1.833 ft/3) (32.2 ft/52) F= 71.311 1b P= F.V=(71,311)(1.833+)=130,710+ 16.ft

t >405 P=(3000)(73.3) = 400 hp4

P= 130,710 t/550=238t (hp)

FOR t < 40s

13.54



GIVEN:

ME = 3000 kg, ELEVATOR HASS MW=1000 kg, COUNTER WEIGHT

(a) P(kw) DELIVERED BY HOTOR WHEN VELOCITY OF E, VE= 3m/S DOWN AND CONSTAUT (QE=0)

(b) P(kw) WHEN NE= 3M/S UPWARD QE=0.5 M/S' DOWN

VE=3m/s

(a) ACCELERATION = O COUNTERWEIGHT ELEVATOR HOTOR

+1 ZF=ZTZ+TW-ME9=0 -1ZF=Tw-Hug=0 Tw=(1000 &g)(9.81 m/s2) 2T=(-98101)+(3000 &g)(9814) Tw= 9810 N Tc= 9810 N

KINGHATIC S

2x==xc 2x==xc Uc=2U==6W/s

P=T-Uc=(9810N)(6M/s)=58,860 J/s P(LW)= 58.9

(b) a==0.5 m/s 1 U== 3 m/s 1

COUNTERWEIGHT ELEVATOR = W Mwaw MEGE aw=a== 0.5 m/s COUNTER WEIGHT ZF=MQ EF= Tw-Mwg= Mwaw

Tw=(1000kg)(9.81 m/s2)+(0.5 m/s2)]

Tw= 10310 N

ELEVATOR ZF= Ma +1 ZF= ZTC+TW-MEg= -MEGE

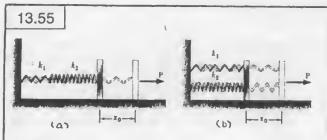
2Tc = (3000 leq) [(9.81 m/s2-10.5 m/s2)]-10310 N

TC= 8810 N NC=6m/5 (SEE LA)

P= Tc. Vc = (8810N) (6m/5) P=52,860 T= 52.860 &T=52.86 &W

P(kw)= 52.9

P=238t hp =



P CAUSES DEFLECTION to, IS SLOWLY APPLIED

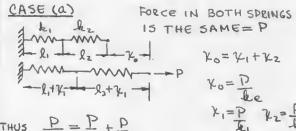
(a) SPRINGS & AND & IN SERIES

(b) SPRINGS & AND & IN PARALLEL

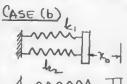
FIND:

SINGLE EQUIVALENT SPRING Re WHICH CAUSES THE SAME DEFUECTION

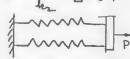
SYSTEM IS IN EQUILIBRIUM IN DEFLECTED XO POSITION.



THUS De le Lez



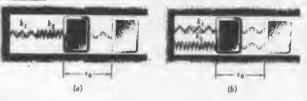
DEFLECTION IN BOTH SPEINGS IS THE SAME = 120



P=k, YotkzXo P=(h,+lez)Xo P=keXo

EQUATING THE TWO EXPRESSIONS FOR P = (k,+hz) ko= ke xo ke=k,+kz

13.56



GIVEN:

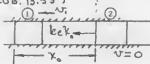
BLOCK OF MASS M BLOCK HOUSD TO XO AND RELEASED FROM REST.

FIND:

MAXINUM VELOCITY, UMAX

13.56 continued

WE WILL USE AN EQUIVALENT SPRING CONSTANT



CHOOSE () AT INITIAL UNDEFLECTED POSITION

V=0 T = 1 m V, 2

CHOOSE 3 AT X0 WHERE U=0 V2=1 ke X0 T2=0

THUS V = UMAX = 40 / ke

CASE (a) Re= k, hz VHAX= Ko VM(k,+kz)

CASE (b)

Re= k1+k2 VHAX= YOV k1+k2

13.57



 $k_1 = 12kN/m$ $k_2 = 8kN/m$ m = 16kq

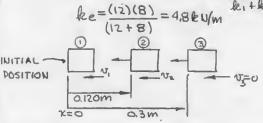
INITIAL POSITION, 300MM

FIND:

(a) HAXIMUM VELOCITY, THAX

(b) VELOCITY 120 mm FROM INITIAL POSITION

FORM SPRINGS IN SERIES , Re= kilz



(a) ATO, SPRING DEFLECTION, O V=0, T== 1 m V= 8V2

AT (3), U3=0 T3=0 V3=1 1 4.2=/4800)(0.3)=214 Nom

 $V_3 = \frac{1}{2} k_e \mu_3^2 = \frac{4800}{2} (0.3) = 216 \text{ N·m}$ $V_3 = \frac{1}{2} k_e \mu_3^2 = \frac{4800}{2} (0.3) = 216 \text{ N·m}$ $V_3 = \frac{1}{2} k_e \mu_3^2 = \frac{4800}{2} (0.3) = 216 \text{ N·m}$

DHAX +0= 0+ 216

(b) $T_2 = \frac{1}{2} m V_2^2 = 8 V_2^2$ $V_2 = \frac{1}{3} k_e k_1^2 = \frac{(4800)}{2} (0.120)^2 = 34.56 \text{ N·m}$

 $T_2+V_2=T_3+V_3$ $80^2_2+34.56=0+216$ $V_2=4.76$ m/5



W=61b

\$_1=51b/in.

\$_2=101b/in.

\$_3=201b/in

INITIAL DISPLACEMENT,

\$_0=1.811.TOLEFT

FROM UNSTRETCHED

POSITION

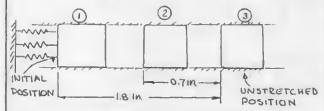
\$_0=0

FIND:

(A) MAXIMUM UELOCITY, UHAX

(b) VELOCITY AT O.7 IN. FROM INTIAL POSTION

EQUIVALENT le= k,+k2+k3 (SEE P13.55 (b))



(a) MAXIMUM VELOCITY OCCURS AT 3 WHERE THE SPRINGS ARE UNSTRETCHED $T_3 = \frac{1}{2} \text{ m U}_{HAx}^2 = \frac{3}{9} \text{ U}_{MAx}^2 \qquad V_3 = 0$ $T_1 = 0 \qquad V_1 = \frac{1}{2} \text{ ke } \chi_0^2 = \left(\frac{470 \text{ lb/ft}}{2}\right) \left(\frac{1.8 \text{ lm.}}{12 \text{ ln./ft}}\right)^2$

V1= 4.725 1b.ft

 $T_1 + V_1 = T_3 + V_3$ $0 + 4.725 = \frac{3}{3} U_{HAX}^2 + 0$ $U_{HAX}^2 = (\frac{32.2 \text{ ft/s}^3}{3 \text{ lb}})(4.725 \text{ lb.ft}) = 50.715$

(b)
$$T_2 = \frac{1}{2} m v_2^2 = \frac{6}{2q} v_2^2 = \frac{3}{2} v_2^2$$

 $V_2 = \frac{1}{2} k_e V_2^2 = \frac{42016/ft}{2} \left(\frac{.7m.}{12 \text{ in } /ft} \right)^2$
 $V_2 = 0.7146 \text{ ib ft}$

$$T_1+U_1=T_2+V_2$$

$$0+4.725 = \frac{3}{9} V_3^2 + 0.7146$$

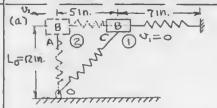
 $V_3^2 = (32.2 \text{ ff/s}^2) (4.010 \text{ lb ff}) = 43.05$

GIVEN:

WB = 10-16
COLLAR B PUSHE
TO RIGHT, &= SIM.
AND RELEASED.
UNDEFORMED
LENGTH OF
EMCH SPRING, LFRM
k=1.6 16/10. FOR
EACH SPRING

FIND:

(a) MAXIMUM VELOCITY, WHAX
(b) HAXIHUM ACCELERATION, QHAX



MAXIMUM VELOCITY OCCURS AT A WHERE THE COLLAR IS PASSING THROUGH ITS EQUILIBRIUM POSITION

POSITION (1) $T_1 = 0$ k = (1.61b/in.)(12 in./ft) = 19.2 lb/ft $L_{0c} = \sqrt{5^2 + 12^2} = 13 in$ $\Delta L_{0c} = 13 in. -12 in. = 110. = \frac{1}{12} ft.$ $\Delta L_{AC} = 5 in. = \frac{5}{12} ft$

V1= 1.733 16.ft

$$\frac{\text{Position}(2)}{T_2 = \frac{1}{2} \text{mU}_2^2} = \frac{1}{2} \left(\frac{10}{9}\right) U_{\text{HAX}}^2 = \frac{5}{9} U_{\text{HAX}}^2$$

V2= 0 (BOTH SPRINGS ARE UNSTRETCHED)

$$T_1 + V_1 = T_2 + V_2$$
 0 + 1.733 = $\frac{5}{9}$ $V_{MAX}^2 + 0$

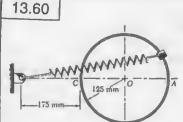
$$V_{MAX}^2 = \frac{(1.733 \text{ lb.ft})(32.2 \text{ f/s}^2)}{(5 \text{ lb})} = 11.16 \frac{\text{ft}^2}{52}$$

$$V_{MAX} = 3.34 \text{ fls}$$

(b) HAXIMUM ACCELERATION OCCURS AT C WHERE THE HORIZONTAL FORCE ON THE COLLAR IS A MAXIMUM

$$\Sigma F = MQ$$
 $F_1 \cos \Theta + F_2 = MQ_{MAX}$
 $R_2 \log \cos \Theta + R_3 \log \Theta = MQ_{MAX}$
 $(19.2 lb/ft) \left(\frac{1}{12} ft \right) \left(\frac{5}{13} \right) + \left(\frac{5}{12} ft \right) = \frac{10 lb}{9} Q_{MAX}$
 $8.615 = \log Q$
 $Q_{MAX} = \frac{8.615 \cdot lb}{(10 lb)} \left(\frac{32.2 ft / 5^2}{10 lb} \right)$

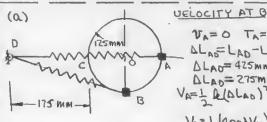
QHAX = 27.7 ft/s2



HASS OF COLLAR m= 1.5 kg. R = 400 N/m. UNDEFORMED LENGTH OF SPRING LO= ISOM COLLAR RELEASED FROH REST AT A

FIND:

(a) VELOCITY OF THE COLLAR AT B, VB () VELOCITY OF THE COLLAR AT C. UC



VA=0 TA=0 ALAS-LAD-LO DLAD= 425mm-150mm DLAD= S75MM=1275M VA=1 R(DLAD)

VA= 1 (400 N/m) (0.275m)2 To= 1 mug= (15 lg) (ve) = (0.75) UB2

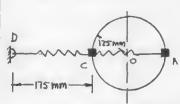
 $L_{BD} = (300^{2} \text{mm} + 125^{2} \text{mm})^{2} = 325 \text{mm}$ $\Delta_{BD} = L_{BD} - L_{0} = (325 \text{mm} - 150 \text{mm}) = 175 \text{mm} = 0.175 \text{m}$

VB= 1 k(DBD)= 1 (400 N/m) (-175m)= 6.125 J

TA+VA=TB+VB 0+15.125=0.7502+6.125

 $V_8 = (15.125 - 6.125) = 17.00 \frac{M^2}{52}$

(b) VELOCITY AT C



TA= 0 VA= 15.125 T (SEE (a))

UB= 3.46 M

Tc= = mv2= = (1.5kg) v2=0.7502

DLoc= Lo-Loc=(150mm-175m)=-25mm

Vc= 1 k(DLoc) = 1 (400 U/m) (-0.025 m)2

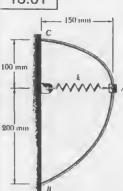
Vc= 0.125 I

TA+VA=Tc+Vc

0+15.125=0.7502+0.125 N2=15/0.75 = 20

Uc= 4.47 m/s





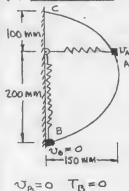
GIVEN:

HORIZONTAL PLANE MASS OF COLLAR M=500-9 UNDEFORHED LENGTH OF SPRING, LO = 80 mm k= 400 le N/m

FIND:

(a) VELOCITY AT A, VA FOR VELOCITY AT B = 0 (b) VELOCITY AT C, UC

(a) VELOCITY AT A



TA = 1 M VA = (0.569) UA TA=(0,25) UA

ALA= 0.150 M-0.080 M ALA = 0.070 m

VA= 1 & (DLA)

VA= 1 (400×103 H/m)(0,070m)

VA= 980 J

DLB= 0.700M-0.080M = 0.120M

VB = = 1 & (ALB) = 1 (400×103 N/m) (0.120 m)2

VB= 2880 J

TA+VA=TB+VO

0.2502+980=0+2880

UA=(2880-980)/0.25)

UA = 7600 M752

UA= 87.2 m/s

(b) VELOCITY AT C



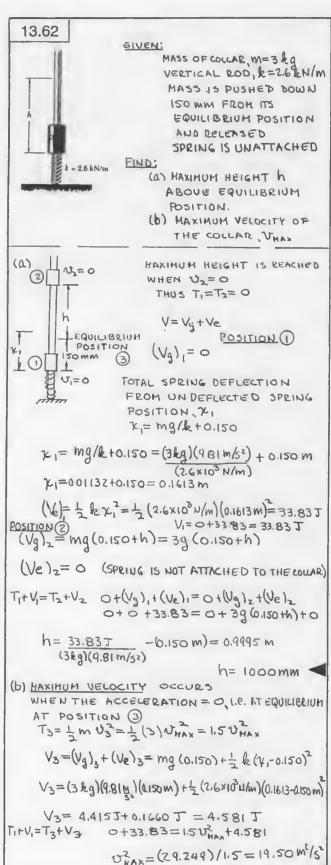
SINCE SLOPE AT B IS POSITIVE THE COMPONENT OF THE SPRING FORCE FP. PARALLEL TO THE ROD, CAUSES THE BLOCK TO HOUS BACK TOWARD A TB=0 , VB= 2880 J (FROH PARTIE)

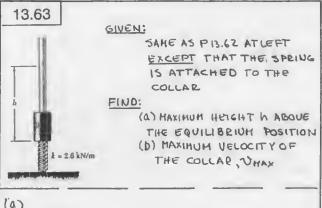
Te= 12 m v2= (0.549) v2=0.25 v2

ALC= 0.100m-0.080m=0.020m Vc= 1 k(OLc) = 1 (400×103N/m/0.020m)=80.0T

TotVo=Tc+Vc 0+2880=0.2542+80.0 Ve2= 11200 m2/52

UC= 105.8 M/5





V,=0, T,=0

SPRING POSITION

POSITION

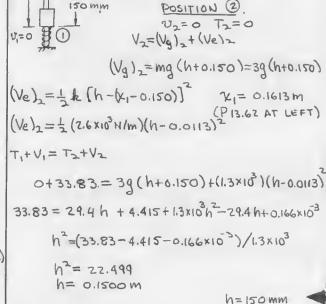
150 mm

POSITION (1)

(SAME AS PIB.62

AT LEFT)

V= 33 83J

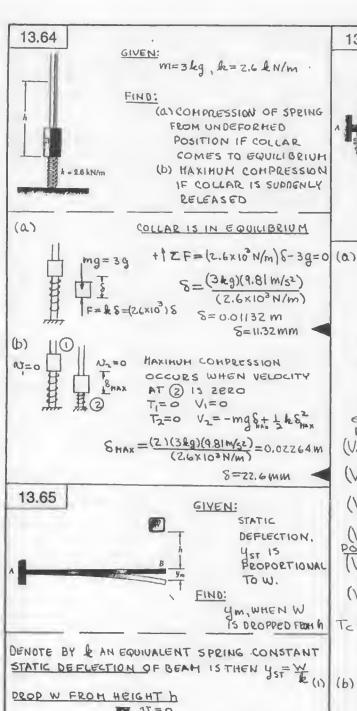


(b) HAXIHUM VELOCITY SEE (b) AT LETT

(2)

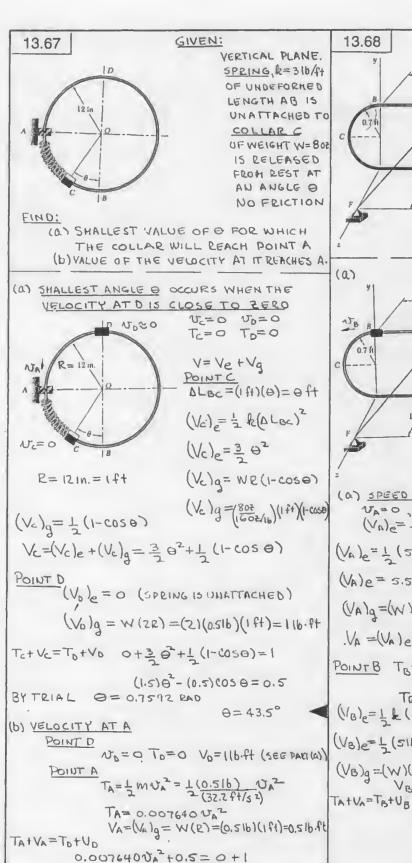
UHAX = 4AZMIS

UHAX = 4.42 M/S



U,=0 B N2=0 Tz=0 Vz=-Wym+= leym Ti=0 Vi= Wh 0+Wh=0-Wym+2kym $T_1+V_1=T_2+V_2$ FROM EQUATION (1), W= kyst. kyst (h+ym)= 1 kym ym -2ysrym-2ysh=0 ym=ysr(1+12n ysr)

13.66 GIVEN: VERTICAL PLANE SPRING, R= 310/ft, UNDEFORMED LENGTH AB, IS UNATTACHED TO COLLAR W= 802. 0=30° (V=0) NO FRICTION FINO: (a) MAXIMUM HEIGHT H ABOUE B REACHED BY THE COLLAR (b) HAXIHUM VELOCITY, UHAY OF THE COLLAR MAXIMUM HEIGHT ABOUE B IS BEACHED WHEN THE VELOCITY AT E IS ZERO Tc = 0 TE=0 V=0 V= Ve + Vg POINTC 0=30°= 17/6 RAD DLBC=(Ift)(IPRAD) P= 1211, = 1 ft DrBC= It t+ (Vc) = 1 le(OLOC) (Vc)e= = (316/ft)(Ift) = 0.4112 16.9t (N°) = MB(1-co20)=(805) (14)(1-co2300) (Ve) g = 0.06699 lb.ft (VE) = 0 (SPRING IS UNATTACHED) $(V_{\varepsilon})_g = WH = \left(\frac{8}{1L}\right)(H) = \frac{H}{2}$ (16.ft) Tc + Vc = TE + Ve 0+0.411Z+0.06699=0+0+H H= 0.956ft (b) THE MAXIMUM VELOCITY IS AT B WHERE THE POTENTIAL ENERGY IS ZERO, UP = NHAN Vc= 0.4112+.06699 = 0.4782 1b.ft To= 12 mV2==1 (1 1b/32.2ft/s2) VHAX TB= 0.07640 V2 VB=0



V2= 64.4 ft3/52

13.68 GIVEN: COLLAR W=2.71 UNDEFORHED LENGTH OF ELASTIC COLD Lo= 0.9ft k = 516/ft VA= 0 FIND: SPEED OF COLLAR LLR (a) AT B (b) ATE (a) LAF= V(1.6)2+(1.4)2+(1.1)2 LAF= 2.394 ft VA= 0 AJB LBF=V(1.4)2+(1.1)2 LBF = 1.780 ft LFE=V(1.6)2+(1.1)2 LFE= 1.942 ft V= Ve + Va (a) SPEED AT B UA=0 , TA=0 VA=0, TA=0 2 POINT A (VA)e=12 R(DLAF) DLAF=LAFLO= 2.394-0.9 (Va)e=1 (516/ft)(1494ft) & DLAF= 1.494 ft (VA)e= 5.580 lb.ft (VA)q=(W)(1.4)=(2.716)(1.4ft)=3.7816.ft .VA = (VA)e+(VA)q = 5.580+3.78 = 9.360 16.ft POINTB TB= 1 MUO= 1 (2.716) VB TB= 0.04193 V62 (1/B)== Lk(DLBF)2 DLB==LBFL=1.780-0.9 ALBF = 0.880 ft (VB)e= 1 (516/ft) (0.880ft) = 1.936 16.ft (VB)q=(W)(1.4)=(2.716)(1.4ft)=3.7816.ft VB=(VB) e+(VB) = 1.936+3.78=5.716 16.ft

0+ 9.360= 0.0419322+5.716

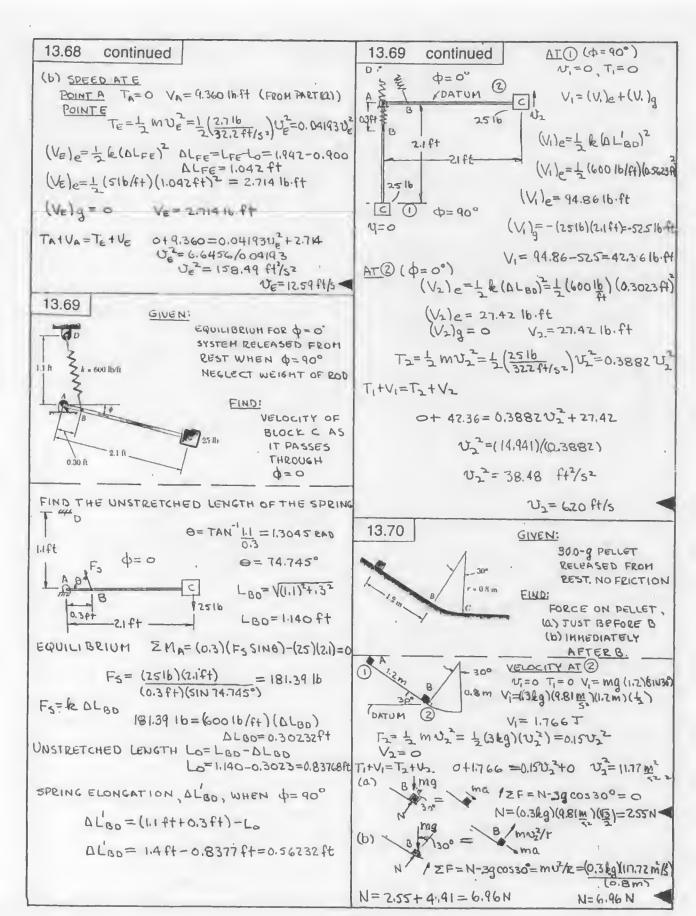
UZ= (3.644) (6.04193)

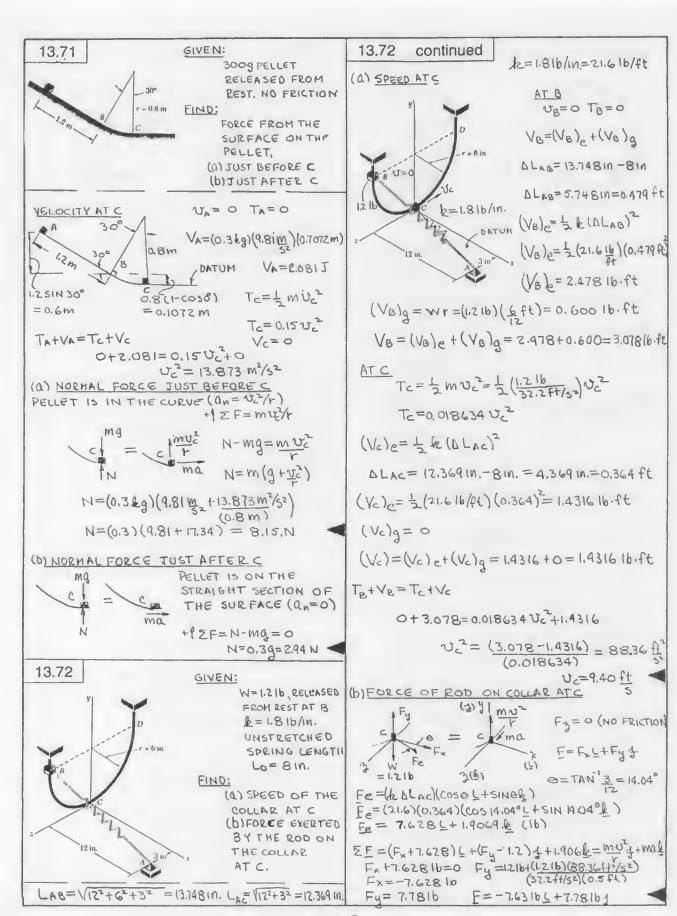
Us= 9.32 ft/s

NB=86.91 A1/52

(CONTINUED)

Un= 8,02 ft/s





13.73

GIVEN:

VERTICAL PLANE

SPRING, &= 31b/ft

UNDEFORMED

LENGTH = ARC AB

UNATTACHED TO

COLLAR.

COLLAR WEIGHT

W= 802.

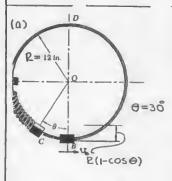
0=30°.

COLLAR RELEASED

FROM REST AT C.

FIND:

(a) YELOCITY AT B, VC (b) FORCE ON THE COLLAR FROM ROD AT B



 $V_c = 0$, $T_c = 0$ $T_0 = \frac{1}{2} m V_B^2$ $T_0 = \frac{1}{2} \frac{(802)}{(1602/16)(32241)} V_0^2$ $T_B = 0.07764 V_B^2$

 $V_c = (V_c)_e + (V_c)_g$ ARC BC = $\Delta L = R\Theta$ $\Delta L_{BC} = (1 ft)(30^3(\pi))$ $\Delta L_{Bc} = 0.5236 ft 180^\circ$

(/c)e=1/2 (DLBC)2

 $(V_c)_e = \frac{1}{2}(31b/f_t)(0.5236ft)^2 = 0.41121b.ft$

(1/c)g=WR(1-cose)=(1802)(1ft)(1-cos30°)

(Vc)g= 0.06699 1b.ft

Vc = (Vc)e+(Vc)g= 0.4112+0.06699=0.4782169

VB = (VB)+(VB)g = 0+0=0

 $T_c + V_c = T_B + V_B$ 0+0.4782=0.07764 V_B^2 $V_B^2 = 61.59 \text{ ft}^2/\text{s}^2$

(b) W=0.51b $mv_8/2$ ma

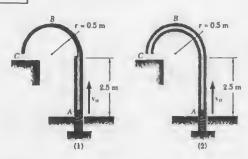
+ = FE-W= mv3/R

 $F_R = 0.5 lb + (0.5 lb) (61.59 ft^2/52) (61.59 ft^2/52)$

FR = 0.516+ 0.9564 1b = 1.456 1b

FR= 1.456 1b

13.74



GIVEN:

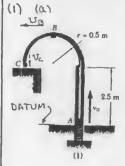
PACKAGE, MASS M= 200-9 INITIAL VELOCITY, US FRICTION LESS TUBE

(1) TUBE IS OPEN ALONG CIRCULAR ARC

FIND:

(a) SHALLEST VELOCITY US FOR PACKAGE TO REACH POINT C

(b) FORCE EXERTED BY THE PACKAGE ON THE TUBE.



THE SHALLEST VELOCITY AT B WILL OCCUR WHEN THE FORCE EXECTED BYTHE TUBE ON THE PACKAGE IS ZERO.

mg = 0.2 g mg = 0.2 g r

+ 1 2F = 0+ mg = m UB

 $V_{B} = g r = (9.81 \text{ m/s}^2)(0.5 \text{ m})$ $V_{B} = 4.90 \text{ 5.} \text{ m}^2/\text{s}^2$

TA = 1 m No VA = 0

 $T_{B} = \frac{1}{2} m V_{B}^{2} = \frac{1}{2} m (4.905) = 2.45.3 m$

VB = mg (2.5 + 0.5) = 3 mg TA+VA=TB+VB

 $\frac{1}{5}$. $m\sqrt{6} + 0 = 2.459m + 3 mg$ $\sqrt{6} = 2[(2.453) + 3(9.81)] = 63.77$

TC= 1 MUZ Vc=mg (25m) Vo= 7.99 m/5

TA+VA=Tc+Vc

1 mv2+0=1 mv2+2.5mg

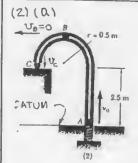
UC= [63.77- (5.0)(9.81)]

Uc2= 14.72 m2/52

(b)

No in the second s

13.74 continued



THE VELOCITY AT B CAN BE NEARLY EQUAL TO ZERO SINCE THE WEIGHT OF THE PACKAGE IS SUPPORTED BY THE TUBE. THUS, Ug=0 TB=0

VB=mg (2.5m+0.5m) VB=3mg

TA= 1 MUo VA= 0

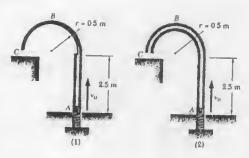
TA+VB=TA+VA

(6) Tc= 1 m vc Vc= mg (2.5 m)

 $T_A + V_A = T_c + V_c$ $\frac{1}{2} m v_o^2 + 0 = \frac{1}{2} m v_c^2 + z.smg$ $v_c^2 = 6g - 5g = 9.81 m^2/s^2$

muc2 -ZF= Nc= MUc2/r Nc=(0.2 kg)(9B/m/s2)/0.5m) PACKAGE ONTUBE, NC= 3.92N-

13.75



GIVEN:

VEYPCITY AT C , < 3.5 m/s (REQUIRED)

EIND:

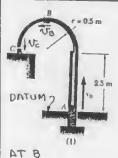
(a) LOOP (2) BUT NOT LOOP (1) CAN SATISFY REQUIREHENT THAT U, <3.5 m/s (D) LARGEST ALLOWABLE VELOCITY TO WHEN LOOP (2) IS USED AND UCC3.5 11/15.

(4) LOOP (1) , THE SHALLEST ALLOWABLE VELOCITY AT B WILL OCCUR WHEN THE FORCE EXERTED BY THE TUBE ON THE PACKAGE IS

+ 2F= 0+mg= mv3/r

 $V_B^2 = 9r = (9.81 \text{ m/s}^2)(0.5 \text{ m}) = 4.905 \text{ m}^2/\text{s}^2$ UB= 2.215 W/S

13.75 continued



THE VELOCITY AT B CANNOT B LESS THAN 2.215M/S IF THE PACKAGE IS TO HAINTAIN CONTACT WITH THE

FOR UCTO BE AS SHALL AS POSSIBLE, VB MUST BE AS SHALL AS POSSIBLE: THAT IS UB= 2,215 m/s TB= 1 m NB=1 m (2.215)2

TB= 2453 M VB= mg(2.5+0.5)= 3mg

TC= - MUZ

Vc = 2.5 mg

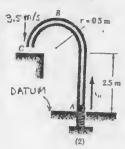
TB+VB=Tc+Vc

2.453 m+3mq = 1 mV2+2.5mg V= 2 (2.453+0.5(9.81m/s2))

152=14.72 m2/52

Uc= 3.836 m/s > 3.5 m/s THUS, LOOP (1) CANNOT HEET THE REQUIRENENT

(b) LOOP (2)



VA= 0

ATC Uz=3.5 m/s Tc=1 m (3.5)2 Tc= 6.125m

Vc= 2.5 mg

TA+1/A=Tc+Vc

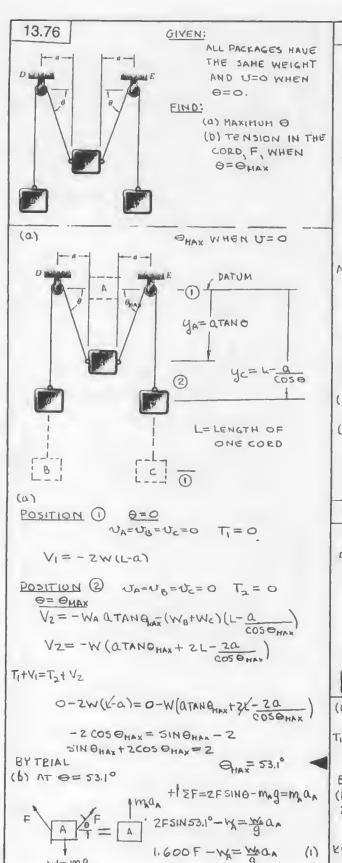
1 m vo2 + 0 = 6.125m +2.5mg

Vo2 = 2 (6.125+2.59) = 61.3 m2/52

Un= 7.83 m/s

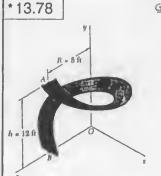
NOTE:

A LARGER VELOCITY AT A WOULD RESULT IN A VELOCITY AT C, GREATER THAN 3.5 W/S



13.76 continued +1 ZF=F-mcq=-mcac (2) meg an=-4A KINEMATICS YA=CITANO LET f (0)=-asec20 ye= aTANO & YA= 1 (0) 0+ fA(0) 0 LET fo= aTANO ÿ=f= 0+f=(0)0 AT OHAX , 0=0 , 0 HAX = 53.1 THUS YA = fa (53.1°) = - a sec2(53.1°) ye f (53.1°) a TAN (53.1°)/0553.1° a= 1,2500c (3) CA = 1.25P REPLACE QAIN (1) BY 1.25.000 FROM (3) WA=WB=WE=W 1.600F-W= W (1.2500c) 1.600 F-W =-1.250 (F-W) 2,850 F= 2.250W F=0.789W 13.77 GIVEN: WA= 216 WB=WC= 316 U=O,WHEN 0=O FIND: (a) HANHUH O (DITENSION F AT OMAX REFER TO FIGURE IN P13.76 (a) AT LEFT (a) 0=0 T=0 V=-(WB+Wc)(L-a)=-6(L-a) Q=QHAX T2=0 V2=-ZQTANQHAX-6 (L-Q COS QHAX) TITVI=TITYZ 0-6(K-a)=0-Zatanq 6K- a cosons -6005 @HAX = 251N BHAX -6 BYTRIAL (b) REFER TO (b) PROB 13.76 2F 51N 36.90-WA= WA QA 1.201F-Z= 3 ax F-3=-3 ac (1) F-Wc=-Weac (2) KINEHATICS QA = 40 = SEC 36.9 TAN 36.9/C0536.9 = 1.665 SOLVE (1),(2), AND (3) FOR F (3)

F= 2.31 16

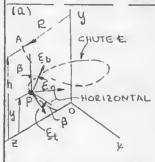


PACKAGES RELEASED FROM REST AT A CHUTE IS BANKED SO THAT PACKAGES DO NOT TOUCH ITS EDGES. NO FRICTION PACKAGE WEIGHT. W=2010. CHUTE IS A HELIX WITH PRINCIPAL NORMAL DUA JATHOSIBOH DIRECTED TOWARD Y DXIS.

FIND:

(Q) ANGLE & FORHED BY THE NORMAL TO THE SURFACE OF THE CHUTE AN THE PRINCIPAL NORHAL (b) MAGHITUDE AN DIRECTION OF THE CHUTE ON THE

PACKAGE AT B



AT POINT A UA= O TA= O

VA= man

AT ANY POINT P Vp=Wy=mgy

En, ALONG PRINCIPAL AND DIRECTED TOWARD 4 axis

TA+VA=Tp+Vp NORMAL, HORIZONTAL O+mgh=1mv2+mgy v= 29(h-4)

EE, TANGENT TO CENTERLINE OF THE CHUTE

ED, ALONG BINORMAL

3 - TAN = TAN (12 ft) 211(8 ft) B= 13.427°

mab=0 SINCE QUE O

NOTE: FRICTION IS ZERO

EFL = mat masing = mat at gsing

ZFb= mab No-WCOSB=O NB=WCOSB

ZFn=man Nn=mu=mzq(n-y)=zw(h-y)

THE TOTAL NORMAL FORCE IS THE RESULTANT OF No AND NM, LIES IN THE b-M PLANE AND FORMS ANGLE & WITH MAXIS.

* 13.78 continued

17-717 B tan = Nb/Nn tanp= wcos B/2(wh-4) tand= (P/2(n-y)cosB

GIVEN: C= R[1+(1/2)]=R(1+tanp)=R

THUS: $\tan \phi = \frac{e}{2(h-y)} \cos \beta = \frac{R}{2(h-y)\cos\beta}$

tan φ = 8 ft / 2(12-y) cos μ.327° = 4.113

OR COT \$ = 0.743(12-4)

(b) At POINT B y= O FOR X, y, Z AXES WE WRITE, WITH W= 2016 Nx= Nb 51NB= WCOSBSINB=(2016)COS 14.327°SINA.327 Nx= 4.517 lb

Ny= No cosβ= Wcos2β= (2016) cos214.3276 Ny=18.922 16

Nz= - Nn=-2W N-4=-2 W h-4

Nz= 2 (2016) (12ft-0) cos214.327° Nz= -56.76516

N=V(4.517)2+(18.922)2+(-56.765)2 N=60.016

COS 0x = Nx = 4.517

0x=85.7°

cos ey = Ny = 18.922

Oy= 71.6

 $\cos \theta_2 = \frac{N_2}{N} = -\frac{56.742}{60}$

Oz=161.1°

13.79 GIVEN:

> F(1,4,2) IS CONSERVATIVE SHOW THAT:

> > $\frac{\partial F_y}{\partial F_y} = \frac{\partial F_y}{\partial F_y}$ $\frac{\partial F_y}{\partial z} = \frac{\partial F_z}{\partial y}$

FOR A CONSERVATIVE FORCE, EQ (13.22) HUST BE SATISFIED

 $F_x = -\frac{\partial V}{\partial x}$ $F_y = -\frac{\partial V}{\partial y}$ $F_{z} = -\frac{\partial V}{\partial z}$

WE NOW WRITE DE = DE

FINCE DZV $rac{3}{2} = \frac{3}{3} \frac{3}{2} \times \frac{3}$

DFX - DF4
DY DX

WE OBTAIN IN A SIMILAR WAY

2F4 = 2F2

*13.80

GIVEN:

F=1924+2x3+xy\$1/xyz

SHOW:

(9) F IS a CONSERVATIVE FORCE

EIND:

(b) THE POTENTIAL FUNCTION ASSOCIATED WITH E

(a)
$$F_x = \frac{yz}{xyz}$$
 $F_y = \frac{zx}{xyz}$
 $\frac{\partial F_y}{\partial y} = \frac{\partial U(x)}{\partial y} = 0$ $\frac{\partial F_y}{\partial x} = \frac{\partial U(y)}{\partial x} = 0$
Thus $\frac{\partial F_x}{\partial y} = \frac{\partial F_y}{\partial x}$

THE OTHER TWO EQUATIONS DERIVED IN PROB. 13. BO ARE CHECKED IN A SIMILAR WAY

(b) RECALL THAT
$$F_x = -\frac{\partial V}{\partial x}$$
, $F_y = -\frac{\partial V}{\partial y}$, $F_z = -\frac{\partial V}{\partial z}$

$$F_x = \frac{1}{x} = -\frac{\partial V}{\partial x}$$
 $V = -\ln x + f(y,z)$ (1)

$$F_y = \frac{1}{y} = \frac{\partial V}{\partial y} V = -lny + g(z_1x)$$
 (2)

$$F_2 = \frac{1}{2} = -\frac{2}{2}$$
 $V = -\ln 2 + h(x,y)$ (3)

EQUATING () AND (2)

$$-lnx+f(y,z)=-lny+g(z,x)$$

THUS
$$f(y,z) = -\ln y + k(z)$$
 (4)
 $g(z,x) = -\ln x + k(z)$ (5)

EQUATING (2) AND (3)

$$-\ln z + h(x,y) = -\ln y + g(z,x)$$

FROM (5)
$$g(z,x) = -lmz + l(x)$$

 $g(z,x) = -lmx + l(z)$

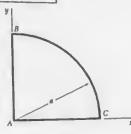
THUS

$$l(x) = -lnx$$

FROM (4)

SUBSTITUTE FOR f(4,2) IN (1)

* 13.81



GIVEN:

PARTICLE P(X,4)
ACTED UPON BY FORCE

FIND:

WHETHER F IS A CONSERVATIVE FORCE, AND COMPUTE THE WORK OF F WHEN P(x,y) DESCRIBES A PATHABOCA, CLOCKWISE FOR, (a) F=ky L
(b) F=k(yL+x1)

(a)
$$F_x = ky$$
 $F_y = 0$ $\frac{\partial F_x}{\partial y} = k$ $\frac{\partial F_y}{\partial x} = 0$
Thus $\frac{\partial F_x}{\partial y} = \frac{\partial F_y}{\partial x} = 1$ Is NOT CONSERVATIVE

S= 0, F IS PERPENDICULAR TO THE PATH

FROM BTOC THE PATH IS A QUARTER CIRCLE WITH ORIGIN AT A.

THUS x2+y2= 02

ALONG BC
$$\int dx dx = \int dx \sqrt{\alpha^2 - x^2} dx$$

$$= \frac{\text{Tka}^2}{4}$$

$$\int ky \cdot dx = 0 \quad (y = 0 \text{ on } cA)$$

$$U_{ABCA} = \int_{A}^{B} + \int_{B}^{C} + \int_{C}^{A} = 0 + \frac{\pi ka^{2}}{4} + 0$$

$$U_{ABCA} = \frac{\pi ka^{2}}{4}$$

SINCE ABOA IS A CLOSED LOOP AND F IS CONSERVATIVE,

UABCA = 0



POTENTIAL
FUNCTION
V(x,4,2)=-(X²+4²+2²)^{1/2}
ASSOCIATED
WITH FORCE P.

EIND:

(a) xy, 2

components

of P

WORK DONE
BY P FROM O TO D BY
INTEGRATING ALONG
THE PATH OABD, UGABB
SHOW THAT UGABG -AV

(p)
$$A = -\frac{\partial x}{\partial A} = -\frac{\partial x}{\partial -(x_5 + A_5 + S_5)_{1/5}} = \frac{\partial x}{\partial A} = -\frac{\partial x}{\partial -(x_5 + A_5 + S_5)_{1/5}} = \frac{\partial x}{\partial A} = \frac{\partial x}{\partial A} = -\frac{\partial x}{\partial -(x_5 + A_5 + S_5)_{1/5}} = \frac{\partial x}{\partial A} = \frac{\partial$$

O-A Py AND Px ARE FERPENDICULAR TO O-A
AND DO NO WORK
1-150, ON O-A X=y=0 AND Pz=1
THUS UO-A= Pz dz= fdz = a

A-0 P_z AND P_y ARE DERPENDICULAR TO A-B AND DO NO WORK 1-LSO ON A-B y=0, z=a AND $P_x=x/(x^2+a^2)^{1/2}$

THUS
$$V_{A-8} = \int_{0}^{\infty} \frac{x dx}{(x^2 + 3^2)^{1/2}} = \alpha (\sqrt{2} - 1)$$

B-D PX AND PZ ARE PERPENDICULAR TO

S-D AND DO NO WORK

OH B-O X=a, Z=a Rg=y/(y²+za²)/2

THUS
$$U_{80} = \int_{0}^{4} \frac{y}{(y^2+za^2)^{1/2}} dy = (y^2+za^2)^{1/2} \int_{0}^{a} U_{80} = (a^2+za^2)^{1/2} (20^2)^{1/2} a (15-15)$$

$$\Delta V_{00} = V(a,a,a) - V(o,o,o) = -(a^2 + a^2 + a^2) \frac{1}{2} o$$

13.83

REFER TO FIG. PIB. BZ ON THE LEFT

GIYEN:

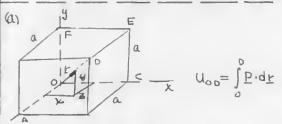
FROM SOLUTION TO (a) OF PROB. 13.82 $P = \frac{x_1 + y_2 + z_2^2}{(x^2 + y^2 + z^2)} V_2$

FIND:

(4) WORK DONEBY P AWNG THE DIAGONAL

VERIFY:

(b) THAT WORK DONE AROUND THE CLOSED PATH OABDO IS ZERO.



ALONG THE DIAGONAL X=4=2

THUS
$$P: dr = \frac{3x}{(3x^2)} V_2 = \sqrt{3}$$

$$U_{0:0} = \int \sqrt{3} dx = \sqrt{3} a$$

(b) USABDO = UOABD+ UDO

FROM PEOB 13.82

UDABO = V3 2 AT LEFT

THE WORK DONE FROM DTO O ALONG THE DIAGONAL IS THE NEGATIVE OF THE WORK DONE FROM OTOD

THUS 40ABDO = 130 - 130 = 0

* 13.84

GIVEN: E= (x+4++2k)/(x2+42+22)3/2

PROYE:

(a) F IS CONSERVATIVE

FIND:

(DITHE POTENTIAL FUNCTION V(X,Y,Z) ASSOCIATED WITH F

(a)
$$F_x = \frac{1}{2} ((x^2 + y^2 + z^2)^{3/2}) F_y = \frac{1}{2} ((x^2 + y^2 + z^2)^{1/2})$$

$$\frac{\partial F_{x}}{\partial y} = \frac{\chi(-\frac{3}{2})(2y)}{(x^{2}+y^{2}+z^{2})^{\frac{3}{2}}} \frac{\partial F_{y}}{\partial x} = \frac{y(-\frac{3}{2})2y}{(x^{2}+y^{2}+z^{2})^{\frac{3}{2}}}$$

THE OTHER TWO EQUATIONS DERIVED IN PROB. 13.79 ARE CHECKED IN A SHILLAR FASHION

(b) RECALLING THAT
$$F_x = -\frac{\partial V}{\partial x}$$
, $F_y = -\frac{\partial V}{\partial y}$, $F_z = -\frac{\partial V}{\partial z}$

$$V = (x^2 + u^2 + 2^2)^{\frac{1}{2}} + f(u,z)$$

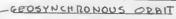
THE UNKNOWN FUNCTION f(x,y) IS A CONSTANT

13.85 GIVEN:

3600-RY LAUNCHED FROM A CIRCULAR ORBIT AT BOOK M ABOVE THE EARTH. ALTITUDE OF GEOSYNCHRONOUS (CIECULAR) ORBIT = 35770 km

FIND:

- (a) ENERGY NEEDED TO PLACE THE SATELLITE INTO GEOSYNCHRONDUS ORBIT . FROM 300 RM
- (b) ENERGY NEEDED TO PLACE THE SATELLITE INTO A GEOSYNCHRONOUS ORBIT FROM THE EARTH LEXCLUDE AIR RESISTANCE



12 = 6370 km + 35770 km = 42.140 x10 m

ORBIT AT 300 km

r= 6370km+300km= 6.67x10°m

Re=6370 Rm

FOR ANY CIRCULAR OPENT OF PADIUS & THE TOTAL ENERGY E = T+V = 1 mu2 - GMM

M= MASS OF THE EARTH M=3600 lg = SATELLITE HASS 13.85 continued

NEWTON'S SECOND LAW F=man: GMm = mu2

 $T = \frac{1}{2} m v^2 = \frac{M}{2r}$

E=T+V= 1 GMM - GMM = 1 GMM

GN=gRE E=-1 9REM

E=-1 (9.81 m/52)(6370×103m)(3600 kg)

E= - 716.15 × 10 (N·m)

FOR A GEOSYNCHRONOUS ORBIT (1=42,140 x10 m)

 $E_{GS} = \frac{-716 \times 10^{15}}{42.140 \times 10^{6}} = +7.003 \times 10^{9} = -17.003 \, GJ$

(a) AT 300 Em (r= 6.67 x10 m)

E300 = -716×1015 = +07.42×109 J=-107.42 GJ

ADDITIONAL ENERGY AE300 = EGS - E300

DE300 - 17.003+107.42

ΔE300=90.46J◀ (b) LAUNCH FROM THE EARTH (R=6370 Rm)

AT LAUNCH PAD E = V = - GHM = - 9 REM E== - (9.81m/52) (6370×103m) (3600 Ag)

EE= - 224.96×109 J = -224.96 GJ

ADDITIONAL ENERGY DE== E - ES

DEE= -17.003+224.96=2086J€

13.86

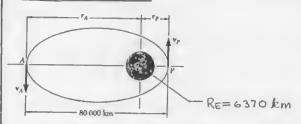
4/Up=Tp/TA ra+ro=80000 km ELLIPTIC ORBIT

FIND:

ENERGY PER UNIT MASS E/M REQUIRED TO PLACE THE SATELLITE IN ORBIT.

DETERHINE THE TOTAL ENERGY PER UNIT HASS FOR THE ELLIPTIC ORBIT AND SUBTRACT FROM IT THE ENERGY PER UNIT MASS ON THE EARTH TO GET THE ENERGY PER UNIT HASS NEEDED FOR PROPULSION. (EXCLUDING, AIR RESISTANCE, THE WEIGHT OF THE BOOSTER ROCKET AND HANEUVERING.)

13.86 continued



TOTAL ENERGY PER UNIT MASS FOR THE ORBIT

$$E_0 = T_A + V_A = T_P + V_P$$

$$E_1/M = \frac{V_A^2 - GM}{2} - \frac{U_P^2}{r_A} = \frac{U_P^2 - GM}{2} - \frac{GM}{r_P}$$

$$V_A^2 \left(1 - \frac{V_P^2}{V_A^2}\right) = 2GM \left(\frac{1}{r_A} - \frac{1}{r_P}\right)$$
(1)

$$V_{A}/U_{p} = r_{P}/r_{A} \quad (GIVEN)$$

$$V_{A}^{2}\left(1 - \frac{r_{A}^{2}}{V_{P}^{2}}\right) = ZGM\left(\frac{r_{P} - r_{A}}{r_{A}r_{P}}\right)$$

$$V_{A}^{2}\left(\frac{r_{P} - r_{A}}{V_{P}}\right)\left(\frac{r_{P} + r_{A}}{r_{A}r_{P}}\right) = ZGH\left(\frac{r_{D} - r_{A}}{V_{A}V_{P}}\right)$$

$$U_A^2 = 2GM \frac{r_D}{r_A} \left(\frac{1}{t_D + r_A} \right)$$
 (2)

SUBSTITUTING UAIN(2) IN (1)

$$E_0/M = GH \frac{1}{r_A} \left[\frac{r_P - (r_P + r_A)}{r_P + r_A} \right] = -\frac{GM}{r_P + r_A}$$

GM= 9 Pe2 = (9.81 m/s2) (6370 x 103 m)

VP+ TA = 80.000 × 103 m (GIVEN)

$$E_0/m = \frac{-(4.81 \text{ m/s}^2)(6370 \times 10^3 \text{ m})^2}{80000 \times 10^3 \text{ m}}$$

Edm = 4.9765 ×106 N-m =-4.9765 MJ

TOTAL ENERGY PER UNIT MASS ON THE EARTH

$$E_{e} = T_{e} + V_{E}$$
 $V_{e} = 0$ $T_{e} = 0$ $V_{e} = -\frac{mGM}{P_{e}}$

$$E_{e}/m = -\frac{9P_{e}^{2}}{P_{e}} = -\frac{(9.81 \text{ m/s}^{2})(6370 \times 10^{3} \text{ m})}{P_{e}}$$

$$E_{e}/m = -62.490 \times 10^{6} \frac{N-m}{M} = -62.49 \text{ HT/kg}$$

$$E_{e}/m = -62.490 \times 10^{6} \frac{N-m}{M} = -62.49 \text{ HT/kg}$$

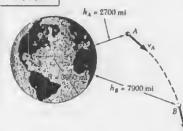
$$E_{e}/m = -62.490 \times 10^{6} \frac{N-m}{M} = -62.49 \text{ HT/kg}$$

ENERGY PER UNIT HASS NEEDED FOR PROPULSION, Ep/m = Eo/m - Ee/m

Ep/m=-4.9765 HJ/kg+62.490 HJ/kg

Ep/m = 57.5 115

13.87



GIVEN:

he AND he 17 = 50 5× 103 MT

FIND:

ra= hat R = 2700 m 1 + 3960 m 6 rA = 6660 mL - 10=h8+R=7900ML+3960M B 1860 Mc

AT A, UA = 20.2×103 mc = 29627 ft/s TA= 1 m (29627ft)= 438.87x106m VA = - GMM = - 9 Rm. R= 3960 mc = 20.909 x 106 ft VA= - (32.2 ft/52) (20.909 × 106ft) 2 =-400.3 × 106 m

TB= 1 m UB VB=-GMM -- QRM

> TB = 11860 ML = 62.621×106ft $V_B = -\frac{(32.2 \text{ ft/s}^2)(20.909 \times 10^6)^2 \text{ m}}{(62.621 \times 10^6 \text{ ft})}$

Va=-224.8 x 106 m

TA+ VA= TB+VB

438.87×106m-400.3×106m= -muy 2-224.8×106m UB= 2[43887X10-400.3X10+224.8X106]

UB2 = 526.75 × 106 ft2/52

Un= 22.951×103 ft/5= 15.65×103 mc/h

VB= 15.65×103mi

13.88 GIVEN: LUNAR EXCURSION MODULE (LEM) FIND: ENERGY PER POUND NEEDED TO ESCAPE MOON'S GRAVITATIONAL FIELD STARTING FROM (a) MOON'S SURFACE (b) CIECULAR ORBIT SOML. ABOVE THE MOON'S SURFACE NOTE: GM HOON = 0.0123 GM EARTH BY EQ. 12.30 GM MOON = 0.01239 RE AT @ DISTANCE FROM MOON; 12=0, ASSUME U2=0 E2= T2+V2= 0- GMMM = 0-0= 0 (a) ON SURFACE OF MOON PH= 1081 ML = 5.7077x10 ft =0 R==3960ml=20.909x10ft E=T,+V=0-0.0123 g R=m V,=0 T,=0 V=-GMMM E,=-(0.0123)(32.2 ft/s2) (20,909×10 ft) m (5.7077×106ft) WE = WEIGHT OF LEH ON THE EARTH E1=(-30.336 x10 ft?)m m= We E= (-30.336 x 106 ft 3/5) WE Δ E = E2-E1 = 0+(942.1×103f+16) We ΔE = 942×10 flb ENERGY PER POUND: (b) r= RH+50ML r,=(1081 mL+50mL)=1131mL=5.9711x10 ft OES NEWTON'S SECOND LAW: VI = GMm TI=1mVI=1mGMm E1= - 1 GMmm = -1 0.0123 9 R2 m

E=-1 (0.0123) (32.2 ft/s2) (20,909 x10 ft) m E1 = (14.498×10 ft2/52) WE = 450.2×10 ft.16 WE ΔE= E2-E1= 0+450.2×103 ft.16 WE

ΔE = 450 × 103 ft.16

ENERGY PER POUND

13.89 GIVEN:

> SATELLITE OF HASS M CIRCULAR ORGIT OF PADIUS Y ABOUT EACH

FIND:

(a) ITS . POTENTIAL ENERGY (b) ITS KINETIC ENERGY (C) ITS TOTAL ENERGY

(a) POTENTIAL ENERGY V= - EMM = - qem + constant

CHOOSING THE CONSTANT

(cf Eq 13.17)

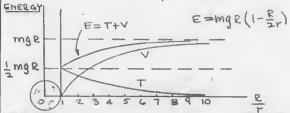
SO THAT V= O FOr= R:

V= mgR(1- R)

P) KINETIC ENERGY

NEWTON'S SECOND LAW F=man: GMM=mv Va CM - GR

(C) TOTAL ENERGY E=T+V = = = m &P2+mg (1- P)



13.90 GIVEN:

SATELLITE IN A CIRCULAR ORBIT

FIND:

ENERGY REQUIRED TO PLACE IT INTO ORBIT-AT (a) 600 km, (b) 6000 km

BEFORE LAUNCHING: 1= R= 6.37x10 m; U,= 0 E = T + V = 0 - GMM = - 9R2M = - mgR

IN CIRCULAR ORBIT OF PADIUS 12: [cf. EQ 12.30]



NEWTON'S SECOND LAW F= man GHM = M 022. V2 = GM = 982

E2=T2+V2= 1 m 02 GHM

E2 = 1 M 9P2 - 9E2M = - 7 9E2M

DE=E2-61= - 1 2 2 2 - (-mg P) = Rmg(1-12 ENEEGY DER KC 15

DE/M = Rg (1-R) (a) 12=6370+600=6970 km

DE/M = (6.37×106)(9.81)(1-(6370))= 33.9 MJ (b) r2= 6370+6000= 12370 km

ΔE/M=(6.37×106)(9.81)(1-6370) = 46.4 NT

13.91

GIVEN:

EQ (13.17'), Vg= - WRZ DISTANCE ABOVE CARTH'S SURFACE, Y

SHOW:

(a) Vg = Wy (FIRST ORDER APPROXIMATION

DERIVE

(b) A SECOND ORDER APPROXHATION

$$V_{g} = -\frac{WR^{2}}{r} \quad \text{SETTING} \quad r = R + y : V_{g} = -\frac{WR^{2}}{R + y} - \frac{WR}{1 + \frac{y}{2}}$$

$$V_{g} = -WR \left(1 + \frac{y}{R} \right)^{-1} - WR \left[1 + \frac{(1)}{1} \frac{y}{R} + \frac{(1)(1+y)}{1 \cdot 2} \left(\frac{y}{R} \right)^{2} + \cdots \right]$$

WE DOD THE CONSTANT WR, WHICH IS EQUIVALENT TO CHANGING THE DATUM FROM 1=00 TO TER:

(a) FIRST ORDER ARPROXIHATION:

(b) SEZOND ORDER APPROXIHATION:

$$\sqrt{d} = MA - MA_{S}$$

$$\sqrt{d} = MA - MA_{S}$$

13.92

GIVEN CELESTIAL BODY IN CIRCULAR ORBIT rADIUS 1= 60 LIGHT YEARS

VELUCITY U= 1.2×106 Mc/h ABOUT A POINT OF MASS, MB

FIND:

PATIO MB/MS, WHERE MS IS THE HASS OF THE SUN

1 v= 1.2 × 10 m1/h= 1.76 × 10 ft/s

Y= GO LIGHT YEARS

I LIGHT YEAR IS THE DISTANCE TRAVELED BY LIGHT IN ONE YEAR SPEED OF LIGHT = 186,300 ML/s

+= (60 YR)(186,300 ML)(5280 ft) (365 DAYS) (24 h) (3600 S)

r= 1.8612×10 18 ft

MB F= GMBM = MUZ F= GMBM = MUZ MB

 $M_B = \frac{rv^2}{G}$ $GN_{ENRTH} = gR_{ENRTH}^2 = (322ft)(3960 MLX5280ft) = 4.017 XIII$

Moun=330,000 ME: CHOUN=330,000 GHEARTH G1150N=(330,000 X140771015) = 4.645×1021 ft3/52

G= 4.645X10 /Msuu

 $M_{0} = \frac{rv^{2}}{6} = rv^{2}M_{50N}/4.695X10^{3}$ $M_{0} = \frac{rv^{2}}{6} = rv^{2}M_{50N}/4.695X10^{3}$

13.93

GIVEN:



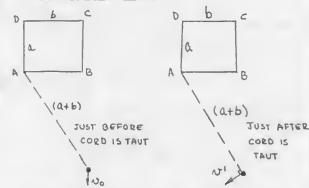
FRICTIONLESS PLATE FIRMLY ATTACHED TO A HORIZONTAL PLANE CORD ABC ATTACHED TO THE PLATE AT A AND TO A SPHERE ATC Un= INITIAL VELOCITY OF SPHERE CAUSES IT TO HAKE A COMPLETE CIRCUIT AND RETURN

FIND:

VELOCITY OF THE SPHERE AS IT STRIKES CIF (a) to 15 PARALLEL TO BC

(b) TO IS PERPENDICULAR TO BC.

IA) No PARALLEL TO BC



ANGULAR HOHENTUM IS CONSERVED ABOUT A

$$bv_o = (a+b)v'$$
 $v' = \frac{bv_o}{(a+b)}$

AS THE SPHERE CONTINUES ITS CIRCUIT TO POINT C ITS VELOCITY IS ALWAYS PERPENDICULAR TO THE CORD AND ENERGY IS CONSERVED THUS V= V'

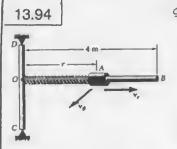
D= 10+ P)

(b) No PERPENDICULAR TO BC

AS THE SPHERE HAKES A COMPLETE CIRCUIT AROUND THE RATE ITS YELOCITY IS ALWAYS PERPENDICULAR TO THE CORD AND ENERGY IS CONSERVED

THUS V=V.

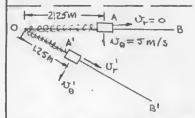
Vc=Vn



k= 750 N/m UNDEFORMED SPRING LENGTH, TO = 1.5M COLLAR HASS, M=2.4/10 INITIALLY, r=225m, Up=5m/s Vr=0

FIND:

U' AND U' WHEN r= 1.25m



CONSERVATION OF ANGULAR MOHENTUM (ABOUT O)

(2.25 m)(m)(5 m/s) = (1.75 m)(H)(V')

NO FRICTION

CONSERVATION OF ENERGY T+V=T'+V'

T= 30.0 J

V= 1 (r-r0)= 1 (750 N/m) (2.25m-1.5m)

V= 210,9 J

V= 9.00 m/s UL

T'= 1.2 U12+ 97.2

V'= 1 1 (r'-ro)= 1 (750 H/m) (1.25m-1.5m)

V'= 23.44 J

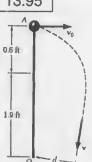
T+V= T'+V'

30+210.9=1.20+97.2+23.44

V'= 10.01 m/s

U;= 10.01 M





GIVEN:

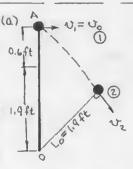
ELASTIC CORD FIXED AT O k= 101b/ft UNDEFORMED LENGTH, LO=19#

WEIGHT OF BALL W=1.516 HORIZONTAL FEICTIONLESS PLANG INITIAL VELOCITY VO

PERPENDICULAR T OA

FIND:

(A) SMALLEST ALLOWABLE UD IF CORD DOES NOT BECOHE SLACK (b) CLOSEST DISTANCE & FOR U EQUAL TO HALF VALUE FOR UD FOUND IN (A)



THE CORD WILL NOT GO SLACK IF No 15 PERPENDICULAR TO THE UNDEFORMED CORD LENGTH LO. AT @

CONSERVATION OF ANGULAR HOHENTUM

POINT () U,= V. T,= 1 W U. = 0.7 U.

V= 1 le (L-Lo)= 1 (10 16/14) (2.5 ft-1.9ft)2

V1= 1.80016.ft

POINT @ T2= 1 4 02 = 9002

AL=0

 $T_1+V_1=T_2+V_2$ 0.75 No +1.800 = 0.75 N2 +0

FROM CONS OF ANG HOH V2=1.3158V0

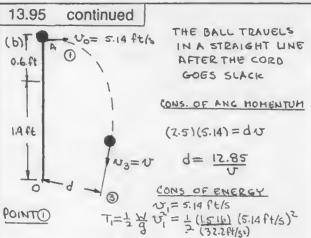
0.75 002 ((1.3128)2-1] = 1.800

 $O_0^2 = (\frac{1.8 \text{ (b.ft)}(32.2 \text{ ft/s}^2)}{(0.75 \text{ lb})(0.7313)}$

Vo= 105.67 だる

Vo= 10.28 ft

(CONTINUED)



Ti= 0.6154 ft.16

Vi= 1 k(1-10)= 1/2(1016/ft)(2.5ft-1.9ft)=1.80016ft

POINT (3)
$$T_3 = \frac{1}{2} \frac{1}{9} U_3^2 = \frac{0.75}{9} U^2$$

$$V_3 = 0$$

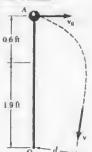
$$T_1 + V_1 = T_3 + V_3$$

0.6154+1.800=0.75v3+0

V= 10.18 ft/s

FROM CONS. OF ANG HOH

13.96



ELASTIC COED FIXED AT O R= 10 1b/ft UNDEFORMED LENGTH LEIGH WEIGHT OF BALL, W= 1516 HORIZONTAL FRICTIONLESS DIANE NO DERDENDICULAR TO OA d= 0.8ft AFTER CORD BECOMES SLACK

FIND:

(a) INITIAL SPEED UO (b) HAYIMUM SPEED, W.

(a) · (1) 060 1.9.6

CONSERVATION OF ANGULAR HOHENTUH ABOUT O

2.5 Vo=0.8V V= 3.125 V CONSERVATION OF ENERGY

V=V0 T=1 4 V0= 0.75 V0

V1= 1 & (LiLo)= 1 (10 16/ft) (2.5ft-19ft)

V1 = 1.800 lb.ft

13.96 continued

POINT @ N2=V T2= 12 W V2 0.15 V2 V2= 0 (CORD IS SLACK)

 $T_1+V_1=T_2+V_2$

0.75 10 +1.800 = 0.75 12 +0

FROM CONS, OF ANG. HOH. , V=3.125 VO

0.75 00 [(3.125)-1]=1.800

Vo= (1.800 16.ft) (32.2 ft/52)

12= 8.816 H2/52

U= 2.97 ft (b) MAXIMUM VELOCITY OCCURS WHEN THE BALLS IS AT ITS HINIMUM DISTANCE FROM O (WHEN d=0.8ft)

Um = 3.125 U0 = (3.125) (2.97) = 9.28 (1/5 Um= 9.2811

13.97



GIVEN:

SPHERE OF MASS, M=0.6 FORCE BETWEEN A ANDO DIRECTED TOWARD O OF MAGNITUDE F=(80/F) N VA= ZOM/S HORIZONTAL FUCTIONLESS PLANE

FIND:

(a) MAXIMUM AND MINIMUM DISTANCES FROM O (b) CORRESPONDING VALUES OF THE SPEED

(0)

THE FORCE EXERTED ON THE SPHERE PASSES THROUGH O. ANGULAR HOHLINTUM ABOUT O IS CONSERVED HINIMUM VELOCITY IS AT B WHERE THE DISTANCE FROM O IS WAXIHUM HAXINUM VELOCITY IS A C WHERE DISTANCE FROM O IS MINIHUM KAMUASINGO= KMMUM (0.5 m)(0.6kg)(20 m/5)(1 N60= rm(0.6kg)(4) Vm= 8.66/rm (1)

CONSERVATION OF ENERGY

TOSI = (& MOS) (\$240.0) = 1201

V= (Fdr= 180 dr= -80 VA= -80 =-160)

AT DOINT B TO= 1 m Um= 1 (0.6kg) Um= 0.3 Um (AND POINT C) VB= -80

TAT VA = TAT VB

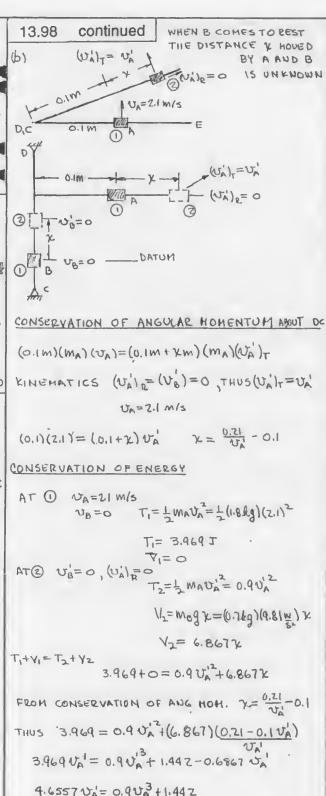
SUBSTITUTE

(1) INTO (2)

120-160=030前- 學 (2) -40 = (0.3) (8.66) 2-80

1m - 2 m + 0.5625= 0 (CONTINUED)

13.97 continued rm= 0.339m AND Tm= 1.661m VHAX 1.661 M YHIN = 0.339 M (b) SUBSTITUTE I'M AND I'M FROM RESULTS OF PART (A) INTO (I) TO GET CORRESPONDING HAXIHUH AND HINIHUM VALUES OF THE SPEED N' = 8.66 = 25.6 M UHA = 25.6 M UMF 8.66 = 5.21 M UHIN= 5.21 14 13.98 GIVEN: ma= 1.8 kg MB= 0.7 kg INITIALLY U = 2.1m AND UB= 0 A STOP IS SUDDENLY REMOVE AT B FIND: (a) VA, WHEN MA 15 0.2 m FROM 0 (b) V', WHEN V=0 (a) U= 21 m/5 CONSERVATION OF ANGULAR HOHENTUH ABOUT DO (0.1m)(ma)(va)=(0.2m)(ma)(va)_ $(v_A^i)_T = (\frac{0.1}{0.2})(2.1)_T = 1.05 \text{ m/s}$ CONSERVATION OF ENERGY 1) UA= 2.1 M/S T=1 (1.8 kg) (2.1 M/S)= 3.969 J VB= O, CHOOSE DATUM FOR B AT ITS INITIAL POSITION AND NOTE THAT THE POTENTIAL ENERGY OF A DOES NOT CHANGE THUS WE TAKE VIO @ (Va) = 1.050 m/s (Va) = Va (KINEMATICS) T= 1 ma [(va) 2+(va) 2) + 2 mg (vg)2 Tz = 1/1.8 kg) [(1.050 m/s) + (V'A) =] + 1/2 (0.7 kg) (V'A) = Tz = 0.9923 + 1.25(V1)2 V= mog (0.1 m) = (0.7kg)(4.81 m/s2)(0.1m)=0.6867 J T+V=T2+Y2 3.969+0=0.9923+1.25(VA)2+0.6867 $U_{A}^{1} = V(U_{A}^{1})_{T}^{2} + (U_{A}^{1})_{R}^{2} = [(1.05)^{2} + (1.354)^{2}]_{T}^{2} = 1.713\frac{M}{3}$ $U_{A}^{1} = V(U_{A}^{1})_{T}^{2} + (U_{A}^{1})_{R}^{2} = [(1.05)^{2} + (1.354)^{2}]_{T}^{2} = 1.713\frac{M}{3}$ $U_{A}^{1} - 26$ $\theta = TAN (V_{A}^{1})_{R}^{2} = TAN (1.354)^{2} = 37.8^{\circ}$



BY TRIAL

5.173VA = U3 + 1.602

UN = 0.316.M/S

13.99

16,900 km/h Earth Maximum altitude

GIVEN:

SATELLITE LAUNCHED AS SHOWN

FIND:

HAXINUM ALTITUDE, USING CONSERVATION OF ENERGY AND CONSERVATION OF MOMENTUM



R=6370 km Vo=500 km+6370 km Vo=6870 km = 6.87 × 106 m Vo=36,900 km/h =36,9×10 m 3.6×1035

= 10.25x103 m/s CONSERVATION OF ANGULAR MOMENTUM

$$V_{A'} = (\frac{r_0}{r_1}) V_0 = (\frac{6.870 \times 10^6}{r_1}) (10.25 \times 10^3)$$

$$V_{A'} = \frac{70.418 \times 10^9}{r_1} (1)$$

CONSERVATION OF ENERGY

TA = 10.25x103 M/s

TA = 12 M 10 = 12 M (10.25 × 103)2

 $V_{A} = \frac{GMm}{r_{o}} \qquad GM = 9R^{2} = (9.81 \text{ M/s}^{2})(6.37 \times 10^{6} \text{ m})^{2}$ $GM = 398 \times 10^{12} \text{ m}^{3}/5^{2}$ $V_{A} = -(\frac{398 \times 10^{12} \text{ m}^{3}/5^{2}) \text{ m}}{(1.87 \times 10^{6} \text{ m})^{2}} = -57.93 \times 10^{6} \text{ m} (J)$

POINT A' TA' = 1 M VA'

 $\Lambda^{VI} = -\frac{k'}{e^{11}M} = -\frac{k'}{348 \times 10_{15} M} (1)$

 $T_A + V_A = T_{A'} + V_{A'}$

57.53×10° M-57.93×10° M=1 MU2. -398×10° M

SUBSTITUTING FOR UN FROM (1)

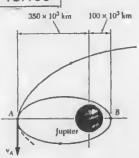
 $-5.402 \times 10^{6} = \frac{(70.418 \times 10^{9})^{2}}{(21(r_{1})^{2}} - \frac{398 \times 10^{12}}{r_{1}}$ $-5.402 \times 10^{6} = \frac{(2.4793 \times 10^{21})}{r_{1}^{2}} - \frac{398 \times 10^{12}}{r_{1}}$

(5.402x 100) 1,2-(398x1012) 1, + 2.4793x 1021=0

r= 66.7 x 10 m, 6 87 x 10 m

rnx=66,700 lem

13.100



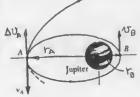
GIVEN:

Va=26.9 Lm/s
mass of Jupiter
MJ= 319 Me

FIND:

AUA, TO BRING THE SPACE CRAFT TO WITHIN 100X103 km AT B

CONSERVATION OF ENERGY



 $T_{A} = \frac{1}{2} W (U_{A} - \Delta U_{A})^{2}$ $V_{A} = -\frac{G M_{7} M}{M_{A}}$

 $Q_{E} = 6.37 \times 10^{6} \text{ M}_{J} = 319 \text{ GM}_{E} = 319 \text{ g R}_{E}^{2}$ $Q_{E} = 6.37 \times 10^{6} \text{ M}_{J} = (319)(4.81 \frac{\text{W}}{52})(6.37 \times 10^{6} \text{ M})^{2}$ $Q_{E} = 6.37 \times 10^{6} \text{ M}_{J} = (319)(4.81 \frac{\text{W}}{52})(6.37 \times 10^{6} \text{ M})^{2}$

 $V_{A}=350\times10^{6} \text{m}$ $V_{A}=-\frac{(126.98\times10^{15} \text{m}^{3}/\text{s}^{2})}{(350\times10^{6} \text{m})}$ $V_{A}=-\frac{(362.8\times10^{15} \text{m}^{3}/\text{s}^{2})}{(362.8\times10^{15} \text{m}^{3}/\text{s}^{2})}$

 $T_{B} = \frac{1}{2} m U_{B}^{2}$ $V_{B} = -\frac{G N_{5} m}{V_{B}} = -\frac{(126.98 \times 10^{\frac{15}{3}} \% s^{2}) m}{(100 \times 10^{6} m)}$ $V_{B} = -(1269.8 \times 10^{6}) m$

TA+VA=TB+VO

 $\frac{1}{2}M(V_A-\Delta V_A)^2-3628\times10^6M=\frac{1}{2}MV_B^2-1269.8\times10^6M$ $(V_A-\Delta V_A)^2-V_B^2=-1814\times10^6$ (1)

COUSERVATION OF ANGULAR MOMENTUM

VA=350X106M VB=100X106M

VAM (UB-AUB)=VBMUB

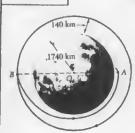
 $v_{B} = \left(\frac{r_{A}}{r_{B}}\right) \left(v_{A} - \Delta v_{A}\right) = \left(\frac{350}{100}\right) \left(v_{A} - \Delta v_{A}\right) \quad (2)$

SUBSTITUTE UB IN (2) INTO (1)

(UA-DUA) [1-(3.5)] =- 1814×106

(TALE + ROOT , - ROOT REVERSES FLIGHT DIRECTION) 5 VA= 26.9710 M (GIVEN) DD= (26.9x10 M - 12.698x10 M)

13.101



GIVEN:

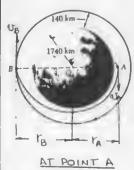
AT ENGINE SHUTOFF AT A VA= 1740+= 1748 km AT B, TB=1740+140=18801M COMMAND HODULE IN A CIRCULAR DEBIT

FIND:

(a) SPEED AT A AT ENGINE SHUTOFF.

(b) RELATIVE VELOCITY LEM NORCHES COMMAND MUDGE AT A

CONSERVATION OF ANG, HOHENTUR



MIAUA=MIBUB UB = TA UA = 1748 UA UB= 0.9298 UA (1)

CONSERVATION OF ENERGY

TA = 1 M VA VA = -GMMOON M

MHOON = 0.0123 HENRTH

6HMOON = 0.0123 GHENRIH = 0.0123 9 RENETH

GM moon = (0.0123) (9.81 M) (6.37×10 m)

GM = 4.896×1012 m3 TA= 1748 X103 M

 $V_A = \frac{-(4.896 \times 10^{12} \text{m}^3 5^7) \text{ m}}{(1748 \times 10^3 \text{m})} = -2.801 \times 10^6 \text{m}$

AT POINT B TB= 1 MVB rB=1880X10 M $V_B = -\frac{GM_{MON}M}{V_B} = -\frac{(4.896 \times 10^{12} \text{ m}^3/\text{s}^2)M}{(1880 \times 10^3 \text{ m})} = -2.604 \times 10^6 \text{ m}$

TA+VA=TB+VB; 1 MUA-2.801x16m=1 MUB-2104x16M

1) = UB+ 393.3×103 (M2)

(a) SPEED AT A

SUBSTITUTE UB IN (1) INTO (2) UA2 (1- (0.9298)2) = 393.3×103

VA= 2.903×106 VA= 1.704×10 M VA= 1704 M

(b) AT POINT B

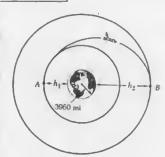
FROM (1) AND RESULT IN (a) Up=(0.9298)(1704)

UB= 1584.0 3 COHMAND HODULE IS IN CIECULAR O BIT, 18=188XID M (EQ 12.44)

Veire = V GHHOW = V 4.896x1012 = 1613.8 M

RELATIVE VELOCITY= UCIECUB= 1613.8-1584.0=29.8 \$

13.102



GIVEN:

h=200mi h2=500 m4.

FIND:

FOR A SPACECRAFT TRANSFERRING FROM A CIRCULAR ORBIT TO A CIRCULAR ORBIT AT B

(a) INCREASES IN SPEED AT A AND

(b) TOTAL ENERGY PER UNIT HASS TO EXECUTE THE TRANSFER



ELLIPTICAL OPBIT BETWEEN A AND B

CONS. OF ANG. HOHENTUM

MIAVA= MIBUB UA = 18 UB = 23.549 VB

1= 3960 ML+200 ML= 4160 ML 7= 21.965×106fe

UA=1.0721UB (1)

18=3960 mi+500mi=4460 mi TA = 23,549 X106ft

R=(3960)(5280)=20909×10°ft

CONSERVATION OF ENERGY GM=gR=(32,2 ft)(20,909 NOTE)

GM= 4.077 × 1015 ft3/52 POINT A:

 $T_A = \frac{1}{2} \text{ m } V_A^2 \quad V_A = \frac{GHm}{r_A} = -\frac{(4.077 \text{ k } 10^5) \text{ m}}{(21.965 \times 10^6)}$

 $T_B = \frac{1}{2} \text{ mUB}^2 V_B = \frac{\text{GHM}}{r_B} = \frac{(14.07710)^6 \text{ m}}{(73.540 \text{ M})^6} = -5977900 \text{ m}$

TATVA = TBTVB

1 mu 2-640.89 x 106 m= 1 mu -597.79 x 106 M VA-Va= 86.219×106

JB (1.0721)-1)=86.219x106 FROM (1) NA= 1.0721 UB UB= 576.98 ×106 ft2/52 V= (1.0721)(24.020×103) UB= 24,0204ft/s

UA= 25 752.6ft/s CIRCULAR OPBIT AT A AND B

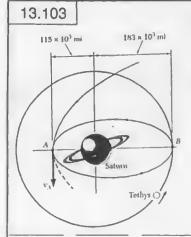
(EQ. 12.44) (VA) = VGH = VIA.071X1015 = 2531 6 ft/s

(VB)c = \ GH = \ \ \ \frac{14.077\ \text{VO}S}{23.54\ \text{23.54\ \te (a) INCREASES IN SPEED AT A AND AT B

AVA=VA-(VA)c='25753-25316 = 437 ft/s AUB=(UB)-UB)= 24449-24020 = 429 ftb_ (b) TOTAL ENERGY PER UNIT MASS F/m = \frac{1}{2}[U_A^2 - (U_A)_c^2 + (U_B)_c^2 - (U_B)^2]

E/m= = [(25759)2-(25313)]+(2450)2-(24020)]

E/m=216x10°ftb2.



(a)

F_A = 115 × 10³ mi

GIVEN:

UA=68.8x103 ft/s UTH=37.7x10 ft/s IN CIRCLAR OPERT

FIND:

CO DECREASE IN

SPEED, DVA OF A

SPACECRAFT AT A

TO ACHIEVE AN

ELLIPTICAL ORBIT

THROUGH A AND B

(b) THE SPACECRAFT

AS IT REACHES B

10=607.2×10 ft

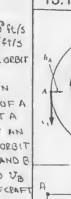
ra=966.2×106ft

JA = SPEED OF EPACECRAFT IN

THE ELLIPTICAL

ORBIT AFTER ITS SPEED HAS BEEN

DECREASED



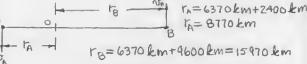
13.104



GIVEN: ha= 2400 km

hB=9600 km

SPEED, VA



CONSERVATION OF HOHENTUM FAMUL = FBMUB

$$U_{B} = \frac{V_{A}}{V_{B}}U_{A} = \frac{8770}{15970}U_{A} = 0.5492U_{A} \quad (1)$$

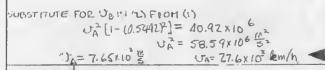
$$CONSERVATION OF ENERGY$$

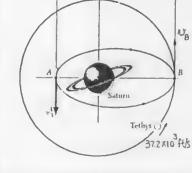
$$T_{A} = \frac{1}{2} m^{A}T_{A}^{2} V_{A} = \frac{GMm}{V_{B}} \quad T_{B} = \frac{1}{2} mU_{B}^{2} V_{B} = \frac{GHm}{V_{B}}$$

$$GM = 9R^{2} = (1.9) \frac{M}{52} (6370 \times 10^{3} m)^{2} = 398.1 \times 10^{12} \frac{m^{3}}{52}$$

$$V_A = \frac{-(398.1 \times 10^{12})}{8770 \times 10^3} = 45.39 \times 10^6 \text{ m}$$

TA+VA=TA+VO = 1 W U2 + 45.39 X:06 M= 1 W U2 - 24.93 KIO M(2)





ELLIPTICAL ORBIT BETWEE! A AND B

 $\Gamma_{\rm B} = 183 \times 10^3 \, \rm mi$

POINT A TA= 1 m 0'2 VA= GHSATM

MSA = HASS OF SATURM, DETERMINE GMSA FROM THE SPEED OF TETHYS IN ITS CIRCULAR ORBIT

(EQ 12.44)
$$V_{clec} = \sqrt{\frac{GM_{cat}}{F}}$$
 $GM_{SAT} = r_{B}V_{clec}^{2}$
 $GM_{SAT} = (966.2 \times 10^{6} ft)(37.2 \times 10^{3} ft/s)^{2} = 1.337 \times 10^{18} ft^{3}/s^{2}$
 $V_{A} = -\frac{(1.337 \times 10^{18} ft^{3}/s^{2}) m}{(607.2 \times 10^{6} ft)} = -2.202 \times 10^{3} m$

 $\frac{PDINTB}{V_{B}=1.384\times10^{9}}V_{B}=\frac{GH_{SAT}M}{r_{B}}=\frac{(1.337\times10^{18}f(^{3})^{2})M}{(166.2000641)}$

TA+VA=TB+VB; = MU'2=2.202x10M=1MU'2-1,384 X10M

CONSERVATION OF ANGULAR HUMAN LUM VA WWA = VB WB "UB= VA VA = 607.2 KIO UA=1.6784 UA WA [1-(0.6784)]=1.636 X 109 VA = 57005 ft/s DVA = VA-VA = 68800-52005=16795 ft/s

 $U_B = \frac{t_A}{t_B} N_A = (0.6284)(52005) = 32700 + 1/5$

h_A O h_B B

+13.105

GIVEN: U=26.3×10 2m/h U=185×102m/h FINO:

ALTITUDE, HB

 $V_{\rm A} = 26.3 \times 10^{3} \, \text{km/h}$ $V_{\rm A} = 7.31 \times 10^{3} \, \text{m/s}$ $V_{\rm B} = 18.5 \times 10^{3} \, \text{km/h} = 5.14 \times 10^{3} \, \text{m/s}$

CONSERVATION OF MOHENTUM YAMUA=YBMUB

$$r_A = r_B v_B$$
 $r_A = \frac{v_B}{v_A} r_B = \frac{18.5}{26.3} r_B$ $r_A = 0.7034 r_B$ (1)

13.105 : continued

CONSERVATION OF ENERGY

TA = 1 m UA2 TA = 1 m (7.31x103) = 26.69x10 m

TB= 1 m UB TB= 1 m (5:14×103)= 13.20×106 m.

VA = - GHM GM = gR = (9.81 5) (6370×10)2 CH= 398.1x1012 m3/52

VA=-398.1×1012

VB=-GHM = -398.1 X1012

TATVA=TRTVB

26.69 X10 W - 398.1X10 W = 13.20 X10 W - 398.1X10 W SUBSTITUTE FOR M FROM (1) $\frac{398.1 \times 10^{12}}{r_{\rm B}} \left[\frac{1}{(0.7034)} - 1 \right] = 13.49 \times 10^{6}$

 $\frac{1}{r_0} = 80.37 \times 10^{-9}$

ra= 12.4 42×10 m = 12442 km

hB= rB-P= 12442km-6370km- 6070km

* 13.106

GIVEN:

COMMAND MODULE IN CIRCULAR OFSIT AT AN ALTITUDE OF 140 2% ATTACH ED LEH CAST ACRIFT AT RELATIVE VELOCITY 0=200m/s

FIND:

UZ AND & AS THE LEN HITS THE MOON

COHHAND HOOULE IN CIRCULAR ORBIT No 13=1740+140=1880 &m=1.88 x10 m K=R GMHOON = 00123GHENETH = 0.0123 2 52 = 0.0123 (9.81) (6.37x 106)2 = 4.896×1012 m3/52

R= 1740 km

ZF= Man GHmm= myo Vo=VGMm=V486×1012

No=1614 m/s VB=1614-200=1414 m/s CONSERVATION OF ENERGY BETWEEN BAND C U= UB+2GMm (FB-1)

Uc2=(1414 m/s)2+2(4.896×1012m2/52)(1.88x106-1)

12=1,999×10+0.4191×106= 2.418×106 # V= 1555 %

* 13,106 continued

CONSERVATION OF ANGULAR HOMENTUM

ramus=RMUsINO SING = TOUR = (1.88 x 10 cm)(1414 m) = 0.98249

(1.74 ×106m)(1555 (4) ф= 79,26° 0=79.3°

13,107

GIVEN:

SATELLITE PROJECTED AT VELOCITY TO AT AN ANGLE & WITH ITS INTENDED CIRCULAR

FIND;

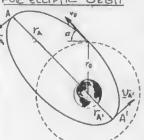
THAX AND THIN



VO FOR CIRCULAR OPBIT OF PADIUS to F=man GHM = m 362

BUT NO FORMS AN AUGLE OF WITH THE INTENDED CIRCULAR PATH

FOR ELLIPTIC OPBIT



CONS OF ANG HOMENTUM tomuo cosa= ramua

VA= (to cosa) vo

CONS OF FIFERSY

1 mvo- GMm =

U2-VA2 = 26H (1-10)

SUBSTITUTE FOR UA FROM (1)

BUT 102= Et, THUS 1- 103 205 x = 2 (1-12)

cos 2 (to 12 2 (to)+1=0

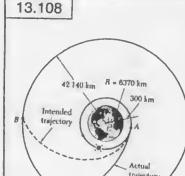
SOLYING FOR. to

 $\frac{r_0}{r_A} = + 2 + \sqrt{4 - 4\cos^2\alpha} = \frac{1 + \sin\alpha}{1 - \sin^2\alpha}$

1 = (1+51Nd)(1-51Nd) 10=(1751Nd) to C ALEO YALID FOR POINT

THUS

MAX = (I+SINA) TO MIN= (+SINA) TO



COMMUNICATION

SATELLITE AT A

13 LAUNCHED

WITH A YELOCIN

RELATIVE TO A

SPACE PLATFORM

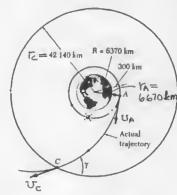
IN CIRCULAR

ORDIT OF

(VA) = 3.44 cm/s

EIND:

ANGLE Y AT WHICH THE SATELLITE CROSSES THE CIRCULAR ORBIT AT C.

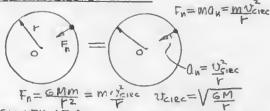


R=6370km TA=6370km+300km TA=6.67X106M

Tc=42.14x106m

GM= 9 R2 G11=(9.81 \$)(6.37x15m) GM= 398.1X10¹² M/52

FOR ANY CIRCULAR OBBIT



(NA) CIRC = VGH = VG98.1X1012 m3 = 7.726X10 m/5

JA=1.77AKIRC+(VA) R=-7.726X103+3.44X10=11.165X103M

YELOCITY ATC

CONSERVATION OF ENERGY TATVA=TC+VC

V= VA+2GH(+2-+A)=(11.165x10)+2(398.1x10)/124x16660mb

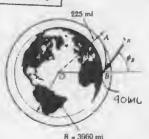
 $U_c^2 = 124.67 \times 10^6 - 100.48 \times 10^6 = 24.19 \times 10^6 \frac{m^2}{52}$ $U_c = 4.919 \times 10^3 \text{ m/s}$

CONSERVATION OF ANGULAR HOHENTUM

COS X = 16.67x10)(11 165x103)

COS X = 16.67x106)(4.919x103)

COSX=0.35926 Y=68.9° 13.109

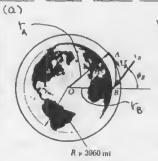


GIVEN:

VEHICLE IN CIRCULAR ORBIT AT ALTITUDE OF 225 ML. SPEED DECREASED AT A SO THAT IT REACHES ALTITUDE AT B OF 40 ML AT AN ANGLE ϕ_B^2 60°

FIND:

(0) UA, AS VEHICLE LEAVES ITS CIRCULAR OPENT (b) UB



YA=3960 ML+225 MC=4185Mi YA=4185MCX5280 Ft=2209 NJ

18=3960mc+40mi=4000mi 18=4000x5280=21120x103ft

R=3960mc= 20909x103ft

GM=g E= (32.2 11) (20909 non)

GN= 14.077X10 5 ft3/52

CONSERVATION OF ENERGY

TA= 1 M JA VA = -GHM = -14.077X10 M= 637.1X10 M

TB= 1 MUB VB=-GHM = -14.077x103 M=-6665x10m

TA + VA = TB + YB

1 MU2-637.1×10 M= 1 MU2-666.5×106 M

 $V_{A}^{2} = V_{B}^{2} - 58.94 \times 10^{6}$

CONSERVATION OF ANGULAR HOHENTUH

 $V_B = \frac{(r_A) \ U_A}{(r_B)(SIN\phi_B)} = \frac{4185}{4000} \left(\frac{1}{SIN\phi_B}\right) U_A$ $V_B = \frac{(r_B)(SIN\phi_B)}{1.208} = \frac{4185}{4000} \left(\frac{1}{SIN\phi_B}\right) U_A$

SUBSTITUTE UB .FROM (2) IN (1)

UA=(1208UA)2-58.94X106

 $U_{A}^{2}[(1.208)^{2}-1] = 58.94 \times 10^{6}$

UA2= 128.27×106 ft3/52

(a) UA=11.32x103945

(b) FROM (2)

UB=1.208 UN=1.208 (11.32×106) = 13.68×103/t/s

UB=13.68x 103 PEG.

(5)

*13.110

R=3960 mc

GIVEN:

VEHICLE AT A IN

CIRCULAR ORBIT IS

GIVEN AN INCREMENTAL

VELOCITY AUT TOWAR

O. ALTITUDE'S AS SHOWN

ENERGY EXPENDITURE

IS 50% OF THAT

USED IN PROB 13.109

FIND:

UB AND \$8



TA = 3960 MI + 225 MI

LALIRE FA=4185 MI = 22.097 XIO FT

FB=3960 MI + 40MI = 4000 MI

FB=21.120 XIO FT

GM= $g R^{2}(37.2 \frac{f}{5}) / (3960) / (3280) / f$ GM = $14.077 \times 10^{15} \text{ ft}^{3}/\text{s}^{2}$

3960 mi

VELOCITY IN CIRCULAR GRBIT AT 275M ATTRIBE



F= GMM

Tale Tale

Man an = MUalciec

Tale

F= man $\frac{GMM}{K^2} = \frac{MEWIONS}{K} = \frac{MEWIONS}{K} = \frac{MW}{K} = \frac{MO71710^{15}}{22.047706} = 25.2410^3 \text{ ft/s}$

ENERGY EXPENDITURE

FROM PROB. 13.109 NA= 11.32×103 ft/s ENERGY, DE109 = 1 M(VA) cine - 1 M NA

1 = 100 = 7 m(ss. 54x10) = 1 m(11.32x10) =

 $\Delta E_{110} = (0.50)\Delta E_{109} = (254.46 \times 10^6 \text{ m ft.} 16)$ $\Delta E_{110} = (0.50)\Delta E_{109} = (254.46 \times 10^6 \text{ m})/2 \text{ ft.} 16$ THUS, ADDITIONAL KINETIC ENERGY AT A 15 $\Delta E_{110} = (254.46 \times 10^6 \text{ m})/2 (1)$

CONSERVATION OF EMERGY BETWEEN A AND B

TA=1 M(Vaking+ (DVA)2) VA=-GHM

TB= 1 mVB VB= -GHM

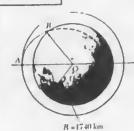
 $\frac{1}{2}$ m (25.24x10) $\frac{3}{12}$ $\frac{2}{11}$ $\frac{2}{120}$ $\frac{15}{16}$ $\frac{1}{120}$ $\frac{1}{12$

 $V_B^2 = 637.4 \times 10^6 + 25446 \times 10^6 - 1274.1 \times 10^6 + 1333 \times 10^6$ $V_B^2 = 950.4 \times 10^3$

COUSE PUATION OF ANGULAR HOMENTUM BETWEEN AAMOR

 $= \ln \phi_{B} = \frac{|Y_{A}|}{|Y_{B}|} \frac{(U_{A})_{GR}}{(U_{B})} = \frac{(4185)(25.24 \times 10^{3})}{(4000)(30.8 \times 10^{3})} = 0.8565$

13.111



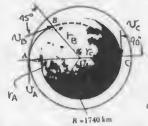
GIVEN:

LEMATANALTITUDE OF 140 &M IS SET ADRIFT FROM A CIRCULAR ORBIT AND ITS SPEED IS REDUCED

EIND:

(a) SHALLEST REDUCTION
OF SPEED TO MAKE SURE
THE LEM WILL HIT THE
MOON

WHICH WILL CAUSETHE LEH TO HIT THE HOON AT 450



1/4=1740km+140km=1880x10m 1/3=12=R=1740km=1740x10m

GM_{HOON} = 0.0123 GMe = 0.0123 g P² = (0.0123)(9.81 m/s²)(6.3 m/s²) GM_{HOON} = 4.896×10¹² m²/s²

VELOCITY IN A CIRCULAR ORBIT AT HOLEM ALTITUDE

UCIEC VEN MOON = V4.896X1012 W/32 = 1.6138×103 m/3

(a) AN ELLIPTIC TRAJECTORY BETWEEN A AND C, WHERE THE LEM IS JUST TANGENT TO THE SURFACE OF THE HOON, WILL GIVE THE SMALLEST REDUCTION OF SPEED AT A WHICH WILL CAUSE IMPACT COURSERVATION OF ENERGY (A AND C)

TA= 1 m NA VA= -GKM = -4.896X10 M = -2.604X10 M

TC= 1 M V2 Vc= GHmm = 4.896x10 m - 2.814x10 m

TA+VA=TC+VC & mv=2.604x10m= & mv-2-2.8410x10m

CONFEDERATION OF ANGULAR MOMENTUM (AANOC)

ramua = rc muc vc= ra va= 1880 Na=1.0805 Na

REPLACE UC IN (1) BY (2)

 $V_{A^{2}} = (1.0500 V_{A})^{2} + 419.1 \times 10^{3}$ $V_{A^{2}} = (1.0805)^{2} = 419.1 \times 10^{3}$ $V_{A^{2}} = 2.502 \times 10^{6}$

UA= 1582 M AVA=(UA) 12- VA=164-1582=31.5M

(b) CONSERVATION OF ENERGY (AAND B)

SINCE TB = YZ COND OF ENERGY IS THE SAME AS BETWEEN A AND C.

THUS FROM (1) UAZ= UBZ-419.1X103 (1)

CONSERVATION OF ANGULAR HOHENTUM (A AND B)
tamva=tomvasin \$ \$=450

UB = TAVA = 18800A = 1.528 UA (3)

REPLACE UB IN (11) BY (3)

NA= (1.528 VAZ) - 419×103 NA= 313.48×103 VA= 560 M DVA= (VA KIRZ VA= 1614-560= 1053 M *13.112

GIVEN:

SPACE PEOBE IN CIRCULAR ORBIT OF RADIUS NP., WITH VELOCITY U. ABOUT A PLANET OF RADIUS P.

SHOW THAT:

(a) PROBE WILL HIT THE PLANET AT AN ANGLE Θ WITH THE VERTICAL, IF ITS VELOCITY IS REDUCED TO $\propto U_0$ WHERE $\alpha = \sin \theta \frac{2(n-1)}{\eta^2 \cdot \sin^2 \theta}$ PLANET IT $\alpha > \sqrt{\frac{2}{1+n}}$



(a) DUSERVATION OF ENERGY AT A Ta=1 m(~Vo)?

> VA= - GMM NR

$$T_0 = \frac{1}{2} M v^2$$

M= MASS OF PLANET M= MASS OF PROBE

VB=-GMM R

 $T_A + V_A = T_B + V_B$ $\frac{1}{2} M(\alpha v_0)^2 = \frac{1}{NR} M v^2 = \frac{1}{2} M v^2 = \frac{GHM}{R} (1)$

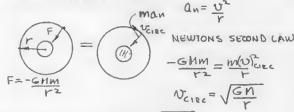
CONSERVATION OF ANGULAR HOMENTUM

NRM & NO = EMUSINO

REPLACE OF IN (1) BY (2)

$$\left(\chi N_0\right)^2 \frac{26M}{NR} = \left(\frac{N\chi N_0}{51N0}\right)^2 - \frac{2GM}{R} \qquad (3)$$

FOR ANY CIRCULAR OPBIT



FOR Y= NR U0= VCIEC = VEH

SUBSTITUTE FOR U_O IN (3) $X^{2} GM - 2GM = \frac{N^{2} X^{2}}{N R} \left(\frac{GM}{N R} \right) - \frac{2GM}{R}$ $X^{2} \left[1 - \frac{N^{2}}{SIN^{2}\theta} \right] = 2(1-N)$ $X^{2} = \frac{2(1-N)(SIN^{2}\theta)}{(SIN^{2}\theta - N^{2})} = \frac{2(N-1)SIN^{2}\theta}{(N^{2}-SIN^{2}\theta)}$ $X = SIN\theta \sqrt{\frac{2(N-1)}{N^{2}-SIN^{2}\theta}} (QED)$

(b) PROBE WILL JUST HISS THE PLANET IF 0 > 90°

NOTE: $N^2 - 1 = (N-1)(N+1)$

13.113



GIVEN:

Vp AND VA AS SHOWN SHOW THAT:

 $V_{A}^{2} = \frac{2 GM}{\Gamma_{A} + \Gamma_{P}} \frac{\Gamma_{P}}{\Gamma_{A}}$ $V_{P}^{2} = \frac{2 GM}{\Gamma_{A} + \Gamma_{P}} \frac{\Gamma_{A}}{\Gamma_{P}}$

CONSCEVATION OF ANGULAR HOMENTUM

 $V_A m v_A = V_P m v_P$ $V_A = \frac{V_P}{V_P} v_P$ (1)

CONSERVATION OF ENERGY

1 M Up - GHM = 1 M UA - GHM (2)

SUBSTITUTING FOR UN FROM (1) INTO (2) $V_p^2 - \frac{C_p}{T_p} = \left(\frac{T_p}{V_A}\right) U_p^2 - \frac{C_p}{T_A}$

$$\frac{\left(1-\left(\frac{r_{p}}{r_{A}}\right)^{2}\right)}{\left(\frac{r_{p}}{r_{A}}\right)^{2}} = 2GM\left(\frac{1}{r_{p}}-\frac{1}{r_{A}}\right)$$

$$\frac{r_{A}^{2}-r_{p}^{2}}{r_{A}^{2}}U_{p}^{2} = 2GM\left(\frac{1}{r_{p}}-\frac{1}{r_{A}}\right)$$

WITH 12-12 = (14-14) (14+14)

$$V_p^2 = \frac{2GH}{r_{A1}r_{B}} \frac{r_{A}}{r_{B}}$$
 (2)

EXCHANGING SUBSCRIPTS PAND A

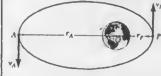
GIVEN:

13.114

EARTH SATELLITE OF HASS M
DESCRIBING AN ELLIPTIC OBSIT
TA IS HAXIMUM AND TO IS MINIMUM
DISTANCES TO EARTHS CENTER

SHOW THAT:

TOTAL ENERGY E = - GMM , WHERE IN- MASS OF THE ENERTH



SEE SOLUTION TO PROB 13.113 (FBOYE) FOR DERIVATION OF EQUATION (3)

TOTAL ENERGY AT POINT P 15

$$E = T_{p} + V_{p} = \frac{1}{2} M U_{p}^{2} - \frac{GMM}{r_{p}}$$

$$= \frac{1}{2} \left[\frac{2GMM}{r_{A} + r_{D}} \frac{r_{A}}{r_{p}} \right] - \frac{GMM}{r_{p}}$$

$$= \frac{GMM}{r_{p}(r_{A} + r_{p})} \frac{V_{A}}{r_{p}} - \frac{1}{r_{p}} = \frac{GMM}{r_{p}(r_{A} + r_{p})}$$

NOTE: BECALL THAT GRAVITATIONAL POTENTIAL OF A SATELLITE IS DEFINED AS BEING ZERO AT AN INFINITE DISTANCE FROM THE EARTH



SPACECRAFT OF HASS M IN CIRCULAR OPBIT OF PADIUS K ABOUT THE EARTH

SHOW THAT:

6) ADDITIONAL ENERGY DE TO TRANSFER IT TO A CIRCULAR OPEIT OF LARGER RADIUS & DE= EMM(5-17)

(b) AHOUNTS OF ENERGY AT A AND B ARE $\Delta E_A = \frac{r_2}{r_1 + r_2} \Delta E$, $\Delta E_0 = \frac{r_1}{r_1 + r_2} \Delta E$

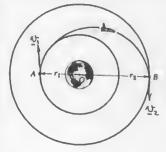


(A) FOR A CIRCULAR OPBIT OF PADIUS + F=man: GHM = M V

$$V^{2} = \frac{GM}{r}$$
 $V^{2} = \frac{GMm}{r} = -\frac{1}{2} \frac{GMm}{r}$ (1)

THUS DE REQUIRED TO PASS FROM CIRCULAR ORBIT OF PADIUS 1. TO CIRCULAR OBBIT OF PADIUS 12 15 DE=ErEz=-1GHm+1GHM

(b) FOR AN ELLIPTIC ORBIT WE RECALL EQ (3) DERIVED IN PROBLEM 13.113 (WITH Up= U1) 26M 12 11+12 11



AT POINT A: INITIALLY SPACECRAFT IS IN A CIRCULA ORBIT OF PADIUS Y VCIRC = GH

AFTER THE SPACECRAFT ENGINES ARE FIRED AND IT IS PLACED ON A SEMI-GLLIPTIC PATH AB, WE RECALL

TI= 1 MO1 = 1 MITHER TO IN ENERGY IS

RECALL EQ (2): DEA = 12 DE (Q.E.D)

A SIMILAR DERIVATION AT POINT B YIELDS, DEBTINHO

13.116

GIVEN:

HISSILE FIRED FROM THE GROUND WITH VELOCITY TO AT AN ANGLE O WITH THE VERTICAL, REACHES A MAXIMUM ALTITUDE & P. WHERE P IS THE PECIUS OF THE EARTH

SHOW THAT:

(a)
$$\sin \phi_0 = (1+\alpha)\sqrt{1-\frac{\alpha}{1+\alpha}\left(\frac{\mathcal{V}_{esc}}{\mathcal{V}_0}\right)^2}$$

WHERE VESC = ESCADE VELOCITY

(b) PANGE OF ALLOWABLE VALUES OF VO



CONSERVATION OF ANG. MON.

RMU.SINO. = VE MUB

B = R+XR = (1+X)R UB= PUOSINDO = VOSINDO

TATVA=TBTVB IMUOZ GMM = IMUOZ GMM

$$v_0^2 - v_0^2 = \frac{2 \text{GMM}}{R} \left(1 - \frac{1}{1 + \alpha}\right) = \frac{2 \text{GHM}}{R} \left(\frac{\alpha}{1 + \alpha}\right)$$
SUBSTITUTE FOR v_0 FROM (1)

$$N_{s}^{o}\left(1-\frac{(1+\alpha)_{s}}{2!N_{s}\varphi^{o}}\right)=\frac{S}{SCHM}\left(\frac{1+\alpha}{\alpha}\right)$$

FROM EQ. (12,43): UES = 26M

$$V_0^2 \left[1 - \frac{31N^2 \dot{\Phi}_Q}{(1+\alpha)^2} \right] = V_{ESC}^2 \left(\frac{\alpha}{1+\alpha} \right)$$

$$\frac{51N^2\phi_0}{(1t\alpha)^2} = 1 - \left(\frac{\upsilon_{esc}}{\upsilon_0}\right)^2 \frac{\alpha}{1+\alpha}$$
 (2)

(b) ALLOWABLE YALVES OF VO (FOR WHICH HAXIMUM ALTITUDE = XP)

FOR SINDO = O FROM (2)

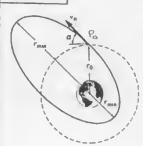
$$0 = 1 - \left(\frac{\sqrt{N_0}}{N_0}\right)^{\frac{1}{2}} \frac{1+\alpha}{\alpha}$$

FOR SINDO=1, FROM Z

$$\left(\frac{\mathcal{Q}_{rsc}}{\mathcal{Q}_{o}}\right)^{2} = \frac{1}{\alpha} \left(1 + \alpha - \frac{1}{1 + \alpha}\right) = \frac{1 + 2\alpha + \alpha^{2} \cdot 1}{\alpha \cdot (1 + \alpha)} = \frac{2 + \alpha}{1 + \alpha}$$

$$\mathcal{Q}_{o} = \mathcal{Q}_{esc} \sqrt{\frac{1 + \alpha}{2 + \alpha}}$$

+13.117

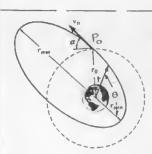


GIVEN:

FROM PROB. 13.107 Tmin= 10 (1-51NX) TMAX = TO (1+SINA)

: TAHT WOULS

INTENDED CIRCULAR OFBIT AND RESULTING *FULIPTIC ORBIT* THIERSECT AT THE ENDS OF THE MINOR AMS OF THE ELLIPTIC ORBIT AT Po



IF THE POINT OF INTERSECTION PO OF THE CIRCULAR AND ELLIPTIC OPBITS IS AT AN END OF THE HINDR AXIS THEN UO 'S PARALLEL TO THE HAJOR AXIS. THIS WILL BE THE CASE ONLY IF X+90°= OO, THAT IS IF COS 0= -SINK WE HUST THEREFORE PROVE THAT

COS 0=-21NX (1)

WE RECALL FROM ED (12.39):

$$\frac{1}{T} = \frac{GM}{N^2} + C\cos\Theta \qquad (2)$$

WHEN 6=0, r= rmin AND rmin= ro(1-SINX)

FOR 0=180° r= rmax= ro (HSING)

$$\frac{1}{\kappa_0(1+\sin k)} = \frac{611}{k^2} - c \tag{4}$$

ADDING (3) AND (4) AND DIVIDING BY Z:

SUSTRECTING (4) FROM (3) AND DIVIDING BY 2:

$$C = \frac{1}{2r_0} \left(\frac{1}{1 - 51NQ} - \frac{1}{1 + 51NQ} \right) = \frac{1}{2r_0} \frac{251NQ}{1 - 51N^2Q}$$

SUBSTITUTE FOR GET AND C INTO EQ (2)

$$\frac{1}{r} = \frac{1}{r_0 \cos^2 \alpha} \left(1 + SIN\alpha \cos \theta \right) \tag{5}$$

LETTING Y= TO AND 0 = 00 IN EQ (5), WE HAVE

COE X = 1+ 5'NO COS BO

$$\cos\theta_0 = \frac{3523 - 1}{511104} = -\frac{51824}{5184} = -5184$$

THIS PROVES THE VALIDITY OF EQ (1) AND THUS PO IS AN END OF THE MINOR AXIS OF THE ELLIPTIC DEBIT

* 13.118

GIVEN:

SPACE YEHICLE UNDER GEAVITATIONAL ATTRACTION OF A PLANET OF HASS H (FIG. 13.15, SHOWN BELOW)

EIND:

(a) TOTAL ENERGY PER UNIT HASS, E/M, IN TECHS OF I'MIN AND UMAX AND THE ANGULAR HOMENTUM PER UNIT HASS, h.

(C) ECCENTRICITY E= /1+ ZE (h)2

(d) TRAJECTORY IS a. HYPERBOLA IF E70 FLLIPSE IF E = 0 PARABOLA IF ECO

(A) POINTA ANGULAR MOMENTUH PER UNIT KAS h= Ho = rmin m Wmax

h= 1min Uma (1) .

$$E/m = \frac{1}{m} (T+V)$$

E/m= m (1 m vmax GMm) = 1 v2 - GM

(b) FROM EQ. (1): Umax= h/rmin. SUBSTITUTING INTO (2)

$$E/M = \frac{1}{2} \frac{h^{2}}{r_{min}^{2}} - \frac{GM}{r_{min}}$$

$$(\frac{1}{r_{min}})^{2} - \frac{2GM}{h^{2}} \cdot \frac{1}{r_{min}} - \frac{2(E/M)}{h^{2}} = 0$$

SOLVING THE QUADRATIC: Thin = EM + VEM) + Z(E/M)

$$\frac{1}{r_{min}} = \frac{GM}{h^2} \left[1 + \sqrt{1 + \frac{2E}{m} \left(\frac{h}{GM} \right)^2} \right]$$
 (3)

(c) ECCENTRICITY OF THE TRAJECTORY EQ (12.39') $\frac{1}{r} = \frac{GM}{N^2}$ (1+ E cos 0)

$$EQ(12.39') \frac{1}{r} = \frac{GM}{N^2} (1 + E \cos \theta)$$

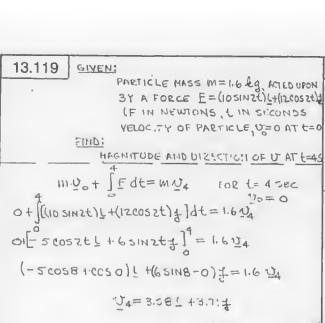
WHEN $\Theta=0$, $\cos\theta=1$ AND $r=r_{min}$, THUS $\frac{1}{r_{min}}=\frac{GM}{h^2}\left(1+\varepsilon\right)$

(d) RECALLING DISCUSSION ON PAGES 708,709 AND IN YIEW OF LQ. (5)

1. HYPERBOLA IF E > 1, THAT IS IF E > 0 2. PARAZOLA IFE=1, THAT IS IF E=0 3. ELLIPSE IF ECI, THAT IS IF ECO

NOTE: FOR CIRCULAR OPBIT E= O AND 1+2E (A 120 OR E = - (GH) 2 M

BUT FOR CIEVIAR ORBIT U= GH AND h=U'r= GMr THUS E = - 1 M (GH)2 = -1 GHM (CHECKS WITH (1) FOUND IN 13-115)



OH- 5 COSZEL +651NZET = 1.674 (-50058+0050) 1 +(651N8-0) 1=1.6 1 U4= 3.58(+3.7:4

13.120

GIVEN:

5-16 PARTICLE ACTED UPON BY A FORCE E = - Zt2 L+ (3-t) + (F IN POUNDS AND tIN SECONDS YELOCITY OF THE PAGNICLE 15 No=(16 16/5) L AT t= 0.

FIND:

(a) TIME AT WHICH VELOCITY IS PARALLEL TO THE Y. A.VIS (b) THE CONSESPONDING VELOCITY

(a) MU, 1) Edt= m(N, 110y 1)

BUT Ux=0, IF VELOCITY IS PARALLEL TO Y-AXIS

1622 14 (3-1) + ldt = 3 1/1 + でたしてまです(のでで)は一色のりま (1) (8-5/3) [1 (31-1)] = 3 Ny #

SINCE THE X COMPONENT OF THE VELOCITY IS ZERO +3= 2.329

t=1.3265

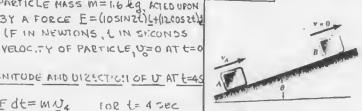
3.098 = 0.1553 Ny 3

(b) SUBSTITUTE L=1.326 IN (1) CL + [3(1.326)-(1.326)] == 5 22 by }

Vy= 19.95 ft/s

13,121

FOR t= 4 SEC



GIVEN;

Un= 30 14, 46=0.30

FIND: TIME FOR THE

BLOCK TO REACH B WHERE UB=O, IF (a) 0=0, (b) 0=20°

(1) 6=0 MUB=0 FF=HEWE

(32.294)(30)

(b) 0= 20° =120° Nt=Wtcos200

THPULSE-HOHENTUH IN Y DIRECTION + MUA-4kWtcoszo"-WtsINZO= 0 9(4xCO520°+SIN20°) (37.2 ft/52)(.1000520, SIN20)

t= 1.493 S

13,122



NA= 10m/s, 4x=030 0= 30°

FIND: TIME FOR THE BLOCK

TO REACH U= 10 M/S DOWN AND TO THE LEFT

-WESIN 300 MFC0330° Ft=4KNt NF= MFC02300

UPTHE PLANE TO B

MNA-4KW+COS 300-W+ SIN30=0 g(4 60230 +2 (N3O) (481 / 10.30 1/2 +1)

PLANE TO C Uc = 10 m/s 77300 Vp=0

JB+11KMF8502300-MF211130=-M JC 10 m/s =4.744 5 t BC = g(sin 30°-4x (05 30°) (1.81m)(1-,3(3))

t= tAB+tBc=1.342+4.244= 5.59 s

13.123

GIVEN:

CEAR (DRIVE) WHEELS OF A CAR SLIP FOR FIRST LOFE WITH FRONT WHEELS JUST OF F THE GROWND. $M_k = 0.60$ WHEELS POLL WITHOUT SLIPPING FOR THE REMAINING 1260 FE WITH 60 % OF THE

REMAINING 1260 FE WITH 60 % OF THE WEIGHT ON THE REAR WHEELS. 45=0.85
IGNORE AIR AND ROLLING RESISTANCE

FIND:

(a) SHORTEST TIME FOR THE CAR TOTRAVEL THE FIRST GOFT STARTING FROM REST

(b) HINIMUM TIME FOR THE CARTO RUNTHE WHOLE RACE

(a) FIRST 60 ft

VELOCITY AT 60 Ft REAR WHEELS SKID TO

GENERATE THE HAXIMUM FORCE RESULTING
IN MAXIMUM VELOCITY AND HINIMUM TIME
SINCE ALL THE WEIGHT IS ON THE REAR WHEELS
THIS FORCE IS F=4KN=0.60W

TO= 0 U0-60 (F) (60) T60 = 1 mV60

 $0 + (4_{E}N)(60) = \frac{1}{2} \frac{1}{9} \sqrt{60^{2}}$ $1 + (4_{E}N)(60) = \frac{1}{2} \sqrt{60^{2}}$ $1 + (4_{E}N)(60) = \frac{1}{2} \sqrt{60^{2}}$ $1 + (4_{E}N)(60) = \frac{1}{2} \sqrt{60^{2}}$

IMPULSE - HOMENTUM

+ 0+4kWt = $\frac{W}{g}$ $\frac{V_{60}}{V_{60}} = 48.15 \text{ ft/s}$ $\frac{48.15 \text{ ft/s}}{(0.60)(32.2 \text{ ft/s}^2)}$

 $(0.60)(32.2ft/s^2)$ $t_{0-60} = 2.49s$

(b) FOR THE WHOLE PACE

THE HAXIMUM FORCE ON THE WHEELS FOR THE FIRST GOFT IS F=4kW=0.60W
FOR REMAINING 1260 FT THE HAXIMUM FORCE IF THERE IS NO SLIDING AND GO % OF THE WEIGHT IS ON: THE REAR (DRIVE) WHEELS IS

F=Us(0.60)W= 0.85)(0.60)W=0.510W

VELOCITY AT 1320 ft

WORK AND ENERGY TOTY TOTY TOTAL

To=0 U0-60=(0.60 W)(60 ft), U=(0.510 W)(1260ft)

0+36W+(1510)(1260)== 1 4.01320

IMPULSE-MOHENTUM

(m) (48.15) + 0.510W to -1320 = y (209); to -1320 = 40.100 = 17.29 = 4.7955

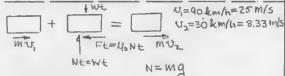
13.124

GIVEN:

TRUCK ON LEVEL ROAD TRAVELING AT 90 km/h
BRAKES ARE APPLIED TO SLOW IT TO 30 km/h
ANTISKID BRAKING SYSTEM LIMITS BRAKING
FORCE SO THAT WHEELS ARE AT IMPENDING
SCIOING, 45 = 0.65

FIND:

SHORTEST TIME FOR TRUCK TO SLOW DOWN



 $m v_1 - 1 < Nt = m v_2$ $m(2 \le m/s) - (0.65) m (9.81 = 1) t = m (8.33 m/s)$

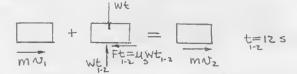
 $t = \frac{25 - 8.33}{(0.65)(9.81)} = 2.615$

13.125 GIVEN:

TRAIN DECREASES SPEED FROM 200 &m/h to 90 &m/h at a constant rate in 123.

FIND:

SHALLEST ALLOWABLE COEFFICIENT OF FRICTION IF A TRUNK IS NOT TO SLIDE



1=200kn/n=55.56m/s

Nz=40lem/h=25.0 m/s

(55.56 m) -45 (9.81 m) (125) = 25 m/s

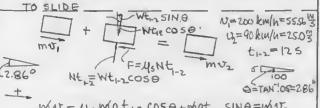
$$\mu_5 = \frac{(55.56 - 25.0)}{(4.81)(12)} = 0.2596$$
 $\mu_5 = 0.260$

13.126 GIVEN:

TRAIN DECREASES SPEED FROM 200 km/h
TO 40 km/h DOWN A 5% GRADE AT A
CONSTANT RATE IN 12 S.

FIND:

SHALLEST COEFFICIENT OF FRICTION IF A TRUNK IS NOT



WUI-USWIGTIZ.COSO+WIGTIZSINO=WUZ

(55.56 \mathrew)-Us(9.81 \mathrew)/125/2005286/HA.81 \mathrew)/125/2007.869)=25 \mathrew 3

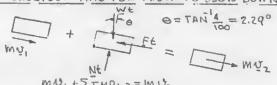
 $45 = \frac{55.56 - 25.014.81)(12)(51N 2.86°)}{(9.81)(12)(052.86°)} = 0.310$

13.127 GIVEN:

TRUCK SLOWS FROM 60 MI/N TO ZOMIND DOWN A 4% GRADE WITH ITS WHEELS JUST ABOUT TO SLIDE US=0.60

FIND:

SHORTEST TIME FOR TRUCK TO TLOW DOWN



+ MU, + X IMP - 7 = MUZ

 $v_1 = 60 \text{ me/h} = 88 \text{ ft/s}$ N=Wcose W=Mg $v_2 = 20 \text{ he/h} = 29.33 \text{ ft/s}$ F= $\mu_s N = \mu_s N \cos \theta$

bix 8 8 ft) HM 132.2 ft) (t) (SIN 229°) - (0.60) (M) (32.2 ft) (COS 2.29°)(t)
= (W) (29.33 ft)

13.128



GIVEN:

INITIAL BOAT

SPEED=

V_= 8 .n./h.

BOAT SPEED

10 SEC AFTER

1.1 INNANKER

1.2 VKISED=

V_= 12 INV/N.

W= 980 ID

FIND:

NET FORCE PROVIDED EYTHE SPINNAKER OVER THE 10 SEC. IN TER-AL

02= 12 mc/h= 17.60 ft/s

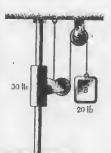
M. U, + TMP = MUZ

 $m (11.73 \text{ ft/s}) + F_N (10 \text{ s}) = m (17.60 \text{ ft/s})$ $F_N = \frac{(1801)(17.60 \text{ ft/s} - 11.73 \text{ ft/s})}{(32.2 \text{ ft/s}^2)(10 \text{ s})} = 17.86 \text{ lb}$

NOTE:

FN IS THE NET FORCE PLOVIDED BY THE SAILS. THE FORCE ON THE SAILS IS ACTUALLY SREATER AND INCOURSES THE FORCE NELDED TO INVERCOISE THE WATER RESISTANCE ON THE HULL.

13.129



GIVEN:

SYSTEM RELEASED FROM

FIND:

TIME FOR ATTO REACH A VELOCITY OF 2 Ft/s

KINEMATICS

LENGTH OF CABLE TO CONSTANT L= ZXA+XB dL = ZVA+VB = 0

NB= -2 VA

Ke.



$$\frac{1}{11} \int (M_A V_A)_2 \frac{(V_A)_2 = 2 \text{ ft/s}}{M_A = \frac{M_A}{9} = \frac{30}{9}}$$

$$\begin{array}{c|c} W_{A} \downarrow & & \\ & \downarrow & \\$$

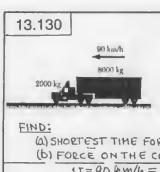
$$[A][mv]_{F_0}$$
 $(T-15)t_{1-2}=\frac{30}{9}$ (1)

 $|| M_{B} V_{B} ||_{1} = 0 \qquad || M_{B} = \frac{20}{9} = \frac{20}{9}$ $|| T_{t_{1}-2} V_{B} ||_{2} = 2(V_{A})_{2} = 4 \text{ ft/s}$ $|| W_{B} V_{A} ||_{2} = 1 \text{ ft/s}$

ADD EQ (1) AND (2) (ELIMINATING T)

$$(20-15)(t_{1-2}) = \frac{(30+80)}{9} = \frac{110}{9}$$
$$t_{1-2} = \frac{22}{32\cdot2} = 0.6835$$

t=0.683 sec.



m_= 2000 Ag

MT = 8000 Ag

INITIAL U= 90 EM/N

FINAL U= 0

TRAILER BRAKES FAIL

45=0.65

(b) FORCE ON THE COUPLING DURING THIS TIME

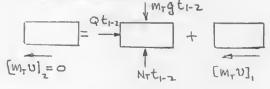
(a) THE SHORTEST TIME FOR THE RIG TO COHE TO A STUP WILL BE WHEN THE FRICTION FORCE ON THE WHEELS IS MAXIMUM. THE DOWNWARD FORCE EXERTED BY THE TRAILER ON THE CAB IS ASSUMED TO BE ZERO. SINCE THE TRAILER BRAKES FAIL ALL OF THE BRAKING FORCE IS SUPPLIED BY THE WHEELS OF THE CAB, WHICH IS HAXIMUM WHEN THE WHEELS OF THE CAB ARE AT IMPENDING SLIDING.

$$[(M_c+M_T)V]_2 = 0$$

$$N_c = \frac{1}{N_T + 1} + \frac$$

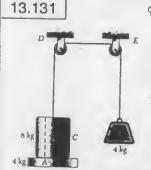
$$0 = -(0.65)(2000 \text{ kg})(9.81 \text{ m/s}^2)(t_{1-2}) = (10000 \text{ kg})(25 \text{ g})$$

(b) FOR THE TRAILER



$$(m_T v)_2 = -Q t_{1-2} + [m_T v]_1$$

$$0 = -Q(19.60 s) + (8000 lg)(25 m/s)$$



GIVEN:

MB= 4 kg

MC= 8 kg

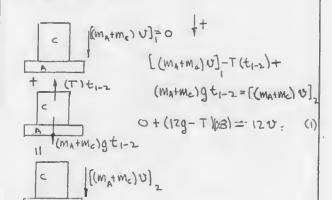
SYSTEM IS RELEASED

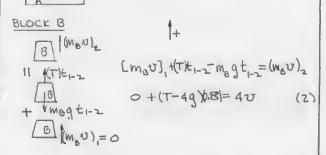
FROM REST

FIND:

(a) VELOCITY OF
BLOCK B AFTER
B SEC.
(b) FORCE EXERTED
BY C ON A

(a) BLOCKS A ANDC





ADDING (1) AND (2) (ELIMINATING T)

$$U = \frac{(8 \text{ kg})(9.81 \text{ m/s}^2)(0.85)}{16 \text{ kg}} = 3.92 \text{ m}$$

$$U_B = 3.92 \text{ m}$$

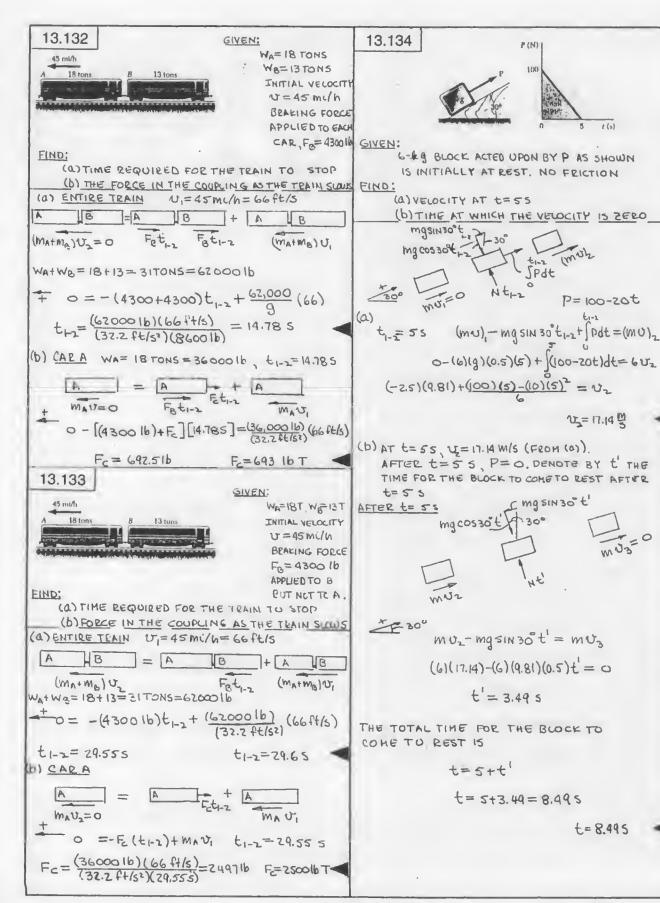
(b) COLLAR A ++

$$Tt_{1-2} \downarrow F_{2}t_{12} \qquad FROM EQ(2) WITH U=3.92 \frac{M}{2}$$

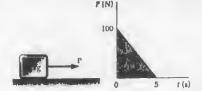
$$T=\frac{4 U}{0.8}+49$$

$$T=\frac{(4 Ag)(3.92 \frac{M}{2})}{(0.85)}+(4 kg)(9.81 \frac{M}{2})$$

T= 58.84 N SOLVING FOR FC IN (3): Fc=(4 kg)(3.92 kg)-(4 kg)(9.81 kg2)+58.84 N=39.2 N (0.85)







A 6-LY BLOCK IS ACTED UPON BY THE FORCE P AS SHOWN AND IS INITIALLY AT REST. COEFFICIENTS OF FRICTION, 4=0.60 4=0.45

EIND:

(a) VELOCITY OF THE BLOCK AT t=5 S (b) HAXIHUM YELOCITY OF THE BLOCK

(a)

CHECK TO SEE IF THE BLOCK MOVES WHEN P

$$F_s = 45N$$
 $F_s = 45N$
 $F_s = 45N$
 $F = (0.60)(64g)(9.81 \frac{M}{52})$
 $F = 35.3 N$

SINCE 35.3 N IS LESS THAN THE INITIAL VALUE OP P=100 N, THE BLOCK HOVES.

P= 100-20t t1-2=55 F=4kmg=(0.45)(6)(g)

$$mv_i = \int_0^{\infty} Pdt - Ft_{i-2} + mv_z$$

 $0 = \int_0^{\infty} (100 - 20t) dt - (0.45)(6)(9.81)(5) = 6.02$

0= 500-250-132.4 +6 Uz

U2=19.59 m/s €

(b) DETERMINE TIME AT WHICH THE VENOCITY

IS A HAXIMUM, WHICH HUST OCCUR AT t<55

0 = (100-20t)dt-(0.45)(6)(9.81) t+60

du=0; 100-20t-26,49=0

t= 3.68 s WHEN U IS MAXIMUM

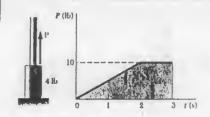
SUBSTITUTE t=3.685 IN EQ.(1)

0=(100)(3.68)-10(3.68)-97.47+6UHAX

U= 134.67 = 22.45 W/S

VHAX= 22.5 M/S

13.136



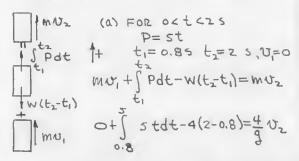
GIVEN:

A 4-16 BLOCK IS ACTED UPON BY THE FORCE P AS SHOWN AND IS INITIALLY AT REST NO FRICTION

FINC:

(a) YELOCITY AT t= 25 (b) VELOCITY AT t= 35

THE BLOCK DOES NOT HOVE UNTIL P=416
FRON t=0 TO t= 25 P= 5t
THUS, THE BLOCK STARTS TO HOVE WHEN
t=4/5=0.85



 $\sqrt{2} = \frac{(32.2 \frac{f+}{52})}{4(1b)} \left[\left(\frac{5}{2} \frac{1b}{5} \right) \left[(25)^{2} (0.8 \text{ s}) \right] - (41b)(25-0.85) \right]$

U2= 28.98 ft/s

· No = 29.0 ft/s

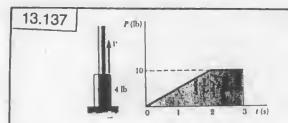
(b) FROM t=25 TO t=35

| MU3 $v_2 = 29.0 \text{ ft/s}, \text{ Feom(a)}$ | P= 10.16 $25 \le t \le 35$ | The second is the second in t

 $V_3 = (29.0 \text{ ft/s}) + \frac{(32.2 \text{ ft/s}^2)}{(4 \text{ lb})} [(6 \text{ lb})(1 \text{ s})] = V_3$

U3= 29.0+48.3=77.3 ft/s

U3= 77.3 ft/s



COLLAR INITIALLY AT REST IS ACTED UPON B A FORCE P(16) AS SHOWN. NO FRICTION

EIND:

(1) THE HAXIMUM VELOCITY OF THE COLLAR, UHAX (b) THE TIME WHEN THE VELOCITY IS ZERO.

(1) DETERMINE TIME AT WHICH COLLAR STARTS TO HOVE

P=st o<t<2s COLLAR HOVES MHEN P=416 02 = $\frac{P}{5}$ = $\frac{4}{5}$ =

MN, + $\int Pdt$ - $\int Wdt = MN_2$ H = $\int Pdt$ FOR t<25 P=5t (16)

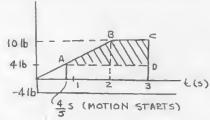
Wdt

25<t<35 P=1016

HNU=0

t>35 P=0

FOR t<3s W=41b
THE HAVINUM MELOCITY OCCURS WHEN THE
TOTAL IMPULSE IS MAYINUM.



 $AREA_{ABCD} = HAX IMPULSE = \frac{1}{2}(61b)(\frac{6}{5}S)+(61b)(1S)$

(b) VELOCITY 13 ZERO WHEN TOTAL IMPULSE
15 ZERO AT THAT

DR 45<+<35, IMPULSE=9.6(16.5), PART (0)

FOR At BEYOND 35 IMPULSE = - 4 Dt (16.3)

TOTAL IMPULSE = $0 = 9.6 - 4\Delta t$ $\Delta t = 2.45$

TIME TO ZERO VELOCITY t= 35+2.45=5.45

13.138 p (MPa) p₀

GIVEN:

20-9 BULLET
10 MM DIAMETER RIFLE:
BARREL
EXIT VELOCITY OF THE
BULLET = 700 M/s
TIME BULLET TO EXIT
= 1.6 Ms
VARIATION OF PRESSURE

AS SHOWN

FIND:

- t (ms)

P(MA) or P= C-Gt AT t=0 p=-po=C1-C2(p) C= 100 AT t= 1.6x103s 10=0 - E(ms) 0=C1-C2(1.6x1035) 1.6×10 3 C2= P0/(1.6x1035) A pdt m=20×103 kg MU,=0 1.6×1035 $A = \pi (10^{-3})$ O+A pdt= MUz A= 78.54 X10 m2 1.6×10 3 0+ A (C,-c2+)dt= 20x103 (78.54×10 6 M2) (C,)(1.6×103)-((2)(1.6×103))= (20x103kg)(100 m/s) 1.6×103C1-1.280×10C2=178.25×10 (1.6x103 m2.5) Po-(1.280x106 m2.52) Po= 178.25x102 kg·m -Pn= 222.8 × 10 6 N/M2

13.139 GIVEN:

25-9 BULLET, 10 mm DIA. PIFLE BARREL EXIT VELOCITY = 520 m/s

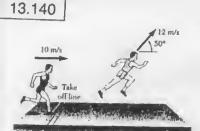
TIME FOR BULLETTO EXIT= 1.44 ms

PLESSURE HODEL

P(t)=(950 HPa)(e

Po= 223 MPa

"/ ERROR IF GIVEN EQUATION FOR PLE) IS USED
TO CALCULATE THE EXIT VELOCITY



INITIAL VELOCITY AT TAKEOFF =10 m/s. VELOCITY AFTER TAKEOFF = 12 m/s AT 58 THPACT TINE

FIND:

= 0.135. JEKTICAL COMPONENT OF THE AVERAGE IMPULSIVE FORCE ON ATHLETES FOOT FROM THE GROUND. (INTERMS OF HIS WEIGHT W)

MV; +(P-W) Dt = MUZ Dt = 0.185

VERTICAL COMPONENTS 0+(Pr-W)(0.18)=(W/g)(12)(SIN 50°)

PV = 6.21W

GIVEN:

13.141

Landing pit

YEL OCITY BEFORE LANDING = 30ft AT 350 IMPACT TIME BEFORE COHING TO A 510P == 0.22 5 WEIGHT=185 D

FIND:

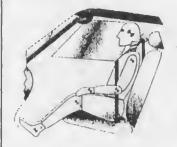
HORIZONITAL COMPONENT OF THE AVERAGE IMPILE 'YE FORCE ON THE ATHLETES FEET

MU, + (P-W) At= M1/2

HORIZONTAL COMPONENTS 以(20)(cos 357) - P4(0,22)= 0

Pu= 642 16

13.142



F (16)

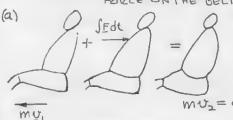
GIVEN:

AUTOHOBILE TRAVELING AT 45 MU/h COKES TO A STOP IN 110 MS. FORCE ACTING ON MAN AS SHOWN HANS WEIGHT = 200 lb

FIND:

(a) AVERAGE IMPULSIVE FORCE EXERTED ON THE BELT AS SHOWN

(b) HAXIMUM FORCE FM EXERTED ON THE BELT FORCE ON THE BELT IS OPPOSITE



TO THE DIRECTION SHOWN

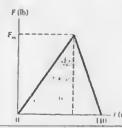
m 12 = 0

N= 45 mi/h= 66 ft/s W= 200 b

MU, - JEdt = MU2 | Fdt = FAVE At

(2001b) (66 fl/s) - FAVE (01105)=0 At= 0.1105

FAVE = (200)(66) = 3727 lb FAVE = 3730 lb (b)



IMPULSE = AREA UNDER F-t DIAGRAM= _ F. (110'S)

FROM (a), IMPULSE= FAVE Dt= (3727 16) (1105)

= Fm (.110)=(3727)(.110) Fm = 7450 lb

13.143 GIVEN:

1.602. GOLF BALL HAS A VELOCITY OF 125 ft/s AFTER IMPACT DURATION OF IMPACT = to=0.5ms FORCE DURING IMPACT F= F_SIN(114)

FIND: HAXIMUM FORCE FM. ON THE BALL 2=125 ft/s MU2 MV,=0 ot | Em 21 MIT qt = (16/16) (152) MU,+ Fdt=MU2 Fm= 122016

13,144





GIYEN:

15-RQ SUITCASE A 40-LQ LUGGAGE CARRIER B INITIAL VELOCITY OF CAPRIER, UB = 0.8 M

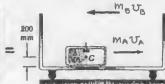
LIND:

(a) VA/B

- (b) UB, AFTER THE SUITCASE HITS THE RIGHT SIDE OF THE CAPRIER WITHOUT PCEOUSED
- (C) ENERGY LOST BY THE "PACT OF THE SUITCASE ON THE FLOOR OF THE CAVELER

(a) SINCE THERE ARE NO EXTERNAL FORCES ACTING ON THE SYSTEM OF THE SUITCASE A AND THE LUGGLEC CARRIER & E. IN THE HORIZONTAL DIRECTION, LINEAR MOHENTUM IS CONSERVED





(MATMO) U= MAUA+MUG N=0 NB=-0.8 m/s NA=VAID+NB MB=40kg MA= 15-R9 0 = (15 kg) (UA/G-0.8 m/s)+40 kg (-8 m/s)

VA/1= (40 kg) (0.8 m/s) + 0.8 m/s = 293 m/s (15 kg)

UA/B = 2.93 M/S -

(b) MOMENTUM IS CONSERVED BEFORE AND AFTER THE SUITCASE HITS THE LUGGAGE CAPRIER



MAUA +MBUB = (MA+MB)U

IT'= MAVA+MOUB (MA+MB)

FROH (a)

NA=NA/8+NB= 2.43-0.8=2.13 m/s U'=(15)(2.13)-(40)(0.8)=0 U'=0

(c) BEFORE SUITCASE FALLS , E = MAG (.7M) AFTER SUITCASE HITS THE BOTTOM OF THE CARRIER E2= 1 MA Up2+1 MB UB2+MAG (0.200M)

E= 15 g (17) ENERGY LOST DE = EI-EZ AE, =(15)(9.81)(0.7)-\(\frac{1}{2}(15)(2.13)^2-\(\frac{1}{2}(40)(0.8)^2-(15)(9.81)(0.7) DEL= 26.7 J

13.145



GIVEN:

LEFORE COUPLING, ZO-mg CAR IS TRAVELING 1- Skin, A AS SHOW! 40-M9 CAR HAS ITS WHEELS LOCKED

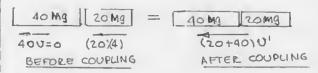
FIND:

(a) VELOCITY OF BOTH CARS IMMEDIATELY AFTER COUPLING

4x=0.30, 10-mg CAR OILY

(b) THE TIME FOR BOTH CARS TO COME TO REST

(a) THE HOMENTUH OF THE SYSTEM CONSISTING OF THE TWO CARS IS CONSERVED INHEDIATELY BEFORE WID WITER COUPLING.



IMV = ZWV' 0+(20Mq)(4&m/h)=(20Mg+40Mq)(U') U = (20)(4) = 1.333 km/h

(b) AFTER COUPLING

$$\frac{60 \text{ Mg}}{60 \text{ U}_{3}=0} = \frac{60 \text{ Mg}}{50 \text{ U}_{1}} + \frac{60 \text{ Mg}}{60 \text{ U}_{1}}$$

THE FRICTION FORCE ACTS ONLY ON THE 40 Mg CAR SINCE ITS WHEELS ARE LOCKED THUS,

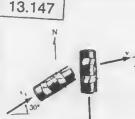
Fr=4 No= (0.30) (40×10 kg) (9.81 M) FC= 117,72 X103 N

FROM (a) 17=v=1.333 km/h=0.3704 m/s

IMPULSE MOHENTUM

ZMVi+ Fedt= ZMV2 (60x10/eg)(.3704 m/s)- (117.72x10 H)dt=0 $t = (60 \times 10^3)(13704)$ = 0.1888 s





WA= 190 16 WB= 1251b RAFT WR = 300 lb VA/0 = 2 ft/3

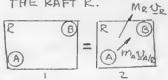
TOWARD B, AFTER THE RAFT BREAKS LOOSE FROM ITS

ANCHOR.

FIND:

(a) SPEED OF THE RAFT, UE, IF B DOGS NOT HOVE (b) SPEED UZ OF B . IF THE RAFT IS NOT TO HOVE

(a) THE SYSTEM CONSISTS OF A AND B AND THE RAFT R.



MOHENTUH IS CONSERVED

Up=0.618 ft/s

(b) FROM EQ (1)

$$O = M_{A}N_{A} + M_{B}U_{B} + O \quad (U_{R} = 0)$$

$$N_{B} = -\frac{M_{A}N_{A}}{M_{B}} \quad N_{A} = V_{A/R} + N_{R} = 2 \text{ ft/s}$$

$$N_{B} = -\frac{(2 \text{ ft/s})(1901b)}{(1251b)} = 304 \text{ ft/s}$$

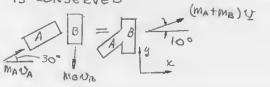
UB= 3.04 ft/s

GIVEN:

MA=1500kg mB=1200 Eg BOTH CARS TOGETHER, SICID AT 10° NORTH OF EAST AFTER IMPACT

FIND: (A) WHO WAS GOING FASTER (b) SPEED OF THE FASTER CAIR IF SLOWER CAR WAS GOING AT 50 km/in

(a) TOTAL MOMENTUM OF THE TWO CARS IS CONSERVED



ZMU, X: MAU, COS 309= (MA+MA)NICOS 10

ZMU, Y: MAN, SIN 30 - MBUB= (MAHM) USINIO DIVIDING (1) INTO (2)

$$\frac{v_0}{v_A} = (0.4010)(1500)(1500)$$

$$\frac{N_B}{N_A} = 0.434 \qquad N_A = 2.30 \, N_B$$

THUS, A WAS GOING FASTER

(b) SINCE UB WAS THE SLOWER CAR UB = 50 km/h



NOTHER AND CHILD TRAVELING AT 7.2 Em/h INITIALLY. Mm=551g Mc=20kg CHILD'S SPEED DECREASES TO 3.6 km/h IN 3 3 AS THE HOTHER PULLS ON THE ROPE

FIND:

(a) MOTHERS SPEED AT THE END OF THE . 3 S INTERVAL

(b) Average value of the tension in THE BODE DURING THE 35 INTERVAL

(a) Consider Hother and Child as a Single system. Assuming the Friction Force on the Jris is negligible Homentum is conserved

$$(m_{c}v_{c}) \qquad (m_{H}v_{H}) \qquad (m_{c}v_{c}') \qquad (m_{H}v_{H}')$$

$$m_{c}v_{c} + m_{H}v_{H} = m_{c}v_{c}' + m_{H}v_{H}'$$

Vc=0n=7.2 km/h Nc=3.6 cm/h

(20)(7.2)+(55)(7.2)=20(3.6)+(55)(UH)

Vn=8.51 km/h

(b) CHILD ALONE

t=35

McVe - FArt = McVe

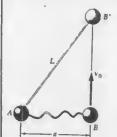
N= 7.2 km/h= 2 m/s N= 3.6 km/h=1 m/s

$$(zolg)(zm/s) - F_{AV}(3s) = (zolg)(1m/s)$$

 $F_{AV} = \frac{(zolg)(1m/s)}{(3s)} = 6.67 lg.m/s^2$

FAV = 6.67 N

13.149



GIVEN

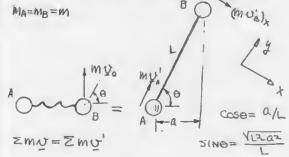
A AND B ON A HORIZONTAL FRICTION LESS PLANE
ARE ATTACHED BY AN INEXTENSIBLE CORD OF LENGTH L
MASS OF A = MASS OF B

NB = Vo , VA = O
INITIALLY

FIND:

(a) U_a and U_b after the cord becomes taut (b) the energy lost as

(a) FOR THE SYSTEM CONSISTING OF BOTH
BALLS CONNECTED BY A CORD THE TOTAL
HOMENTUM IS CONSERVED (Mu').



 $X: -mN_0\cos\theta = m(V_B^i)_X \qquad (1)$ $(V_B^i)_X = -N_0\cos\theta = -V_0\frac{\alpha}{L}$

y: $m v_0 SIN \Theta = m v_A^{\dagger} + m (v_B^{\dagger})_y$ (2)

SINCE THE CORD IS INEXTENSIBLE

OA' = (Ub)y (3)

THUS FROM (2) VOSINO = 2VA

FROM (3)
$$(V_{B}^{1})_{y} = V_{A}^{1} = (V_{0}/2L)\sqrt{L^{2} - \Omega^{2}}$$

$$V_{B}^{1} = \sqrt{(V_{0}^{1})_{x}^{2} + (V_{0}^{1})_{y}^{2}} = V_{c}\sqrt{\frac{n^{2}}{L^{2}}} + \frac{(L^{2} \cdot \Omega^{2})}{4 \cdot L^{2}}$$

$$V_{B}^{1} = (V_{0}/2L)\sqrt{L^{2} + 3\Omega^{2}}$$
(b)

INITIAL $T = \frac{1}{2} m n J_0^2$ $T' = \frac{1}{2} m (U_A)^2 + \frac{1}{2} m (U_B)^2 = \frac{1}{2} m (U_0 / 2L)^2 [(L^2 a^2) + (L^2 3 d)]$ $T' = \frac{1}{2} (M U_0^2 / 4 L^2) (Z L^2 + 2 a^2) = (M U_0^2 / 4 L^2) (L^2 + a^2)$ $\Delta T = T - T' = \frac{1}{2} m U_0^2 - (M U_0^2 / 4 L^2) (L^2 + a^2)$

DT=(mvo/412)(12-02)



-0.5 m

GIVEN:

2-kg SPHERE CONNECTED BY AN INEXTENSIBLE CORD OF LENGTH 1.2M TO POINT O ON A HORIZONTAL FRICTIONLESS PLANE INITIAL VELOCITY VO PEPPENDICULAR TO OA

FIND:

MAXIMUM ALLOWABLE Vo IF IMPULSE OF THE FORCE EXECTED ON THE COLD IS NOT TO EXCEED 3N.S.

FOR THE SPHERE AT A INHEDIATELY BEFORE 1: L AFTER THE CORD BECOHES TAUT

$$mv_{o}$$
 | 0 + 0 = 0
 $\Delta e = \cos^{-1}(.5)/(1.2) = 65.38^{\circ}$

MNO + FAt = MYA

10 MUSCINO-FLt=0 FAt= 3 N.S 11.= 2-20 3 N.S. (22.0 XSIN 65.38°)

Uo= 1.650 m/s

13.151

GIVEN:

MASSES ALLINITIAL VELOCITY OF THE BALL AS SHOWN, NO ENERGY LOST IN THE IMPACT



EIND:

(A) VELOCITY OF THE BALL IMMEDIATELY AFTER IMPACT (b) IMPULSE OF THE FORCE EXECTED BY THE PLATE ON THE BALL

(a) FOR THE SYSTEM VIHICH IS THE ZALL AND THE PLATE MOHENTUH IS CONSERVED

ALL FORCES ARE NON-IMPULSIVE EXCEPT THE EQUAL AND OPPOSITE FORCES BETWEEN THE PLATE AND THE BALL

+ (mu)=-(wv')+(mu')p

(:.125kg)(3m/s)=(0.125kg)(vo)+(0.250kg)vp

$$\nabla_{\rho}^{i} = 0.5 N_{B}^{i} + 1.5$$
(1)

SINCE THERE IS NO ENERGY LOST THE KINETIC ENERGY OF THE SYSTEM IS CONERVED

(CONTINUED)

13.151 continued

BEFORE IMPACT, T= 2 MBUB= 2 (0.125 Rg) (3 M/s)=0.5637

AFTER IMPACT T= 1 mg(vi) + 1 mp(vi)2 SUBSTITUTE FOR US FROM (1)

T=2(0.125kg)(v's)3+(2)(0.250kg)[0.50kg+1.5]2

T= 0.09375/06)2+0.1875UB+ 0.2813

T=T 0.563=0.09375(vg)2+0.1675 vg+0.2813

$$v_{6}^{12} + 2v_{6}^{2} - 3 = 0$$
 $v_{6}^{12} = -2 + \sqrt{4+12} = -1 + 2 = -3$

(Na=-3 m/s BEFORE IMPACT) Va=1 m/s

(b) BALL ALONE

+ (0.125kg)(-3m/s)+FDt = (0.125kg)(1 m/s)

FAt=0.5 N.S

13,152

MM MARA

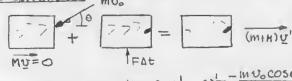
GIVEN:

BULLET FIRED INTO THE BLOCK AS SHOWN

FIND:

UNE JATHOSISON VERTICAL COMPONENTS OF THE IMPULSE ON THE BULLET.

FOR THE SYSTEM WHICH IS THE BULLET AND THE BLOCK, MOHENTUM IN THE HODIZONTAL DIRECTION IS CONSERVED



U=-INVocose -mvocose=(H+m)U (M+m)

BULLET ALONE

- muocoso + Px ot = mu Px At = mv cose [1 - M+m]

+1-muosino+PyAt= 0

 $P_{k} \Delta t = \frac{m M}{m+H} v_{o} \cos \theta$

Py At = MUOSINO

13.153



GIVEN:

RIGID BEAM WEIGHS ZAO Ib BLOCK WEIGHS 60 16 INITIAL VELOCITY OF THE SLOCK = O AND IT IS DROPPED FROM 5ft.

FIND:

INITIAL IMPULSE EXERTED ON THE CHAIN AND THE ENERGY ABCORBED BY THE CHAIN IF THE SUPPORTING COLUMNS ARE, (a) PIGID, (b) EQUIVALE NT TO TWO ELASTIC SPRINGS

13.154



GIVEN:

WB= 5-02 INITIAL SPEED OF THE BALL = 90 mi/h AVERAGE SPEED OF THE GLOVE DURING IMPACT = 30 ft/s OUER A 6 In. DISTANCE

FIND:

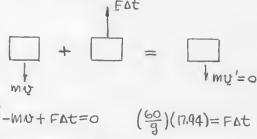
AVERAGE IMPULSIVE FORCE EXERTED ON THE PLAYERS HAND

$$F_{AV} = \frac{mv}{t} = \frac{(5/16 \text{ lb})(132 \text{ fVs})}{(32.2 \text{ fH/s}^2)(1605)} = 76.9 \text{ lb}$$

VELOCITY OF THE BLOCK JUST BEFORE INDACT Ti=0 Vi=Wh=(601b)(5ft)=300 1b.ft 72= 5MU2 V2=0 T+1 V= T2+12 0+300= \frac{1}{2}(\frac{60}{9}) U^2

U=V(600)(32.2)/60 = 17.94 ft/s

(a) PIGID COLUMNS



FAt= 33.43 lb.5 ON THE BLOCK FAt= 33.4 16.5

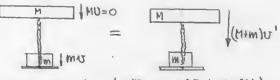
ALL OF THE KINETIC ENERGY OF THE BLOCK IS ABSORBED BY THE CHAIN

T= 1 (60) (17.94) = 300 (t.16

E=300 ft.16

(b) ELASTIC COLUMNS

HOHENTUH OF SYSTEM OF BLOCK AND BEAM 15 CONSERVED



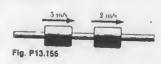
 $V' = \frac{m}{m+M}$ $V = \frac{60}{300} (17.94 \text{ ft/s})$ v= 3.59 ft/s

REFERRING TO FIGURE IN PART (a) -mo+FAt=-mo!

FAt = m(v-v') = (60/9)(17.94 - 3.59) = 26.7 lb.5

E= 1 mu2 - 1 mv2 - 1 Mv2 60 [(11.44)2-(3.59)] - 240 (3.59)2 E= 240 fts

13.155



GIVEN:

IDENTICAL COLLARS WITH VELOCITIES AS SHOWN. C=0.65, M=1.2 kg NO FRICTION

FIND:

W. U, ANDUB AFTER IMPACT (D) ENERGY LOST DURING IMPACT

(A) TOTAL MOHENTUM IS CONSERVED

$$U_A = 5 \text{m/s}$$
 $U_B = 2 \text{m/s}$ U_A' U_B' $U_B = 1.2 \text{kg}$

+ (+5 m/s)+(+ 2 m/s) = W VA+ M VB 7M/s=NA+NO

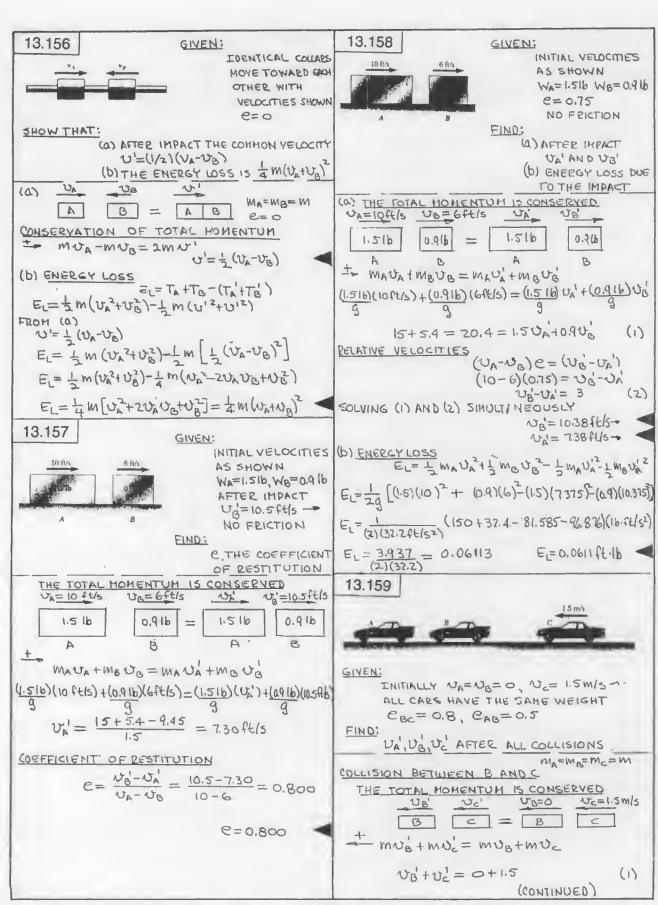
RELATIVE VELOCITIES ALONG LINE OF IMPACT U' - N' = e (VA-NB) e= 0.65

UB'- UA'= (0.65) (5 m/s-2 m/s) = 1.95 m/s (2)

ADDING (1) AND (2) 200 = 8.95 00=4.48 5 FROM (1) WITH UR=448M/S

Un= 7m/5-4.48m/5 = 2.53m/5-(b) ENERGY LOST DURING IMPACT

EL=TA+TA-TA-TB EL= = (1.2kg)[52+22-(4.475)2-(2.525)2] EL = 1.559 N.M



13,159 continued

RELATIVE VELOCITIES

$$(U_{B}-v_{c})(e_{Bc}) = (v_{c}'-v_{b}')$$

$$(-1.5)(0.8) = (v_{c}'-v_{b}')$$

$$-1.2 = v_{c}'-v_{b}'$$
(2)

SOLVING (I) AND (Z) SINULTANEOUSLY

Un'= 1.35 m/s

U-= 0.15 M/S-SINCE UD > UC, CAR B COLLIDES WITH CAR A COLLISION BETWEEN A AND B

$$V_A^1 + V_B^{\parallel} = 0 + 1.35$$
 (3)

RELATIVE YELOCITIES

$$U_{A}^{i} - U_{B}^{ii} = 0.675$$
 (4)

SOLVING (3) AND (4) SIMULTANEOUSLY

UA= 1.013 m/s -UB= 0.338 m/5_

SINCE NE NB CUA THERE ARE NO FURTHER COLLISIONS

13.160

GIVEN:

SPHERES A, B, C OF EQUAL WEIGHT INITIAL VELOCITY OF A IS VO AND BAND C ARE AT REST. E IS THE SAME FOR ALL SPHERES

EIND:

(a) U' AND UB AFTER THE FIRST COLLISION (b) UB AND UC AFTER

THE SECOND COLLISION

(C) FOR IN SPHERES, THE VELOCITY U' AFTER IT IS HIT FOR THE FIRST TIME

(d) USING THE RESULT FROM PART (C) THE

VELOCITY OF THE LAST SPHERE FOR 11=6 P=015

(a) FIRST COLLISION (BETWEEN A AND B)

THE TOTAL HOMENTUM IS CONSERVED ひゃしゃ ひゅ=0

OO=OO MUA+MUB=MUA'+MUB

13.160 continued

RELATIVE VELOCITIES

$$(U_A - V_B) \mathcal{C} = (V_B' - V_A')$$

$$V_O \mathcal{C} = V_B' - V_A'$$
(27)

SOLVING EQUATIONS (1) AND (2) SIMULTANEOUSLY

Un = U0(1-6)/2 UB=Vo(1+e)/2

(b) SECOND COLLISION (BETWEEN BAND C) THE TOTAL HOHENTUH IS CONSERVED NB' UC=O UB" UC!

(B) (C) = (B) (C)

MUB+MUC-MUB+MUC

USING THE RESULT FROM (A) FOR UR

$$V_0(1+e)/2 + 0 = V_B^{"} + V_C^{"}$$
 (3)

PELATIVE YELOCITIES

SUBSTITUTING AGAIN FOR N' FROM (a)

$$v_0 \frac{(1+e)(e)}{2} = v_c - v_B^{"}$$
 (4)

SOLVING EQUATIONS (3) AND (4) SIMULTANEOUSLY

(C) FOR N SPHERES

$$\begin{array}{c|cccc}
 & \nabla_{n-1} & \nabla_{n} & \nabla_{n-1} & \nabla_{n} \\
\hline
 & \bigcirc & \bigcirc & \bigcirc
\end{array}$$

N-1 TH COLLISION 0 0 = 0 0 n-1 n n-1 n

WE NOTE FROM THE ANSWER TO PART

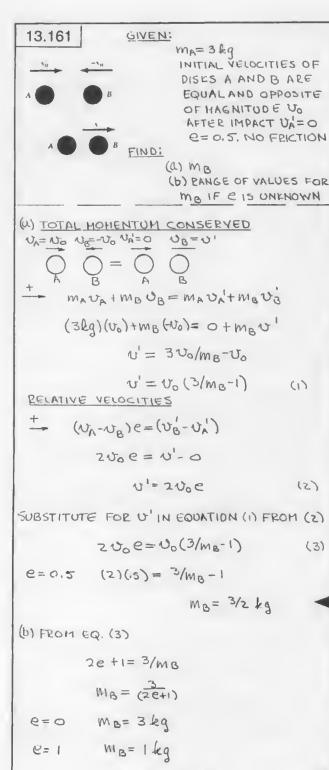
(b), WITH n=3 $v'_{n}=v'_{3}=v'_{c}=v_{0}(1+e)^{2}A$ OR N'2 = No (He)(3-1)/(2 (3-1)

THUS FOR N BALLS

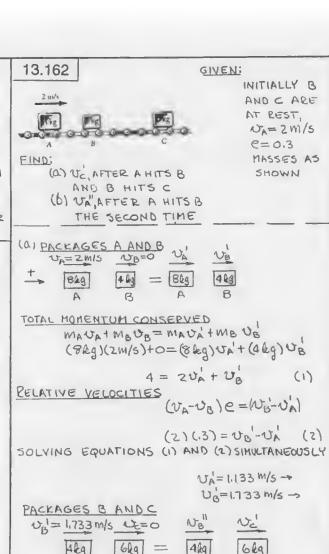
(d) FOR N= 6 AND @= 0.95

FROM THE ANSWER TO PART (C) WITH N=6

N,=0.881Vo



1kg < mo < 3kg



PACKAGES B AND C $v_{B}^{1} = 1.733 \, \text{m/s} \quad v_{E} = 0 \quad v_{B}^{11} \quad v_{C}^{11}$ 4kg 6kg = 4kg 6kg

+ m_B v_B + m_C v_C = m_B v_B + m_C v_C

(4kg)(1.733 m/s) + 0 = 4 v_B + 6 v_C

6.932 = 400 + 600 (3)
PELATIVE VELOCITIES (UB-NE) PE UC - UB

(1.733)(1.3)=0.519=02-04 (4)

SOLVING EQUATIONS (3) NON SIMULTANFOUSLY

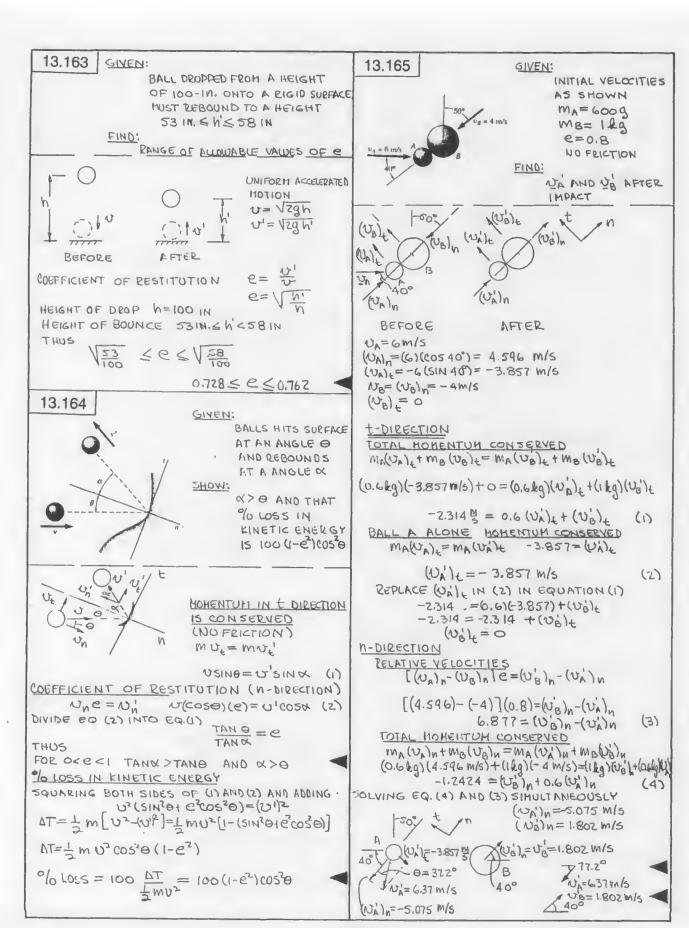
Uz=0,901M/5-

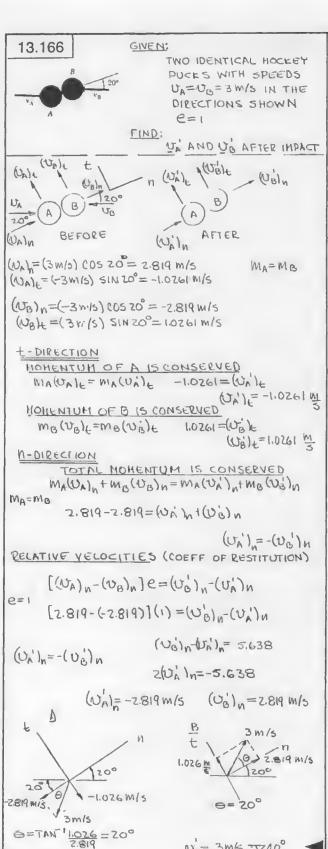
(8)(1.133)+(4)(0.381)= 8 UA"+4 UB"

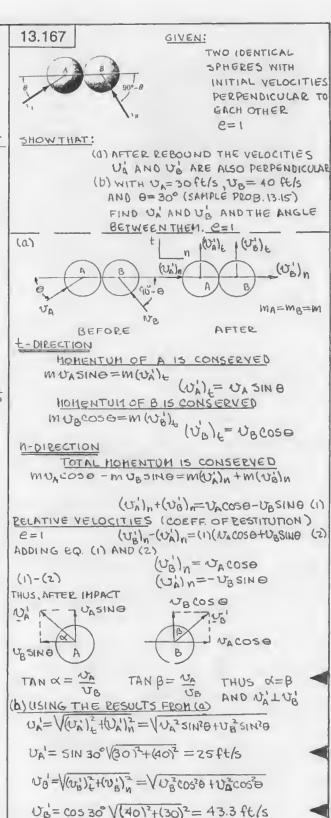
10.588 = 804"+406" (5)

 $(V_A' - V_B'') = V_B'' - V_A''$ $(1.133 - 0.381)(0.3) = 0.727 = U_B'' - V_A''$ (6) 50LYING (5) AND (6) SIMULTANEDOSLY,

UA = 0.807 W/S ->



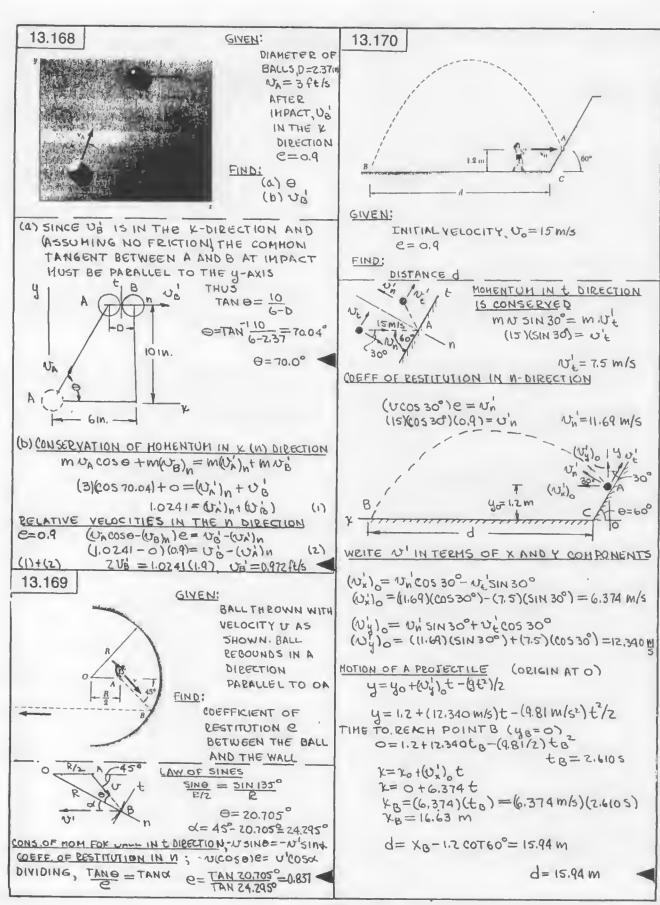


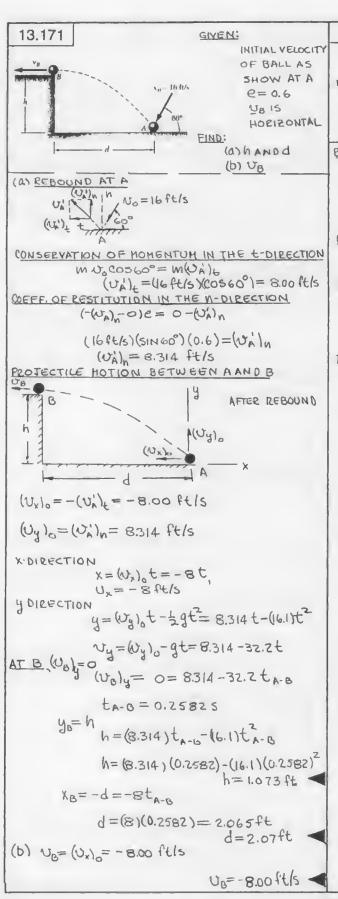


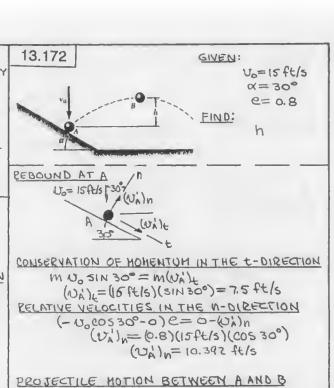
(= β = TAN UA = TAN 30 = 36.9°

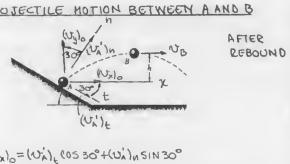
×90-β 8= 180-(α+90-β)] = 90°

NA 3M/5 740°

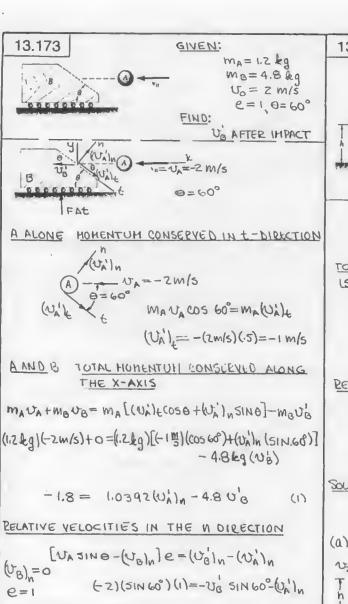


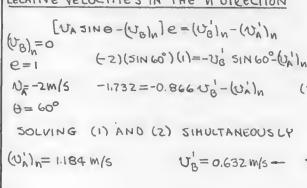


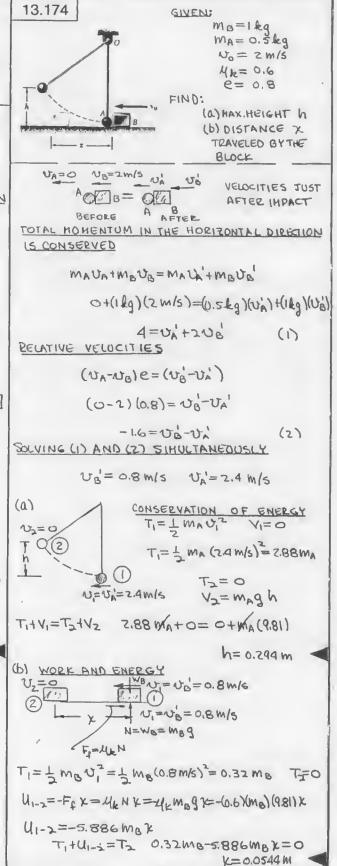




 $(v_{x})_{o} = (v_{h})_{t} (os 30^{\circ} + (v_{h})_{h} sin 30^{\circ}$ $(v_{x})_{o} = (7.5)(cos 30^{\circ}) + (10.392) sin 30^{\circ}$ $(v_{x})_{o} = 11.691 \text{ ft/s}$ $(v_{y})_{o} = -(v_{h})_{t} sin 30^{\circ} + (v_{h})_{h} cos 30^{\circ}$ $(v_{y})_{o} = -(7.5)(sin 30^{\circ}) + (10.392) cos 30^{\circ}$ $(v_{y})_{o} = 5.2497 \text{ ft/s}$ $X \text{ DIRECTION } X = (v_{x})_{o} + v_{x} = (v_{x})_{o}$ $X = 11.69 + v_{x} = 11.69 \text{ ft/s} = v_{B}$ $y \text{ DIRECTION } y = (v_{y})_{o} + \frac{1}{2} \text{ gt}^{2}$ $v_{y} = (v_{y})_{o} - \text{gt}$ $AT A \quad v_{y} = 0 = (v_{y})_{o} - \text{gt}_{AB}$ $t_{AB} = v_{y} \log \frac{1}{3} = \frac{5.2497 \text{ ft/s}}{32.27 \text{ ft/s}^{2}}$ $t_{AB} = 0.1630 \text{ s}$









m=m=1.5kg INITIALLY. U=5 m/s, U=0 NO FRICTION (1) e=1 (2) 0=0

FIND:

(a) MAXIMUM DEFLECTION OF THE SPRING

(b) FINAL VELOCITY OF BLOCK A UB UA= 5M/S UB' UA

PHASE I IMPACT

CONSERVATION OF TOTAL HOMENTUM MAVATMOND = MANA + MOVE MA=MB

5+0= NA + VA (1)

RELATIVE VELOCITIES (UA-UB) = (UB-UA)

> (5-0) e= Va-VAL (2)

(S) ONA (I) DNIDDA

5(1+e)=No

SUBTRACTING (2) FROM (1) 5(1-e) = NA

e=1 UB=5 m/s UA=0 e=0 UB=2.5 m/s UA=2.5 m/s

(a) CONSERVATION OF ENERGY PHASE II

ONE B B B A REBONIM

TI= 1 mo (00)2= 1 (1.5kg) (5m/s)= 18.75 J

AT X= XHAX, T2=0; V2= = & (XMAX)= (40)(XMAX)2 TI+1= T2+12

18.75+0= 0+40 Kmax

0=1 THAX = 0.685 M

E=2 BOTH A AND B HAVE THE SAME VELOCITY INITIALLY AT 0 < 2.5 m/sTHUS $T_1 = \frac{1}{2} (m_A + m_B) (V_A)^2 = (3 + 4)(2.5 \text{ m/s})^2$

T= 9.375 J V= 0 AT X=XHAX T=0 V2=1 & XMAX = 40 KHAX

Tity1=T2+Y2 9.375+0= 40x2max

0=0 KHAX = 0.484 M

13.175 continued

(b) e=1, BLOCK B IS RETURNED TO POSITION () WITH A VELOCITY OF 5 W/S - SINCE ENERGY IS CONSERVED, AND IMPACTS BLOCK A WHICH IS AT REST, IN THE IMPACT TOTAL HOHENTUH IS CONSERVED AND PHASE I IS REPEATED WITH THE VELOCITIES OF A AND B INTERCHANCED THUS JA" = 5m/s - AND NO NO SINCE THERE IS NO FRICTION THESE VELOCITIES ARE THE FINAL VELOCITIES OF A AND B

0=1 NA"= 5m/s-

6=0 BLOCKS A AND B ARE RETURNED TO POSITION () WITH THE SAME YELOCITY OF 25 M/S - SINCE LINERGY IS CONSERVED. THERE IS NO ADDITIONAL IMPACT AND THE SPRING SLOWS BLOCK B DOWN AND A AND B SEDARATE WITH A CONTINUING WITH A VELOCITY OF 2.5 M/S TO THE RIGHT

Un"=2.5W/c= " 0=0

13.176



ma=mB=1.5 % INITIALLY UA=5m/5, UB=0

4,6=0,3,45=0.5

FINAL POSITION OF (a) BLOCKA (b) BLOCK B

IMPACT SEE PHASE I OF SPOB 13.175, R=1 AFTER IMPACT NA = 0 NB = 5 Mils

No=5m/s N=0

N= MB9 HAXIMUM DEFLECTION XHAX, OF THE SPEING

WORK AND ENERGY $T_1 = \frac{1}{2} M_0 (V_0^*)^{\frac{3}{2}} = \frac{1}{2} (1.5 \log) (5 m/s)^{\frac{3}{2}} : 8.75 \text{ J}$ T2=0 U== 1-1-xax-J4KMB9dx U1-2=-1 (80) (XHAY)-103)(1.5 kg) (9.81 M) X

T,+U1-2=T2 1875-40 KHAX-4,4145 XHAX =0

XHAX 0.632 M (5) PHASE III PETURN OF B TO POSITION () BEFORE IMPACT WITH A

T2=0, U2-1=+1 8(XMAY)2-4KMB9KHAX

 $T_1 = \frac{1}{2} m_B J_B^{(2)} = (0.75) N_B^{(1)} U_{2-1} = (40)(0.632)^2 (4.4145)(0.632)$ U2-1= 15.977- 2.790= 13.173 0+13.173=(0.75) Ue UB= 4.191 m/s-

AFTER IMPACT WITH A AT POSITION (1)

RELATIVE VELOCITIES

(NH-V/)(e) = (NA-1)B (4.191-0)(1)= UA"-UE (1)

(CONTINUED)

13.176 continued

CONSERVATION OF TOTAL HOMENTUM AT () mauatma u" = maua t maua"

o + 4.191 = va"+va" MA=MB ADDING EQUATIONS (1) AND (2)

2(4.191) = 2UA"

Un = 4.191 m/s FROM 60 (2) NB= 4.191 -4.191=0

(a) PHASE IX (VELOCITY OF B = O AT (1))



T== ma(v) = (0.75 kg)(4.191 m/s)2 N=MAG T=13.173 T U1-3=-4 KMB 9 XA=-(0.3)(1.5/g)(9.81 N)

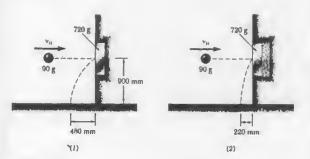
T1+ U1-3=T2 13.173-4.415 KA = 0

FINAL POSITION OF A

(b) NB = O AT IMPACT POINT AND THE SPRING IS UNDEFLECTED AT THIS POINT.

13.177

BLOCK A

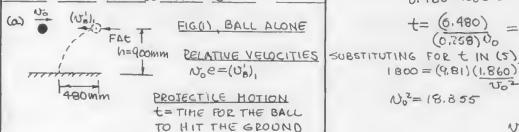


GIVEN:

BALL REBOUNDS AS SHOWN IN FIGURES (1) AND (2), FOAM RUBBER BEHIND PLATE IN (2)

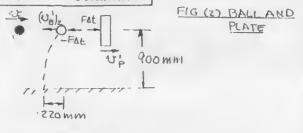
FIND:

(a) COEFFICIENT OF RESTITUTION e BETWEEN THE BALL AND THE PLATE (b) THE INITIAL VELOCITY US



0.480 m = voe t

13.177 continued



RELATIVE VELOCITIES

$$\frac{+}{U_B - U_P} (U_B - U_P) e = U_P + (U_B)_Z$$

$$U_B = U_0 \qquad U_P = 0$$

$$U_0 e = U_P + (U_B)_Z \qquad (2)$$

CONSERVATION OF HOHENTUM

$$\begin{array}{ll}
+ & M_B U_B + M_P U_P = M_B (-U_B')_2 + M_P (U_P') \\
(0.09 \text{ kg}) (U_0) + O = (0.09 \text{ kg}, (-U_B')_2' (0.720 \text{ kg}))_4' \\
U_0 = (-U_B')_2 + 8 U_P' \\
\end{array}$$
(3)

SOLVING (2) AND (3) SIMULTAMEOUSLY FOR

$$(U_{B})_{2} = U_{1}(5(-1))$$

PROJECTILE HOTION

$$0.220 \, \text{m} = V_0 \, (\underline{8e-1}) \, t$$
 (4)

DIVIDING FQ (4) BY EQ. (3)

$$\frac{0.220}{0.480} = \frac{8e^{-1}}{4e}$$

$$4.125e = 8e^{-1}$$

C=0.258

(b) FROM FIG (1)

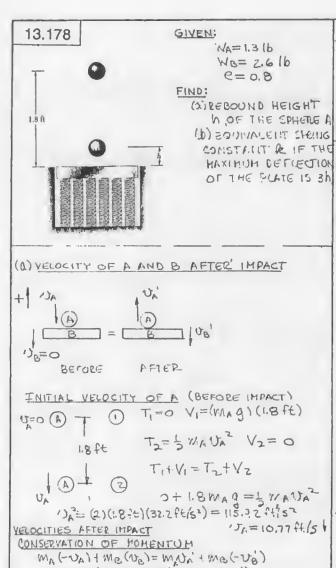
PROJECTILE HOTION r= T 0+5

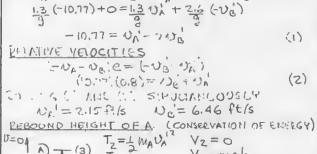
EQUATION (1)

1800 = (9.81) (1.860)

No2=18.855

V0= 4.34 m/s





 $V_4 = \frac{1}{3} k (3)^{\frac{1}{3}}$ $V_4 = \frac{1}{3} k (3)^{\frac{1}{3}}$

13.179 GIVEN:

FIGURE AS SHOW IN 13.178 (LEFT)
WA=1.3 lb, WB= Z.6 lb
EQUIV SINGLE SPRING R=51

FIND:

(a) VALUE OF E FOR WHICH IN IS A HAXIMUM (b) CORRESPONDING VALUE OF IN (C) CORRESPONDING HAXIMUM DERECTION OF IS

(2) INDIAL VELOCITY OF A (BEFORE IMPACT)
FROM SOLUTION TO PROB 13.178, VA = 10.77 ft/s V
VA=10.77 ft/s | A A T.

CONSERVATION OF HOHENTUM

+ MA (-UA)+ MB UB = MA UA'+ MB (-U')

 $\frac{1.3}{9}(-10.77) + 0 = \frac{1.3}{9}v_1' + \frac{2.6}{9}(-v_0')$ $-10.77 = v_0' - 2v_0'$ (1)

RELATIVE VELOCITIES

 $(-N_A - U_B) e = (-N_B - U_A')$ 10,77 e = U_B' + UA'
(2)

SOLVING (1) AND (2) SIMULTANEOUS LY FOR UA

3 NA=(10.77)(2e-1)

h is haxinum when un' is haxinum, that is when e=1

0=1

(b) FOR C=1 VA= 10.77/3 = 3.59 ft/s

FOR A ALONG

UN=0 CONSERVATION OF ENERGY

 $\frac{1}{2} \frac{\sqrt{a}}{9} (v_A^2)^2 + 0 = 0 + \sqrt{a} h$ $h = \frac{1}{2} \frac{(3.59 \text{ ft/s})^2}{(32.2 \text{ ft/s}^2)} = 0.200 \text{ ft}$

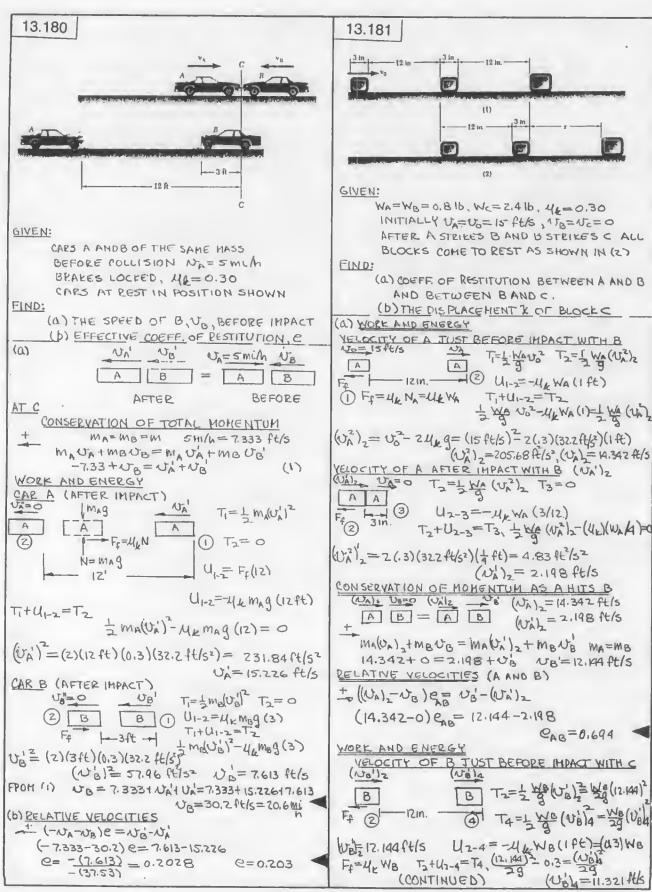
(C) FOR B ALONE

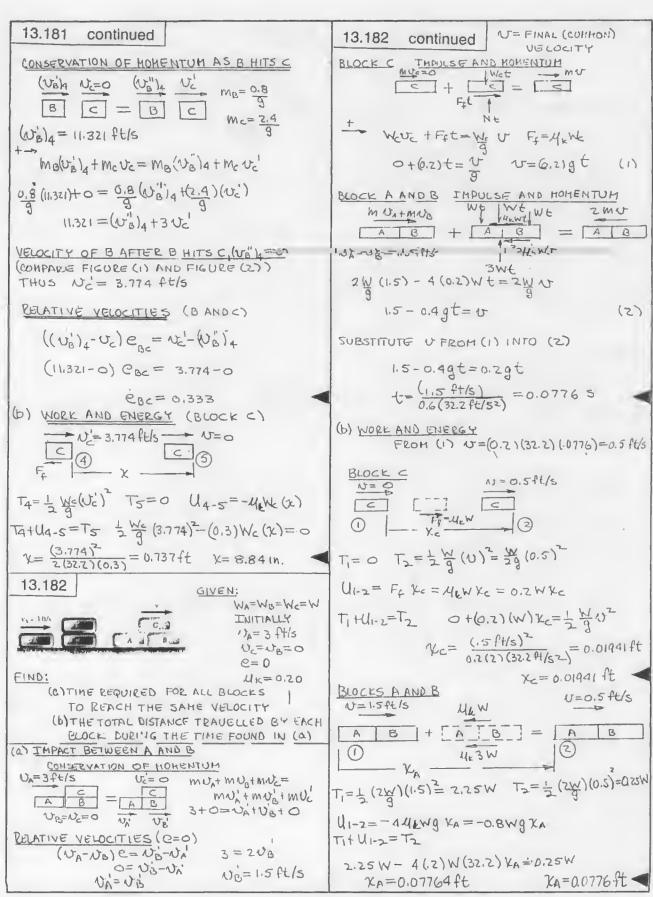
UB CONSERVATION OF ENERGY

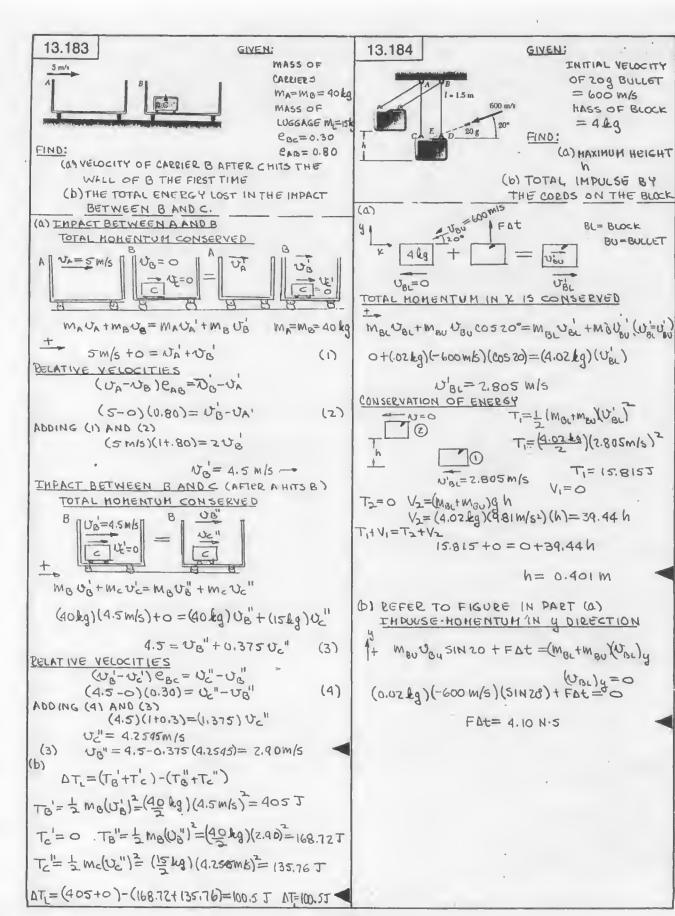
From (a), EQ (2) $V_{B}^{+} = |0.77| - V_{A}^{+}$ $V_{B}^{+} = (10.77)(1) - 3.59$ $V_{B}^{+} = 7.18 \text{ ft/s}$ $V_{B}^{-} = \frac{1}{2} \frac{W_{B}}{9} \left(V_{B}^{+} \right)^{2} = \frac{1}{2} \frac{(2.6 \text{ lb})}{(32.2 \text{ ft/s})} \left(7.18 \text{ ft/s} \right)^{2}$

 $T_1 = 2.08 \text{ ft.} \text{ b}$ $V_1 = C$ $T_3 = 0 \quad V_3 = \frac{1}{2} \text{ k} \times^2 \text{ k} = 5 \text{ lb/in.} = 60 \text{ lb/ft}$

 $V_3 = \frac{1}{2}(60)(x)^2 = 30x^2$ $T_1 + Y_1 = T_3 + V_3$ $7.09 + 0 = 0 + 30x^2$ x = 0.263 ft









GIVEN:

MASS OF BALL MB=709 ho=2108 BALL DROPS FROM B h2=0.25m MA=M=2109 FOAH RUBBER SUPPORT ATC

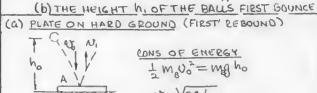
13.186

GIVEN:

MB= 7009, MA=3509 A STRIKES B WITH VELOCITY UO AT 60° AS SHOWN CORD BC ATTACHED TO B IS INEXTENSIBLE NO FRICTION

FIND:

VELOCITY OF EACH BALL AFTER IMPACT, CHECK THAT NO ENERGY IS LOST IN THE



FIND:

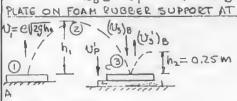
CONS OF ENERGY I movo = ma ho U= 129 ho:

(1) COEFFICIENT OF RESTITUTION BETWEEN THE

PELATIVE YELOCITIES

U= e Vzgho we=v, PLATE ON FOAH PUBBER SUPPORT ATC

BALL AND THE PLATES



CONSERVATION OF ENERGY

POINTS () AND (3) V = V3 = 0

1 mov = 1 mo(N3)B

(U3)3= e 129ho

CONSERVATION OF HOHENTUM

1+ AT 3 MB (-U3) B+ MPUP = MB (U1) B-MPUP Mp = 210 = 3 -evigho =(N3) 2-3 Up (1)

PRATIVE YELOCITIES [(-N3)8-(Dp)]C = -U' - (U')B

EVZghotó = Up+ (U3)B HULTIPLY (2) BY 3 AND ADD TO (1) 4(U3) B=VZgho (3e2e) (2)

CONSERVATION OF ENERGY AT 3, (N3) 8= 129 h2 THUS 4 /29 h2 = /29 ho (3e2-e)

36-6-1.633=0

e= 0.923

b) FROM (a), N=e Vzghp POINTS () AND (2) CONS. OF ENERGY & MBUZ = MBQ h, = ezzgho=ghi

NI= (0.923)2(1.5)=1.278M

UB IS IN THE X DIRECTION

BALL A ALONE

MOMENTUM IN & DIRECTION IS CONSERVED

MAWALE MAWA'LE, WALEO THUS (UA) = O AND (UA') = UA A-600

BALLS A AND B TOTAL MOMENTUH IN X DIRECTION CONSERVED

MAUSINGO= MA(VA) nSINGO+MBUB (0.350) 1 No=(0.350) 1 Un +0.700 NB Un= NA+ 2.309 NB

RELATIVE VELOCITIES (N-DIRECTION)

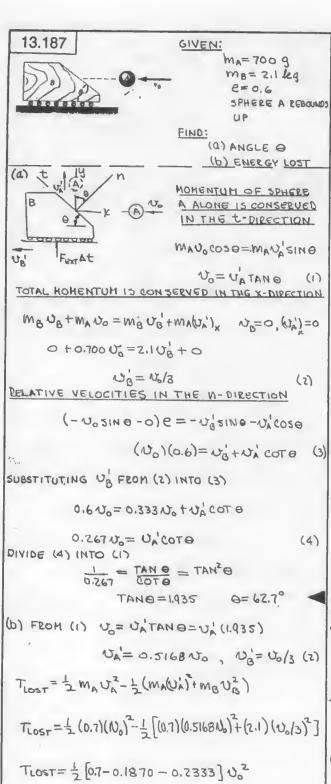
[(UA)n-(VA)n]e=(Ué)n-(VA)n (No-0](1= UR SINGO-NA

Vo= 0.866 Ng - VA (2) (S) ONA (I) DNIDOA

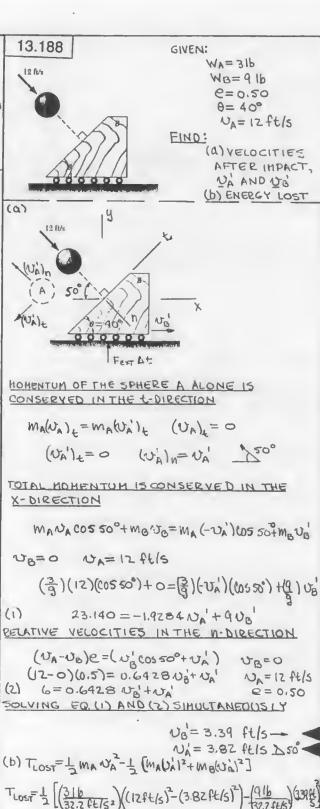
> 2 Vo= (2.309 +0.866) UB NB= 0.63000 -

FROM (1) NA'= No-(2.309)(.630 No) = -0.455 No UA'= 0 A55No &

ENERGY AT = 1 MAU2 - 1 MAVA - 1 MOVE $\Delta T = \frac{1}{2} \left[(0.350) \left(v_0^2 - (0.455 \, v_0)^2 \right) - (0.700) \left(0.630 \, v_0 \right) \right]$ DT=1 [0.350(1-0.2065)-.700 (0.3969)]V02 AT= 1 (0.278 - 0.278) 0=0 (CHECK)



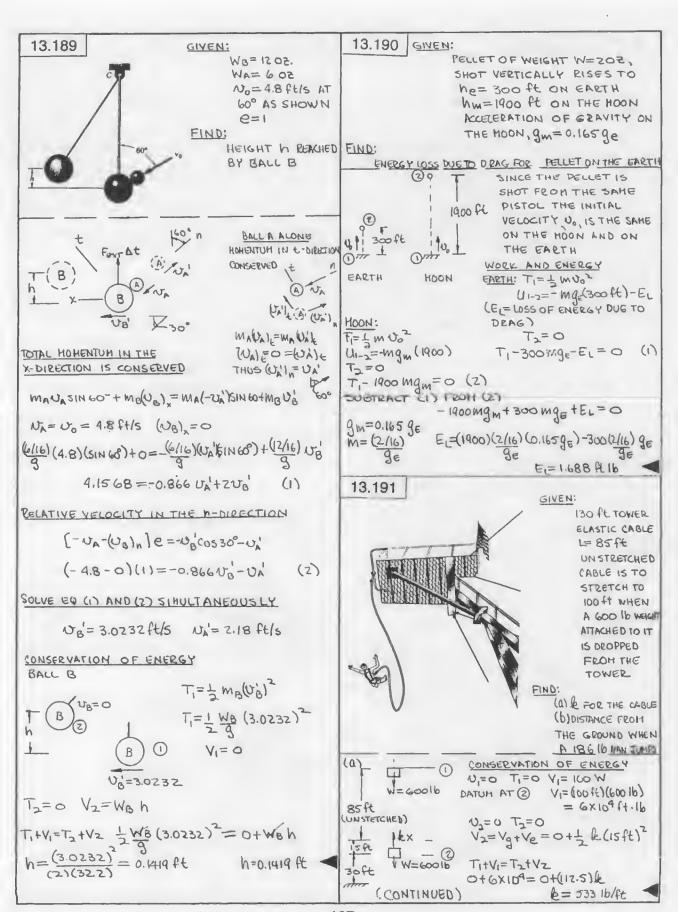
TLOST = 0.1400 No T

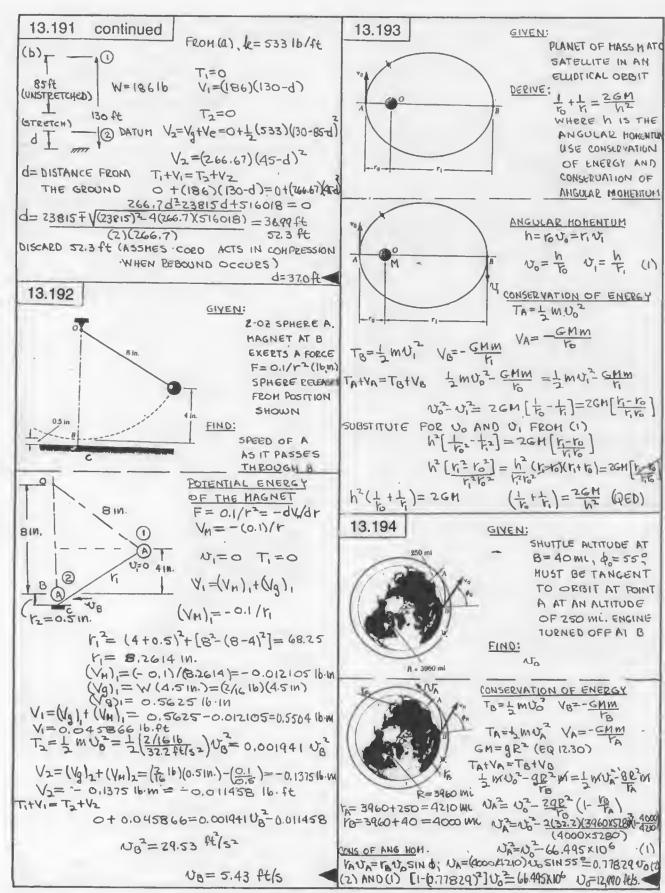


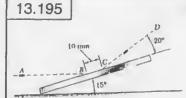
TLOST = 2 [12.064 - 3.212] = 4.42 ft.16

TLOST = 4.42 ft.16

TLOST = 0.140002







GIYEN:

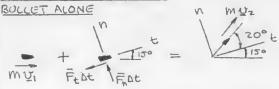
25-9 BULLET
FNITIAL YELOCITY

Ni= 600 m/s,
HORIZONTAL
RICOCHET
VELOCITY Uz
= 400 m/s
AT 20°.

BULLET LEAVES A 10-MM SCRATCH ON THE PLATE AT AN AVERAGE SPEED OF 500 M/S FIND:

THE HAGNITUDE AND DIRECTION OF THE AVERAGE INDUCTIVE FORCE EXERTED BY
THE BULLET ON THE PLATE

EMPULSE AND HOMENTUM



MUI + FLt = MUZ

<u>t εινεστιον</u> μοι εος 15° - Fe Δt = νι υς σος 20°

F_t Δt = (0.025 kg): [600 m/s cos 15° - 400 m/s cos 20°]

F_t Δt = 5.092 kg - m/s

 $\Delta t = \frac{5pc}{V_{NV}} = \frac{0.010 \text{ m}}{500 \text{ m/s}} = 20 \times 10^{-6} \text{ 5}$ $F_{t} = (5.092 \text{ kg} \cdot \text{m/s}) / (20 \times 10^{-6} \text{s}) = 254.6 \times 10^{3} \text{ Jg/m}$

FE= 254.6 RN

N DIRECTION

/*. - MUI SINIS° + FNAt = MUZEIN 20°

For N= (0.025 ig) [600 m/s SIN 15°+400 m/s SIN 28]

Fust = 7.3025 Ag.m/s St=20x10"5

Fn=(13025 / 3 m/s)/207 5 = 365.1 X10 kg m

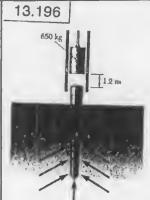
Fn= 365.1 KN

 $F = \sqrt{754.6}^2 + (365.1)^2 = 445.1 \text{ kn}$ $F = \sqrt{150}$ $\Theta = 700 \cdot \frac{F_1}{F_0} = 700 \cdot \frac{254.6}{365.1}$ $\Theta = 34.9^\circ$

X=34.9+15°=49.9°

FORCE OF THE BULLET F= 445 &N





GIYEN

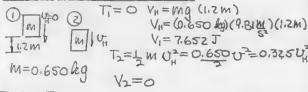
650 kg HAHNER
DEOPS 1.2 M AND
DEIVES A 140 kg
PILE 110 MM INTO
THE GEOUND
E=0

FIND:

AYERAGE RESISTANCE OF THE GROUND TO PENETRATION

VELOCITY OF THE HANNER AT IMPACT

CONSERVATION OF ENERGY



Tit V1 = T2 + V2

0+7.652=0,32502 0=23.54 m/s2 N=4.852 m/s

VELOCITY OF PILE AFTER IMPACT

SINCE THE IMPACT IS PLASTIC (C=0), THE VELOCITY OF THE PILE AND HANGE ARE THE SAME AFTER IMPACT

CONSERVATION OF HOHENTUM



THE GROUND REACTION AND THE WEIGHTS ARE NON-IMPULSIVE

THUS $m_H v_H = (m_H + m_P) v_I^{\dagger}$ $N = \frac{m_H v_H}{(m_H + m_P)} = \frac{(650)}{(650 + 140)} (4.852 \text{ m/s}) = 3.992 \text{ m/s}$

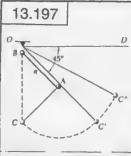
WORK AND ENERGY d=0.110 M

 $T_2 + U_{2-3} = T_3$ $T_2 = \frac{1}{2} (M_H + M_P) (U')^2$

 $T_3 = 0$ $T_2 = \frac{1}{2}(650 + 140)(3.992)^2$ $T_3 = 6.295 \times 10^3$ J

U2-3= (MH+MP) 9d-FAYd= (650+140)(9.81)(110)-FAY(110) U2-2= 852.49 - (0.110) FAY

T2+U2-3=T3 6.295×103+852.49-(0.110)FAV= 0 FAV= (7147.5)/(0.110)=64.98×103N FAV=656N



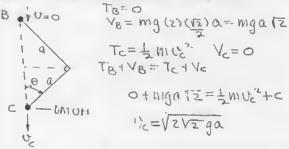
GIVEN:

SPHERE RELEASED FROM REST AT B. CORD OF LENGTH ZA BECONES TAU? AT C

FIND:

VERTICAL DISTANCE
FROM OD TO THE
HIGHEST POINT C"
REACHED BY THE SMICH

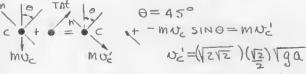
VELOCITY AT POINT C (BEFORE THE CORD IS TAUT)



VELOCITY AT C (AFTER THE CORD BECOMES TAUT)

LINEAR MOHENTUM PERPENDICULAR TO THE CORD

(5 CONSERVED



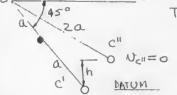
NOTE: THE MEIGHT OF THE SPHERE IS A NON-IMPULSIVE FORCE

VELOCITY AT C' (CONSERVATION OF ENERGY) $T_c = \frac{1}{2} w(N_c)^2 V_{ci} = 0$ $T_c = \frac{1}{2} w(N_c)^2 V_{ci} = 0$

E DATUM TC+VC=TC+VC1

1/2 m(vc) +0=1/2 m(vc) +0

C'TO C" (CONSERVATION OF ENERGY)



Tc1 = 1 m(vc1)2

Tc== 1 m (24 vga)

DATUM VCI = 0

Ve= 24ga Tc+Vc=Tc+Vc"

T_ = 0

Emgato= otmgh

Ven= mgh

h= 1= a

h= 0.707a

13.198

GIVEN:

MA AND MO SCIDING ON A
FRICTIONLESS SURFACE
INITIALLY, UG = 0

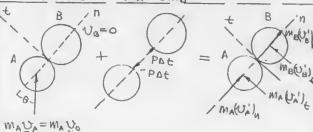
VA = VO AT ANGLE O
COEFFICIENT OF
RESTITUTION, C

SHOW:

THAT N COHPONENT OF THE VELOCITY OF A AFTER IMPACT IS,

(a) POSITIVE IF MAZEMB

(c) ZERO IF MA= ema



DISKS A AND B (FOTAL HOHENTUM CONSERVED)

mAUA+MBUB=MAUA+MBUB

MORMAL DIRECTION:

MANOCOSO + O = MA(C') , + MB(C'), ()

PELATIVE VEWSITIES

$$[V_{A}\cos \theta - (V_{B})_{n}] = [V_{B}']_{n} - (V_{A}')_{n}$$

$$V_{0}(\cos \theta) = [V_{B}']_{n} - (V_{A}')_{n}$$
(2)

MULTIPLY (2) BY MB AND SUBTRACT IT FROM (1)

$$V_0 \cos \Theta (M_A - eM_B) = (M_A + M_B)(U_A')_N$$

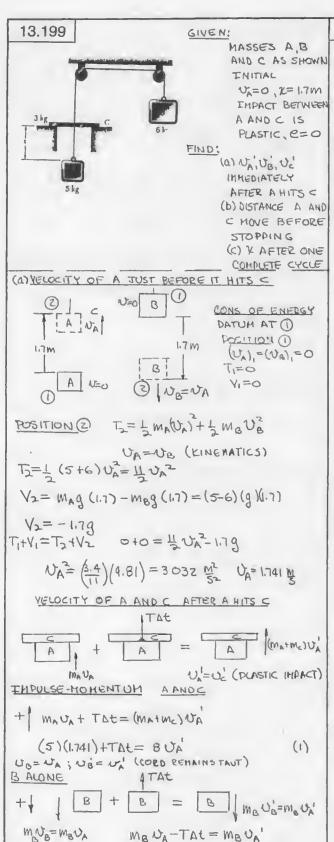
$$(V_A')_N = (V_0 \cos \Theta) (\frac{M_A - eM_B}{(M_A + M_B)})$$
(3)

FROM EQUATION (3)

(a) MAZEMB (VA) n POSITIVE

(b) MA < emb Waln NEGATIVE

(c) ma= emo (va)n = 0



13.199 continued

TNITIAL

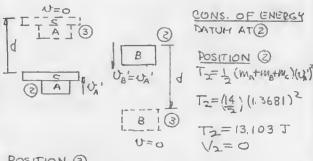
UA=0, X=1.7M

ADDING GOURTIONS (1) AND (2), 11(1.741)= 404

UA=0, X=1.7M

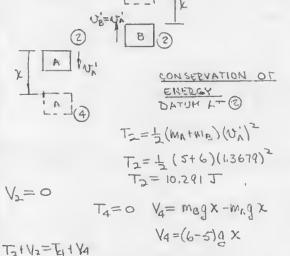
UA=0, X=1.7M

THPACT BETWEEN DISTANCE A AND C MOVE CEFORE STOPPING



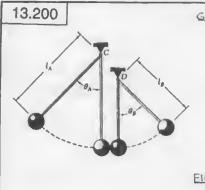
POSITION (3) $T_3 = 0$ $V_3 = (M_A + M_c)gd - M_B gd$ $V_3 = (8 - 6)gd = 2gd$ $V_3 = (8 - 6)gd = 2gd$ $V_3 = (8 - 6)gd = 2gd$

(b) AS THE SYSTEM RETURNS TO LOSITION (2), AFTER STOPPING IN POSITION (3), ENERGY IS CONSERVED AND THE VELOCITIES OF A B AN C BEFORE THE COLLAR AT C. IS REMOVED, ARE THE SAME AS THEY WERE IN (A) ABOVE WITH THE DIRECTIONS REVERSED. THUS, NA " " " " " " " " 1.3679 M/S. AFTER THE COLLAR C IS REMOVED THE VELOCITIES OF A AND B REMAIN THE SAME SINCE THERE IS NO IMPULSIVE FORCE ACTING ON EITHER. (4) " " "



10.291+0=(1)(9.81) X

(6)(1.741) - TAt = 6NA' (2)

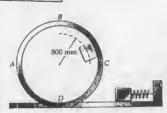


GIVEN:

SPEERE A IS EFLEASED FROM PEST AT AN AHGLE OA. SPHERE B 15 AT REST, IS HIT BY A, AND RISES TO A HONIXAH ANGLE 0 = 9

EIND:

OB IN TERMS OF RB/LA AND C.



G". EM:

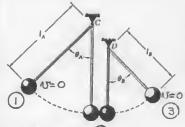
300-9 BLOCK SPRING OF CONSTANT R= 600 N/IN 15 INITIALLY COMPRESED 60 MM WHEN THE BLOCK IS RELEASE D. NO FRICTION

FIND;

13.201

FORCE EXERTED BY THE LOOP ABOD ON THE BLOCK AS IT PASSES THPCUGH. (a) POINT A

(b) POINT B, (c) POINT C



CONSERVATION OF ENERGY (1) - (2) DATUM AT (2) SPHERE A POSITION (1)

UA=0 T=0 VI= Mg.la(1-cos e)

MA=MB=M OA= OB Ti+V1=T2+ V2

POSITION (2) T2=1 MUA2 V2=0

0+ mg. (1-cose) = = m v2+0 1) = 292A (1-cos 0A) (1)

CONSERVATION OF MOHENTUM AT (2)

MUA + MUB= MUA+ MUB (A)(B) = (A)(B)aum Aum OFOU AUM NA+0= UA+ UB (2)

PELATIVE VELOCITIES AT (2)

(UA-NB) e= UB-VA NAE=UB-UA (3)

ADDING EQUATIONS (2) AND (3) AND SOLVING

UB= = (Ite)UA (4)

CONSERVATION OF ENERGY @-3

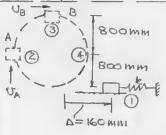
POSITION (2) T2=1 m(v3) V2=0

13=mg(B(1-cos 08) 1 m(v/s)2+0=0+mg(g(1-coses)

SUBSTITUTE UB FROM EQ.(4) INTO EQ. (5)

4 (1+e) NA = 29 (6) (1-cos 0A)

DIVIDE (1) INTO (6) AND SET 0,= 0B 1 (1+e) 2 U2 = 290B(1-cosob) 4 (1-cosob) 28/2= (1+e)2/4



VELOCITIES AT A AND B CONTERVATION OF ENERGY, DATUM AT () POSITION () J=0 T=0

V1= 4 (600 N/m) (0.160 m) V= 7.68 J

POSITION (2) $T_2 = \frac{1}{2} M U_A^2 = \frac{1}{2} (0.3) U_A^2 = 0.15 U_A^2$ Vz= mg(0 800m)=(0.3kg)(g)(0.8m)=0,249 T1+V1=T2+V2 0+ 7.68 = 0.15 UA2+0.249 U2= 7.68-(0.24)(9.81)

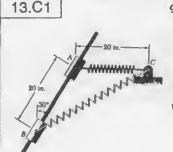
UA 35.50 m3/52 POSITION3 T3=1 m UB= = (0.3) UB=0.15 UB2 V3= mg(1.6m)=(0,3)(9)(1.6)=0.48q 0+7.68 = 0.1506 +0.489 UB= 7.68-6.48 (9.81)=19.81 m2

POSITION (4) SINCE Y4=V2 THE VELOCITY UA= NC U= 35.50 m2/52 NEWTONS SECOND LAW

A TA(D) EF= NA= Man an= NA = (35.50 M/51) (0.8 M) NA= (0.3 kg) (35.50 m3/52) (6.8 M)

(P) NA= 13.31 N-AT B INB Mae ZFn= No+ma= man an= NB2 = (111.81 m2/5) (UB)= 29(B(1-COSOB) (5) NB=(0.3kg)(19.81 M3/52)-(0.3kg)(9.81 M3) (0.8M) NB= 4.49 N

> [Mat I ZFn=Nc=man an=v2 Nc=(0.3 kg)(35.50 m/s2)/(0.8 m) Nc= 13.31 N



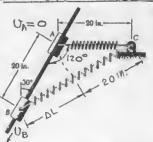
GIVEN

COLLAR WA= 12 lb
SPEING IS
UNSTRETCHED WHEN
COLLAR IS AT A.
COLLAR RELEASED
FROM REST AT A

FIND:

YELOCITY AT B
FOR &= 0.11b/in
TO 2.0 1b/in IN
0.11b/in INCLUMENTS

WRITE EQUATION FOR UB IN TERMS OF A



ANALYSIS

(20+AL)=20+20-2(20)05120

(20+AL)=800+400=1200

AL=14.64 IN=1.220ft

CONSERVATION OF ENERGY

UA=0 TA=\frac{1}{2}mVa^2=0

TB=\frac{1}{2}mVb^2=\frac{6}{3}Vb^2

VA=0 (DATUM AT A)

VB= -(121b)(20 ft)(005 30)+1(12(16/14)(1214/ft)(1.220ft)(2)

VB=(-17.32 +8.932 le)(16.ft),(INPUT le IN 16/In.)

Trat VA = TB+VB 0+0 = (32.2 AVS2) UB-17.32+88326

UB = [92.95-47.933 k.] 1/2 (ft/s) (1)

OUTLINE OF PROGRAM

INPUT & IN (1) IN 16/11 IN 0.1 16/11 INCREMBER
AND STOP WHEN &= 2.0 16/11
PRINT VALUES OF UB (IN Ft/S)
NOTE: COLLAR NEVER BLACHES B FOR

k> (92.95)/47.933)= 1,939 lb/in

PROGRAM OUTPUT

13.Cl

K (LB/IN)	VELOCITY	(FT/S)
0.10	9.39	
0.20	9.13	
0.30	8.86	
0.40	8.59	
0.50	8.31	
0.60	8.01	
0.70	7.71	
0.80	7.39	
0.90	7.06	
1.00	6.71	
1.10	6.34	
1.20	5.95	
1.30	5.54	
1.40	5.08	
1.50	4.59	
1.60	4.03	
1.70	3.39	
1.80	2.58	
1.90	1.37	

13.C2 | GINEN:

CAR WEIGHT, W= 2000 lb

FOR FIRST GOFT ALL WEIGHT IS ON

THE REAR WHEELS WHICH ARE SUPPING

FOR ECHAINING 1260 FT, 60 % OF THE

VIEIGHT IS ON THE REAR WHEELS

WITH JLI PAING IMPENDING.

US=0.60 4R=0.85

AERODYNAMIC DRAG FT=0.0098 VZ

VIITH U'IN FT/S AND FT IN 16.

FIND:

VELOCITY AND ELAPSED THE WITH AND WITHOUT DRAG.
EYERY 5 ft FOR THE FIRST 60 ft AND
EYERY 90ft FOR THE REHAINING 1260 ft.

ANALYSIS USE WORK AND ENERGY IN INCREMENTS OF AX = 0.1 ft CRETWEEN CTH AND CHITH INTERVAL

TOGET ULLI

TOGET ULLI

TOGET ULLI

(ULLO FOR LEO)

 $T_{i} + (I_{(c) \rightarrow (i+1)} = T_{i+1} + \frac{1}{2}mN_{i}^{2} - (F_{0}+F_{i})\Delta X_{i} = \frac{1}{2}mN_{c+1}^{2}$ $V_{i+1} = [N_{i}^{2} + \frac{2q}{W}(F_{0}-F_{0})\Delta X_{i}]^{\frac{1}{2}} \quad F_{0} = 0.0098V_{i,0}^{2}$

 $\Delta t_i = \frac{2\Delta \chi_i}{\langle N_i, N_{i+1} \rangle}$

FIRST 60 ft Ff = 4KW = (0.85)W FOR REMAINING 1260 FE Fr = (0.60)46W = 0.36W

0 x = 0.1 ft g = 32.2 ft/s W = 2000 lb

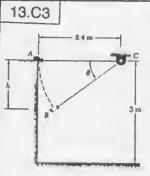
OUTLINE OF PROGRAM

IDENTIFY N, AND VO AS THE UELOCITIES IN THE CITH INTERVAL WITHOUT AND WITH DRAG, WITH UZ==0 AND F+=0.85W USE A LOOP TO SOLVE FOR UZ+1 AND TO SOLVE FOR ti. SUM AX. TO FIND X; AND SUM ALL TO FIND ti.PRINT UZ+C, X; AT 5 ft INTERVALS REPEAT FOR REHAINING 1260 ft WITH F+=0.36W.

PRINT YL, UZ+C, AT 90 FT INTERVALS

13.C2

ISTANCE	(FT)		T(S)		T(S)
		NO	DRAG	DRAG	
5.		13.90	0.719	13.89	
10.		19.66	1.017	19.64	
15.		24.07	1.246	24.05	1.247
20.		27.80	1.439	27.76	1.440
25.		31.08	1.609	31.02	1.610
30.		34.05	1.762	33.97	1.764
35.		36.78	1.903	36.67	1.905
40.		39.31	2.035	39.19	2.037
45.		41.70		41.55	2.161
50.		43.95	2.275	43.78	2.278
55.		46.10	2.386	45.90	2.390
60.		48.15	2.492	47.92	2.496
60	FT. TO	1320 FT A	T 90 FT.	INTERVALS	
150.		72.65		71.76	4.000
240.				89.00	5.119
330.			6.002	103.02	6.057
420.			6.803	115.02	6.882
510.			7.524	125.60	7.630
600.			8.184	135.09	8.320
690.		151.70	8.798	143.71	
780.		161.15	9.373	151.63	9.575
870.		170.08	9.917	158.95	10.155
960.		178.56	10.433	165.75	10.709
1050.		186.66	10.926	172.10	11.242
1140.			11.398	178.06	
1230.		201.88	11.852		
1320.		209.07	12.290	188.96	12.736



GIVEN:

5- Lg BAG

ROPE = 2.4 M LONG

INITIAL VELOCITY ZERO

EIND:

FOR VALUES OF MAXIMUM
TENSION FM FRON 40
TO 140 M IN 5-N
INCREMENTS, THE
(a) DISTANCE h
(b) DISTANCE d FROM
THE WALL TO THE
POINT WHERE THE
BAG HITS THE

h B W 3m

BAG HOVES ALONG A CIRCULAR ARC AB UNTIL THE LOPE BLEAKS (LADIUS, L) NEWTONS LAW

FLOOP.

Fm
$$mv^2 = a_n$$
 $\sum_{k=0}^{\infty} \sum_{k=0}^{\infty} ma_k$
 $\sum_{k=0}^{\infty} \sum_{k=0}^{\infty} ma_k$
 $\sum_{k=0}^{\infty} \sum_{k=0}^{\infty} ma_k$

$$|SIN\theta = h| U = |Egh| (1)$$

$$|SIN\theta = h| |Fm = 2mg | h + mg | h = 3mg | h$$

$$|\Theta = SIN^{-1} | h| (2) |h = |Fm| | h| (3)$$

FROM B TOD (PROJECTICE TRAJECTORY) $V_H = V \leq IN\Theta \Rightarrow d = (l - l\cos\Theta) + V_H + D$ $U_{U} = V \cos\Theta \neq (3-h) = V_{U} + g + g + 2 \cdot 2 \cdot 3 - h$ $U_{D} = V \otimes I + V \otimes I + 2 \cdot 3 - h$ (6) $U_{D} = V \otimes I + V \otimes I + 2 \cdot 3 - h$ (5)

OUTLINE OF PROGRAM!

WITH L= 2.4 m, m=5kq, q=q.81m/s² IN EQUATION (3),

AND FOR FM IN SN INCREMENTS FROM 40TO HO N, SOLVE FOR h.

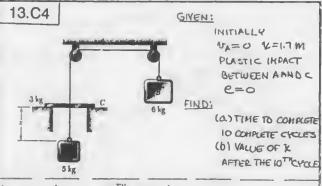
FOR EACH h, SOLVE FOR U (EQ 1), AND O (EQ 2).

SOLVE FOR UH AND UD (EQ 6) AND WITH UD AND h.

SOLVE FORTD (EQ.5) AND WITH O, h, to SOLVE FOR d IN (EQ 4). PEINT h AND d FOR EACH FM.

PROGRAM OUT PUT

FORCE (NEWTONS)	H (METERS)	d (METERS)
40	0.652	0.503
45	0.734	0.585
50	0.815	0.668
55	0.897	0.752
60	0.979	0.839
65	1.060	0.927
70	1.142	1.017
75	1.223	1.109
80	1.305	1.203
85	1.386	1.300
90		
	1.468	1.401
95	1.549	1.505
100	1.631	1.615
105	1.713	1.731
110	1.794	1.854
115	1.876	1.989
120	1.957	2.137
125	2.039	2.306
130	2.120	2.504
135	2.202	2.753
140	2.283	3.101



ANALYSIS (FOR THE LTH CYCLE)

REFER TO FIGURES IN THE SOUTION TO PROB. 13,199

FROM OTO CONSERVATION OF ENERGY (REFORE IMPACT)

T,=0, V1=0, T2= 1/2 (516) (VA)= 11/2 (VA)/2

Y2=(5-6) g(Y1)=-gxi

T,+V1=T2+V2 O+0=11 (VA)-gxi, (VA)=V1/1 gxi (1)

THE (±1-2);

ACCELERATION FROM () TO (2) 15 CONSTANT,

THUS AVERAGE VELOCITY IS (VA)=0+(VA);

AND $(\overline{U}_{A})_{c} = \frac{\nu_{c}}{(\pm_{1-2})_{c}} (\pm_{1-2})_{c} = \frac{2\nu_{c}}{(U_{A})_{c}} = \frac{2\nu_{c}}{\sqrt{\frac{2}{11}9\nu_{c}}} (\pm_{1-2})_{c} = \frac{1.498\sqrt{\nu_{c}}}{(2)}$

(AFTER IMPACT) AT (2)

THRUSE-HOHENTUH FOR A AND C

T(Va) + TAL= (8 Wa') {

INDIVISE-HOHENTUH

TOO B

IMPULSE-HOHENTUM FOR B

6(VA)(-TAt=6(VA)).

ADDING $1/(U_A)_i = 14(U_A)_i$ $(U_A)_i = 1/(U_A)_i = \sqrt{\frac{119}{98}}i$ (3) FROM (2) TO (4), (SEE (D) IN SOLUTION TO PROB. 13.199) CONSERVATION OF ENERGY DATUM AT (8) $T_2 = 1/(5+6)(U_A)^2 = 11/(9)$ 1/(9) 1/(9

Ta=0 V4= mag ki+1-mag ki+1=6-5)g xi+1=g xi+1
Ta+V2=Ta+V4 121gxi+0=0+gxi+1

196 Xi+1=121 xi (4)

TIME FROM (2) TO (3) AND FROM (3) TO (2) $T_2 = \frac{1}{2} (M_A + M_B + M_C) (U_A^1)_i^2 = 7 (\frac{119 \chi_i}{98}) = \frac{779}{98} \chi_i$ $T_3 = 0 \qquad \forall_3 = (M_A + M_C) q d_i - M_B q d_i = (8-6) q d_i = 29 d_i$

 $T_2+V_2=T_3+V_3$ $\frac{779}{98}$ $V_1+0=0+29$ di di= $\frac{77}{77}$ V_1

 $(2-3)i = \frac{2 dc}{(0-1)i}$

(t2-3) = 2 (77 ki) / 119 ki = 0.7488 \u03b4 (6)

(7)

(t3-2); = (t3-4)c = 0.7488 VFi

13.C4 continued

TOTAL TIME TO COMPLETE THE LTH CYCLE EQS. (2)+ (6)+(7)+(5) tu=(t12) + (t2-3)(+(t3-2)i+(t2-4))

ti=(1.498 +0.7488+0.7488 +1.1766) \\Zi

ti= 4.172 \xi

OUTLINE OF PROGRAM SET X = 1.7 M ((=1)

- (a) CALCULATE YELL FROM EQUATION (4) FOR L= 1 TO L= 10. FOR EACH VALUE OF X USE EQUATION (B) TO DETERMINE & FOR THE L' CYCLE, SUM t'S TO OBTAIN THE TOTAL TIME THROUGH THE 10TH CYCLE.
- (b) FOR C=10 OBTAIN & FOR THE TENTH

PRINT TOTALTIME AND & FOR THE 10TH CYCLE. PROGRAM OUTPUT

TOTAL TIME-23.1 SECONDS

X FOR THE TENTH CYCLE-0.01367 METERS

13.C5 GIVEN:



MB= 700 g, MA= 350 9 U0=6 m/s UB=0 0 = 20° TO 150° IN 10° INCREMENTS

FIND:

UA AND UB AFTER IMPACT AND ENERGY LOST FOR,

(a) e=1

(b) e=0.75

(c) e=0

ANALYSIS:

DEVELOP FORHULAS FOR UN AND UB IN TERMS OF & AND C

MAUA

MOHENTUH OF A ALONE IS CONSERVED IN THE t DIRECTION

MAUA) = MUAlt (UA)=0 THUS WA'L=0 AND W' IS ALONG THE N AXIS

KINGHATICS

(NB) = UB

13.C5 continued

(8)

CONSERVATION OF MOHENTUM IN THE & DIRECTION FOR A AND B TOGETHER

MAUASINGO = MAUASINGO+ MOUB MA= 0.350 kg MB=0.700 kg UA= Ub= 6 M/S 6 SINO = WASINDOT ZUB

PELATIVE VELOCITIES IN THE M DIRECTION (NA-0) @= NB SING - NA NA= 00 = 6 M/S (2) 6e = NBSINDO-NA

HULTIPLY (2) BY SIND, AND ADD TO (1) TO GET UD UB = 651NO0 (1+e) (2+51N2 Q)

SUBSTITUTE (3) IN (2) FOR UA' (4) UA'= 651N20-120 (2+51N200)

FOR 00 790°, TOL= O, AND BALL A AT A VELOCITY OF 6 M/S HITS BALL B WHICH IS AT O VEWCITY AND IS NOT CONSTRAINED BY THE COOD. THUS IF ONLY HAGNITUDES ARE CONSIDERED WA ANDU'B HAVE VALUES FOR 110° < 9 > 90° WHICH ARE THE AS FOR 0 = 90°

ENERGY LOST DE= 1 MUR - 1 (MAUL + MBUE) DE= 1 (0.350)[V2-V1]-1(0.700) U6 (5)

OUTLINE OF PROGRAM

INPUT O INTO EQUATIONS (3) AND (4) FROM 20° TO Q0° IN INCREHENTS OF 5° FOR E=1. C=0.75 AND C=0 TO OBTAIN U' AND U'S. SUBSTITUTE NA AND US IN (5) TO OBTAIN DE. PRINT C, OS, UA, UB, DE PROGRAM OUTPUT

13.C5

e	THETA (DEG)	VEL A (M/S)	VEL B (M/	S) VE LOS
1.00	20.	-5.337	1.939	0.0
1.00	30.	-4.667	2.667	0.0
1.00	40.	-3.945	3.196	0.0
1.00	50.	-3.278	3.554	0.0
1.00	60.	-2.727	3.779	0.0
1.00	70.	-2.325	3.911	0.0
1.00	80.	-2.081	3.979	0.0
1.00	90.	-2,000	4.000	0.0
0.75	20.	-3.920	1.696	41.3
0.75	30.	-3.333	2.333	38.9
0.75	40.	-2.702	2.797	36.3
0.75	50.	-2.118	3.109	33.8
0.75	60.	-1.636	3.307	31.8
0.75	70.	-1.284	3.422	30.4
0.75	80.	-1.071	3.482	29.5
0.75	90.	-1.000	3.500	29.2
0.00	20.	0.332	0.969	94.5
0.00	30.	0.667	1.333	88.9
0.00	40.	1.027	1.598	B2.9
0.00	50.	1.361	1.777	77.3
0.00	60.	1.636	1.890	72.7
0.00	70.	1.838	1.956	69.4
0.00	80.	1.959	1.990	67.3
0.00	90.	2.000	2.000	66.7
		OP 90 TO 150	DEGREES AR	E THE SAME

FIND:



SOUTTUA NA TA OP ONA OU OF 40 MI FOR ENERGY EXPENDITURE OF 5 TO 100 % DF THAT USED IN PROB. 13.109 IN 5% INCREHENTS

GIVEN:

INITIAL CIPCULAR OPBIT OF 225 ML ABOVE THE SURFACE OF THE LARTH INCREHENTAL VELOCITY DUA TOWARD THE CENTER OF THE GARTH

ANALYSIS



CONSERVATION OF ENERGY TA= 1 W(VA) CIRC+(AY) VA = - GMM

> AT POINT B TB=1 mUB2

R = 3960 ml

NB= - EMM ta=3960+225=4185 ml VB=-rB=3960+40=4000 mi TA+VA=TB+VB (VA)218c= TA

= m [(va)ciectava] - GHM = 1 mv3 GHM UB= (UA) CIEC HAU) + ZEM[to-ta] (1)

ENERGY EXPENDITURE IN PROB. 13.109 LET NA = UELOCITY AT A IN PEOB. 13.109 TO BRING THE VEHICLE TO B AT \$=60° FROM 13.109, UA= 11.32×103 ft/s ENERGY EXPENDITURE, E= 12 m [(1) time (VA)]

ENERGY EXPENDITURE IN THIS PROBLEM KE=1 M (AUA)2, WHERE K IS THE % ENERGY

USED IN PROB 13.109. SOLVING FOR (AUX) AND REPLACING E BY EQUATION (2)

$$\left(\Delta U_{A}\right)^{2} = \frac{K}{100} \left[\left(V_{A} \right)_{\text{CIRC}}^{2} - \left(V_{A} \right)^{2} \right]$$
 (3)

EQUATION FOR Up (SUBSTITUTE (3) INTO (1)) UB= {(VA) = (VA) = (VA) = (VA) = 2 CM [+ -1] } = 2 CM

 $\frac{\text{CONSTANTS:}}{\left(V_{A}\right)_{CIRC}^{2}} = \frac{9R^{2}}{V_{A}} = \frac{(32.7)[(3960)(5280)]^{2}}{(4185)(5780)}$

 $(N_A)^2_{\text{CIRC}} = 637.07 \times 10^6 \text{ ft}^2/\text{s}^2$ FROM 13.109, $(N_A)^2 = (11.32 \times 10^3)^2 = 128.14 \times 10^6 \text{ ft}^2/\text{s}^2$ 26H[+ +]=29R2[+ +]=6)(32.7 (3960)(5280))(+004185) (CONTINUED)

13.C6 continued

2GM (+ + 1= 58.93×106 ft3/52

EQUATION FOR DO

CONSERVATION OF ANGULAR HOHENTUM

TAUA LET TO UB SIN 4B OB = SIN' [ra(VA Iciec / ra (VB)]

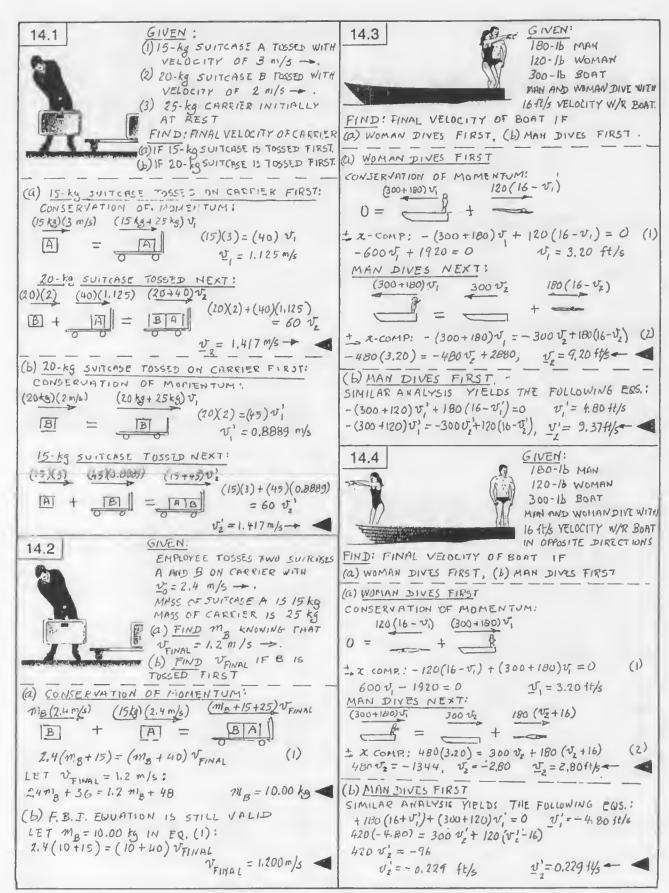
(5)

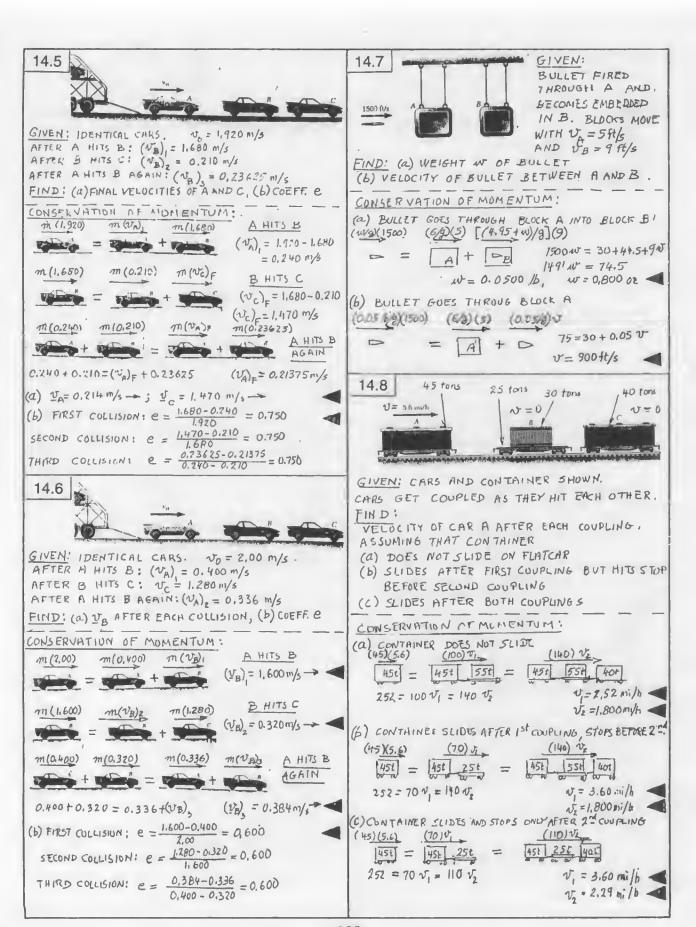
OUTLINE OF PROGRAM INPUT CONSTANTS INTO EQUATION (4) AND SOLVE FOR UB FOR VALUES OF K OF 5% TO 100 % AT INTERVALS OF 5% FOR EACH VALUE OF UB AND USING THE GIVEN CONSTANT VALUES OF (VALUEC, YA AND YB, USE EQUATION (5) TO SOLVE POR DB. PRINT K, UB AND DB.

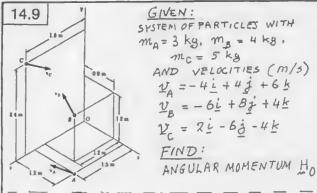
PROGRAM OUTPUT

13,C6

K (%)	VB (FT/S)	PHI (DEGREES)
5. 10. 15. 20.	26860. 27329. 27791. 28245.	79.5 75.1 71.8 69.2
25. 30. 35.	28692. 29132. 29566.	67.0 65.0
40.	29993. 30414.	63.3 61.7 60.3
50. 55. 60.	30830. 31240. 31644.	58.9 57.7 56.6
65. 70. 75.	32044. 32438. 32828.	55.5 54.5
80. 85. 90.	33214. 33595.	53.6 52.7 51.8
95. 100.	33971. 34344. 34712.	51.0 50.3 49.5







= 1 × m V + 2 × m V + 2 × mc x USING DETERMINANT FORM FOR VECTOR PRODUCTS AND FACTURING MASSES! 15 2 K | 16 2 E H = (3kg) 1.2m 0 1.571 +4 0.9 1.2 1.7 +5 0 2.4 1.8 K 1-4 m/s 4 m/s 6 01/s -6 8 4 2 -6 -4 =-18 i-39.6 i + 14.4 k-19.2 i-43.2 j+57.6 k+6 i+18 j-24k $H = -(31.2 \text{ kg} \cdot \text{m}^2/\text{s}) = -(64.8 \text{ kg} \cdot \text{m}^2/\text{s}) + (48.0 \text{ kg} \cdot \text{m}^2/\text{s}) k$

14.10 GIJEN: SYSTEM OF PARTICLES OF PRUB, 14,9

FIND:

(a) POSITION VECTOR & OF MASS CENTER G.

(b) LINEAR MOMENTUM OF SYSTEM.

(c) ANGULAR MOMELITUM HG OF SYSTEM.

ALSO: VERIFY THAT ANSWERS TO PROES. 14,9 AND 14.10 SATISFY EUUATION

(a) EQ.(14.12):

m == Zm; E;

(3+4+5) = 3 (12i+15k)+4(0.9i+1.2j+1.2k)+ 5(2.41+1.8E)

12 = 7.2 6 + 16.8 j + 18.3 k

E = (0,600m) i+(1,400m) j+(1,525m) k

(b) $L = \sum m_i \underline{v}_i = 3(-4\underline{i} + 4\underline{j} + 6\underline{k}) + 4(-6\underline{i} + 8\underline{j} + 4\underline{k}) +$ 5 (2i-6j-4K)

L=(-26.0 kg.m/s)i+(14.00 kg.m/s)j+(14.00 kg.m/s)k

(c) HG = 2NG × MA VA + EBBXMB VB + 2C/6 × MC -C WHERE ZNG = 1 - 2 = 1.21 +1.5k - (0.61 +1.4 1 + 1.525 k) = 0.6 i - 1.4 j - 0.025 k

 $\frac{2}{8}B/6 = \frac{2}{6} - \frac{2}{6} = 0.3i - 0.2j - 0.325k$ 20/6 = 2 - 1 = -0.6i + j + 0.275 k,

a K 1 H=(3kg)(06m-1.4m-Q025m+40.3-02-0.25+5-0.6 1 0.275 -4N/3 4m/3 6 N/S 1-6 0 4 2 -6 -4

=-24.9i-10.5j-3,6k+7.7i+3j+4.8k-11.75i-925j+8t

H = - (29.45 kg. m/k) i - (16.75 kg.m/k) 1 + (3.20 kg.m/k) & (CONTINUED)

continued 14.10

WE COMPUTE EXMT:

2×M= = TxL = (0.6i+1.4j+1.525k)×(-26i+14j+14k) = a6 1.4 1.525 = -1.75 i -48.05 j +44.8 t 14 14

THUS: Exm + Hg = -1,75: - 48.05j+44.8k -29.45: -16,75j+300k =-31,2i-64.8j+48.0k

VIHICH IS THE EXPRESSION OBTAINED FOR HOIN PRUB. 14.9.

14.11

GIVEN; SYSTEM OF PARTICLES WITH m=3 kg, m= 4kg, mc=5kg AND VELOCITIES (m/s) N=-41+4j+6k ひゃ - ひょ + ひょう十4上 V = 21-63-4K FIND:

(a) V, FOR VIHICH HO IS PARALLEL TO & AXIS (b) CORRESPONDING HO.

H= +xmv + + Bxmv + + xmcvc =(3kg) 1.2m 0 1.5m + 4 0.9 1.2 1.7 + 5 0 2.4 1.8 2 1.2 1.7 + 5 0 2.4 1.8

=-18 = -39.6 = +14.4 + + (19.2-4.8) = + (4.8 15 -14.4) = + (3.6 Vy - 4.8 V,) + + 6 i + 18 j - 24 K

H = (7.2-4.8 Vy) i + (-36 + 4.8 Vx) j + (-9.6 + 3.6 Jy - 4.8 Vx) k

(a) FOR HO TO BE/ZAXIS: Hy = -36 +4.8 V = 0 H=7.2-4.8 Vy =0

U = 7.50 m/s, U = 1.500 m/s

(b) Ho = Hak = (-9.6+3.6 × 1,500 - 4,8 × 7,50) K H = - (40.2 kg·ni/s) k

14.12 GIVEN: SAME SYSTEM OF PARTICIES WITH SAME VELOCITY DATA AS IN PROE. 14.11

FIND: (a) UZ AND UZ FOR WHICH HO IS PARALLEL TO YANS. (b) CORRESPONDING HO.

SEE SOLUTION OF PROB. 14.11 FOR DERIVATION OF EQ. (1):

H = (7.2-4.8 1/y) = +(-36+4.8 V2) + (-9.6+3.6 V, -4.8 V2) +

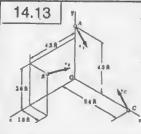
(a) FOR HO TO BE // " AXIS!

H = -9.6 + 3.6 Vy - 4.8 V = 0 H, = 7.2 - 4.8 Vy = 0 -7.6+3.6(1,500)-4.8 v, = 0 1 = 1,500 m/s V,=-0.875 m/s

v = -0.875 m/s, v = 1.500 m/s

(b) H = Hy] = [-36+4.8×(-0.875)] j

H = - (40.2 kg. mi/s) j A. 16



GIVEN: 5YSTEM OF PARTICLES WITH $W_A = 9.66 \text{ lb}$, $W_B = 6.44 \text{ lb}$, $W_C = 12.88 \text{ lb}$ AND VELOCITIES (ft/s) $U_A = 4i + 2i + 2k$ $U_B = 4i + 3i$ $U_C = -2i + 4i + 2k$

FIND: ANG. MOMENTUM Ho.

 $\frac{H}{b} = \frac{\pi}{a} \times \frac{m}{a} \frac{V}{4} + \frac{\pi}{b} \times \frac{m}{B} \frac{V}{B} + \frac{\pi}{c} \times \frac{m}{c} \frac{V}{c}$ USING DETERMINANT FORM FOR VECTOR PRODUCTS AND FACTORING MASSES; $H = \frac{9.66}{32.2} \begin{vmatrix} \dot{i} & \dot{j} & \dot{k} \\ 0 & 4.5 & 0 \\ 4 & 2 & 2 \end{vmatrix} + \frac{6.44}{32.2} \begin{vmatrix} \dot{i} & \dot{k} \\ 1.8 & 3.6 & 4.5 \\ 4 & 3 & 0 \end{vmatrix} + \frac{12.85}{32.2} \begin{vmatrix} \dot{i} & \dot{j} \\ 5.4 & 0 & 0 \\ 32.2 \end{vmatrix} = 0.3(9\dot{i} - 18\dot{k}) + 0.2(-13.5\dot{i} + 18\dot{j} - 9\dot{k}) + 0.4(-10.8\dot{j} + 21.6\dot{k})$ $H = -(0.720 \, ft \cdot |b \cdot s)\dot{j} + (1.440 \, ft \cdot |b \cdot s)\dot{k}$

14.14 GIVEN: SYSTEM OF PARTICLES OF PROB. 14.13.

FIND: (a) POBITION VECTOR & OF MASS CENTER G.
(b) LINEAR MOMENTUM OF SYSTEM.

(C) ANGULAR MOMENTUM H OF SYSTEM.
ALSO: VERIFY THAT HASWERS TO PROBS. 14,13 AND
14.14 SATISFY EQUATION

HO= EXMV+ HG

(a) EQ.(14.12): $m = \sum m_i \underline{v}_i$ WHERE $m_A = 0.3$, $m_B = 0.2$, $m_C = 0.4$, m = 0.9 $0.9\underline{v}_C = 0.3(4.5\underline{j}) + 0.2(1.8\underline{v}_C + 3.6\underline{j} + 4.5\underline{k}) + 0.4(5.4\underline{l})$ $\underline{v}_C = (2.80 + 1)\underline{v}_C + (2.30 + 1)\underline{j}_C + (1.00 + 1)\underline{k}$

(b) $L = \sum m_i \underline{v}_i = 0.3 (4\underline{i} + 2\underline{j} + 2\underline{k}) + 0.2 (4\underline{i} + 3\underline{j}) + 0.4 (-2\underline{i} + 4\underline{j} + 2\underline{k})$ $L = (1.200 \text{ lb·s})\underline{i} + (2.80 \text{ lb·s})\underline{j} + (1.400 \text{ lb·s})\underline{k}$

(c) $H_G = 2_{A/G} \times m_A v_A + 2_{B/G} \times m_B v_B + 2_{C/6} \times m_C v_C$ WHERE $\frac{1}{2_{A/G}} = \frac{1}{2_A} = \frac{1}{2$

Hg=-(0,420 ft.lb.s) i+(2,00 ft.lb.s) j-(3,64 ft.lb.s) k

 $\frac{COMPUTE}{2 \times m \overline{V}} : \frac{2}{2} \times m \overline{V} : \frac{1}{2} \times m \overline{V} = \frac{2}{2} \times L = (2.8 \cdot 1 + 2.3 \cdot 1 + 1.4 \cdot 1.$

= -0,720; +1,440 k WHICH IS THE EXPRESSION ORTAINED TO A

WHICH IS THE EXPRESSION OBTAINED FOR HO IN PROB. 14.13.

14.15 | GIVEN :

AS IT PASSES THROUGH O AT t=0, IT EYPLOTES INTO A (450 lb), B (300 lb), C (150 lb)

AT t = 45, POSITIONS OF A AND B ARE A(3840ft, -960ft, -1920ft) B(6480ft, 1200ft, 2640ft)

FIND: POSITION OF C AT THAT TIME

MOTION OF MASS CENTER:

SINCE THERE IS NO EXTERNAL FURCE,

= Vot = (1200 ft/s) i (45) = (4000 ft) i

EQUATION (14.12)

m = = \(\times_{\frac{1}{2}};\frac{1}{2};\frac{1}{2}(90\g)(4800i)=(450\g)(3840i-960j-1920k)+
(300\g)(6480i+1200j+2640k)+
(150\g)\frac{1}{2}c

150 t = (900×4000 - 450×3840 - 300×6480) i +

(450×960 - 300×1200) j + (450×1920 - 300×2640) k

= 648,000 i + 72,000 j + 72,000 k

= (4320 ft) i + (480 ft) j + (480 ft) k

14.16 GIVEN:

30-16 PASSES THROUGH O WITH

VELUCITY U = (120 ft/s) L WHEN IT EXPLODES

INTO FRAGMENTS A (12 16) AND B (18 16).

AT t = 3 5, POSITION OF A 15 A (300 ft, 24 ft, -48 ft).

FIND: POSITION OF B AT THAT TIME

ASSUME: ay = -g = - 32.2 ft/s2

MOTION OF MASS CENTER:

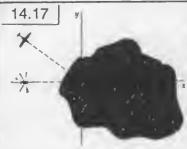
IT MOVES AS IF PROJECTILE HAD NOT EXPLODED.

 $\overline{Z} = \sqrt{t} i - \frac{1}{2}g t^{2}j$ $= (120 \text{ ft/s})(3s)i - \frac{1}{2}(32.2 \text{ ft/s}^{2})(3s)^{2}j$ = (360 ft)i - (144.9 ft)j

 $\frac{E(0ATION (14.12))}{m^{\frac{7}{2}} = \sum m_{i}^{\frac{7}{2}} \frac{1}{2}};$ $m^{\frac{7}{2}} = m_{i}^{\frac{7}{2}} \frac{1}{2} + m_{i}^{\frac{7}{2}} \frac{1}{2}$ $\frac{30}{3} (360 i - 144.9 i) = \frac{12}{3} (300 i + 24 j - 48 k) + \frac{18}{9} \frac{1}{2} \frac{1}{8}$

=7200i -4635j + 576k

2 = (400 11) i - (2581c) j + (32.011) k



GIVEN:
AIRPLANE: MA= 1500kg
HELICOPTER: MH= 3000kg
CULLIDE AT 1200 M
ABOVE U.
4 MIN BEFORE:
1+ELICOPTER WAS

HELICOPTER WAS B.4 km WEST OF O; PLANE WAS 16 km WEST AND 12 km NORTH OF O.

AFTER COLLICION, HELICOPTER BREAKS INTO

H, (1000 kg) AND H2 (2000 kg)

FIND: POINT A WHERE WRECKAGE OF PLANE
WILL BE FOUND KNOWING THAT FRAGMENTS OF HELICOPTER WERE AT H, (500 M, - 100 M) AND H2 (600 M, - 500 M).

MOTICH OF MASS (ENTER G.

AT COLLISION: $U_{H} = (6400m)! = (35,00 m/s)!$ $U_{A} = \frac{(16000m)! - (12000m)!}{4(60s)} = (66.67m/s)! - (50m/s)!$ VELOCITY OF HASS CENTER: $(m_{H} + m_{A})! = m_{H}! U_{H} + m_{A}! U_{A}$ 4500 $u_{A} = 3000 (35,000!) + 1500 (66.67! - 50!)$ $u_{A} = (15.556 m/s)! - (16.667m/s)!$

VERTICAL MOTION OF G: $k = \frac{1}{2}gt^2$ $t = \sqrt{\frac{2h}{g}} = \sqrt{\frac{2(1200m)}{9.81m/s^2}} = 15.6415$ POSITION OF G AT TIME OF GROUND IMPACT: $\frac{7}{2} = \frac{1}{2}t = (45.556\frac{1}{2} - 16.667\frac{1}{2})(15.641)$ $\frac{7}{2} = (717.55m)\frac{1}{2} - (260.69m)\frac{1}{2}$ (1)

 $\frac{F_{ROM} E_{R,(14,12)}}{(m_H + m_A) \bar{t}} = m_{H_1} t_{H_1} + m_{H_2} t_{H_1} + m_A t_A$ (2)

4500 (712.55 i - 260.69 j)=
1000 (500 i - 100 j) + 2000 (600 i - 500 j) + 1500 2 1.52 = (4.5 × 712.75 - 500 - 2 × 600) i + (-4.5 × 260.69 + 100 + 2 × 500) j

2-(1004m)i-(40.7m)j

14.18 GIVEN: SAME AS FOR PROB. 14.17.

FIND: POINT WHERE FRAGMENT HZ WILL BE FOUND, KNOWING THAT WRECKAGE OF PLANE WAS FOUND AT A (1200 m, 80 m) AND FRAGMENT H, AT H, (400 m, -200 m).

SEE SOLUTION OF PROB. 14.17 FOR DERIVATION OF & = (712.55 m) i - (260,69 m) j (1)

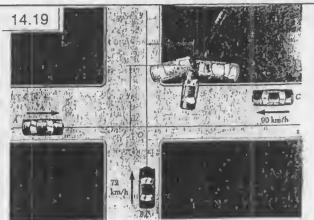
(mn+mn) = mn, + mn = + mn = 2 + mn = (2)

SUBSTITUTING DATA:

4500(712.55 i - 260.69 i) = 1000(400i - 200j)+2000 2H+1500(1200i+80j)

22 H2 = (4.5 × 712,55 - 400 - 1.5 × 1200) i + (-4.5 × 260.69 + 200 - 1.5 × 80) j

4-1-(703 m) i - (547 m) j



GIVEN: CARS A (1500kg), B (1300kg), AND C (1200kg).

WERE TRAVELING AS SHOWN WHEN CAR A HITS CARB.

AT THAT INSTANT CAR C IS AT $x_c = 10 \text{ m}$, $y_c = 3 \text{ m}$.

CAR C HITS A AND B, AND ALL CARS HIT $P(z_p, y_p)$.

FIND: (a) TIME & FROM FIRST CULLISION TO STOPAT P.

(b) SPEED V_A OF CARA

KNOWING THAT 2p = 18 m, yp = 13.9 m

MOTION OF MASS CENTER

FINAL POSITION OF MASS CENTER OF SYSTEM IS THE SAME AS IF THE CARS HAD NOT CULLIDED AND HAD KEPT MOVING WITH THEIR CRIGINAL VELOCITIES,

 $(m_A + m_B + m_C) \stackrel{?}{=} p = m_A (v_A t) \stackrel{!}{=} + m_B (v_B t) \stackrel{!}{=} + m_C (z_C \stackrel{!}{=} + y_C \stackrel{!}{=} - v_C t \stackrel{!}{=})$

WHERE $U_B = 72 \text{ km/h} = 20 \text{ m/s}$, $U_C = 90 \text{ km/h} = 25 \text{ m/s}$ $4000 U_p = 1500 U_A \text{ti} + 1300(20 \text{t})j + 1200(10i+3j-25 \text{ti})$

 $z_p = (0.375 \, v_A - 7.5) t \, i + 3 \, i + 6.5 t \, j + 0.9 \, j$ THUS: $x_p = (0.375 \, v_A - 7.5) t + 3 \, , y_p = 6.5 t + 0.9$ (1) (a) MAKINE $y_p = 13.9 \, m : 13.9 = 6.5 t + 0.9$

(b) MAKING Zp = 18 m AND t = 25: 18 = (0.375 \(\text{T}_A - 7.5 \) 2 + 3 \(\text{T}_A = 40 \text{ m/s} = 144 \text{ km/h} \)

14.20 GIVEN: SAME AS FOR PROB. 14.19.

FIND: COORDINATES OF POLE P, KNOWING THAT

VA = 129.6 km/h AND THAT TIME FROM FIRST

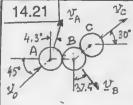
 $V_A = 129.6 \text{ km/h}$ AND THAT TIME FROM FIRST COLLISION TO STOP HT P 13 t = 2.45.

SEE SOLUTION OF PROB. 14.19 FOR DERIVATION OF $\mathcal{R}_D = (0.375V_A - 7.5)t + 3$, $\mathcal{Y}_D = 6.5t + 0.9$ (1)

MAKING $V_A = 129.6 \text{ km/h} = 36 \text{ m/s}$ AND t = 2.4 s IN EQS. (1):

 $\chi_p = (0.375 \times 36 - 7.5)(2.4) + 3 = 17.40$ $y_p = 6.5(2.4) + 0.9 = 16.50$ m

2p=17.40m, yp=16.50m



GIVEN: 3 BALLS OF SAME MASS,
BALL A STRIFES B AND C
WHICH ARE AT REST.
BEFORE IMPACT, Vo= 12 ft/s
AFTER IMPACT, Vc = 6.29 ft/s
EIND:

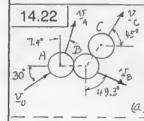
10 13.4 UB EIND:
(a) VA, (b) VB AFTER IMPACT

CONSERVATION OF LINEAR MOMENTUM

IN X DIRECTION: $m(12ft/3)\cos 45 = mv_{A}\sin 4.3 + mv_{B}\sin 37.4 + m(6.29)\cos 30^{\circ}$ $0.07498 v_{A} + 0.60738 v_{B} = 3.0380$ (1)

IN & DIRECTION: $m(12ft/3)\sin 45 = mv_{A}\cos 4.3 - mv_{B}\cos 37.4 + m(6.29) + 1n30^{\circ}$ $0.99719 v_{A} - 0.79441v_{B} = 5.3403$ (2)

(a) MULTIPLY (1) BY 0.79441, (2) BY 0.60738, AND APD: $0.6652 + v_{A} = 5.6570$ $v_{A} = 8.50 ft/s$ (b) MULTIPLY (1) BY 0.99719, (2) BY -0.07498, AND ADD:



0.66514 Vp = 2.6290

EVEN: 3 BALLS OF SAME MASS

BALL A STRIKES B AND C

WHICH ARE ATREST.

BEFORE IMPACT, Vo = 12 ft/s

AFTER IMPACT, Vo = 6.29 ft/s

FIND:

(a) VA, (b) VB AFTER IMPACT.

VR = 3.95ft/s

CONSERVATION OF LIMERS MOMENTUM
IN X DIRECTION:

m (12 ft/s) cos 30 = m VA sin 7.4°+ m VB sin 49,3°+ m (6,29) cos 45° 0.12880 VA + 0.75813 VB = 5.9446 (1)

IN y DIRECTION: $m(12 ft/s) sin 30 = m U_A cos 7.4^{\circ} - m V_B cos 49.3^{\circ} + m (6.29) sin 45^{\circ}$ $0.99167 V_A - 0.65210 V_B = 1.5523$ (2) (2) MULTIPLY (1) BY 0.65210, (2) BY 0.7583, AND HDD: $0.83581 V_A = 5.0533$ $V_A = 6.05ft/s$

(b) MULTIPLY (1) BY 0. 99167, (2) BY - 0, 12880, HAD ADD: $0.63581^{-1}E = 5.6951$. $V_B = 6.81 \text{ ft/s}$

14.23 GIVEN: 3-kg BIRD FLYING 15m ABOVE GROUND WITH $Y_B = (10 \text{ m/s}) i$ IS HIT BY 50-g ARROW WITH $Y_A = (60 \text{ m/s}) i + (80 \text{ m/s}) k$.

FIND: DISTANCE FROM 0 UNDER FOINT OF IMPACT TO

CONSERVATION OF MODIENTUMI

(3000g)(10ms): + (50g)(60j+80K) = (3050g) V

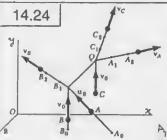
VELOCITY OF BIRD MYD HELLY AFTER IMPACT:

V = (9.8361 m/s): + (0.98361 m/s); + (1.3115 m/s) L

P WHERE BIRD HITS THE GROUND.

VERTICAL MOTION: $y = y_0 + \sqrt{y_0} t - \frac{1}{2}gt^2$ Make y = 0: $0 = 15m + (0.98361442) t - \frac{1}{2}(9.814442) t^2$ (CONTINUED) 14.23 continued

t' - 0.20053t - 3.0581 = 0 $t = \frac{0.20053 + \sqrt{(0.20053)^2 + 4(3.0581)}}{2} = 1.8519 \text{ s}$ $\frac{1}{2} + \frac{1}{2} + \frac$



BEFORE COLLISIONS ALPHA

PARTICLE A MOVED WITH

L=-(480 MS) i+600 j-640 E,

NUCLEI B AND C MOVED

WITH L=(480 MS) j. ...

AFTER COLLISIONS, THEY

MOVED ALONG PATHS WHERE

A(240,220,160), A(320,300,200)

B, (107,200,170), Bz(74,270,160)

(DIMENSIONS IN mm)

C1(200,212,130), Cz(200,260,115)

FIND: SPEED OF EACH PARTICLE AFTER COLLISIONS.

MASS OF OXYGEN NUCLEUS = M, HAS OF & PARTICE = 1/m
BEFORE COLLISIONS:

OX PARTICE: 11 = -480 i + 600 f - 640 f
NUCLEI B AND C: 1/6 = 480 f

AFTER COLLISIONS:

 $\frac{V_{A} = V_{B}}{A_{A}A_{A}} = \frac{80i+80j+40K}{120} V_{A} = (0.6667j+0.6667j+0.3331k) V_{A}$ $\frac{V_{B}}{B_{B}} = V_{B} \frac{8.3}{8.8} = \frac{-33i+70j-10k}{78.63} V_{B} = (-0.4229i+0.8971j-0.12816k) V_{B}$ $\frac{V_{C}}{C_{C}C_{C}} = \frac{48j-15k}{50.29} V_{C} = (0.9545j-0.2983k)$

CONSERVATION OF MOISENTUM:

-120 L + 150 j - 160 k + 960 j = (0.16 67 i + 0.1667 j + 0.08333 K) VA (-0.4229 i + 0.8971 j - 0.120 16 k) VB + (0.4545 j - 0.2983 k) VB EQUATING THE CUEFFICIENTS OF THE UNIT VECTORS:

 $\begin{array}{lll}
0.1667 V_A - 0.4229 V_B & = -120 & (1) \\
0.1667 V_A + 0.8971 V_B + 0.9545 V_C = 1110 & (2)
\end{array}$

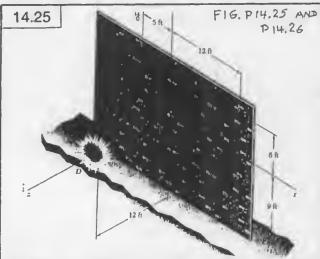
 $0.08333 \, V_A - 0.12816 \, V_B - 0.2983 \, V_C = -160$ $MUITIPLY (2) RY 0.9983 \, (3) RY 0.9585 \, AND 3 PD 1$

MULTIPLY (2) EY 0.2983, (3) BY 0.9545 AND 7 DD 1 0.12926 VA + 0.14528 VB = 178.39 (4)

HULTIRY (1) BY 0.14528, (4) BY 0.4279 AND ADD: 0. $7888 V_A = 58.01$ $V_A = 735.4 \text{ m/s}$ $V_A = 735 \text{ m/s}$

FROM (1): $0.1667(735,4) - 0.4229V_8 = -100$ $V_8 = 573.6 \text{ m/s}$ $V_8 = 574 \text{ m/s}$

FROM (3): 0.08333 (735.4) - 0.12816 (573.6) - 0.2983 7 = -160 1 = 495.4m/s V = 495 m/s



GNEN: 12-16 SHELL EXPLODES ATD INTO FRAGMENTS, A(5/6), B(4/6), AND C(3/6), WHICH HIT WALL AS SHOWN. VELOCITY OF SHELL WAS Yo = (40ft/s) i - (30ft/s) i - (1200ft/s) k. FIND: SPEED OF EACH FRAGMENT

CONSERVATION OF MOMENTUM!

 $(12/9) \underbrace{v_0}_{0} = (5/3) \underbrace{v_A}_{0} + (4/3) \underbrace{v_B}_{0} + (3/3) \underbrace{v_C}_{0}$ $12(40 \underbrace{i-30}_{0} - 1200 \underbrace{k}) = 5(-\frac{5}{13} \underbrace{i-\frac{12}{13}}_{13} \underbrace{k}) \underbrace{v_A}_{0} + \underbrace{v_C}_{0}$

4(31+51-3K) +3(-31-4K) TC

EQUATE COEFFICIENTS OF UNIT VECTORS: $(1) - \frac{25}{13} V_A + \frac{9}{3} V_B = 480$

 $\frac{4}{8} \sqrt{g} - \frac{9}{2} \sqrt{c} = -360$ (e) $\frac{4}{8} \sqrt{g} - \frac{9}{2} \sqrt{c} = -14,400$ (3)

Solving these Equations SIMULTANEOUSLY: $V_B = 1389,84$, $V_C = 1229,51$

T = 1678 ft/s; V = 1390 ft/s; V = 1730 ft/s

14.26 SEE FIGURE AT TOP OF PAGE

GIVEN: 12-16 SHELL EXPLOSES AT DINTO FRACTO A (416), B(316), AND C(516), WHICH HIT WALL IS SHOWN. VELOCITY OF SHELL WAS 95-(+05+1/5):-(30+1/5)/2-(1200+1/5)/K

FIND: SPEED OF EACH FRAGMENT

(12/4) $v_0 = (4/9) v_A + (3/8) v_E + (5/4) v_C$ 12 (40 i - 30 j - 1200 k) = 4 (-\frac{5}{13} \dots - \frac{12}{5} \dots - \frac{1}{5} \dots - \frac{1}{5} \dots \dots

EQUATE COEFFICIENTS OF UNIT VECTORS:

 $\sqrt{E} = \sqrt{V_B} - 3 V_C = -360$ (2)

SOLVING THESE EQUATIONS SIMULTANEOUSLY: V_A = 2097.05, V_B =1953.11, V_C = 737.705

V = 2097 ft/s; V = 1853 ft/s; V = 730 ft/s

14.27 DERIVE $H_0 = \overline{z} \times m\overline{v} + H_G$, WHERE $H_0 = \overline{z} (\underline{z}_i \times m_i \underline{v}_i)$ (14.7) $H_0 = \overline{z} (\underline{z}_i \times m_i \underline{v}_i)$ (14.24)

AND m = TOTAL MASS OF SYSTEM, $\overline{z} = \text{POSITION VECTUR OF G}$; $\overline{v} = \text{VEWCITY OF G}$.

MAKING 1: = = + 1: IN = Q.(14.7): H = \(\bar{2} + \bar{2}_{\bar{1}} \right) \times m; \(\bar{2}_{\bar{2}} \right) \times m; \(\bar{

 $= \Xi \times \Sigma m_i \underline{v_i} + \Sigma \underline{z_i} \times m_i \underline{v_i}$

BUT Im, V. = my

AND, BY (14.24): \(\Sigma\); \(\mathbf{x} \mathbf{m}; \mathbf{y} = \mathbf{H}_6\)

THEREFORE: $H_0 = E \times m V + H_G$ (Q.E.D.)

14.28 DERIVE EM = H (14.23)

DIRECTLY FROM EM - H (14.11)

BY USING EQUATION DERIVED IN PROB. 14.27.

Y EM SET WE REDUCE THE FORCES
TO THE VECTORS SHOWN.

IT FOLLOWS THAT $\sum_{n=2}^{\infty} X \sum_{r=0}^{\infty} F + \sum_{r=0}^{\infty} (1)$

FROM PROB 14.27: $H = \frac{1}{2} \times mV + \frac{H}{G}$ DIFFERENTIATE: $H = \frac{1}{2} \times mV + \frac{1}{2} \times mV + \frac{H}{G}$ $= \frac{1}{2} \times mV + \frac{1}{2} \times 400 + \frac{H}{G}$

BUT VXMV=0 AND MA= ZF. THUS H = E X ZF+ HG (2)

SUBSTITUTE FUR EMOTRUM (1) AND HOTROM (2)

 $\overline{z} \times \Sigma F + \Sigma M_G = \overline{z} \times \Sigma F + U_G$ $\Sigma M_G = H_G \quad (O.E.D.)$

14.29

GIVEN:

N=1-1111-11 FRAME OXYZ

AND FRAME AZ'Y'Z' IN

TRANSLATION W/R TO OXYZ.

LET H' = \(\frac{1}{2} \) \(\text{X''} \) \(\frac{1}{2} \) \(\frac{1}{2}

WHERE TE AND TO DENNE VELUCITIES W/R AZ'I'Y' AND OZYE, RESPECTIVELY

SHOW THAT HA = HA GIT GIVEN INSTANT

IF, AND ONLY IF, ONE OF THE FOLLOWING
CONDITIONS IS SHISFIED AT THAT INSTANT:

(a) VA = O WITH RESPECT TO OZUZ,

(b) A COINCIDES WITH MASS CENTER & 119 SYSTEM OF PARTICLES.

(C) IA IS DIFFICTEL ALONG AG.

(CONTITUES)

14.29 continued WE RECALL:

H' = \ E e'; x m; V. (1) (2) HA= Σ全: ×加口

LET U: = + 1, IN EQ. (2):

 $\vec{H}_{A} = \sum \vec{\epsilon}_{i}^{2} \times m_{i} \left(\vec{v}_{A} + \vec{v}_{i}^{2} \right) = \left(\sum m_{i} \vec{\epsilon}_{i}^{2} \right) \times \vec{\gamma}_{i}^{2} + \sum \vec{\epsilon}_{i}^{2} \times m_{i} \vec{v}_{i}^{2}$ BUT, BY (14.12): ZM(2: = m 2' = m AG RECALLING EQ.(1) WE WRITE

HA = MAGX VA + HA

THIS EQUATION REDUCES TO HA = H' IF (a) 1/A = 0, (b) A = G, (c) 1/A // AG (Q.E.D.)

GIVEN: 14.30

FRAME AZYZ' IN TRANSLATION WITH RESPECT TO NEWTONIAN FRAME OZYZ. LET H'= \(\frac{1}{2}\) xm v?

WHERE I AND 11 ARE DEPINED W/R FRANE AZY'E AND LET ZMA BE THE SUM OF THE MOMENTS OF THE EXTERNAL FURCES ABOUT A.

SHOW THAT THE RELATION EM = H

IS VALID IF, AND ONLY IF, ONE OF THE FOLLOWING CONDITIONS IS SATUFIED:

(a) Az'y'z' IS A NEW TON! AN FRAILE OF REFERENCE (b) A CONCIDES WITH MASS CENTER G OF SYCTEH OF PARTICLES.

(c) a 15 DIRECTED ALONG AG

DIFFERENTIATE EQ.(1):

 $H'_{A} = \sum \dot{z}_{i}^{2} \times m_{i} \, \underline{v}_{i}^{2} + \sum \dot{z}_{i}^{2} \times m_{i} \, \underline{v}_{i}^{2}$ = EV, x m, v, + E+ x m; a;

BUT VixVi= 0 AND ai = ai - aA

THUS: H'= \(\begin{array}{c} \dag{\text{t}} \dag{\t BUT, BY (14.12): [mi] = mi = mAG

AND, SINCE Q. IS ACCELERATION WIR NEW TONIAN

PRAME, WE HAVE, BY EQ. (14,5),

 $\Sigma(\xi_i^* \times m_i a_i) = \Sigma(\xi_i^* \times F_i) = \Sigma M_A$

THEREFURE HA = EM - mAG X a

THIS EQUATION REDUCES TO HA = EM IF

- (a) a= OSFRAME Ax'y'z' IS IN UNIPERM TRANSLATION WR NEWTONIAN FRAME DZYZ AND IS ITSELF A NEWTONIAN FRAME,
- (b) AG=0; A COINCIDES WITH G,
- (c) AGX a = 0; a IS DIRECTED ALONG AG (Q.E.D)

GIVEN: REFERRING 70 PROB. 14.1. ASSUME THAT

(1) 15-kg SUITCASE PIRST TOWED WITH W = 3 m/s -

(2) 20-to SUITCASE THEN TOSSED WITH U= 2 m/s > (3) 25-kg CARRIER INITIALLY AT REST.

FIND: ENERGY LOST AS

(a) FIRST SUITCHEE HITS CARLIER

() SECOND SUITCASE HITS CARRILR

a) BEFORE FIRST SUITCASE HITS CARRIER! $T_0 = \frac{1}{2} m v_0^2 = \frac{1}{2} (15 \text{ kg}) (3 \text{ m/s})^2 = 67.50 \text{ J}$

FIRST INPACT: CONSERVATION OF MARK INTO IN (15kg)(3m/s) = (25+15) V, V,=1.125 m/s

T, = = (25kg + 15kg)(1.125m/s) = 25.313J

EN, LOST = To - T, = 67.50 J - 25.313J = 42.2 J

(b) JUST BEFORE SECOND SUITCASE HITS: T, = T, + 1 (20kg)(2m/s)=25.313J + 40J

= 65.313 J SECOND IMPACT: CONSERVATION OF MINIENTUM (25kg+15kg)(1,125m/s)+(20kg)(2m/s)=(60kg) V2

0,=1,416/m/s

72 = \$ (60 kg)(1.4167 m/s) = 60.208 J ENLOST = T1 - T, = 65,313 J-60,2083 = 5,10 J

GIVEN: CULLISIONS DESCRIPED 14.32 IN PROB. 14.5. WE RECALL THAT INITIAL VELOCITY OF CAR A WAS VA = 1,920 11/5

AFTER A HITS B: $(V_B)_1 = 1.680 \,\text{m/s}$ AFTER B HITS C: $(V_B)_2 = 0.210 \,\text{m/s}$

AFTER A AGAIN HITS B. (DB) = 0.23625 M/s MASS OF EACH CAR = 1500 kg

FIND! ENERGY LOST HFTER ALL COLLISIONS HAVE TAKEN PLACE.

PROM SOLUTION OF PRIB. 14.5 WE HAVE THE FOLLOWING FINAL VELOCITIES!

VA = 0,21375 m/s, VB = 0,23625 m/s. Vc = 1,470 m/s

To = = 1 m1 V02 = = (1500kg)(1,920 m/s) = 2764.8J

FINAL ENERGY:

Ty = - 1 m V 2 + 1 m12 2 + 1 m1 V = = 1 m1 (VA+VB+VC) = 1 (1500 kg) [(0.21375 m/s)+(0,23625 N/s)+(1,470 m/s)] = 1696.8 J

ENERGY LOST

 $= T_A - T_I = 2764.8 J - 1696.8 J = 1068 J$

14.33 GIVEN:

180-16 MAN AND 120-16 WOMAN OF PROB.

14.3 JUMP FROM SAME END OF 300-16 BOAT
WITH VELOCITY OF 16 ft/s WITH RESPECT TO BOAT.

FIND:

WORK DONE BY WOMAN AND BY MAN IF WOMAN DIVES FIRST.

TOTAL KE. AFTER WOMAN DIVES

FROM PART a OF SOLUTION OF PROB. 14.3:

VEL. OF BOAT = $(V_B)_1 = 3.20 + t/s$ THUS, VEL. OF WOMAN = $(V_W)_1 = 16 - 3.20 = 12.8 + t/s$ K.E. = $T_1 = \frac{1}{2} m_W^2 (v_W^2)_1^2 + \frac{1}{2} (m_B t m_M) (v_B)_1^2$ $T_1 = \frac{1}{2} \frac{120}{32.2} (12.8)^2 + \frac{1}{2} \frac{120}{32.2} (3.20)^2 = 381.61 + 1.16$ WORK OF NOMAN = $T_1 = 982 + 1.16$

TOTAL K.E. AFTER HAN DIVE

FROM ANSWER TO PARTO OF PROP. 14.3:

VEL. OF EOA! = $(V_B)_2 = 9.20 \text{ ft/s}$ THUS, VEL. OF KAN = $(V_M)_2 = 16 - 9.20 = 6.80 \text{ ft/s}$ K.E. = $T_z = \frac{1}{2} m_W (V_W)_1^2 + \frac{1}{4} m_M (V_M)_2^2 + \frac{1}{4} m_B (V_B)_2^2$ = $\frac{1}{2} \frac{120}{32.1} (17.8)^2 + \frac{1}{2} \frac{180}{32.2} (6.80)^2 + \frac{1}{2} \frac{300}{32.2} (9.20)^2$ $T_z = 828.82 \text{ ft/b}$ WORK OF 11AN = $T_z - T_z = 829.82 - 381.61 = 447 \text{ ft/b}$

14.34 GIVEN:

BULLET OF PROB. 14.7 FIRED WITH $V_0 = 1500 \text{ ft/s}$ THROUGH 6-16 BLOCK A BECOMES EMPEDDED IN 4.95-16 BLOCK B. BLOCKS MOVE WITH $V_A = 5 \text{ ft/s}$ AND $V_B = 9 \text{ ft/s}$.

FIND:

ENERGY LOST AS BULLET

(a) PASSES THEOUSH PLOCE A

(b) BECOMES EMBEDDED IN BLOCK B

FROM AUGULKE TO PLOB. 14.7: .
WEIGHT OF EULLET = W = 0,800 OR = 0.0500 16
VEL. OF BULLET BETWEEN BLOCKS = V = 900 ft/s

(a) ENERGY LOST AS PULLET PASSES THROUGH A

INITIAL K.E. = To = 1 W Vo = 1 0.0500 15 (1500 ft/s)2

To = 1746.89 ft. 16

K.E. OF SYSTEM AFTER BULLET PASSES THROUGH H.

= $T_1 = \frac{1}{2} \frac{44}{3}$, $\pi_1^2 + \frac{1}{2} \frac{44}{3} \pi_A^2 = \frac{1}{2} \frac{0.0500}{37.2} (900)^2 + \frac{1}{2} \frac{6}{37.2} (5)^3$ $T_1 = 628.80 + 2.33 = 631.21 \text{ H·lb}$

EN. LOST = 75-T, = 1746.89-631.21 = 1116 ft.16

(b) ENTRGY LOST AS YULLET LEWISE ENERDEDINE FINAL KE. = $\frac{1}{2} = \frac{1}{2} \frac{WA}{g} R_a^2 + \frac{1}{2} \frac{(W_B + a^2)}{g} V_B^2$ $T_2 = \frac{1}{2} \frac{6}{322} (5)^2 + \frac{1}{2} \frac{1.95 + 0.05}{32.2} (9)^2 = 8.616 \text{ flib}$

EN.1.057 = T-T = 631,21-8.618 = 623 ft. 16

14.35 GIVEN: AUTUMOBILE A, OF MASS MA, COLLIDES WITH AUTUMOBILE B OF MASS MB WE ASSUME PLASTIC IMPACT AND THAT ENERGY ARGRED BY EACH AUTOMOBILE EQUALS ITS K.T. WITH RESPECT TO HOVING FRAME ATTACHED TO MASS CENTER OF SYSTEM



(a) SHOW THAT $E_A/E_B=m_B/m_B$, WHERE (E_A AND E_B ART ENERGIES ABSORBED BY A AND E_B (b) FIND E_A AND E_B IF $m_A=1600$ kg, $m_B=900$ kg, $V_A=90$ km/h, $V_B=60$ km/h.

BEFORE CULUSION. VELOCITY TO OF MASS CENTER G:

(MATMB) T = MAVA - MBVB

T = MAVA - MBVB

MATMB

MOTION OF AUTOS RELATIVE TO G: $V_{A/G} = V_A - \bar{V} = V_A - \frac{m_B V_A - m_B V_B}{m_{A+m_B} v_B} = \frac{m_A V_A - m_B V_B}{m_{A+m_B} v_B} = \frac{m_A V_A - m_B V_B}{m_{A+m_B} v_B} = \frac{m_A V_A - m_B V_B}{m_{A+m_B} v_B v_B} = \frac{m_A V_B}{m_{A+m_B} v_B} = \frac{m_A V_B}{m_A v_B} = \frac$

 $T_{B/G} = \frac{1}{2} m_B V_{B/G}^2 = \frac{1}{2} \frac{m_A m_B}{(m_A + m_B)^2} (V_A + V_B)^2$ (2)

AFTER COLLISION.

SINCE THERE IS NO EXTERNAL PORCE, G KEEPS
MOVING WITH VELOCITY V.

SINCE IMPACT IS PLASTIC! V' = V' = V

AND VAIG = Vaig = 0. THUS: TA' = TB' = 0

IT POLLOWS THAT EA = TA' AND EB = TB' G

(a) DIVIDING (1) BY (2): $\frac{E_0}{Z_0} = \frac{T_{A/b}}{T_{B/c}} = \frac{m_B}{m_A}$ (0. E. D)

(b) SUBSTITITING (N (1) AND (2) THE GIVEN DATA, $m_A = 1600 \text{kg}$, $m_B = 900 \text{kg}$, $v_B = 900 \text{kg}$, $v_B = 60 \text{kg}$.

14.36 GIVEN: CAR COLLISION OF TROB. 14.35
DEFINE: SEVERTY OF A COLLISION = E/ED

WHERE E = ENERGY ARSOFRED BY SAME CAR IN COLLISION,

AND E = EN. ABSORBED BY SAME CAR IN A TEST

WHERE IT HITS AN IMMOVABLE WALL WITH VELOC. VO

SHOW THAT COLLISION OF PROB. 14.35 IS (MA/MB)

TIMES MORE SEVERE FOR CAR & THAN FOR CAR A.

ENERGIES ABSORBED IN TESTS OF A AND 8: $(E_A)_0 = \frac{1}{2} m_A V_0^2 \qquad (E_B)_0 = \frac{1}{2} m_B V_0^2 \qquad (3)$

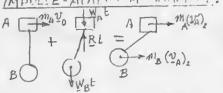
SEVERITY OF COLLISION FOR CAR A = EA/EA)
SEVERITY OF COLLISION FOR CAR B = EB/EB)
RECALLING ERS. (3) AND PROM PROB. 14.35 THAT

ENTERITY OF COLL FOR B FR(FA)

SEVERITY OF COLL. FOR B = $\frac{E_B(E_A)_0}{E_B(E_B)_0}$ = $\frac{m_A}{m_B} \frac{\frac{1}{2}m_Bv_0^2}{\frac{1}{2}m_Bv_0^2} = \left(\frac{m_A}{m_B}\right)^2$ (Q.E.D.)



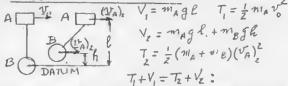
(a) VELOCITY OF B AT HAXIMUM ELEVATION IMPULSE-ALMENT M CHINCH'S



WE MOTE THAP WHEN B REICHES HA: MUM HEIGHT (VB) = (TH) =

Imv, + ZExt Imp = Zmv2 to x COMP: MA V = (MA + ME)(VA). $(V_B)_2 = (V_A)_2 = \frac{m_{IA}}{m_{IA} + m_{IB}} V_0 \rightarrow (1)$

(P) HUXIMUM HEIGHT KEACHED BY B CONSERVATION /FENELGY:



1 m A V + magt = + (nin + m B (VA) 2 + ningt + me] ti SUBSTITUTING FOR (VA) FREM (1).

I MAN = 1 MA VO + MBG? P1 - 10 2 21 + 11 H HE - 14 H h = 11/4+ 11/2 23

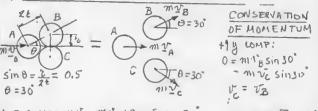
(SAME ANSWER AS FUR PART & OF SP14.4) 14.38

GIVEN: BALL A HITS WITH 1 BALLS B AND C WHICH ARE AT REST. ASSUME CONSERVATION OF ENERGY.

(CONTINUED)

FIND: FINAL VELOCITY OF EACH BALL, IF (1) A STRIKES B AND C SIMULTANEOUSLY, (b) H HITS & BEFORE IT HITS C

(a) A STRIKES B AND C SIMULTANELUSLY



1 Z (CAIP: MIND= MINA +2m VB cos 30,

15-15=VV3 CONSERVATION OF ENERGY: Jo- JA = 21E 2 m Vo2 = 1 m va + 1 (2 m v B) · Vo + VA = = = 3 VB DIVIDE (2) BY (1): If DD (1) AND (3): $2V_0 = \frac{2+3}{\sqrt{3}} V_B$ FROM (1): $V_A = V_0 - \frac{6}{5} V_0 - \frac{7}{3} - \frac{7}{4} V_0 / 5$ VB= 3 5 3 VO VA = 0,200 V, -; V = 0.693 V, - 30"; 1 = 0.6931 5 30"

(b) A HITS B BEFORE C 14.38 continued my my

CONS OF MIDMENTUM FROM 1 TO'R: (UA) = 0 - Vg 00330 to X COMP. MV = m (VA) + mVB cos30° (va) = 1/2 sin20° 9 3 cox =: 0 = m(vi) +m vesin300 SMITH BOTH MENBERS OF (4) AND (5) AND ADD: (6)

CONS. OF ENERGY FRUM 1 TO Z: 15 H = U - VE + ni10 = + ni20 = + = mi12 CARRYING IN TO (6) AND SOLVING FOR UE: UB = VO COS 30 (7)

CONS. OF HUMENTUM FROM 1 713:

+ 2 (01 F: m 1 = m (TA) + m v cos 30"+m v cos 30" thy course o = m(vh) + + + = sin30 - mvcsin30° CUBSTITUTE FOR TO FRAM (7) AND SOLVE FOR (17) AND (16) (VA)y = - Vo sin 30 cos 30 + Vo sin 30 SQUARING AND ADDINE: 172 = 0.25 15-0.866 15 1/2 + 16 (10)

CINS, OF INEKRY FROM 1 TO 3: 5 = 15 + VB+VE ラガイー= 1 mvA+ 1m12+ 1mve SLESTITITE FIR IT AND US FRAM (10) AND (7) INTO (1): Vo= 0.35 Vo- 0.866 Vo Vc + Vc+ 0.75 Vo+ 16, V= 0.433 Vo CARRY 116 1 273 (8) AND (7): (Un) = 0.2510 - 0.433 (0.30° = - 0.1250 20 (+ A) = -1,433 = +0,435 Vo sin 30 = -0.2165 Vo THUS: 1-0.250 NO F 60; UB = 0.866 No 230; U=0.433 V. 530



GIVEN: A HITS BWITH V = 15ft/s. ASSUME CONS. OF ENERGY FIND: MAGNITUDES OF UA, TB, AND TC.

CONS, OF HEARNTUM! +, I COMP: my Vo cos45" - my sing of my cas 30" (1) + 1 y aup: 1117 5:1145 = m vg - mv gos30 + mv 5:1130 (2) MULTIPLY (1) BY SIN 30°, (2) BY COS 30°, SUBTRACT, AND (3) VP = 0.8660 VA - 0.2588 VO

CARRY INTO (1) AND SCLUE FOR VC! V_ = 0,8165 (- 0.57735(0.8660 VA - 0.2588 50) VC = - 0.5 VA +0.9659 VA CONS. OF ENERGY:

1 m vo = 1 m va+ 1 m ve, vo= va+ va+ ve SUBSTITUTE FOR UB AND VC FROM (3) AND (4). 10 = Un + (0.8660 VA - 0.25 88 VD) + (-0,5 VA +0. 1657 VD) 2VA - 1,4141 VO VA = 0 VA = 0.7071 VO TRIM (3) ANG(4): UB- 0.3536 UO, VC=0.6124 VO GIVEN DATA: 10 = 15 +1/s. THEREFURE:

VA = 10.61 ft/s; VB = 5.30 ft/s; VC = 9.19 ft/s



A HITS B WITH U = 15ft/s ASSUME CONSERVATION DF ENERGY. .

FIND: MAGNITUDES OF VA. VB, AND VC.

CONS. OF MOMENTUM:

\$ 2 COM P. M VO COS 30" = M VB Sin45" + 21 VE COS 45" (1) thy comp. " muto sin30" = mot - nivo cos45" + motesin45"

SUBTRACT (2) FROM (1, AND DIVIDE BY M:

VB = 0.707/15+0.258845 (3) 0.3660 Vo = - VA +1.4142 VE ADD (1) AND (2) AND DNIDE BY MI

Vc = -0.7071 2/ + 0.9659 5 (4) 1.3660 No = VA + 1.4142 25

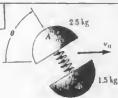
CONS. OF ENERGY:

グランカナップンナンC = n1v== = 1 mV++ 121-12+ 1 mV SUBSTITUTE FOR UB HADVE FROM (3) AND (4): Vo = VA+(0.7071VA+0.2588 V5)+(-0.7071VA+0.859V5)

2252 - VO VA =0 VH = 0.516 FECH(2) AUD(4): 1=0.6124 Jo, Vc=0.6124 Jo

GIVEN DATA: 10 = 15 ft/s. THEREFORE: U =7.50 ft/s; VB = 9.19 ft/s; Vc = 9.19 ft/s

14.41



GIVEN: 15 = 8 m/s. POTENTIAL ENERGY OF SPRING = 120 J. CORD CUT WHEN B = 30.

FIND! MY AIL TU AFTER CORD IS CUT.

(1)

CONS, OF LINEAR MON. WR FRAME GZ'Y'E



B & CTAHENLATS: O= MAY - nig Ta 13 = 114 T' = 2.5 V' いるるか

CONS. OF ENERGY W/R FRAME GZ'1'2'; $120J = \frac{1}{2}(2.5)V_A^{12} + \frac{1}{2}(1.5)V_P^{12}$

SUBSTITUTE FOR (B FRUM (1) INTO (2): 17A = 36 5 VA +3 (517) = 480 10 = 6 11/5 20

FROM (1): $V_B^2 = \frac{\pi}{3} (6 \text{ m/s})$ WITH RESPECT TO FIXED FRANE DXYE:

V = V + 17 = 8 m/s -+ + 6 m/s 100 (3)

VB = V0 + V' = 8 m/s → +10 m/s 50 (W)

FOR 0-30: Eu.(3):

\$ 2 COMP: (17) = 8 - 6 COS30 = 2.804 m/s

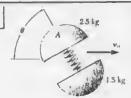
=> } coll: (1 A) = 6 sin 30 = 3 11/6

V= 4.11 m/3 & 46.90

+ X comp: ((B), = 8+10 cos 30 = 16,660 m/s => y comp: (ve) = -10 sin 30 = -5 m/s

V = 17,39 m/3 5 16.7°

14.42



GIVEN: U = 8 m/s. POTENTIAL ENERGY OF SPRING = 120J. CORD CUT WHEN 0=120.

UA AND UB AFTER THE CORD IS CUT.

SEE SOLUTION OF PROB. 14,41 FOR DERIVATION OF EAC. (3) AUD (4). WITH 0 = 120°, WE HAVE

UA = U + VA = 8 m/s + 6 m/s & 60

1 = 1 + 1 = 8 m/s + 10 m/s 7 60°

(3") (41)

Ea:(3'): + 14 COMP: (VA) = 8+6 cos 60° = 6+3= 11 m/s + 14 COMP: (VA) = 6 sin 60° = 5,196 m/s

Va= 12,17m/s & 25,3° EQ.(4'): ± X COMP: (VB) = 8-10 CUS 60° = 8-5 = 3 m/s + 9 y comp: (Va) = -10 sin 60° = - 8.660 m/s

V = 9.17 m/s \$ 70.9°

14.43

GIVEN:

THREE SPHERES EACH OF MASS M. A AND B ARE CONNECTED BY TAUT, INEXTENSIBLE CORD. C STRIKES B AS SHOWN. ASSUME CONS. OF ENERGY. FIND:

VELOCITY OF EACH SPHERE AFTER IMPACT.

EFFECT ON CONSTRAINTS ON FINAL VELOCITIES

m 15 C + C = = 1 = VA 260

BECAUSE CORD AB IS INEXTENSIBLE COMPONENT OF 1/B A LONG AB MUST BE EWVAL TO YA.

U = VAZ 60+ VB/A530 (1)

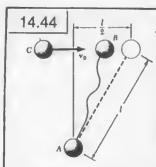
CONS. OF MOMENTUM FOR SYSTEM;

から=かでナスかかみ+かである HY y COMP: 0 = 2 m 1/A sin60 - mV sin30 VB/A = 213 VA ± 2 COMP:

mv = mv + 2 mv cos 60°+ mv 8/4 cos 30 DIVIDING BY M AND SUBSTITUTING FOR UBYA FROM (2): No= VC + VA + 4 (2 V3 VA) 15 = V0 - 4 VA

CONS. OF ENERGY: 1 1 V02 = 1 m V2 1 m V2+1 m VC, U=2UA+UB/A+Vc (4) SUBSTITUTE FOR UB/A AND UC FROM (?) AND (3) INTO (4): 1 = 2 VA+ 17 VA+ B2-8 VOVA+ 16 VA UA = 4 V 260° FROM (3): U= U0-16 V0=-15 V0 Vc = 15 Vo+

FROM(1) AND (2): U= " v &60+ 853 v + 30, U=0.961 v 13.9"



THREE SPHERES, FACH OF MASS M.

A AND B ARE CONNECTED BY
INEXTENSIBLE CORD WHICH IS
SLACK. C STRIKES B AS SHOWN
WITH PERFECTLY ELASTIC HYPACT.
FIND:

(A) VELOCITY OF EACH SPHERE AFTER CORD BECOMES TAUT.
(b) PRACTION OF INITIAL K.E.
LOST WHEN CORD BECOMES TAUT.

(a) DETERMINATION OF VELUCITIES

CONS. OF ENERGY (PERFECTLY ELASTIC IMPACT):

 $\frac{1}{2}mv_0^2 = \frac{1}{2}mv_0^2 + \frac{1}{2}mv_1^2 \qquad v_2^2 + v_1^2 = v_0^2 \qquad (2)$ SQUARE (1): $v_2^2 + v_1^2 = v_0^2$

SQUARE (1): $V_{i}^{*}+2V_{i}V_{i}+V_{j}^{*}=V_{0}$ SUBTRACT (2): $2V_{i}V_{j}=0$

N=0 CORRESPONDS TO INITIAL CONDITIONS AND SHOULD BE ELIMINATED. THEREFORE N=0

FROM (1): V; = V

A A A BECOMES TAUT

BECAUSE CORD IS

THEXTENSIBLE, COMPONENT

OF UB ALONG AB MUST

BE EQUAL TO UA.

CONS. OF MOMENTUM:

mv = 2mv + m v B/A

TEYA = 2V3 VA (3)

\$\frac{1}{2} \text{Z COMP: m V_0 = 2 m VA COS 60 + m V B/A COB 30°} \text{DIVIDING BY m AND SUBSTITUTING FOR VB/A FROM (3):

 $V_0 = 2 V_A(0.5) + (2\sqrt{3} V_A)(\sqrt{3}/2)$ $V_0 = 4 V_A$ $V_A = 0.250 V_0$ $V_A = 0.250 V_0$

CARRYING INTO (3): UB/A = 2 U3 (0,250 U) = 0.866 Vo

B $U_A = 0.250 \text{ T}$ $U_B = U_A + V_{B/A}$ $U_B = 0.250 \text{ T}$ $U_B = 0.250 \text{ T}$

 $v_{\rm E} = 0.875 \, v_{\rm 0} \, i - 0.2165 \, j$

2 = 0,4012915 = 13.90° VB=0.9015 \$13.9°

(b) FRACTION OF K.E. LOST

 $T_{O} = \frac{1}{2} \pi V_{0}^{2}$ $T_{FINAL} = \frac{1}{2} m V_{A}^{2} + \frac{1}{2} m V_{B}^{2} + \frac{1}{2} m V_{C}^{2}$ $= \frac{1}{2} m (0.250 V_{0})^{2} + \frac{1}{2} (0.90139 V_{0})^{2} + \frac{1}{2} m (0)$ $= \frac{1}{2} m (0.875) V_{0}^{2}$ $k.E. LUST = T_{0} - \frac{1}{FINAL} = \frac{1}{2} m (1-0.875) V_{0}^{2} = \frac{1}{2} \frac{1}{8} m V_{0}^{2}$ $fRACTION OF k.E. LUST = \frac{1}{8}$

14.45 GIVEN:

360-kg SPACE VEHICLE WITH $\mathcal{N} = (450 \text{M}) \text{k}$, AS IT PASSES THROUGH D, EXPLOSIVE CHARGES SEPARATE IT INTO 3 PARTS: A (60 kg), B (120 kg), AND C (180 kg). SHORTLY INTER, THE POSITIONS OF THE 3 PARTS ARE A (72 m, 72 m, 648 m), B (180 m, 396 m, 972 m), C (-144 m, -288 m, 576 m). VELOCITY OF B IS $\mathcal{L}_B = (150 \text{ m/s}) \dot{i} + (330 \text{ m/s}) \dot{j} + (660 \text{ m/s}) \dot{k}$. X-comp of VELOCITY OF C IS $(\mathcal{N}_C)_Z = -120 \text{ m/s}$. FIND: VELOCITY OF A.

CONSERVATION OF AIVGULAR MOMENTUM ABOUT O SINCE VEHICLE PASSES THROUGH O, HO = 0, OR

Ho = 1A × mAVA + 1B × mB VB + 1c×mc Vc = U USING DETERMINANT FORM:

 $H_0 = 60 \begin{vmatrix} i & i & k \\ 72 & 72 & 640 \\ (Va)_2 (Va)_3 (Va)_3 \end{vmatrix} + 120 \begin{vmatrix} i & 0 & 12 \\ 150 & 330 & 660 \end{vmatrix} + 180 \begin{vmatrix} -144 & -280 & 576 \\ -120 & (Va)_2 (Va)_3 \end{vmatrix} = 0$

EQUATING TO ZERO THE COEFF. OF i, i, k, AND DIVIDING BY 60:

 $(72(V_A)_g - 648(V_A)_g - 1128 \times 10^3 - 264(V_C)_g - 1728(V_C)_g = 0$

(1) 648(Va)2-72(VA)=+54.0×103-207.36×103+432(VE)2=0

(F) 72 (VA)y - 72(VA)2 +0 - 452 (VC)y - 103.68×103 = 0

OR, AFTER REDUCTIONS:

 $(V_A)_3 - 9(V_A)_y - 12(V_C)_2 - 24(V_C)_y = 1650$ (1)

 $-(V_A)_2 + 9(V_A)_2 + 6(V_C)_2 = 2130$ (2)

 $(V_A)_y - (V_A)_z - 6(V_C)_y = 1440$ (3)

CONSERVATION OF LINEAR MUMENTUM

 $m \underline{v}_{h} = m_{h} \underline{v}_{h} + m_{h} \underline{v}_{h} + m_{c} \underline{v}_{c}$ $360(450 \underline{k}) = 60[(\sigma_{h}, \underline{i} + (v_{h}), \underline{j} + (v_{h}), \underline{k}] + 120[170\underline{i} + 330\underline{j} + 660\underline{k}] + 180[-120\underline{i} + (v_{c}), \underline{k}]$

EWVATING THE COFFT OF THE UNIT VECTORS HAD DINIDINGBY 60:
(1) $(V_{\rm H})_z + 300 - 360 = 0$ $(V_{\rm H})_z = 60 \, \text{m/s}$ (4)

(1) (1/n) +660 +3 (1/c) = 0 (1/n) = -660-3(1/2) (5)

(V_A)₂ + 1320 + 3(V_C)₂=2700 (V_A)₃=1380 - 3(V_C)₂ (6)

SUBSTITUTING FRUM (4), (5), (6) INTO (2) AND (3): - 1380 + 3 (V_c), + 9(60) + 6(V_c), = 2130 (V_c), = 930 m/s - 660 - 3 (V_c), - 60 - 6 (V_c), = 1440 (V_c), = -240 m/s

SUBSTITUTING FOR (VC), AND (VC), INTO (S) AND (6):

 $(V_A)_1 = -660 - 3(-240) = 60 \,\text{m/s}$

(VA)2= 1380 - 3 (330) = 390 m/s RECALLING FROM (4) THAT (VA)x=.60 m/s, WE HAVE

Un= (60.0 m/s) i + (60.0 m/s) j+(390 m) k

CHECK

SINCE EQ. (1) WAS NOT USED IN OUR SOLUTION, WE CAN USE IT TO CHECK THE ANSWER. SUBSTITUTING THE VALUES OBTAINED FOR $(V_A)_1$, $(V_A)_2$, $(V_C)_2$, AND $(V_C)_3$ INTO THE LEFT-HAND MEMBER OF EQ. (1), WE OBTAIN 390-9(60)-12(330)-14(-240)=340-540-3960+5760=1650 O.K

GIVEN: 14.46 IT IS KNOWN THAT PARTICE A IS PROJECTED FROM A, (260, -20,340) AND COLLIDES WITH C AT Q(200,180,140) COURDINATES OF BO WHERE PATH OF BINTERSECTS 27 PLANE =60i-200j+200k QB=(02)1+(0y); +(02)+ QB1=2B-20=(107i+200+170+)-(2001+180+1401) = -93 i +20 j +30 k 40 = -4801 +600g - 640k / AND, FROM SOLUTION OF PROB. 14.24in = -242.61 + 514.6 j - 73.51 K k | 1½ à ±1 -480 600 -640 0 480 0 (i) 1/4 (8000) - 48007 = -16408

IN SCATTERING EXPERIMENT OF PROB. 14.24

CONS. OF ANGULAR MOMENTUM ABOUT Q: SINCE PATHS OF A HETER CULLISIONS AND OF C BEFORE AND AFTER COLLISION PASS THROUGH Q THE CORRESTONDING ANG, MOHENTA ARE ZERO (FIG. P14.24). CONS. OF ANGULAR MOMENTUM OF ALL PARTICLES ABOUT Q IS EXPRESSED AS

$$\widehat{Q}\widehat{A}_0 \times \frac{m}{4} \frac{u}{t_0} + \widehat{Q}\widehat{B}_0 \times m \underline{v}_0 = \widehat{Q}\widehat{B}_1 \times m \underline{v}_g \tag{1}$$

WHERE
$$Q \tilde{h}_0 = \frac{L}{A_0} - \frac{L}{A_0} = (260i - 20\frac{1}{0} + 340i) - (200i + 180\frac{1}{0} + 140i)$$

$$= 60i - 200\frac{1}{0} + 200\frac{1}{0}$$

$$Q \tilde{h}_0 = \frac{L}{A_0} - \frac{L}{A_0} = (260i - 20\frac{1}{0} + 340i) - (200i + 180\frac{1}{0} + 140\frac{1}{0})$$

$$= 60i - 200\frac{1}{0} + 200\frac{1}{0}$$

$$Q \tilde{h}_0 = \frac{L}{A_0} - \frac{L}{A_0} = (260i - 20\frac{1}{0} + 340i) - (200i + 180\frac{1}{0} + 140\frac{1}{0})$$

$$= 60i - 200\frac{1}{0} + 200\frac{1}{0} + 200\frac{1}{0}$$

$$Q \tilde{h}_0 = \frac{L}{A_0} - \frac{L}{A_0} = (260i - 20\frac{1}{0} + 340i) - (200i + 180\frac{1}{0} + 140\frac{1}{0})$$

$$= 60i - 200\frac{1}{0} + 1200\frac{1}{0} + 1200$$

$$V_{B} = V_{B} 2_{B} = 573.6(-0.4229i + 0.8971j - 0.12816k)$$

SUBSTITUTING INTO (1) AND USING DETERMINANTS:

$$\frac{m}{4} \begin{vmatrix} \frac{1}{60} & \frac{1}{200} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{200} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{600} & \frac{1}{600} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \end{vmatrix} + \frac{1}{60} \begin{vmatrix} \frac{1}{60} & \frac{1}{60} & \frac{1}{60} \\ \frac$$

EQUATING THE COEFFICIENTS OF L AND KS

$$\frac{1}{4}(8000) - 48007 = -16908 \qquad \Delta Z = 39.39 \text{ mm}$$

$$\frac{1}{4}(-60000) + 480007 = -43006 \qquad \Delta Z = -58.35 \text{ mn}$$

$$\frac{1}{4}(-60000) + 4800\chi = -43006$$
 $\Delta x = -58.35mn$

280 = 50+05=140+39.4

28, = 179,4 mm



GIVEN: 5-16 SPHERE A AND 2-16 SPHERE B CONNECTED BY RIGID ROD REST ON HORIZONTAL, FRICTION LESS SUPTRICE WHEN A IS GIVEN VELOCITY 1 = (10.5 ft/s) i.

FIND ! (a) LINEAR MOM. AND ANG. HCM. HE (b) VA AND VE AFTER 180 KOTATION.

FOSITION OF MASS CENTER AG+ 8G=711. AG (516) = BG(216), BG = 2.5 AG, 3.5AG=7, AG=2in.

(a) LINEAR AND A NG. MOMENTUM.

L= mA V = 516 32,2 +1/52 (10,5 +1/5) = (1.63 04 16.5)i L=(1.630 16.5)

$$\frac{H}{G} = \overrightarrow{GA} \times m_A \underbrace{v}_{8} = (2 \text{ in.}) j \times (1.6304 \text{ lb.s.}) \underline{i}$$

$$= -(3.2669 \text{ in. lb.s.}) \underline{k}$$

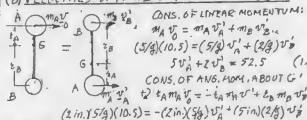
$$= -(0.27174 \text{ ft.lb.s.}) \underline{k}$$

$$\underbrace{H}_{6} = -(0.272 \text{ ft.lb.s.}) \underline{k}$$

(CONTINUED)

14.47 continued

(b) VELOCITIES OF A AND BAFTER 180 ROTATION



MULTIPLY BY & AND DIVIDE BY 2: -50 +50B = 52.5 ADD (1) AND (2): 70 = 105 VB=+15,00 ft/s.

FROM (1): 5VA+2(15)=52.5 VA = + 4.50 ft/s υ' = (4.50 ft/s) ! ; υ' = (15.00 ft/s) !

14:48 G

GIVEN: 5-16 SPHERE A AND 2-16 SPHERE B CONNECTED BY RIGID ROD REST ON HORIZONTAL, FRICTIONLES SURFACE WHEN B IS GIVEN VELOCITY U = (10.5 H/s) L (a) LINEAR MOM, AND AND, MOM.H. (D) VAMB V AFTER 180° ROTATION .

POSITION OF MASS CENTER AG + BG = 7 in. AG(516)=3G(26) BG=2.5AG, 3.5AG=7, AG=2 in

(a) LINEAR AND ANG. MOMENTUM

$$L = \mathcal{L}_{BV} = \frac{2/b}{32.0 \text{ fH/s}^2} (10.5 \text{ fH/s})^2 = (0.6522 \text{ fb/s})^2$$

$$L = (0.652 \text{ fb/s})^2$$

$$\frac{H_{G} = GB \times m_{B} v_{G} = -(5i\vec{n}.)\dot{j} \times (0.6522 \text{ lb.s})\dot{k}}{= +(3.261 \text{ in.lb.s})k} = +(0.272 \text{ ft.lbs})k$$

(6) VELOCITIES OF A AND B AFTER 180° ROTATION.

CONS. OF LINEAR MOHENTUM 4 mov = m v +mov B (2/g)(10.5)=(5/g) VA + (2/g) OB T 6 t8 50 +20 = 21 CONS. OF ANG. MOM. ABOUT G: MATA +) EmBO = 2 MAVA - LEMBUS

(5 in.)(2/g)(10.5) = (2 in.)(5/g) VA - (5 in.)(2/g) V3

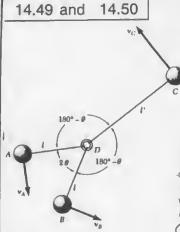
HULTIRY BY & AND DIVIDE BY 2!

$$5v_A^2 - 5v_B^2 = 52.5$$
 (2)

SUBTRACT (2) FROM (1): V= = - 4.50 ft/e 70 = -31,5

Va = + 6.00 ft/s FROH(1): 5VA+2(-4.50)=21

1 = (6.00 ft/s) 2; 1 = - (450ft/s) 2



CONNECTED TO RING D AT THEIR MASS CENTER

SLIDE ON HORIZONTAL,
PRICTIONLESS SURFACE

(1°= 2 (cos 0).

V_A = N_B = V_O WHEN
CORD CD BREAKS.
FIND AFTER CORDS

AD AND BD BECOME TAUT

(2) SPEED OF RING D

(b) RPIATIVE SPEED AT

THREE IDENTICAL SPHERES

(b) RELATIVE SPEED AT WHICH A AND B ROTATE ABOUT D

(C) PERCENT ENERGY OF

DRIGINAL SYSTEM LOST WHEN AD AND BD BECOME TAUT.

PROB. 14.49: ASSUME 0 = 30°.

PROB. 14.50; ASSUME B = 45°.

WE CONSIDER THE FOLLOWING TWO POSTIONS OF THE SPHERES A AND B AND THE RING D. POSITION 1: IMMEDIATELY AFTER CORD CD BREAKS

A Republication of the state of

LINEAR MOMENTUM:

L=(2m vo coso) i (1)

ANGULAR MOMENTUM ABOVI G:

HG=2((sin p)(m vo sinb));

HG = (2 Pm v sint 8) (2)

POSITION 2: AFTER COUDS AD AND BD BECOME TAUT!

(a) SPEED OF HASS CENTER (NOW LOCATED AT D).

RECALLING (1):

 $L = (2m)\overline{v} - (2mv_i\cos\theta)i \qquad \overline{v} = (5\cos\theta)i$

 $v_D = \bar{v} = v_0 \cos \theta \qquad (3)$

(b) RELATIVE SPEED V'AT WHICH A AND B ROTATE

ABOUT D AMIN'S

6=D

2

ANG. HOMENTUH ABOUTG: Hc=(2mv))

RECALLING (2):

 $2mv'l = 2lm v_0 sin'0$ $v' = v_0 sin'0$ (4)

C) ENERGY LOST:

CONSIDERING SYSTEM OF 3 SPHERES:

INITIALLY, VC = (e'/e) VA = (2008) VG. THEREFORE

10 = 2 m VA+ 2 m VB2+ 2 m Vc = m Vo2 (1+2cus20)

 $T_S = \frac{1}{2} (2m) V_D^2 + 2 (\frac{1}{2} m v)^2) + \frac{1}{2} m V_C^2$

= m [vo cos 0 + m (vo sin 0) 2 + 2 vo cos 0] = m vo 2 (3 cos 0 + sin 4)

 $\frac{7005}{700} = \frac{100}{100} \frac{1 + 20050 - 30050 - 30050}{1 + 20050} = 100 \frac{\sin \theta \cos \theta}{1 + 20050}$

PROB 14.49: MAKING 8=30" IN ENS. (3), (4), AND (5)

(a) 0.866 vo. (b) 0.250 vo. (c) 7.50 %

PROB. 14.50: MAKING 0 = 45° IN EQS. (3),(4), AND(3):

(a) 0.707 vo. (b) 0.500 vo. (c) 12.50%



C Vo VA B

GIVEN: TWO SHALL IDENTICAL SPHERES A AND B. CONNECTED BY A CURD SLIDE ON A HORIZONTAL FRICTION.
LESS SURFACE. INITIALLY THEY ROTATE WITH D-Brod/S ABOUT G, AND G HAS VELOCITY U = Voi.
AFTER CORD BREAKS, SPHERES MOVE ALONG PATHS WITH KA=2, KB=1, AND d=625 mm.
FIND:

(a) SPEEDS No, JA, AND NB, (b) length 20 of cord

CONSERVATION OF LINEAR MOMENTUM
BEFORE BREAK: L = (2m) \$\frac{7}{2}\$ L = 2m \nabla_0\$

SETTING L = LO AND EQUATING COEFF. OF UNIT VECTORS:

SUBTRACTING (2) FROM (1): \$\frac{1}{\sqrt{5}} \varV_A = 2\varV_0 \quad \text{3} \quad \text{7} \quad \quad \text{7} \quad \text{7} \quad \text{7} \quad \text{7} \quad \quad \text{7} \quad \text{7} \quad \text{7} \quad \quad \quad \text{7} \quad \

SUBSTITUTING FOR MAINTO (2): 5 = 2 VE (2 V5 V6) = 4 VE V6 (4)
CONSERVATION OF ANGULAR HOMENTUM

BEFORE BREAK: (HG) = 2 mc = 2 mc (8 mu/s) = 16 mc

ATTER BREAK: HG=H = m(V5) d=m = (4 V2 V6) (0.625 m) - 25 m V.

SETTING HG = (HG) = 2.5 m V0 = 16 m c2 V0 = 6.40 C (S)

CONSERVATION OF ENERGY

BEFORE BREAK! To = 1/2(2m) Vo+ 1/2(2m)(ch) = m (V+ ch)

LETTING $\theta = 8 \text{ rod/s}$ AND ASING(5): $T_0 = m(40.96c^4 + 64c^2)$ AFTER BREAK: $T = \frac{1}{L} m V_B^2 + \frac{1}{2} m V_B^2$

RECALLING (3), (4), AND (5): $T = \frac{1}{2} m (20 v_0^2 + 32 v_0^2) = 26 m v_0^2 = 1064.96 m c^4$

SETTING T = To: 1064, 9664 = 4096 c+ 646

10240 = 64 C=-0,0625 C=0250M

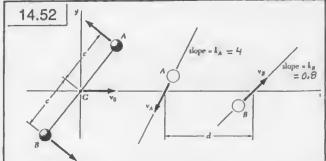
FROM (3): Vo = 6.40(0.062=) = 0.400 m/3 FROM (3): VA = 2 V5(0.4) = 1.789 m/s

FRUM (4): 25 = 4 12 (0.4) = 2.26 =/s

ANSWERS:

(a) Vo = 0.400 m/s; VA = 1.789 m/s; VB = 2.26 m/s

(b) LENGTH OF CORD = 2C = 500 mm



GIVEN: TWO SMALL IDENTICAL SPHERES A AND B. CONNECTED BY A CORD OF LENGTH 2C = 600mm

SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE.
INITIALLY THEY ROTATE WITH &= 12 rod/s ABOUT

G, AND G MOVES WITH U = U L. APTER COLD

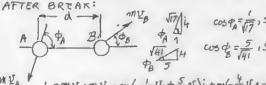
BREAKS, SPHERES MOVE ALONG PATHS WITH K_B = 4

AND K_B = 0.8.

FIND: SPEEDS NO, VA, AND VB, (b) DISTANCE d

CONSTRUATION OF LINEAR HOMENTUM

BEFORE BREAK: Lo=(2m) Vo Lo=2m V_L



THE L= $mV_A + mV_B = m\left(-\frac{1}{17}U_A + \frac{5}{\sqrt{41}}V_B\right)\dot{v} + m\left(-\frac{7}{17}V_A + \frac{7}{\sqrt{41}}V_B\right)\dot{g}$ SETTING L= L_O AND EQUATING COEFF OF UNIT VECTORS:

 $(\underline{\hat{U}}) - \frac{1}{\sqrt{17}} \underline{V}_A + \frac{5}{\sqrt{14}} \underline{V}_B = 2 \underline{V}_0$ (1)

SUBSTITUTE FOR US INTO (1):

 $-\frac{1}{\sqrt{17}}v_A + \frac{5}{\sqrt{17}}v_A = 2v_0 \qquad v_A = \frac{117}{2}v_0$ (3)

 $V_{B} = \frac{\sqrt{17}}{\sqrt{17}} \frac{\sqrt{17}}{z} \frac{\sqrt{$

CONSERVATION OF ANGULAR MOMENTUM

BEFORE BREAK: $(H_G)_0 = 2mC^6\dot{\theta} = 2m(0.3 m)^2(12 rad/s) = m(2.16)$ AFTER BREAK: $H_G = H_A = m(V_B)_g d = md\frac{H}{V_H}\frac{1}{2}V_0 = 2mdV_0$ SETTING $H_G = (H_G)_0$: $2mdV_G = m(2.16)$ $V_G d = 1.08$ (5)

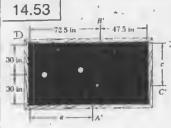
EFORE PREAK: $T_0 = \frac{1}{2}(2m)v_0^2 + \frac{1}{2}(2m)(c\dot{\theta})^2$ = $mv_0^2 + m(o.3 \times 12)^2 = m(v_0^2 + 12.96)^2$

HFTER BREAK: $T = \frac{1}{2}m V_A^2 + \frac{1}{2}m V_B^2$ RECALLING (3) AND (4): $T = \frac{1}{2}m V_0^2 (\frac{17}{4} + \frac{41}{4}) = 7.25 m V_0^2$ SETTING $T = T_0$: $7.25 V_0^2 = V_0^2 + 12.96$ Vo = 1.440 m/s

FROM (3): $V_A = \frac{17}{2}(1.440) = 2.969 \text{ m/s}$ FROM (4): $V_B = \frac{1}{2}(1.440) = 4.610 \text{ m/s}$ FROM (5): $d = \frac{1.09}{1.440} = 0.750 \text{ m}$

ANSWERS: (a) vo= 1.440 m/s; va= 2.97 m/s; vB = 4.61 m/s

(b) DISTANCE d = 0.750 m = 750 mm



GIVEN:

BALL A HITS BALL B

WITH UD = (12 ft/s), THEN

C, THEN SIDE OF TABLE

AT A' (WHERE a = 66 in.)

WITH UA = - (5,76 ft/s) j

FIND:

(a) VELOCITIES OF BAND C

(b) DISTANCE C WHERE

BALL CHITS SIDE

-

1

CONSERVATION OF LINEAR MOMENTUM

MTO i = - MVAj + M(Ve), i + m(Ve), j + mVci

EQUATING COEFF. OF UNIT VECTORS!

(i) $mV_0 = m(v_B)_2 + mV_c$ $(V_B)_2 + V_c = V_0 = 12 \text{ ft/s}$ (1) (i) $0 = -mV_A + (V_B)_3$ $(V_B)_3 = V_A = 5.76 \text{ ft/s}$ (2)

CONSERVATION OF ANG. MUNENTUM A BOUT CURIVER D (301n.) Vo = -(661n.) VA + (72.5in.) (VB), + C VC

 $30(12) = -66(5.76) + (72.5)(5.76) + CV_C$ $CV_C = 322.56$ (3)

CONSERVATION OF ENERGY

\[\frac{1}{2}mV_0^2 = \frac{1}{2}mV_0^2 + \frac{1}{2}m\left[(V_B)_3^2 + \left(V_B)_3^2] + \frac{1}{2}mV_0^2
\]
THURSDAY AV M. WILLTENSING BY 2 AND SUBST

DIVIDING BY M, MULTIPLYING BY 2, AND SUBSTITUTING FOR \$, VA, (VB), THEIR VALUES AND (VE)= 12-VE FROM (1): (12) = (5.76) + (12-VE) + (5.76) + VE

DIVIDING BY 2: VC= 12 VC+ (5.76) = 0, VC = 6-1.68 WITH VC= 6-1.68= 4.32, EQ. (3) YIELDS C= 74.7 (IMPOSSIBLE)

THEREFORE: VC = 6+1.68 = 7.68 V=7.68 ft/s-

TROM (3): C(7.68) = 32256. C=42.0 in.

14.54 (SEE FIGURE OF PROB. 14,53)

GIVEN: BALL A HITS B WITH Y=(15-R/S) &

THEN C; BALL C HITS SIDE AT C=48 In. WITH Y=(9.6+16) L

FIND: (A) YA AND YB, (B) DISTANCE Q.

CONSERVATION OF LINEAR MOMENTUM

mv = mvAj+m(VB), i+m(VB), j+m vc i

(NB) = 15-9.6=5.40 ft/s (1)

(VE) = UA (2)

CONSERVATION OF AWEULAR MOMENTUR ABOUT OUR HER D +) (30in.) Vo=- a VA + (72.5in.) (VB) y + C VC

SUBSTITUTING GIVEN DATA AND USING EQ. (2): 30(1541/4) =- a VA + 72.5 VA + 48/9.6 41/5)

 $(a-72.5)\dot{v}_{A}=10.8$

CONSERVATION OF ENERGY

\[\frac{1}{2}m\mathbf{V}_0^2 = \frac{1}{2}m\mathbf{V}_A^2 + \frac{1}{2}m\mathbf{V}_B^2 + (\mathbf{V}_B)_y^2 \right] + \frac{1}{2}m\mathbf{V}_C^2

DNIDING BY M, MULTIPLYING BY 2, AND SUBSTITUTING I

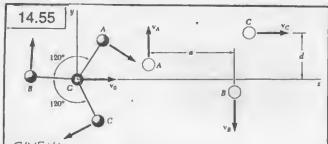
(15)\frac{1}{2} = \mathbf{V}_A^2 + (\frac{1}{2}, \frac{1}{2}) + \frac{1}{2} + (\frac{1}{2}, \frac{1}{2}) + (\frac{1}, \frac{1}{2}) + (\frac{1}, \frac{1}{2}) + (\frac{1}, \frac{1}{

 $V_A^2 = 51.84$ $V_A^2 = 7.20 \text{ fl/s}$ $V_A = 7.20 \text{ fl/s}$

PROH (1) AND (2): YB=(VB)2+(VB)yj=(5.40f4/5)i+(7.20f4/5)j OR VB=-9.00f4/5 Z 53.1°

FROM(3): (a-72,5)7.20 = 10,8

a=72,5+1.5 a=74.0 in.



GIVEN: THREE SMALL IDENTICAL SPHERES CONNECTED BY 200-MM-LONG STRINGS TO RING & SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE.

INITIALLY, SPHERES ROTATE ABOUT & WITH O,8 m/s RELATIVE VELOCITY AND RING MOVES WITH 5= (0.4 m/s) SUDDENLY RING BREAKS AND SPHERES MOVE FREELY HS S'HOWN WITH & = 346 mm.

FIND:

(a) VELOCITY OF EACH 5 PHERE, (b) DUTANCEd.

CONSERVATION OF LINEAR MOMENTUM

BEFORE BREAK: L= (3m) = 3m (0,4i)=m(1.2ms)i APTER BREAK: L= mvaj - mvB ; +mvc = [=]: mvi+m(vA-vB)j=m(1.2m/s)i

THEREFORE: (1)

> 12c= 1,200 m/s → (2) V = 1,200 m/5

CONSERVATION OF ANGULAR MUMENTUM

BEFORE BREAK: +) (Ho) = 3 m (v = 3 m (0,2 m) (0.8 m/s) = 0.480 1

+2 Ho=-m25A2A AFTER BREAK! mvc m UA 1 0.346A + mvA (36+0,346) + mvcd Ho = (Ho) : FROM(1): OB 0.346 m V + m V cd = 0.480 m RECALLING (2): 0,346 VA+ 1,200 d = 0,480 d=0,400-0,28833Va

CONSTRUATION OF ENERGY

BETORE BREAK:

 $T_0 = \frac{1}{2} (3m) \overline{U}^2 + 3 (\frac{1}{2} m U)^2)$ $= \frac{3}{2} m (V_0^2 + V)^2) = \frac{3}{2} [(0.4)^2 + (0.8)^2] m = 1.200 m$

AFTER BREAK:

T= 1 m vA+ 1 m vB2+ 1 m vc

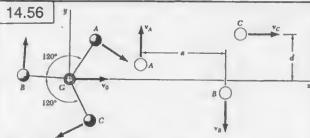
: SUBSTITUTING FOR VB FRUM (1) AND V FROM (2): $\frac{1}{2} \left[U_A^2 + V_A^2 + (1.200)^2 \right] = 1.200$ VA2 = 0.480 $V_A = V_B = 0.69282 \, m/s$

(a) VELOCITIES:

U = 0,693 m/s 1; U = 0.693 m/s 1; U= 1.200 m/s→

(b) DISTANCE d:

PROH (3): d=0.400-0.28833(0.69282) = 0.20024 m d=200 mm



GIVEN:

THREE SMALL IDENTICAL SPHERES CONNECTED BY STRINGS OF LENGTH & TO RING G SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE. INITIALLY, SPHERES KOTATE ABOUT G AND KINI. MOVES ASSHOWN. SUDDENLY RING BREAKS AND SPHERE MOVE PREELY IN ZY PLANE. WE KNOW THAT v = (1.039 m/s) ; v = (1.800 m/s) i, a = 416 mm, d = 240 mm.

(a) VEL. V OF RING. (b) LENGTH & OF STRINGS (C) RATE IN rad/S AT WHICH SPHERES WERE ROTATING

CONSERVATION OF LINEAR MOMENTUM

(3m) v = mvA + mvB + mv

3 m Vo L = m (1,039 m/s) j - n1 VB j + m (1,800 m/s) i

EQUATING COEFF. OF UNIT VECTORS:

(1) 5 Vo = 1.800 m/s (a) v = 0,600 m/s

(1) 0 = 1.039 m/s - VB (1) · V= 1.039 m/s

CONSERVATION OF ANGULAR MOMENTUM BEPORE BREAK: +) (Ho) = 3 m l2 0

AFTER BREAK: +2 H = - n1 VAZA mvc mvA+ 0.416m + 2 VA (ZA + 0, 416) 0,240m. + m V (0,240) = m (1.039 X 0.416) +m (1.800)(0.240) = m(0,864224)

 $(H_0) = H_0: \Im m \ell^2 \dot{\theta} = M(0.864224)$ $\ell^2 \dot{\theta} = 0.28807$ (2)

CONSERVATION OF ENERGY

BEFORE GREAKS

To = {(3m) v+3({mv)}=3mv+3m(le) = 3 m (0,600) + 3 m (202 AFTER BREAK:

T= = mNA+ = more + = m Nc

 $= \frac{1}{2} m \left[(1.039)^2 + (1.039)^2 + (1.800)^2 = \frac{1}{2} m (5.399)^2 \right]$

T=To: 1 m (5,399) = 3 m (0,600) 2+3 m 22 62 l'0 = 1.4397 (3)

DIVIDING (3) BY (2): 0 = 4397 = 4,9976

(b) FROM(2): L2 = 0.28867 l=0.2401 m L= 240 mm

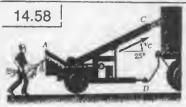
(C) RATE OF RUTATION = 0 = 5,00 rad/s



VEL. OF STREAM = 25 m/s A = 300 mm

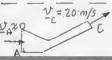
FIND: FORCE EXERTED BY STREAM ON MAILBOX.

NOTE: FORCE P SHOWN ON SKETCH IS FURCE APPLIED BY HAILBOX ON STREAM. FURCE EXERTED BY STREAM ON MAILBOX IS 187.5 N ->



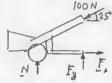
GIVEN:
TREE LIMBS ARE FED
INTO SHREDDER AT
RATE OF 5 kg/s AND
CHIPS HRE SPEWED
WITH VC= 20 m/s.

FIND: HORIZ. COMP. OF FORCE EXERTED ON HITCH AT D.



Eq. (14.3b): (\(\Delta m \) \nu_A + \(\Sigma \) Fot = (\(\Delta m \) \nu_C \(\Sigma \) \nu_A = (\(Skg/s \) (20 \m/s \(\Sigma \) 25°

FORCE EXERTED ON CHIPS = E F = 100 N 425



FREE BODY: SHREDDER +, E F = 0: Fx - (100 N) cos 25°=0

Fx = 90.6 N -

14.59

GIVEN: WATER DISCHARGED AT RATE OF 2000 gal/min WITH YEL. OF 150 tt/s.

FIND: THRUST OF ENGINE TO KEEP BOAT STATIONIARY.

EQ.(14.38): (DM) YA + \(\Sigma\) TE OT = DM) UB

WHERE UA = 0, UB = 150 ft/s \(\mathbb{Z}\) 35°

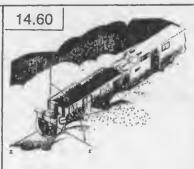
FORCE EXERTED ON STREAM!

\(\Sigma\) = - AM At - (2000 dal) \(\sigma\) \(\sigma\) \(\sigma\) \(\sigma\) \(\sigma\)

 $\sum F = \frac{\Delta m}{\Delta t} \frac{dt}{dt} = (2000 \frac{9al}{min}) \frac{1 \text{ ft}^3}{7.48 \text{ gai}} \frac{1 \text{ min}}{60\text{ s}} (62.4 \frac{16}{5t^3}) \frac{1}{322 \text{ ft}/s} (150 \frac{\text{ft}}{\text{s}})$ $\sum F = 1295.4 \text{ 16 } 2 35^{\circ}$

THRUST OF ENGINE = (ZF) = (1295.4 16) cos 350

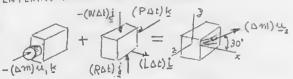
THRUST = 1061 16



GIVEN:
ENGINE PROPELS FLOW
HT SYETD OF 12 mi/h.
PLOW PRUJECTS 180 TONS
OF SHOW FEA A ATTE
WITH VELOCITY OF
40 ft/s W/R TO CAA.
FIND:
(a) FORCE EXERTED

(a) FORCE EXECTED BY ENGINE ON CAK
(b) LATERAL FORCE
EXERTED BY TRACK.

WE MEASURE ALL VELOCITIES WAR PLOW CAL AND APPLY THE IMPULSE-HOMENTUM PRINCIPLE TO THE PLOW CAR, THE SNOW IT CONTAINS, ACID THE SNOW ENTERING IN THE TIPE INTERVAL AT.



-(Om)~, k+(PDt) k+(ROt) j-(WOt) j= (Om)~, k+(PDt) k+(ROt) j-(WOt) j= (Om)~, k+(PDt) k+(ROt) j-(WOt) j=

EQUATING THE COFFE OF THE UNIT VECTORS:

(1)

(1)

 $\frac{u_1 = 12 \text{ mi/h} = 17.60 \text{ ft/s}}{\Delta m} = (180 \text{ fons/min}) \left(\frac{1 \text{ min}}{600}\right) \left(\frac{2000 \text{ lb}}{1 \text{ ton}}\right) \left(\frac{1}{32.2 \text{ ft/s}}\right) = 186.34 \text{ lb·s/ft}$

(a) EQ.(1): $P = (186.34 lb \cdot s/st)(17.60 ft/s)$ P = (328)(b) EQ.(2): $1 = (186.34 lb \cdot s/st)(40 ft/s) cus 30$ L = (6.11)

P=(32801b)k

14.61

GIVEN:

V = 30 m/s

STREAM SEPARATED

INTO TIVO STREAMS

WITH Q = 100 L/min

AND Q = 500 L/min

FIND: (a) 8, (b) TOTAL FORCE EXERTED BY STREAM ON PLATE

WE NOTE THAT $(i = Q_1 + Q_2)$ (1) $(2\pi)^{2} V = (1)$ $P\Delta t = (1)^{2} V = (1)$

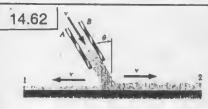
IMPULSE-MOMENTUM PRINCIPLE: $\pm x \cos \theta$: $(\Delta m) v \sin \theta = (\Delta M_L) v - (\Delta m) v$ $(QQ \Delta t) v \sin \theta = (QQ_L \Delta t) v - (QQ_L \Delta t) v$ $Q \sin \theta = Q_L - Q_L$ (2)

+1 y COMP.: $-(\Delta m)$ v cos θ + P Δt = 0 $P = \frac{\Delta m}{\Lambda F}$ v cos θ P= $\theta \Delta v$ cos θ (3)

(a) FROM (1): Q = 100 + 500 = 600 L/min.FROM (2): $\sin \theta = \frac{Q_2 - Q_1}{Q_2} = \frac{500 - 100}{Q_2} = \frac{2}{3}$ $\theta = 41.8^{\circ}$

(b) FROM (3): P=(1kg/L)(600 L/min)(1min)(30 m/s)cos 41.2° P= 124 N1

PORCE EXERTED BY STREAM ON PLATE = 224 N



GIVEN:

U = 40 m/s, 8 = 30

TUTAL PORCE
EXERTED SY STREAM
ON "LATE = 500N L.

FIND: Q, MI/D Q.

OF RESULTING

WE NOTE THAT $Q = Q_1 + Q_2$ (1)

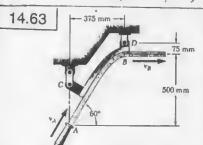
SEE SOLUTION OF 1 FUE, 14,61 FOR DERIVATION OF

Q Sin $\theta = Q_2 - Q_1$ (2) $P = Q \cup COS\theta$ (3)

FROM (3): $Q = \frac{P}{Q \cup COS\theta} = \frac{500 \text{ N}}{(1 \text{ kg/L})(40 \text{ m/s}) \cos 30^\circ} = \frac{14.43 + L/s}{16.43 + L/s}$

= 866.03 L/min ADDING (1) AND(2): Q(1+sin 0) = $2Q_2$ $Q_2 = \frac{1+\sin 0}{2}Q = \frac{1+\sin 30}{2}(866.03 L/min) = 649.57 L/min$

FROM (1): $Q_1 = Q - Q_2 = 866.03 - 649.52 = 216.57 L/min$ $Q_1 = 217 L/min; Q_2 = 650 L/min$



GIVEN!

WATER DISCHARGED

AT RATE Q=1.2 m/nin

WITH VA=VB=25 m/s

FIND:

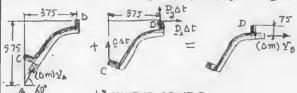
COMPONENTS OF

REACTIONS AT (AND D.

(NEGLECT WEIGHT

OF VANE).

WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE BLADE, WATER IN CONTACT WITH THE FLATE, AND WATER STEINING THE BLADE IN INTERVAL DE.



60 + MOMENTS ABOUT DI

575(Om) $V_A \cos 60^\circ - 375(\Delta m) v_A \sin 60^\circ - 375 C \Delta t = 75(\Delta m) v_B$ 375 $C = \frac{\Delta m}{\Delta t} (25 \text{ m/s})(575 \cos 60^\circ - 375 \sin 60^\circ - 75)$ $C = -7.484 \frac{\Delta m}{\Delta t}$

BUT $\Delta m = (Q = (1000 \text{kg/m}^3)(\frac{12 \text{ m}^3}{605}) = 20 \text{ kg/s}$ THUS: C = -7.484(20) = -149.66 M $C_2 = 0, C_3 = 149.7 \text{ M} + 1000 \text{ M}$

 $\frac{1}{2} 2 \cot_{1}P_{1} : (\Delta m) V_{A} \cos 60^{\circ} + D_{2} \Delta t = (\Delta m) V_{B}$ $D_{2} = \frac{\Delta m}{\Delta t} (25 \text{ m/s}) (1 - \cos 60^{\circ}) = (20 \text{ kg/s}) (25 \text{ m/s}) (1 - \cos 60^{\circ})$ $D_{1} = 250 \text{ N} \rightarrow$

+ 9 y comp.; (Δm) $^{\circ}H$ sinh $^{\circ}H$ C $\Delta L + D_{g} \Delta L = 0$ $D_{g} = -\frac{\Delta m}{\Delta L}(25 \text{ m/s}) \sin 60^{\circ} - (-149.7 \text{ N})$ $= -(20 \text{ kg/s})(25 \text{ m/s}) \sin 60^{\circ} + 149.7 \text{ N}$ = -433.0 N + 149.7 N = -283.3 N $D_{g} = 283 \text{ N} + 149.7 \text{ N} = -283.3 \text{ N}$

14.64 HASUME THAT BLADE ABOF SAMPLE PROBLET.

SHOW THAT RESULTANT FORCE F EXERTED BY THEBLADE ON THE STREAM IS APPLIED AT MIDPOINT C OF ARC AB.

WE APPLY THE IMPULSE-HUMENTUM PRINCIPLE TO THE PORTION OF STREAM IN CONTACT WITH THE BLADE AND ENTERING IN CONTACT IN INTERVAL DE.

 $\frac{1}{A} = \frac{1}{A} = \frac{1}$

WE RECALL THAT UA = UB = M

+) MOMENTS ABOUT O' R(AM) 4 + MOH. OFF = R(AM) 4
THUS: MOH. OF F ABOUT O = 0; LINE OF ACTION OF F PASSE
THROUGH O.

12 COMP: (Dm) u - (Fat) sind = (Dm) u cos o

av.: $F(\Delta t) \sin \alpha = (\Delta m) \alpha (1 - \cos \theta)$ (1) Early: $0 + F(\Delta t) \cos \alpha = (\Delta m) \alpha \sin \theta$ (2)

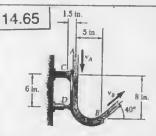
 $t^{\dagger}g corr.: .0 + F(\Delta t) cos \alpha = (\Delta m) u sin t$ (2) DIVIDE (1) BY (2):

 $tan \alpha = \frac{1 - \cos \theta}{\sinh \theta} = \frac{2 \sin \frac{\theta}{2}}{2 \sin \frac{\theta}{2} \cos \frac{\theta}{2}} = \tan \frac{\theta}{2} \qquad \alpha = \frac{\theta}{2}$

THUS: LINE OF ACTION OF E BISSECTS & AOB,

E K APPLIED HT MIDPOINT C OF ARC AB.

(Q.E.D.)



GIVEN:

STREAM OF WATER WITH

G=150 gal/min

VA = VB = 60 ft/s

REACTION AT D HORIZONTAL

COMPONENTS OF REACTIONS AT C AND D (NEGLECT WEIGHT OF VANE.)

WE APPLY THE IMPULSE-HENERTUM PRINCIPLE TO THE VAME, THE WATER IN CONTACT WITH IT, AND THE HISS AN OF WATER ENTERNY AND LEAVING THE SYSTEM IN THE INTERVAL A t. WE NOTE THAT AM = CQAt = \frac{62.4}{32.2 ft/s} \left(\frac{150 gal}{665}\right) \frac{15t^3}{2.483a} \alpha = (0.6977 \frac{165}{32}) \text{At}



+ Hom. about C: - (Δm) va (1.5in.) + D Δt(6in.) = ... ,=(Δm) vB cos 40°(8in.) + (Δm) vg rin 40 (6.5in)

Dot(6in)=(0.6477/6.5) Dt(609/3)(11.806in.) D=76.47/6

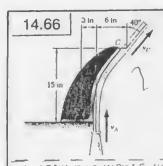
BR COMP.: CLAt+DAt=(Am) \$ cus 40°

C2 = (0.6477 (60 ft/s) cus 40° - 76.47 16 = - 467 1.

+1 y comp; - (Dm) VA + Cy Dt = (Dm) VB sinto

Cy = (0.6477 16.5)(60 ft/s)(sin + 0°+1) = 63.8 6

 $C_x = 46.7 \text{ lb} \longrightarrow$, $C_y = 63.8 \text{ lb}^{1}$ $D_x = 76.5 \text{ lb} \longrightarrow$, $D_y = 0$

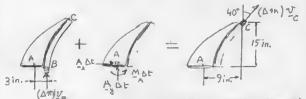


GIVEN: STREAM OF WHITER WITH Q = 200 gal/min AND UR = V = 100 ft/s

FIND:

FORCE-COUPLE SYSTEM APPLIED TO VANE AT A. (NEGLICT WEIGHT OF VANE)

WE APPLY THE IMPIRSE-AND 19:17 11 1 WILLIAM TO " HE VATO THE WATER INCONTA . INTO I AND THE HASS AM OF WATER BUTCH 1 - - . . PRINTER THE SYSTEM IN Dt. WE HOTE THAT 62.416/-F (20080) 7.40g at DE = (0.863616.5) DE DM1=00000=



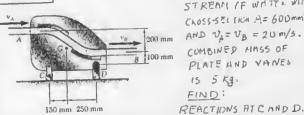
= X CCAP: A DE = (Om) VC sin 40°= (0.8636 Dt)(100 ft/s) sin 40° A=55.5/b-A = 55.5116

+ 9 COMP. : (AM) UB + Ay At = (AM) VE COS 40"

Az = (0.8636)(100 ft/s)(cos40-1) = - 80.2 lb. Az=20.2 lb.

+ 3 MOMENTS AFOUT A I (Dm) 1 (3 in.) + MADE=-(Dm) 2 sin40 (15 in.) + (Dm) v cos40 (9 in.) MA = (0.863616.5/ft)(1009th)[-(15in)sinyo+(9in)cos40-3in.] M = 49616.in.) = -496.3 lb.in.

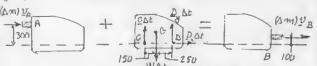
14.67



GIVEN:

STREAM IF WHITE WITH CHOSS-SEL [KIN H= 600mm2 AND VA = VB = 20 m/5. COMBINED MINSS OF PLATE AND VANES 15 5 59. FIND:

WE APPLY IMPULSE-MUMENTUM PRINCIPLE TO PLATE, VANES MATER IN CONTACT WITH PLATE, AND MASS DON OF WATER ENTERING AND LEAVING SYSTEM IN INTERVAL DE.

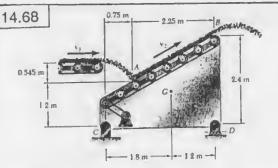


1 = (Q D = (A I) D = (1000 kg/m3 X600x10 m3 X20 m) DE = (12 kg/s) DE =2 COMP: (AM) VA + D2 At = (AM) VB, Dx = (1,200)(20-20) = 0 (100) MOM. ABOUT D: (An) of (300) + CAt (400) - WAT (250) = (An) of (100) 1100 C = (12 kg/s)(20 m/s)(100-300)+ (5 x 9.81 N)(250) = -35,738 C = 89.3 N& C = -89.344 N

Py COMP: (-89,344 - 5 x 9,81 + Dy) WE = 0

 $D_0 = +138.37$ RECALLING THAT DA =0.

D = 138.4 N

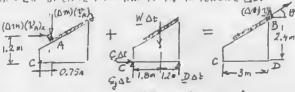


GIVEN: COAL DISCHARGED PROM FIRST TO SECOND CONVEYOR BELT AT RATE OF 120 kg/s WITH U,= 3 m/s AND V2 = 4.25 m/s. MASS OF SECOND BELT ASSEMBLY AND COAL IT SUPPORTS IS 472 kg. FIND: COMPONENTS OF REACTIONS AT CAND D.

MASS OF COAL ENTERING AND LEAVING SYSTEM IN AT: DM = (120 kg/s) DE VELOCITY IN WITH WHICH COAL 1,=3 m/s HITS SECOND BELT: (2)

(14) = 1 = 3 m/s -= (1/A)2 (VA) = 129h = 12 (9.81) (0.545) (II) = 3.27 m/s d (3)

WE APPLY THE IMPULSE-MONEINTUM PRINCIPLE TO THE SECOND BELT ASSENBLY, THE COAL IT SUPPORTS, A113 THE MRCS ENT OF LIME IN TIME IT IN INTERVAL OCT



WE NOTE THAT can 0 = 1.25m

t) 11011. 14 BOUT C: $(\Delta \pi i)(V_{h}^{*})_{2}(1.2m) + (\Delta \pi i)(\tau_{h}^{*})_{4}(0.75m) + (WDt)(1.8m) - (DDt)(3m)$

=(Dn)(vacus 8/2,4m)-(0 m/vasin)(3m) D(3 m) = 472 kg)(9.81 m/3)(1.8 m) + (120 kg/s) [(3 m/s)(1.2 m) + (3, 27 m/s)(0.75 m)] -(120 kg/s)(4.25mg) [2.4m) cos 28.07-(3 m)sin 28.07")]

D(3m) = 8334.6 N.m+726.30 N.m-360,09 N.m = 8700 N.m D=0, D= 2900 N

1,2 CUTIP: (DM)(VA), + C, Dt = (DM) VB CUS D

C = (120 kg/s)(4,25 m/s) cos 28.07 - (120 kg/s)(3 m/s) = 90.0 N C=90.0N->

+94 COMP: - (OM)(Va)y + Cy St+DOL-WA = = (Om) Va sint Cy=-2900 N+ (472kg X4.81 m/s)+(120 13/6)(3.27+ 4.25 sin 28.07) m/s C = 2360N1



WHEN BELT IS AT PEST:) EH, = 01 D(3m) - W(1.8m) = 0 3D-(472×9.81)(1.8)=0 D = 2778 N D = 2730 N) C= 1852 N 1 C = 471 x 9,81 - 2778

GIVEN!

PLANE CRUISES AT 900 km/1. SCOOPS AIR AT RATE OF 90 kg/s AND DISCHARGES IT AT 660 M/S RELATIVE TO PLANE,

FIND; TOTAL DRAG DUE TO AIR FRICTION

WE APRY ZQ. (14.39); ZF = dm (v - vA)

WITH RESPECT TO PLANE. WE HAVE: ZF = D = TOTAL DRAG,

V = 660 m/s, V = 900 km/h = 900 1000 m = 250 m/s

EQ. (14.39): D = (90kg/s) (660 m/s - 250 m/s)

D= 36.9 KN

GIVEN: 14.70

PLANE IN LEVEL PLIGHT AT 570mi/h. DRAG DUE TO AIR FRICTION = 7500 16 EXHAUST VEL. = 1800 ft/s RELATIVE TO PLANE

FIND: RATE IN 16/5 AT WHICH AIR PASSES THRU ENGINE

WE APPLY EU. (14.39): EF = din (v - V) WITH RESPECT TO PLANE.

WE HAVE EF = DRAG = 7500 16

U = 1800 ft/s, U = 570 mi/h = 636 ft/s

EU. (14.39): 7500/b = dm (1800 ft/s - 836 ft/s)

do = (7.780 · (b.s) (32.2 ft/s) = 251 16/s

14.71



ENGINE SCOUPS IN AIR AT A AT RATE OF 200 Ib/s AND DISCHARGES IT AT B AT 2000 Ft/S W/R PLANE FIND:

THRUST OF ENGINE

WHEN AIRPLINE SPEED IS (a) 300 mi/h, (b) 600 mi/h.

WE APPLY IMP. - MOM. PRINCIPLE USING VELOC. W/R PLANE



\$2 COMP; (DM) WA + FDt = (DM) WB

F = \frac{\Delta m}{\Delta t} (u_B - u_A) = \frac{200 lb/s}{32.2 ft/s} (2000 ft/s - v) (1)

+> MOM, ABOUT B: - (DM) 4 (12 ft) + (FDT) d = 0

Fd = Am (125t) up = 200 11/2 (125t) V (z)

(a) v = 300 mi/h = 440 ft/s

 $EQ(1): F = \frac{200}{32.2} (2000 - 440) = 9,689 B$

Pa.(2): $Fd = \frac{200}{37.2}(12)(440) = 32,795 lb.st$

DIVIDE (2) BY (1): d = 3.38ft

MNSWER: 9690 16, 3,38 ft BELOW B

(b) v = 600 mi/h = 880 ft/s

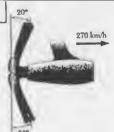
 $FQ.(1): F = \frac{200}{32.2}(2000 - 880) = 6,956 / 6$

EQ.(2): Fd = 200 (12)(880) = 65,590 lb. ft

DIVIDE (2) BY (1): d = 9.43 ft

ANSWER: 696016, 9.43 ft BELOW B

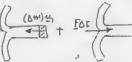
14.72



GIVEN:

IN REVERSE THRUCT. ENGINE SCOOPS AIR AT RATE OF 120 kg/s AND DISCHARGES IT AS SHOWN WITH VELOCITY OF 600 m/s RELATIVE TO ENGINE. FIND! REVERSE THRIST WHEN PLANESPEED IS 270km/h.

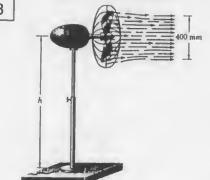
WE APPLY IMPULSE-FIGHENTIN PRINCIPLE WIR PLANE



4: 0: 270 km/h =75 m/s .41 = 600 mb

(F IS OPPOSITE TO REVERSE THRUST OF ENGINE) + x conp: - (Dm) u, + Fat = 2 [1/2 (Dm) u, Sin 20"] F= Om (11, + 11, sin 20) = (120 kg/s)(75 + 600 sin 20) m/s F= 33,6 KN

14.73



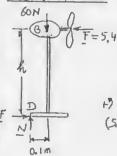
GIVEN: FLUOR FAN DELIVERS AIR WITH SPEED OF 6 m/s. IT IS SUPPORTED BY A 200-MIN - DIAMETER CIRCULAR BASE AND ITS TOTAL WEIGHT IS 60 N.

FIND: MIX. HEIGHT & IF PAR IS NOT TO TIP OVER. (USE P = 1.21 kg/m3 FOR HIR AND ASSUME UA = U.)

THRUST:

FROM EQ. (14.31): F = dm (VB-VA) = (Q(V-U) = (1.31) $F = (1.21 \text{ kg/m}^3) \frac{\pi}{4} (0.400 \text{ m})^2 (6 \text{ m/s})^2 = 5.474 \text{ N}$

FREE BODY: FAN



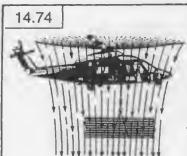
FORCE EXERTED ON : . / BY HIR STREAM IS EQUAL AND OPPOSITE TO THRUST.

WHEN PAN IS ABOUT TO TIP OVER, NORMAL FORCE N IS APPLIED AT D.

17 IMD = 0:

(5474 N) & - (60 N)(0,1 m)=0

h= 1.096 m



MAX, DOWNWARD AIR SPEED PRODUCED BY HELICOPTER 15 BC ft/s. WEIGHT OF HELICOPTER AND CREW 15 3500 1b.

FIND! MAX. LOAD THAT HELICOPTER CAN LIFT WHILE HOVERING, (ASSUME 8 = 0.076 1b/ft3 FOR AIR.)

WE USE EQ (14,39) TO DETERMINE THE THINKT F: F = din (VB-VA) = QQ(V-0) = QAJ2 = & AJ2 F = 0.076 16/52 T (30 10)2 (80+ t/s)2 = 10,678 16

THE LIFT PROVIDED BY THE BLADE IS EQUALAND CIPPOSITE, THAT IS TO, 670 16 T. WE WRITE

+1 2Fy = 0: 10,678 16 - W - 3500 16 = 0 W= 7/7816

W=718016

14.75



GIVEN:

AIRLINER CRUISES AT 600 mill WITH EACH OF ITS THREE ENGINES DISCHARGING AIR IT 2000 11/5 RELATIVE TO PLANE.

SPEED OF PLANE AFTER IT HAS LUST THE USE OF (a) ONE ENGINE (b) TWO ENGINES

(ASSUME THAT DRAG IS PROPRIEDURAL TO UZ.)

WE USE EQ (14.39) TO DETERMINE THE TOTALTHRUST OF THE ENGINES

 $F = \frac{dm}{dt}(v_B - v_A)$ WHERE $U_B = 2000 \text{ ft/s}$ V = SPEED OF PLANE

74US: $F = \frac{dm}{dr}(2000 - v)$ THE DRAG IS DEAU

EQUATING THRUST AND DRAG: dm (2000- v) = kv

WITH THERE ENGINES V = 600 mi/h = 880 H/s

SUBSTITUTING IN EQ. (1):

 $\left(\frac{dm}{dt}\right)_{3}(2000-880) = 4(880)^{2}$ (dm) = 691.73 4

(a) WITH TWO ENGINES:

(dm) = = (691,43 k) = 460,95 R

SVESTITUTING III EU. (1):

460.95 4 (2000-ひ)=ねか

12+460.95 V-921.9 × 103=0

1) = -460.95+V(460.95) + 4 (921.9 × 103) = 756.96 14/3 v = 516 mi/h

(b) WITH ONE ENGINE:

 $\left(\frac{dm}{dt}\right)_1 = \frac{1}{3}\left(\frac{dm}{dt}\right)_3 = \frac{1}{3}\left(691.43 \text{ k}\right) = 230.48 \text{ k}$

SUBSTITUTING IN EQ. (1):

230,484 (2000-0)= 4000

13+.230.48 V - 460.95 × 103 =0 15- - 230.48+ ((230.48) + 4 (460.95 × 103) = 573.41 ft/s

15=391 mi/h 41

14.76



GIVEN:

16-Mg PLANE MAINTAINS U = 774 km/h WITH X = 18°. IT SCOUPS AIR AT RATE OF 300 kg/s AND DISCHARGES IT AT 665 m/s RELATIVE TO PLANE

FINDI (a) INITIAL ACCELERATION IF PILOT CHANGES TO HORIZONTAL FLIGHT WITH SAME ENGINE SETTING (b) MAX. HORIZONTAL SPEED THAT WILL BE ATTAINED. (ASSUME THAT DRAG IS PROPORTIONAL TO V.)

DETERMINATION OF THRUST

SINCE AIRPLANE IS ACCELERATED IN HURIZUNTAL FUGHT WE USE A REPERENCE FRAME AT REST WITH RESPECT TO THE ATMOSPHERE WHEN USING EQ. (14.39) TO DETERMINE THE THRUST F (CF. FOUTHOTE, PAGE 860). F = dm (VB - VA)

WHERE VA = 0, VB = VDISCH! VPLANE = 665 m/s - 774 km/h (1000m) = 665 m/s - 215 m/s = 450 m/s

F=(300kg/s)(450 m/s-0) = 135,0 kN AIRPLANE CLIMBING (NO ALLELERATION)



EFA 18 = 0 1350KN-D-Wsin 18°=U

D = 135,0 kN - (16Mg)(9,31 0/3) sin 19 = 135.0 KN - 48.50 KN = 86.50 KN

WAT START OF HORIZONTAL FLIGHT THRUST AND DRAG ARE STILL THE SAME

IF = ma F-D=ma(135.0-86.5) ×103N=(16×10 kg)a a = 3.03 m/5"

(6) AT MAX, SPEED IN HURICONTAL FLIGHT

WE HAVE a=0 (1) $F_m - D_m = 0$

Fm = dm (41-Vn) = (300 kg/s) (665 m/s - Vm) (2)

ON THE OTHER HAND

$$D_m = k \ V_m^2 \tag{3}$$

BUT, INITIA LLY, WE HAD D= 86.50 KN AND V=774 km/h = 215 m/s AND, THEREFORE

D = 8 152 86.50×10°N = \$ (2/5m/s)2

DIVIDING (3) AND (4) MEMBER BY MEMBER:

86.50×103 = (215)" Dm = 1.8713 Vm

SUBSTITUTING FOR FM FROM (2) AND FUR D FROM (5) INTO (1):

300 (665-Vm) - 1.8713 Vm = 0 Vm + 160.32 Vm - 106.61 x18=0 $V_{m1} = -160.32 + \sqrt{(160.32)^2 + 4(106.61 \times 10^3)} = 256.05 \text{ m/s}$ = (256.05 m/s) 3600 s -= · 921.78 k=/h

V = 922 km/h

GIVEN:

WIND TURBINE-GENERATOR'S OUTPUT-POWER RATING IS 5KW FOR 30 Km/h WIND SPEED.

FIND FOR THAT WIND SPEED (a) KINETIC ENERGY OF AIR PARTICLES ENTERING CIRCLE PER SECOND.

(b) EFFICIENCY OF THIS
ENERGY-CONVERSION SYSTEM.
(ASSUME C=1.21 bg/m³ FOR AIR.)

(a) KINETIC ENERGY PER SECOND
$$= \frac{1}{2} \frac{\Delta m}{\Delta t} v^{2} = \frac{1}{2} (Qv^{2} = \frac{1}{2} (Av) v^{2} = \frac{1}{2} (Av)^{3}$$

$$= \frac{1}{2} (1.21 \frac{1}{6} / m^{3}) \frac{17}{4} (7.50 m)^{2} (\frac{30 \times 10^{3} m}{3.6 \times 10^{3} s})^{3} = 15.47 kJ/s$$
(b) EFFICIENCY
$$= \frac{5 kW}{15.47 kJ/s} = 0.323$$

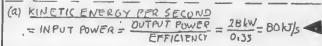


GIVEN:

WIND TURBINE GENERATOR PRODUCES 28 KW OF ELECTRIC POWER WITH AN EFFICIENCY OF 0.35 AS AN ENERGY -CONVERSION SYSTEM

FIND:
(a) KINETIC ENERGY OF AIR
PARTICLES ENTERING CIRCLE
PER SECOND
(b) WIND SPEED

(ASSUME P=1.21 kg/m pur AIR.)



(6) WIND SPEED

K.E. PER SECON = $\frac{1}{2} \frac{\Delta m}{\Delta t} v^2 = \frac{1}{2} (0) v^2 = \frac{1}{2} (Av) v^2 = \frac{1}{2} (A$

14.79 | GIVEN:

PLANE CRUISING IN LEVEL FLIGHT AT GOOMING SCOOPS IN AIR AT RATE OF 200 16/s AND DISCHARGES IT AT 2200 AT/S RELATIVE TO PLANE.

FIND: (A) POWER USED TO PROPEL PLANE,

(b) TOTAL ENGINE POWER (C) EFFICIENCY OF PLANE

(a) PROM EQ.(14.39): THRUST = $F = \frac{dm}{dt}(V_B - V_B)$, WHERE $V_B = 2200 \text{ ff/s}$. $V_A = 600 \text{ mi/h} = 880 \text{ ft/s}$ $F = \frac{200 \text{ lb/s}}{32.2 \text{ ft/s}}$, (2200 - 880) ft/s = 8,198,816

TROPULSIVE FOWER = FV = (8,198.816)(880 ft/s) = 7,21,49 × 106 ft/16/5 = 13,120 hp

(b) POWER DOST IN EXHAUST = 1 am Ver = 1 200 (2200 - 880) = 5.4 112 × 106 ft. 16/5 = 9,838 hp

TOTAL POWER = 13,120 hp+9.838 hp= 22,960 hp

(c) EFFICIENCY = 13.120 hp = 0,571

14.80 GIVEN:

PROPELLER OF SMALL PLANE HAS 6-SEDIAMETER SLIPSTREAM AND PRODUCES 800-ID THRUST
WHEN PLANE IS AT REST ON GROUND.

FIND: (a) SPEED OF THE AIR IN THE SLIPSTREAM.

(b) VOLUME OF AIR PASSING THROUGH PROPELLER PERSECUND.

(C) KINETIC ENERGY IMPARTED TO THE MIR PER SECUND.

(ASSUME & = 0.076 ID/SE) FOR AIR.)

(a) SPEED V OF AIR

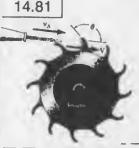
APPLY EQ.(14.39), ASSUMING AIR ENTERS SLIPSTREAM WITH ZERO VELOCITY:

THRUST = $F = \frac{dm}{dt} v = \rho Q v = \frac{1}{8} (Av) v = \frac{7}{8} Av^2$

800 1b = $\frac{0.076 \text{ lb/ft}^3}{32.2 \text{ ft/s}^2} \frac{9}{4} (6 \text{ ft})^3 v^2$ 800 b = $(0.066734 \text{ lb·s/ft}^3) v^2$, $v^2 = 11.988 \text{ ft/s}^2$ v = 109.49 ft/s v = 109.5 ft/s

(b) VOLUME OF AIR PER SECOND Q = AV = # (6ft)*(109.49ft/s) Q = 3100ft/s

(C) KINETIC ENERGY IMPARTED TO AIR PER SECOND $\frac{\Delta T}{\Delta t} = \frac{1}{2} \frac{dm}{dt} v^2 = \frac{1}{2} ((\Delta v)) v^2 = \frac{1}{2} (\frac{1}{3} \Delta v^2) v = \frac{1}{2} F \sigma$ $= \frac{1}{2} (800 \text{ lb}) (109.49 \text{ ft/s}) = 43.800 \text{ ft-lb/s}$



GIVEN: PELTON-WHEEL TURBINE.

RATE AT WHICH WHITER IS DEFLECTED BY BLADES EQUILS RATE
AT WHICH WATER ISSUES FROM

NOZZLE: AM/AL = PAVA.

PIND: (a) VELOCITY V OF

BLADES FOR MAXIMUM POWER.

(b) MAXIMUM POWER.

(C) MECHANICAL EFFICIENCY.
(USE NOTATION OF SP 14.7)

(1)

IMPULSE-MOMENTUM PRINCIPLE

AS IN SAMPLE PROB. 14.7:

\$\frac{1}{2} \comp. (OM) \(\omega - \overline{\text{F}} \) = (OM) \(\cos \overline{\text{C}} \)

BUT NOW \(\Delta m = \text{ACV_A OF} \) \(\text{I-cos } \text{O} \)

THUS: \(F_Z = ACV_A \left(\sum_A - \reft) \left(\frac{1}{2} - \reft) \right) \)

OUTPUT POWER = \(F_Z \right) = ACV_A \left(\sum_A - \right) \left(\frac{1}{2} - \right) \right) \)

CR: OUTPUT POWER = ACV_A \(\sum_A \right) - \sum_A^2 \right) \(\left(\frac{1}{2} - \right) \right) \)

(a) FOR MAX, POWER: d(POWER)/dV = 0: $A P V_A (V_A - 2V)(1-\omega s v) = 0$ $V = \frac{1}{2} V_A$

(b) MAX. POWER: MAKE V= 1 VAIN EQ. (1): MAX. POWER = APVA(VA- 2VA)(1-650) LVA = 14AP(1-650) VA

(C) EFFICIENCY

IN PUT POWER = $\frac{1}{2} \frac{\Delta m}{\Delta t} U_A^2 = \frac{1}{2} (90) U_A^2 = \frac{1}{2} (9AU_A) U_A^2$ $= \frac{1}{2} A 9 U_A^3 \qquad (2)$

DIVIDE (1) BY (2). $\mathcal{D} = \frac{\text{OUTPUT POWER}}{\text{INPUT POWER}} = \frac{\text{ACVA}(V_A - V)(1 - \cos\theta)V}{\frac{1}{2} \text{ACVA}}$ $\mathcal{D} = 2 \frac{V}{V_A} (1 - \frac{V}{V_A}) (1 - \cos\theta)$

NOTE: HAXIMUM EFFICIENCY IS OBTAINED WHEN Y= 1 NA.

 $\mathcal{I}_{\text{MAY}} = 2\left(\frac{1}{2}\right)\left(\frac{1}{2}\right)(2) = 1$



GIVEN:

CIRCULAR REENTRANT

ORIFICE (BORDA'S

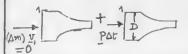
MOUTHPIECE)

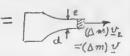
V_= 0, V_2 = V = V2 gh

SHOW THAT:

d = D/V2

WE APPLY IMPULSE-MOMENTUM PRINCIPLE TO SECTION OF WATER INDICATED BY DASHED LINE AND TO MASS OF WATER DM ENTERING AND LEAVING IN QE.





 $\pm x comp.$: $O + PDE = (DM) v = (PQDE) v = (PA_2 vDE) v$ THUS: $P = (PA_2 v^2 = (T d^2 v^2) (I)$

BUT, RECALLING THAT THE PRESSURE AT A DEPTH h 15 12= 84 h. WE HAVE

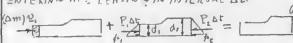
SUBSTITUTING THIS EXPRESSION IN (1) ALL THE GAPRESSION GIVEN FOR V:

(Q.E.D.)



GIVEN:
HYDRAULIC JUMP.
CHANNEL VIIDTH = b.
EXPRESS RATE OF FLOW
QIN TERMS OF b,d,d

WE APPLY THE IMPUISE-MONENTUM PRINCIPLE TO THE WATER SECTION SHOWN AND TO THE MASS OF WATER AND ENTERING ALL LEHING IN INTERVAL DE.



 $\pm x \text{ CUMP: } (\Delta m) v_1 + P_1 \Delta t - P_2 \Delta t = (\Delta m) v_2$ $(? Q \Delta t) v_1 + P_1 \Delta t - P_2 \Delta t = (PQ \Delta t) v_1$ $PQ (v_1 - v_2) = P_2 - P_1$ (1)

BUT
$$Q = A_i v_i = b d_i v_i$$
 $v_i = Q/b d_i$ (2)

AND
$$Q = A_2 v_2 = b d_2 v_2$$
 $v_2 = Q'/b d_2$ (3)

ALSO:
$$P_{i} = \frac{1}{2}p_{i}, A_{i} = \frac{1}{2}(8d_{i})(6d_{i}) = \frac{1}{2}86d_{i}^{2}$$
 (4)

SIMILARLY:
$$P_2 = \frac{1}{2}\delta b d_2 \qquad (5)$$

SUBSTITUTE FRIM (2), (3), (4), (5) INTO (1):
$$CQ\left(\frac{G}{bd_1} - \frac{G}{bd_2}\right) = \frac{1}{2} \delta b \left(d_2^2 - d_1^2\right)$$

$$CQ^2 \frac{d_2 - d_1}{bd_1d_2} = \frac{1}{2} \delta b \left(d_2 + d_1\right) \left(d_2 - d_1\right)$$

DIVIDING THROUGH BY dz-d, AND RECALLING THAT 8=Pg:.

$$\frac{\partial^{2} Q}{\partial d_{1} d_{2}} = \frac{1}{2} \frac{1}{2}$$

* 14.84

GIVEN: FOR CHANNEL OF PRUB. 14.83: b=12 ft, d=4ft, d=5-ft FIND: RATE UF FLOW.

SEE SOLUTION OF PRUB. 14.83 FOR DERIVATION OF

Q = bV = g d de (d+dz) SUBSTITUTING THE GIVEN DATA:

 $Q = (12 \text{ ft}) \sqrt{\frac{1}{2} (32.2 \text{ ft/s})(4 \text{ ft})(5 \text{ ft})(9 \text{ ft})}$ $Q = 646 \text{ ft}^3/5$

14.85



GIVEN:
GRAVEL FALLS ON CONVEYOR BELT WITH NO VELOCITY
AND AT THE CONSTANT RATE q = dm/dt

- (a) FIND MAGNITUDE OF PORCE PREQUIRED TO MAINTAIN A CONSTANT BELT SPEED.
- (b) SHOW THAT K, E. REQUIRED BY GRAVEL IN GIVEN TIME INTERVAL IS HALF THE WORK DUNE BY P. WHAT HAPPENS TO THE OTHER HALFOF WURK OFP?
- (a) WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE GRAVEL ON THE BELT AND TO THE MASS DY OF GRAVEL HITTING AND LEAVING BELT IN INTERVAL DE

TO MY + PAE = MY (AM)Y

 $P = \frac{\Delta m}{\Delta t} v = qv \qquad P = qv$

(b) KINETIC FIVERBY ACQUIRED PER UNIT TIME:

$$\frac{\Delta T}{\Delta t} = \frac{1}{2} \frac{\Delta m}{\Delta t} v^2 = \frac{1}{2} q v^2 \tag{1}$$

WORK DONE PER UNIT TIME:

$$\frac{\Delta U}{\Delta t} = \frac{P\Delta z}{\Delta t} = PU$$

RECALLING THE RESULT OF PART a:

$$\frac{\Delta U}{\Delta t} = (q v) v = q v^{2} \tag{2}$$

COMPARING ERS. (1) AND (2), WE CONCLUDE THAT

$$\frac{\Delta T}{\Delta t} = \frac{1}{2} \frac{\Delta U}{\Delta t} \qquad (Q.E.D.)$$

THE OTHER HALF OF THE WORK OF P IS
DISSIPATED INTO HEAT BY FRICTION AS
THE GRAVEL SLIPS ON THE BELT BEFORE
REACHING THE SPEED V.





GIVEN: CHAIN OF LENGTH & AND MASS on FALLS THROUGH SMALL HOLE IN PLATE. CHAIN IS AT REST WHEN Y IS VERY SMALL.

PIND IN EACH CASE SHOWN:

(A) ACCELERATION OF FIRST LINK A AS FUNCTION OF Y.

(b) VELOCITY OF CHAIN AS LAST LINK PASSES THRU HOLE

IN-CASE 1; ASSUME THAT EACH LINK IS AT REST UNTIL

IT FALLS THRUHOLE

IN CASE 2, ASSUME THAT ALL LINKS HAVE THE SAME SPEED AT ANY GIVEN INSTANT

CASE1. WE APPLY THE IMPULSE-MOVIENTUM PRICEPLE TO THE PORTION OF CHAIN WHICH HAS ALREADY PASSED TROUGH THE HULE AT TIME & AND TO THE PROTON WHICH WILL PASS IN INTERVAL AT.

$$\frac{1}{2} \frac{\delta}{\delta} \frac{\delta}{m^{3}} \frac{\delta}{n} + \frac{\delta}{\delta} \frac{1}{m^{3}} \frac{1}{\delta} \frac{$$

DIVIDE BY ST AND LET ST +0: gy = y dy + v dy at = d(yv)

MULTIPLY BOTH SIDES BY y vdt AND NOTE THAT vdt = dy:

SET YVEL AND INTERNATE!

$$\int_{0}^{3} g y^{2} dy = \int_{0}^{3} v du$$

$$\frac{1}{3} g y^{3} = \frac{1}{2} (3 v)^{2} \qquad v^{3} = \frac{2}{3} g y \qquad (1)$$

(a) DIFFERENTIATE (I) WITH RESPECT TO t:

$$2 \frac{dV}{dt} = \frac{3}{3} g \frac{dy}{dt}$$
 OR $2 Va = \frac{3}{3} g V$ $a = \frac{1}{3} g$

(b) A! LAST LINK PASSES PHROUGH HOLE, 4 = P AND EQ ()
YIELDS V= = QP V= V=V=GP

CASE 2. (a) AT time t, THE PORCE CAUSING THE ACCEL-BRATION OF THE ENTIRE CHAIN IS THE WEIGHT OF THE LENGTH 14 OF CHAIN WHICH HAS PASSED THEOUGH

$$\operatorname{ang}(\frac{1}{2}) = \operatorname{ma} \quad a = \frac{31}{2}$$

(b) SETTING a=v dv, WE HAVE

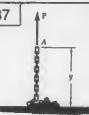
vdr = & gdy

INTEGRATING IN & FROM 0 TO U AND IN J FROM)

UTO 2: $\frac{1}{2}v^2 = \frac{1}{2}\frac{3}{2}\frac{2}{2}$

v=Vge

14.87



GIVEN:

CHAIN OF LENGTH & ANDMASS M 15 LYING INAPILE ON FLOOR. IT IS RAISED AT A CONSTANT V. FIND FOR ANY 4;

FIND FOR ANY Y:

(a) MAGHITUDE OF FURCE P

(b) REACTION OF FLOUR.

(a) WE APPLY THE IMPULSE-HOMENTUM PRINCIPLE TO THE LENGTH & OF CHAIN WHICH IS OFF THE FLOORA AND TO THE LENGTH DY WHICH WILL BE SET IN MOTION DURING THE TIME INTERVAL DE.

$$\frac{1}{2} \int_{\mathbb{R}^{n}} P\Delta t = \int_{\mathbb{R}^{n}} \frac{1}{2} \int_{\mathbb{R}^{n}} \frac{1}{2}$$

+14 COMP: M& U+Pat-mg & at = m. 3+04 v

PAt = m (940E-40+34+004)

DIVIDING BY At:
P=m(gy+vat)

NOTING THAT $\Delta z/\Delta t = v$, $P = \frac{m}{\ell}(q\dot{y} + v^2)$

(b) THE REACTION OF THE FLUOR IS EQUAL TO THE WEIGHT OF CHAIN STILL ON THE FLOOF:

 $R = mg - mg \frac{H}{\ell}$ $R = mg \left(1 - \frac{H}{\ell}\right)$

14.88 AP

GIVEN:

CHAIN OF LENGTH & AND MASS M IS LOWERTD INTO A PILE ON THE FLOOR AT CONSTANT OF FIND FUR ANY &: (a) MAGNITUDE OF FORCE P. (b) REACTION OF THE FLOOR.

(1) P IS EQUAL TO THE WEIGHT OF CHAIN STILL OFF THE FLOOR: P=mgy/l

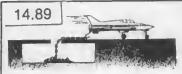
(DWE APPLY THE INIPULSE HOMENTUM PRINCIPLE TO THE LENGTH C-Y OF CHAIN ON THE FLOOR IN DT:

41; COMP: - m & U - = (l-y + ay) Dt + R Dt = 0 SOLVING FOR R;

 $R = \frac{m}{\ell} \left[q \left(\ell - y + \Delta y \right) + U \frac{\Delta y}{\Delta \ell} \right]$

BUT \$ = V AND BY -O WHEN BE -O.

THEREFORE! R = m [g(l-y)+v2]



AS PLANE OF MASSM LANDS WITH NO ON CARRIER, ITS TAIL HOOKS INTO END OF CHAIN OFLENGTH &.

FIND: (a) MASS OF CHAIN REDUIRED TO REDUCE FLANE SPEED TO BY (WHERE B < 1), (b) MAX. FORCE EXERTED BY CHAIN ON PLANE

LET M'= MASS OF CHILIN PER UNITLENGTH X = DISTANCE TRAVELED AT TIMET

CONSERVATION OF LINEAR MOMENTUM (n1+m2)0

to x coma: mv = (m+m'x)v (a) WE WANT V = BUO FOR X = P. SUBSTITUTE: mvn = (m+m, E) B vo m v, (1-B) = m'lB vo

MASS OF CHAIN = m'l = 1-13 m (2)

(b) SOLVE EQ. (1) FOR V: (3)

 $a = \frac{dv}{dt} = -\frac{mv_0}{(m+m'x)^2}m'\frac{dx}{dt} =$

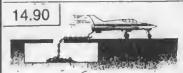
OR, RECALLING (3): $\alpha = -\frac{m^2 m^3 v_0^2}{(m_1 m_1^2 x)^3}$ (4)

DECELERATION IS HAXIMUM FOR 2C=0. WE HAVE $(-a)_{mn} = \frac{m^2 m^3 v_0^2}{m^3} = \frac{m^3 v_0^2}{m^3}$

WRITING | Flow = m | almax AND RECALLING (2):

1F1 max = m' v 2

1F1 may = 1-13 my



GIVEN:

AS 6000- Kg PLANE LANDS AT IBUKM/h ON CARRIER, ITS TAIL HUDKS INTO END OF

BO-M-LONG CHAIN OF MASS OF 50 Kg/m.

FIND: (a) FINI, DECE EVALUAD OF FLANS. (6) I EIGEITY WHEN ENTIRE CHAIN IS PULLED HUT

SEE SULVTION OF PRUB, 14.89 FOR DERIVITION OF EQS. (3) AND (5).

(a) FROM EQ. (5); MAX. DECEL. = $(-a)_{max} = \frac{m!}{m!} \sqrt{s^2} = \frac{50 \text{ kg/m}}{6000 \text{ kg}} \left(\frac{180 \text{ m/s}}{3.6}\right)^3$

MAX. DECEL. = 20.8 m/52 (b) FROM EU. (3), FOR x = l = 80 m: Vous = mvo = (6000 kg)(180 km/h) (80m) Umax = 108.0 Km/h

14.91

GIVEN:

EACH OF THE THREE ENGINES OF SPICE SHUTTLE BURNS PROPELLANT AT RATE OF 340 kg/s AND EJECTS IT WITH A RELATIVE VELOCITY DF 3750 m/s

FIND: TOTAL THRUST PROVIDED BY THE THREE ENGINES

FROM EQ. (14.44) FOR EACH ENGINE P= dm u = (340 kg/s)(3751) m/s) = 1.275×106× FOR THE 3 ENGINES: TOTAL THRUST = 3(1.275 x10 N) = 3.53 MN

14.92

GIVEN:

THE THREE ENGINES OF SPACE SHUTTLE PROVIDE . A TOTAL THRUST OF 6 MN. PROPELLANT IS EJECTED WITH A RELATIVE VEL. OF 3750 m/s.

FIND: RATE AT WHICH PROPELLANT IS BURNED BY EACH OF THE THREE PNGINES.

THRUST OF EACH ENGINE: P= 1 (GMN) = 2 X106 N EQ. (14.44): P= dm u 2 × 106 N = dm (3750 m/s) dm - 2×106 N = 533 kg/s

GIVEN: 14.93

ROCKET FIRED VERTICALLY FROM GROUND WEIGHT OF ROCKET (INCLUDING FUEL) = 2400 16 WEIGHT OF FUEL = 2000 16

FUEL EJECTED AT RATE OF 25 16/5 WITH RELATIVE VELOCITY OF 12,000 fts.

FIND; ACCELERATION OF ROCKET

(a) AS IT IS FIRED.

(6) AS LAST PARTICLE OF FUEL IS BEING CONSUMED

EQ. (14. 44): P = dm u = 2516/5 (12,000 ft/s) = 300 x 103 +15F=ma: $a = \frac{P}{m} - \frac{W}{m} = \frac{(300 \times 10^3)/9}{W/9} - 9$ $a = \frac{300 \times 10^3}{W} - \frac{3}{9}$

(a) AS ROCKET IS FIRED:

 $a = \frac{300 \times 10^{9}}{2400} - 32,2 = 125.0 - 32,2 = 92,8$ W=2400 16 PROM(1): a = 92,8 ft/s2 \$

(b) AS LAST PARTICLE OF FUEL IS BEING CONSUMED: W= 2400-2000 = 40016

FROM(1): $\alpha = \frac{300 \times 10^3}{400} - 32.2 = 750 - 32.2 = 717.8$ a = 718 ft/s 1

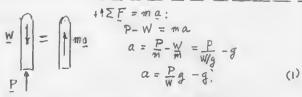
GIVEN:

ROCKET FIRED VERTICALLY PROM GROUND WEIGHT OF ROCKET (INCLUDING PUEL) = 9000 16 WEIGHT OF FUEL = 2500 16.

FUEL CONSUMES AT RATE OF 30 16/s. ACCELERATION IN CREASES BY 750 ft/s" FROM TIME ROCKET IS FIRED TO TIME WHELL LAST PARTICLE OF TUEL IS CONSUMED.

FINDE

RELATIVE VELOCITY WITH WHICH IS EJECTED



AS ROCKET IS FIRED; EQ. (1) YIELD.

$$a_0 = \frac{pg}{3000/b} - g \qquad (2)$$

WHTN LAST PARTICLE IS FIRED !

$$a_0 + 750 ft/s^2 = \frac{p_2}{50016} - g$$
 (3)

SUBTRACT (2) FROM (3): 750 = Pg (1/50) - 1/300)
750 = (1.6667 × 10-3) Pg BUT PROM EQ. (14.44):

450 × 103 = 30 16/5 m nc = 15,000 ft/s

14.95



GIVEN:

FIND:

SATELLITE IS FIRED TO INCREASE ITS VELOCITY BY 8000 ft/5. WEIGHT OF SATELLITE (INCLUDING FUEL) = 10,000 lb. FUEL CJECTED WITH RELATIVE VEL. OF 13,750 H/s

WEIGHT OF FUEL CONSUMED

WE APPLY IMPULSE-MOMENTUM PRINCIPLE TO SATELLITE AND FUEL EXPELLED IN INTERVAL AL

$$\frac{\Delta n(u-v-\Delta v)}{mv} + 0 = \frac{\Delta n(u-v-\Delta v)}{mv}$$

to mv = (m-Am)(v+Av) - Am(u-v-Av) mov = priv - tram + man - uam + van + secont-erdreterns mot = u om

BUT Am = q At AND 111 = 111 - qt THUS (Mo-9t) DV = ug st

 $V = \int_0^t \frac{uq}{m_0 - qt} = -u \left[\ln \left(m_0 - qt \right]_0^t \right]$ $V = u \ln \frac{m_0}{m_0 - qt}$ (1)

(CONTINUED)

14.95 continued

EXPRESSING EU.(1) IN EXPONENTAL FORM:

$$\frac{m_0}{n_0 - q_L} = e^{V/n} \tag{2}$$

SETTING Mo= (10,000 1b)/g, U= 8000 ft/s, 4= 13,150 ft k, AND EXPRESSING Q IN 16/s, WE HAVE

$$\frac{10,000/9}{(10,000-9t)/9} = e^{\frac{0.52100}{13,750}} = e^{0.52102} = 1,7893$$

WEIGHT OF PUEL EXPANDED = 9t = 4410 16

14.96 , GIVEN: CONMUNICATION SATELLITE OF PROB 14.95 FIND: INCREASE IN VELOCITY AFTER 2500 16 HAS BEEN CONSUMED.

SEE SULUTION OF PROB 14,95 FOR DERIVATION OF ED. (1). $v = u \ln \frac{m_0}{m_0 - qt}$

FROM DATA OF PROBS, 14, 95 AND 14.96:

m = 13,750 ft/s, m = (10,000.16)/g, 9t = (2,500 16)/g SUBSTITUTE IN (1):

U= (13,750 ft/s) ln 10,000/g = (13,750 ft/s) ln (1,3333) v=3960 ft/s

14.97

GIVEN:

A 540-Kg SPACECRHET IS MOUNTED ON TOP OF ROCKET OF MASS OF 19 Mg, INCLUDING 17.8 Mg OF FUEL. FUEL IS CONSUMED AT THE RATE OF 225 kg/s AND EJECTED WITH A RELATIVE VELOCITY OF 3600 m/s.

MAXIMUM SPEED OF SPACECRAFT IF ROCKET IS FIRED VERTICALLY FROM THE GROUND.

SEE SAMPLE PROB. 14.8 FOR DERIVATION OF (1) v=uln mo-gt

u = 3600 m/s, q = 225 kg/s, movel = 17 800 kg m = 19 000 kg + 5 40 kg = 19 540 kg

WE HAVE MS (17800 kg = (225 kg/s) E t = 17800 kg = 79.1115

MAX. YEWCITY IS REACHED WHEN ALL FUEL HAS BEEN CONSUME, THAT. IS, WHEN gt = MSud. EO. (1) YIELDS

$$V_{m} = u \ln \frac{m_{0}}{m_{0} - m_{fuel}} - gt$$

$$= (3600 \text{ m/s}) \ln \frac{19540}{19540 - 17600} - (4.81 \text{ m/s}^{2})(79,111 \text{ s})$$

$$= (3600 \text{ m/s}) \ln 11.230 - 776.1 \text{ m/s} = 7930.8 \text{ m/s}$$

$$V_{m} = 7930 \text{ m/s}$$

GIVEN:

A 540-Kg SPACECRAFT IS MOUNTED ON A TWO-STAGE ROCKET: EACH STAGE ITHS A MASS OF 9.5 Mg, INCLUDING B.9 Mg OF FUEL. FUEL IS CONSUMED AT A RATE OF 215 kg/s AND EJECTED WITH A RELATIVE VELOCITY OF 3600 m/s. AS STAGE A EXPELS ITS LAST PARTICLE OF FUEL ITS CASING IS JETTISONED. FIND: (a) SPEED OF RUCKETAT THAT INSTANT,

(b) MAXIMUM SPEED OF SPACECRAFT

SEE SAMPLE PRUB. 14.8 POR DERIVATION OF v=4 ln mo-9t-gt (1)

(a) FIRST STAGE u = 3600 m/s . 9 = 225 kg/s, MASS OF FUEL = 11/4 = 8900 kg mo = 2 (9 500 kg) + 540 kg = 19540 kg IVE HAVE my = 9t, t = ms = 8900th t,= 39.556 s SUBSTITUTE INTO (1): v = (3600 m/s) ln 19540 -8900 -(9.81 m/s) (39.5565) = (3600 m/s) low 1.8365 - 388.04 m/s = 1800.3 m/s N = 1800 m/s

(b) SECOND STAGE 11 = 3600 m/s, 9 = 225 kg/s, HASE OF PUZLE ong = 8900 kg 11, = 9500 kg + 540 kg = 1040 kg t, = 39,556: REPLACING V BY V2-V, AND MOBY M, IN EQ. (1): 12-0, - 46 ln m, -9t2 -9te = (3600 m/c) ln 10040-8900 - (9.81 m/s) (39.5565) = (3600 m/s) En 8.8070 - 388.04 m/s = 7444 m/s 1= 1 +7444 = 1800+7+44 = 9244, V= 9240 m/s

GIVEN: SPAKECRAFT OF PROB. 14,97. 14.99 FIND: ALTITUDE REACHED WHEN ALL THE FUEL OF THE LAUNCHING ROCKET IS CONSUMED.

WE RECALL DATA FROM PRUB. 14.97 AND EQ(1). SETTING v = dy/dt, WE HAVE

 $dy = \left(u \ln \frac{m_0}{m_0 - qt} - gt \right) dt$ fay = 5 (nen = 10 - 9t) de = u 5 ln mo-qt dt - 29t SETTING MO- Qt - WE FIND THAT uc ∫ ln elo dt = u∫(-ln z)(-modz) = u mo [z ln z - z] THUS: h= shay = um (2 ln 2-2-0+1)- 1 gt2 $h = u \left[t + \frac{m_b - qt}{q} \ln \frac{m_b - qt}{m_b} \right] - \frac{1}{2} g t^2$

GIVEN DATA: 41 = 3600 m/s, 9 = 225 kg/s, m = 19540kg $\xi = 79.1115$, $9t = m_3 = 17800 \text{ Kg}$; $m_1 - 9t = 19540 - 17800 = 1740 \text{ Kg}$ $h = 3600 \left[79.111 + \frac{1740}{225} \ln \frac{1740}{19540} \right] - \frac{1}{2} (9.81) (79.111)^3$ = 3600 (79.111 - 18.704) - 30698 = 186770 mh=186.8km [NOTE THAT & WAS ASSUMED CONSTANT]

GIVEN: SPACECRAFF AND TWO-STAGE 14.100 LAUNCHING ROCKET OF PROB. 14.98. FIND ALTITUDE AT WHICH

(1) STAGE A IS RELEASED.

(b) FUEL OF BOTH STAGES HAS BEEN CONSUMED.

SEE SULUTIONS OF SHMPLE PROB. 14:8 AND PROB. 14.99 FOR DERIVATION OF EU. (2): h= u[t+ mo-qt en mo-qt]- ist (2)

(a) FIRST STAGE

PRUM PROB 14.98 WE HAVE u=3600 m/s, 9=225 kg/s, mo=19 540 kg, t,=39,556 s 9t, = mg = 8900 kg, mo - 9t, = 19540 - 8900 = 10640 kg $h_1 = (3600)[39.556 + \frac{10640}{225} lm \frac{10640}{19540}] - \frac{1}{2}(9.81)[39.556]^6$ EQ.(2) YIELDS

= (3600) 39.55+ -28.744) - 76747 = 31248 m h, = 31.2 km

(6) SECOND STAGE USING AGAIN EU. (Z) AND ADDING h, AND UT to IT. hz=h,+v,t,+u[tz+m,-gta en m,-gta]-1gt2

FROM PROB. 14.98, WE HAVE V = 1800.3 m/s, W = 3600 m/s, 9 = 225 kg/s, t = 39,556 s m1 = 10 040 kg, 9t2 = m3 = 8900 kg, m1-95 = 1140 kg

h2 = 31248 + (1800.3)(37,556) + 3000 (37,556 + 1140 64 1140) -- 1 (9.81)(39.556)2.

h2 = 31248+71213+3600 (39.556-11.023)-7675 h= 197,5 km = 197 500 m

GIVEN: 14.101

COMMUNICATION SATELLITE UF PROB. 14.95 FUEL CONSUMED AT RATE OF 37.5 16/s.

FIND: DISTANCE FROM SATELLITE TO SHUTTLE AT t = 603.

SEE SULUTION OF PROB 14.95 FOR DERIVATION OF (1)

v=u ln mo-96 SETTING U=dx/dt, WE HAVE

 $dz = \left(u \ln \frac{m_0}{m_0 - q_E} \right) dt$ (1) X=5 (u la 10-9t) dt = - u 5 ln 10-9t dt SETTING MO-9t = WE HAVE dt = - Mode AND

 $x = \frac{m_0 u}{9} \int_{z}^{z} \ln_2 dz = \frac{m_0 u}{9} \left[z \ln_2 - z \right]_{z}^{z} = \frac{m_0 u}{9} \left(z \ln_2 - z + \right)$ $= \frac{m_0 u}{9} \left(\frac{m_0 - qt}{m_0} \ln_2 \frac{m_0 - qt}{m_0} - 1 + \frac{qt}{m_0} + 1 \right)$ $x = u \left(t + \frac{m_0 - qt}{9} \ln_2 \frac{m_0 - qt}{m_0} \right)$ (2)

GIVEN DATA: 9=(37.516/6)/9, t=605,

AND PROH PRUB. 14.95: n= 13,750 tt/s, m= (10,0001b)/g THUS: Mo-qt = (10,000/g) - (37.5/g)(60) = (7750 16)/g

AFTER SUBSTITUTION, TQ. (2) YIELDG 2=(13,750 ft/s)(60 + 7750 (n 7750) =(13,750)(60-52,678) = 100,680 ft $=(100,680 \text{ ft}) \frac{1 \text{ ni}}{5280 \text{ ft}} = 19.068 \text{ mi}$

X=19,07 mi

GIVEN:

ROCKET OF PROB. 14.93.

PIND: (a) ALTITUDE AT WHICH ALL FUEL IS CONSUMED.

(b) VELOCITY OF ROCKET AT THAT TIME.

SEE SAMPLE PROB. 14.8 FOR DERIVATION OF V=uln mode - gt (1)

AND SOLUTION OF PROB. 14. 99 PUR DERIVATION OF $h = M \left[t + \frac{m_0 - 9t}{9} \ln \frac{m_0 - 9t}{m_0} \right] - \frac{1}{2}gt^2 \qquad (2)$

PROM STATEMENT OF PROB. 14. 93, WE RECALL u = 12,000 H/s, $m_0 = (2400 \text{ Ib})/g$, $q = m_s = (2000 \text{ Ib})/g$ q = (25 Ib/s)/g $t = \frac{m_s}{9} = \frac{2000 \text{ Ib}}{25 \text{ Ib/s}} = 80 \text{ s}$

(A) ALTITUDE AT WHICH ALL FUEL IS CONSUMED SUBSTITUTING DATA IN EQ (2):

 $h = (12,000 + t/s) [80 + \frac{2400 - 2000}{25} e_n \frac{2400 - 2000}{2400}] s - \frac{1}{2} (32.2 + t/s^2) (80 s)^2$

h. = (12,000)(80-28,668)-103,040=512,944 ft . h = $\frac{512,944}{5280}=97.148$ mi h = 97.1 mi

(b) VELOCITY OF RUCKET AT THAT TIME

SUBSTITUTING DATA IN ER. (1):

 $V = (12,000 \text{ ft/s}) \ln \frac{2400}{2400-2000} - (32.2 \text{ ft/s}^{\circ})(805)$ = 12,000 \ln 6 - 2576 = 18,925 ft/s

v = 18, 930 ft/5

14.103

GIVENI JET AIRPLANE WITH

SHOW THAT MECHANICAL EPAKIENCY IS $\eta = \frac{2U}{2}$

EXPLAIN WHY 1 = I WHEN U = U.

THRUST P 13 OBTAINED FROM Eu. (14.39):

ZF = dm (v - v) WHERE VA = V = AIRPLANE SPEED
V=1 = EXHAUST VEL. REL, TO PLANE

THUS: F=dm(u-v)

USEPUL POWER = FV = dm (u-v)v

WASTED POWER = K.E. IMPARTED PER SECOND TO EXHAUS
GASES WHOSE ASSOLUTE VEL. 13 a-v.

= 1 dm (u-v)2

TUTAL POWER = USE FUL POWER + WASTED POWER $= \frac{dm}{dt} \left[(u-v)v + \frac{1}{2} (u-v)^2 \right] = \frac{dm}{dt} \left(uv - v^2 + \frac{1}{2}u^2 + \frac{1}{2}v^2 - uv \right)$ $= \frac{1}{2} \frac{dm}{dt} \left(u^2 - v^2 \right) = \frac{1}{2} \frac{dm}{dt} \left(u + v \right) \left(u - v \right)$

EFFICIENCY = $g = \frac{USEFUL POWER}{TOTAL POWER} = \frac{(u-v)^2v}{\frac{1}{2}(u+v)(u-v)}$

 $\mathcal{D} = \frac{2N}{n+v} \qquad (Q.E.D.)$

WHEN 41 THE ABSOLUTE VELOCITY LI-V OF THE EXPELLED GASES IS ZERO. THUS, INTEREST IS IMPARTED TO THE EXPELLED GASES AND IND POWER IS WHITED.

14.104

GIVEN:

ROCKET WHTH SPEED O, EXPELLING FUEL WITH RELATIVE SPEED &.

SHOW THAT RECOPPLICAL EFFICIENCY IS 3 = 240 (4.0).
EXPLAIN WHY D=1 WHEN 41 = V.

WE RECALL EN. (14.44) FOR THEWIF POF RUCKET!

. F = dm at

USEFUL POUTER = Por = diment

WASTED POWER = K.E. ENERGY INTERPLIED PER SECUND
TO EXPELLED FUEL WHOSE APPOINTE

VELICITY IS U. - U.

= \frac{dm}{dt} (U-U)^2

TOTAL POWER = USEFUL FOWER + WASTED FOURR

= $\frac{\partial m}{\partial t} uv + \frac{1}{2} \frac{\partial m}{\partial t} (u - v)^2$ = $\frac{1}{2} \frac{\partial m}{\partial t} (2uv + u^2 + v^2 + 2uv)$ = $\frac{1}{2} \frac{\partial m}{\partial t} (u^2 + v^2)$

 $EPFICIENCY = D = \frac{UCEFUL POWER}{TOTAL POWER} = \frac{dP uly}{dM}(u^2+\theta^2)$ $\eta = \frac{duv}{u^2+v^2} \qquad (O, E.D.)$

WHEN HEV, THE ABSOLUTE VELOCITY IL- V OF THE EXPELLED FUEL IS ZERD. THUS, NO ENTRGY IS IMPARTED TO THE EXPELLED FUEL AND NO POWER IS WASTED.

14.105



GIVEN:

30-9 BULLET FIRED
WITH 00 = 48000/5 INTO
5-kg BLOCK A, WHICH
RESTS ON 4-kg CATC.

Mk = 0.50 BETWEEN
BLOCK A AND CARTC.

FIND (a) FINAL VELOCITY US OF CARTAND BLUCK, (b) FINAL POSITION OF BLOCK ON CART.

CONSERVATION OF LINEAR MOMENTUM

mo vo = (mo + ma) v' = (mo + ma + mc) v+

(0,030kg)(480 m/s) = (5,030 kg) v; = (9.030 kg) v;

 $U' = \frac{0.030}{5.030} (480 \, \text{m/s}) = 2.863 \, \text{m/s}$

(6) WORK - ENERGY PRINCIPLE

JUST AFTER IMPACT:

 $T' = \frac{1}{2} (m_0 + m_A) v^2 = \frac{1}{2} (5.030 \text{kg}) (2.863 \text{ m/s})^2 = 20.615 \text{ J}$

FINAL KINETIC = NERGY: == 12 (9,030kg) (1.5947m/s) = 11.482 J

WORK OF FRICTION FORCE:

 $F = \mu_k N = \mu_k (m_0 + m_A) g = 0.50(5.030)(9.81) = 24.672 N$ WORK = U = -Fx = -24.672 x

T'+U= I: 20,615-24,672x=11,482 x=0.370 in



GIVEN: 80-Mg ENGINE A WITH Vo=6.5 km/h STRIKES 20-Mg FLATCHR C WHICH IS AT REST AND CARRIES 30-Mg LOAD B. A AND C ARE COUPLED UPON IMPACT. B CAN SLIDE ON C WITH Mk = 0.25. FIND VELOCITY OF CAR C
(a) IMMEDIATELY AFTER IMPACT
(b) AFTER B HAS SLID TO A STOP RELATIVE TO C.

CONSERVATION OF LINEAR MOMENTUM

PIRST NOTE THAT & WILL HOT MINE DURING COUPLINGS OF HAND G, SINCE THE PRICTION FORCE EXECTED ON B BY C IS NONIM PULSIVE: FAL- M, NOT 30.

MAV = (MA+mc) v' = (MA+mc+me) V; (80 Mg)(6.5 km/h) = (100 Mg) v'= (130 Mg) Vf

(a)
$$v' = \frac{80}{100} (6.5 \text{ km/h})$$

(b)
$$v_f = \frac{80}{130} (6.5 \, \text{km/h})$$

14.107



GIVELL: THREE IDENTICAL CARS WITH VELOCITIES SHOWN, CAR B IS FIRST HIT BY CAR A. FIND FINAL MELOCITY OF EDUCATION IF (a) ALL CARS GET AUTOMATICALLY COUPLED, (b) A AND B GET COUPLED, BUT B AND C BOUNCE OFF EACH OTHER WITH C = 1 (i.e. NO EMERGY LOSS).

(a) ALL CARS AUTOMATICALLY COUPLED CONSERVATION OF LINEAR MOMENTUM: mava+meve+meve+meve=(ma+me+me)vs m(6 mi/h)+0-m(4.8 mi/h)=(3 m)vs vs=6-4.8=+0.4 vs=0.400 mi/h-

(b) CARS A AND B ONLY GET COUPLED

CONSTRUCTION OF LINEFF FORTENTUM FOR A MOB:

m (6 milh) (2m) v.

$$\begin{array}{c}
M(6mi/h) = (2mi) V' \\
V' = 3mi/h
\end{array}$$

CAR C HITS AND BOUNCES OFF FAS A AND B

2711 (3 mi/h) m (4,8 m/h) (211)U" m vc

A B C - A B C

CONS. OF LINEAR MOMENTUM:

 $2m(3) - n_1(1.8) = 2m v'' + n_1 v_c^2$ $2v'' + v_c^2 = 1.2 \text{ mi/h}$ (1)

(CONTINUED)

14,107 continued

CONSERVATION OF ENERGY (C=1):

RELATIVE VELOCITY AFTER AND BEFORE IMPACT ARE

EWAL: $U'_{c} - U'' = (3 + 4.8) \text{mi/h}$ SUBTRACTING (2) FROM (1): 3 U'' = 1.2 - 7.8 $U''_{c} = -9.20 \text{mi/h}$ THUS: $U''_{c} = -9.20 \text{mi/h}$ $U''_{c} = -9.20 \text{mi/h}$

14.108 GIVEN:

QUOO-16 HELICOPTER A 13 TRAVELING

DUE EAST AT 75 m/h AT ALTITUDE OF 2500 FT WHEN

TO 15 HIT BY 12,000-16 HELICOPTER B. THEIR

ENTANGLED WRECKAGE FALLS TO THE GROUND IN IZ S

AT POINT LOCATED 1500 FT EAST AND 384 FT SOUTH

OF POINT OF IMPACT

FIND VELOCITY COMPONENTS OF HELICOPTER &
JUST BEFORE CULLISION. (NEELECT AIR RESISTANCE.)

VELOCITY OF WRECKAGE IMMEDIATELY AFTER

COLLISION

UP 18

$$v' = v_{k}'\dot{t} + v_{b}'\dot{g} + v_{c}'\dot{k}$$

BUT:

 $x = v_{k}'\dot{t}$
 $v'_{z} = \frac{\pi}{t} = \frac{1500 ft}{12 s} = 125 ft/s$
 $z = v_{k}'\dot{t}$
 $v'_{z} = \frac{\pi}{t} = \frac{384 ft}{12 s} = 32 ft/s$
 $v'_{z} = \frac{\pi}{t} + \frac{\pi}{2} gt$
 $v'_{z} = \frac{\pi}{t} + \frac{\pi}{2} gt$
 $v'_{z} = \frac{2500 ft}{12 s} + \frac{\pi}{2} (32.2 ft/s^{2})(12s)$
 $v'_{z} = -15.133 ft/s$

THUS: U' = (1256t/s) + - (15.133 ft/s) + (32 ft/s) +

IMPACT: CONSERVATION OF LINEAR MOMENTUM

MAUA + MBUB = (MA+MB) U'

AFTER SUBSTITUTING DATA AND EXPRESSION FOUND FOR U', AND NOTING THAT VA = 75 mi/h = 110 ft/s,

9000 lb (110 ft/s) i + 12,000 lb YB

31 000 lb (110 ft/s) i (12 000 lb)

= 21,000 15 [(125 ft/s)i-(15, 133 ft/s)j +(32 ft/s)x

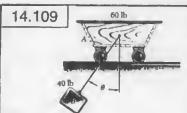
SOLVINE FOR 1/2:

1.75 [(125 +t/s)i-(15,133 ft/s)j+(32 5t/s)k-(82.5ft/s)i

IT FOLLOWS THAT $(N_8)_{2} = 1.75(125ft/s) - 82.5 ft/s = 136.25tt/s = 92.90 mi/b$

 $(v_B)_3 = -1.75(15.133 \text{ ft/s}) = -26.48 \text{ ft/s}$ $(v_B)_2 = 1.75(32 \text{ ft/s}) = 56.0 \text{ ft/s} = 38.18 \text{ mi/h}$

ANSWER: 92.9 mi/h EAST, 30.2 mi/h SOUTH, 26.5 ft/s DOWN

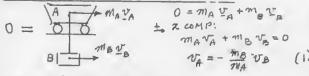


GIVEN:
BLOCK B IS SUSPENDED
FROM 6-FT CORT
ATTACHED TO CART H
SYSTEM IS RELEASED
FROM REST WHEN 8 = 35.
FIND:

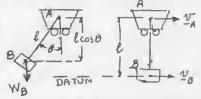
VELOCITIES OF A AID

& WHEN B = O.

CONSERVATION OF LINEAR MOMENTUM



CONSERVATION OF FHERGY



INITIALLY: $T_0 = 0$ $V_0 = W_B \ell(1-\cos \theta)$ = $m_B g \ell(1-\cos \theta)$

AS B PASSES UNDERA:

$$T = \frac{1}{2} M_A V_A^2 + \frac{1}{2} m_B V_6^2$$
 $V = 0$

To+Vo = T+ V:

$$m_{B}gl(1-\cos\theta) = \frac{1}{2} m_{A} v_{A}^{2} + \frac{1}{2} m_{B} v_{B}^{2}$$

 $m_{A} v_{A}^{2} + m_{B} v_{B}^{2} = 2 m_{B} g l'(1-\cos\theta)$

SUBSTITUTING FOR VA FROM (1):

$$m_{A} \left(\frac{m_{B}}{m_{A}}\right)^{2} v_{B}^{2} + m_{B} v_{D}^{2} = 2 m_{B} g l (1 - \cos \theta)$$

$$\frac{m_{B}}{m_{A}} (m_{A} + m_{B}) v_{B}^{2} = 2 m_{B} g l (1 - \cos \theta)$$

$$\frac{m_{A} + m_{B}}{m_{A}} v_{B}^{2} = 2 g l (1 - \cos \theta)$$

$$v_{B}^{2} = \sqrt{\frac{z m_{A}}{m_{A} + m_{B}}} g l (1 - \cos \theta)$$

GIVEN DATA:

$$\frac{m_A}{M_{A}+m_B} = \frac{W_A}{W_A+W_B} = \frac{60 \text{ lb}}{60 \text{ lb} + 40 \text{ lb}} = 0.6$$
 $\ell = 6 \text{ ft}, \quad \theta = 35^{\circ}$

To = \(\frac{1}{2}\left(0.6)\left(32;2\ft/s^2)\left(6\ft)\left(1-cos 35°) = 6.4752\ft/s

CARRYING THIS VALUE INTO (1):

$$V_A = -\frac{a_0}{a_{IA}}V_B = -\frac{W_B}{W_A}V_B = -\frac{40/b}{60/b}(6.4752 \text{ ft/s})$$

= -4.3168 ft/s

ANSWER:



GIVEN:

9-kg BLOCK B STARTS FROM REST AND SLIDES DOWN 15-kg IVEDGE A. FIND:

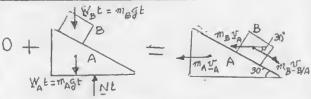
(a) VELUCITY OF B
RELATIVE TO A ATTER
IT HAS SLID 0.6 m

(b) CURRESPONDING VELOCITY OF WEDGE A. (MEGLECT FRICTION.)

WE RESOLVE YB INTO 175 COMPONETT: YA

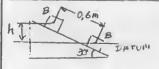


IMPULSE - MOMENTUH PRINCIFLE



 $\sum m v_{D} + \sum \bar{r} t = \sum m v_{D}$ $+ \chi conv.; \qquad 0 + 0 = m_{B} v_{B/A} \cos 30^{\circ} - m_{A} v_{A} - m_{B} v_{A}$ $1\bar{r}_{A} = \frac{m_{B} \cos 30^{\circ}}{m_{A}^{2} + n_{B}} v_{B/A} = \frac{(9 k_{A}) \cos 30^{\circ}}{15 k_{A}^{2} + 9 k_{B}} v_{A}^{2} = 0.32476 t_{B/A}^{2}$ (1)

CONSERVATION OF ENERGY



 $T = \frac{1}{2} n_{\mu} v_{A}^{\alpha} + \frac{1}{2} m_{B} v_{B}^{L} \qquad \qquad \forall = 0$

REFERRING TO VELOCITY TRIANGLE SHOWN ABOVE AND USING THE LAW OF CUSINES:

 $T = \frac{1}{2} \cdot H_A v_A^3 + \frac{1}{2} \cdot H_B \left(v_A^3 + v_{B/A}^4 - 2 v_A^7 v_{B/A} \cos 30^{\circ} \right)$

 $+\frac{1}{2}(4)[(0.32476)^{2}+1-2(0.32476)\cos 30]U_{BA}^{2}$ $=0.79102 U_{BA}^{2}+2.44336 U_{BA}^{2}=3.2344 U_{BA}^{-2}$

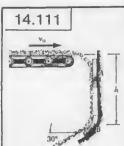
 $T+V=T_0+V_0:$ 3.2344 $V_{8/A}=26.487 J$ $V_{8/A}=2.8617 m/s$

1 = 2,86 m/s 30°

(b) FROM EO. (1): $V_A = 0.32476(2.8617 m/s)$ = 0.92936 m/s

v=0,929 m/3

(a)



MASS Q OF SAND DISCHARGED PER UNIT TIME FROM CONVEYUR BELT AND DEFLECTED BY PLATE AT A SO THAT IT FALLS IN . A VERTICAL STREAM UNTIL IT 15 DEPLECTED BY PLATE AT B. FIND FORCE REWULKED TO HOLD (a) PLATE A. (b) PLATE B. NEGLECT FRICTION BETWEEN SAMD APAD

(a) IMPULSE-MUMENTUM PRINCIPLE FOR PLATE A AND SAND

FLATES.)





Am = q bt

+2 (OMP : (OM) 10 - A Dt = 0 A = AM Vo = q Vo

(b) LMPULSE-MOMENTUM PRINCIPLE FOR PLATE BANDSAND 11 = 11 = V2gh



An = qat

1, x comp: 0 - B, at = - (am) u, cos 30°

$$B_2 = \frac{\Delta \pi 1}{\Delta t} u_2 \cos 30^\circ = 9 \sqrt{28 h} \frac{13}{2}$$

$$B_2 = \frac{1}{7} 9 \sqrt{69 h}$$

+ 1 y comp: (Am) u, - By At = (Am) uz sin 30°

$$B_y = \frac{\Delta m}{\Delta t} \left(\omega_1 - \omega_2 \sin 30^\circ \right) = 9 \sqrt{2gh} \left(1 - \frac{1}{2} \right)$$

$$B_y = \frac{1}{2} g \sqrt{2gh}$$



 $B' = B_A + B_J$ =(9)*(6gh+2gh) = 29*gh

$$\tan \theta = \frac{B_J}{B_A} = \frac{\sqrt{2gh}}{\sqrt{6gh}} = \frac{1}{\sqrt{5}}, \quad \theta = 30^\circ$$

B=9/29/1 >30





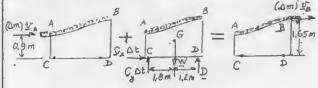
GIVEN:

SAND RECEIVED AT A AND DISCHARGED AT B AT A RATE OF 100 kg/s AND WITH U = 0 = 4.5 m/s. COMBINED WEIGHT OF COMPONENT AND SAND IT SUPPORTS IS W = 4 KN.

FIND:

REACTIONS AT C AND D.

WE APPLY THE IMPULSE-MONENTUM PRINCIPLE TO THE COMPONENT, THE SAND IT SUPPORTS AND THE SAND IT RECEIVES IN THE INTERVAL AT.



$$\pm 2 \text{ GMP}$$
: $(\Delta m) v_A + C_x \Delta t = (\Delta m) v_B$

$$C_x = \frac{\Delta m}{\Delta t} (v_B - v_A) = (100 \text{ kg/s}) (4.5 \text{ m/s} - 4.5 \text{ m/s}) = 0$$

+9 MOMENTS A BOUT C:

 $-(\Delta m)V_{A}(0.4m)-(W\Delta t)(1.8m)+(D\Delta t)(3m)=-(\Delta m)V_{B}(1.65m)$ $3D = 1.8W + \frac{\Delta m}{\Delta t}(0.9V_A) - \frac{\Delta m}{\Delta t}(1.65V_B)$ = 1.8 (4000N) + 0.9(100 kg/s) 4.5 oVs) - 1.65(100 kg/s)(4.5m)

= 6862,5 N D = 2287.5 N

D=2.29 KN7

+ 13 COMP .:

RECALLING THAT C, -0:

C= 1.712 kN

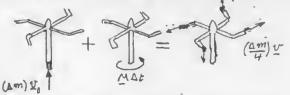
NOTE. IF COMPONENT WAS STOPPED AND THE SAND WAS NOT MOVING, WE WOULD HAVE

GIVEN:

EACH OF THE FOUR ROTATING ARMS OF SPRINKLER CONSISTS OF TWO STRAIGHT PORTIONS OF PIPE FORMING 120 ANGLE. EACH ARM DISCHARGES WATER AT THE RATE OF 20 L/min WITH RELATIVE VELOCITY OF 18 m/s. FRICTION IS EQUIVALENT TO COUPLE M = 0.375 N·m. FIND:

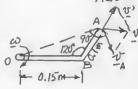
CONSTANT RATE AT WHICH SPRINKLER ROTATES.

WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE
TO THE SPRINKLER, THE WATER IT CONTAINS, AND
THE MASS AM OF WATER ENTERING IN INTERVAL DE.



EQUATING MOMENTS ABOUT AXIS OF RUTATION $\frac{1}{2}$: $0 + M \Delta E = 4 \left[MONENT OF \left(\frac{\Delta m}{4} \right) \mathcal{V} \right]$

MAT = MOMENTOF (AM) & (1)



THE VELOCITY V

OF THE WATER LEAVING

AN ARM IS THE

RESULTANT OF THE

VELOCITY U' RELATIVE

TO THE ARM AND OF THE

VELOCITY U_A OF NUZZLE:

WHERE U'=18 m/s AND VA = (OA) W BUT APPLYING THE LAW OF COSINES TO TELANGLE OAB:

 $(0A)^8 = (0B)^8 + (BA)^6 - 2(0B)(BA) \cos 120^9$ = $(0.15m)^7 + (0.10m)^6 - 2(0.15m)(0.10m) \cos 120^9$ $(0A)^8 = 0.0475 m^8$

THEREFORE:

+) MOM. OF Y ABOUT D = MOM. OF Y'+ MOM. OF Y

= (0.15 m) & cos 30- (OA)(OA) W

= (0.15m)(18 m/s) cos 30° - (0A)200

= $2.3383 \text{ m}^{3}/5 - (0.0475 \text{ m}^{2}) \omega$

SUBSTITUTING-INTO EO.(1) AND RECALLING THAT M = 0.375 N.m:

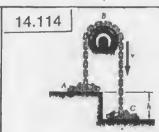
(0.375 N.m) At = (Am) [1,3383075-(0.047501) W)

DIVIDING BY Dt, AND NOTING THAT

AM = PQ = (1 kg/L)(80 L/min) 1min = 4 Kg/s

 $0.375 \text{ N·m} = \left(\frac{4}{3} \text{ kg/s}\right) \left[2.3383 \text{ m/s} - (0.0475 \text{ m}^2) \omega\right]$ $2.33.83 \text{ m/s} - \left(0.0475 \text{ m}^2\right) \omega = 0.20125 \text{ m/s}$

w = 43.306 rad/s W = 414 rpm

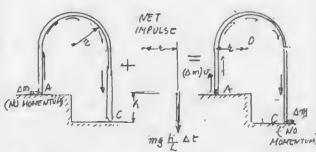


GIVEN:

WHEN GIVEN AN INITIAL
SPEED of, THE CHAIN
KEEDS MOVING OVER
THE PULLEY.
FIND:
HEIGHT h.

(NEGLECT PRICTION.)

WEAPPLY THE MAPULE - MODERATION FRANCIPLE TO THE PORTION OF CHAIN OF HASS ON AND LENGTH L IN MOTION ATIME L AND TO THE ELEMENT OF LENGTH DE AND HASS ON = MLDZ WHICH WILL BE SET IN MOTION IN THE TIME LINTERVAL LE.



WE NOTE THAT THE ELEMENT AT A ACCURES A LINEAR HOMENTUM (AM) V WHICH IS ADDED TO THE SYSTEM, WHILE THE MOMENTUM OF THE ELEMENT AT C IS LOST TO THE SYSTEM.

EMUNTING MUNICIPIS ABOUT D:

+2 0 + (n/9 \(\text{\text}\) \(\text{\tint{\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\tinter{\text{\tert{\text{\text{\text{\text{\text{\texi}\text{\text{\texict{\text{\text{\text{\text{\text{\text{\text{\texi}\text{\text{\texit{\ti

 $= \frac{\Delta^2}{\Delta^2} \frac{v}{g} = v \frac{v}{g} \qquad h$

14.115



GIVEN:

RAILROAD CAR OF MASS

mo AND LENGTH L

APPROACHES CHUTE AT

SPEED VO TO BE WADED

WITH SAND AT RATE

dm/dt = q.

FIND: (a) MASS OF CAR AND LOAD AFTER CAR HAS PASSED.
(b) SPEED OF CAR AT THAT TIME.

CONSEPTATION OF MOMENTURE IN HORISONTAL DIRECTION WE CONSIDER THE CAR AND THE MASS OF SAND OF WHICH FALLS INTO THE CAR IN THE TIME E.

$$\frac{m_b \, \psi_o}{m_0 \, \psi_o} = \underbrace{ \left[(m_0 + 9 \, t) \, \psi \right]}_{m_0}$$

 $x = \frac{4n_0 v_0}{m_0 + q t}$ $v = \frac{4n_0 v_0}{m_0 + q t}$ (1)

LETTING V = dz IN (1):

 $dx = \frac{m_0 v_0 dt}{m_0 + qt} \qquad x = m_0 v_0 \int_0^t \frac{dt}{m_0 + qt}$ (CONTINUED)

14.115 continued

$$z = \frac{m_0 t_0}{q} \left[\ln \left(\dot{m}_0 + q t \right) \right]_0^t = \frac{m_0 t_0}{q} \ln \frac{m_0 + q t}{m_0} \tag{2}$$

USING THE EXPONENTIAL FORM: mo + qt = mu e qx/movo

WHERE MIT OF REPRESENTS THE MASS AT TIME L, AFTER THE CAR HAS HOVED THROUGH Z.

(a) MAKING X=L IN (2), WE OBTAIN THE FINAL MASS. ms = mo + 9ts = mo e 9L/mos

(b) MAKING [= C; IN(1), WE OBTAIN THE FINAL SPEED: v = mo 10 = mo v = ve-91/mov.

14.116





GIVEN:

SPACE VEHICLE DESCRIBING CIRCULAR ORBIT ABOUT THE THRIH AT SPEED OF 15,000 milh RELEASES AT ITS FRONT END A CAPSULE WITH A GROSS WEIGHT OF 1200 1b, INCLUDING 800 1b OF FUEL , WHICH IS CONFUNED AT THE RATE OF 40 16/5 AND EJECTED WITH RELATIVE VELOCITY OF 9000 ft/s.

FIND:

(a) TANGENTIAL ACCELERATION OF CAPSULE AS 17 IS FIRED,

(b) MAX, SPEED ATTAINED BY THE CAPSULE.

PROMER. (14.44): THRUST = $P = \frac{dm}{dt} = \frac{40 \text{ W/s}}{32.2 \text{ ft/s}} (9000 \text{ ft/s})$ = $\frac{360 \times 10^3}{32.2} \text{ /b}$

(a) $P = m_0 a_t$: $\frac{360 \times 10^3}{32.2} \% = \frac{1700\%}{32.24\%}, a_t$

 $a_{t} = \frac{360 \times 10^{3}}{1.2 \times 10^{3}} ft/s^{2}$ $a_{t} = 300 ft/s^{2}$

(b) HAX, SPEED OF CAPSULE RELATIVE TO SPACE VEHICLE I DETAINED PROM EXPRESSION DERIVED IN PRUB, 14.95 OK FROM EXPRESSION OBTAINED IN SAMPLE PROB. 14.8 BY MITTING THE TERM DUE TO FRAVITY.

$$V_{c/v} = u \ln \frac{m_0}{m_0 - qt}$$

WHERE AL = (1000ft/s) mo = 1200 1b, mo - 9t = 1200 1b - 800 1b = 400 1b

THUS:

VC/V = (9000 ft/s) 2113 = (9000 ft/s)(1.0986) = 9887.5 ft/s = 6741 mi/b

Vc = 15,000 ni/h + 6741 n 1/0 = 21,741,5 mi/h

v = 21,700 mi/h

14.C1



GNEN:

WOMAN OF WEIGHT WASTANGS READY TO DIVE WITH VEWCTY VIOL RELATIVE TO BOAT OF WEIGHT Wb . MAN OF VEIGHT W'M READY TO DIVE FROM OTHER END OF BOAT WITH RELATIVE VELOCITY Now

VELOCITY OF BOAT AFTER BOTH SNIMMERS HAVEDINED IF (a) WOMAN DIVES PIRST, (b) MAN DIVES FIRST USE W = 1201b, W = 1801b, W = 3001b, AND (PROB. 14.4): Now = Nom = 16 ft/s

(i) No = 14 ft/s, NA = 18 ft/s.

(Li) Vw=18+t/s, Vm =14+t/s

ANALYSIS

(a) WOILEN DIVES FIRETS

UN = VEL. OF BOAT AFTER WUMAN DIVES Uh = VEL OF BOAT AFTER BOTH SWIMMERS HAVE DIVED

CONSERVATION OF MOMENTUM! War (Var-Vb) (Wb+Wm) Ub U's = Was var (1) 0 = - Wy (Vy - Vb) + (Wb+ Wo) VB

 $(W_b + W_{a_1}) V_b^i = W_b V_b + W_m (V_b + V_b^i)$ $V_b = V_b^i - \frac{W_b V_{a_1}}{V_b V_{a_2}}$

SUBSTITUTING POR UB FROM (1): $\frac{1}{4} \quad V_b = \frac{W_{40} \quad V_W}{W_W + W_W + W_b} - \frac{W_m \quad V_m}{W_m + W_b}$ (2)

(b) MAN DIVES FIRST:

INTERCHANGE SUB'W AND SUB MIN (2) AND CHANGE ALL SIGNS

+ War Var Vb = - Wm Vm Vm Wb

OUTLINE OF PROGRAM INPUT War, Wm, Wb, Var, Vm, AND EUS, (2) AND (3).

PROGRAM OUTPUT

PROB. 14.6 (a) Woman dives first Velocity of boat = -2.800 (b) Man dives first Velocity of boat = -0.229

(a) Woman dives first (L) Velocity of boat = -3.950 (b) Man dives first Velocity of boat - -1.400

(a) Woman dives first Velocity of boat = -1.650 (b) Man dives first Yelocity of boat = 0.943 (ii)

14.C2 GIVEN:

SYSTEM OF IN PARTICLES A: OF MASS IN:
COURDINATES X; 3; 4; , WITH VELOCITIES OF COMPONENTS
(V2); (Vy); (V2):
FIND:

COMPONENTS OF ANGULAR HOMENTUM OF SYSTEM ABOUT ORIGIN O. USE PROGRAM TO SOLVE PROES. 14.9 AND 14.13.

ANALY515

$$H_{0} = \sum_{i=1}^{m} \underline{v}_{i} \times m_{i} \underline{v}_{i} = \sum_{i=1}^{m} m_{i} \left| \begin{array}{ccc} \underline{i} & \underline{j} & \underline{k} \\ \overline{z}_{i} & \underline{y}_{i} & \overline{z}_{i} \\ (\underline{v}_{k})_{i} & (\underline{v}_{y})_{i} & (\underline{v}_{y})_{i} \end{array} \right|$$

$$H_{x} = \sum_{i=1}^{n} m_{i} \left[y_{i}(v_{x})_{i} - z_{i}(v_{y})_{i} \right] \tag{1}$$

$$H_{i} = \sum_{i=1}^{n} m_{i} \left[\frac{1}{2} \left(\left(\frac{\nabla}{z} \right)_{i} - \chi_{i} \left(\frac{\nabla}{z} \right)_{i} \right]$$
 (2)

$$H_{z} = \sum_{i=1}^{l} m_{i} \left[z_{i}(v_{y})_{i} - y_{i}(v_{z})_{i} \right]$$

$$(3)$$

OUTLINE OF PROFRAM

ENTER PROBLEM NUMBER AND JYSTEM OF UNITS USED IF SI UNITS, ENTER FOR i = 1 TO i = n: $m_i(ke); z_i, y_i, z_i (m); (v_z)_i, (v_j)_i, (v_z)_i (m/s)$ IF U.S. CUSTOMARY: UNIS, ENTER FOR i = 1 TO i = n: $W_i(lb); z_i, y_i, z_i (ft); (v_x)_i, (v_y)_i, (v_z)_i (ft/s)$ AND COMPUTE $m_i = W_i/32.2$

COMPUTE THE SUMS (1), (2), AND (3).

PRINT PROBLEM NUMBER

PRINT VALUES OBTAINED FOR H, Hy, Hz.

IF SI UNITS, RESULTS ARE EXPRESSED IN kg·m/s.

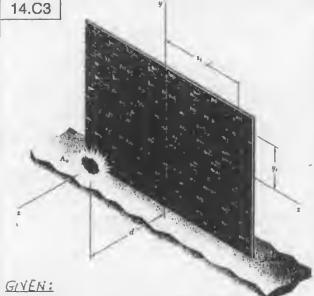
IF U.S. CUSTOFIARY UNITS, RESULTS ARE

EXPRESSED IN ft·16·5.

Probert nutrut

Problem 14.09 Hx = -31.2 kg*m*2/s Hy = -64.8 kg*m*2/s Hz = 48.0 kg*m*2/s

Problem 14.13 Hx = 0.000 ft*lb*s Hy = -0.720 ft*lb*s Hz = 1.440 ft*lb*s



SHELL MOUNTS WITH VELOCITY OF COMPANIONS

V_X, V_Y, V_Z EXPLOSES IN THREE FRAMENTS OF WEIGHTS

W₁, V_X, W₃ AT POINT A₀ AT DISTANCE & FROM WHILL.

FRAMENTS HIT THE WALL AT POINTS A₁ (2=1,2,3)

OF CONDINATES X₁ AUSY;

FIND: CORT CORE FROM FRAMENT OF THE EXPLOSION

FIND: SPEED OF EACH PRAGMENT AFTER EXPLOSION USE PROGRAM TO SOLVE (a) PROB. 14.25, (b) PROB. 14.26.

ANALYSIS

DETERMINE DIRECTION COLINES OF PATH A A: (i=1,2)
FIRST CONPUTE & = 1/2; + 3; + d2 (1)

THEN $(\lambda_x)_i = x_i/\ell_i$, $(\lambda_y)_i = y_i/\ell_i$, $(\lambda_s)_i = -d/\ell_i$ (2)

CONSERVATION OF LINEAR MOMENTUM:

 $\frac{1}{9}(W_1 + W_2 + W_3)(V_2 + V_3 + V_3 + V_4 +) = \frac{W_1}{9}V_1 + \frac{N}{9}V_2 + \frac{N}{9}V_3 +$

2. COMP: $W_1(\lambda_2)_1 U_1 + W_2(\lambda_2)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (3) 4. COMP: $W_1(\lambda_3)_1 U_1 + W_2(\lambda_3)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (4) 2. COMP: $W_1(\lambda_3)_1 U_1 + W_2(\lambda_3)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (5) THESE 3 EQS. ARE SOLVED SIMULTANEOUSLY FOR V_1, V_3, U_3

CUTLINE OF PROGRAM
ENTER PROBLEM VICTORER
ENTER VALUES OF VI, VI, VI, AND d
ENTER VALUES OF WI, XI, YI FOR i=1,2,3
COMPUTE DIRECTION COSINES FROM EOS. (1) AND (2)
COMPUTE COEFF. IN EQS. (3), (4), (5) AND SOLVE FOR VI, VI, VI
BY COMPUTING

AND $V_1 = D_1/D$, $V_2 = D_2/D$, $V_3 = D_3/D$ PROGRAM OUTPUT

> (A) Problem 14.25 VA = 1678 ft/s VB = 1390 ft/s VC = 1230 ft/s

(b) Problem 14.26 VA = 2097 ft/s VB = 1853 ft/s VC = 738 ft/s

14.C4



GIVEN: AS 6000-Ky PLANE LANDS ON CARRER AT 180 Km/h, 175 TAIL HUDRS INTO END OF 80-m LONG CHAIN OF AIM 175 TAIL HUDRS INTO END OF 80-m LONG CHAIN OF AIM 175 PER UNIT LENGTH MI' = 50 Kg/m LYING BELOW PECK.

TRAVELED BY THE PLANE AND THE CORRES PONDING VALUES OF THE TIME, THE VESSITY, AND THE ACCELERATION OF THE PLANE.

D1 = 1 11

CHESTIAN OF LINE " LONG THE TIME



$$+ \qquad n | V_0 = (m + n | Z) v \qquad (1)$$

LETTING v = dz/dt:

$$t = \int_{0}^{z} \frac{m+n'z}{mv_{4}} dz = \left[\frac{(n+n'z)^{2}}{2mn'v_{0}^{2}} \right]^{2} = \frac{(m+m'z)^{2}-n'}{2mm'v_{0}} (2)$$

SOLVING (1) FOR V:
$$v = \frac{\pi i \, \nabla_0}{m + m^2 z}$$

DIFFERENTINTING (1) WITH RESPECT TO t:

0 = m d2 v + (m + m 2) dv at

NOTING THAT du/tt =
$$v$$
 AND $dr/At = a$?
$$0 = m'v'' + (m+m'z)a \qquad \alpha = -\frac{m'n''}{m+m'z}$$

OUTLINE OF PROGRAM

ENTER 11 = 6000 kg, 11 = 50 kg/m, v = 180 km/h = 50 m/s
FOR X = 0 TO X = 80 M AND USING 5-M INCLEMENTS
CALCULATE t, V, AND a FRON EQS. (2), (3), (4) AND TABULATE

PROGRAMOUTPUT

Distance (m)		Velocity (km/h)	Acceleration [m/s^2]
0.000 5.000 10.000 15.000 20.000 25.000 30.000 40.000 45.000 50.000 55.000 60.000 65.000	0.000 0.102 0.208 0.319 0.433 0.552 0.675 0.802	180.000 172.800 166.154 160.000 154.286 144.966 144.000 139.355 135.000 130.909 127.059 123.429 120.000	-20.833 -18.432 -16.386 -14.632 -13.120 -11.809 -10.667
75.000 80.000	1.969		-4.855 -4.500

14.C5



GIVEN:

A 16-Mg PLANE MAINTAINS A CONSTANT SPEED OF 774 km/h
WHILE CLIMBING AT AN ANGLE & = 18.
PLANE SCOOPS IN AIR AT RATE OF 300 kg/s AND-

PLANE SCOOPS IN AIR AT RATE OF 300 kg/s AND-DISCHARGES IT AT A RELATIVE SPEED OF 665 m/s. PILOT THEN CHANGES ANGLE OF CLIMB & WHILE MAINTAINING THE SAME ENGINE SETTING.

FIND FOR VALUES OF & PROM O TO 20 USING 1° INCREMENTS:

(b) MAXIMUM SPEED THAT WILL BE ATTAINED.

(ASSUME DRAG TO BE PROPORTIONAL TO U.)

ANALYSIS

FROM EQ. (14.39): THRUST = P = dm (u-v) DENOTING RATE dm/dt by R:

 $P = R(u - v) \qquad (1)$ $\text{WHICE CLIMBING AT } v_0 \text{ AND } \alpha_0:$ $P_0 = R(u - v_0) \qquad (2)$ $D_0 \text{ SINCE PLANE IS IN EQUILIBRIUM:}$ $D = P_0 - mg \sin \alpha \qquad (3)$

Po W-right

(a) PLANE CLIMBING AT ANGLE & AND SPEED 5:

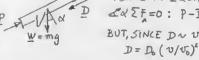
 $P_0 - D_0 - mg \sin \alpha = ma$ $P_0 - D_0 - mg \sin \alpha = ma$ $Q = (P_0 - D_0 - mg \sin \alpha)/m$ $Q = (P_0 - D_0 - mg \sin \alpha)/m$ $Q = (P_0 - D_0 - mg \sin \alpha)/m$ $Q = (P_0 - D_0 - mg \sin \alpha)/m$ $Q = (P_0 - D_0 - mg \sin \alpha)/m$

(b) MAX, SPEED WHILE CLIMBING AT ANGLEX:

PLANE IN EQUILIBRIUM.

AND FEO: P-D-mgsing=0 (5)

BUT, SINCE DN UZ, WE HAVE



SUBSTITUTING FOR D INTO (5) AND FOR P FROM (1): $R(u-v) - \left(\frac{D_0}{v_0^2}\right)v^2 - mg \sin \alpha = 0$ $v^2 + \frac{Rv_0^2}{D_0}v + \left(v_0^2/D_0\right)(mg \sin \alpha - Ru) = 0$

SET $B = R v_0^3 / D_0$, $C = (v_0^3 / D_0) \times mg \sin \alpha - Ru$ (6)

MAX, SPEE D: $v_{aa} = v = \frac{1}{2} \left(-B + \sqrt{B^2 + C} \right) \tag{7}$

OUTLINE OF PROGRAM

R= 300 kg/s, 12= 665 m/s, g= 9.81 m/s. Q= 18°, USE EUS. (2) AND (3) TO CALCULATE PO AND DO.

CALCULATE B = R vo /Do.

FOR & FROM O TO 20, WITH 1° INCREMENTS

(a) USE ED. (4) TO CALCULATE a

(b) USE EQS. (6) AND (7) TO CALCULATE UMAY

(CONTINUED)

14.C5 continued

PROGRAM O'ITPUT

-1-6		
	celeration	
degrees	m/s^2	km/h
0.000	3.031	921.796
1.000	2.860	913.933
2.000	2.689	906.020
3.000	2.518	898.060
4.000	2.347	890.053
5.000	2.176	882.002
6.000	2.006	873.907
7.000	1.836	865.770
8.000	1.666	857.594
9.000	1.497	849.378
10.000	1.328	841.126
11.000	1.160	832.839
12.000	0.992	824.518
13.000	0.825	816.166
14.000	0.658	807.785
15.000	0.492	799.375
16.000	0.327	
17,000		790.940
18.000	0.163	782.481
	0.000	774.000
19.000	-0.162	765.499
20.000	-0.324	756.981

14.C6 GIVEN:

ROCKET OF WEIGHT 2400 16, INCLUDING 2000 16 OF FUEL, IS FIRED VERTICALLY FROM GROWN, IT CONSUMES PUEL AT RATE OF 25 16/5 AND EJECTS 1T WITH RELATIVE VELOCITY OF 12,000 ft/s.

FIND FROM TIME OF IGNITION TO TIME WHEN LAST PARTICLE OF FUEL IS CONSUMED, AND AT 4-5 TIME INTERVALS:

- (A) ACCELERATION IL OF ENCHET IN FE/52,
- (b) ITS VELOCITY VINI ft/s,
- (c) ITS ELEVATION H ABOVE GROUND IN MILES.

ANALYS 15

WE RECA: L TROM SAPPLE PROB. 14.8 THAT

$$v = u \ln \frac{m_0}{m_0 - qt} - gt \tag{1}$$

WHELE UE VELOCITY OF ROCKET

MO T INITIAL WEIGHT OF ROCKET AND FUEL

Q = RATE AT WHILL FUEL IS CONSUMED

W = RELATIVE VELOCITY AT WHICH FUEL IS EJECTED

LEITING 1 = dy/dt ANT INTEGINING Y FROM OTO h: $h = \int_0^h dy = u \int_0^t ln \frac{m_0}{m_0 - qt} dt - \frac{1}{2}gt$ (2)

TO CALCULATE THE INTEGRAL, WE SET MO-96= 2

AND BETHIN dt = - MO dz. THEREFORE:

$$\int_{0}^{t} \frac{m_{0}}{m_{0}-qt} dt = \int_{0}^{z} -\ln z \left(-\frac{m_{0}}{q}dz\right).$$

$$= \frac{m_{0}}{q} \int_{0}^{z} \ln z dz - \frac{m_{0}}{q} \left[z \ln z - z\right]^{z} = \frac{m_{0}}{q} \left(z \ln z - z + 1\right)$$

THUS, EQ. (2) YIELDS

$$h = \frac{m_0 u}{q} \left(\frac{m_0 - qt}{m_0} ln \frac{m_0 - qt}{m_0} - 1 + \frac{qt}{m_0} + 1 \right) - \frac{1}{2} gt^2$$

$$h = u \left(t + \frac{m_0 - qt}{q} ln \frac{m_0 - qt}{m_0} \right) - \frac{1}{2} gt^2$$
 (3)

REWRITING EQ. (1) AS

AND DIFFECTIVITY NEWITH CESPECT TO t,

$$a = \frac{dv}{dt} = -u \frac{-\eta}{\eta \eta_0 - qt} - g$$
 $a = \frac{u q}{\eta \eta_0 - qt} - g$ (4)

(CONTINUED)

14.C6 continued

OUTINE OF PROGRAM

ENTER q = 31.2 ft/s, mo = 2400/g, mg = 2000/g.

q = 25/g, u = 12.000 ft/s

COMPUTE FINAL TIME = tg = mg/q = 2000/25 = 80 s

POR E PROM O TO 80 s AT 4-S INTERVALS,

COMPUTE

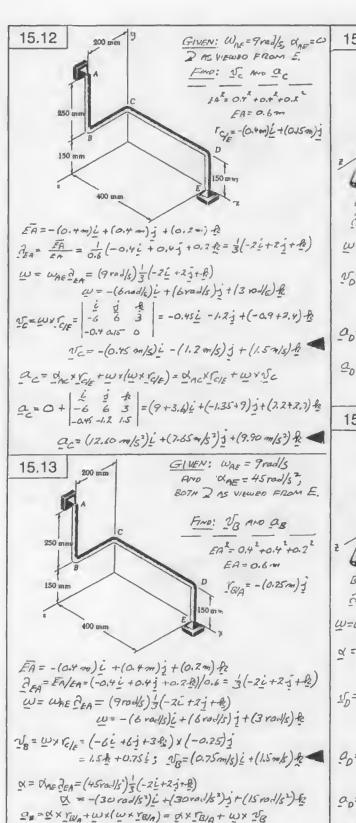
- (a) ACCELERATION a FROM EQ. (4)
- (b) VELOCITY 'U FROM EQ. (1)
- (c) ELEVATION & FROM EQ. (3), DIVIDING RESULT BY 5280 TO OBTAIN & IN MILES.

PROGRAM OUTPUT

٤	a	v	h
	ft/s^2	10^3 ft/s	mi
0.000	92.800	0.000	0.000
4.000	98.235	0.382	0.143
8.000	104.164	0.787	0.584
12,000	110.657	1.216	1.341
16.000	117.800	1.673	2.434
20,000	125.695	2,159	3.883
24.000	134.467	2.679	5.714
28.000	144.271	3.236	7.952.
32.000	155.300	3.035	10.628
36,000	167.800	4.481	13.775
40,000	182,086	5.180	17,431
44.000	198.569	5.940	21.639
48.000	217.800	6.772	26.449
52.000	240.527	7.688	31.921
56.000	267,800	8.702	38.122
60.000	301,133	9.838	45.137
64.000	342.800	11.123	53.066
68.000	396.371	12.596	62.037
72.000	467.800	14.317	72.213
76.000	567.800	16.376	83,814
80.000	717.800	18.925	97.148

GIVEN: 0=1.5t3-4.5t2+10 GIVEN: 0= 0, Sin(Tt) - 0.50, SIN(TT) 15.1 15.4 WHERE Bo= 6 rad, T=45. FIND: O, W, AND X WHEN (a) t=0, (b) t=45. Fino: O, W, AMOD WHEN (a) t=0, (b) t=25. $\omega = \frac{d\Theta}{dt} = \Theta_0 \frac{\pi}{T} \cos\left(\frac{\pi t}{T}\right) - 0.5\Theta_0 \frac{2\pi}{T} \cos\left(\frac{2\pi t}{T}\right)$ w= de/dt = 4.5t 2-9t d = JW/dt = 9t -9 $\alpha = \frac{d\omega}{dt} = -\Theta_0(\frac{\pi}{T}) \sin(\frac{\pi t}{T}) + 0.5\Theta_0(\frac{2\pi}{T}) \sin(\frac{2\pi t}{T})$ (a) t=0: @= 10 nad w=0 $\omega = 6 \frac{\pi}{4} - 0.5(6) \frac{2\pi}{4}$ $\alpha = 0$ (a) t=0: 0=0 a = - grad/s2 (b) t=45: G=1.5(4)3-45(4)+10 6=34 rad W= 4.5(4) 2-9(4) 0=6 sin(211)-0.5(6) sin(41)=6-0, 0=6 red W= 36 nad/s a = 9(4)-9; a = 27 rad/52 W= 6(#) cas(21) - as(6) 21 cos(41) GIVEN: 0-1.5t3-4.5t2+10 = 6 \frac{\pi}{4}(0) - 0.5(6) \frac{2\pi}{4}(-1) = \frac{6\pi}{4} 15.2 w=4.71 nad/s FINO: t, O, AND O WHEN W=0 $X = -6\left(\frac{\pi}{4}\right)^2 \sin\left(\frac{2\pi}{4}\right) + 0.5(6)\left(\frac{2\pi}{4}\right)^2 \sin\left(\frac{4\pi}{4}\right)$ w=de/dt = 45t2 -9t $= -6\left(\frac{\pi}{4}\right)^{2}(1) + 3\left(\frac{2\pi}{4}\right)^{2}(0) = -\frac{3}{8}\pi^{2}$ a = dw/dt = 9t -9 FOR W=0: 4.5t2-92=0 d = -3.70 nod/s" t= O ANO Z=Z. GIVEN: 0=0, sin (Tt) -0.50, sin (Tt) 15.5 0 = 10 nod, x = -9 nod/52 WHERE 60=6 red, T=45 t= 25: 0=1.5(2)-4.5(2)+10, FIND: O, W, AND ON WHEN t= 15 d= grad/s2 0 = 9(2) - 9w= do = 0 + cas(7) -0.50 27 cos(27) GNEN: 0=00(1-e-4) WITH 6=0.40 rod 15.3 $\alpha = \frac{diw}{dt} = -\Theta(\frac{T}{T})^2 \sin(\frac{TT}{T}) + \alpha SO(\frac{2\pi t}{T})^2 \sin(\frac{2\pi t}{T})$ FIMO: 6, W, AND OX WHEN (a) t=0, (b) t=35, (x) t=0 £=15: @=65in(#)-0.5(6) sin(2#) 0=0.40 (1-e-44) $=6\frac{\sqrt{2}}{2}-0.5(6)(1),$ Ø=1.243 nod € $\omega = \frac{d\Theta}{dt} = \frac{1}{4}(0.40)e^{-\frac{t}{4}} = 0.10 e^{-\frac{t}{4}}$ (w= 6(1) cos(1) - 0.5(6)(21) cas(21) $\alpha = \frac{d\omega}{dt} = -\frac{1}{4}(0.10)e^{-\frac{t}{4}} = -0.025e^{-\frac{t}{4}}$ = 6(\frac{\pi}{4}) \frac{\sqrt{2}}{2} - 0.5(6) (\frac{\pi}{2})(0), \cus 3.33 nod/s a=-6(#) sn(#)+0.5(6)(211) sin(211) @= 0.40(1-e°) (a) t=0: 0=0 = -6(T) = VZ + 0.5(6)(2)(1), X = 4.79 rod/s2 w= 0,10 e0 w=0.1 rad/s d=-0.025e0 d= - 0.025 ned/52 15.6 GIVEN: t=0, W=0 0=0.40(1-E t=65, w= 3300 yn= 1107 ned (b) t = 35: =0.40(1-0.4724). 0=0.211 nal THEN COASTS TO PREST IN 805. W=0.10 e-3/4 FIND: NUMBER OF REYOLUTIONS W=0.0472 nod/s = 0.10(0.4724) $\alpha = -0.025 e^{-3/4}$ (a) TO REACH SPEED OF 3300 APM. (b) TO CONST TO REST. = -0.025 (0.4724) A = -0.01/81 nod/s UNIFORMLY ACCELERATED MOTTON: Ub=0, &=65. (e) t= D: 0=0.40(1-e-0) (a) w= w0+dt; 110T = 0+d(6), d= 110 Tr nad/s2 = 0.40 (1-0) A=0.4 rod 0= w,t+ \frac{1}{2} at = 0+ \frac{1}{2} \left(\frac{110}{2} \pi \right) \left(\delta s \right)^2 = 330 TT ned W=010e-0 0 = (330 11 nod) 1 rev WEO d = -0.025e-00 d = 0 (b) W,=110 Trads, W2=0 WHEN E=805 W= W, +dt; 0=11017+d(80s), d=-11017 rad/52 @= W, t + 1 dt = (1107)(80) - 1 (1107)(80s) = 8800 TT - 4400TT = 4400TT and €= 2700 rev 8 = (4400 TT) 1 rey

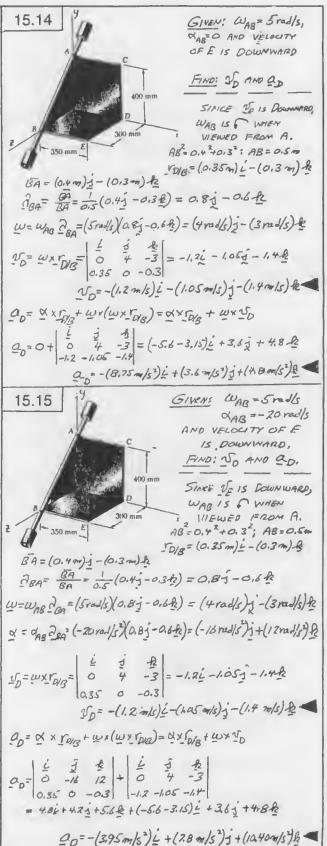
15.7 GIVEN: ROTUR CONSTS TO REST IN 4 mm. 15.10 GIVEN: Up= 7,5 nods FROM RATED SPEED OF US= 6900 mm. DAS VIEWED FROM B. FOR UNIFORMLY ACCELERATED MOTION, CAR =O. FIND: (a) ANG. ACEL. C. (b) NUMBER OF REVOLUTIONS FIND: NE AND a Wo= 6900 apr (20) = 722.57 rad/s, t=4min=240s (a) W= Wb+ dt; 0=722.57+d(240) AB = 20+9+122 a=-3.0107 rad/s, &=-3.01 rad/s2 AB-25 in. (b) = wot + 1 xt2 = (722.57)(240) + 1 (-3.0107)(240) 0=173,416-86,708 = 86,708 nal @=86,708 and (100), @=13,800 nov Am= AB = 1/201-95+12-R) 15.8 GIVEN: X=-RO. W= WAST = (7.5 mod/s) 1/25 (20i -91+12A) FIND: (a) VALUE OF & FOR WHICH W= 8 ned/s INHER Q=OAND Q=4 no d/s WHEN W=O. W= (brad/s)i- (2.7 rad/s)j+ (3.6 rad/s)-fe (b) ANGULAR VELOCITY INHEN @= 3 nad. TE/B=-(12 in.)& x=-10 VE = WXTER = (61-2.7)+3,6+)x(-12+) $\int \omega d\omega = -\int d\theta d\theta ; \left| \frac{1}{2} \omega^2 \right| = -\left| \frac{1}{2} d\theta^2 \right|$ = 72 5+37.46 · 2[=(32.4 m/s) i+(72 m/s) j 1(0-82)=-1-R(42-0) af = dxrE/B+ wx/wxrE/a) = dxrE/B+ wx VE $\left| \frac{1}{2} \omega^2 \right| = - \left| \frac{1}{2} (4s^{-1}) \omega^2 \right|$ [wdw = - | Rede ; 0=0+6-2.7 36=-25726+1166j+(43+87.7)+6 8 rolls $\frac{1}{2}(\omega^2-8^2)=-\frac{1}{2}(4)(3^2-0)$ a=-(259,2 m/s2) i+(116,6 m/s3)-j+(519 m/s3) & w=-64=-36; w=64-36=28; w=5.79 nd GIVEN: WAB= 7.5 mod/s 15.11 GIVEN: d=-0.25 w; WHEN t=0 W=20 mg/s 15.9 DAS VIEWED FROM B. FIND: (a) REVOLUTIONS BEFORE W=0. 0/AB = - 30 marys 2 (W) TIME WHEN W=0. (R) TIME WHEN W= 0.01 WO. FINO: V. AND Q d=-0.25w; w dw =-0.25db; dw=-0.25db A8 = 20+9 + 122 [dw = -0.25 | d6 ; (0-20) = -0.250 AB = 25 in. 0= 80 med AB AB G=(80 rad) rev @= 12.73 rey Yel= - (20 m.) L 7A8 AB = 15(201-95+12-B) 10=-0.25w; dw =- 0.25w; W = WAS 7 = (25 rads) = (20i-9j+12f2) $\int \frac{d\omega}{\omega} = -0.25 \int dt$ lnw =-0,25t w= (6 nods)i-(202001/s) + (36 rod/s) A d = das 200 = (-30 rad/s2) = (20i-9j +12A) t=-1 (lnw-ln 20) = 4(ln 20-lnw) d=-(24rads) + (10.8 rads) j-(4.4 rads) /2 No-wx Yelk= (66-2.75+3.6A)x(-20i) (1) t=4 ln 20 = -54R-725 NG=-(72m/s)j-(54m./s)k t=4 ln 30=4 ln 00 FOR W=0 ac= dxrelg+wx(bxrelg) = dxrelg+wxxe ac= -24 10.8 -149 + 6 -2.7 3.6 (C) FOR W= 0.01 W0=0.01(20) = 0.2 nad USE EQ(1): t=4ln(20)=4ln 100=4(4605) a= 2883 +216 \$ + (157.2+259.2) +324 -5 -432 -B +=18.42 S a= (405 m, 15) i+(612 1 152) j-(216 m, 153) 12

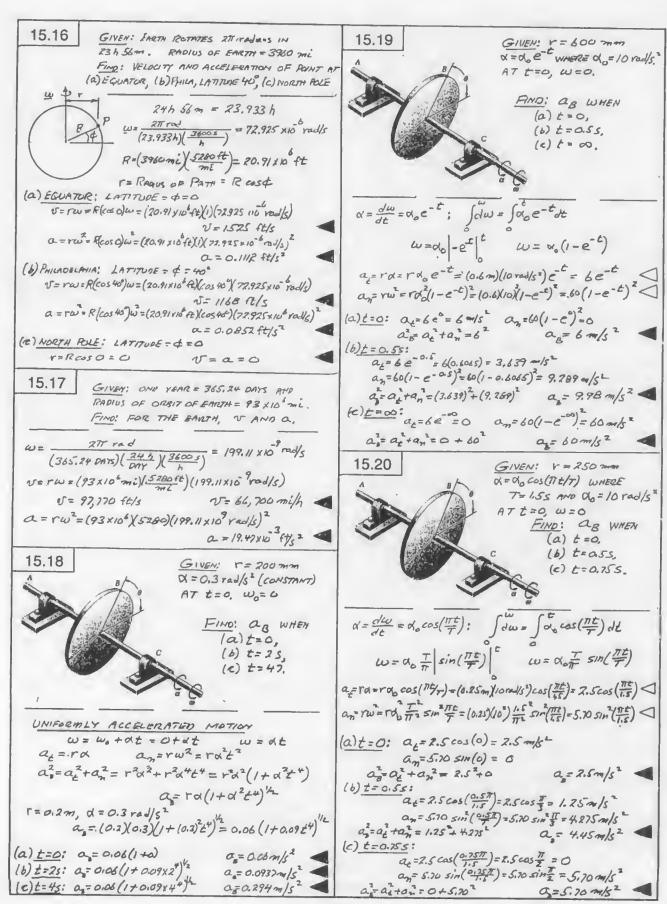


Q = -30 30 15 + -6 6 3 = 3.75 L+25-B

0-0250 0.15 0 1.5 +9i+(2.25+9)j-4.5k

a = (12.75 m/s2) i + (11.25 m/s2) 5 + (3 m/s2) 4.





15.21 FIND: (a) W AND of OF PULLET. (b) aB (a) 5=1510 = 0=910.152 v=rw 1511/5=(611)w; W= W = 25 nad/s) 971.15 = (bin) x a=rd; a=1.5 nad/52) Tam at = 91m/52 (b) an=rw=(6in.)(2. Smad/s) an = 37.5 cm/52 + ay= 910/52 a= 38.6 19/52 5765 an=37.51n/s2-15.22 GIVEN: W= 4 rad/s 2 FIND: & FOR WHICH aB= 120 in./52 = ax=rd = (6in.) X an= rw= (bin) 4rad/s) = 96in./52 6in: $a_R^2 = a_c^2 + a_n^2$ (1201n/s") = (60) + (96 in./s") x = 12 med/5 2) $\alpha = \pm 12$ 15.23 GIVEN: WHEN t=0, W=0 FIND: ac WHEN

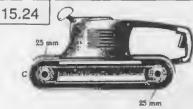


of = 120 rays2)

(a) t=0.55, (b) t=25.

a=rd=(0.025m)(120ral/s) d=120 rod/52 at= 3 m/s2+ (a) t=0.55: w= xt = (120 rad/s2)(0.55) = 60 rad/s a = rw2 = (0.025 m)(60 rod/s)2 an= 90 m/s2a= a+ a= 3+902 aB= 90,05m/52

(b) t=25: W= xt=(126 rod/52)(25)= 240 rod/s an ru= (0.025m)(240 rad/s)2 an= 1440 m/s2 a= ata= 3+14402 ag=1440 m/s2



GIVEN: RATED SPERD OF DRUMS 15 2400 rpm SAHOER CONSTS TO REST 111 105. FIND: VE AND Q (a) BEFORE POWER

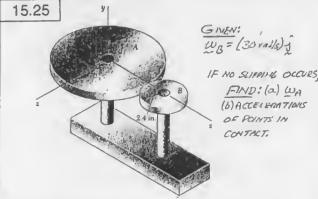
IS CUT OFF, (b) 95 LATER.

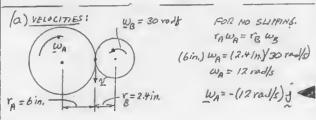
r=01025 m Wo = 2400 rpm = 251,3 nalls N= rw= (0.025m)(251.3 md/s); N=6.28 m/s a = rw=(0.025 m)(251,3 nad/s)? a=1579 m/52

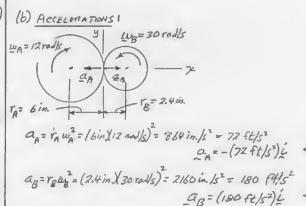
(b) WHEN ==105, W=0. (= w + dt; 0=251.3 mds + d(106); x=-25.13 rols INNEN £ = 95:

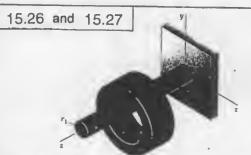
w= W+at; w= 251.3 rad/s-(25.13 rad/s)(95)=25.13 7/s N= ruj= (0.025m/25.13 rad/s); N=0.678 m/s (ac) = rx = (0.025 m) - 25.13 ral/5); (at) = 0.628 m/52 (ac) = ru; = (0.025 m)(25. B radk)2; (ac) = 15.79 m/s

a = 15.80 m/s2 a=(a)+(ac)=(0.628-15)+(15.79 +15)









FIND: (a) ANG. VELOCITY OF BING WB

(b) ACCELERATIONS OF POINTS SHAFT AND RING WHICH ARE IN CONTACT.

PROB. 15.26: IN TERMS OF WA, T, T2, AND Y3.
PROB. 15.27: WHEN WA = 25 rodb, Y, = 12mm,

T2= 30 mm, AID Y3= 40 mm ALSO, FIND ACCEL. OF POINT ON OUTSIDE OF B.



PROB. 15,26
(a) AT POINT OF CONTACT

TWA = 12WB WB - TE WAY

(b) ACCEL OF POINTS OF CONTACT

SIMFT A: $a_A = r_1 \omega_A^2$ RING B: $a_B = r_2 \omega_B^2 = r_2 \left(\frac{r_1}{r_2}\omega_A^2\right)^2$

28 = 12 WA

ACCEL OF POINT D ON OUTGIOE OF PLANE

 $a_{D} = r_{3} \omega_{B}^{2} = r_{3} \left(\frac{r_{1}}{r_{2}} \omega_{A} \right)^{2}; \quad a_{D} = r_{3} \left(\frac{r_{1}}{r_{2}} \right)^{2} \omega_{A}^{2}$

PROB. 15.27 WA= 25 rad/s, Y,= 12 mm
Ya= 30 mm, Y3= 40 mm

(a) WB = \frac{\gamma_1}{\gamma_2} w_A = \frac{12 mm}{30mm} (25 rad/s); WB = 10 rad/s)

(b) $a_A = r_i \omega_A^2 = (12 \text{ mm})(25 \text{ rad/s})^2 = 7.5 \times 10^3 \text{ ma/s}^2$ $a_A = 7.5 \text{ m/s}^2$

ag= \frac{\range 1^2}{\range 2} \omega_A = \frac{(12 mm)^2}{(30 mm)} (25 rad/s) = 3 \tilde{10} mm/s^2

 $a_{D} = r_{3} \left(\frac{r_{1}}{r_{2}}\right)^{2} \omega_{A}^{2} = (4\omega_{mm}) \left(\frac{12 \text{ mm}}{30 \text{ mm}}\right)^{2} (25 \text{ rad/s})^{2}$ $a_{D} = 4 \times 10^{3} \text{ mm/s}^{2}$

a0=4m/s2

15.28



GIVEN:
WHEN \$ = 9 M = 9 M |
BRAKE IS AFFLED
AND BLACK CAMES
TO REST AFFLED
MOVING 18 FL.
ASSUMING UNIFORM
MUTION, FIND:
(a) OL OF DRUM
(b) TIME TO
COME TO REST

BLOCK A:

 $N^2 - V_0^4 = 2aS$ $O - (9ft/s)^2 = 2a(18ft)$

a=-2.25 fWs2; a= 2.25 fys2

DRUM:



 $V_A = r\omega_0$ $9 \text{ ft/s} = (0.75 \text{ ft})\omega$ $\omega_0 = 12 \text{ ind/s}$

a=rx

-(2.25Pt/s2) = (0.75Ft) X

d=-3rad/s2; d=3rad/s2)

W= Wo + at: 0= (12 red/s)-(3 red/s=)Z,

2,=45

15.29



WHEN t=0, T=0.

WHEN t=5, BLOCK

HAS MOVED 16tt ASSUMING UNIFORM

MOTION, FIND:

(a) d or DRUM

(b) W OF DRUM

WHEN t=45

BLOCK A:

 $5 = \sqrt{5t + \frac{1}{2}at^2}$ $16ft = 0 + \frac{1}{2}a(5s)^2$

a=+1.28 ft/s2 a=1.28 ft/s2

DRUM:

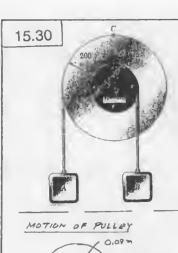


UNIFORM MOTION W= 0 WHEN &= 0

W=Wotat

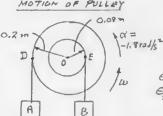
when t = 45: $\omega = 0 + (1.707 \text{ rad/s}^2)(45)$ $\omega = 6.83 \text{ rad/s}$

W=6.83 rad/s)



GIVEH: FOR PULLEY. Wo= 0.8 mad/s). d = 1. Brad/5").

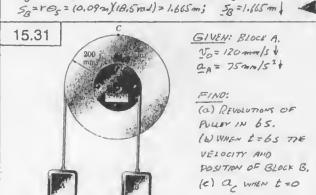
PINO: WHEN t= 55. THE VELOCITY AND POSITION OF 10) BLOCK A. 16) BLOCK B.



UNIF. ACCEL. MOTION rw=wo+at a=+(0,8 rad/s)-(1.8 rad/s2)(5s) w==-8.2 rad/s W= 8.2 rad/s ? 0=a+wst+ 2dt2

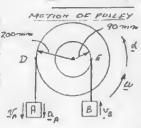
62=0+(0.8 mys)(55)- = (1.8 rod/s) (55) 0,=-18.52ad 0=18.5 neil]

NA = No = r Ws = (0.2m)(8.2 rolfs)=1.64 m/s; VA=1.640 m/s S= + Os= (0.2 m)(18, 5red) = 3.70 m; S= 3.70 m BLOCK B: N= 1= FW== (0.09m)(&2 red/s)=0.738m/s; N=0.738m/s+



GIVEN: BLOCK A. V= 120 mm/s V an = 75mm/52+

(a) REVOLUTIONS OF PULLEY IN 65. (WHEN Z = 65 THE VELOCITY AND POSITION OF BLOCK B. (e) a WHEN t=0



UNIF, ACCEL. MOTHON No= + Was 120 ma/s = (200 ma) W ub= 0.6 + ad/5) axra, 75mm/s = (200 ma) V 0=0.375 m= 1/3

Ter laspuller: WHEN t= 65 (a) 0= 0+ wot + fat 2 @= 0+(0.6 red/s)(5s)+ \(\frac{1}{2}\)(0.37500d/2)(6s) 6=10.35 rad

0=10.35 rad (= 1.647 rev

a = w + at = 0.6mlb + (0.375 rad/s)(65); W= 2.85 rad/s)

BLOCK B: WHEN t= 65 VB= rw= (90, mm)(2.85 rod/s) UR= 256 mm / 1 58=932 mm + Se=10=(9 = 1)(10.35 nod) POINT C, WHEN E=0 at=rx=(20000, 1375 radt)=75 mg/5 -] =104.0 mg/5 743.6 02=10 = (200 mmy 3.6rad/s) = 7200/5=1

15.32

15.33

GIVEH: WHEN E = 0, (4a) = 450 rpm 7 (Q) = D AFTER SLIPPAGE, WHEN t=65 WA= 140 MM 2

FIND: DURING SLIPPAGE DIA AND DIB

DISKA: (Un) = 450 rpm = 47.124 madk] INNEH 1=6: WA = 140 rpm = 14.661 mads) WA = (WA) + WA t 14.661 nod/s = 47.124 val/s + 0/A(65) 0 = 5.41 rad () da= -5.41 mad/s

DISK B: WU = 0 WHEN E=651 (END OF SLIPPALE) H Tawa = Ygwg! (3in) (19.661 mal/s) = (5in) (WE) Wg = 8.796 naul/s) WR=(W)+CART 3.796 nod/s = 0+00 (65)

OLR= 1.466 rad/s") Olg=1,466 mad/52

GIVEN: DISKA: (WA)=50UTPS) WILL COAST TO REST IN 60s DISKB: (Wa) =0 dB= 2.5 rad/s2) FIND: (a) WHEN DISKS CAN BE BROUGHT TOGETHER WITH NO SLIMME (b) FIMAL WA AND WB.

DISK A: (WA) = 500 rpm = 52, 36 rad/s 7 DISK A WILL COAST TO REST IN 605 WA = (WA) + dA t; 0=57.36 mod/s + dA (600 da= -0,87266 rod/s AT TIME t:

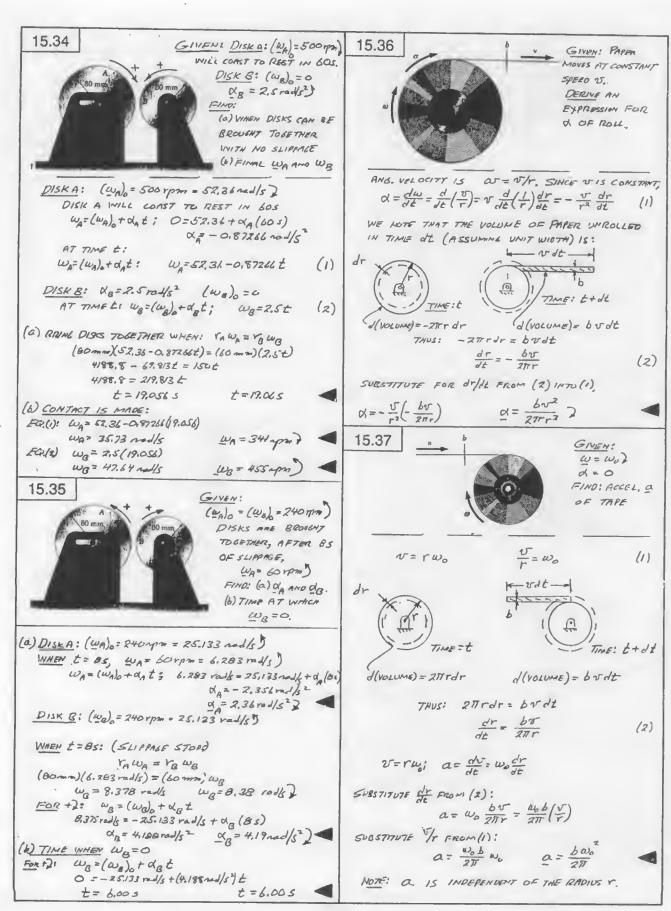
4x=52.36-0.97266 t (1) WA = (WA) + dAt;

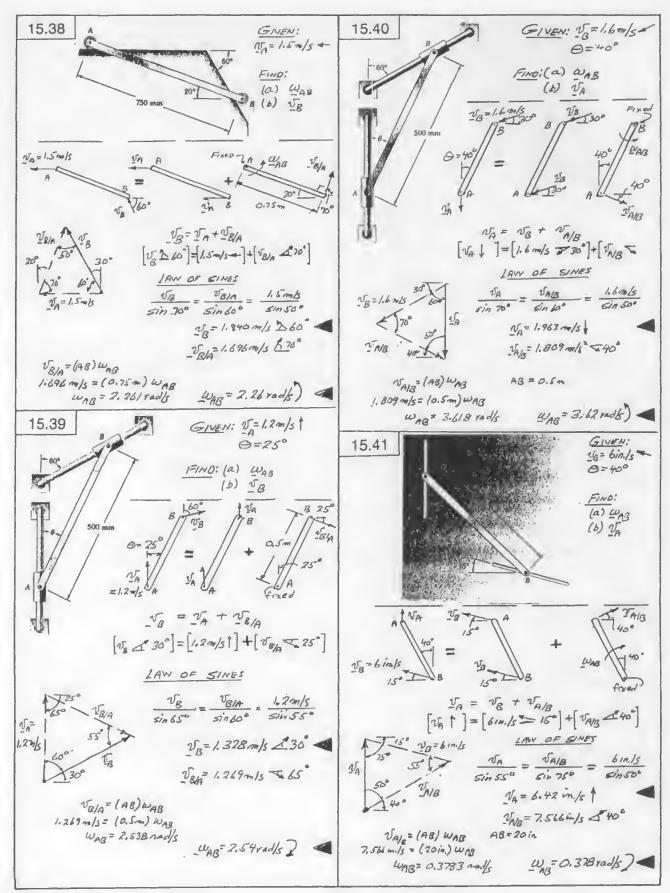
DISK B: dB=2.5 ned/s2 (WB) =0 WB = 2.5 t (2) AT TIME E: Wa= (Wa) + dati

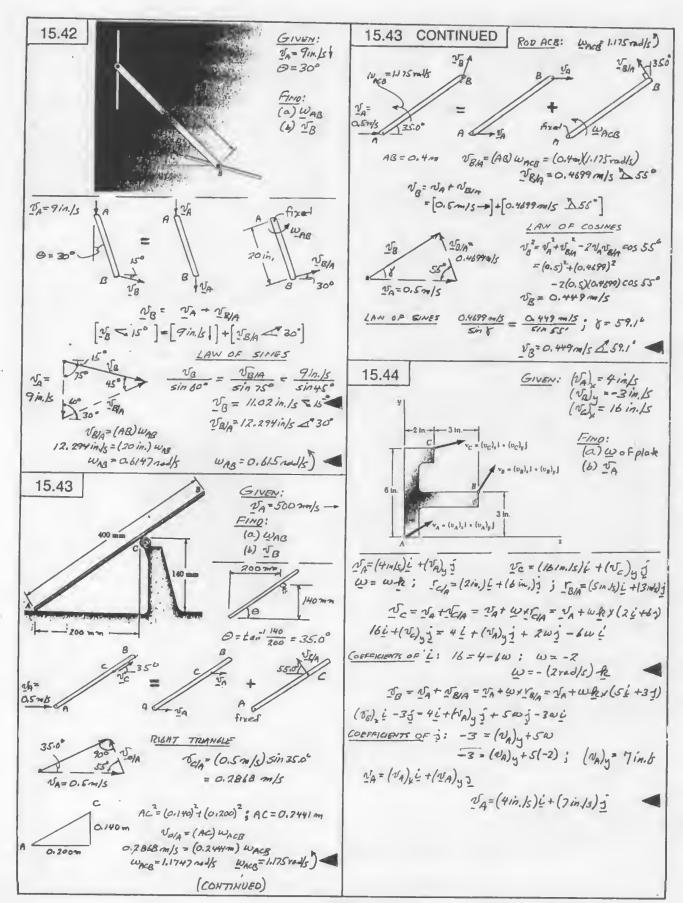
(a) BRING DISKS TOGETHER WHEN: YAWA - YBWB (3 in (52.36-0.877/1t)=(510.1)(2.5t) 157.08 = 2.618t = 12.5t 157.08 = 15.118t t=10.395

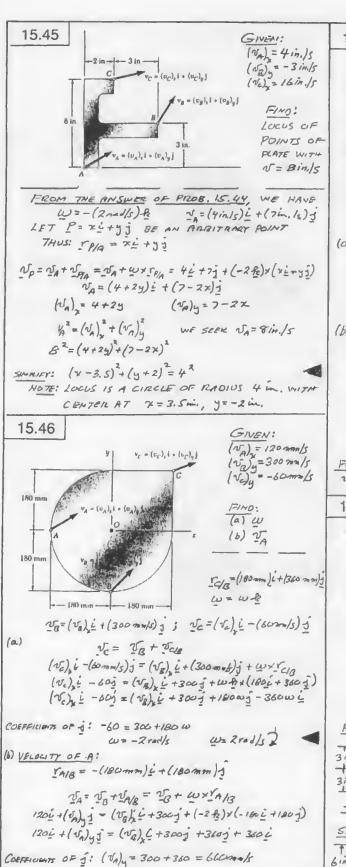
(b) WHEN CONTACT IS MADE (t=10.395) EQ.(1): Wa= 57.36 -0.87266(10.39) WA = 43,29 rad/5 Wa = 413 11000 }

E0(2): WB = 2.5(10.39) WB=24AApm) WR= 25,975 redls

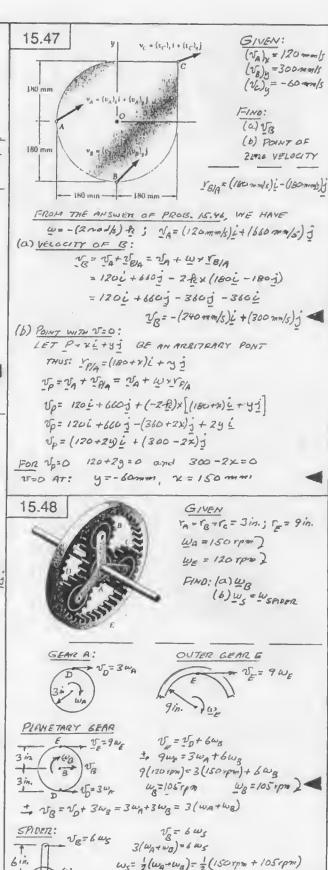






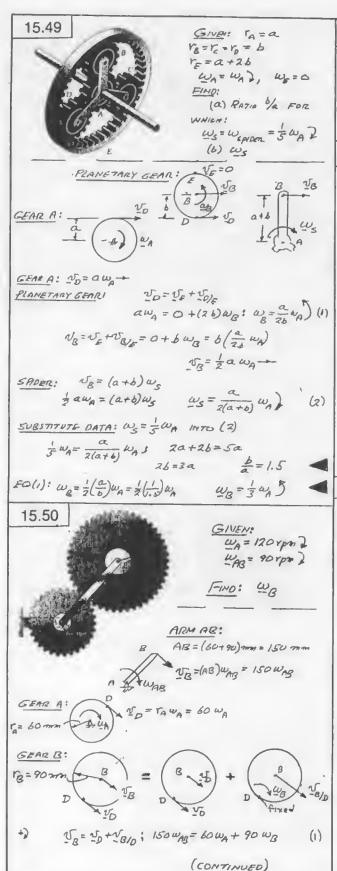


VA= (120 mak) & + (660 mm/s) 5



W,=127,5 rpm

W=127.5 rpm)



15.50 CONTINUED

EQU): +3 150 WAR = 60 WA + 90 WB

DATA! WA = 120 rpm ? WAG= 90 mm } 150(90 rpm) = 60(120rpm) +90 avg UB=+70rpm WB= 70 (64)



GIVEN: WAB = 42 mpm 2 FIND: (a) WA FOR WHICH wass 20 mm) (b) WA FOR WHICH WB=0 (CURVILINEAR TRANSLATION

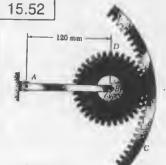
SEE FIRST PART OF SOLUTION OF PROB 15.50 FOR DERIVATION OF (1)

+) 150 WAB = 60 WA +90 WB

(a) For w8=20 m), mB=-20 Mm EQ(1): +) 150(42 mm) = 600, + 90(-20 mm) WA= 135. pm] WA= + 135 mpm

(b) FOR WB=0:

EG(1): +2 150 (42 rpm) = 60 WA +0 ma= 105 rpm) WA=+105 mpm



GIVEN: WAR = 20 md/s) FINO:

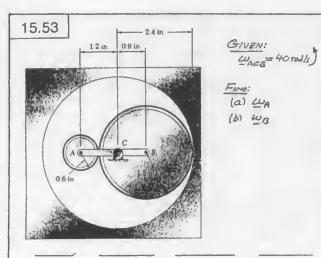
(a) wa (b) VD

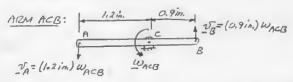
ARM AB A VB= (120mg/201/k) = 2,4m/st

GEAR B: UB= 2.4 m/s

V8= N5+N8/8= 0+(BE) W8 (a) BE=0.05m:: 2.4 m/st = 0 + (0.05 m) wg t We = 48 radis WB=48 rad/s)

(b) DE = (0.05 VI): TD = VE + VOIE = O + (DE) WB V= 0+ (0.05 V2)(48) Vp= 2.39 mls Up = 3.39 m/s 145° €



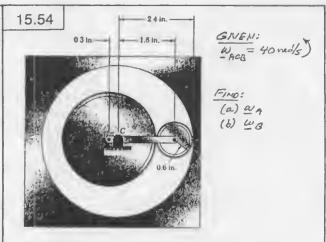


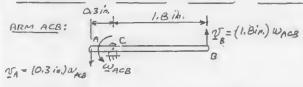
DISIC ROLLS ON D:
$$V_D = V_B + V_{DIB} = V_B + (BD) w_B$$

+1 $O = (0.9 \text{ ii.}) w_{ACB} - (1.5 \text{ ii.}) w_B$
 $w_B = 0.6 w_{ACB} = 0.6 (40 \text{ vad/k})$
 $w_B = 24 \text{ vad/s}$

DISK A

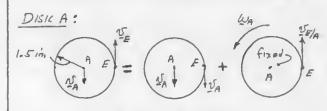
$$U_{A}$$
 U_{A}
 U_{A}





Disk ROLLS ON D:
$$\nabla_D = V_B + \nabla_{D/B} = V_g + (60) w_B$$

+1 $O = (1.8 \text{ in}) w_{ACB} - (06 \text{ in}.) w_B$
 $\dot{w}_B = 3 w_{ACB} = 3 (40 \text{ rad/s})$
 $w_B = 120 \text{ rad/s}$

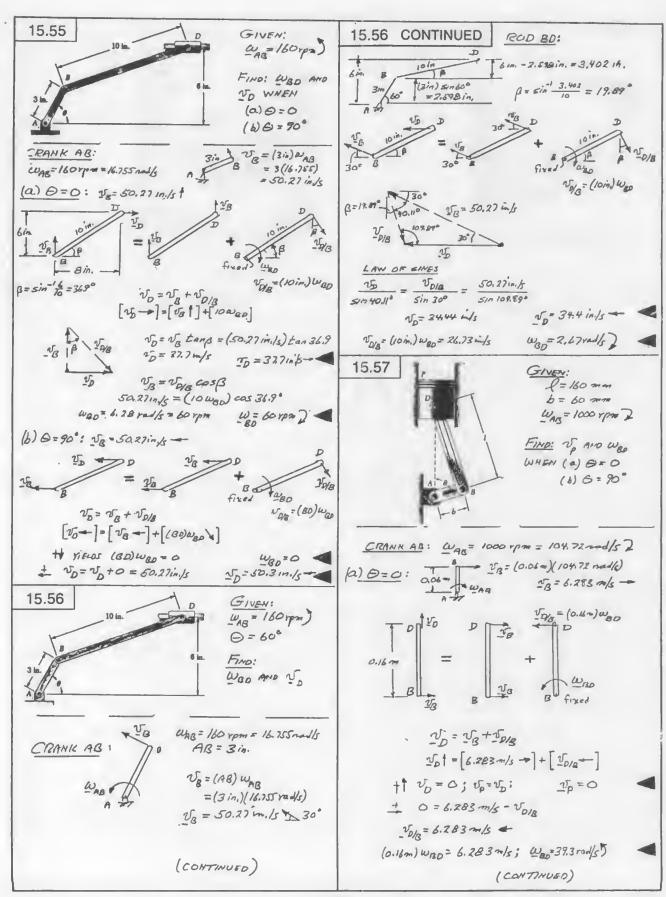


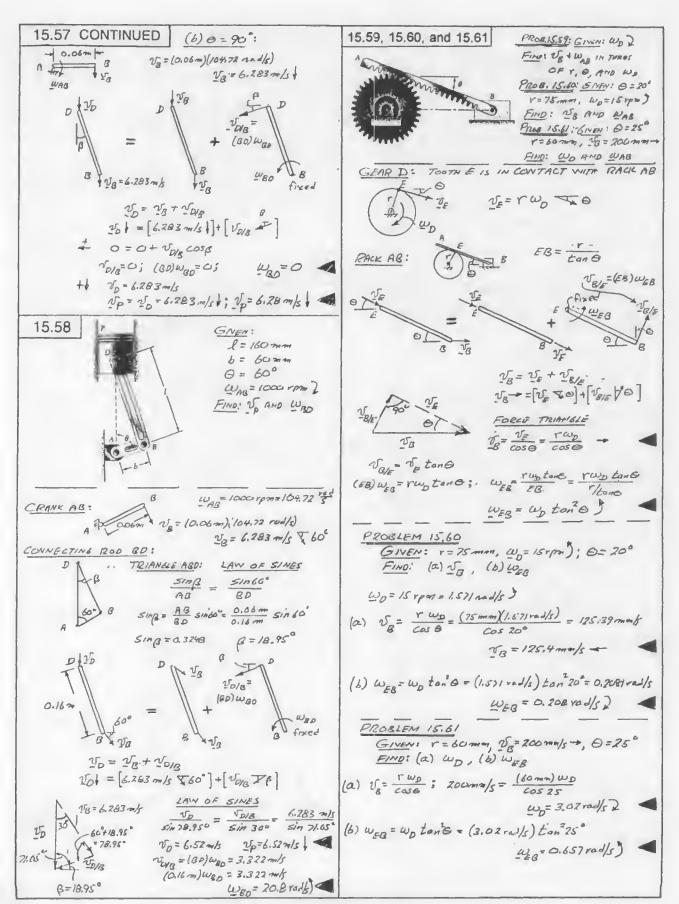
$$\tilde{V}_{E} = \tilde{V}_{A} + \tilde{V}_{EA} = \tilde{V}_{A} + (AE) \omega_{A}$$
+) $(3.6i...) \omega_{ACB} = -(0.3 i...) \omega_{ACB} + (1.5 in.) \omega_{A}$

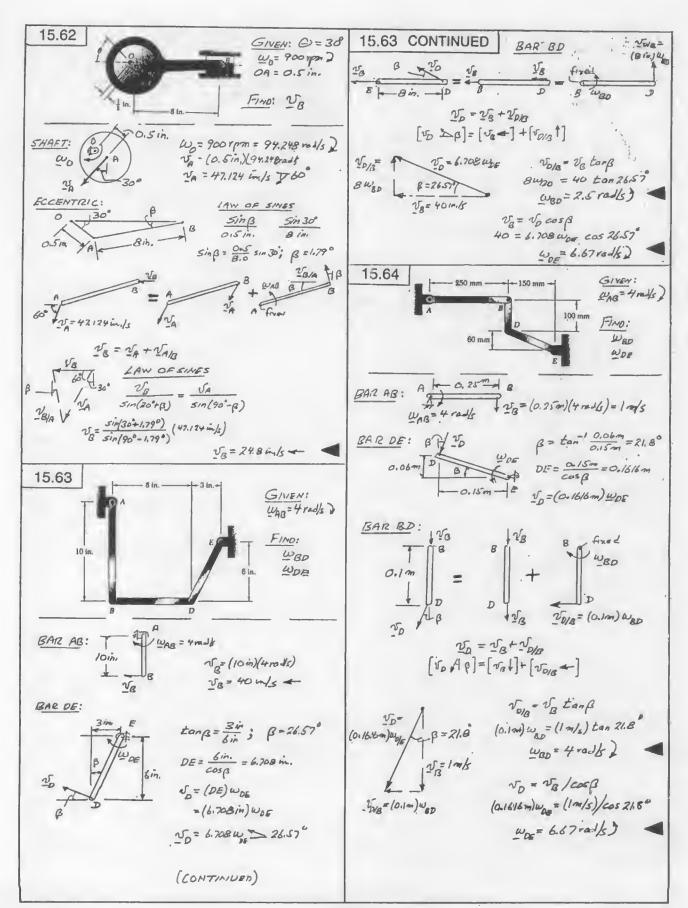
$$\omega_{A} = \frac{3.6 + 0.3}{1.5} \omega_{ACB} = 2.6 \omega_{ACB}$$

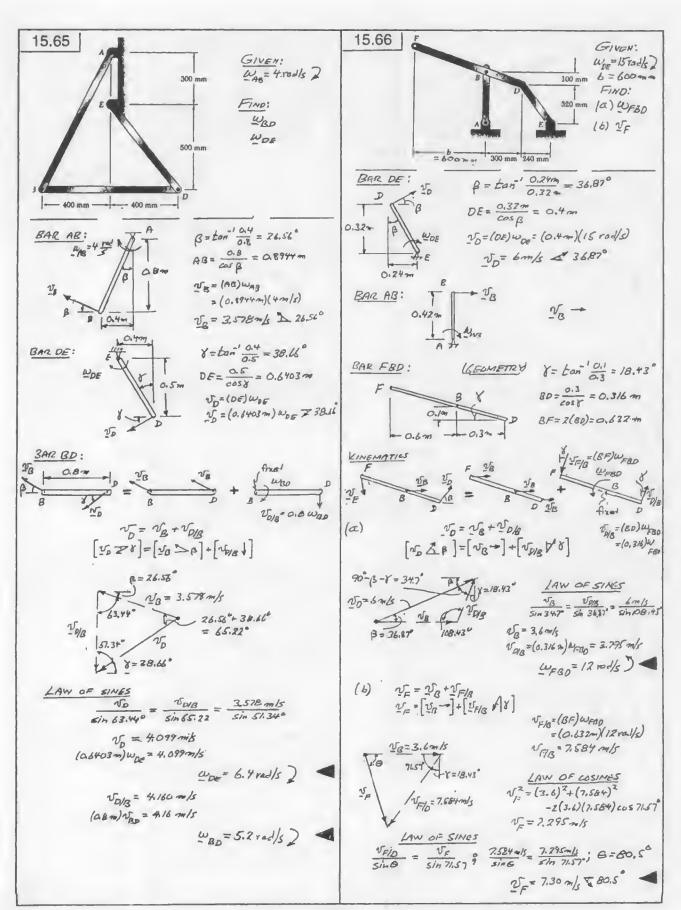
$$\omega_{A} = 2.6 (40 \text{ rad/s}) = 104 \text{ rad/s}$$

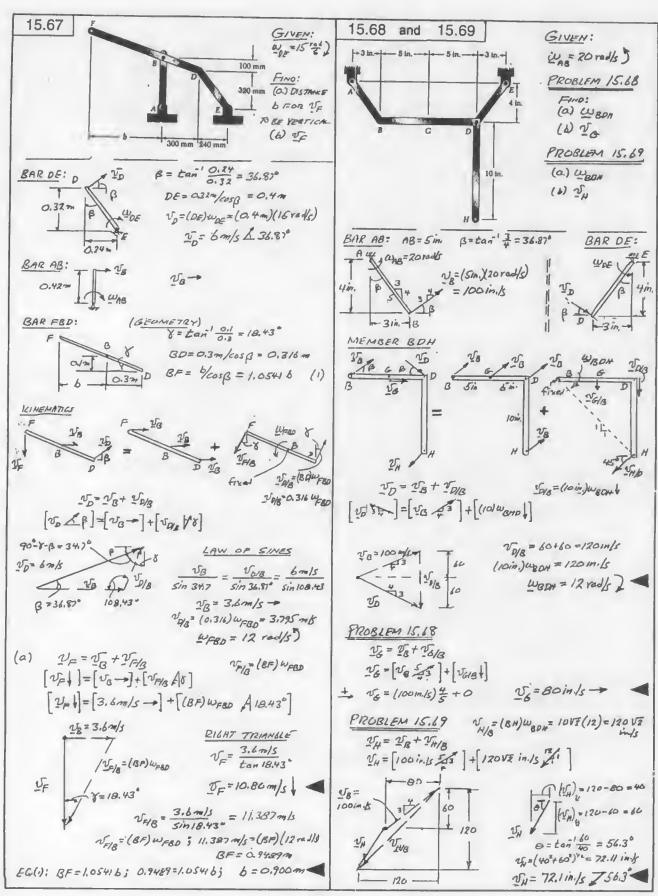
$$\omega_{A} = 104 \text{ rad/s}$$

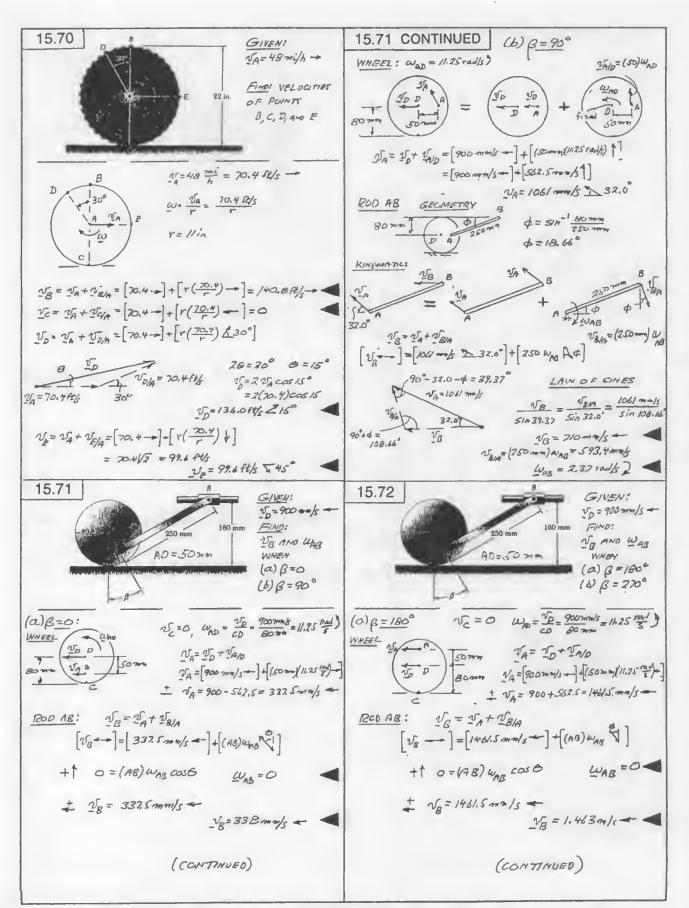


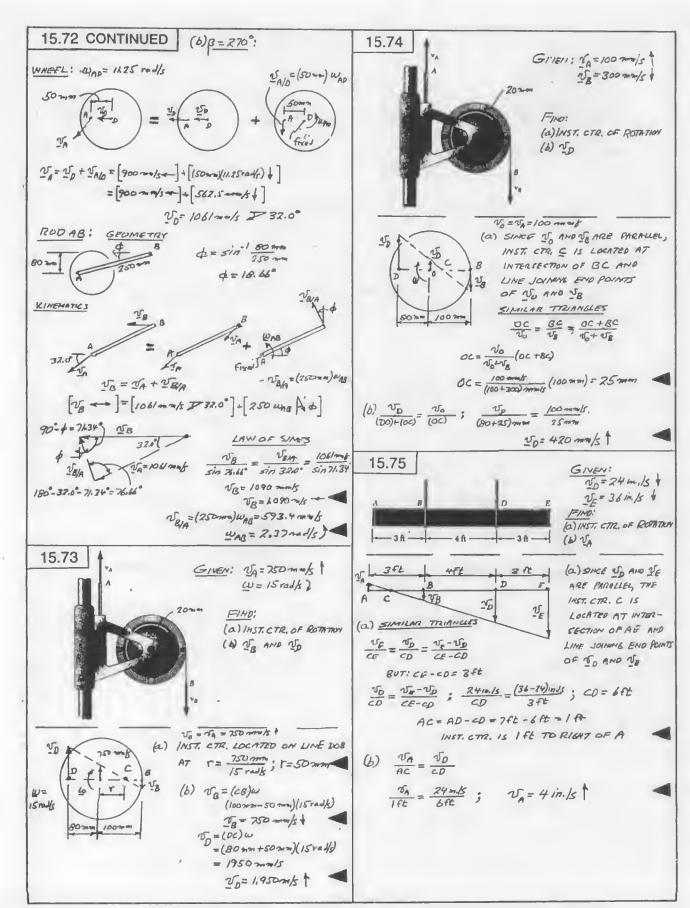


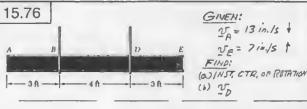


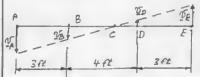












(a) SINCE NA AND NE ARE
PARALLEL, THE INST.

CTR. C IS LOCATED

AT INTERSECTION OF

AE AND LINE JOINING

END POINTS OF NA AND NE

(a) SIMILAR TRIANGLES

$$\frac{AC}{N_A} = \frac{CE}{N_E} = \frac{AC + CE}{N_A + N_E}$$

$$AC = \frac{v_A}{v_A + v_E} \left(AC + cE \right) = \frac{13i \cdot 1/5}{(13+7)i \cdot 1/5} \left(10ft \right) = 6.5 ft$$

CD = ND - AC = 7ft - 6.5ft = 0.5ft

INST. CTIR. IS O.S FE TO LEFT OF D

(b)
$$\frac{\mathcal{V}_D}{CD} = \frac{\mathcal{V}_A}{AC}$$
; $\frac{\mathcal{V}_D}{0.5R} = \frac{13 \text{ in. b}}{6.5 \text{ R}}$; $\mathcal{V}_D = J \text{ in./s}$

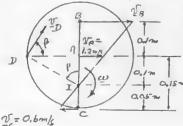
15.77 $v_{A} \approx 1.2 \text{ m/s}$ $r_{1} \approx 150 \text{ mm}$

SIVEN:

VA=1.2 m/s -
VELOCITY OF LOWER

RACK IS V=0.6 m/s -
FIND: (a) W

(b) VR AND VO



SINCE NA AND NO.

ARE PARALLEL THE

MIST. CTA. OF ROTATION

IS AT THE INTER
SECTION OF BC AND

THE LIME JOINNA THE

END POINTS OF

NA AND NO.

(a) ANGUAR VELOCITY 2 = (A1) W
1.2 m/s = (0.15m) W
W= 12 radf]

(b) UPPER RACK VR = VB = (BI) W

VR = (0.2m)(12 red/s)

VE = 2.4 m/s -

VELCUTY OF POINT D: $\beta = \tan^{-1} \frac{0.15 \, m}{0.1 \, m} = 563^{\circ}$ $DJ = \frac{DA}{\cos \varrho} = \frac{0.15 \, m}{\cos 56.3^{\circ}} = 0.1603 \, m$

VD= (DI)W VD= (0.1803~)(12 ralls) VD= 2.16 m/s 8.56.3 15.78 200 mm/s

COVERN INNER RABUS = 30 mm

COUTER PARIUS = 60 mm

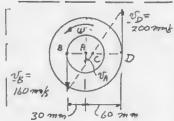
PINO: (a) INST. ETT., OF ROTATION

(b) IBLOCK = VA

(c) LENGTH OF CORD

WRAFPED OR UNWRAPPED PER SECOND

ON EACH PULLEY.



(a) SINCE I'S MO OF ARE PARALLEL, INST. CTR. IS LOCATED AT THE INTERSECTION OF BE AND LINE JOINING END POINTS OF YET YOU

BC = CD = BC+CD ; BUT BC+C0= 90 mm

160 = 70mm; BC=40mm; AC=BC-AB=40mm-30mm=10mm
160 = 360; BC=40mm; AC=BC-AB=40mm-30mm=10mm

b) VBLOCK = VA W= 16 = 160 mm/s w=4 rods)

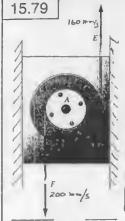
NA=(AC) W=(10mm)(4rolls)-4Kmg/ Nova 40mm/s

SINCE Not AND WAL, CORD IS UMWEATHED AT RATE (NA+VO)/IS

NA+VO=40+200=240 ma/s; 240 mm, UMMARTED/S

IMMAR POLLEY: Nat >NAt, CORD IS UMMRAPED AT RATE(No-VA)/S

WHEN PULLEY: VE > VA +, CORD IS UMMRAPPED AT RATE (VE VE - VA = 160-40 = 120 mm/s; 120 mm, CHWRAPPED S



FIND: (a) INST. CTR. OF ROTATION
(b) ISLOCK = VA
(4) LENGTH OF COLD WIRMHED OIL

UNUIRAMED PER SECOND ON GACH PULLEY

GIVEN: INNER PULLEY = 30 mm

25 = 200 mg/s

(10mm

(a) SINCE & AND NO ARE AMPLIEL, INST. CTT2 IS LOCATED AT THE
INTERSECTION OF BD AND LINE JOINN'S END POINTS OF & AND NO.

BC CO BCCO; BUT BC+CD=90mm

BC 90-70 BC-CD ; BUT BC+CD=90mm

BC 90-70 BC-CD ; BC-EC-AG-SU'M-30mm=20mm

BC = 50 mm; AC=BC-AB = 50 mm = 20 mm =

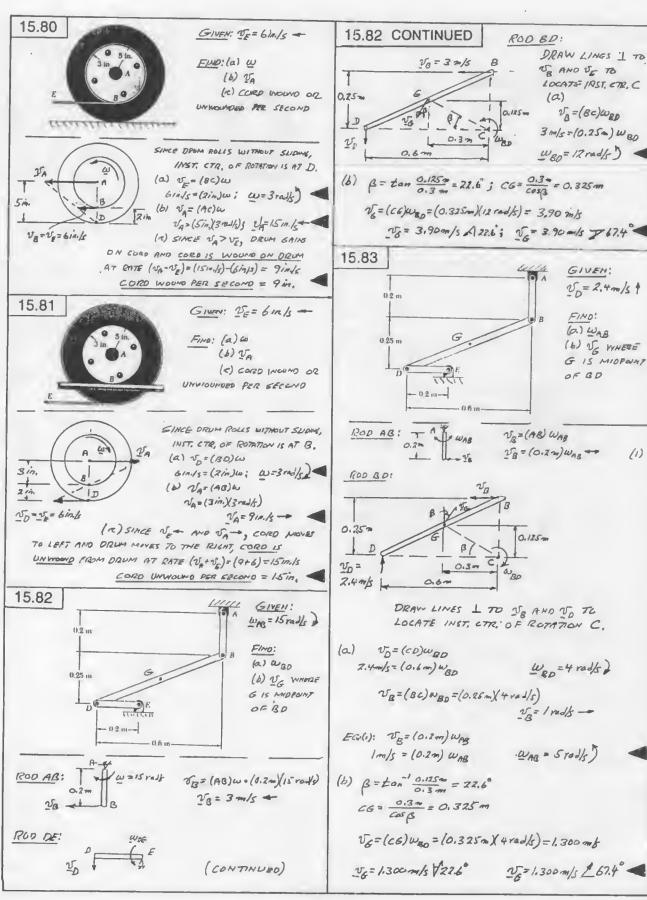
(b) \$\int_{\text{Blog} = \text{Va}} : \omega = \frac{26}{8} \text{BC} = \frac{200 mm/s}{150 mm}; \omega = 4 rad/s)\$

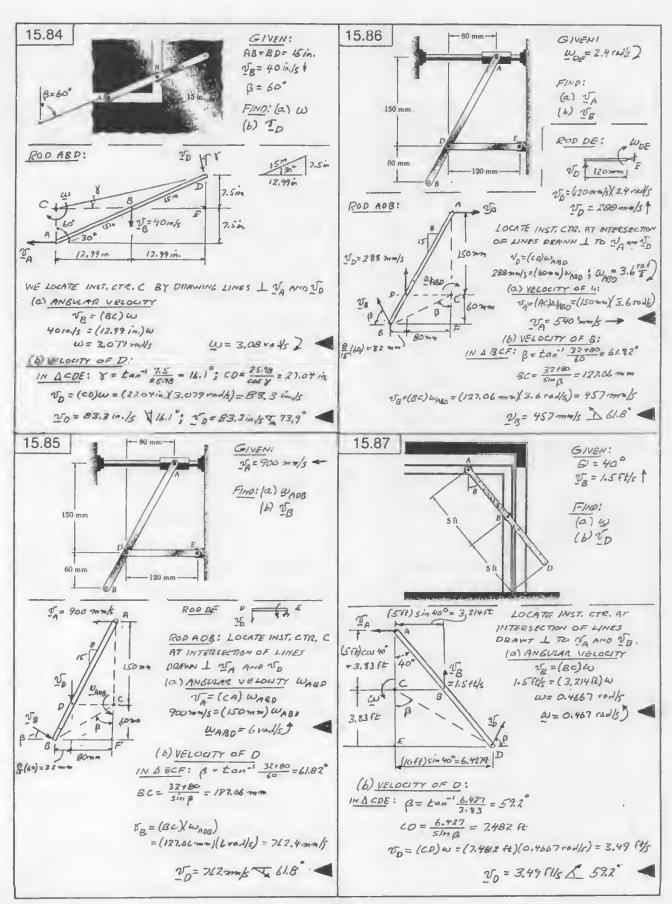
(1) OUTER PULLEY: Not AMONA & CORD IS UNWRAMER AT (1/4 MA)/S

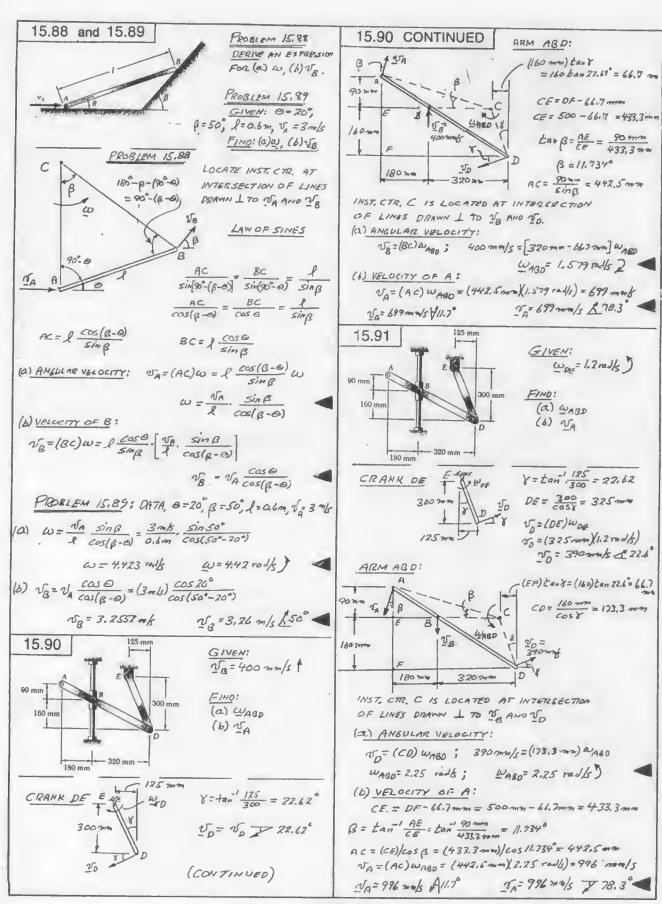
NOTURE 160+80= 240 mm/s; 240 mm, UNWRAMED/S

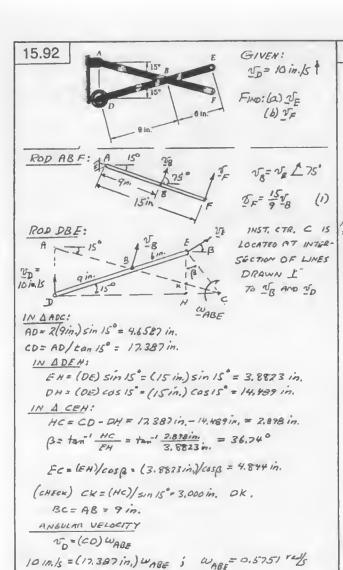
IMMER PULLEY: Not > NA +, CORD IS UNWRAMED AT (No - NA)/S

VB-V= 200-80= 120 mak; 120 mm, UNWRAPIAD/S





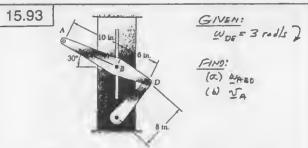


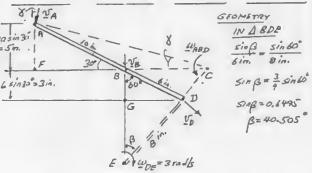


(a) VELOUTY OF E: $\sigma_E = (EC) w_{ABE} = (4.8 \text{ W} in X0.5751 \text{ red/s}) = 2.79 in./s$ $V_E = 2.77 in./s 236.7°$

(b) VELOCITY OF F;

 $T_B = (Bc) \omega_{ABE} = (9 in, \chi_0.575) \text{ rad/s} = 5.176 in. ls$ $EQ(i): \quad x_F = \frac{15}{9} x_B = \frac{15}{9} (5.176 in. ls) = 8.63 in. ls$ $x_F = 8.63 in. ls \stackrel{4}{\triangle} 75^{\circ}$





IN Δ 6 DE: FG = (OE) α Sβ = (Bin) cos β = 6.083 in.

IN Δ 8 CE: BC = (BE) tenβ = [86+FG] tenβ
= (3 in. + 6.083 in.) tenβ = 7.759 in

EC = (BE)/cos β = (3 in. + 6.083 in.)/cusβ = 11.54/β in.

FB = (AB) cos 30° = (10 in.)/cos 30° = 8.660 in

FC = FB + BC = B, 660 in. + 7.759 in. = 16.449 in.

IN A AFC:
$$X = \tan^{-1} \frac{AF}{FC} = \tan^{-1} \frac{Sin.}{16.419in.} = 16.937.$$

$$AC = \frac{FC}{\cos X} = \frac{16.419in.}{\cos X} = 17.163in.$$

MEMBER ABD: THE INST. CTR. C IS LOCATED AT INTERESCTION OF LINES DRAWN I TO AND TO.

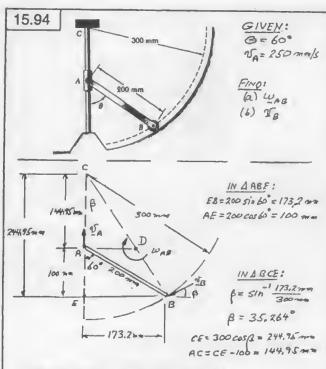
CD=EC-ED= 11.946 in. -8 in. = 3.946 in.

(0) ANGULAR VELOCITY WARD:

(b) VELOCITY OF A:

 $N_A = (Ac) \omega_{ABO} = (17.163 in.)(6.087 rad/c)$ $N_A = 104.4 in./s$ $N_A = 104.4 in./s$ | 1044 in./s | 16.9°

VA = 104.4 in./5 7 73.1°



THE INST. CTR. IS LOCATED AT POINT D WHICH IS THE POINT OF INTERSECTION OF LINES DRAWN I TO DA ATTO UB

(a) ANGULAR VELUCITY WAS ?

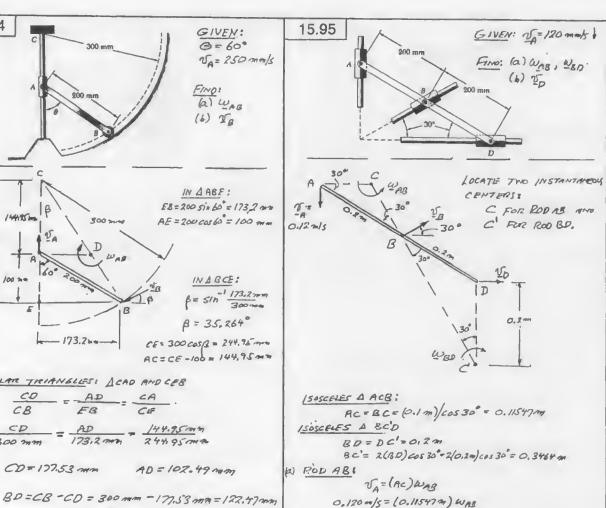
$$V_A = (AD) w_{AB}$$
 ?

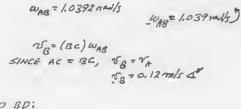
250ma/s = (102,49 mm) was

 $w_{AB} = 2.439 \text{ rad/s}$ $w_{AB} = 2.44 \text{ rad/s}$

(b) VELOCITY OF B;

$$V_B = (BD) \omega_{AB} = (122.47 nm) (2.439 ned/s)$$
 $V_B = 298.7 ma/s$
 $V_B = 299 ma/s 735.3°$

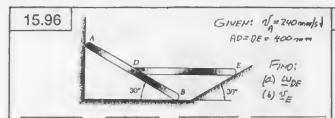


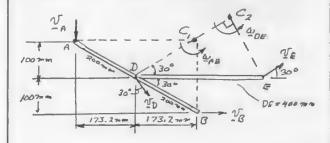


(b) VELOCITY OF D:

$$V_D = (DC') \approx_{BD}$$

 $= (0.2 \approx 10.3464 \text{ radf})$
 $V_0 = 0.06928 \text{ mass}$
 $V_D = 69.3 \text{ mass} \rightarrow$





WE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS:

C; FOR ROD AB, DRAW LINES I TO \$ + YE C; FOR ROD DE, DRAW LINES I TO SO SE

GEONETTZY: AC = (400 mm) cos 30° = 346,4 mm

BC, = (400 mm) sur 30° = 200 mm

DC, = AO = 200 mm

DC2=(DE) cos 30=(400 mm) cos 30° = 346,4 mm

EC2=(DE) SIN 30° = (400 mm) sin 30° = 200 mm

ROD AB: $V_A = (AC_1) \omega_{AB}$; 240 mm/s = (346.4 mm) ω_{AB} $\omega_{AB} = 0.69284$ ral/s)

VD=(DC,) WAB = (200 mm) (0.69284 md/s)
N=138.57 md/s A 30°

ROD DE: $T_0 = (DC_2) \omega_{DE}$ 138,57 mm/s = (346.4 mm) ω_{DE}

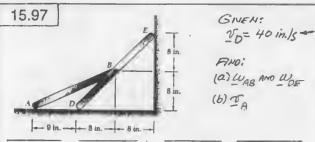
(a) Wor = 0.400 reds Wor = 0.4 rads)

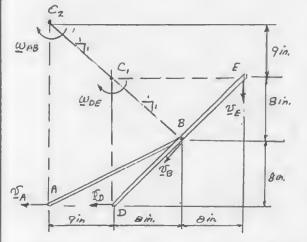
(b) V_E = (EC₂)W_{DE} = (200 mm)(0.400 rad/s)

V_E = 80 mm/s

V_E = 80 mm/s

L 30° ✓





NE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS:

Cy: FOR RUD DE, DRAW LINES I TO UD AND YE

GEOMETRY: 'BC,= (8in.) VZ = 8VZ in.
0C,= 16in.

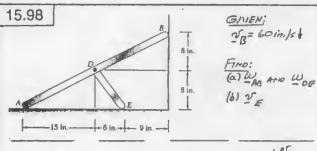
BC2= (9in + Bin)V2 = 17V2 in. AC2= 25in.

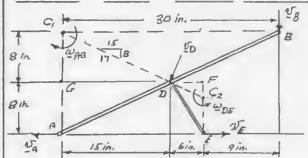
 $\omega_{DE} = 0.4 \text{ rad/s}$ \bullet (a) ROD DE: $\omega_{DE} = (DC_{i}) \omega_{DE}$ $40 \text{ in/s} = (16 \text{ in.}) \omega_{DE}$ $\omega_{DE} = 2.5 \text{ rad/s}$

T_B=(BC,) ω_{OE} =(8 V2 in.)(2.5 rod/s) T_B = 20 V2 in./s 7 45°

 $\frac{1200 \text{ AB:}}{20 \text{ Visin/s}} = (8C_2) \omega_{AB}$ $20 \text{ Visin/s} = (17 \text{ Visin}) \omega_{AB}$ $\omega_{AB} = \frac{20}{17} \text{ rad/s} = 1.1765 \text{ rad/s}$ $\omega_{AB} = 1.176 \text{ rad/s}$

(b) $V_A = (AC_2) \omega_{AB}$ = (25 in, X1.186 rad/s) $V_A = 29.41$ in./s $V_A = 29.4$ m/s





WE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS! C,: FOR 1200 AB DRAW LINES LTO VARIONS Cy : FOR ROD DE DRAW LINES ITE TO AND VE

GEOMETRY: OC, = (82+152) = 17 in. SINCE A GOG AND A DFC, ARE SIMILAR, CzF = CzD = 6in. 8in. = 17in. = 15ln.

> CoF= 3.2 in. C2 D= 6.8 in. EC,=8in,-C,F=8-3.2=4.8in.

NB=(BC,)WAB (a) RODAB: 60 in. 15 = (30 in.) WAB WAB = 2 rad/s WAB = 2 rad/s]

> VD=(DC)WAR VD = (17 in.)(2 rad/s) = 34in./5

1200 DE: ND = (04) WDE 34in/5 = (6.8in) WOE

WOE = 5 rad/s WOF = 5 rad/s)

T=(EC2) WOE (6) VE = (4.8in,)(5 rad/s) v= 24 in./s 5=24 in./s -> 15.99



GIVEN: AB = BD = 15 in.

DESCRIBE THE SPACE CENTROOF AND BOOY CENTROOF OF ROD ABD.

LET: AB = 1= 15 in.

SPACE CENTRODE: COORDINATES OF INST. CTR. 22+y2=12(cos26+sin26)

23 y2= -8" SPACE CENTROOF IS A QUARTER CIRCLE OF L= 15in, PAROIUS CENTERED AT INTERSECTION OF TRACES IN WHICH INHEELS A ANIO B MOVE

BODY CENTRODE: DRAW LINE CE WHICH CONNECTS INST. CTR. C AND POINT E LOCATED MIDWAY BETWEEN A AND B. SINCE CE = AE = 1 & = 7. Sin., WE NOTE

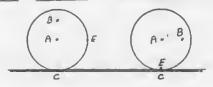
THAT BODY CENTRUGE IS A SEMICIACLE OF 7.5- m, RADIUS CENTERED AT E.

15.100



GIVEN: GEAR ROLLS ON STATIONARY LOWER IZACK.

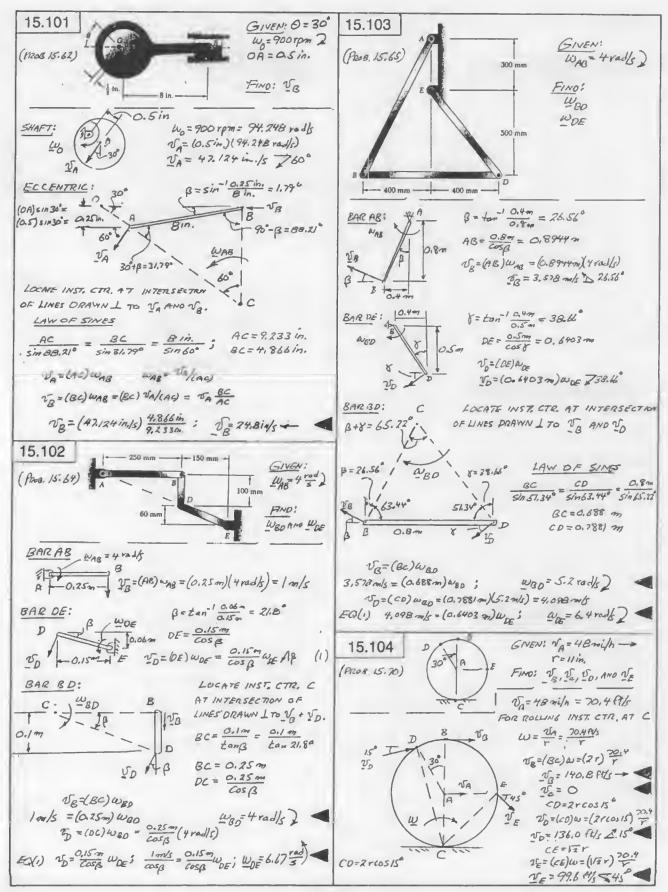
DESCRIBE THE SPACE CENTROOF AND BOOY CENTRODE OF THE GEAR.

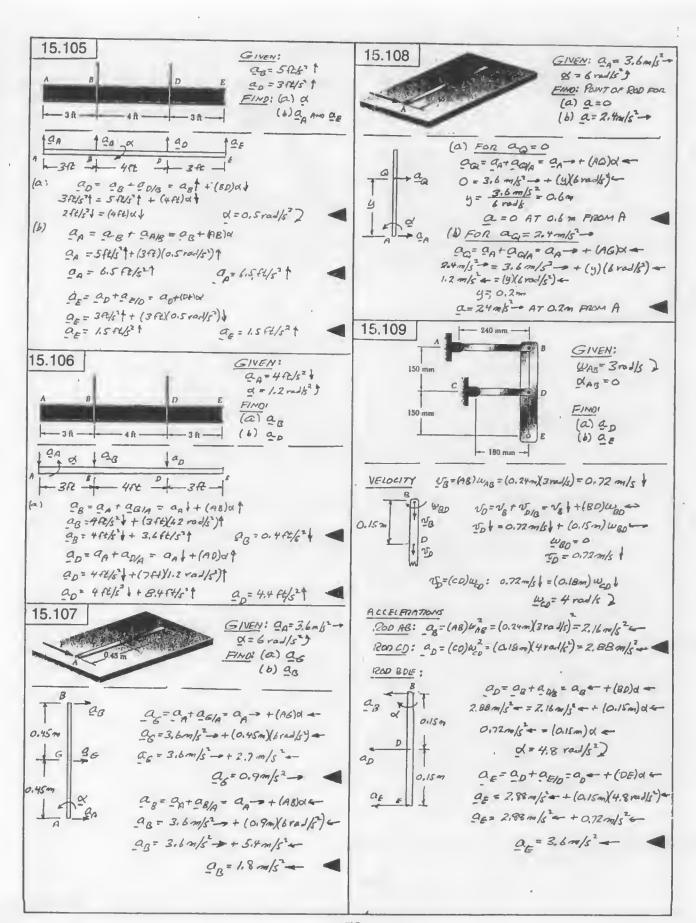


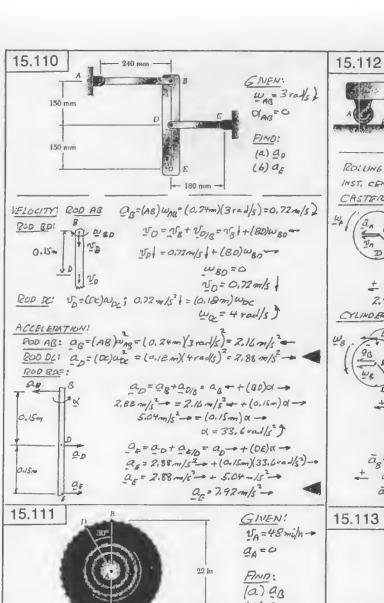
SINCE GEAR ROLLS ON LOWER RACK, THE INST. CTR. IS ALWAYS AT POINT OF CONTACT BETWEEN GEAR AND LOWER RACK.

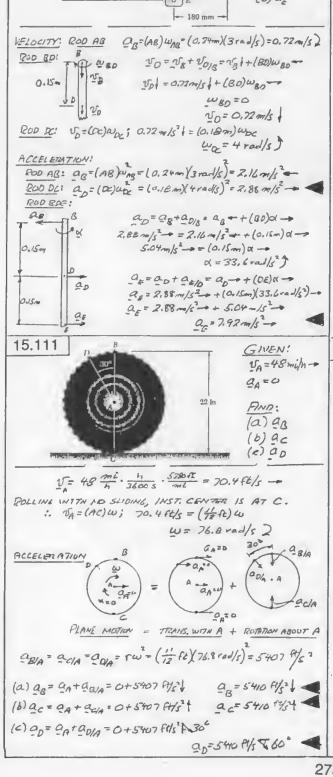
SPACE CENTRUDE: L'OWER RACK

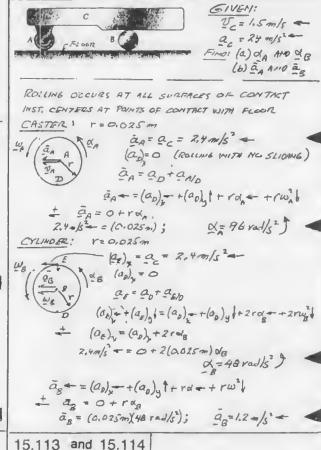
BOOT CENTRODE: CIRCUMFERENCE OF GEAR



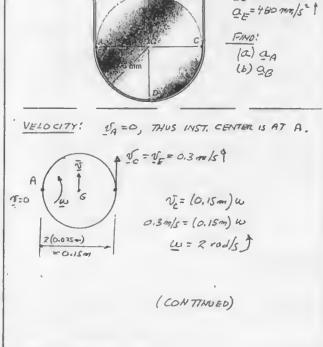




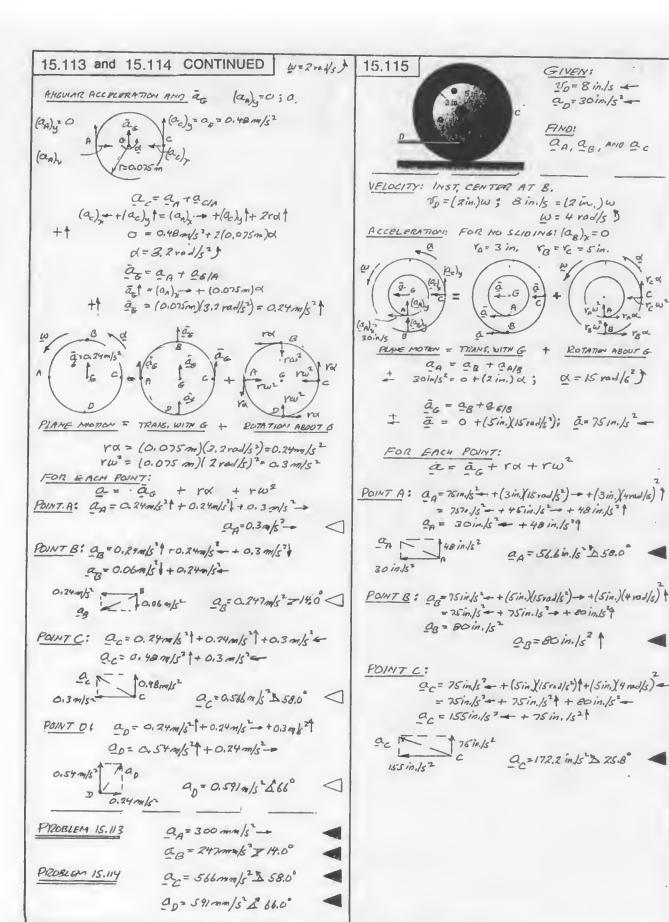




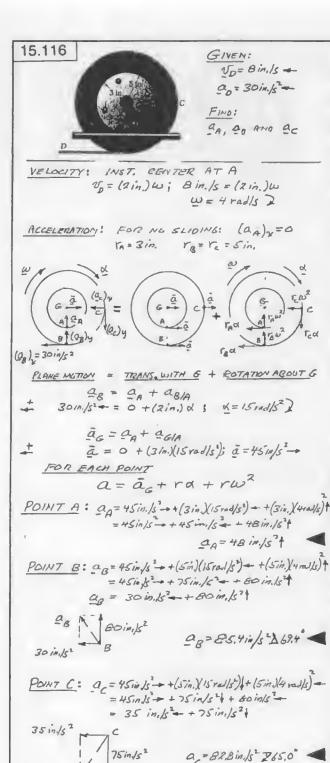
CASTER AND CYLINOFIL FACH OF 50- may DIAM.

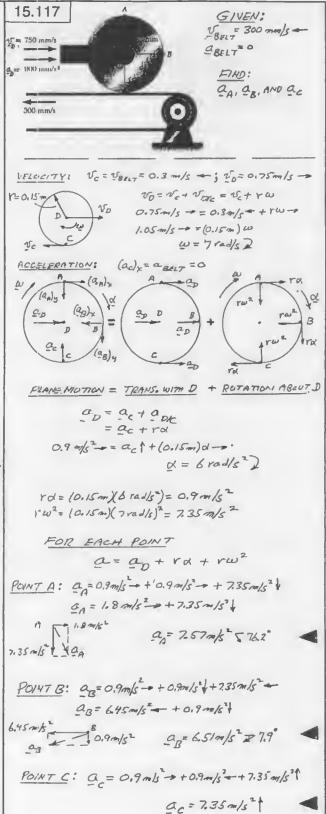


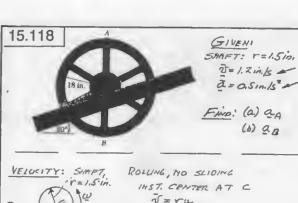
GIVEN: N== 300 ma/s 1

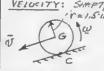


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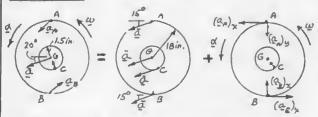




N=ra 1.2in./s = (1.5in.) W W= 0,8 rad/s)

ACCELERATION!

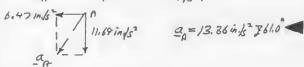
411



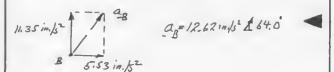
PLANE MOTION = TRANS. WITH G + RUTATION ABOUT G a= ac + a6/c a= 0+ rx 0.5in,/s2= 0+(1.5in.)d; x=1/3 rad/s2) + J 154

(a) POINT A: a=(0.511.6) Vro +(18in, / 'grad/2) -+ (18in, /0.8rad/s) = 0.470 in. 15 - +0.171 in. 15 + 611.152 + 11.52 in 153

an = 6.47 in/s2 + 11.69 in/s2+

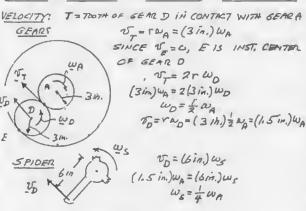


(b) POINT B: an= (0.5 in. /5) 720 + (18in.) (3 rad/5) = + (18in.)(0.8 rad/5) = 0.470 in./5 = +0.17/11./5 + 6 in/5 - + 11.5211./57 0B= 5.53 mils=+ 11.35 mils





GIVEN: ra= ra= re= 3in; (= 9in. Wa= 150 rpm), d=0 W=0 FIND: MACHITUDE OF ACCELERATION OF TOOTH OF GEAR D IN CONTACT WITH (a) GEAR A, (b) GEAR E.



Ws = 4 MA = 3.927 rad/s 2 ACLELERATION Ws= 3.927 radk SPIDER: a,=(A0) w=(6in. X3927 mods)

Wa= 150 rpm = 15.708 rad/s 2

wp = 2 wp = 7.854 rad/s)

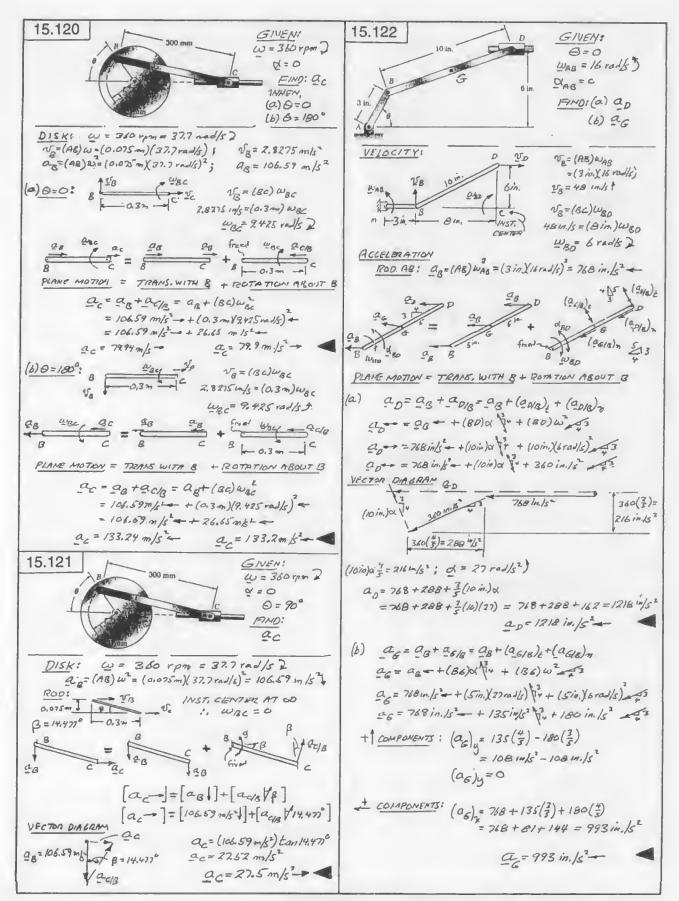
GEAR D:

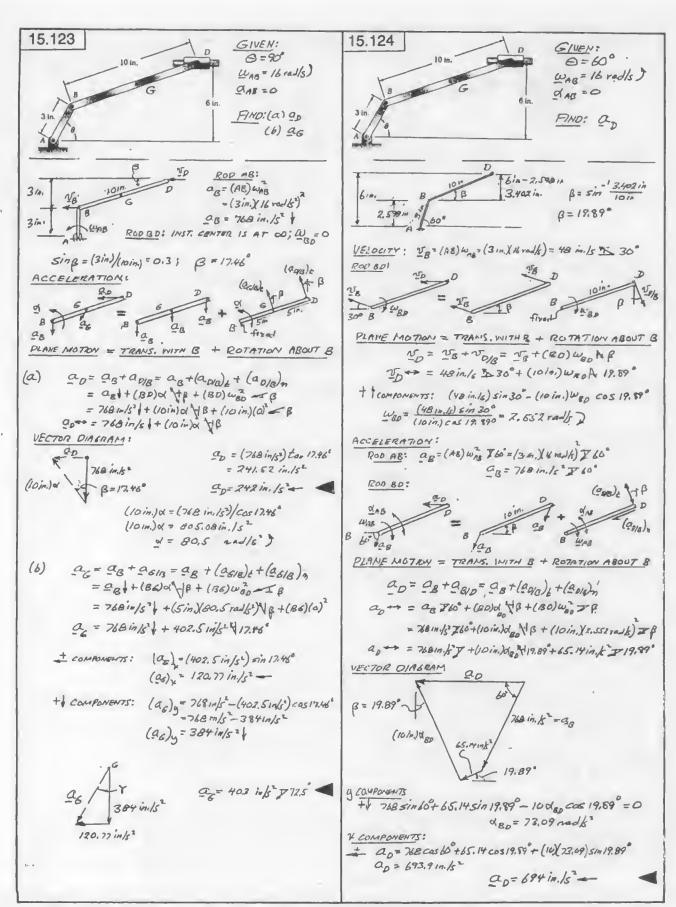
a = 92.53 in./s2/ Ú

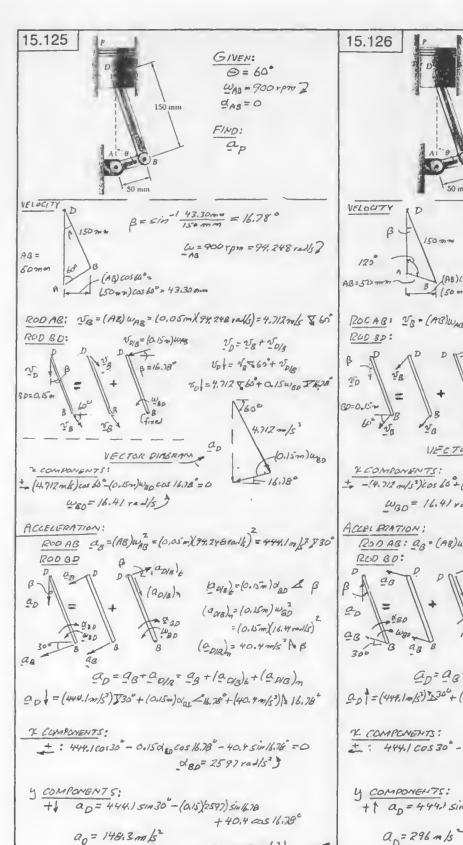
(a) TOOTH T IN CONTACT WITH GEAR A ay = 20 + a T/0 = ap + (07) wo = 92.53 in/s2 /+ (3in) (2.854 rad/s) 2 = 92,531m/s 1+ 185,06 cm/s 2 a= 92.53 in/52 a= 92.5 in./5

PLANE MOTION = TRANS, WITH D + ROTATION ABOUT D

(b) TOOTH E IN CONTACT WITH BEAR 5 a= a0+a= a0+ (20)00 = 92,53 in 15 / + (3 in,) 2854 rad/s) = 9253 in/5 10 + 185.06 in/5" a = 277.6 ln/5. a = 278 in/s2

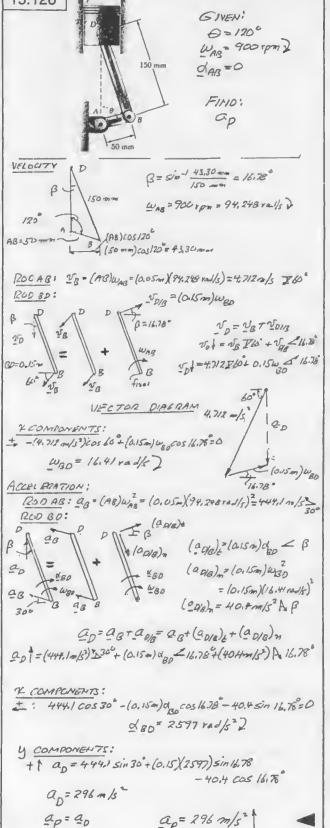


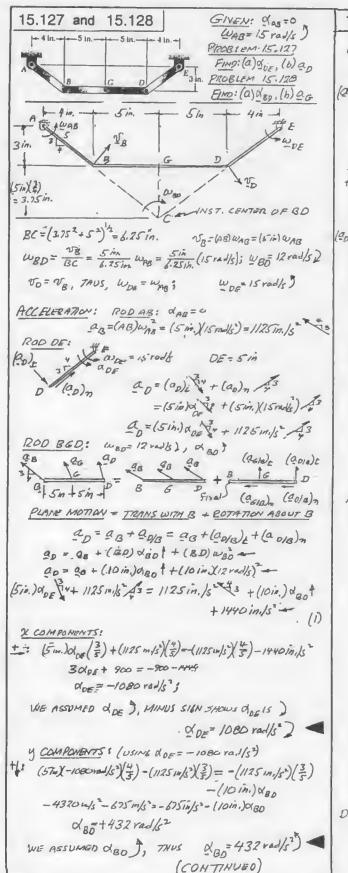




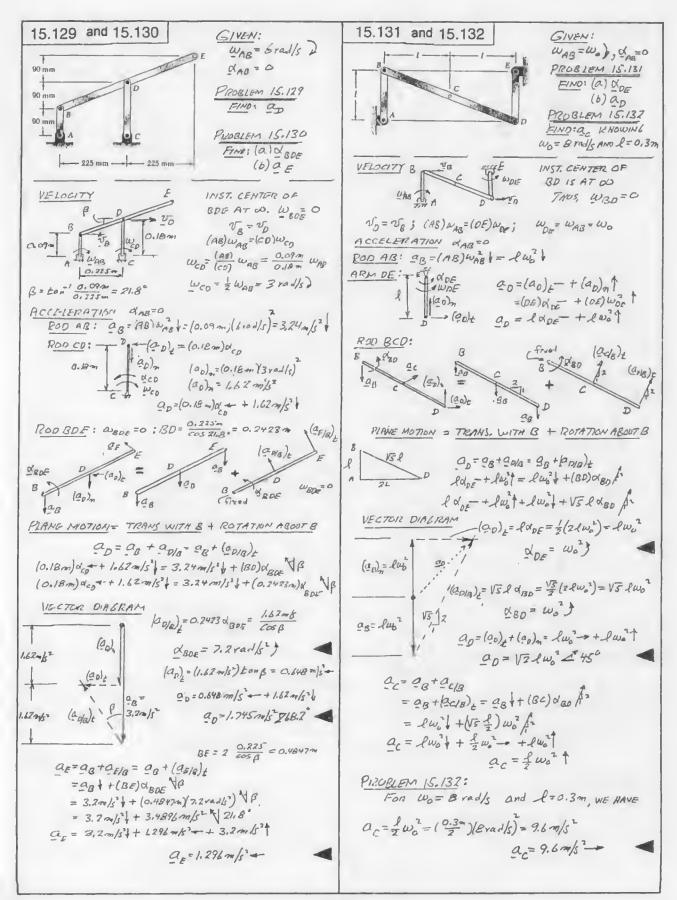
ap=148.3m/s2

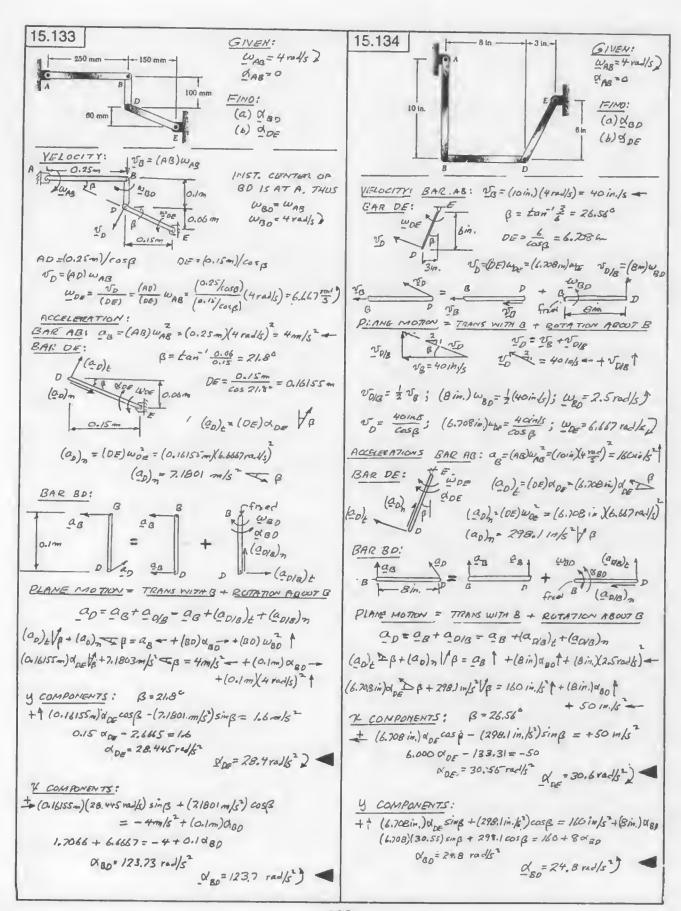
ap=ap

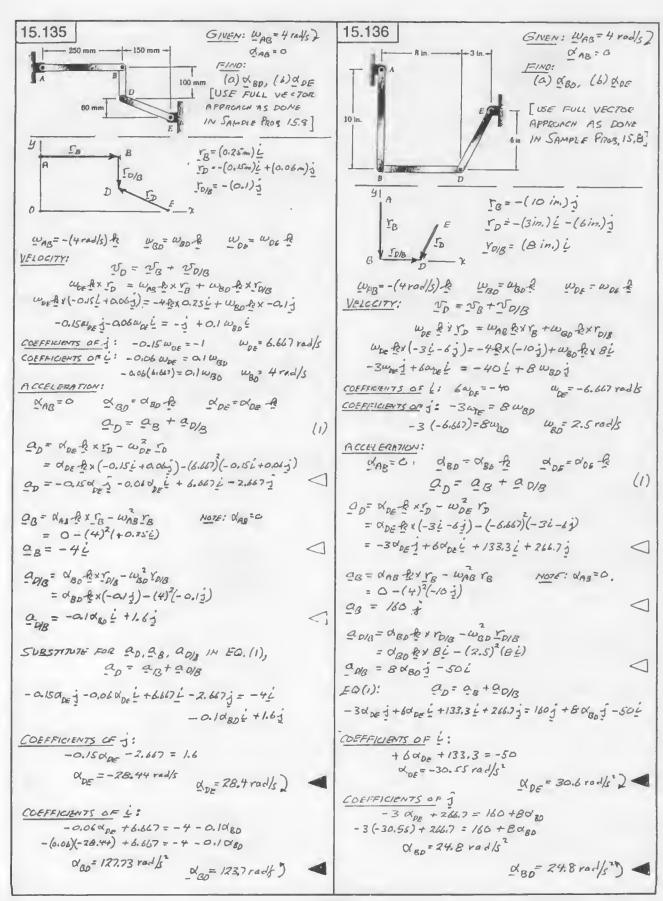




15.127 and 15.128 CONTINUED ACCELERATION OF D: WE KNOW &= 1080 rad/5" AND WOE = 15 rod/s> ap= (ap) 70 + (00) 15 Sim Jave = (DE) NOE -+ (DE) W2 - - 3 Maply 90=(5×1080) + (5×15) 00=5400 W/3 44 + 1/25 W/3 - 55 $(a_0)_{\chi} = 5400(\frac{3}{5}) - 1/25(\frac{4}{5}) = +2340 \text{ m/s}^2$ +1 (00)y=540(x)+1125(3) = +4995 m/s 20 1 (20) = 4995 ap=5520 in/52 \$64.9° (ap) = 2340 = D ACCELERATION OF 6: WE AGAIN USE THE FREE- 8001' a= a + a 6/8 = a + (a 6/2) + (a 6/8) m a= as+ (B6)de0 + (BD) wBD + PECALL INE FOUND OF BO = + 432 rad/s2 AND USE THIS CALLE HERE TOGETHER WITH WED = 12 rads AND CAB = 1175 in 15 12 ax= 1125m/s243+ (5in. 10432 rad/s) ++ (5in x121016) -= [900 in. 5 + +625 in/5] + 2160 in/s] + 720 in/s . a= 2835 in/s + 1620 in/s2= ag 1 2835 in./s2 Qg=3270 mg 1 60.3° € 1620 In/52 La VECTOR DIAGRAM OF EQ() 1=1NIS# 4320 lada) t 675 1440 DIMBUSIONS IN: in./62







15.137



INSTANTANEOUS CENTER
OF RUTATION AT C
(a) SHOW THAT

(b) SHOW THAT ac=0, IF, AND OWLY IF, $Q_{p} = \frac{\alpha}{w} \tilde{v}_{A} + w v \tilde{v}_{A}$

SINCE W I TO PLANE CONTAINING (YA-Y),
CROSS MULTIRYING TWICE BY W IS EQUIVALENT
TO MULTIRYING (YA-Y) BY W2 AND ROTATING
17 THROUGH 1800. THUS,

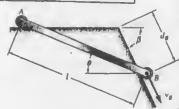
BUT N= ale AND W= WA, AND SMEE LIVA

$$Q_{C/A} = \frac{\alpha}{\omega} \left[f_{X} \times (f_{X} \times f_{A}) \right] - \omega \times V_{A}$$

$$= -\frac{\alpha}{\omega} V_{A} - \omega \times V_{A}$$

SUBSTITUTING INTO (1) AND SOLVING FOR DAY
WE HAVE FOR Q=0

*15.138 and 15.139



PROBLEM 15.138

EXITESS WOF

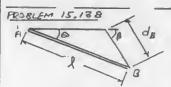
ROD IN TERMS

OF V₈, O. L. AM, B

PROBLEM 15.139

EXPRESS & OF ROD

IN TERMS OF V₈, O. L. + B



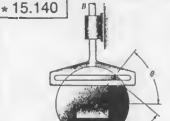
 $\frac{L_{AW} \text{ OF SIMES}}{\frac{d_{B}}{\sin \Theta}} = \frac{l}{\sin \beta}$ $\frac{d_{B}}{\sin \theta} = \frac{l}{\sin \theta} \sin \theta$

N3 = d(da) = 1 case de = l case w

PROBLEM 15.139 NOTE THAT QB = dNB = O.

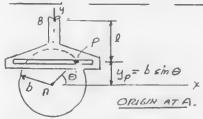
$$C = \frac{d\omega}{dt} = \frac{N_B \sin \beta}{\ell} \cdot \frac{\sin \Theta}{\cos^2 \Theta} \cdot \frac{dC}{dt}$$

$$C = \frac{N_B \sin \beta \sin \Theta}{\ell \cos^2 \Theta} \cdot \frac{N_B \sin \beta}{\ell \cos \Theta}$$



GIVEN: FOR DISK,

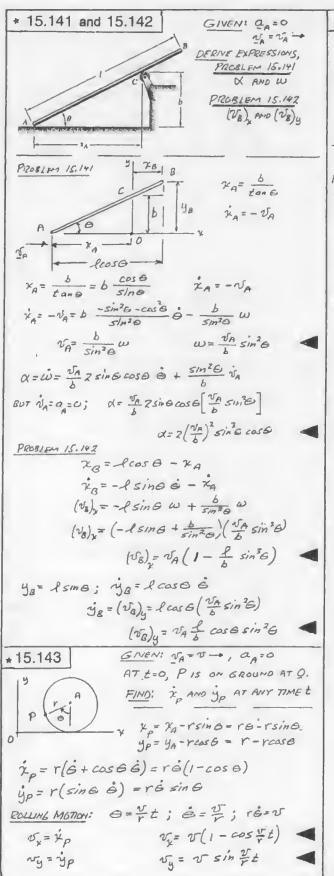
DERIVE EXPRESSIONS FOR V8 AND AB

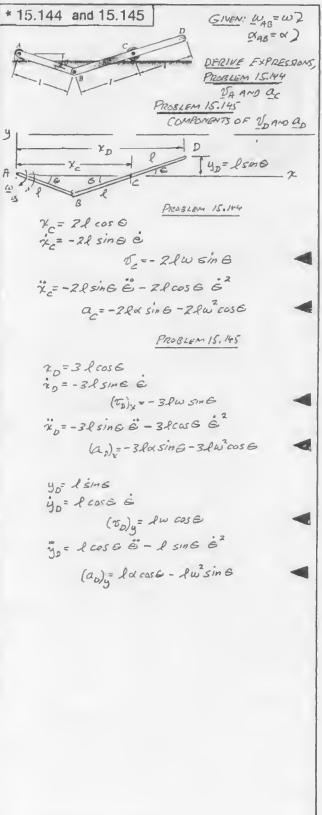


 $y_B = l + y_P = l + b \sin \epsilon$ $v_B = y_B = b \cos \epsilon = b \cos \epsilon \omega$ $v_B = b \omega \cos \epsilon$ $v_B = b \omega \cos \epsilon$ $v_B = b \omega \cos \epsilon$

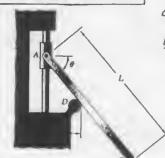
an = - bsing & + b cose &

ag = bdece - bwsine





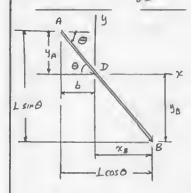
+ 15.146 and 15.147



GIVEN: $N_A = V_A$ $Q_A = O$ DERIVE EXPRESSIONS,

PROBLEM IS. 146

(a) w_{AB} (b) COMPONENTS OF V_B PROBLEM IS, 147 V_A AB



POSTIVE 615 2

Problem 15.146 $y_A = b tan6$ $v_A = \dot{y}_A = b \frac{1}{\cos^2 6} \dot{6} = \frac{b\omega}{\cos^2 6}$ $\omega = \frac{\sqrt{4}a}{b} \cos^2 6$

 $\gamma_g = L\cos \theta - b$ $\dot{\gamma}_g = -L\sin \theta \dot{\theta} = -L\omega\sin \theta$ $= -L\left(\frac{\sqrt{h}\cos^2 \theta}{b}\right)\sin \theta$ $\dot{\gamma}_g = \dot{\gamma}_g = -\sqrt{h}\frac{L}{b}\sin \theta\cos^2 \theta$

 $y_{B} = L \cos \theta - y_{A} = L \sin \theta - b \tan \theta$ $\dot{y}_{B} = L \cos \theta \dot{\theta} - b \frac{1}{\cos^{2} \theta} \dot{\theta}$ $= \left(L \cos \theta - \frac{b}{\cos^{2} \theta}\right) \left(\frac{V_{A}}{b} \cos^{2} \theta\right)$

+1 $(\mathcal{N}_B)_g = \mathring{\mathcal{J}}_B = \mathcal{N}_A \left(\frac{L}{6} \cos^3 \Theta - 1\right)$

PROBLEM 15.147

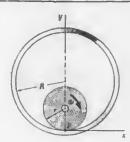
RECALL THAT $\alpha_{A} = \delta_{A} = 0$ and $\omega = 0 = \frac{\pi_{A}}{b} \cos^{2}\theta$

 $\alpha = \hat{\omega} = \frac{\sqrt{a}}{b} \left(- z \cos \theta \, \sin \epsilon \right) \, \hat{\theta}$

NOTES SINCE POSITIVE @ 15 2, THE DIRECTION

 $d = 2\left(\frac{\tau_A}{b}\right)^2 \sin \theta \cos^3 \theta$

*15.148 and 15.149



GIVEN: POSITION SHOWN IS WEEN t=0

W = CONSTANT (x=d)

PROBLEM 15.148

DERIVE EXPRESSIONS

FOR $(\mathcal{N}_p)_T$ AND $(\mathcal{N}_p)_y$

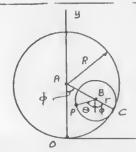
PROBLEM 15.149

WHEN Y=R/2 SHOW

THAT PATH OF P IS & AND

AND DERNIE EXPRESSIONS

FOR Vp AND ap



D= L OAB

G = ANGLE BP FORMS

WITH THE VERTICAL

 $6=\omega t$; $6=\omega$ (1) $v_{B}=(AB)\dot{\phi}$ $v_{B}=(R-r)\dot{\phi}$

SINCE C IS INSTANTANEOUS CENTER, UB= TW
EXWATING THE TWO EXPRESSIONS OFTAINED FOR UB

(R-r) &= YW &= YW (2)

 $\chi_p = (R-r) \sin \phi - r \sin \Theta$ $y_p = R_1 - (R-r) \cos \phi - r \cos \Theta$

DIRFERINTATING AND USING (1) AND (2): $\dot{x}_{p} = (R-r)\cos\phi \ \dot{\phi} - r\cos\theta \ \dot{\theta}$

yp= (R-1) sin¢ ¢ + r sino €

 $\dot{\gamma}_{p} = (R-r)\cos\phi\left(\frac{r}{R-r}\right)\omega - r\cos\theta \ \omega$ $\dot{\gamma}_{p} = (R-r)\sin\phi\left(\frac{r}{R-r}\right)\omega + r\sin\theta \ \omega$

ip = rω(cos φ - cos 6) ip = rω(sinφ + sin 6)

(Vp) = xp = rw [cos rwt - cos wt]
(Vp) = yp = rw [sin rwt | + sin wt]

PROBLEM 15.147 FOR Y = R/2

'ip = rw(cos wt - cos wt) = 0

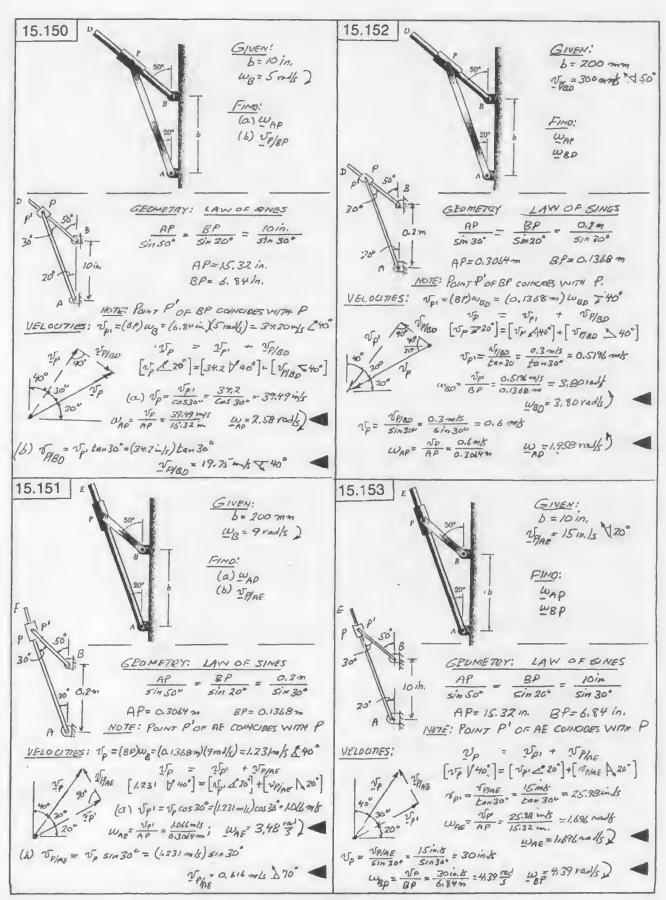
THUS P MOVES ALONG THE Y 1XU

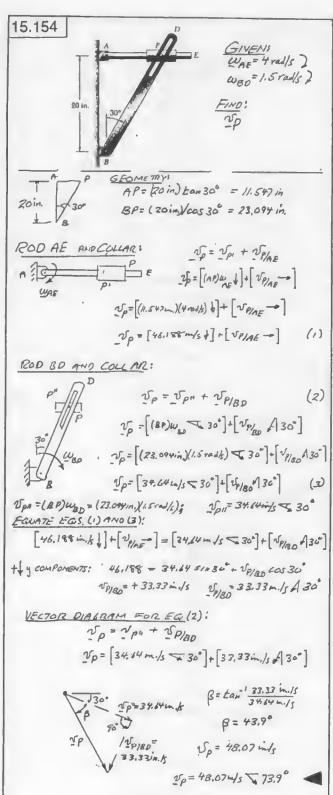
V= yp= Yw (sinut+sinut)
V= 2rwsinut

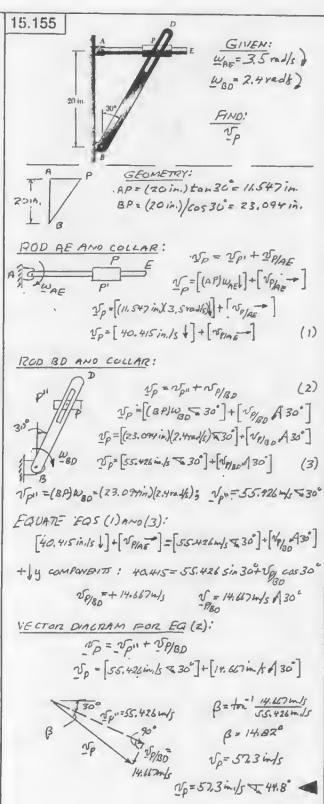
I=(Rwsinut) j

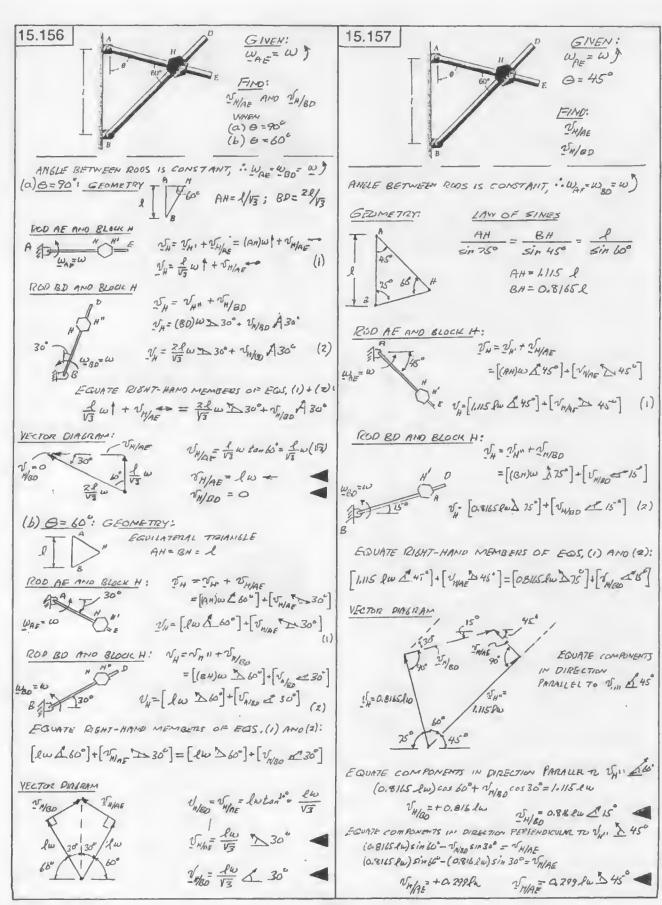
 $a = \frac{dv}{dt} = 2rw(w\cos \omega t)$ [RECALL W=CONSMAT]

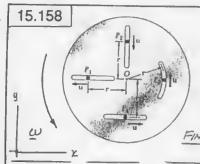
a = (Rw cas wt) j











GIVEN: CONSTANT AHOLIAG VELOGITY = W)

CONSTANT SPEED OF PINS RELATIVE TO PLATE = UL

FIND: ACCEUMATION OF EACH PIN.

ap=ap+app+ae FOR EACH PIN: ACCELERATION OF CONCIDING POINT P':

FOR EACH PAY! api = YW TOWARD CENTER D

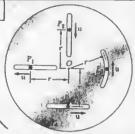
ACCREMATION OF PIN WITH RESPECT TO PLATES FOR P, PZ, AND P4: appa = 0

FOR P4: ap/g= U/r TOWNED CENTER O

CORIOUS ACCELERATION FOR EACH PIN Q = 2 U W, WITH OR IN A DIRECTION OBTAINED BY ROTATING U THROUGH 90° IN THE SENSE OF W.

$$q_{z} = [r\omega^{2}] + [z + \omega];$$

15.159



GIVEN: CONSTANT ANGULAR VELOCITY = W]

CONSTANT SPETO OF PINS RELATINE TO PLATE = LL

FOR EACH PIN: ap = api+apy + ac ACCELERATION OF COINCIDING PONT P'I

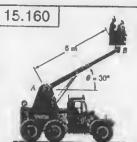
POREACH AN: ap = rw2 TOWARD CENTER O ACCRETATION OF PIN WITH RESPECT TO PLATE:

FOR P, P2, AND Px: ap/g:=0

FOR Py: appy = u2/r TOWARD CENTER O.

COMOUS ALLERATION; FOR EACH PIN, Q = 244 WITH QE IN A DIRECTION OBTAINED BY ROTATING UL THROUGH 90° IN THE SENS OF W.

$$a_3 = [rw +] + [\frac{u^2}{r}] + [2uw -]; \quad a_3 = -(rw^2 + \frac{u^2}{r} - 2uw)L$$



GIVENI: WAS = 0.08 rays) dag=C V8/A = 0,2 m/s 7 30° 98/A = 0

FINO: (a) NB (b) aB

6) VELOCITY: B

(b) ACCEL ERATION:

V8/5- = V6/4 = 0.2 m/s 736 TB = TB, + T8/2

0.2 mb N60 10.48 3/5 B=22.6° VE= 0.52m/s VE26

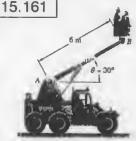
VR=[(AB)w \$60]+[0.2m/s 230] =[(6m)(0.08mb) \ 60°]+[0.2m/5 \ 30°] 28 = 0.48 m/s \$ 60 + 0.2 m/s \$ 30°

QB' = (AB) w= (6m) (0,08 vad/s) = 0.0384 m/s = 38.4 m/s = 30° ac= 2 uw = 2(0.2 m/s)(0.08 roll)=0.032 m/s= 82 m/s \$ 60° Qn=[38.4mm/s 730]+0+[32mm/s 16]



B = 39.8°

az=50.0 mm/s a = 50,0 ma/s > 9.8°



GIVEN: WAB = 0.08 rad/s 2 MA8 = 0 TEM = 0,2 m/s 2 30° ABIA = 0 FMO: (a) VB

(a) VELOCITY

0,2015

VB/g= TB/A = 0.2 m/s 7 30°

VB = VB, + VEV8 VB=[(ARN 760°]+[VB/2 7 30°]

(b) aB

=[600)(0.08 m/s) \$60°] +[0.2 m/s = 30°]

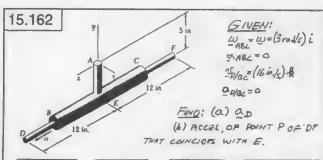
NB = 0.48m/5 5 60° + 0.2m/5 2 30° 8=90-30-22.6=37.4°; U=0.52m/s \$ 37.4°

(D) ACCELEDATION a= 98, + 08/3+0c; 08/2=0

az = (AB)w=(6m)(0.08 ras/s)=0.038+m/s=38.4m m/s 230° a = Zuw = 2(0.2 m/s)(0.08 rode) = 0.032 m/s = 32 m m/s \$ 60° aB=[38.4mm/s 730]+0+[22771/52 566]



B=39.8° ag=50,0 mm/s an = 500 ma/s2 7 69.8°



(a) PONT D: $\nabla D_{g} = \nabla D_{g} = (16 in / s) \cdot R$; $\Delta_{D/g} = 0$ $A\bar{D} = -(5 in) \cdot \frac{1}{2} + (12 in) \cdot R$ $\Delta_{D} = (12 in) \cdot \frac{1}{2} + (12 in) \cdot R$ $\Delta_{D} = (12 in) \cdot \frac{1}{2} \cdot \frac{1}$

(b) POINT P OF DF THAT CONCIDES WITH E

Note = Usin);

ap = W rw AE = - W AE = (3rold) AE = (45 in/s);

ar = 2wx vojg = 2[(2rold) | x(16 in/b) | x = - (96 in/b)]

ap = ap + apg + ar

= [(45 in/s)];

ap = -(5/ in/s)]

15.163

5 in.

σης = ω = (3 rad/s) δ

σης = ω = (3 rad/s) δ

σης = ω

σης

(a) POINT D: VD/g= VD/Bc= (1610. k) &: Aprile= 6

AB = -(5111.) j + (12 in.) f.

ND = W x RD = (3101/5) j x [-(511) j + (12 in.) f.)] = (3611/5) j.

QD = W x ND = (3101/5) j x (36 in./5) i = -(108 in./5) f.

QC = 2 W x ND/g = (3101/5) j x (16 in./5) j.

QD = QD + QU/D + QC

= -(108 in./5) f. + O + (96 in./5) j.

QD = (96 in./5) i - (108 in./5) f.

(b) POINT P OF DF THAT COINCIDES WITH E

**SPIQ= **SPIQC = (16 im./s) &; ap/g=0

AE=-(5in.); ; Np. = w x AE = (2 rad/s) &x (-5in./s) &=0

ap = an NE/Q = 2 (3 rad/s) &x (16 im./s) &= (96 im./s) &

ap = ap + ap/g + ac

ap = 0 + 0 + (96 m/s) & ap = (96 im./s) &

15.164

GIVEN: ELEVATOR MOVES DOWNWARD AT 40 FG

FIND: CORDULIS ACCELETATION OF ELEVATOR IF

IT IS LOCATED AT: (a) EQUATOR, (b) 40° HORTH, (c) 40° SOUTH.

EARTH MAKES ONE REVOLUTION IN 23 h 56m = 23,933 h

W = 771 rad = 72,92 × 10° frad/s *

W = (72,92 × 10° frad/s) \(\frac{1}{2} \) \(\frac{1}{2}

(b) 97 40 NORTH: U = 40 FUS (-cos 40 L - sin 40 j) O = 2 M M = 2 (72.97 x 0 rad/s) j x (40 FU) (-cos 40 L - sin 40 j) O = 4.47 FUS WET

(e) AT 40 SOUTH: U = 40 Ft/s (-cos 40° i + sin 40° j) Q= 2 WIM = 2 (72.92 x 10° rad/s) j x (40 f2/s) - cos 40° i + sin 40° j) Q= (447 x 10°) (1/s° Q= (447 x 10°) (2/s° WEST

* NOTE: EARTH ROTATES COUNTER CLOCKWISE WHEN OBSERVED FROM ABOVE THE HORTH POLE,

15.165

GIVEN: TEST SLED MOVING DUE MORTH
AT 900 2. // h, AT 40° MORTH LATITUDE.

FINO: COMOUS ACCELERATION OF
SLED

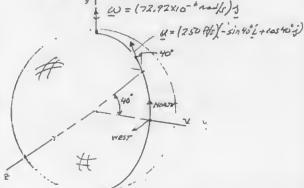
EARTH MAKES ONE REVOLUTION IN 234 56m
OR 23,933 h,

[20 red] 1 - [779211 - red]]

w= \frac{(23,933 h)(3600 \frac{5}{h})}{(23,933 h)(3600 \frac{5}{h})} = (72,92 \times 10^{-6} \text{ rad/s}) \frac{1}{3}

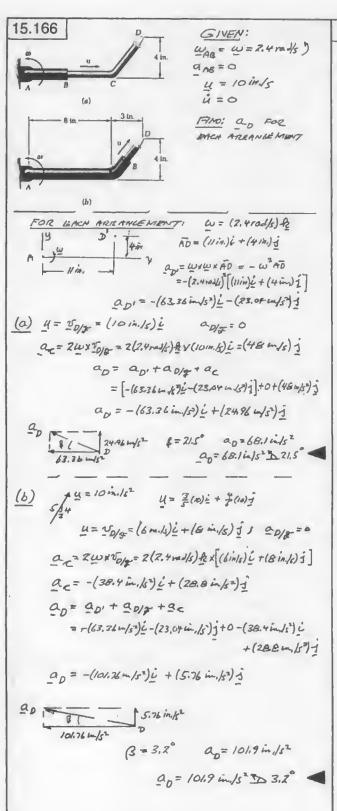
u = 900 \text{km/h} = 250 \text{m/s}

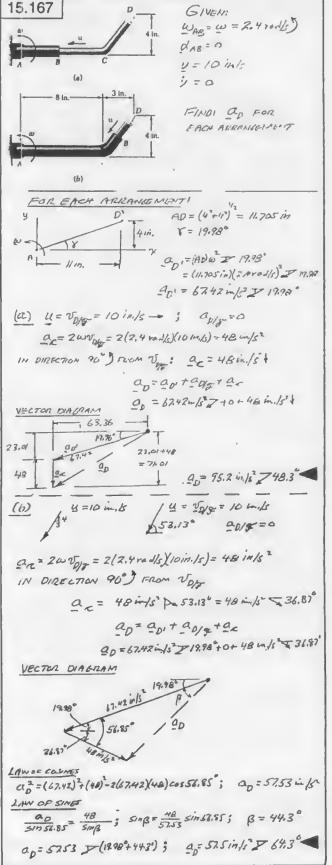
NOTE: EARTH ROTATES COUNTRICLOCKWISE WHEN WIEN THE HOLTH POLE

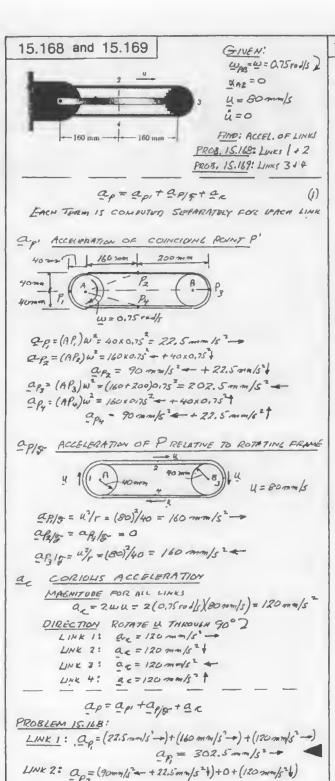


2c = 20 xu = 2(72.92 110 rad/s) \(\frac{1}{3} \omega \left\) - sinto i + costic)

0c = (23.4 x16 3) \(\text{als}^2 \\ \frac{1}{2} = (23.4 x16 3) \(\text{als}^2 \) \(\text{TOF DED} \)







ap = 90 mm/s = + 142.5 mm/s = 168.5 mm/s 757.7"

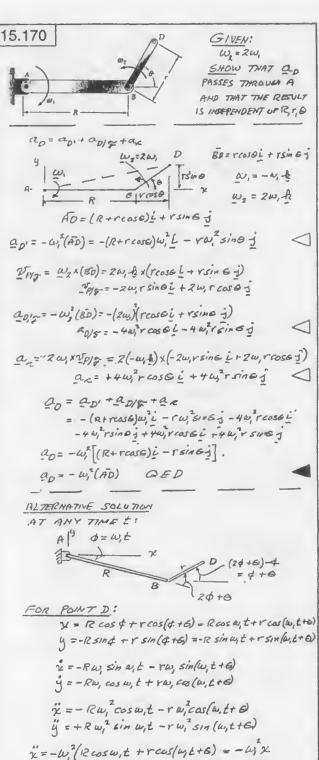
ap = 90 mak - + 142.5 mak 1 = 168.5 mak 2527

ap = 482.5 mm/s2

LINK 3: ap3 = (202,5 mm/s =)+(160 mm/s =)+(120 mm/s =)

LINK 4: apy = (90 ma/s = +22.5 mm/s +) = 0+(120 mm/s +)

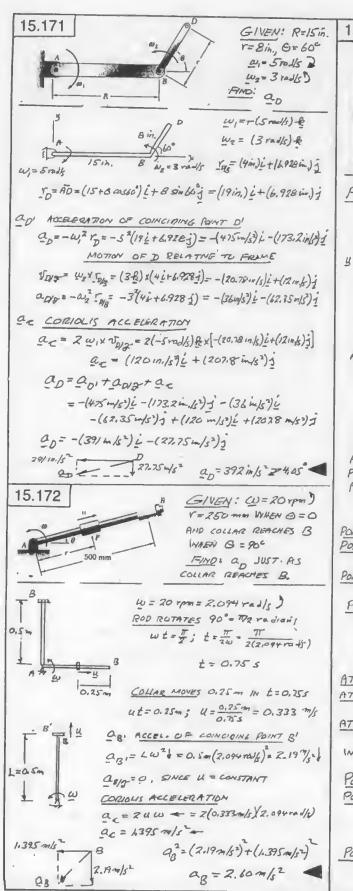
PROBLEM 15.169:



WHEN W = 2W, OD PRISSES THROUGH POINT A
DURING ENTIRE MOTION

ig = -w,2(-Rsinut + Ysin(w, ++6) = - w, y

:. ap=-w,2(AD)



15.173 and 15.174

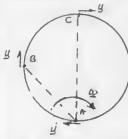


GIVEN: (U= 3ral/s) U= 90 mm/s, u=0

PROBLEM 15.173: FOR d =0 Find: Qp WHEN PIN IS AT (a) POINT A, (b) POINT B, (c) POINT C.

PROBLEM 15.174: SOLVE SAME PROBLEM IF Q = Smad/s²) AS PIN IS AT POINTS A, B, + C.

PROBLEM 15.173: AB = 0.1m1+0.1m=; AC=0.2m1



ACCELERATIONS OF COINCIDE POINTS $Q_{R1} = 0$ $Q_{R1} = -\omega^{2}(AR) = -(3)(AR)$ $= 0.9 \text{ m/s}^{2} \rightarrow +0.9 \text{ m/s}^{3}$ $Q_{C1} = -\omega^{2}(AC) = -(3)^{2}(AC)$ $= 1.8 \text{ m/s}^{3}$

ACCELERATIONS OF PIN RELATIVE TO THE

ROTATING FRAME = U²/r = (0.09 m/s)²/(0.1 m) = 0.081 ⁹/s²

WE HAVE:

Q_A/g = 0.081 m/s²

Q_{C/g} = 0.081 m/s²

Q_{C/g} = 0.081 m/s²

CORIOLIS ACCELERATIONS

FOINT A: SE = ZULU = 2(0.09 m/s) 3 m J/s) = 0.54 m/s² ?

POINT B: SAME MAGNITUDE QC = 0.54 m/s² +

FOINT C: "

QC = 0.54 m/s² +

ap= ap+ app+ ac

POINT A: Q= 0 +0.081 m/s +0.54 m/s = 0.621 m/s 1 POINT B: QB = 0.9 m/s +0.9 m/s +0.081 m/s -0.54 m/s -= 1.521 m/s -0 +0.9 m/s + = 1.767 m/s = 30.6 POINT C: Qs = 1.8 m/s +0.081 m/s +0.57 m/s + 2.42 m/s +

PROBLEM 15.174 WE NOW ALSO HAVE & = 5 rad/s)

THIS ADDITION CHANCES ONLY THE

ACCELERATIONS OF THE COINCIONS POINT BY

HODING THE TERM XXI

AT POINT A: r=0 and dxr=0

AT POINT B: dr=d(AB)=(5rals^2)AB

= 0.5 m/s^2 + 0.5 m/s^2 +

AT POINT C: dr=d(Ac)=(5ral/s^1(0.2n)=1 m/s^2 +

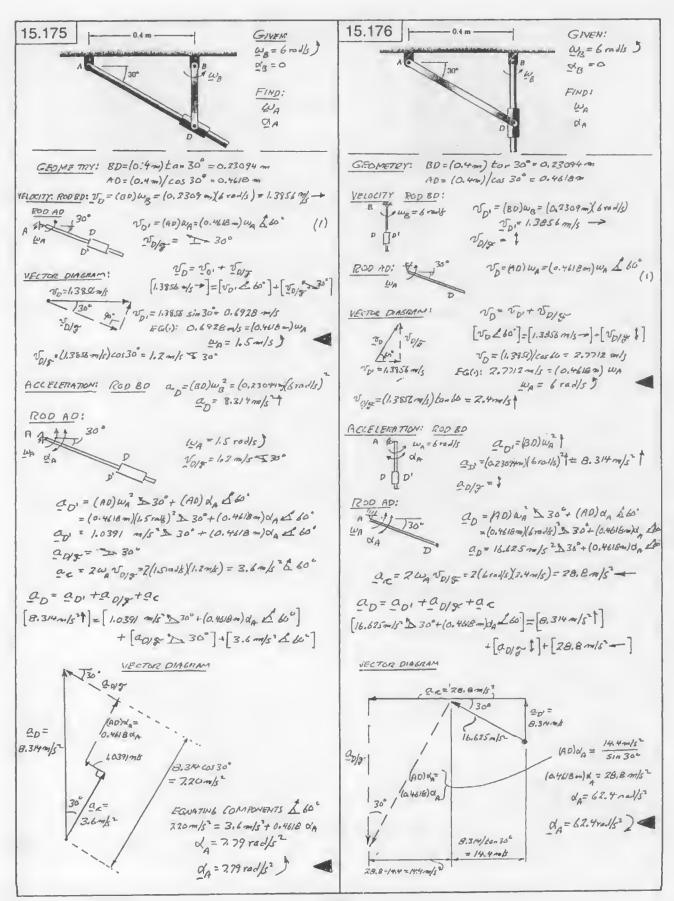
INE NOW ADD OF TO RESULTS OF PILOS, 15.173

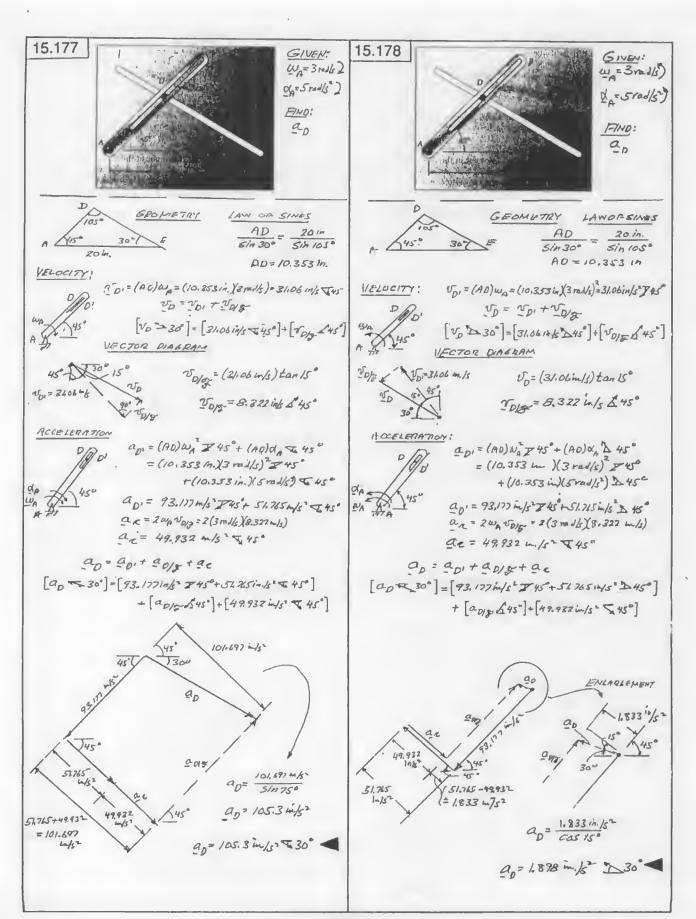
POINT A: a=0+0.621 m/s + 0.5 m/s +1.521 m/s + 0.9 m/s +

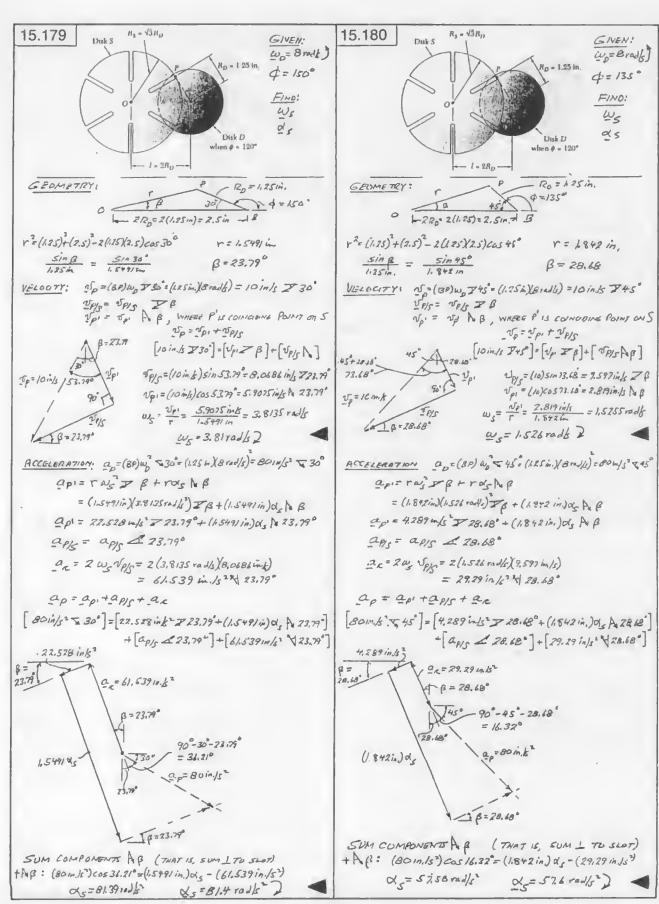
= 1.021 m/s - + 1.4 m/s - +

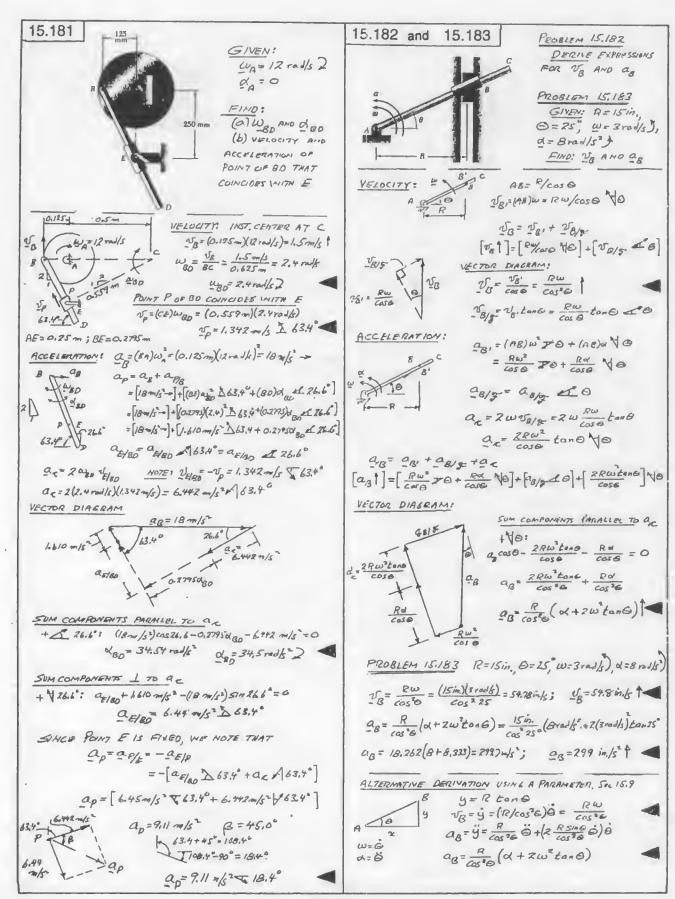
a=1.733 m/s = \$\frac{7}{3} = 1.733 m/s = \$\frac{7}{3} =

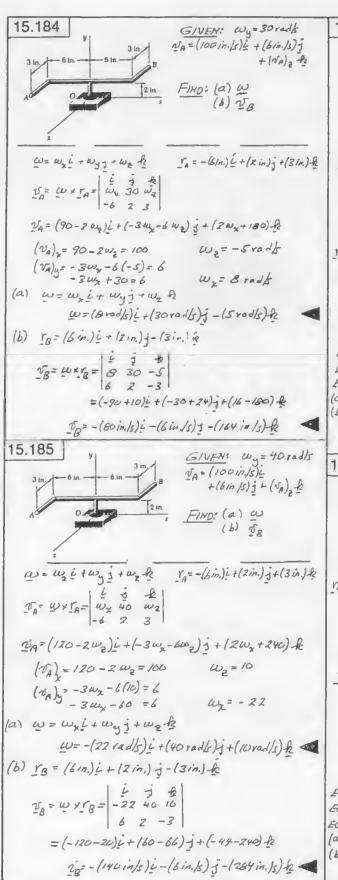
POINT C: Qc= 1 m/s + 2.42 m/s 2 676 = 2.62 m/s 7 676

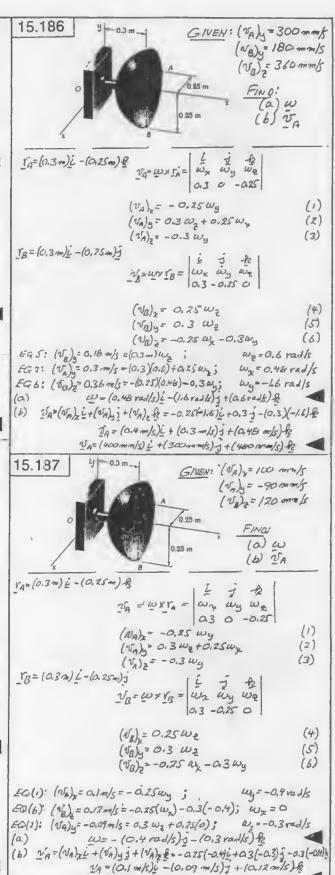




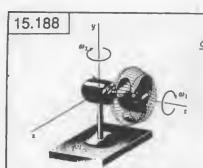








VA = (100 mm/s) i - (90 mm/s) 3 + (120 mm/s) &



GIVENI $\omega_1 = (360 \text{ rpm})L$ $\alpha_1 = (360 \text{ rpm})L$ $\alpha_2 = (2.5 \text{ rpm})J$ $\alpha_2 = 0$ EIND: FOR HOUSING $\omega_F MUTOR$ $(a) \omega_H$ $(b) \alpha_H$

ω,=-(360 rpm) L = -(1211 rad/s) ξ

ωz=-(2,5 rpm) δ = -(17/12 rad/s) δ

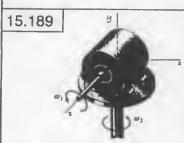
ωz = ROTATION OF FRAME Ozy2

X=(ω,+ω,=)=(ω,+ω,2)_{0xy2}+ω2x(ω,+ωα)

X=ω2xω,=(-17/12 rad/s) δ x(-/217 rad/s) δ

Q=-(9.8696 rad/s) δ

X=-(9.87. rad/s) δ



GIVEN: W = 1800 rpm

01, = 0

W2= 677m

01 = 0

FINO: FOR ROTOR OF MOTOR, OL

ω, = (1800 γρπ) & = (60 π rad/s) &

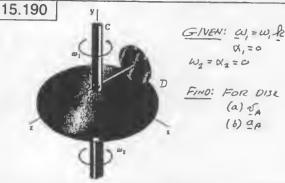
ω2 = (6 γρπ) j = (π/5 rad/s) j

ω2 = 10 γρπ ο ε ι επανιε Ο χ ε

α = (ω, + ω2) = (ω, + ω2) ο χ χ + ω2 χ (ω, + ω2)

α = ω2 χωι = (π/5 rad/s) j χ (60 π rad/s) &

α = (118, 4 rad/s) i α = (118 4 rad/s) i



DISK A: (IN ROTATION ABOUT D) SINCE Wy=W, WA= Wx L+W, J+ Wz-&

15.190 CONTINUED

STACE W = 0, 50 = 0

EACH COMPONENT OF TO IS ZERO $\{V_0\}_{\gamma} = r \omega_{\gamma} = 0;$ $(V_0)_{\gamma} = -R \omega_1 + r \omega_{\gamma} = 0;$ $\omega_{\gamma} = \omega_{\gamma} + (R/r) \omega_{\gamma}$ $\omega_{\gamma} = \omega_{\gamma} + (R/r) \omega_{\gamma}$

(b) DISK A: ROTATES ABOUT & AXIS AT RATE W,

CI = dwa = wyxw = wrix(w, 1 + Rw, 1)

Q= FWZ L



GIVEN: $\omega_1 = \omega_1 \cdot k$ $\omega_2 = \omega_2 \cdot k$ $\omega_3 = \omega_3 \cdot k$ $\omega_4 = 0$

FINO: FOR DISK,

(a) \$\sum_{A}\$

(b) \$\dag{A}\$

DISK A: (IN ROTATION ABOUT O)

STINCE Wy W, WA W, L+ W, J+WZ &

POINT D IS POUNT OF CONTACT OF WHEEL AM DISK

TOB = -rj - RA

10= WAX TOO = W2 W1 W2

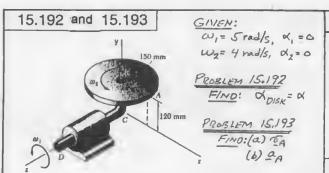
0 -r-R

50=(-Rw,+rw) + Rw, j-rwx & (1)

DISK B: $\omega_8 = \omega_2 \stackrel{\checkmark}{J}$ $\Sigma_D = \omega_B \times \Sigma_{Dh} = \omega_2 \stackrel{\checkmark}{J} \times (-r \stackrel{\checkmark}{J} - R \stackrel{\checkmark}{L}) = -R \omega_2 \stackrel{\checkmark}{L}$ (2)

From Ecs 1 and: $\sqrt{D} = \sqrt{D}$: $(-Rw_1 + rw_2)i + Rw_1 - rw_2 k = -Rw_2 i$ $\frac{COVE.OF}{i}: (-Rw_1 + rw_2) = -Rw_2;$ $w_2 = \frac{R}{r}(w_1 - w_2)$ (a) $w_4 = w_1 j + \frac{R}{r}(w_1 - w_2)k$

(b) DISKA ROTATES ABOUT Y AXIS AT RATE ω , $\alpha = \frac{d\omega_n}{dt} = \omega_y \times \omega_A = \omega_y \hat{j} \times \left[\omega_y \hat{j} + \frac{R}{r} (\omega_y - \omega_z) \hat{k} \right]$ $\alpha = \frac{R}{r} \omega_z (\omega_y - \omega_z) \hat{c}$



DISK: W = W, & +W, & = (4 rad/s) & + (5 rad/s) & PROREM 15.192:

DISK ROTARS ABOUT & AXIS AT RATE W, = W, &

X = 0, yw = (5rad/s) Ax[(+rad/s) + (5rad/s) A]

X = -(20raj/s²) i

PROBLEM 15.193:

$$N_A = (0.15 \text{ mm}) \hat{L} + (0.12 \text{ mm}) \hat{J}$$

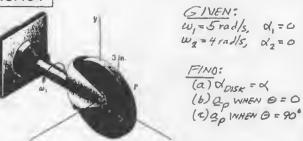
 $N_A = (0.15 \text{ mm}) \hat{L} + (0.12 \text{ mm}) \hat{J}$
 $N_A = (0.15 \text{ mm}) \hat{L} + (0.12 \text{ mm}) \hat{J}$
 $N_A = (0.15 \text{ mm}) \hat{L} + (0.12 \text{ mm}) \hat{J}$

1-A=-(0.6 m/s)+(0.75 m/s)]-(0.6 m/s)A

$$Q_{A} = \times \times Y_{A} + \times Y_{A}$$

$$= -20i \times (0.18i + 0.12j) + \begin{vmatrix} i & j & -k \\ 0 & 4 & 5 \\ -0.6 & 0.75 & -0.6 \end{vmatrix}$$

15.194



 $\omega = \omega, \dot{c} + \omega_2 - \dot{R} = (5 \text{ rad/s}) \dot{c} + (4 \text{ rad/s}) \dot{R}$ $\alpha = \omega, \times \dot{\omega} = (5 \text{ rad/s}) \dot{c} \times \left[(5 \text{ rad/s}) \dot{c} + (4 \text{ rad/s}) \dot{R} \right]$

(a) $\Delta = -(20 \text{ rad/s}^2) \frac{1}{2}$ (b) $\Delta = 0$: $\Gamma_p = (3/n) \frac{1}{2}$

 $\mathcal{D}_{\rho} = \omega \times r_{\rho} = (SL + r_{\rho}) \times 3L;$ $\mathcal{D}_{\rho} = (12 \text{ in.}/s) \cdot j$ $\alpha_{\rho} = \alpha \times r_{\rho} + \omega \times r_{\rho}$

 $= -20\cancel{5} \times 3\cancel{i} + (5\cancel{i} + 4\cancel{k}) \times 12\cancel{j}$ $= 60\cancel{k} + 60\cancel{k} - 48\cancel{i} = -48\cancel{i} + 120\cancel{k}$ $Q_{p} = -(48 \text{ in.}/5^{2})\cancel{i} + (120 \text{ in.}/5^{2})\cancel{k}$

(LONTINUED)

15.194 CONTINUED

(c)
$$\Theta = 90^{\circ}$$
: $r_p = (3in)\frac{1}{2}$

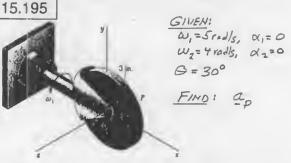
$$1f_p = N \times f_p = (5i + 4 + 2) \times 3j; \quad N_p = -(12ii./s)i + (15in./s)k$$

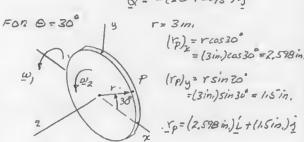
$$a_p = d \times f_p + w \times N_p.$$

$$= -20\frac{1}{2} \times 3\frac{1}{2} + (5i + 4 + \frac{1}{2}) \times (-12i + 15 + \frac{1}{2})$$

$$= 0 = 75\frac{1}{2} - 48\frac{1}{2} = -123\frac{1}{2}$$

$$\alpha_p = -(123in./s^2)\frac{1}{2}$$





$$\int_{p} = \omega \times \Gamma_{p} = \begin{vmatrix} \dot{L} & \dot{g} & \dot{Q} \\ \dot{S} & 0 & 4 \\ 2.598 & 1.5 & 0 \end{vmatrix}$$

$$= -6\dot{L} + 10.392\dot{g} + 7.5\dot{Q}$$

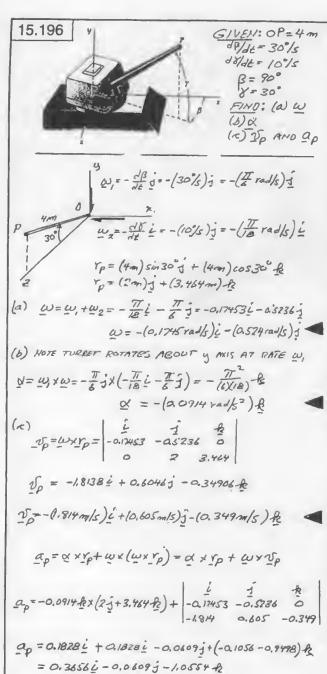
$$\int_{p} = -(6in.k)\dot{L} + (10.392in./s)\dot{g} + (7.5in./s) \cdot \dot{Q}$$

$$a_{p} = \omega \times \Gamma_{p} + \omega \times \nabla_{p}$$

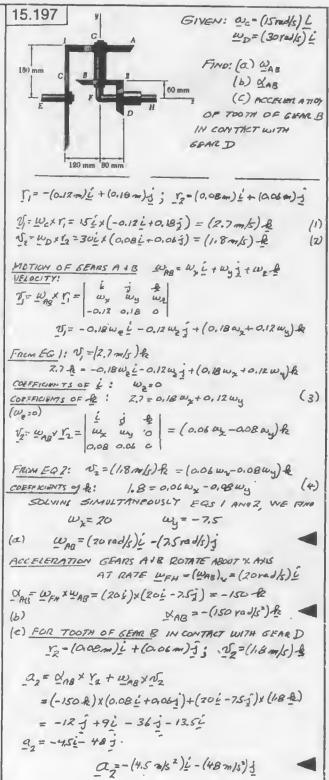
$$a_p = -20j \times (2598i + 1.5j) + 5 0 4$$
 $-6 10.392 7.5$

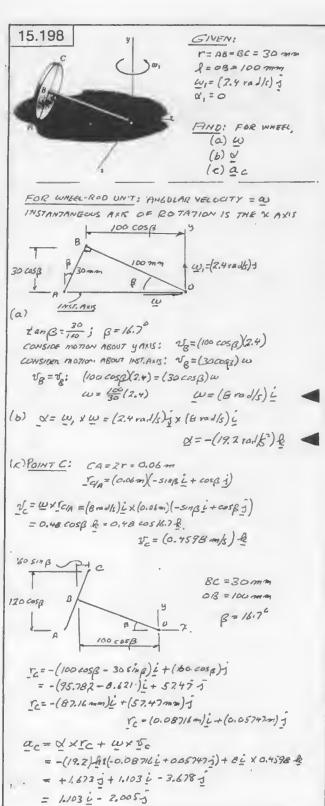
= 51.96 & -41.57 L+(-24-325) +51.96 & ap = -41.57 L - 61.5 + 103.92 &

ap=-(41.6 in./s2)i-(61.5 in./s)j+(103.9 in./s2)-&



ap= (0.366m/5) i - (0.0609m/5) j - (1.055m/5) &



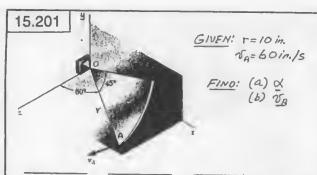


(aB) =0 PROBLEM 15.199 FINO: (a) W (6) 25 ProgLEM 15,200 FINO: (a) & (b) ac PROBLEM 15.199 13 = - (6 in.) i 12A)4 (VA)=-2(VA)4 TA = (8 in.) & W= Wzi+wyj+wz A V8= Wx 18 = (wzi+ wyg+ az k) x (-6h) i - (15 in/s) &= (6 in) wy & - (6 in) w= 1 -157 p/s = (6im) way Wy = -2.5 rad/s COEF, OF AC: COEF. OF j: a = (6in) wz W2=C VA = WYVA = (W2 1- 7.5) X 8-R (1) 5/ = -(2010/s)i - (Bin.)wy 5 BUT (NA) =- 2(2/A) : . - 20in/s = -2(-Bin.) wz Wx = -1.25 rad/s w=-(1.25 rad/s) =- (2.5 rad/s) r= (4in.) 3 1 = ax 5 = (-1.25 i - 2.5 j) x 4j = - 5 A Vc=- (5in/s) & VA=WXFA = (-1.25i -2.5j) y 8 + = - (2011./s)i + (1011./s)j PROBLEM 15.200 SINCE (08) =0, QB = (08) 21 QQ = X YB + WX YB = (\(\si \cdot \alpha \gi + \alpha \gi \frac{1}{2} \rightarrow \frac{1}{2} \ = +604 & -602 & -18.75 j +375i COEF. OF A: O= Ny COEF. OF -j: 0 = -60 = -18.75 0 = -3.125 rad/s2 an= XXIA+WXJA = (4,2 i - 3.125-2) x 8 12 + (-1.25 i - 2.5 j) x (-20 i +10 j) a= -8x27 -12.5-12-50-12 (an) x=0 (an) y=-801x (an) 2=-62.5 SINCE (VA) = - 2(VA)y, (aA) = - 2(aA)y 0= -Bdx X = - (3.125 rad/5) & rc= (4in.) 1 ac= XXYc+w+Vc = (-3.125 h)x4-j+(-1.25 i-2.5j)x(-5-b) = 12.51-6.25 5+12.54 a = (25 in./52) 6 - (6.25 in./5) 5

GIVEN: 05 = - (15 in./s) &

15,199 and 15,200

a = (1.103 m/s) 1- (2.005 m/s) 1



FIND OB: OA = (10in.) sin 60 L + (10in.) cas60 & TA = OA = (8.6603 im.) i + (5in.) B OB = ((8) 1 + (18) 4 1.

SCALMR PRODUCTS

OA . OB = (OA)(OB) 5in 45° (8.6603 i + 5-12) + (10) i + (18) i) = (10)(10) sin 450 8.8603 (rg) = 70.7/1 (rg) = 8.165 in (r3)4 = 082-(r8)2 = 102-8.1652; (r8)4 = 5.773 in TB= (8.165 in.) i + (5.773 in.) j

SINCE Of I OA, NA FORMS 30 ANGLE WITH ZAXIS 8A=(6010/5)(-SIN30 + COS 603)

5 = - (30in/s) + (51.96in/s) & PLATE OAR: W= Wxi+ wyj+ wzfe

Na= WXYA= Wy my we 8.6603 0 5

-30 i +51.96 h = 5 wy + (8.6603 wg -5 wg) - 8.6603 wy A

COFF i: -30 = 5 Wy -> Wy=-6 rad/s COEF o: 0 = 8.6603 W2 - 5W2 - 602 = 0.57735W2 coet, h: 57.96 = -8,6603 any -> 10y=-6 rad/s

NB= MALE WX MY WE 8.165 6.773 0

VB = - 5.773 Wz + 8.16502 + (5.773 Wz - 8.165 Wy) A

SINCE POINT B MOVES IN 24 PLANE (VB) = 0 = 5.773 wz - 8.165 wy 0 = 5.773 Wx -8.165(-6)

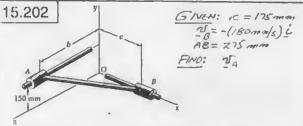
Wx = -8.486 rad/s

EG(1): W= - 057735(-8.486) = -4.899 rad/s

w = -(8.49 rad/s)i - (6 rad/s) - (4,90 rad/s) &

(UB) = -5.773 (-4.899) = 28.3 m/s (58)4= 8.165 (-4,899) = -40,0 in/5 (VB)= 0

25= (28.3 in./s)i-(400 in./s)j



K=175mm; 275=150+175+6; 6=150mm UB = - (180 ma/s) 1; VA = VA & ; VAVA = - 175 1 + 150 \$ + 150 & NA = NB + TAB = NB + WXYAB TAR=-1801+ Wy Wy

V. A = -180' + (1500) - 1500) i + (-1750) - 15000) + (15000 + 17500) &

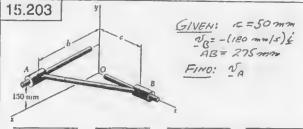
COEF OF i: +180 = +150Wy - 150Wz

COFF. OF 5: 0 =-150 Wg (2) COER OF R: VA = 150 Wx + 17504 (3)

(FO2+FC3) = 60/7=0 +150 Wy - 150 WZ

(4) E01-EG4 180-684/2=0; 8= 210 ma/s &

FOR USE IN PEOR IS. 214; WE CALCULATE A POSSIBLE W. INE SHALL ASSUME Wy = 0: FROM EQ(2), WE HAVE Wy = Q. EQ(1): 180 = 0 +150 W4 eu=+(1.2 rad/s)]



(C=50mm; 275=1502+502+62; b=225mm NB=-(180 mm/s)i; fa= NA to ; YA/B=-501+1501+225-12 NA = NB+ JAIB = NB+WXYAIB

VA Pr = -1801 + (225Wy -150W2) 1+ (-50W2-275W2) 1+ (150W2+50W3) 8

COEF OF L: +180 = 225 mg - 150 mg (1)

0 = -225w2 - 50W2 COSF OF 1: (2) COEF OF R: (3)

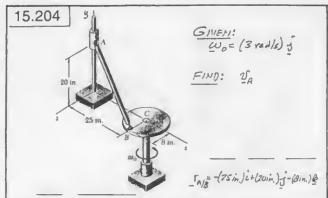
VA = 150 W2 +50 W4 [EQ.1 + 3x EQ.2 - 4.5 EQ.3]: 180-4.58, =0 VA=+(40mm/s) & €

FOR USE IN PROB. 15.215:

WE SHALL CALCULATE A POSSIBLE W. ETIME W IS MOETERMINATE, WE CAN ASSUME A VALUE FOR ANY COMPONENT OF U. LUE ASSUME W2=0, FIRM EC(2), WE AND WE SO

EQ(1): +180 = 225 mg

W=(0,8 rad/s) 5



DISK: UB = WOY (Ble = (3rails) gy (8in.) & = (24 in. ls) L VA=VB+ JAB=JB+WXYBIC 2/ j= (2+/m/di + w. w. w. -25 +20 -8

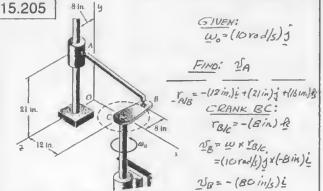
1/2) = 24 i + (-Eny-20w) i + (-25w2+8w2) 3+ (20w2+25w2) &

CLEF OF
$$i$$
: $-24 = -8\omega_y - 20\omega_z$ (1)
CLEF OF j : $V_A = 8\omega_x - 25\omega_z$ (2)

$$\frac{CCEF \text{ of } 1:}{COEF \text{ of } 4:} O = 20w_x + 25w_y$$
(3)

SINCE DETERMINANT OF WX, WY, WZ IS ZERO, W IS INDETERMINATE. WE CAN ASSUME ANY ONE COMPONENT ASSIME Wy=0, EQ.3 NEW Wy=0.

Eq.1:
$$-24 = 0 - 20\omega_2$$
; $\omega_2 = 1.2 \text{ rod/s}$
Eq.2: $v_A = 0 - 25(1.2) = -30$; $v_A = -(30 \text{ m/s})$

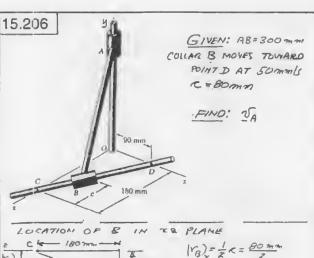


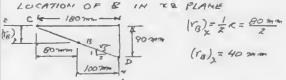
ROOAB: NA = VB + VAIR = NB + W * TELC Jaj=- 801 + wy wy we

0 = 21 wz + 12 mg COST, OF A:

SINCE DETERMINIANT OF Wa, Wy WE IS ZERO, THE ANGULAR VELOCITY IS INDETERMINATE, WE CAN ASSUME VALUE OF ANY ONE COMPONENT.

ASSUME $w_{\chi}=0$. EQ.2 YIELDS $w_{\psi}=0$ EQ.(11): $-80=0-2/w_{2}$; $w_{\chi}=-\frac{80}{27}$ rad/s EQ.(2): VA=0-12(-27); VA=(45.7in./5)5





1200 AB = 300 mm (300mm)2=(100mm)+(40mm)+(40) 10 = (280 mm) 1 7 = 1-(-2++4)

IB= 258 = 50 (-2A+i) = - (44,22 mm/s) & + (22,36 mm/s) L · VA/a = - (40mm) + (250mm) + - (100mm) &

Taj = -4472-12+22,36i+(-100 mg-2000)i + (-40 m2 + 100 m2) 3 + (200 m2 + 40 my) &

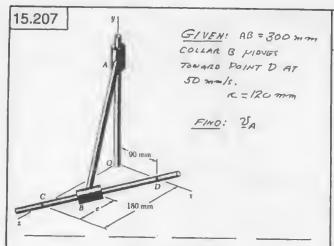
COEF. OF
$$j$$
: $\sqrt{A} = 100 \, \text{W}_{\text{K}} - 40 \, \text{W}_{\text{E}}$ (2)
 $COEF. OF P: +44.72 = 280 \, \text{W}_{\text{K}} + 40 \, \text{W}_{\text{Y}}$ (3)

SINCE DETERMINANT OF WY, WY, WZ IS ZERY, THE ANGULAR VELOCITY IS INDETER MINANT, WE CAN ASSUME VALUE OF ANY COMPONENT.

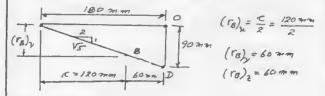
ASSUME WL=0 my=1.112 rad/s EG. 3: 44,72 = 0 + 40 404; EG. 1: - 22,36 = -100(1.118) - 280 W2 w= -0,3194 radt

W= (1.118 rad/s) g-(0.314 rad b) &

EG. 2: Va= 0-40(-0.3194)= 12.777 mm/s VA= (12.78 mm/s) 3 .

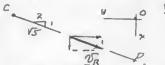


LOCATION OF B W WE PLANE



ROD AB = 300 mm (300 mm) = (60 mm) + (60 mm) + (rA)2 TA = 287.75 mm

TAIR = - (80 mm) + (287.75 mm) j - (60 mm) &



VR = 50 mm/s 2== (-2+i)

VB = VB = 50 (1-2-3) = + (22,36 mm/5)1 - (44,72 mm/s) A

NAj = 22.36i-44.72 f2 + (-60 mg - 207.75 We) i + (-60 wz +60 wz) j + (267,75 wz +60 wz) &

-22.36 = COEF- OF L: -60 wy - 287.75 Wz (1) COEF, OF J VA = 60002 - 50 W2 (2) COEF. OF &: 44.72 = 28275W +6000g (3)

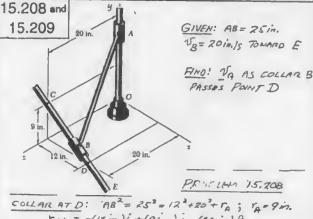
SINCE DETERMINANT OF WE, WY WE IS ZERO, THE ANGUAR VGLOCITY IS INDETERMINANT. INE CAN THUS ASSUME THE VALUE OF ANY COMPONENT.

ASSUME Wy=0: EQ.3: 44.72=0+60wy; Wy = 0.7453 rad/5 -22.36 = -60(0.7453)-287.75w2

W3 = -0.0777 rad/5

W= (0.7453 rad/s) = (0.077 rad/s) & FG. 2: NA = 0-60(-0.0777) = 4.66

1/A= (4,66 mm/s) 5



rala = - (12ini) + (9in.) j - (20 ini) &

3 4 NB=20 IN/S NB= NB NO = NB (0.86 -0.69) NR= (16 in. 15) i - (12 in./5) i Va= VB+VAIR= VB+WY SAIB

(DEF. OF L: -16 = -20mg -9me 10

COFF. OF j: NA +12 = 20 az (2) COFF. OF A: 0 = 9 az + 12 wy (31

SINCE DETERMINANT OF WY, Wy, 12 IS ZERD, THE ANSULAR VELOCITY IS INDETERMINATE, WE CAN THUS

ASSUME THE VALUE OF ANY COMPONENT ASSUME WE = 0, EQ 3. YIELDS Wy = 0 we = 16 rads EG.1: -16 + 0 - 942

FO. 2: 8+12=0-12(16); VA+12=-21.33 VA = - (33,3 in/s) 1

Frace By 15.209 COLLAR AT C: A8 = 25 = 20 + 54; YA= 15in. [A/B = (15in.)] - (20 in) A VR = (16 in.15) = - (12 in.15) j

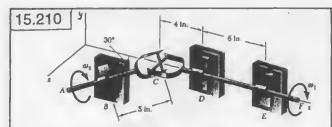
DA = DB + DAID = DB + WX SAB NAj= 16i-12j+ wx wy we

COEF. OF UI 16 = -20mg - 15 w2 (2) COFF OF T! VA+12 = 20Wz

COFF. OF A: 0 = 15 Wz (3)

EG.3: Wx =0 EQ. 2: VA+12 = 0 VA=-12 m/s N=-(1210.15)1

NOTE: WIS INDETERMINATE, ANY VALUE CAN BE CHOSEN FOR EITHER WY ORWZ FOR EXAMPLE, If waso, THEX EQ. 1: 16 = - 20 wy Wy = -0.8 rad/s



WHEN ARM OF CROSSPIECE ATTACHED TO SURET CF IS HORIZONTAL, FIND CUZ OF GHAFT AC.

PLACE ORIGIN AT CENTER OF CROSSPIECE AND DENOTE BY & THE LENGTH OF EACH ARM.

$$r_{H} = -R \sin 30i + R\cos 20j$$

$$w_{I} = -w_{I}\hat{u}$$

$$w_{Z} = -w_{Z}\cos 30\hat{u} - w_{Z}\sin 30\hat{j}$$

SMAFT CF: $V_{h} = \omega_{1} \times I_{e} - \omega_{1} \hat{I} \times \hat{I}_{h} = \hat{I} \omega_{1} \cdot I_{h}$ (1) $SMAFT AC: V_{h} = \omega_{2} \times I_{h}$ $= (-\omega_{2} \cos 30 \hat{i} - \omega_{2} \sin 30 \hat{i}) \times (-\hat{I} \sin 30 \hat{i} + \hat{I} \cos 30 \hat{i})$ $V_{h} = -\hat{I} \omega_{2} \cos^{2} 30 \hat{i} + 2\omega_{3} \sin^{2} 30 \hat{i} \hat{I}$ $V_{h} = -\hat{I} \omega_{2} (\cos^{2} 30 \hat{i} + \sin 30 \hat{i}) \hat{I}_{h} = -\hat{I} \omega_{2} \cdot \hat{I}_{h}$ (2)

CROSSPIECE W= Wx i + wy j + wz le Va= Wxxz = (ax i + wy j + wz le) x l le

Vo= ωχνζ = (axi+ky j+ky k) × lh Vo= -lw, j+lwy i (3)

EQ1=F03: $v_8 = v_8$ $lw, j = -lw_x j + lw_y i$ coeff. of j: $lw, = -lw_x$ $w_x = -w$, (4) coeff. of i: $w_y = 0$ (5)

TH=WXYH= WX Wy WE -LSIM3W LCOS30° O

> = -lwecos30° i - lw2 sin30° j +(lw2 cos 30°+ lw2 sin30°) R

SUBSTITUTE FROM EGS. 4 AND 5: Wx= AND Wy=0

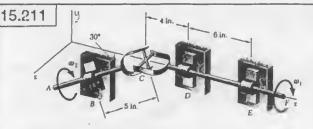
TH=- LW2 COS 30" -- LW2 sin 30" - LW, COS 30" A

SUBSTITUE FOR Ny FROM EG. 2.

-lw2-=-lw2cos30i-lw2sin30j-lw,cos364

COEF OF B: J=- LW2 SIN36 W2=C

wz=w, cos30"



VIHEN ARM OF CROSSPIECE ATTACHED TO SHAFT CF IS VERTICAL, SIND WIZ OF SHAFT AC.

PLACE ORIGIN AT CENTUR OF CROSSPICE AND DENOTE BY & THE LENGTH OF EACH ARM

2/5- Jung 100 1 + 1 was m 20 1

<u>v. (ε) 1 ire</u>: ω= ωχ <u>i</u> + ωχ <u>f</u> + ωχ <u>f</u> <u>σε</u>= ωχ <u>r</u>ε = (ωχ <u>i</u> + ωχ <u>f</u> + ωχ <u>f</u>) χ <u>f</u> <u>f</u> <u>νε</u>= lωχ fε - ρω <u>i</u> (2)

S_μ= ωχγ = (ωχ ½ + ως ½ + ως ½ () x(-lk)

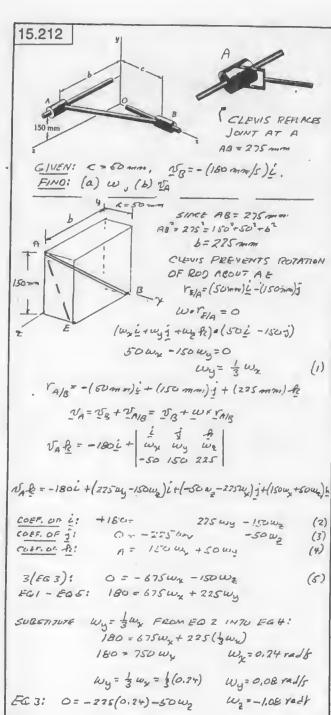
σ_μ= lωχ ½ - lωy ½

(4)

Ec. 1 = Ec. 3: $v_6 = v_6$ $-lw, k = lw_x k - lw_2 i$ $ccep.of k: -lw, = lw_x$ $w_2 = 0$ (4) $ccep.of i: w_2 = 0$ (5)

FRUM EQ.5: $\omega_{\chi} = -\omega_1$ THUS, $\omega_2 = -\frac{(-\omega_1)}{\cos 3\omega_2}$

W2= cos 300



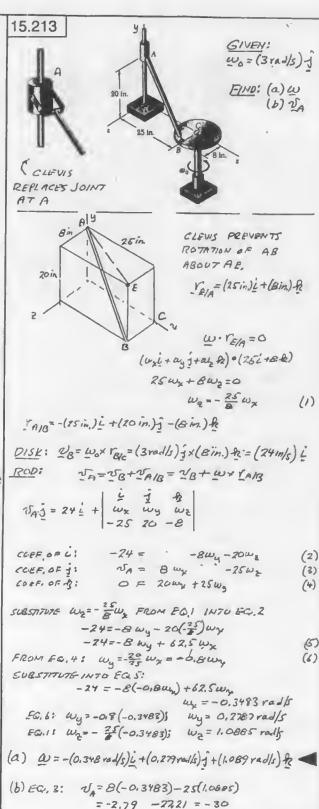
W= (0,24 rad/s) i + (0,000 vad/s) j-(1Ne rad/s) &

S= (40 mm/s)-k

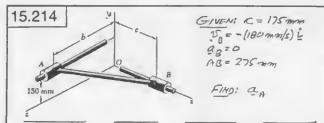
(b) EG 4: VA= 150(0.74) +50(0.08)

= 36 +4

(a)



NA=- (30 in./s) j



FROM SOLUTION OF PROB. 15.202 INE RECALL

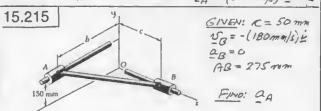
b = 150 mm; YA/B = - (15 mm) L + (150 mm) ft (150 mm) &

w = + (1.2 rad/s) f

WE NOW CALCULATE!

$$\frac{\text{COEF OF is}}{\text{COFF OP js}} = -252 = 150 \text{My} - 150 \text{Mz}$$
 (1)
$$\frac{\text{COEF OF js}}{\text{COEF OP js}} = -1750 \text{Mz}$$
 (2)

\[
 \lambda \] IS INDETERMINANT: ASSUME \(\pa_x = 0 \), FROM \(\tilde{c}_2 \), \(\pa_2 = 0 \)
 \[
 \tau \) FQ. 1, YIELOS: -252 = 150 \(\alpha_y \); \(\alpha_y = -1.68 \)
 \[
 \lambda \]
 \[



FROM SOLUTION OF PROB 15.203, WE RECALL:

b = 275 mm; YAB = - (50 mm) L+ (150 mm) j+ (27 mm) k

w = (0.8 rad/s) j

VAIB = WXYAB = 0.8 j X (-50 L+150 j+225 k)

TAB = (40 mm/s) k + (160 mm/s) L

an 1 = (22504 - 1500 =) + (-5002 - 225012) + (1500 + 5004) + 32 i - 144 12

(CONTINUED)

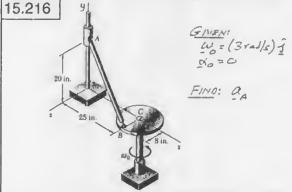
15.215 CONTINUED

 $\frac{\text{COFF. OF i:}}{\text{COFF. OF i:}} = -32 = 225 \text{dy} - 150 \text{dz}$ = -225 dy - 50 dz(2)

COSF, OF &: 00+144= 150 dy +50dy (3)

THEN EQ. 1, YIELDS - 32 = 225014 Ny= - 32/25 EQ. 3: Qx + 144 = 0 + 50(-32/225)

 $a_A = -144 - 2.111$ $a_A = -(15/1) - 2.111$



FROM PROB 15. 204, WE RECALL: UB = (24in, 8) =

[A/3 = -(25in) \(\delta + (20in) \) \(\delta - (8in) \) \(\delta \) \(\delta - (12 in) \) \(\delta - (12 in

2,1=+72++(-80,-200)+(-250+80,)++(200x+2500)+2 + 36 6-28.83

$$\frac{COEF, OF \stackrel{.}{=} : -36 = -8 \alpha_{y} - 20\alpha_{z}}{COFF, OF \stackrel{.}{=} : \alpha_{A} + 28.8 = 8\alpha_{z}} - 25\alpha_{z}$$
 (1)

$$\frac{\cos p, \ oF \ j}{\cos p, \ oF \ R}; \quad \alpha_A + 28.8 = 8\alpha_R \qquad -25\alpha_R \qquad (2)$$

$$\frac{\cos p, \ oF \ R}{\cos p, \ oF \ R}; \qquad 12 = 2\alpha_R + 25\alpha_R \qquad (3)$$

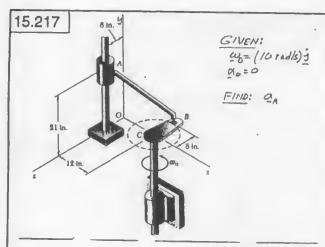
d IS INDETERMINATE: ASSUME dy=c:

FQ. 1: -36 = 0 -2002 d2=1.8

FO.2: OA + 28.8 = 8(3.6) - 25(1.8)

QA = -28.8 + 28.8 - 45

an = -(45 m/s2) j



WE NOW CALCULATE: UB= WOXYB= 10jx-8k=-(8011/3) &

aB= WOXYB= 10jx-80i=(80011/3) &

d IS INDETERMINANT:

ASSUME dy = 0 Eq. 1: -174,15 = 0 -21 dz

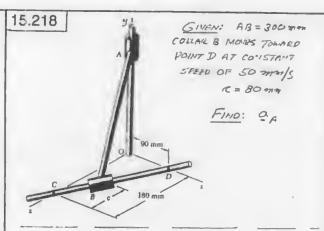
d2=8.293

EQ.3: -800 = 210x+0

dy = -38,095

EQ, 2) $Q_A = 30\%, 76 = -16(-38,095) - 12(8,293)$ $Q_A = 30\%, 76 = 609.53 - 99.50$

an= (815 in./5")j



C= 80 mm: From Pros. 15.206 WE RECALL:

15 = +(22.36 mm/s) & -(44.72 mm/s) fe

60 = (1.118 rad/s) & -(0.3194 rad/s) &

5 A/B = -(40 mm/s + (280 mm/s) -(100 mm/s) fe

1/A/B= - (22,37 mm/s) + (12,776 mm/s) + (44.72 mm/s) -

and = (-100 dy -280 ap) i + (-40 dp +100 dx) i + (280 dx +40 dy) fx + (50.0 +4.08) i +7.15 i +25 fx

d IS INDETERMINANT: ASSUME dy=0

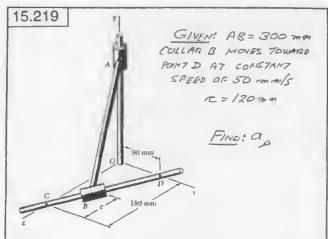
Eq. 1: -54:00'=0-280 % 02 20.19314

 $FQ3: -25 = 2800_2 + 0$ $V_2 = -0.0893$

x=-(0.0893 rad/53) + (0.19314 rad/53) &

EG. 2: $Q_A - 7.15 = 100(-0.0893) - 40(0.19314)$ $Q_A - 7.15 = -6.93 - 7.73$ $Q_A = -9.51$

an = - (9.5/mm/s2) j



K= 120 mm: FROM PROB. 15.207 INE EECALL: 0/ = + (22.36 mm/s) L - (4472 mm/s)-8 w = (0.7453 rails)j - (0.0777 rad/s) -A TA/B= - (60 mm) L+ (287,75 mm, j-(60 mm) R UA/R = WXYA/B = 0 0.7453 -0,0777 -60 = (44.718 + 22.308) [+ 4.662 - + 44.718-8

SAIR =-(22.36 mads) & + (4.662 mads) + (44.718 mads) - R a= 0

$$a_{AJ} = a_{B} + a_{X} r_{A|B} + a_{X} r_{A|B}$$

$$a_{AJ} = 0 + \begin{vmatrix} L & j & -\frac{1}{2} \\ a_{A} & a_{3} & a_{2} \\ -40 & 287.75 & -60 \end{vmatrix} + \begin{vmatrix} L & j & \frac{1}{2} \\ 0 & 0.7453 & -0.0777 \\ -22.36 & 4.612 & 44.718 \end{vmatrix}$$

anj = (-600/y-28?750/2)i+(-600/2+600/)j+(287,50/+600/)/2 +(33.33+0.363)4+1.7373+16.66-4

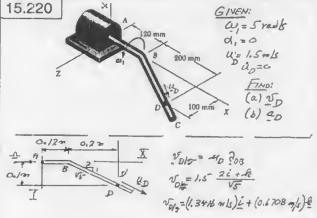
-33,69 = -60dy -287.75d2 COEF, OF L: COEF. OF 1; ag-1.737 = 600 - 60 dz -16.66 = 287,750g +6004 COGF, OF .PL:

& IS INDETERMINANT: ASSUME Xy = 0 EQ.1: -33,69=0-287,7502: az = a//7/ EQ.3: -16.66 = 287.75 d2 +0 ; d = - a0579

Q=-(0.0519 radk) =+ (0.1171 radk)-&

ay - 1,737 = 60(-0.057); -60(0.1171) EQ. 2: ay -1.737 = -3.474 - 7.026 ay =-8,76

ay=-(8.76 ma/s2)5



Yola = (0.320 m) + (0.1m) & -A = W, = (5 radk) L

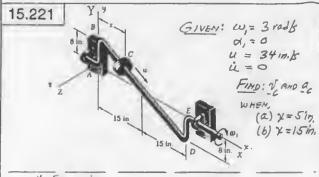
(a) VELOCITY OF D: Up=1xxpy=5ix(0,32i+0.12) 201 = - (0.5 m/s) 2

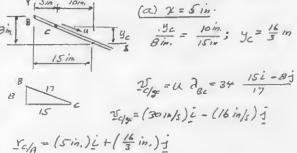
20 = 201+ 2018 = -0.53 +1.3416 i +0.628 A Tp= (1.342m/s)i-(0.5m/s) + (0.67/m/s) &

(b) FICCELLERATION OF D: app=0; 1=0 an = ix Toya + ix x x x Toya = 0+ 1×101 = 5'L x (-0.5j) = - (2.5 m/s2) &

ar= 21x Toy= 2(5i)x(1.34/6i+0.6708 +)=-(6.7/~/5)) 20= a0, + a0/3 + ac

=-2.5.4 +0-6.715 an=-(6.71m/s)j-(2.5m/s)-R

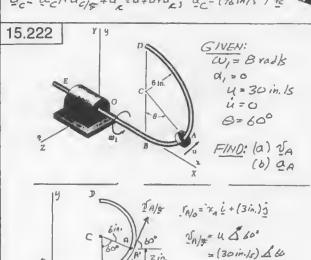




_n= w, i =- (3 rad/s) i

(CONTINUED)

15.221 continued 2=51n. VELOCITY 20, = - 1 × re/A=(-31045) i x [(5/n.) i + (16/n.)) = - (16/n./5) & 7e= vc1+ vc/g= -(61m/s) ++ (30m/s)=-(161m/s) ; Vc= (30 in. b)i - (16 in/s) j - (16 in/s) &. ACCELERATION: 00/8=0; 1=0 ac = Ux Ldu + Ux Tx Ldu = Ux 2014 + Ux x CI ac1 = 0 + (-3 rad/s) by (+16 in./s) & = - (48 in./s)) ar= 2-1x 26/5= 2/- 3 rad/s) Lx (3014/5) i-(1614/5) j a= (96 in./5) & ac= ac+ ac/ + + ac = - (48 17./53) 1+0+ (96 11/53)-12 a=-(48 in 152) j +(96 in/52) & (b) FOR, X = 15 in, (COLLAR C IS IN X2 PLANE) VELOCITY; FROM PART a: Vely= (3017/s) i-(16 11/s) & TO(= (15 in) L; VC1= 1 x [4A = - 3 L x 15 + = 0 2= 20+ 2== 0+24x; V= (30 in. ls)i-(16 in/s)] ACCELERATION: 00/2=0; 1=0 aci= 1x rea + 1x ve = 0+0; aci=0 ax is same AS IN PART a: QE = (96 in./s) & Oc= aci+aci+a=0+0+0; ac=(9611/52) & 15.222 GIVEN: W,= Bradk 4=30 in. 15 u=0 B= 600 FIND: (a) VA



1= w, L = (8 rod/s) i VAI = -1 × (A/6 = & E × (xA E+3j) = (24 m./s) & (a) VELOCITY: 2/A = NA, +NA/5 = 24/8+15/+25.98 j VA = (15in/s) & + (26.0 in/s)) + (24in/s) &

VA/2= (15 in./s) + (25.98 in/s) j

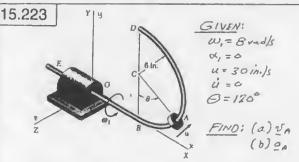
(CONTINUED)

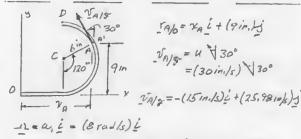
15.222 continued (6) ACCELETATION: 1 = 6

an/7= L D30= (3014/2) D30= 12014/2 D30. Q-A/g==-(127910/52) + (7510/52) 5

QAI = NYTONO + DX-DXTAB = NXTONO + DX VAI QA: = 0 + (8 rad/s) i x (24 in 1s) & = - (192 in/s2)j ac = 2 - 1 x VA/ = 2 (0 rad/s) L x [(1510/s) + (25.9810/s) j] a= (415,7 m/s) &

DA= QAI+ QA/3++ak = - (192 in./5°)-j - (129.91 m/5°) + (75 m/5°) + (415.7 in/5°) & a = - (129.9 lu/s) i - (11 in/s) j + (416 in/s) &



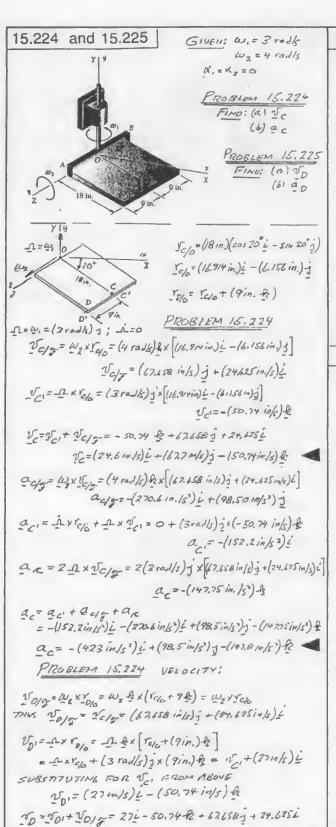


VAI = IL x rate = 8 ix (2 a i +91) = (72 in/s) & (a) VELOCITY: NA = 2/A1 + 2/A/2 = 72-8 -152 + 25.88 j

VA = - (15 in. k) i + (26.0 in/s) + (72 in/s) &

(1) ACCELERATION: _L=0 any = 42 730 = (3011/5) 730 = 15011/5 7 an/3 = - (129.9 in/s2) i - (75 in./5) j THI = - UX LAP + - UX LAP = - UX LAP + - UX RAP an=0+(Brad/s)ix(72 in./s) = -(576 in/s)) a=2.1 x V/3= 2 (& rad/s) & x [-(15iv/s) & + (25.98 in/s)] a = (415.7 in. 152) &

an= an+anytae = - (57611/5) j - (129,9 in/5) i- (75 in./5) j + (415.714/5) & a=-(129.9 in./s') =-(651 in./s') + (416 in./s') }



Vp = (57.6 in/s) + (67.7 in/s) = -(50.71n/s) &

(CONTINUED)

15.225 continued

ACCELERATION 1:0

2013= (4 rad/s) + x[(62658 in/s)]+ (24.625 in/s)]

2013= -(270.63 in/s)]+ (98.5 in/s)]

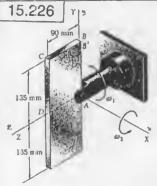
NOTE ! 40/3 = 443 SINCE . VOTE : VOTE

 $Q_{D} = \frac{1}{12} \times \frac{1}{10} + \frac{1}{12} \times \frac{1}{10}$ $= O + (3 \text{ rod/s}) \frac{1}{2} \times \left[(27 \text{ in.k}) + (52.77 \text{ in/s}) \right]$ $Q_{D} = -(81 \text{ in/s}^{2}) + (152.7 \text{ in/s}) + (152.7 \text{ i$

QR = 2.1 \$ \$\frac{1}{2} BUT ING KNOW THAT \$\frac{15}{2} \rightarrow \frac{1}{2} \rightarrow \frac{1}{2

an=an+an+ar =-812-152.21-270.631+98.53-147.75fz

ap=-(423 in/s'16+(98.5 in/3)5-(229 in.13)&



GIVEN: $\omega_1 = 9 \text{ rad/s}$ $\omega_2 = 12 \text{ rad/s}$ $\alpha_1 = \alpha_2 = 0$

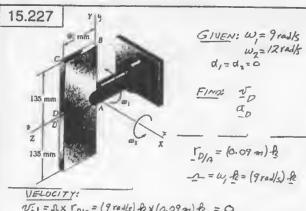
FINO: VB

VELOCITY: VELOCITY: VELOCITY: VELOCITY: $V_{EV} = -\Omega \times I_{B/A} = (9 \operatorname{radk}) & \times (0.135 \text{ m/s}) = -(1.215 \text{ m/s}) \\ \dot{U}_{EV} = -\Omega \times I_{B/A} = (12 \operatorname{rad/s}) \\ \dot{U} \times (0.175 \text{ m/s}) + (1.62 \text{ m/s}) \\ \dot{U}_{EV} = -\Omega_{2} \times I_{B/A} = (12 \operatorname{rad/s}) \\ \dot{U} \times (0.175 \text{ m/s}) + (1.62 \text{ m/s}) \\ \dot{U}_{EV} = -(1.215 \text{ m/s}) \\ \dot{U}_{EV}$

ACCEL FRAZION:

 $a_{g} = \sum_{n} \sum_$

 $= -(10.935 \text{ m/s}^2) \frac{1}{3} - (19.44 \text{ m/s}^2) \frac{1}{3}$ $\alpha_B = -(30.4 \text{ m/s}^2) \frac{1}{3}$

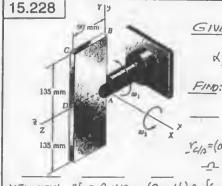


 $\frac{\sqrt{ELOCITY!}}{\sqrt{D}} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} \times \frac{1}{2} = \frac{1}{2} \times \frac$

GDI = T- X T- X LOW = -T- X JDI = TYO = O

apy = w2 x w2x x DA = w2 x DD/g ap/ = (12.8 m/s) & = -(12.8 m/

a = a 0 + a 0/3 + a x = 0 - (12,96 m/s²) - 2 + (19.44 m/s²) i ac = (19.44 m/s²) i - (12.96 m/s²) fe



GIVEN: $\omega_1 = 9 \text{ rad/s}$ $\omega_2 = 12 \text{ rad/s}$ $x_1 = \alpha_2 = 0$

FIND: To

1/d/2=(0.135-) j+10.09m) & -1 = W, &=(9rad/s) &

VELOCITY! U, = 1-x YCA = (9 rod/s) & x (0.135m) & + (0.09m) &]

Y(1 = - (1.215m/s) L

Ic/y= W= X Ic/n = (12 rod/s)ix [(0.135 m) i+(0.09 m) A]

Ic/y= (1.62 m/s) A - (1.08 m/s) j

\$ = \$ c + \$ Cfg: \$ C = -(1.215 m/s) i - (1.08 m/s) j + (1.62 m/s) fr

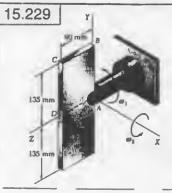
act = mx x mx x c/a = mx x sc/2 = (10.95 m/s) j

act = (12 rod/s) i + [1.62 to -1.08 j] = -(19.44 m/s) j - (12.96 m/s) to

ac = 2 1 x v c/z = 2 (9 rod/s) to x [1.62 m/s) to - (1.06 m/s) j

ac = (19.44 m/s) i

ac= ac+ 1 ac/3+ 0x = -(10.95 m/s) j -(19.44 m/s) j -(1296 m/s) /2+(19.44 m/s) c ac=(19.44 m/s) /2-(30.4 m/s) j -(12.12 m/s) /2



GIVEN: $\omega_1 = 9 \text{ rad/s}$ $\omega_1 = -45 \text{ rad/s}^2$ $\omega_2 = 12 \text{ rad/s}^2$ $\alpha_2 = -60 \text{ rad/s}^2$

FINO: De

 $\frac{\Lambda}{\omega_{R}} = \omega_{1} \frac{1}{L} = (9 \text{ rad/s}) \frac{1}{R} \qquad \frac{\Lambda}{\omega_{1}} = \omega_{1} \frac{1}{L} = -(45 \text{ rad/s}^{2}) \frac{1}{R} \\
\frac{\omega_{R}}{\omega_{2}} = \omega_{2} \frac{1}{L} = (12 \text{ rad/s}) \frac{1}{L} \qquad \frac{\omega_{1}}{\omega_{2}} = \frac{1}{2} \frac{1}{L} = -(60 \text{ rad/s}^{2}) \frac{1}{L} \\
\frac{V_{1}}{\omega_{1}} = (0.135 + 1) \frac{1}{L} + (0.09 \text{ m}) \frac{1}{R}$ (F) O(17):

1200174: V_1 = 1 x [4] = (9 rad/s) & x [(0.135 m) j + (0.09 m)] V_1 = - (1.215 m/s) £

Delig = W24 14A = (12 radis) i x [(0.135m) j+(0.09m)]

Vc = Vc + Vely Vc = -(1.215 m/s) i - (1.08 m/s) f +(1.62 m/s) f

ACCE LERATION

O-CI = -1. x (1/4 + 1 x 1 x (1/4 = 1.) x (1/4 + -1 x VC)

= - (45 red/s) & x (0.135m) + (0.09m) &]

+ (9 red/s) & x (-1.215m/s) &

ac1=+(6.075m/52)i-(10.94m/62)j

acy = w2x (c/4 + w2x w2x (c/4 + w2x 2c/2 - (-60 red/s)) ix [(0.1350) j + (0.09 m) &]
+ (12 rad/s) ix [(1.62 m/s) & - (1.08 m/s) j]
= - (8.10 m/s) & + (5.4 m/s) j - (19.44 m/s) j - (12.96 m/s) k

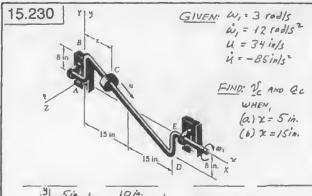
acy = - (14.04 m/s) j - (21.06 m/s) B

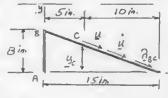
ax = 21 × Vc/= 2 (9 rad/s) Ax [(1.62 0/s) & - (1.08 m/s)]

ar = (19.44 m/s)

= (6.075 m/s) i - (10.94m/s²) j -(14.04 m/s²) j -(21.06 m/s²) f + (19.44 m/s²) j

Q = (25.5 m/s) L - (25,0 m/s) f - (21.1 m/s) A





 $\frac{(a) \chi = 5 \text{ in.}}{\frac{4c}{8 \text{ in.}} = \frac{10 \text{ in.}}{15 \text{ in.}}; g = \frac{16}{3} \approx \frac{17}{15}$

$$\begin{split} & \underline{\mathcal{I}}_{C/p} = \mathcal{U} \, \underline{\mathcal{I}}_{Bc} = \left(34 \, \text{in./s} \right) \frac{15 \, \hat{\iota} - 8 \, \hat{\jmath}}{17} = \left(30 \, \text{in./s} \right) \, \hat{\iota} - \left(16 \, \text{in./s} \right) \, \hat{\jmath} \\ & \underline{\mathcal{I}}_{C/p} = \, \hat{\mathcal{U}} \, \underline{\mathcal{I}}_{a} = \left(-85 \, \text{in/s}^2 \right) \frac{15 \, \hat{\iota} - 8 \, \hat{\jmath}}{17} = - \left(75 \, \text{in/s}^2 \right) \, \hat{\iota} + \left(40 \, \text{in./s}^2 \right) \, \hat{\jmath} \end{split}$$

 $\frac{V \in Locaty;}{\int_{C_{i}}^{C_{i}} = \int_{C_{i}}^{C_{i}} = \int_{C_{i}}$

ACCELERATION: QC/2, SEE ABOVE

QC = 1 x 5 c/A + 1 + 1 x x 1 c/A = 1 x 1 c/A + 1 x 1 C

= (-1200/5°) ± x [(5/n) ± + (4/2/n) 5] + (3 rad/5) ± x (16 in/5) &

QC = - (64 in/5°) Ac - (48 in/5°) 5

an = 2.1. x Volg= 2(-3 10 16) ix (30 in 16) i - (16 in 16) j] = (96 10/5) &

=-(64 in/s3) Pa-(48 in/s3) j-(75 in/s3) +(40 in/s3) + (96 in/s3) 2

0-=-(75in/s*)!-(8in/s*);+(32in/s*)!

VELOCITY: VC/g = SAME AS IN PART a AROVE

V = VC + VC/g = O+V/g: V = (30in/s) L - (Kin/s)j

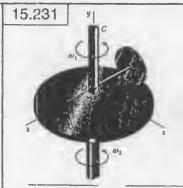
ACCELBRATION:

ac/g = SAME AS IN PART a ABOVE

ac/g = C; SINCE COLLAR LIES ON AXIS OF RUTATION

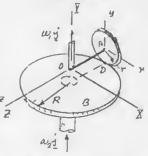
ac = 21 x Telg = SAME AS IN PART a ABOVE

ac=ac1+ac1g+an =0-(75in/s)i+(40in/s)j+(96in/s)+ ac=-(75in/s)i+(40in/s)j+(96in/s)+



GIVEN. W. = W. K. 01, = 0 W2 = W2 /2 02 = 0

FIND: FOR DISK,
(a) WA
(b) & A



MOVING FRAME A 242 ROTATES 41714 ANGUM VELOCITY I = W, J WOISH/= W, L+W, A

10/a = - 13 - RA

(1) TOTAL ANGUAR VELOCITY OF DISK A:

W = W, j + Wask/g = W2 + w, j + w2 ? (1)

DENOTE BY D POINT OF CONTACT OF DISK

CONSIDER DISK R: $V_0 = \omega_2 \int y \left(-R P_1\right) = -R \omega_2 L$ (2)

CONGION SYSTEM OC, OA, AND DISKA.

Vo, = . 1. x (0/4 = w, j x (-rj-Rh) = - Rw, i

Vo/5 = w / No = (w, i+w, h) x (-rj-Rh)

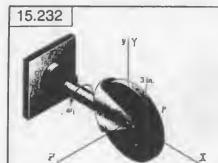
= -ray h + Rw, j + rw, i

$$\begin{split} &\mathcal{L}_{0} = \mathcal{L}_{0} + \mathcal{L}_{0} / \mathcal{L}_{z} = -R \, \omega_{z} \dot{L}_{z} - r \, \omega_{x} \, \dot{R}_{z} + R \, \omega_{x} \dot{J}_{z} + r \, \omega_{z} \, \dot{L}_{z} \, \dot{L}_$$

EG3: $\omega = \omega_1 j + \frac{R}{r}(\omega_1 - \omega_2) + \frac{R}{r}$

(b) DISK A ROTATES ABOUT Y AXIS AT RATE W, $d = w_1 \times w = w_1 \cdot x \times \left[w_1 \cdot \frac{1}{r} + \frac{R}{r} \left(w_1 - w_2 \right) \right] R$

x= w, (w,-w2) R i



GIVEN: \[\overline{\pi_1 = 5_{10} \right|_{5} \, \pi_1 = 0}{\pi_2 = 4_{10} \right|_{5} \, \pi_0 = 0} \]
\[\overline{\pi_2 = 4_{10} \right|_{5} \, \pi_0 = 0}{\pi_0 = 3_0^0} \]

FINO: a

FRAME OXYZ IS FIXED. MOVINE PRAME Dryz ROTATES

WITH AHBURAR VELUCITY __ = W, L = (5 rad/s) L

*Plo = (3in.)cos 30" L + (3in.) sin 30";

= (7.598 in.) L + (1.5 in.) f

- Dsx/g = Anh = (4 rad/s) &

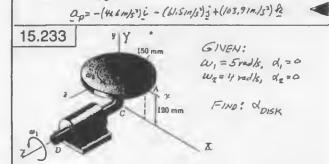
TP/7 = W DISU/5 × 5P/0 = (4 rad/s) & x [2.598 i +1.55]

YP/8 = (10.392 in/s) j - (6 in./s) i

YP/8 = 1 x y P/0 = (5 rad/s) i x [2.598 i +1.55]

NO = (25 in/s) &

ACCELEDATION: $\alpha_{pi} = \Omega + U_{pi} = (Siads)i \times (7.5 m/s) 2 = -(325 in/s^2) j$ $\alpha_{p/g} = \omega_{Disin/g} = \Omega_{p/g} = (4 rads) 2 \cdot [(10.292 in/s)j - (6 m/s)i]$ $\alpha_{p/g} = -(41.569 in/s^2) i - (24 in/s^2) j$ $\alpha_{r} = 2 \Omega \times \Omega_{p/g} = 2 (5 rad/s) i \times [(10.392 in/s)j - (6 in/s)i]$ $\alpha_{r} = (103.92 in/s^2) 2$ $\alpha_{p} = \alpha_{pi} + \alpha_{p/g} + \alpha_{e} = -37.5 j - 41.569 i - 24 j + 103.92 2$



FRAME CXYZ IS FIXED

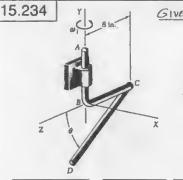
MOVING FRAME BYGG ROTATES INITE ANGULAR VELOUTY _A = W, & = (Srad/s)& AROUT 2 AXIS

work = w, fe + was

X DIZX = T x MDIZX = M' & x (M' + + M' 2) = -M' M' F

doisk = - (sradk) 4 radk) i

Opisk = - (20 rad/s 2) 6



GIVEN: CD = L = 16 in. $\Theta = 30^{\circ}$ $\omega_1 = 4$ rad δ $\omega_2 = \frac{d\theta}{dt} = 3$ rad δ

FIND: 25 AND ap

TOIS - L SING j + (LCOSO &)

VELOCITY: $V_D = \Delta \times Y_{D/B} = 0$, $j \times \left[-1 \sin j + L(\cos 0 - \frac{1}{4}) \cdot \frac{R}{3} \right]$ $V_{D} = L v_1 \left(\cos 6 - \frac{1}{4} \right) \cdot \frac{1}{4}$

VD/7 = 02× 10k = 02i+ (-15in 6 i+ Lcose A)

Voly = -Lw2 sin 6 R - Lw2 cose s

V= V0, + V0/7 √D = LA, (COSO-1)i-Lw20000 j-Lw251000 f

Account tion: $\alpha_D = \Lambda \times V_D = \omega, \hat{\mathbf{j}} \times L\omega, (\cos \theta - \frac{1}{2})\hat{\mathbf{L}}$ $\alpha_D = -L\omega,^2(\cos \theta - \frac{1}{2})\hat{\mathbf{R}}$

20/3 = +Lw2 sin & j - Lay cose & - Luz cose j)

20/3 = +Lw2 sin & j - Lay cose &

ar = 2-1 x v D/3 = 2 w, j x (-Luz sin & L - Luz cose j)

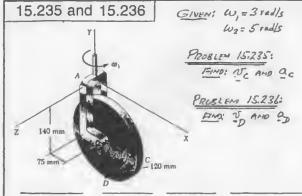
ar=-2LW, Wasing i

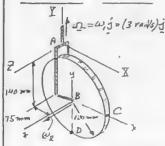
 $Q_{D} = Q_{D} \cdot T Q_{D}/3 + Q_{C}$ $= -L u_{1}^{2} (\cos \theta - \frac{1}{2}) \frac{1}{2} + L u_{2}^{2} \sin \theta - \frac{1}{2} - L u_{2}^{2} \cos \theta - 2 L u_{2} u_{2} \cos \theta - \frac{1}{2}$ $Q_{D} = -2L u_{1} u_{1} \sin \theta + L u_{2}^{2} \sin \theta + \frac{1}{2} + (-L u_{1}^{2} (\cos \theta - \frac{1}{2}) - L u_{2}^{2} \cos \theta) \frac{1}{2}$

DATA: 6=30, L=16 io., $\omega_1 = 4 \text{ rad/s}$, $\omega_2 = 3 \text{ rad/s}$ $\overline{U}_5 = 16(4)(\cos 30^\circ - \frac{1}{3}) \dot{L} = 16(3)\cos 30^\circ \dot{g} = 16(3)\sin 20^\circ \dot{g}$ $\overline{U}_0 = (23.4 \text{ in./s}) \dot{L} = (44.6 \text{ in./s}) \dot{g} = (24 \text{ in./s}) \dot{g}$

20= -2(16)(3)(4)sin 30 + 16(3) 51+30 -5 +(-16(4) (cos30-1)-4(3) cos30)

ap - - (192 in./5) + (72 in./5) + - (218 in.15) +





FRAME AXYZ IS FIXED MOVING FRAME BYZZ
ROTATES ABOUT YAYIS
WITH 1 = (3rad/s) 3
W2=(5rod/s) 8

PROBLEM 15.235: FOR POINT C - [4= (195mm)i-(140mm)j; 5c18= (120mm)i VELOGITY:

TC = 1 x fc/a = (3 rad/s) jx (1950-140j) = - (585 ma/s) &

Vc/z= Wx * rc/z = (5 rad/s) & x (120i) = + (600 ma/s) j

Vc = V1 + Jc/z Vc = (600 ma/s) j - (585 ma/s) &

ACCEL BE ATION

 $Q_{c1} = \underline{\Omega} \times \underline{U}_{c1} = (300 \text{ dls}) \underline{\dot{s}} \times (-586 \text{ mayb}) \underline{\dot{h}} = -(1.755 \text{ m/s}^2) \underline{\dot{c}}$ $Q_{c1} = \underline{U}_2 \times \underline{U}_{c1} = (5 \text{ rad/s}) \underline{\dot{h}} \times (600 \text{ mayb}) \underline{\dot{s}} = -(3.00 \text{ m/s}^2) \underline{\dot{c}}$ $Q_{c2} = 2\underline{\Omega} \times \underline{U}_{c1} = 2(3 \text{ rad/s}) \underline{\dot{s}} \times (600 \text{ mayb}) \underline{\dot{s}} = 0$ $Q_{c2} = Q_{c1} + Q_{c2} + Q_{c2} = -(1.755 \text{ m/s}^2) \underline{\dot{c}} - (3.00 \text{ m/s}^2) \underline{\dot{c}}$ $Q_{c2} = -(4.76 \text{ m/s}^2) \underline{\dot{c}}$

PROBLEM 15.736: FOR POINT D YO/= (75mm)i-(260mm)j; YO/8 = -(12cmm)j

Do = 1 x 10/2 (3 rad/b) j x (75 L - 260 j) = -(225 mank) & ED/2 = W2 x 10/3 = (5 rad/s) & x (-120 j) = (600 man/s) L

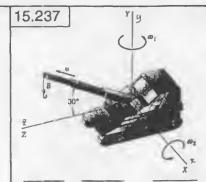
Te = Ve + Ve / 2

V = (600 man/s) L - (225 man/s) &

ACCELERATION $Q_0 = A \times 20_0 = (3 \text{ rad/s}) j \times (-225 \text{ mm/s}) l_0 = -(0.65 \text{ m/s}) l_0 = 0.65 \text{ m/s} l_0 = -(0.65 \text{ m/s}) l_0 = 0.65 \text{ m/s} l_0 = +(3.00 \text{ m/s}) l_0 = 0.00 \text{ m/s} l_0 = +(3.00 \text{ m/s}) l_0 = 0.00 \text{ m/s} l_0 = -(3.60 \text{ m/s}) l_0 = 0.00 \text{ m/s} l_0 = 0$

ac=-(0.675 m/s) i+(3.00 m/s) j-(360 m/s)-&

ac=ac+ +ac/8+ac



GIVEN: $\omega_1 = 0.75 \text{ mad/s}, \alpha_1 = 0$ $\omega_2 = 0.70 \text{ rad/s}, \alpha_2 = 0$ AB = 20 ft U = 1.5 ft/s $\dot{u} = 0$

FINO: No AND QB

TBI = ILX (BIR = (0.25 rad b) jx (10 j + 17.32 R) = (4.33 ftb) L

TBI = U + W2 × TBI = (0.75 + 1.299 R + (0.001) × (10 j + 17.72 R)

= 0.75 0 + 1.299 R + 4 R - 6.97 R J

U3/2=-(6.172 (t/s) j+(5.299 (4)) } V= V8,+28/2= (4.33 (t/s)i-6.178 (1/s)j+(5.299 (1/s)) & V= (4.33 (t/s)i-(6.18 (1/s)j+(5.30 (1/s))) & ACCELERATION:

as=-1×581=(0.25rad) jx(4.33fts)i=-(1.083f4s2)&

CORIOL CORIOL

COINCIDING POWT AS AS OF 8 MOVING AT U. AS ROTATES WITH W2 ABOUT AND AB ROTATES ABOUT & AXIS

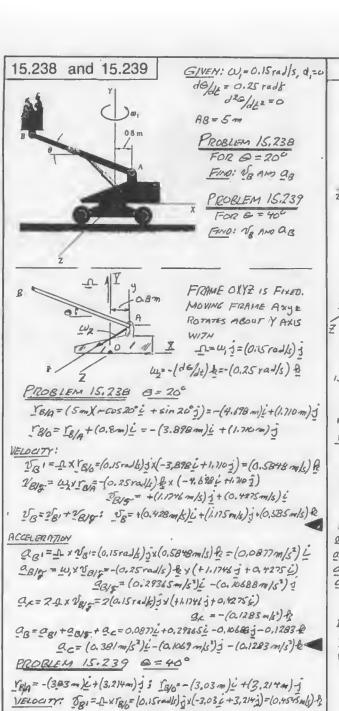
(28/2=(0.4i)x(0.4ix(10j+17.82k))+2(0.4i)x(0.75j+1.29/2) =(0.4i)x(.4+2-6.928j)+0.42-1.0392j =-1.6 j-2.77/2+0.6-2-1.0392j

CR = CORDUS ACCELERATION OF B MOVING WITH
FRAME A RYZ 'IN ROTATION AGOUT Y WITH
ANGULAR VELOCITY ________

 $2c = 2 \Omega \times 28/3$ = $2(0.25 \text{ rad/s}) \frac{1}{2} \times [(6.178 \text{ fHz}) \frac{1}{2} + (5.299 \text{ fHz}) \frac{1}{12}]$ = $(2.650 \text{ ft/s}^2) \frac{1}{6}$

 $\begin{array}{lll}
(2) & = & 281 + & 281 = + & 2/2 \\
& = & -(1.083 \text{ ft/s}^2) + & -(2.639 \text{ ft/s}^2) - (2.171 \text{ ft/s}^2) + \\
& + & (2.650 \text{ ft/s}^2) - & -(2.639 \text{ ft/s}^2) - & -($

aB = (2.65 At/s) i - (2.14+t/s) j - (3.25 At/s2) &

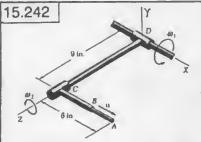


VIELOCITY: 581=1-x186= (0.15 rad/s) (x(-3.036+3,214))=(0.4545 mb)-2 NB/3= W2X[BIA= (0,25rad/s) &x (-3,83 i + 3,214 j) VB/3=+(0.9576~/s)+(0.8035m/s)L NB=NB+NB/5: NB=+(0.803 nk) (+(0.958 m/s) j+(0.455 m/s) & ACCELERATION: QBI = 1xVBI = (0.15 rad/s) (x(0.4545m/s) & = (0.0682 m/s2) = QQq= w, x2Q==(0,25 rad/s) & x(+0.95 76 +0.8035 i) 28/7= (0.239+ m/52)6-(0.2009 m/52)5 Q = ZA x 28/5 = 2(0.15 rad/s) fx (+0,75% j +0.80356) ar=-(0,240m/s)R QB = aBit 98/8+ 19/ = (0.0682m/5) i+(0.259+m/5) i-(0.2009m/5)-j-(0.2410m/5) tz aB = (0.308 m/s2)i - (0,201 m/s2) 5 - (0.24 m/s2) &

15.240 and 15.241 GIVEN: W= Bredly, 0,=0 Wy=12 rad/s, 0/2=0 PROBLEM 15.240 FIND: VA AND CLA PIZOBLEM 15.241 Finn: 15 AND aB FRAME DXYZ IS FIXED. 180 mm MOVING FRAME CZYZ D=2/1 ROTATES ABOUT D Y AXIS WITH IL=Wj=(Brod/s)J w= (12 rad/s) & 360200 PROBLEM 15.240: FOR POINT A TOM= (0.15=) + (0.18 m) j- (0.36 m) A; YA/c=(0.18m)j VAI = 1 X YOVA = (8 rolls) jx(0.51 + 0.18j - 0.36-8) JA'= - (1.20/s) fe - (2.880) L UA/9= WRX Int = (12 rad/s) &x (0.18 m) = - (2.16 m/s) L 7A=1A1+VA17 =- (1.2 m/s) &-(2.88 m/s) i-(2.16 m/s) i V=-(5.0+ m/s) L- (1.2 m/s) to IA= Dxga=(8 rad/s) jx(-1.2-8-2.86)=-(9.6 a/s2) +(23.04 a/s2) A QA/g= W2XVA/g= (12 rad/s)& (-2.4 a/s)i=-(25.92 a/s) Qx = 2.1 x 1/15 = 2(8 rools) jx (-2.16 m/s) i = (34.58 m/s2) /2 Qx Qx + Qx + Qx = -(96 m/s2) i + (230+ m/s) /2 - (25.72 m/s) j + (34.56 m/s2) a= -(9.6 a/s2)i-(25.9 m/s)j+(576 m/s)& PROBLEM 15.241 FOR DOM'T B TELA = (0.15m) i - (0.18m) j - (0.36m) ft; IB/c=- (0.18m)] TBI = 1 x YB/A = (8 ral/s) jx (a.15 i -0.18 j -0.28) TBI=-(1.2 m/s) & -(2.88 m/s) i JE/g= WaxYB/c= (12 rad/s) &x (-0.18m) = (2.16m/s) L NB = 281 + 18/2= - (12mk)&-(2.88m/s) i+(2.16m/s) i VB = - (0.72m/s) - - (1.2m/s) & ACCELERATION: aB1= 2 x of = (8 rad/s) jx (1.2 k - 2.80 i) aB = - (9.6 m/s) + (23.04 m/6) & QB/g= N2X TB/g= (12 rad/s) AX(2.16 m/s) i= (25.92 m/s) 1 a = 21 x1/8/9 = 2(8 rad/s) 1x (2.16 m/s) = - (34,56 m/s') & Q8 = Q8/+28/8

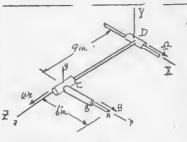
= - (9.6 m/s) + + (23.04 m/s) & + (25.92m/s) + - (34.56 m/s) &

ag= - (9.6 m/s) i+ (25.9 m/s) j - (1/52 m/s)/2



GIVEN: W=1,2 radk, 0,=0 W2=1.5 rad/s, d2=0 4=3in./s, 4=0

FIND: VA AND DA



FRAME DXYZ IS FIRED. MOVING FRAME CZYE ROTATES ABOUT THE YAXE WITH 1= Wi= (1,2 rad/s) i. u= w, & (1.5 rod/s) &

4=4 = (3 in./s) L

YND= (6 in.) + (9 in.) A

VELOUTTI

TAI= 1. x YAID = (1.2 rad/s) ix [(6 in) i + (9 in) to] = -(10.8 in k) j VAJ = W= XTAL + U = (1.5 rad/s) Ax (6in.) i + (3 in/s) i TAI2= + (9 in./s) + (3 in./s) L

TA = 2/A1+2/A1= - (10.8 in/s) i+(9 in/s) j+(3 in/s) 4 TA = (3/m./s) i - (1.8 in./s) 3

ACCELERATION

and NOTE, SINCE POINT A MOVES IN THE ROTATHE FRAME CAYE THERE IS A CORJOIR ACCERETATION.

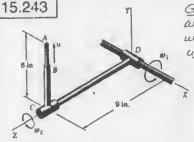
aA/2= w2 x w2 x VAK + 2 WE XIL = W2 x (1- Sradb) & x (6 in,) i + 2 (1.5 radb) & x (3 rads) i = (1.5 radle) & x (+9 in./s) + (9 in/s) 5 2A/2 = - (13.5 in./52) + (9 in/52) +

ac: CORDOLIS ACCELERATION DUE TO A MOVING WITH VELOCITY VALT

ac= 21 y NA/2 = Z(1.2 rad/s) ix [(910.15) j+(310/s) i] ar = (2/6 in./s2) &

a=aAI+aAIz+ac = - (12.96 in 152)-12-(13.5 in. 15) + + (9 in/5) + + (71.6 in/5) 12

Q= -(13,51/s2)i+(9in/s)j+(8.64in./s)A



GIVEN: W,=1.2 rad/s, d,=0 W= 1.5 rad/s, 0=0 4 = 3 in./s, il = 5

NA AND QA

6 in.

FRAME DXYZ IS FIXED. MOVING FRAME CXY2 ROTATES ABOUT THE I AXIS WITH 12-Wi= (1.2 rod/s) i.

w2 = w, R=- (1.5 rolls) & u=uj=(3 in./s) i

TAID = (6in) + 19in.) 4

VELOCITY:

VAI= 02 XYAI= - (1-5 rad/s) & x (6 in.) + + (3 In./5) + 1/A/5= + (9in./s) + (3in./s) +

TA= 1/4+ 2/1= (7.2 m/s) &- (10.8 in/s) = (9 in/s) + (3 in/s) + No = + (9in./s) 1 - (7.8 in./s) 1 + (72 in/s) & ◀

ACCELER ATIONS 071= 1 x 1 x Yalp = 1 x Ma = (1.2 rad/) (x (22 in/s) / - (10.8 in 6)) an = - (8.64 in 153) \$ - (12.96 14/53) &

QAIQ: NOTE SINCE POINT A MIDNES IN THE ROTATING FRANE CZYZ THERE IS A CORIOLIS ACCELERATION

and = waxwax TAK+ 2 wax u = W2 x (1.5 rad/s) 2x y (61m.) 1 + 2 (1.5 rad/s) &x (3 rad/s) 3 = (15 rails) + x (-9 in./s) = - (9 in./s) =

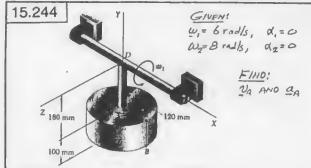
EAIG= - (13,5in,/52) j - (9in,/52) 4

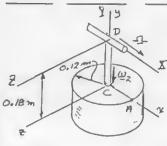
ar: CORIOLIS ACCELERATION DUE TO A MOVING INITH VELOCITY VAIS

an = 2.1. x Jajg = 2(1.2 rad/s) ix [+(9in/s) i+ (3in/s)] ac= (7.2 in/5") &

an = an + an + tax =-(8.64 m./s) 5 -(12.96 m./s2) & - (13.5 in./s2) 1 + 19in/s2) i + (2.2 in/s2) te

an=+(9 in.152) i - (22.1 in/52) 5 - (5.7611.152) k





FRAME DXYZ IS FIXED.

MOVING FRANE, C 245,

ROTATES ABOUT THE

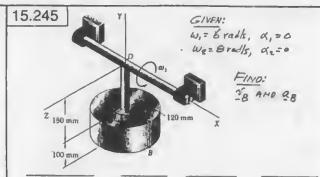
Y AXIS WITH

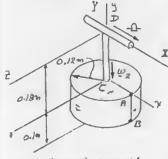
1 = W, L= (brods) L

W= W2j=-(erads) j

VELOCITY:

ACCELERATION:





FRAME DXYZ IS FIXED.

MOVING FRAME, CzyR,

ROTATES ABOUT THE

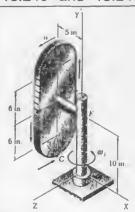
Y AXIS WITH

SL=W, L=(bird/s)L

VELOCITY:

ACCELERATION:

15.246 and 15.247

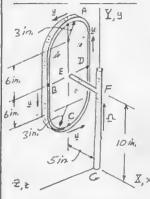


GIVEN: W,=1,6 rad/s, X,=0 LINK BELT MOVES AROUND PERIMETER AT CONSTANT SPEED H=4.5 in./s.

PROBLEM 15.246

FINO: (a) an
(b) an

PROBLEM 15.247 FIMD: (a) ac (b) ab



FRAME GXYZ IS FIXED.
MOVING FRAME, GXYZ,
ROTATES ABOUT THE
Y ANS WITH

\$\Lambda = U_0 = (1.6 \text{ rad/s}) \frac{1}{2}

PROBLEM 15.246: (a) POINT A: 4= (4.5 m./s) & -A/G - (5 in.) L+ (19 in.) j

NA = - 12 × [A/6 = (1.6 rad/s) = x [-(5 in.) + + (19 in.)] = (8 in/s) - 8
2/A/6 = 4 = (4,5 in./s) - 8

VA = VAI + VAIS = (8in.16) & + (4.5in.15) & VA = (125in.15) &

an = - x Jaly = (1.6 rads) jx (8 in, /s) = (12.80 in, /s) L an/y = - u2 j = - (4.5 in./s) j = - (6.25 in./s) j

ax = 2-1 x 4 = 2(1.6 rad/s) jx(4.5 in./s) = (14.4 in./s) i

a= (12.80 in,5) i- (6.75 in,5) + (14.4 in 82) i

QA= (27.2in/s2) = -(6.75 in./s2) j

(CONTINUED)

15.246 and 15.247 continued

PROBLEM 15.2461 (b) POINT B: 4=-(4.5111.16)]

[8/6=-(5in.) + (10in.) + (3in.) +

 $V_{g_1} = \underbrace{1}_{X_{g_2}} \times [g_3 = (\frac{1}{6} \sin \frac{1}{2}) \frac{1}{2} \times [-(\sin \frac{1}{2} + (10 \ln \frac{1}{2}) \frac{1}{2} + (10 \ln \frac{1}{2}) \frac{1}{2}]$ $V_{g_3} = (\frac{1}{6} \sin \frac{1}{2}) \frac{1}{2} \times [\frac{1}{6} \cos \frac{1}{2} + (\frac{1}{6} \sin \frac{1}{2}) \frac{1}{2} + (\frac{1}{6} \sin \frac{1}{2}) \frac{1}{2}]$

VB/0= 4 = - (4.5 11.15) j

 $V_B = V_{B1} + V_{B1} = (8in./s) + (4.8in./s) - (4.8in./s) +$ $V_B = + (4.8in./s) - (4.8in./s) + (8in./s) +$

Q81= 1 × 1/81 = (1.6 ink) j x[(8 in. 15) k + (4.8 in. 16) i]

Q81= 1 × 1/81 = (1.6 in. 15) j x [(8 in. 15) k + (4.8 in. 15) k

 $Q_{S/p} = 0$ $Q_{R} = 2 \cdot 1 \times N_{B/p} = 2(1.6 \text{ in./s}) \cdot j \times (-4.5 \text{ in./s}) \cdot j = 0$ $Q_{R} = Q_{R} + Q_{R/p} + Q_{C} = (12.6 \text{ in./s}^{2}) \cdot i - (2.68 \text{ in./s}^{2}) \cdot k + 0 + 0$ $Q_{R} = (12.8 \text{ in./s}^{2}) \cdot k - (2.68 \text{ in./s}^{2}) \cdot k$

PROBLEM 15.247 (a) POINT C: u=-(4.5 in/s) & Ying = -(5 in/s) + (1 in.) j

 $2[x] = \Delta \times x_{C/6} = (1.6 \ln x_6) \int_{\mathbb{R}} x [-(5 \ln x_6) + (1 \ln x_6)] = (8 \ln x_6) \int_{\mathbb{R}} x_6 (1 + (1 \ln x_6)) dx$ $2[x] = 2[x] \times x_{C/6} = (4.5 \ln x_6) \int_{\mathbb{R}} x_6 (1 + (1 \ln x_6)) dx$

Vc = Vc1 + Vc15 = (8 in./s) &- (4.5in./s) &: Vc = (3.5in./s) &

 $Q_{C_1} = \frac{1}{4} \times \frac{1}{2} = \frac{(1.6 \text{ in./s})}{(3 \text{ in./s})} \times (6 \text{ in./s}) \cdot \frac{1}{12} = (12.80 \text{ in./s}^2) \cdot \frac{1}{2}$ $Q_{C_1} = \frac{u^2}{12} = \frac{(4.5 \text{ in./s})}{(3 \text{ in./s})} \cdot \frac{1}{2} = (6.75 \text{ in./s}^2) \cdot \frac{1}{2}$

ax = 2.1 x 347 = 2(1.6 in. 16) +x (-4.5 in. 15) = - (14.40 in/5) [

ac = ac + ac + ac

 $\alpha_{c} = (12.80 \text{ in } 15^{3})\hat{L} + (6.75 \text{ in } 15^{3})\hat{L} - (14.40 \text{ in } 15^{3})\hat{L}$ $\alpha_{c} = -(146 \text{ in } 15^{3})\hat{L} + (6.75 \text{ in } 15^{3})\hat{L}$

(b) Paint D: 4 = (4,511./s) j YOIG = -(5711.) £ +(1011.) j -(3111.) - R

NDI = -12 x TO/6 = (1.6 in./s) jx [-(5in.) + (10in.) f - (3in.) }]

NDI = -12 x TO/6 = (1.6 in./s) jx [-(5in.) + (10in.) f - (3in.) }]

TO15 = a = (4.5 in/s) =

To = 5/0+ 50/2= (8io./s) & - (48io./s) + (45 w/s) }

V = - (4.8 in./s) + (4.5 in/s) j + (8 in./s) - R

ap= (1.6 rad/s) 1x [(810./s) - (4.8 in./s) -]

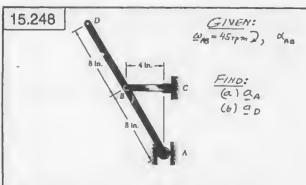
Qp= (12.8 in./s) - + (7.68 in./s) - +

20/5=0

ax = 21 × Volg = 2(1.6 rolls) fx (4.5 in.ls) j = 0

ap = ap + ap/ + ar = (12.8 in./s) + (7.14 in./s) + +0+0

CLO= (12.8 m. /52) + (7.68 in./52) &



CRANK BC:
$$\omega_{gc} = (45 \text{ ypm}) \frac{2\pi}{60} = 4.7124 \text{ rad/s}$$

8 $\omega_{gc} = (80)\omega_{gc}^2 = (4 \text{ in})(4.7124 \text{ rad/s})$
 $\omega_{gc} = 4 \text{ in}$
 $\omega_{gc} = 4 \text{ in}$
 $\omega_{gc} = 4 \text{ in}$

ACCELERATION

D

$$\begin{array}{c}
a_{B} \\
a_{B}
\end{array}$$
 $\begin{array}{c}
a_{B} \\
a_{AB}
\end{array}$
 $\begin{array}{c}
a_{AB} \\
a_{AB}
\end{array}$

15.249 GIVEN: ROTOR IN UNIFORMLY ACCELERATED MUTTON t=0, W= 1800 mm, 0=0

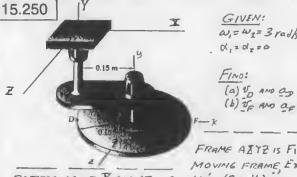
LUED, SE 1550 Yeard whove FINO: (a) &, (b) + REQUIRED TO COME TO MEST.

No=1800 rpm (37) = 188.50 rads 0 = 1550 rev (217) = 9739 mad

(a) AMBULAR ACCELERATION: (USE LAST OF EQS. 15.16) w= w= 2+2d(6-8) 0 = (188.5 rad/s)2+20 (9739 nad -0)

d = -1.824 nad/s2 (b) TIME REQUIRED TO STOP: (USE FIRST OF EQS. 15.16) W= Wo+dt

0 = 188 Snad/s - (1.824 rolls) t t= 103.35



GIVEN: a,= w== 3 radk d, = d, = o FINO: (a) VO AMO ago

FRAME AXYZ IS FIXED. MOVING FRAME EXYE,

ROTATES ABOUT Y AXU AT -1 = W, j = (3 rad/s) j. (a) PONT D: W= W2 j = (3 rad/s) j

TDIA = 0; TDIE = - (a.15 m) & VDI = -1 XTOIA = 0 Vag = WRX TOE = (3 rad/s) jx(-0.15 m) i = (0.45 m/s) R Tp= (0.45 m/s) & で = でかけのりない

ap = 1 x 1 0 = 0 app = w2 x Vo/5 = (3 rad/s) 5x(0.45 m/s) & = (1.35 m/s) L ac= 21 x 1/4g= 2(3 rad/s) jx(0.45 ads) &= (2.20 a/s) i 00= a01+ a0/5+ ap=0+(1.35 m/s) + (2.70 m/s) i ap= (4.05m/s")}

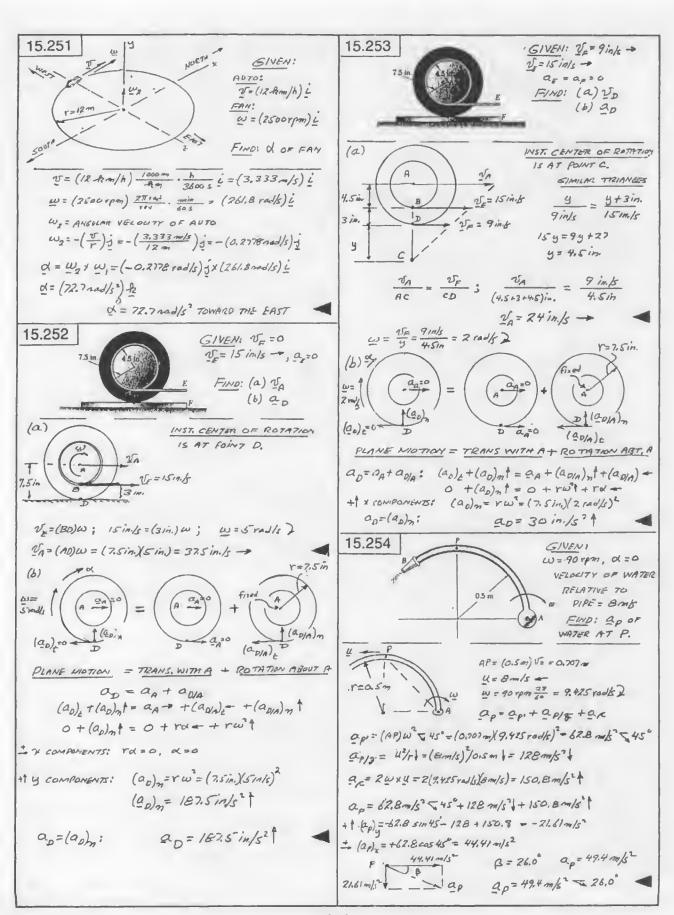
(b) POINT F: 02 = W2 j = (3 rad/s) j TF/A = (0.3 m) ' ; TF/E = (0.15 m) '

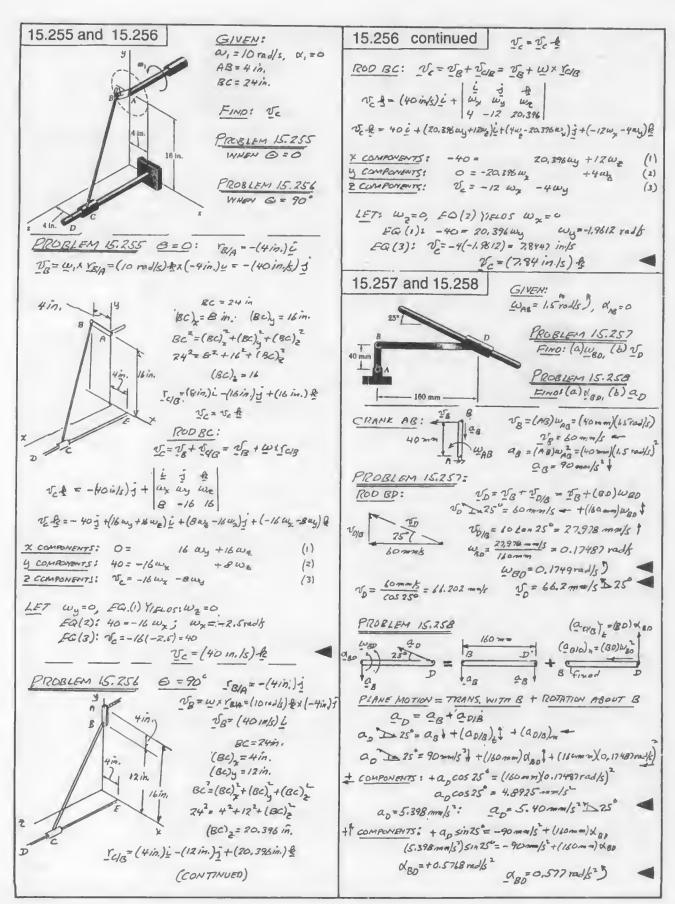
V = 1 × (= 1 = (3 rad/s) jx (0.3 m) i = - (0.9 m/s) & IFIG= W2 X FF/E = (3 rad b) jx(also)i = - (0.45 m/s) - 3

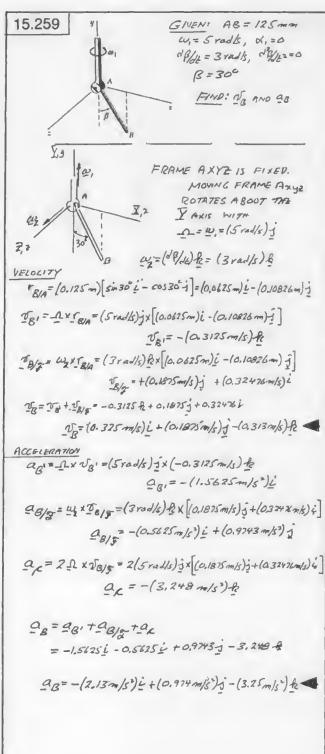
TE= T. + VE/= - (0.9 m/s) A - (0.45 m/s) A V==-(1.35m/s)-R

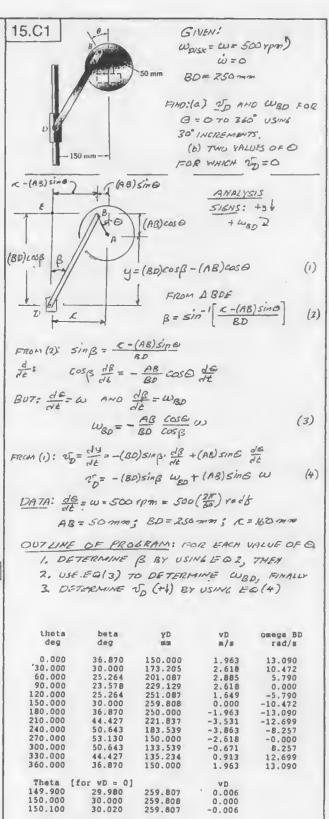
ar = 1 x Vp = (3 rad/s) jx(-0.9 m/s) k= -(2.7 m/s) i aff = w x 1 = (3 rod/s) jx (-0.45 m/s) & = -(1.35 m/s) & ax = 2-1 × 21 F/z= 2(3 rad/s) jx(-0.45m/s) &= -(2.70/s) &

ar= apitafly +ac = - (2.7 m/s) i - (1.35 m/s) i - (2.7 m/s) i a=-(675m/s2)&









Theta

311.400

311.410

[for vD = 0]

48.592

48.590

48.588

132.288

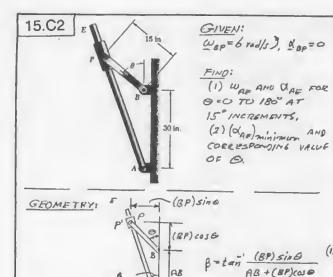
132.288

132.288

VD

-0.001

0.001

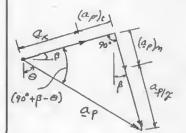


AP=[(BP) SIG] / sing VELOCITYI ROD BP: Up = (BP) WBD TO (3) RUDAF: No = (AP) WAE JA (4) Jp = Vp, + Jp/2 [v, 70]=[v, 78]+[vp/5 A18] VAG= Vpsin(0-p) AR V, = V, cos(6-8) = B

WAE = VP/(AP) 3 .(5) ACCELERATION: P \ ap ap=(8P)WEP 40

Tapi) = (api)=(AP) XAE &B (ap) = (AP) WAS A B ax = 2WAF VP/ AB (7)

ap = ap1 + ap/ + ax



RIGHT TRIANCLE: an+ (ap) = ap cos(90°+B-6) ax + (AP) dAE = ap cos (90 + 18-6) dar = 1 apcos(90+8-0) - ac] (8)

(CON TINUED)

15.C2 continued

DATA: Wap = 6 rads BP = 15 in.; A8 = 30 in

CUTLINE OF PROBRAM:

1. USE EQS. () AND(2) TO FIND & AND AP.

2. USE EQS. (3) AND (4) TO FIND Up AND Up

3. DETERMINE WAR BY USING EQ.(5)

4. USE EQ.(6) TO FIND ap

5. USE EQ(7) TO FIND ax

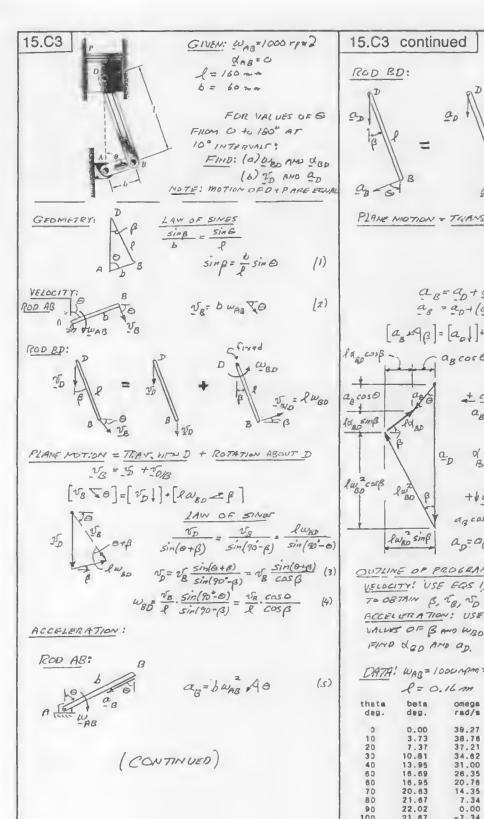
6. DETERMINE das BY USING EG.(8)

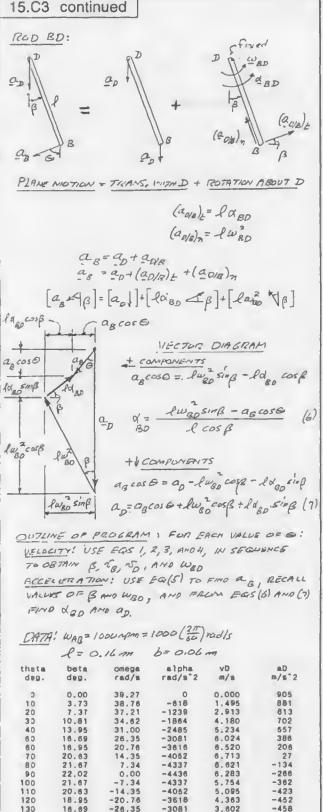
theta	bata	omegaAE	alpha
deg.	deg.	rad/e	rad/6°2
0	0.00	2.000	0.000
15	4.99	1.985 -	0.712
30	9,90	1.937	1.508
45	14.84	1.850	2.492
80	19.11	1.714	3.818
78	23,18	1.509	5.128
90	28.57	1,200	8.640
105	29.02	0.730	13.273
120	30.00	0.000	20.785
135	28.68	-1.144	32.388
150	23.79	-2.860	45.782
185	14.05	-4.920	43.298
180	0.00	-8.000	0.000

theta for maximum alpha

(2)

theta deg.	alpha 'rad/e^2
157.0800	48.58893 48.58894
167.1000	48.58894





-31.00

-34.62 -37.21

-39.27

-1864

-1239

10.61

0.00

2.643

2.103

1.365

0.667

0.000

-451

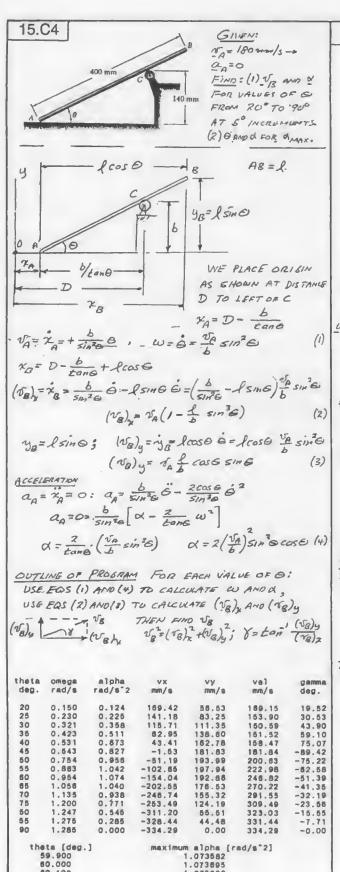
-437

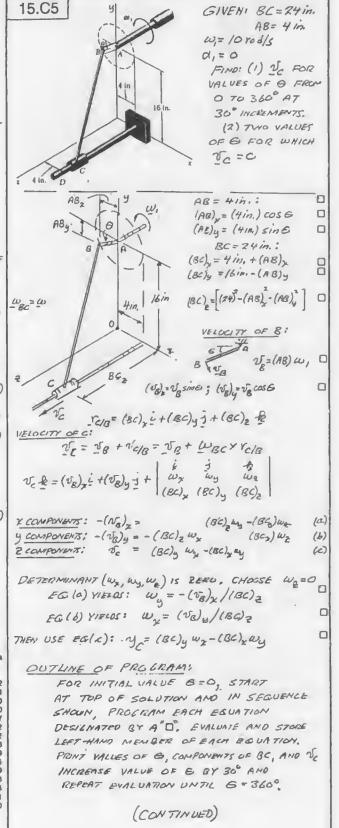
-415

150

160

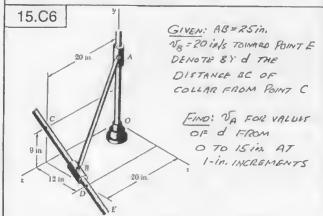
180

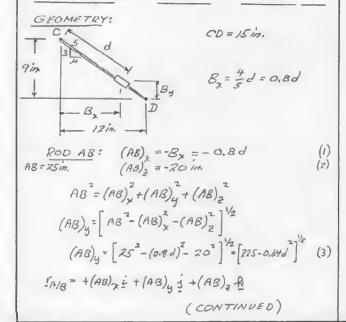




	Come	onents of	and ac	Veleelt
theta	X	Y	700 80	Velocity of C
dag	in.	in.	in.	in./s
		****	****	111.76
0.000	8.000	-16.000	18.000	40.000
30.000	7.464	-14.000	18.008	35.221
60.000	6.000	-12.536	19.507	23.436
90,000	4.000	-12.000	20.396	7.845
120.000	2,000	-12.536	20.368	-8.908
150.000	0,536	-14.000	19.486	-24.338
180.000	0.000	-16.000	17.889	-35.777
210.000	0.536	-18.000	15.865	-39.977
240.000	2.000	-19.464	13.898	-32.995
270.000	4.000	-20.000	12.649	-12.649
300.000	6.000	-19.464	12.694	14.293
330,000	7.464	-18.000	14.010	33.851
380.000	8.000	-16.000	16.000	40.000
Dete	rmination of	values of	theta for vi	C = 0
	Comp	onenis of	rod ac	Velocity
theta	×	V	z	01.0

	Comp	onenis of	rod BC	Velocity
theta	ж	У	Z	of C
104,034	3.030	-12,119	20.492	0.001
104.035	3.030	-12.119	20.492	0.001
104.036	3.030	-12.119	20.492	0.000
104.037	3.030	-12,119	20.492	-0.000
	Comp	onents of	rod BC	Velocity
theta	ж	У	2	of C
264.020	4.969	-19.881	12.492	-0.015
284.030	4.970	-19.881	12,492	-0.008
284.040	4.970	-19.881	12.492	0.003





15.C6 continued	VELOCITY OF B:	
W	NB = 20 10/5	
TC 35	2co = 42 - 3 j	
912.	VB = VB 2c0	
1210	25=(2010/s X = i-3 =)	
/ / / / / / / / / / / / / / / / / / / /	(Ng) = 16 in/s	(4)
	(Na)y = -12 in/s	(5)

 $\frac{\nabla_{A} = \nabla_{B} + \mathcal{D}_{A/B} = \mathcal{D}_{B} + \mathcal{Q} \times \mathcal{D}_{A/B}}{\mathcal{Q}_{A} \cdot \mathcal{Q}_{A} \cdot \mathcal{Q}_{A} \cdot \mathcal{Q}_{A/B}} = \frac{\nabla_{B} + \mathcal{Q} \times \mathcal{D}_{A/B}}{\mathcal{Q}_{A} \cdot \mathcal{Q}_{A/B}} + \frac{\mathcal{Q}_{A/B} \cdot \mathcal{Q}_{A/B}}{\mathcal{Q}_{A/B} \cdot \mathcal{Q}_{A/B}} + \frac{\mathcal{Q}_{A/B} \cdot \mathcal{Q}_{A/B}}{\mathcal{Q}_{A/B}} + \frac{\mathcal{Q}_{A/B} \cdot \mathcal{Q}_{A$

VELOCITY OF A:

DETERMINATE OF $(w_{x_3}, w_{y_3}, w_{y_4})$ is 2600. CHOSE $\omega_{x_2} = 0$ EQ. (a): $-(v_{y_3})_x = 0 + (AB)_y \omega_{y_4}$ $\omega_{y_2} = -(v_{y_3})_x/(AB)_y$ (b)

$$EG.(b): \nabla_A - (\nabla_B)_y = O + (AB)_x \omega_B$$

 $V_{A} = (V_{B})_{y} + (AB)_{\chi} \omega_{Q} \tag{7}$

OUTLINE OF PROGRAM:

FOR INITIAL VALUE & = 0, PROBRAM, IN

SEQUENCES, EQUATIONS (1) THROUGH (7)

EVALUATE LEFT-HAND MEMBER OF EACH

EQUATION AND PRINT VALUES OF

d, COMPONENTS OF YAIB, AND Ve.

INCREASE VALUE OF & BY I'M. AND

REPEAT PROCESS UNTIL &= CD = 15in

	Com	ponents of	A8	Velocity
d	×	У	2	VA
in.	in.	in.	in.	in/s
0.000	0.000	15.000	-20.00	-12.000
1.000	-0.800	14,979	-20.00	-12.855
2.000	-1.600	14.914	-20.00	-13.716
3.000	-2.400	14.807	-20.00	-14,593
4.000	-3.200	14.655	-20.00	-15.494
5.000	-4.000	14.457	-20.00	-16,427
6.000	-4.800	14.211	-20.00	-17.404
7.000	-5.600	13.915	-20.00	-18.439
8.000	-6.400	13.566	-20.00	-19.548
9.000	-7.200	13.159	-20.00	-20.754
10.000	-8.000	12.689	-20.00	-22.088
11.000	-8.600	12.147	-20.00	-23.591
12.000	-9.600	11.526	-20.00	-25.327
13.000	-10.400	10.809	-20.00	-27.394
14.000	-11.200	9.978	-20.00	-29.960
15.000	-12.000	9.000	-20.00	-33.333



GIVEN: W=316 .



PROBLEM 16.1:

FOR P=516,

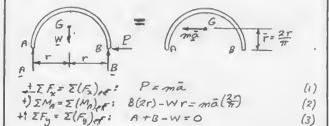
FIND: (a) Q.

(b) REACTION.

PROBLEM 16.2:

FOR A=0,

FIND: (a) P/(b) a.



PROBLEM 16.1: P=51b, W=31b, m=4/9
EQ(1)! P=(w/g) &
a=(P/w)g=\frac{51b}{31b}(32.274/s^2)
\[\bar{a} = 53.1774/s^2; \bar{a} = 53.774/s^2 \]

$$EQ(2): B(2r) - Wr = \frac{W}{3} (\frac{\rho}{W}g)(\frac{2r}{\pi})$$

$$B = \frac{1}{2}W + \frac{\rho}{\pi} = \frac{1}{2}(316) + \frac{516}{\pi}$$

$$B = 3.09216$$

$$B = 3.0916$$

EQ(3): A + 3.0924-316=0 A=-0.09246 A=0.09216

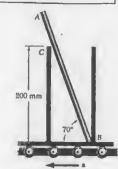
PROBLEM 16.2: A=0, W=316, m= 4/9

$$E_{G}(2): \quad 0 - W r = \frac{W}{g} \bar{a} \left(\frac{2r}{\pi}\right)$$

$$\bar{a} = \frac{\pi}{2} g = \frac{\pi}{2} (32.25 \%^{2})$$

$$\bar{a} = 50.58 \text{ A/s}^{2} \qquad \bar{a} = 50.64 \%^{2} - \frac{1}{2} (32.25 \%^{2})$$

EQ(1): $P = \frac{W}{9}\bar{a}$ $P = \frac{W}{9}(\frac{\pi}{7}9) = \frac{\pi}{2}w = 4.71216$ P = 4.7116 16.3 and 16.4



GIVEN:

Roo: m = 2.5 kg, AB = 300 ma

PROBLEM 16.3: FOR a=1.5m/s = FIND: (a) C, (b) B

TO REMAIN IN POSITION

PROBLEM 16.4:
Fire: amax FOR ROD

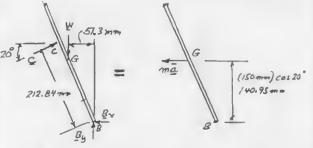
GEOMETRY

ROOMM = 212.84 mm

ROOMM = 212.84 mm

(0.15 m) SIN 70 = 51.3 mm

W=mg=(2.5 Rg) 9.81 m/s - 24.525 N



+) IMB = I(MB)AF; C(212.8+mm) = IM(51.3 mm) = -ma(140.25 mm)

C = 0.241 W - 0.6622 ma

C = 0.241(24.525 N) - 0.6622(2.5Aa)(a)

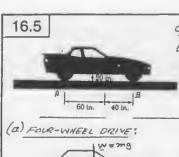
C = 5.911 N - 1.656 a (1)

 $B_{\chi} = (3.43 \text{ N})\cos 20^{\circ} = (2.5 \text{ Re})(1.5 \text{ m/s})$ $B_{\chi} = 3.22 + 3.75 = 6.97 \text{ N}$ $B = 24.4 \text{ N} \quad 573.4^{\circ}$ $6.97 \text{ N} \quad B = 24.4 \text{ N} \quad 573.4^{\circ}$

PROBLEM 16.4: For a may, C=0

EG(1) $C = 5.911 \, N - 1.656 \, \alpha$ $0 = 5.911 \, N - 1.656 \, \alpha \, man$ $a_{man} = 3.57 \, m/s^2$

amoj 3,57 m/s2



[=IND: amax ASSUMINE (a) FOUR-WHEE DRIVE

(6) REAR-WHERE DIRIVE

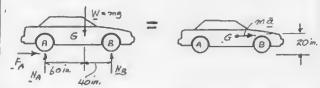
(4) FRONT-WHER DRIVE

+ + ZFy=0: NA +NB-W=0

THUS: FATE = 1/NA+1/N3 = 1/NA+NB) = 1/N = 0.00 mg + IEz = I(F2)eff: FATE3 = mā 0.80 mg = mā

ā=0,00 g=0.80(32.2 fgs) ā= 25.8 ft/s'-

(b) REAR-WHEEL DAINE:



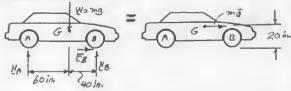
+) $\Sigma M_B = \Sigma (M_B)_{eff}$: (40 in.) $W - (100 \text{ in.}) N_B = - (20 \text{ in.}) \text{ ma}$ $N_B = 0.4W + 0.2 \text{ ma}$

THUS: FA = MNg = 0.80(0.4W+0.2 ma) = 0.32 mg+0.16 ma

 $\pm \sum F_{\chi} = \sum (F_{\chi})_{eff}$: $F_{A} = m\bar{a}$ 0.32 $mg + 0.16m\bar{a} = m\bar{a}$ 0.32 $g = 0.84\bar{a}$ $\bar{a} = \frac{0.32}{127.764c^{3}}$

a= 0.32 (37.2 ft/s) a= 12.27 ft/s->

(c) FRONT-WHEEL DRIVE!



+) IM= Z(Ma)er (100 in.) NB-(60 in.) W=-(20 in.) mā
NB= 0.6W-0.2 mā

THUS: FB = 4/18 = 0.80(06W-0.2ma) = 0.48 mg-0.16 ma

$$\begin{array}{lll}
+ \sum F_{\chi} = \sum (F_{\chi})_{e,f} : & F_{B} = m\bar{\alpha} \\
0.48 \, mg - 0.16 \, m\bar{\alpha} = m\bar{\alpha} \\
0.48 \, g = 1.16 \, \bar{\alpha} \\
\bar{\alpha} = \frac{0.48}{1.16} \left(32.2 \, f \, \xi^{2}\right)
\end{array}$$

a=13.32ft/c2 -

16.6



GIVEN: D=30ft/sFROM SAMPLE PROB 16.1

4/k=0.699

FIND: DISTANCE REGINED
TO STOP IF

(a) REAR-WHEEL BRAKES
FAIL TO OPERATE

(b) FRONT WHEAL BRAKE FAIL TO OPERATE

(a) IF REAR-WHEEL BRAKES FAIL TO OPERATE

+) $\Sigma M_A = \Sigma (M_A)_{AB}$: $N_B (12ft) - W(5ft) = m\tilde{a}(4ft)$ $N_B = \frac{5}{12}W + \frac{1}{3}\frac{W}{9}\tilde{a}$

$$\frac{1}{4} \sum_{n} F_{n} = \sum_{n} F_{n} F_{n} = \sum_{n} F_{n} F_{n} = \sum_{n} F_{n} F$$

$$\tilde{a} = \frac{0.679(\frac{5}{17})(32.2845)}{1-0.233}$$
 $\tilde{a} = 12.227 \text{ fys}^2 +$

 $v^{2} = v_{0}^{2} + 2ax$ 0 = (30 ft/s) - 2(12.227 ft/s)xx = 36.8 ft

(b) IF FRONT-WHEEL BRAKES FAIL TO OPERATE



+) $IM_B = I(M_B)_{eff}$: $W(7ft) - N_A(12ft) = m\bar{a}(4ft)$ $N_A = \frac{7}{12}W - \frac{1}{3}\frac{W}{9}\bar{a}$

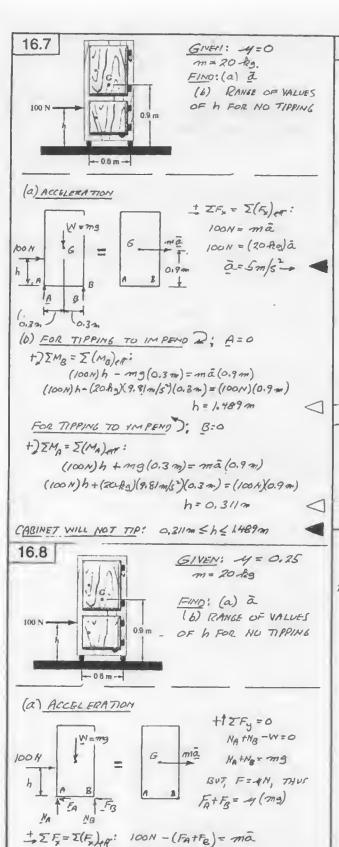
$$+ \sum F_{\chi} = \sum (F_{\chi})_{ef}; F_{A} = m\bar{\alpha}$$

$$- \int_{K} N_{A} = \frac{W}{9}\bar{\alpha}$$

$$0.699 \left(\frac{7}{12}W - \frac{1}{3}\frac{W}{9}\bar{\alpha}\right) = \frac{W}{9}\bar{\alpha}$$

$$\bar{a} = \frac{0.699 \left(\frac{7}{12}\right) \left(32.26t/c^2\right)}{1 + 0.233} \qquad \bar{a} = 10.648 \text{ ft/s}^2 =$$

UNIFORMLY ACCELETATED MOTION $T^2 = V_0^2 + 2ax$ $O = (30 \text{ ft/s})^2 - 2(10.648 \text{ ft/s}^2) \times$ $\chi = 42.3 \text{ ft}$



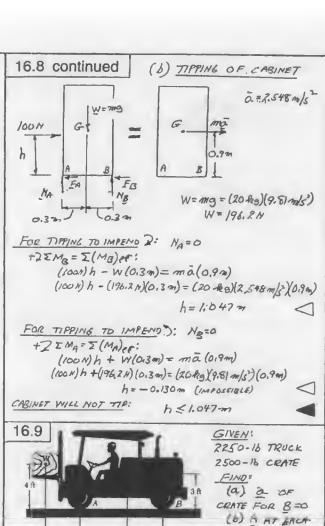
100 H - 4mg = ma

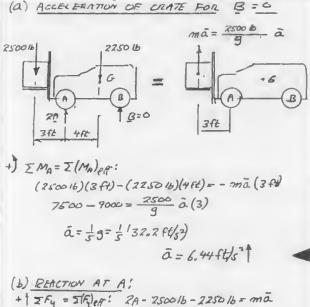
(CONTINUED)

a= 255 m/s'-

100 N-0.25 (20 Ag) (9.8/m/s2) = (20 leg) a

a= 2.548 m/5





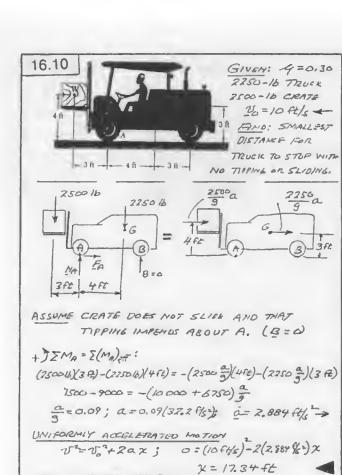
2A - 4750 16 = 2500 16 (9)

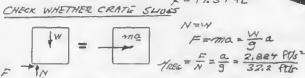
2A= 525015

FOR ONE WHEEL:

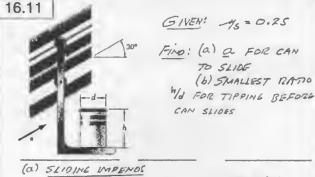
A = 2625 16 1

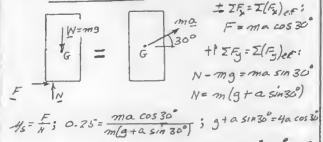
PRONT WHEEL

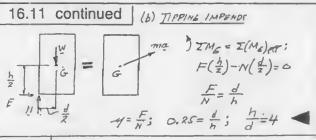


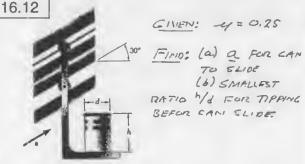


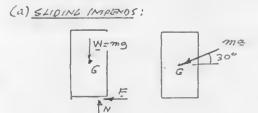
MREG = 0.09 < 0.30, CRATE DOES NOT SLIDE







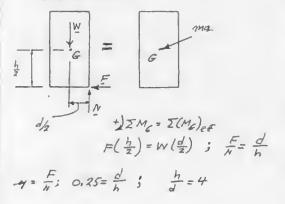


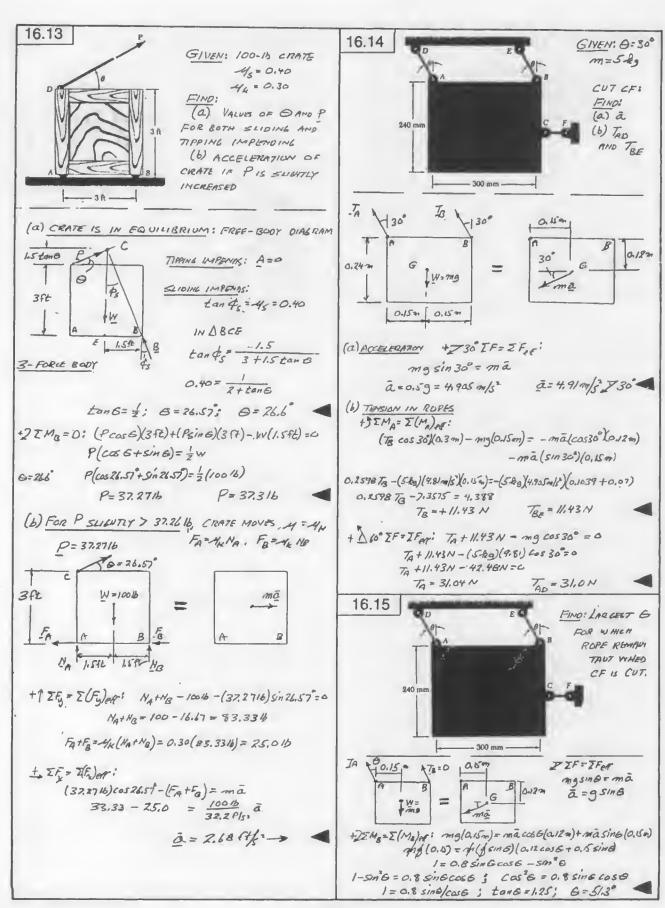


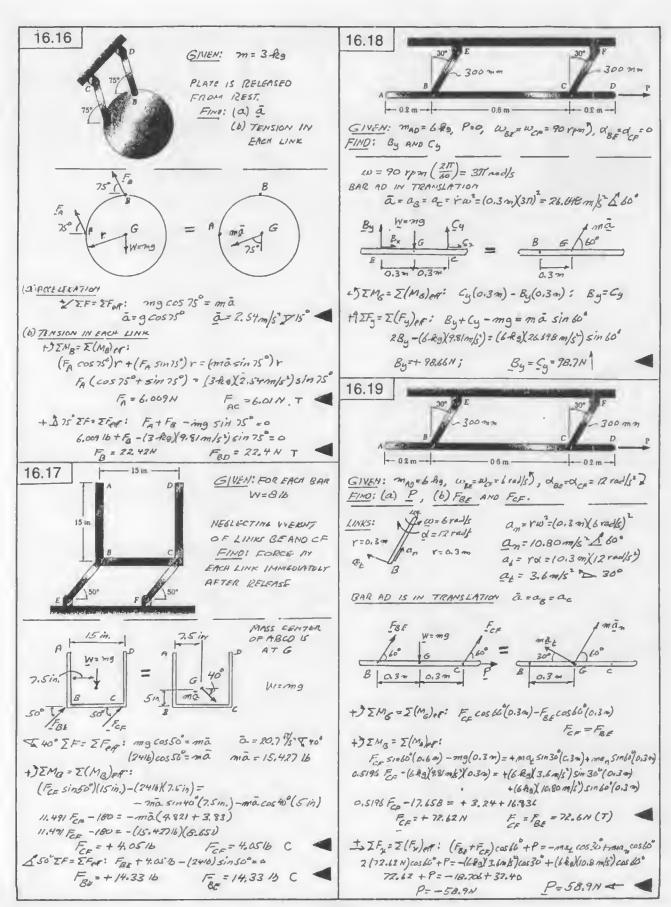
= IFx = I(Fx) = F = ma cos 30°

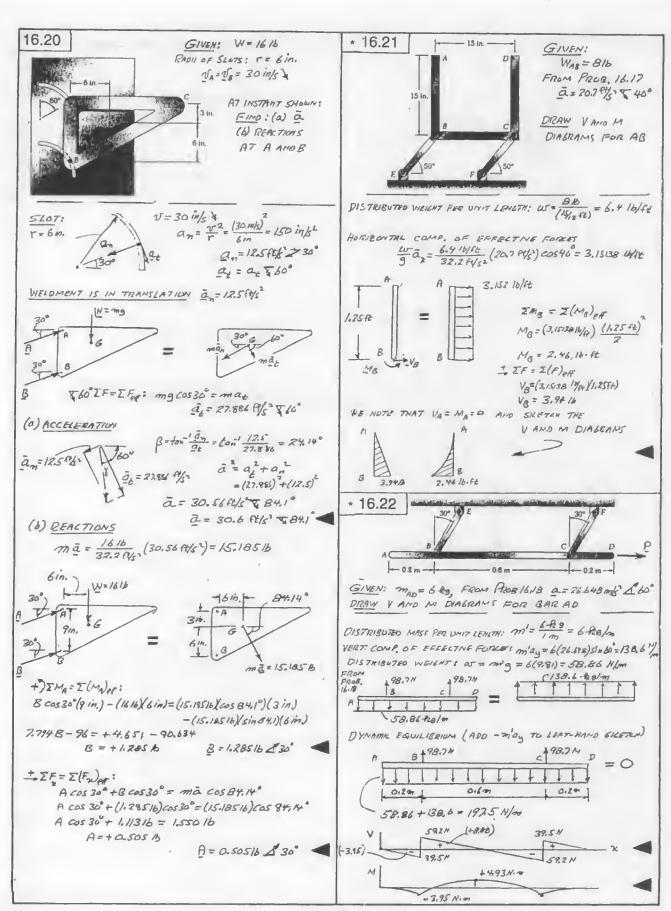
$$+1 \sum_{j=1}^{n} \sum_{i=1}^{n} \sum_{j=1}^{n} \sum$$

(b) TIPPING IMPENDS:











FOR TRANSLATION

SHOW THAT EFFECTIVE

FORCES ARE (DW) A

ATTACHED TO FARTICLES

AND ALE REDUCE TO

MA ATTACHED AT G

SINCE SLAB IS IN TRANSLATION, EACH PARTICLE HAS SAME ACCEPTATION AS G. NAMELY Q.
THE EFFECTIVE FUNCES CONSIST OF (DM) Q.



THE SUM OF THESE VECTORS IS: $\Sigma(\Delta m_i)\hat{a} = (\Sigma \Delta m_i)\hat{a}$ OR SINCE $\Sigma \Delta m_i = m$, $\Sigma(\Delta m_i)\hat{a} = m\hat{a}$ THE SUM OF THE MUMENTS ABOUT & IS: $\Sigma r_i \times (\Delta m_i)\hat{a} = (\Sigma \Delta m_i r_i) \times \hat{a}$

BUT, I M. P. = M. P. = O, BECAUSE & IS THE MASS
CENTER. IT FOLLOWS THAT THE RICHT-HAND
MEMBER OF EQ.(1) IS ZEED. THUS, THE MOMENT
ABOUT 6 OF M. MUST ALSO BE ZERD, WHICH MEANS
THAT ITS LINE OF ACTION PASSES THROUGH & AND
THAT IT MAY BE ATTACHED AT 6.

16.24



FOR CENTROIDAL ROTATION,

SHOW THAT EFFECTIVE

FULCES CONIST OF VECTOUS

-(DM; WY; AND (DM;)(XXY!)

ATTACHED TO PARTICLES AND

REDULE TO A COUPLE IN.

FOR CONTROVAL ROTATIONS a, = (9;), + (a;) = 1 + x; - w2x;

EFFECTIVE FORCES ARE: (Am;) a; = (Am)(xxx;)-(Am;) w2x;

(Am)(xxi)

 $\frac{1}{2} \frac{1}{2} \frac{1}$

 $\sum (\Delta m_i) a_i = \sum (\Delta m_i) (d \times r_i^*) - \sum (\Delta m_i) \omega^2 r_i^*$ $= \alpha \times \sum (\Delta m_i) r_i^* - \omega^2 \sum (\Delta m_i) r_i^*$

SINCE G IS THE MASS CHATER, \$\(\Delta\mi\)\overline{\chi} = 6

i. EFFECTIVE FORCES REDUCE TO A COUPLE,

SUMMINE MOMETATS ABOUT G

$$\begin{split} & Z\left(Y_{i}^{\prime} \times \Delta m_{i} \, \alpha_{i}\right) = \sum \left[Y_{i}^{\prime} \times \left(\Delta m_{i}\right) \left(\alpha_{i} \times Y_{i}^{\prime}\right)\right] - \sum_{i} \chi(\Delta m_{i}) \omega_{i} Y_{i}^{\prime} \\ & \mathcal{B} u \gamma_{i} \, T_{i}^{\prime} \times \left(\Delta m_{i}\right) \omega_{i}^{2} T_{i}^{\prime} = \omega_{i}^{2} \left(\Delta m_{i}\right) \left(Y_{i}^{\prime} \times Y_{i}^{\prime}\right) = G \\ & \mathcal{A} \times \mathcal{A}_{i}, \, Y_{i}^{\prime} \times \left(\Delta m_{i}\right) \left(\alpha_{i} \times Y_{i}^{\prime}\right) = \left(\Delta m_{i}\right) Y_{i}^{\prime} \times \mathcal{A}_{i}^{\prime} \\ & \mathcal{A} \times \mathcal{A}_{i} \times \mathcal{A}_{i}^{\prime} \times \mathcal{A}_{$$

SINCE I (Am) 1/2 = I,
THE MOMENT OF THE COUPLE IS IN

16.25 FLYWHEEL: W=600016 &= 36 in.

AT t=0, W=300 PPM, AT t=10 min, W=0

FING COURLE DUE TO KINETIC FRICTION, (UNIF. ACCEL. MOTION)

 $\vec{I} = m \vec{R}^{2} = \left(\frac{6000 \text{ lb}}{32.2 \text{ R/s}^{2}}\right) (3 \text{ ft}) = 1677.0 \text{ lb. ft. s}^{2}$ $\omega_{c} = 300 \text{ rpn} \left(\frac{2\pi}{60}\right) = 1077 \text{ rad/s}$ $\omega = \omega_{0} + \text{dt}; \quad 0 = 1077 \text{ rad/s} + \alpha(600 \text{ s})$ $\alpha = -0.05236 \text{ rad/s}^{2}$

M=Ix=(1677 16.ft.s) (0.5236 **/(*)=87.81 16.ft
M=67.8 16.ft

16.26 ROTOR: m=50 Rg, R= 180 mm

FRICTION COUPLE: M = 3.5 N·m

6=0, W = 3600 rpm (UNIF. ACCEL. MOTION)

FIND: REVOLUTIONS AS ROTOR CORSTS TO REST

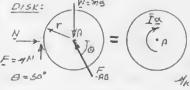
$$\begin{split} \tilde{I} &= m\tilde{R}^2 = (50 \, \text{Ag})(0.180 \, \text{m})^2 = 1.620 \, \text{Ag} \cdot \text{m}^2 \\ M &= Id: \quad 3.5 \, \text{N·m} = (1.620 \, \text{Ag} \cdot \text{m}^2) \, \text{d} \\ d &= 2.1605 \, \text{rad/s}^2 \, \left(\text{DECULETIATION} \right) \\ W_0 &= 3600 \, \text{rpm} \left(\frac{2\pi}{60} \right) = 120 \, \text{Trad/s} \\ W^2 &= V_0^2 + 2 \, \text{d} \, \text{G}: \quad 0 = (120 \, \text{Trad/s})^2 + 2 \left(-2.1605 \, \text{rad/s}^2 \right) \, \text{G} \\ \Theta &= 32.811 \, \text{No} \, \text{rad} \left(\frac{1 \, \text{rev}}{2\pi \, \text{rad}} \right); \quad G = 5234.8 \, \text{rev} \\ \Theta &= 5730 \, \text{rev} \end{split}$$

16.27 G

GIVEN: 4/4 = 0.40

FIND: OF FOR DIRECTION OF MOTION OF BELT SHOWN

BEIT: N F= MAN



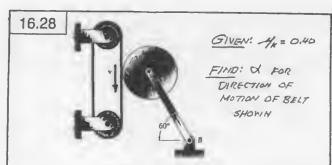
 $\begin{array}{l}
+ \sum F_{x} = \sum (F_{y})_{eff}: \\
N - F_{AB} \cos \Theta = 0 \\
F_{AB} \cos \Theta = N \quad (1) \\
+ + \sum F_{y} = \sum (F_{y})_{eff}:
\end{array}$

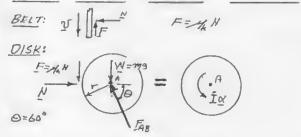
MKH + FAS SING - mg = 0 FAS SING = mg - MKH (2)

 $\frac{EG(2)}{FQ,(1)}: \tan \theta = \frac{mg - M_R N}{N}$ $N = \frac{mg}{\tan \theta + 4\pi} : F = \frac{mg}{\tan \theta + 4\pi} : F = M_R N = \frac{mg}{\tan \theta + 4\pi}$ $f) = \sum_{k=1}^{\infty} \frac{F(k)}{\pi} = \sum_{k=1}^{\infty} \frac{mg}{\pi} : F = \prod_{k=1}^{\infty} \frac{mg}{\pi} + \prod_{k=1$

d= IF = + mg Ac = 29 · ton6 + 4k

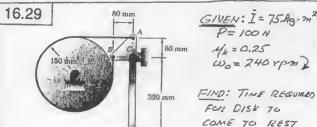
 $\frac{DA7A: \ r = 0.18 \ m_1, \ \Theta = 66^{\circ}, \ 4/\kappa = 0.46}{0.18 \ m} \cdot \frac{0.40}{0.18 \ m} \quad \propto = 20.4 \frac{rad}{5^{\circ}}$

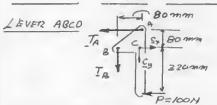




$$+\sum F_{\chi} = \Sigma (F_{\chi})_{AF}: N - F_{AB} \cos \Theta : F_{AB} \cos \Theta = N$$
 (1)
+\ \Sigma F_{\text{g}} = \Sigma (F_{\text{b}})_{AF}: F_{AB} \sin \Omega - \mathre{M} = C

$$\frac{EO.(2)}{EO.(1)}$$
: $tanG = \frac{mg + q_E N}{N}$





STATE EQUILIDATION:

(CONTINUED)

16.29 continued

ON = 240 rpm (21) = STT rads)

(3)

+)
$$IM_g = \Sigma(M_E)_{eff}$$
: $T_E r - T_A r = \bar{I} \propto$

$$T_E - T_A = \frac{\bar{I}}{r} \propto \qquad (z)$$

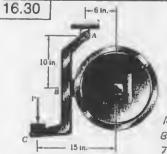
8FLT FRICTION:

$$\beta = 270^{\circ} = \frac{3}{4}\pi$$
 and $\frac{T_{B}}{T_{A}} = e^{4\kappa}\beta = e^{(0.26)\frac{3}{4}\pi} = e^{1.178} = 3.748$

TB = 3.248 TA

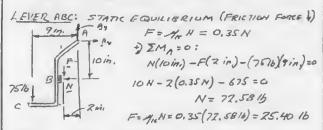
EQ.(2):
$$T_A + T_B = 400N$$
; $T_A + 3.248 T_A = 400 N$
 $T_A = 94.16N$ $T_B = 3.248 (94.16N) = 305.9N$
EQ.(2): $T_B - T_A = \frac{1}{r} d$; $305.9N - 94.16N = \frac{75.8476^2}{0.15m} d$
 $0.423 re J/s^2$

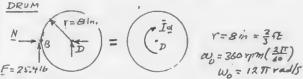
NOTE: IF & IS REVERSED THEN TA AND TO ALE
INTER CHANGED. THIS CAUSEN HO CHANGE IN EG.(1)
AND EG(2). THUS FROM EG(3), & IS NOT CHANGED.

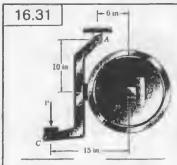


51 NEM: I = 14 b.ft.s² T/k = 0.35 P = 76 16 W₀ = 360 rpm)

FINO: NUMBER OF
REVOLUMENTS OF DRUM
REFORE 17 COMES
TO REST

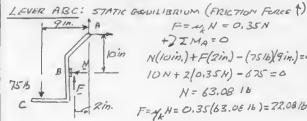




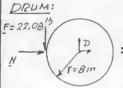


GIVEN: 1 = 14 16. ft.52 4 = 0.35 P=7516 Wo = 360 rpm)

FIND: NUMBER OF REVOLUTIONS OF DRUM BEFORE IT COMES TO REST.



F= N, N = 0.35N +JIMA=0 N(10in.)+F(2in.) - (7516)(9in.)=0 10N+2/0,35N)-625=0 N = 63.08 18 F=4+H=0.35(63.08 16)=72.08/b



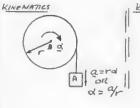
r=811.= = = A $U_0 = 360 \text{ rpm} \left(\frac{2\pi}{L_0} \right)$ w= 12 Tracks

+) IM = EMO) == 0 Fr= IX (22.08 lb)(3/4)= (14.16.ft.5) X X = 1.5015 rad/s (DECELERATION) 0=(1271 rad/s) +2(-15015 tod/s2)& w= w + 2de: 6 = 675.8 rad B = 675.8 nad (27) = 10256 rev; B = 107.6 rev

16.32 r=500 × m=15-kg)

GIVENI FLYWHEEL Mr = 120 Rg A= 375 mm V=0 AT 5=0 FIND: (a) a of BLOCK. (b) I AFTER IT HAS

MOVED 1.5 m.



KINETICS

+ I IMB = I (MB) et: (mg)r = I a+(m,a)r mgr = m, k2(2)+mar ma + mp (+)2

a= (15-89)(9.8/m/s2) 15-Ag+(120Ag)(375mm)2=1,7836 mb (CONTINUED) 16.32 continued

(a) a=1.7836 m/s26

a = a = 1.7836 m/s = 3.567 rad/s2 a = 3.57 md/s)

(b) V= V+ zas FOR S=1.5m: V==0+2(1.1836m/s) (1.5m) VA = 2.313 m/s VA= 2.3/m/s +

16.33 600 711

GIVEN: SYSTEM PELFASED FROM REST: 1. IF ma = 12 fig, BLOCK FALLS 3 MM IN 4.65 2. IF ma= 24-kg, BLOCK FALLS 3m IN 3.15 ASSUME CONSTANT ME DUE TO AXLE FRICTION.

FIND: I

KINEMATICS OR cl= a

KINETICS W=MA3 +) EMR = I(MB) ex: (mag)r - Mr = Ix + (ma)r magr-Mg= I = + mar

CASE 1: 4=3m, 2=4.65 7= 2a 12; 3m= 2a (465); a=0.2836 m/s2 mp= 12-29

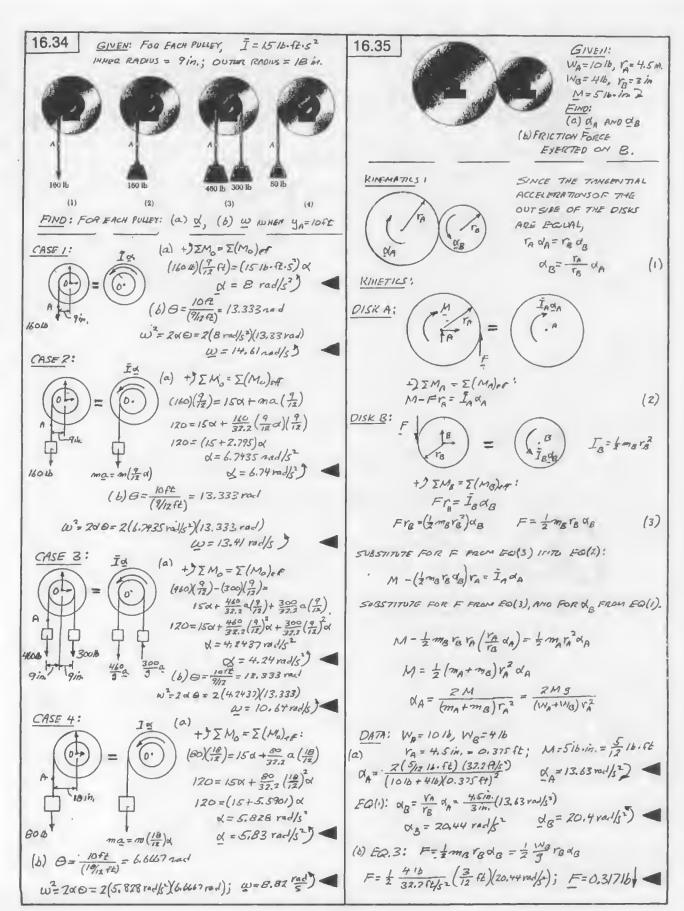
SUBSTITUTE INTO FOO(1) (12.8g)(9.81m/5)(0.6m) - M5 = I/O.2836 m/6) + (1260)0.2836 m/6) 70.632 - NI = I (0.4727) + 2.04/9 (2)

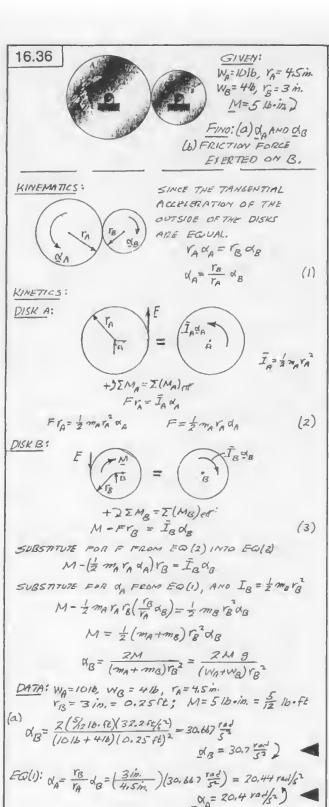
CASE 2: 4=3m, t=3.15 y= = 2at2; 3m = = 2a(3.15)2; a=0.6243 m/s2 ma= 24-Rg

SUBSTITUTE INTO EQ(1): [24.80)(9.81 m/s) (0.6 m) - Mg - I (0.6243 m/s) + (24 kg)(0.6243 m) (0.62) 141.264 -Mg= I (1.0406) +8.9899 (3)

· SUBTRACT EQ(1) FROM EQ(2), TO ELININATE M.F. 70,632 = I (1.0406 - 0.4727) + 6.948 63.684 = I (0.5679) I= 112.14 kg.m

I = 112.1 kg·m

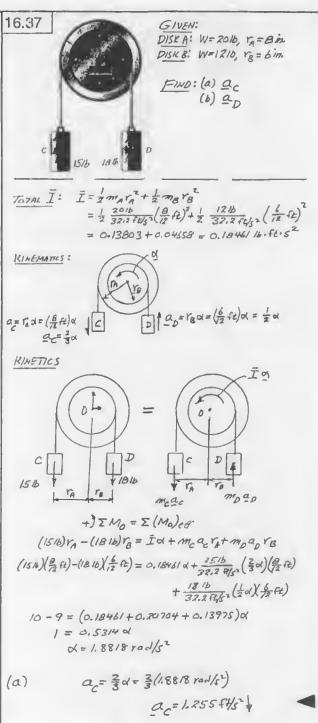


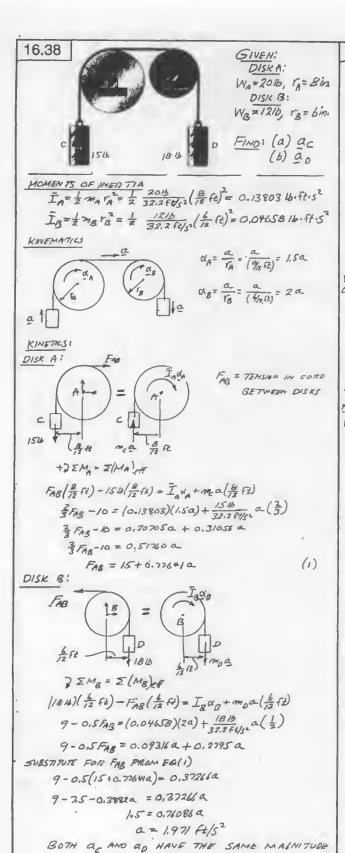


(b) EQ(2) F= 1 marada = 1 Wa rada

F = 1/2 (1016) (4.5) (4.5) (20.44 val/s2) = 1.190/6

FRICTION FORCE ON DISIL B: F=1.19016





ac= 1.97/ 4/52

ap= 1.971 ft/s=

16.39 and 16.40



GIVEN: $m_A = 6.4g; m_B = 2.4g$ N = 20.15 PROBLEM 16.39: $(w_A)_B = 360 pm; ; (w_B)_B = 0$ PROBLEM 16.40: $(w_A)_B = 0; (w_B)_B = 360 pm;]$

FIND: (a) BA AMA &8. (b) FINAL VELOCITIES WA AMA WA

WHILE SLIPPING OCCURS, A PRICTION FURCE FT IS

APPLED TO DISIL A, AND F TO DISK 8.

DISK A: $I_{A} = \frac{1}{2} m_{A} r_{A}^{2}$ $I_{A} = \frac{1}{2} (6 + 9)(0.02m)^{2}$ $I_{A} = \frac{1}{2} m_{A} r_{A}^{2}$ $I_{A} = \frac{1}{2} m$

+) [Ma= [(Ma) = : Fra = Inda (3N/0.08m) = (0.0192 kg·m²) da da = 12.5 rad/s² (Xp = 12.5 rad/s²)

DISK B: F=3M N T_B= OOL m

 $\vec{I}_{8} = \frac{1}{2} m_{8} r_{8}^{2}$ $= \frac{1}{2} (3 k_{9}) (0.06 m)^{2}$ $= 0.005 + k_{9} \cdot m^{2}$

 $(3N)(0.06m) = (0.0054 - 6g \cdot m^{2}) \times g$ $(3R)(0.06m) = (0.0054 - 6g \cdot m^{2}) \times g$ $(3R)(0.06m) = (3.0054 - 6g \cdot m^{2}) \times g$ $(3R)(0.06m) = (3.0054 - 6g \cdot m^{2}) \times g$

PROGLEM 16.39: (WA) = 360 PPO (20) = 17 Tradk); (WB) = 0

DISKS WILL STOP SLIDING, WHEN SC = SOI, THAT IS

WHEN

\[(WA) - AAL] YA = WBYB

[(WA) - AAL] YA = QE YB

 $[(\omega_A)_0 - \alpha_A t] Y_A = \alpha_B t Y_B$ $(12\pi - 12.5 t)(0.08) = (33.33 t)(0.06)$ 3.0159 - t = 2t ; t = 1.0053/5

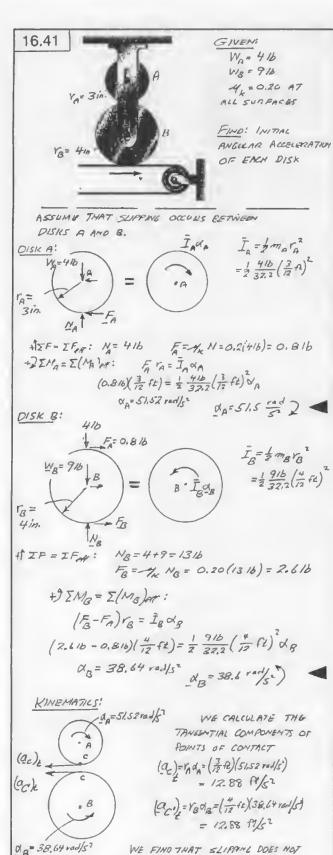
+) $\omega_{A} = (\omega_{A})_{0} - \sigma_{A}t = 12\pi - 12.5(1.00531) = 25.132 \text{ rad/s}$ $\omega_{A} = 25.132 \text{ rad/s} \left(\frac{60}{20}\right) = 240 \text{ rpm} \qquad \omega_{A} = 240 \text{ rpm}$

+) WB = 48t = (33,33)(1.00531) = 33.507 rad/s
WB = 33.507 rad/s (60) = 3201pm WB = 3201pm)

PROBLEM 16.40: $(w_A)_0 = 0$; $(w_B)_0 = 360 \text{ pm} / \frac{27}{6}) = 1277^{-44} \frac{1}{5}$ SLIDNG STOPS WHEN $V_L = V_{L^2}$ THAT IS WHEN $w_A v_A = w_B v_B$ $(\alpha_A t) v_A = [(w_A)_0 - \alpha_B t] v_B$ (12.5t)(0.08) = (1277 - 33.33t)(0.00) t = 2.26195 - 2t; t = 0.253985

+) $W_A = W_A l = (12.5)(0.75398) = 9.4248 \text{ rad/s}$ $W_A = 9.4248 \text{ rad/s}(\frac{60}{2\pi}) = 90 \text{ rpm}$ $W_A = 90 \text{ rpm}$

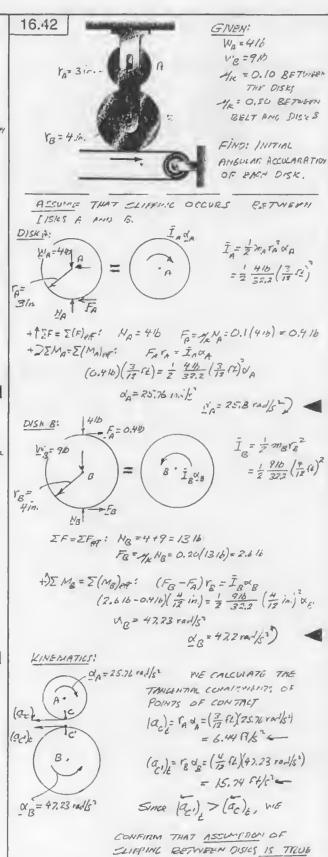
+) $w_8 = (4_0)_0 - \alpha_B t = 12\pi - (33.23)(0.75398) = 12.569 red/s$ $w_B = 12.569 \text{ red/s} (\frac{60}{2\pi}) = 120 \text{ rpm}$ $w_B = 120 \text{ rpm}$

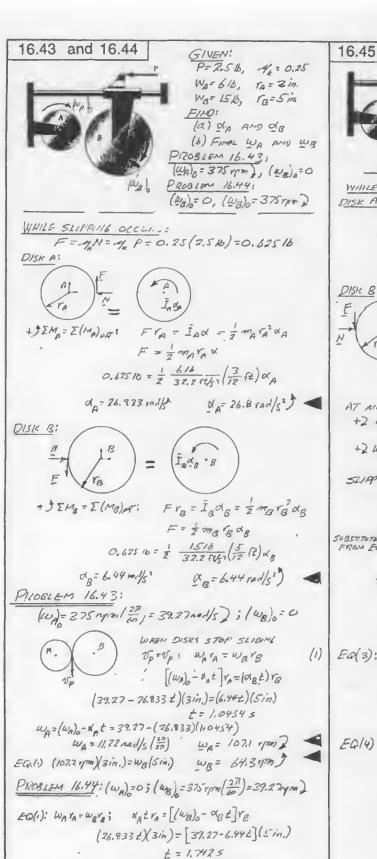


OCCUR BETWEEN DISKS, BUT SINCE

(Qc) = (Qc) & SLIPANG IN PENOS AND THAT FA = 4, NA = 0.816

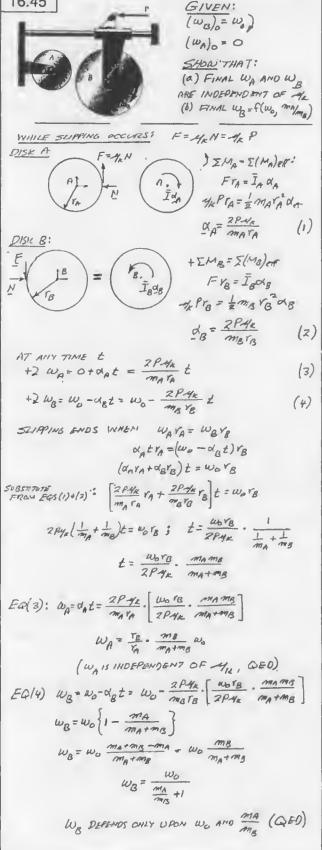
PNO ABONE RESULTS ARE VALID.





WA= dAt= (21.833)(1.742)= 46.74 vo.1/5 (2)

EO(1): (446 npm)(3111) = We(5111)



10 = 446 ypm)

WB = 268 YPM 2



SHOW THAT SYSTEM OF EFFECTIVE FORCES FOR A SLAB REDUCES TO Mã. AND EXPRESS DISTANCE FROM ITS LINE OF ACTION TO GIN TERMS OF AZO, AND X.

WE KNOW THAT THE SYSTEM OF EFFECTIVE FORSELS CAN BE KEYUCED TO THE YECTOR MIG. AT G AND THE COUPLE IN. WE FURTHER KNOW FILM CHASTER 3 OF STATICS TIMT A FORCE-COUPLE SYSTEM IN A PLANE CAN BE FURTHER LEDIKED TO A SINGLE FORCE



THE PERPENDICULAR DISTANCE & FROM G TO THE LIME OF ACTION OF THE SINGE VECTOR ME IS EXPRESSED BY WRITING

$$+) \Sigma M_{G} = \Sigma (M_{G}) e F^{2}$$

$$= \frac{1}{2} M_{G} + \frac{1}{2} M_{G}^{2}$$

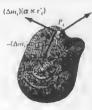
$$\vec{I} \propto = (m\vec{a}) d$$

$$d = \frac{\vec{k} \cdot \vec{k}}{\vec{k}}$$

 $d = \frac{Id}{m\bar{a}} = \frac{\pi k\bar{a}}{m\bar{a}}$



16.47



(Am,) SHOW THAT THE SYSTEM OF EFFECTIVE FORCES OF A PUBLIC SLAB CONSISTS OF THE VECTORS SHOWN ATTACHED TO THE PARTICLES P' OF THE STAB. FULTHER

SHOW THAT THE EFFECTIVE FORCES REDUCE TO MÀ ATTACHED AT & AND A COUPLE IX.

KINEMATICS



THE ACCELERATION OF PI is a: = a = aple a = a + dxr + wx(wxr;) = a + xxi - wzi. NOTE THAT XXY IS I TO Y

THUS THE EFFECTIVE FOILES ARE AS SHOWN IN FIE P 16.47 (also shown above). WE WRITE

(Ami) a:= (Ami) a+(Ami) (XXYi) - (Ami) w ri

THE SUM OF THE EFFECTIVE FORCES IS $\Sigma(\Delta m_i)a_i = \Sigma(\Delta m_i)\bar{a} + \Sigma(\Delta m_i)(\alpha \times \gamma_i) - \Sigma(\Delta m_i) \omega^2 \gamma_i^!$ I(Am) a = a I(Am) + x X [(Am)] - w I (Ami) re

(CONTINUED)

16.47 continued

INE NOTE THAT

[(AM) = M. AND SINCE G IS THE MASS CENTER I (4+;) " = mr, =0 THUS, E (Ani) a: = mā

THE SUM OF THE MOMENTS ABOUT & OF THE EFFECTIVE FORCES IS!

I(r:x Ami a) = Irix Ami a + Irix (Ami)(xx si) - Er X (Am) W'TE

I ([x A m; a;)=(E r Am;) a + [r x (xxy) Am;] - w I (r x y) Am;

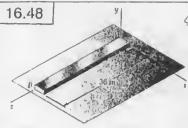
SINCE GISTHE MASS CENTER, IT! AM: = 0 ALSO, FOR EACH PAILTICLE, V'XY' = 0

I (Yex Am; a;) = I [Yi x (a; x ri) Am;

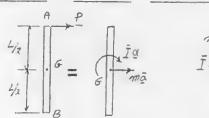
SINCE & I TI, WE HAVE I'X (dxr,) = rad AND $\sum (r_i^2 \times \Delta m_i^2 = i) = \sum r_i^2 (\Delta m_i) \underline{\alpha} = (\sum r_i^2 \Delta m_i) \underline{\alpha}$

SINCE IT'S DAGE I (2) I(r, xsm; a) = Ia

FROM EQS. (1) AND (2) WE CONCLUDE THAT SYSTEM OF EFFECTIVE FOLCES REDUCE TO MI ATTACHED AT & AND A COUPLE IX.



GIVEN: 1.75-16 ROD AB P= 0.2515 L = 3610 FIND: ACCEL EMATION la) OF A. (b) DF B.



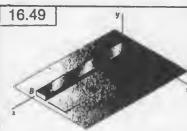
 $\sum F_{x} = \sum (F_{x})_{i} + P = m\bar{a} = \frac{W}{q}\bar{a}$ $\bar{a} = \frac{P}{w}g = \frac{0.25\%}{1.15\%}g = \frac{1}{7}g$

+ 2 Ing = I(MG) of: P= Ix = 1/2 \frac{w}{9} 12 x x=6 \frac{P}{W} \frac{3}{2} = 6 \frac{0.2516}{1.7566} \frac{9}{L} = \frac{6}{79}

(a) + a = a + = x = -1 g + = -1 g = -1 (32.21/6)

EA= 18.40 8/5-

(b) $\pm a_{g} = \bar{a} - \frac{1}{2} \propto = \frac{1}{7} g - \frac{1}{2} \cdot \frac{6}{7} g = -\frac{2}{7} g = -\frac{2}{7} (32,284/5)$ QB= 9.2 At/s2+



GIVEN: 1,25-16 ROO AB

P=0.2516

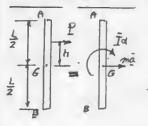
L=3ft

FINDI (a) WHERE P

SHOULD BE APPLIED

FOR QB=0.

(b) CORRESPONDING ACCEL, OF POINT A.



 $\begin{array}{ccc}
\Xi \Gamma_{x} & \Xi (F_{x})_{AF} \\
P & m\bar{a} & = \frac{W}{g}\bar{a} \\
\bar{a} & = \frac{P}{W}g & \longrightarrow \\
+2 IM_{g} & = \Xi (M_{g})_{eff} & = \frac{1}{12} \frac{W}{3} L^{2} \propto \\
N & = \frac{12Ph}{WL^{2}} 3 2
\end{array}$

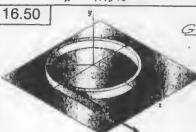
(a) $\pm \frac{\alpha}{18} = \frac{\bar{\alpha} - \frac{L}{2} \times 12Ph}{0 = \frac{P}{W}g - \frac{L}{2} - \frac{12Ph}{WL^2}g}; h = h = \frac{36m}{6} = 6in.$

THUS, P IS LOCATED 12 in. FROM END A.

FOR h= 1/6: \ \ = \frac{12P(1/6)}{WL^2} g = 2 \frac{P}{W} \cdot \frac{9}{L} \frac{2}{L}

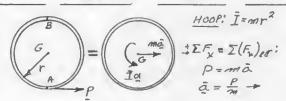
(b) $\pm a_a = \bar{a} + \frac{1}{2}\alpha = \frac{\rho}{w}g + \frac{1}{2} \cdot 2\frac{\rho}{w}\frac{g}{1} = 2\frac{\rho}{w}g$

a= 2 0.25/b (32.2 ft/s); a=9.29t/s=



GINEN: P=311 . m=2.4-hg

> [FINO: (a) a_A (b) a_B



+) $IM_6 = I(M_6) \rho q$: $Pr = \bar{I}\alpha = mr^2 \alpha$ $\alpha = \frac{P}{mr}$

(a)
$$\pm a_{A} = \bar{a} + r\alpha = \frac{P}{m} + r(\frac{P}{mr}) = 2\frac{P}{m}$$

 $a_{A} = 2\frac{3N}{2.4 \cdot 69} = 2.5 \frac{m}{s^{2}}$
 $a_{A} = 2.5 \frac{m}{s^{2}}$

(b) $\pm a_8 = \bar{a} - r\alpha = \frac{P}{m} - r(\frac{P}{mr}) = 0$

an=0

16.51 and 16.52

All the state of t

GIVEN: P=3N m=2.4kg

PROBLEM 16.51

FIND:
(a) aA

(b) aa

FROBLEM 16.52

SHOW THAT FOR 360° ROTATION DISK WILL MOVE DISTANCE TY

 $\frac{D/s\kappa}{}: \bar{I} = \frac{1}{2}mr^{2}$ $+ \sum_{\kappa} \sum_{k} \sum_{k} (F_{\kappa})_{k} r^{k}$ $P = m\bar{a}$ $\bar{a} = \frac{P}{m}$

+) $IM_G = I(M_{der})$ $Pr = \overline{I} \propto$ $Pr = \frac{1}{2}mr^2 \propto$ $\propto = \frac{ZP}{2mr}$

PROBLEM 16.51

(a)
$$\frac{1}{2}$$
, $a_{A} = \bar{a} + r\alpha = \frac{\rho}{m} + r \cdot \frac{2\rho}{mr} = 3\frac{\rho}{mr}$
 $a_{A} = 3\frac{3N}{2.480} = 3.75 \text{ m/s}^{2}$
 $a_{A} = 3.75 \text{ m/s}^{2}$

(b) $\frac{1}{2} \cdot O_B = \hat{a} - Y\alpha = \frac{P}{m} - Y \cdot \frac{2P}{mY} = -\frac{P}{m}$ $\alpha_B = -\frac{3N}{2.4 + 69} = -1.25 \, m/s^2$ $\alpha_B = 1.25 \, m/s^2 = -1.25 \, m$

PIZOBLEM 16.52

LET $t_i = TIME$ REGINED FOR 360 KOTATION $G = \frac{1}{2} \times t_i^2; \quad 2\pi \text{ rod} = \frac{1}{2} \left(\frac{2P}{\pi r}\right) t_i^2$

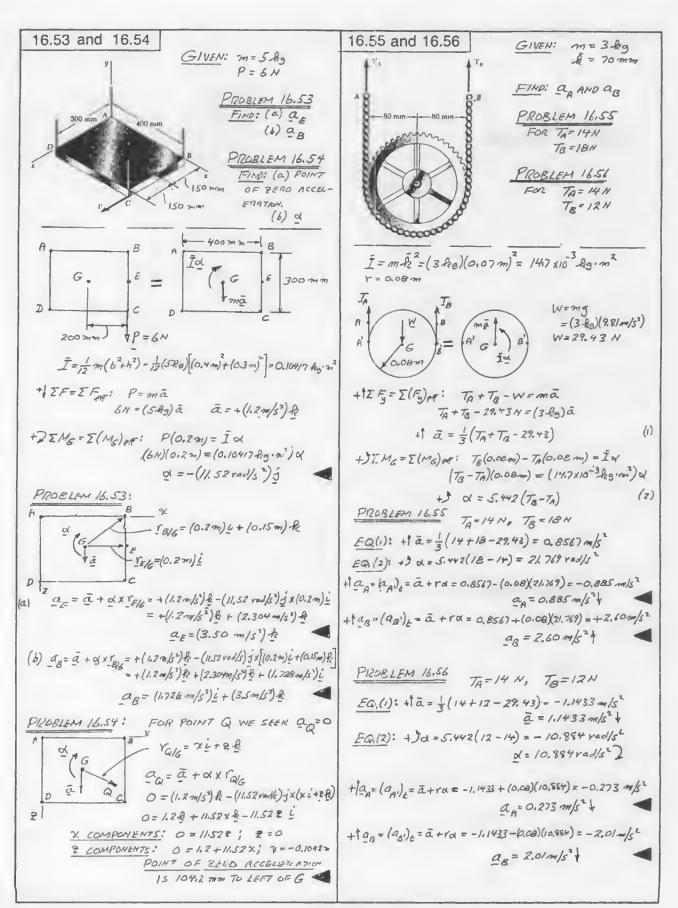
t, 2 = 211 mr

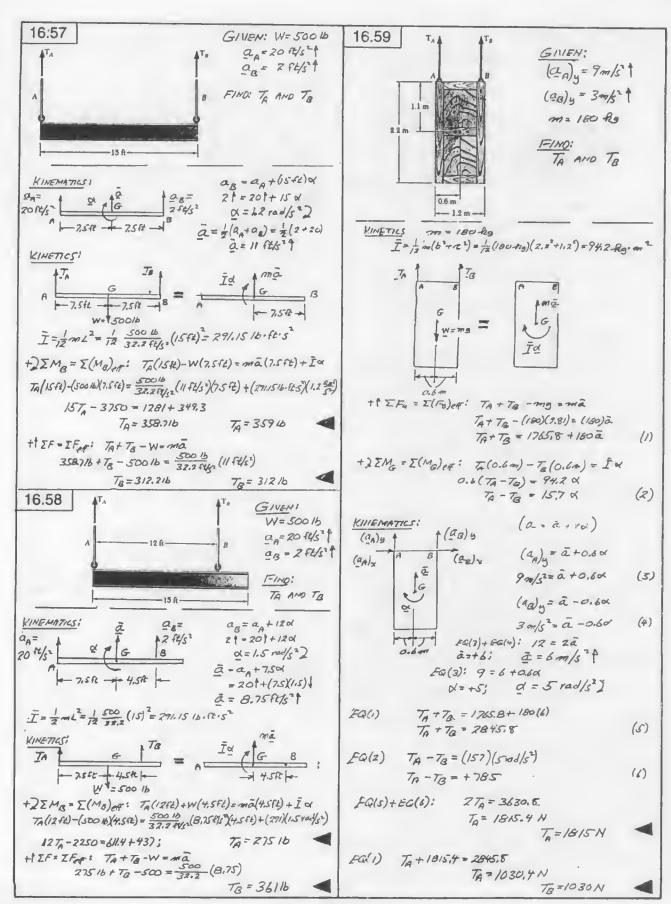
LET 4, = DISTANCE & MOVES

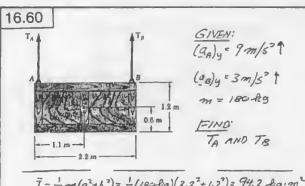
DURING 360° ROTATION

 $\gamma_{i,j} = \frac{1}{2}\bar{a}t_{i,j}^{2} = \frac{1}{2}\frac{P}{m}\left(\frac{2\pi mr}{P}\right)$

x,=TTr Q.E.D.







I= = 1 m(a2-b3)= 1/2 (180 fig)(2,22+1,23)= 94,2 fig.m2

KINETICS:

$$\begin{array}{c|c} |CINEMATICS| \\ (a_{A})_{y} \\ (a_{A})_{y} \\ (a_{A})_{y} \\ (a_{A})_{y} \\ (a_{A})_{y} = \bar{a} + 1.1 \times \\ (a_{B})_{y} = \bar{a} + 1.1 \times \\ (a_{B})_{y} = \bar{a} + 1.1 \times \\ (a_{B})_{y} = \bar{a} - 1.1 \times \\ (a_{B})_{y} = 1.1 \times \\ (a_$$

$$EQ(3) + EQ(4)$$
: $12 = 2\bar{a}$
 $\bar{a} = +6\pi/5^{2}$ $\bar{a} = 6\pi/5^{4}$
 $EQ(3) - EQ(4)$: $6 = 2.24$

$$EQ(3) - EQ(4)$$
: $6 = 2.24$
 $\alpha = 2.727 \text{ rod/s} \qquad \alpha = 2.73 \text{ rod/s}^2$

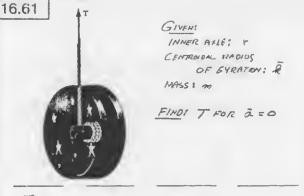
EQ(1):
$$T_A + T_B = 1765.8 + 160(6)$$

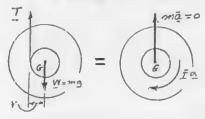
 $T_A + T_B = 2845.8$ (5)

$$EQ.(2);$$
 $T_A - T_B = 85.636(2.727)$ $T_A - T_A = 233.5$ (6)

$$T_A - T_B = 233.5$$
 (6)

$$EG(1)-EG(2)$$
: $2T_B = 2612.3$
 $T_B = 1306.2^{1}$
. $T_B = 1306N$

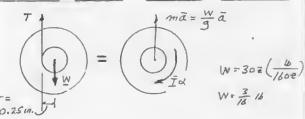




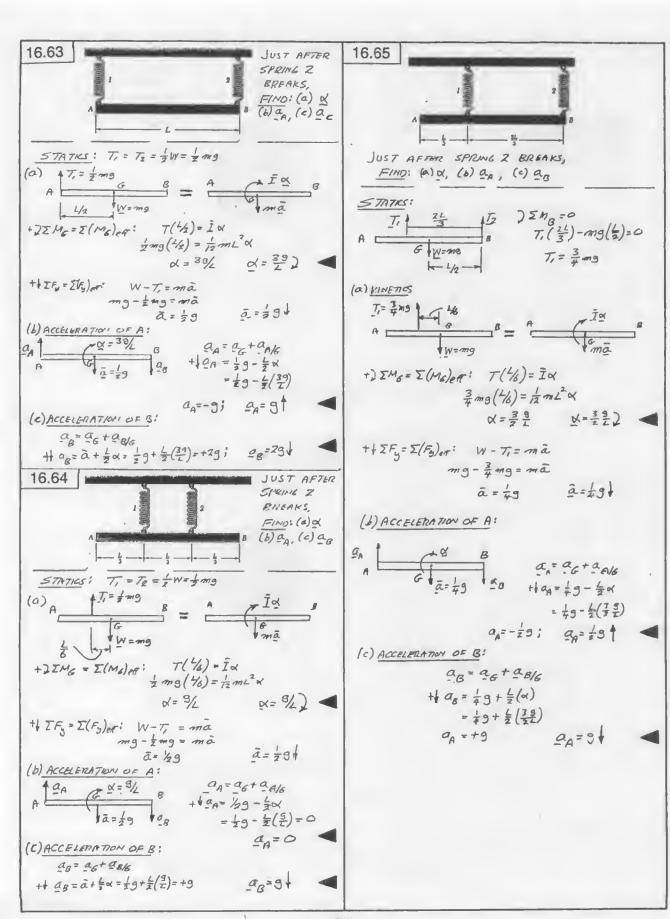
+ 1 IFy = [(Fy)er: T-mg=0; T=mg

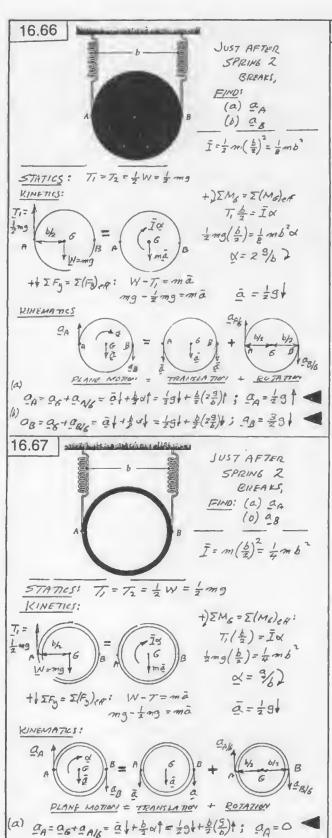
+)
$$IM_6 = I(M_6)_{AF}$$
: $Tr = \tilde{I} \times M_6 \times M_6$



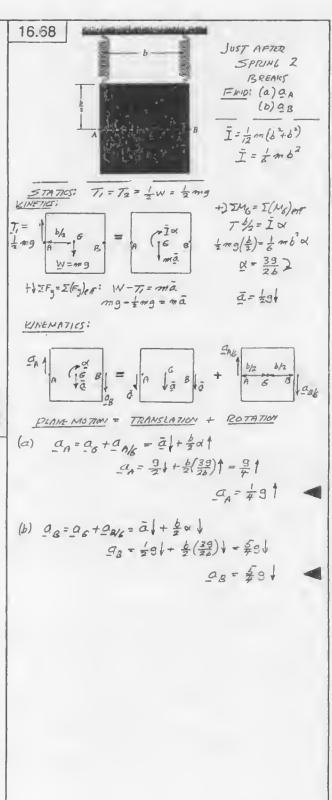


+
$$\sum M_6 = \sum (M_6)_{ef}$$
: $Tr = \overline{1} \ d$
 $(0.205 \ b) \left(\frac{0.25}{72} \ fe\right) = m - \overline{fe}^2 \ d$
 $4.271 \times 10^3 \ lb \ ft = \frac{3/6 \ lb}{32.27 \ fe}_3 \times \left(\frac{1.25}{12} \ fe\right) \ d$
 $\alpha = 67.6 \ rod_6^2 \qquad (2-67.6 \ red_1)$





(b) a3=a6+a8/6= a1+ 2 × = = = 9+ € (96); 08=91



16.69 and 16.70 GIVIEN: \$ = 15 12/5 -V=4in., 4/x=0.10. PROBLEM 16.691 W= 9 rads Prog LAM 16.70: W.= 18 reds FIND: (a) t, WHEN ROLLING STORY 16) V AT E, (C) DISTANCE TRAVELED AT E, KINETICSE I ΣFx = Σ(Fx) pp yng = mā N= mg | F=1 mg + IEM = I(Ma) +: Fr= IX (1/4mg)r===mr? d= 5 1/2) KINE MATICS: WHEN SPHERE ROLLS, INSTANT CENTEL OF ROTATION IS AT C AND WHEN tot, V= ra V=V0-at = V0-1/29 t a=-wo+d1 = -wo + 5 749 t WHEN t=t .: To-1/20t, = (- Wo + 5 - 1/29 t) + EQU: V= rw: Vo-4, gt, =- Nor+5 1/292, Z,= 2 (vo+rwo) PROBLEM 16.81 25= 15 Ft/s, W= 9 rad/s, r=4in = 1 ft (b) FQ.(2): V= Vo -1/9 t, = 15-0.1 (32.2) (1.5972) 8,= 15-5.1429= 2857 Pt/s (c) a= 1/2 9 = 0.1 (3229/5") = 3.22 81/6" -\$ 5, = v, t, - 1 at,2

= (15ft/s)(1.5975) - \frac{1}{2}(3.22ft/s2)(1.5975)2

t, = 1.8635

2,=22.4 € →

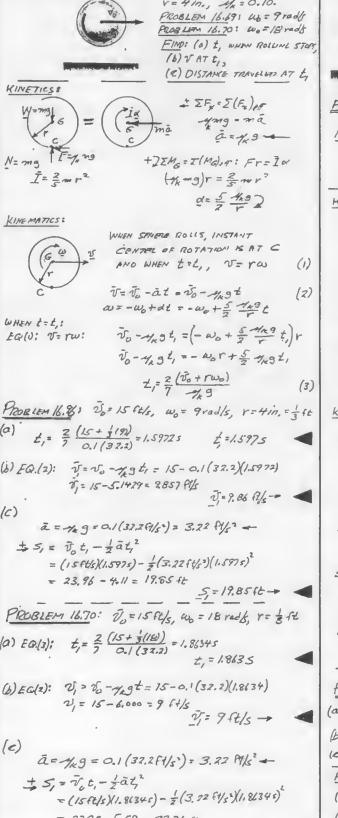
= 23.96 - 4.11 = 19.85 ft

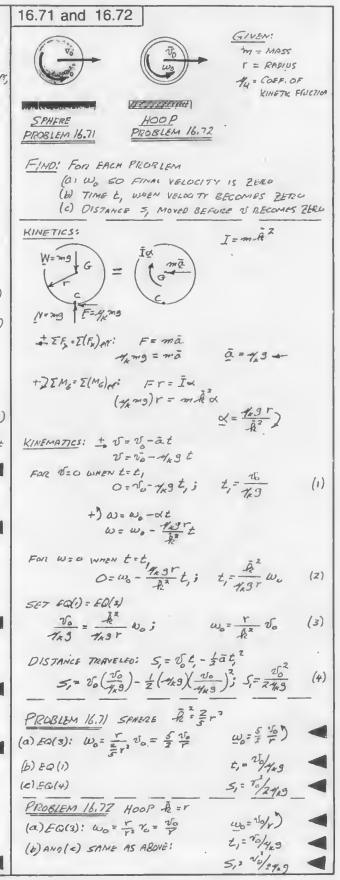
(a) EQ.(3): $t_1 = \frac{2}{7} \frac{(15 + \frac{1}{3}(14))}{Q_1(37.2)} = 1.86345$

+ 5,= v,t,- fat,2

= 27.95-5.59 = 22.36 Ft

(c)





16.73



GIVEN: SPHERE PLACED

SI EULT HITH NO VELOUTY.

T = RADIUS,

A/K = COÉF. KINETIC FALCTON

FIND: (a) t WHEN SPHERE DOLL

1/k = COEF. KINETIC FALCTON

FIND: (a) t WHEN SPHERE ROLLS

(b) V AND WI WHEN t:t,

KINETICS:

Walna G = 6 mid

-4,79 N=79

KINEMATICS: $\pm \bar{v} = \bar{a}t = \gamma_{x}gt$ (1)

12)

C= POUNT OF CONTACT WITH EELT $\frac{1}{2} N_{c} = \overline{U} + WV = \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} r$ $N_{c} = \frac{2}{2} \frac{1}{2} \frac{1$

+) w= dz = 5 -169 t

(a) WHEN SPHENS STARTS ROLLING (t=t,), WE HAVE

V=V; V,= 7-4,9 t,

(b) VELOGIPES WINTER E=t,

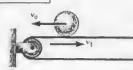
EQ(1): \$= 49 (= 1/4,9)

ਭੋ= = 3 ਹ, →

EQ(2): $\omega = \left(\frac{5}{2} \frac{\gamma_{+} 9}{r}\right) \left(\frac{2}{7} \frac{v_{1}}{\gamma_{+} 9}\right)$

而是在了

16.74



GIVEN: SPHERE WITH IS AND WO = O PLACED

ON RELT. Y = RADIUS

ME = COEF. KINETIC FRICTION

FIND: (a) YO SO THAT

SPHERE WILL HAVE NO

LINEAR VELOCITY AFTER IT STARTS WOLLING ON LITER, (b) t, WHEN SMERE STARTS ROLLING, (c) DISTANCE SPHERE WILL HAVE MOVED LYHEN t=t,

 $\pm \sum F_{g} = \sum F_{g} \cdot F_{g} \cdot F_{g} = ma$ $\Delta = 1/2 \Rightarrow$ $+ \sum F_{g} = \sum F_{g} \cdot F_{g} \cdot F_{g} = \sum F_{g} \cdot F_$

KINEMATICS: $\pm \vec{v} = \vec{v}_0 - \vec{a}t - \vec{v}_0 - \eta_R gt$ (1) $+ \vec{v} = \vec{v}_0 + \vec{$

(CONTINUED)

16.74 continued



C = POINT OF CONTACT WITH BELT $\frac{1}{2} T_C = -\bar{V} + \Gamma \omega$ $v_C = -\bar{V} + \Gamma \frac{5 + 4.9}{2 \Gamma} t$ $v_C = -\bar{v} + \frac{5 + 4.9}{2} t$ (3)

BUT, WHEN t=t, $\overline{V}=0$ AND $\overline{V}_{c}=\overline{V}_{c}$ FG(a): $\overline{V}_{j}=\frac{5-4k^{2}}{2}\overline{Z}_{j}$; $\overline{Z}_{j}=\frac{2\overline{V}_{j}}{5-4k^{2}}$

FQ(1): $\vec{v} = \vec{v}_0 - H_0 \vec{z}$ WHEN $t \cdot t_1$, $\vec{v} = 0$, $0 = V_0 - H_0 \left(\frac{2 \vec{v}_1}{5 H_0} \right)$; $\vec{v}_0 = \frac{2}{5} \vec{v}_1$

DISTANCE WHEN $t = t_i$: $t \le z_0 t_i - \frac{1}{2} a t_i^2$ $S = \left(\frac{2}{5} \sqrt{1}\right) \left(\frac{2 \sqrt{1}}{5 \sqrt{19}}\right) - \frac{1}{2} \left(\frac{4}{19} 9\right) \left(\frac{2 \sqrt{1}}{5 \sqrt{19}}\right)^2$ $S = \frac{{v_i}^2}{4 \sqrt{19}} \left(\frac{7}{25} - \frac{2}{25}\right)$; $S = \frac{2}{25} \frac{{v_i}^2}{7 \sqrt{19}}$

16.75



SHOW THAT I'D (FIGHERS) CAN BE ELIMINATED BY ATTACHING

May AND Man AT POINT P ON OG WHERE GP = \$2/7

F16 16.15 b

| Tax | man | man

06= r a = r x

WE FIRST OBSERVE THAT THE SUM OF THE VECTORS IS THE SAME IN BOTH FIGURES TO HAVE THE SAME SUM OF MOMENTS ABOUT G, WE MUST HAVE

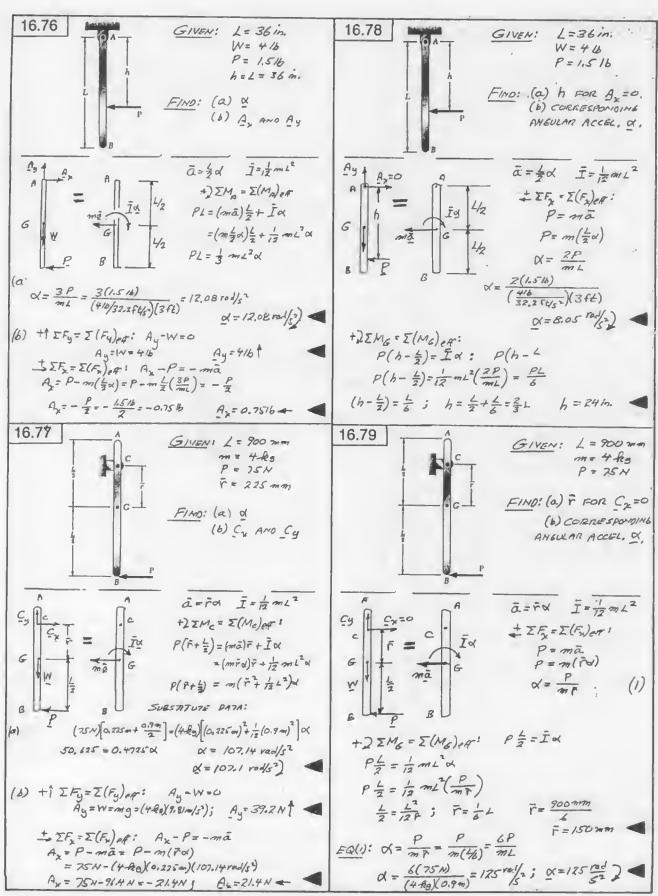
+) $EM_6 = EM_6$: $\tilde{J} \propto = (m\tilde{a}_1)(6P)$ $m\tilde{h}^2 \propto = m\tilde{r} \propto (6P)$ $EP = \frac{\tilde{k}^2}{\tilde{r}}$ (QE.O.)

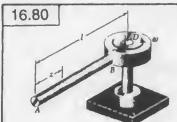
MOTE: THE CENTER OF ROTATION AND THE CENTER OF PERCUSSION ARE INTERCHANGEABLE, INDEED, SINCE OF F, WE MAY WRITE $6P = \frac{3}{60} \quad \text{or} \quad 60 = \frac{1}{6}$

THUS, IF POINT P IS SELECTED AS CENTER.

OF LUTATION, THEN POINT O IS THE CENTER.

OF PERCUSSION.





GIVEN: W= 0.25 16/52 1 = 1.2ft 00 = 150 ypm 2 = 0,9 ft

FIND: TENSION IN ROD (a) IN TERMS OF W, l

2. AND W. (b) FOR GIVEN DATA

a= rw= (1-1/2) w2 IN HORIZONTAL PLANE:

$$\frac{1}{2} \sum_{z} F \cdot \sum_{z} F_{z} = \frac{\omega}{3} z$$

$$= \left(\frac{\omega}{3} z\right) \left(1 - \frac{z}{2}\right) \omega^{2}$$

$$T = \frac{\omega}{3} \left(1 z - \frac{z^{2}}{2}\right) \omega^{2}$$

SUBSTITUTE DATA: : W= 150 rpm (21) = 51T nad/s, 2=0.995

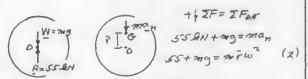
T= 0.25 16/ft [1.274)(0.9 ft) - (0.9 ft)2 (577 rad/s)2 T= 1.293 16



GIVEN: FLYWHEEL, CENTER OF ROTATION AT O, AND MASS CENTER AT G W= 1200 rpm, MAXIMUM FORCE EXERTED ON SHAFT 15 55 ANT AND 85 AN . FIND: (a) MASS OF FLYWHEEL (b) DISTANCE F

W = 1200 rpm (21)= 40 11 rod/s a = rw2

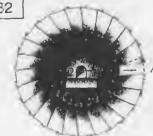
+ TF = ZFeF BSAN-mg=Man 85 - Mg= mF 05



30 kH - 2mg = 0 EQ(2) - EQ(1): 30×103 N = 2 m (9.81 m/53)

m=1529 kg EQ(1) + EQ(2): 140 AN = 2 m rw 140 x103 N = 2(1529 Ag) + (407)2 r= 2.90 x10 m 7=2.90 mm

16.82



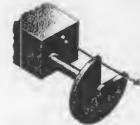
GIVEN: A 45-9 VANE IS THROW OFF FROM BALANCED TURBIN DISK. au = 9600 rpm FIND: REACTION AT O

W=9600 rpm (211) W= 320 11 rod/s CONSIDER VANE BEFORE IT IS THROWN OFF

R=man=nnFw2 = (45110 3 Rg)(0,3 m)(320 71)2 R=13,64-RN

BEFORE VANE WAS THROWN OFF DISK INAS BALANCED (R=0). REMOVING VANE AT A ALSO REMOVUS . TS REACTION, SO DISK IS UMBALAMEND AMO REACTION IS R= 13.64RN-

16.83



GIVEN: 0.125-11 SHUTTER OF RADIUS 0.75 in. W= 24 rades per second

FIM: MAGNITUOE OF FORCE EXELTED ON SHAFT BY SHUTTELL

d=3/1 V = 0.75 in

SEE INSIDE FRONT COVER FOR CENTRUD OF A CIRCULAR SECTOR

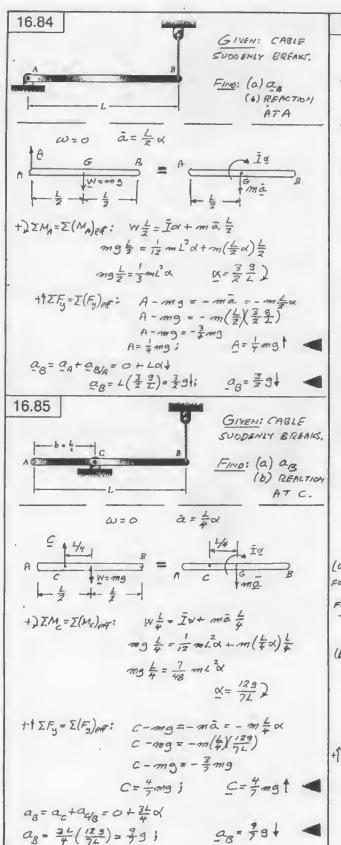
F= 2r Sind T= 2(0.75 in.) sin (37)

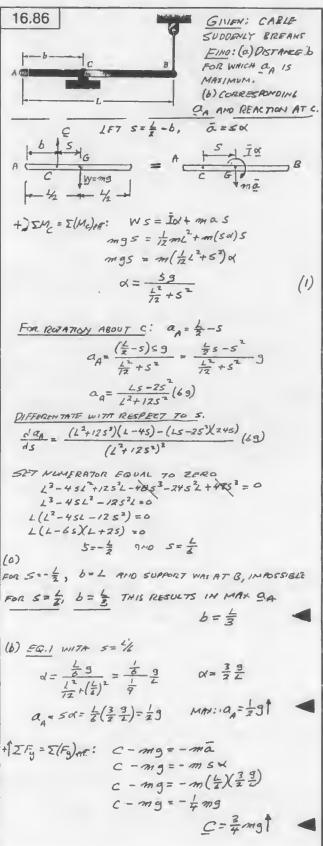
P= 0.15005 in.

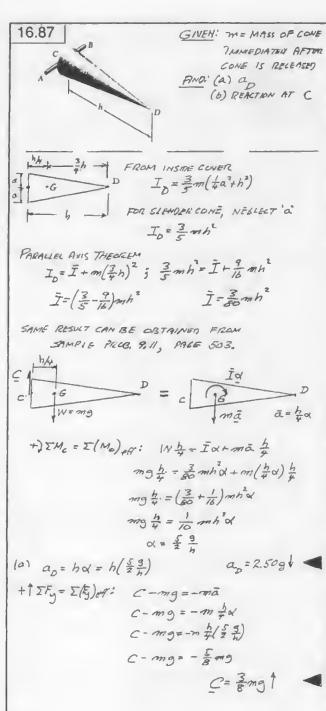
an= Fw w= 24 rov = 24(211) rad W= 150. B radf

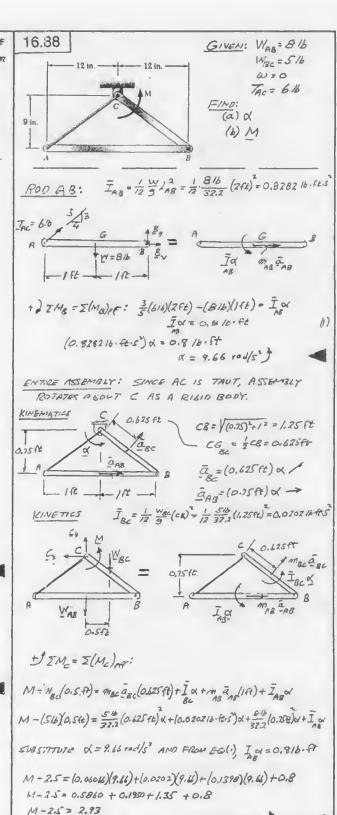
+ 2= = 2 FAT R= man=mrw2 = (0.12516) (0.15005 ft) 150.8 rods) 12= 1.1038 16 ×

FORCE ON SHAFT IS R=1.10416/ MASHITUDE: R=1.10416



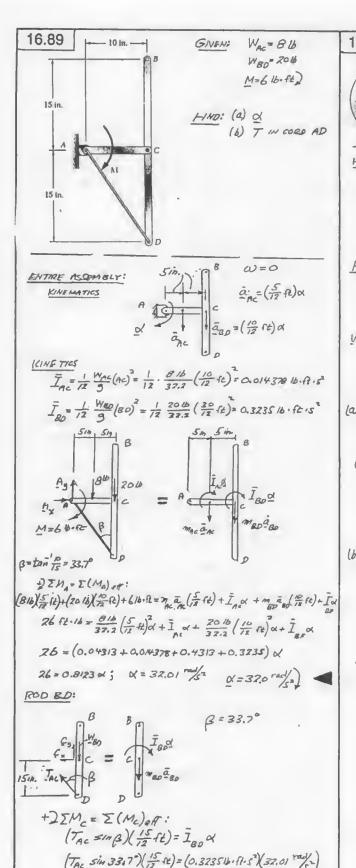




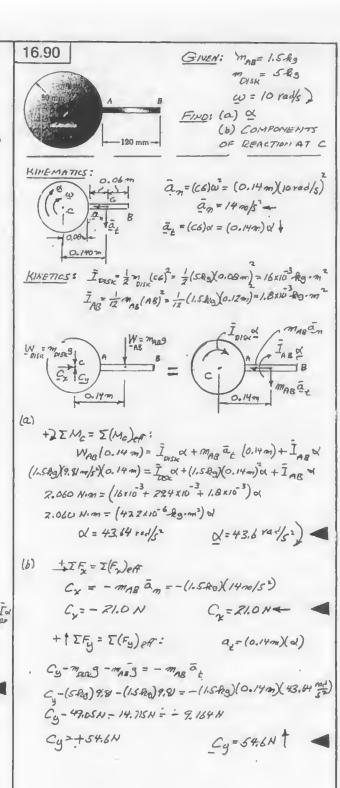


M=5.73 16. Ft)

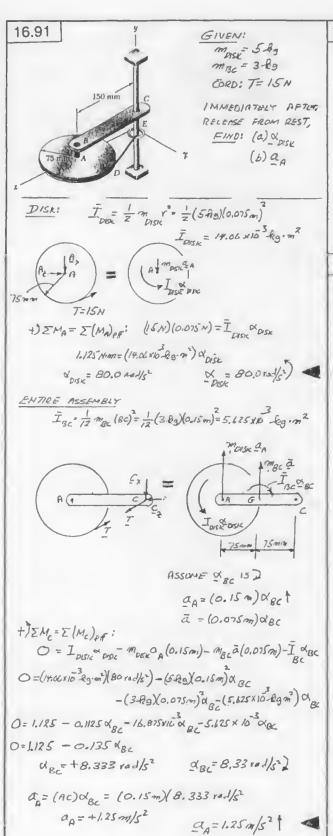
M=+5.43 16-ft



TAC= 14.93 16



TAC= 14.93 16



NOTE: ANSWERS CAN ALSO BE WRITTEN!

a=-(1.25m/52) L

d = (Boredby)j

16.92 DERIVE IM: I'M FOR THE BOLLING DISK OF F16.16.17. W=mg +) IMc = [(Mc)ef: EME= (ma)r+IX = (m rx)8+ I of IM = (mr2+I)d BUT, WE KNOW THAT I = mr + I THUS: EM = ICX (a. E. D.) 16.93 FOR All UNBALANCED DISK SHOW THAT EM = I & IS VALID ONLY WHEN THE MASS CENTER G, THE GEOMETRIC CENTER O, AND

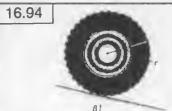
THE INSTANTANEOUS CENTER C HAPPEN TO LIE

EMC = \(\int(M_c)\) A : \(\int(M_c) = \int(M_c) \times M_c \ti

FOIT REDUCES TO IMET ON MEN SOLXMAC = O
THAT IS, WHEN YOU AND A ARE COLLINEAR.

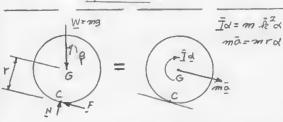
REFERRING TO THE FIRST DIAGRAM, WE NOTE
THAT THIS WILL OCCUR ONLY WHEN
POINTS G,O, AND C LIE IN A STRAIGHT LINE.

(Q,E,D.)



GIVEN: ROLLING WHEEL

FIMS: à IN TERMS OF I, À, B, AM 9.



+) $\Sigma M_c = \Sigma (M_c)_{eff}$; $(W sin \beta) r = (m \bar{\alpha}) r + \bar{1} \propto$ $(mg sin \beta) r = (m r \alpha) r + m \bar{k}^2 \propto$ $rg sin \beta = (r^2 + \bar{k}^2) \propto$

a=rd=r rg sin B

$$\bar{a} = \frac{r^2}{r^2 + \bar{k}^2} 9 \sin \beta$$



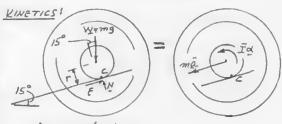
GIVEN: STARTING
FROM REST, FLYWHEEL
MOVES 16 ft M 405
Y=1.5in.

FIND: I

KINEMATICS: $S = \sqrt{5}t + \frac{1}{2}at^{2}$ $16t = 0 + \frac{1}{2}a(40s)^{2}$ $a = 0.02 \text{ ft/s}^{2}$ Since r = 1.5 in = 0.125 ft

THE Y = 1.5 In = 0.125 ft

\[\bar{a} = Ya ; \quad 0.02 ft/s^2 = (0.125 ft) \text{ \text{\$\delta} } \\
\text{\$\delta} = 0.16 \text{ \text{\$rod/s}}^2 \\
\end{a}



+) IMc=[(Mc)eff: (mg sin 15°)r = I & + (mā)r (mg sin 15°)r = mæa + (mra)r gr sin 15° = (m²+r²) &

DATA: r = 0.125 ft, $d = 0.16 \text{ rad/s}^2$ $(32.2 ft/s^2)(0.125 ft) \sin 15^0 = (\frac{1}{12} + r^2)(0.16 \text{ rad/s}^2)$ $\frac{1}{12} + r^2 = 6.511 ft^2$ $\frac{1}{12} + (0.125 ft)^2 = 6.511 ft^2$

 $-\bar{R}^2 + (0.125 \text{ ft})^2 = 6.571 \text{ ft}^2$ $-\bar{R}^2 = 6.4953$

\$ = 2.55 ft

16.96



GIVEN:

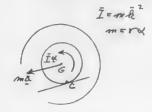
\$\bar{R} = CPATIROIOAL

PAOIUS OF GYIZATIUN

\$\mathre{H}_5 = COEF, STATIC FRICTION

FINDS LARGEST B FOR ROLLING WITHOUT SLIPPING

B W= mg



+) $\sum M_C = \sum (M_C)_{eff}$: $(mg sing) r = \overline{I} \omega + (m\overline{a}) r$ $mg sing r = m \overline{R}^2 \omega + m r^2 \omega$ $\omega = \frac{gr}{r^2 + \overline{a}^2} sing \qquad (1)$

F- mg sin B = -ma

F- mg sin B = -mr x

F= mg sinB - mr x

+ TF= IFAT: N-mgcosp=0 N=mgcosp

IF SLIPPING IMPENOS $F = M_S N$ OR $M_S = \frac{F}{N}$ $M_S = \frac{F}{N} = \frac{mg \sin \beta - mr \alpha}{mg \cos \beta} = \frac{\sin \beta - \frac{c}{3} \alpha}{\cos \beta}$

SUBSTITUTE FOR & FROM EQ(1)

$$M_{S} = ton \beta \left[1 - \frac{r^{2}}{r^{2} + \hbar^{2}}\right] = ton \beta \left[\frac{\hbar^{2}}{r^{2} + \hbar^{2}}\right]$$

$$ton \beta = M_{S} \frac{r^{2} + \hbar^{2}}{\hbar^{2}}$$

ton B=4/1+(1/2)]

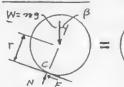


GIVEN: PIPE CYLINDER AMD SPHETCE ARE RELEASED FROM REST. AFTER 4 S, FIND

DISTANCE BETWEEN (a) PIPE AND CYLINDER

(b) CYLHOUR AND SPHERE.

I=mR2 GENERAL CASE!





+ 2 IMC = I (Mc) AT (wsing) r= I a+ mar mgsingr= mfa+ord d= rg sin B

a=ra=r rg sing $\bar{Q} = \frac{r^2}{r^2 + \hat{\beta}^2} g S m g$ (1)

 $\bar{a} = \frac{r^2}{r^2 + r^2} 9 \sin \beta = \frac{1}{2} 9 \sin \beta$ FOR PIPE: BET

FOR CYLINDER: R= 1/2 = = 12 g sing = 2 g sing

FOR SPHERE: $\vec{A} = \frac{1}{5} = \frac{1}{$

(a) BETWEEN PIPE AND CYLINDER

a= = a-a= (2-1)9 sin B = 1 g sin B

Yelp = 1 adpt = 1 (6 9 sing) t

SIUNITS: YC/p= 1/(\$ 9.81 m/5) sin/0 (4 s) = 2,27 m

USUMITS: 7/1/p= 1/2 32.2 Flys) sin 10 (45) = 7.46 FL

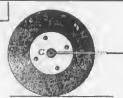
(b) BETWEEN SPHERE AND CYLINDER

ast= as-a= (5-2)9 sing = 1 9 sin B 25/= 1 205/c = 1 (1 9 sing) t2

51 UNITS: 75 = 1/2 (1/21 9.81 m/s2) sin 10" (4 s) = 0.649 m

US ONITS: $\chi_{S/c} = \frac{1}{2} \left(\frac{1}{2}, 32.2 \text{ ft/s}^2 \right) \sin 10^2 (45) = 2.13 \text{ ft}$

16.98



GIVEN: W= 1016 1 = 81n. 1 = 4in. -R = 6 in. P = 516 15=0.25, 41=0.20

FIND: (a) DOES DISK SLIDE, (b) of AND a.

在=ra=(日ft) o ASSUME DISK ROLLS:

W=10161

I= m-R2 = 10/b (6/2 12)2 I = 0,07764 16. ft. 5

+) IN: - [(Mc) pq: (516) B ft) = (ma)r + I x 3.333 /b.R = 10/b (8 ft) x + 0.077640

3.333 = 0.21566 d x=15.456 ro3/52 a=rx=(B 12)(15.456 rad/s")

· X = 15.46 rod/3) ā = 10.30 ft/s2 ->

\$ ΣFx = Σ(Fx)er: -F+516 = mo -F+510= 1010 (10.30 F4/52); F=1.801b N=1015

HTERY = I (Fy) ex: N-1016 = 0

Fm=.45 N=0.25(1016)= 2.516 DISK ROLLS WITH NO SLIDING

16.99



GIVEN: IN=1016 To- 8in, 1:=4in. A=6in, P=515 45=0.25, 1/k=0.20 FIND: (a) DOES DISK SLIDE. (b) & AND a

a=ra=(=ft)x ASSUME DISK ROLLS:

W=1014 ma 12in.

Ī= m 1 2 I=0,07764 164tis

+) IMc = I(Mc) + F: (516)(192) = (ma) + IV 5/6. A= 100 (8 ft) x +0.07764 x

5 = 0,21566 a

0 = 23.184 rad/s2 a=rx=(=ft)(23.184 52)

X = 23,2 rad/52) a= 15.46 11/52-

+ ΣF = Σ(Fx)eR: -F+5/6 = ma

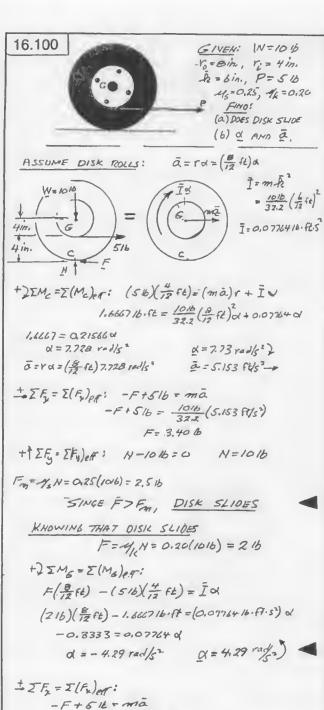
-F+516 = 1016 (15.46 1/32); F= 0.2016

+ TFy= I(Fy) eff: N-1016 =0

N=1016

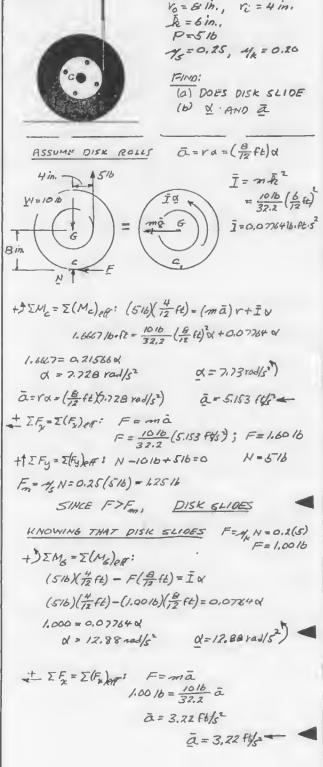
Fm=45/N=0.25(1016)=2.516

SINCE F < FM, DISK ROLLS WITH NUSLIONG



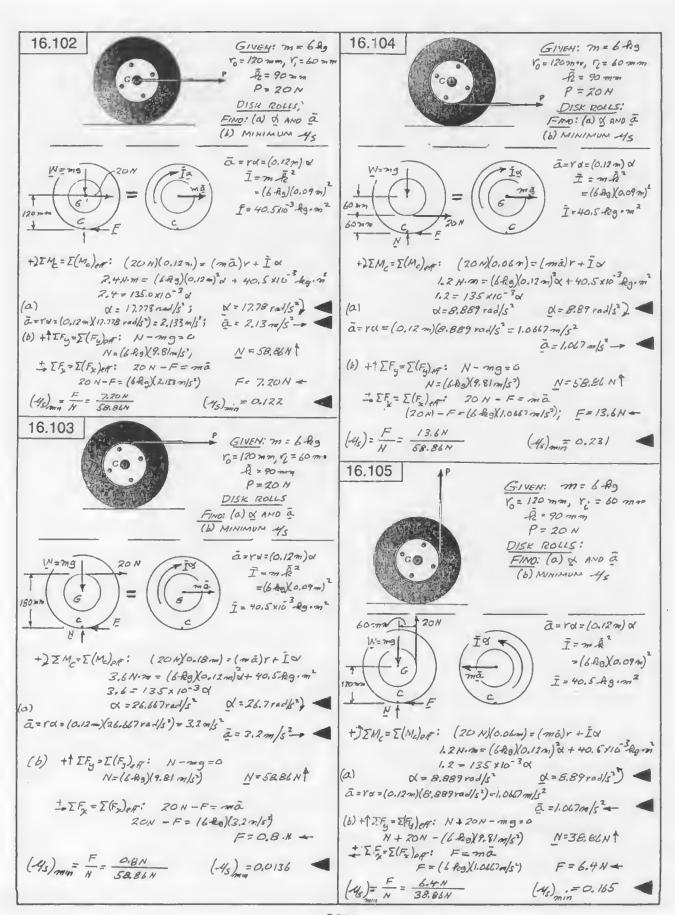
-216+510 = 1016 à

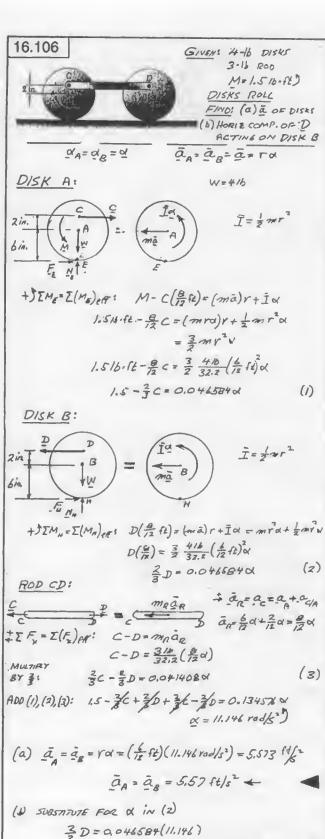
a= 9.66 ft/2 == 9.66 ft/2->

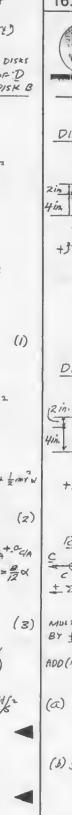


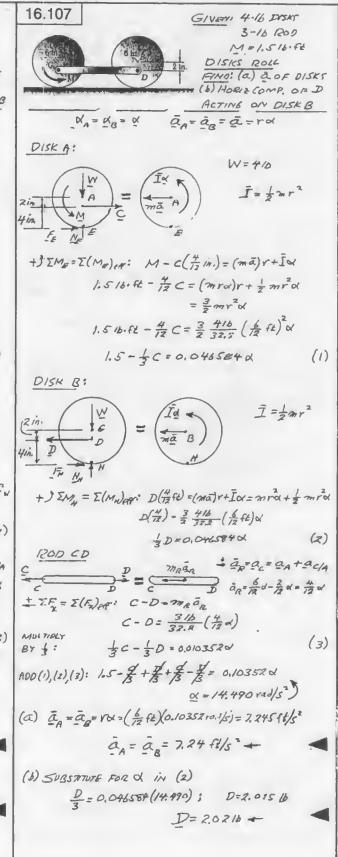
GIVEN: IN=1016

16.101

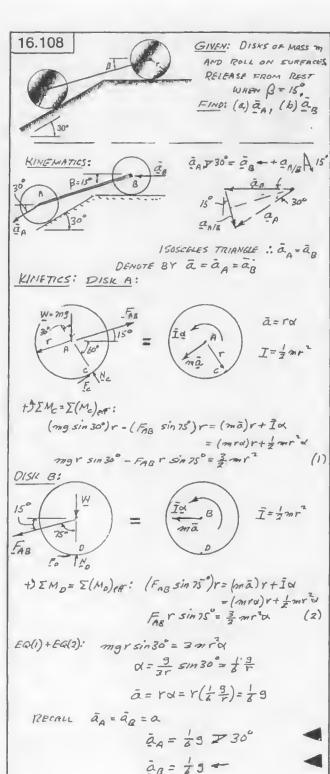


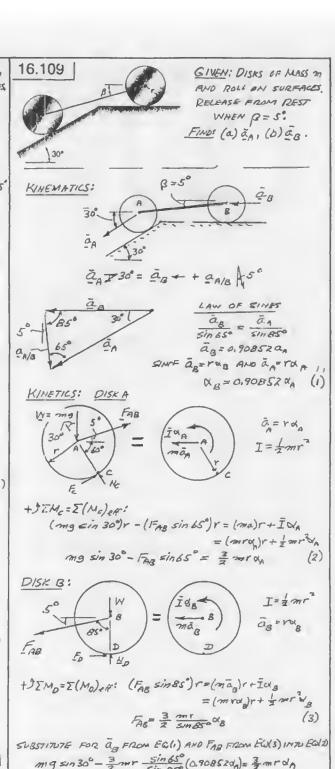






D=0.77916-





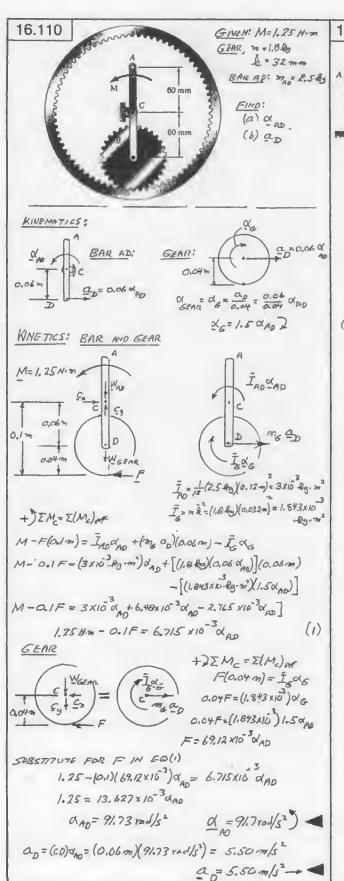
SUBSTITUTE FOR \bar{a}_{g} FROM EQ(1) AND FAB FROM EQ(3) INTO EQ(2)

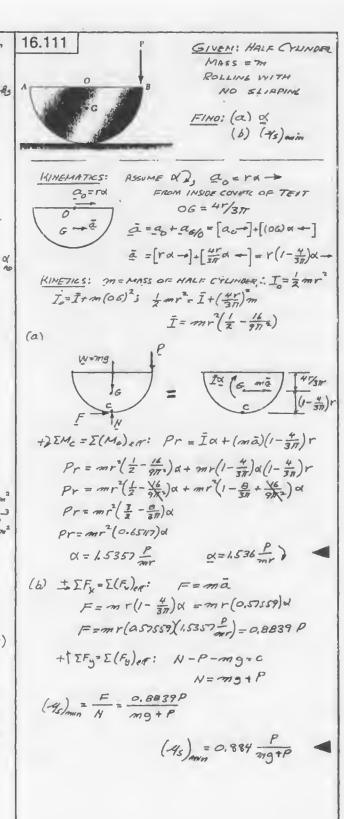
MIG SIN 30° - $\frac{3}{2}$ mr $\frac{\sin 65^{\circ}}{\sin 85^{\circ}}$ (0.90852d_A)= $\frac{3}{2}$ mr d_A

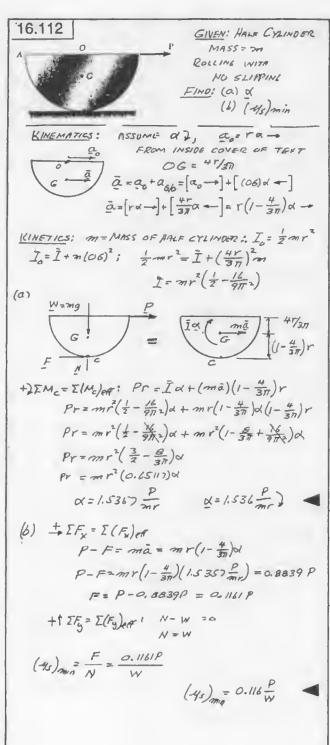
0.5 $\frac{9}{V}$ = $\frac{3}{2}$ (0.8265+1) d_A

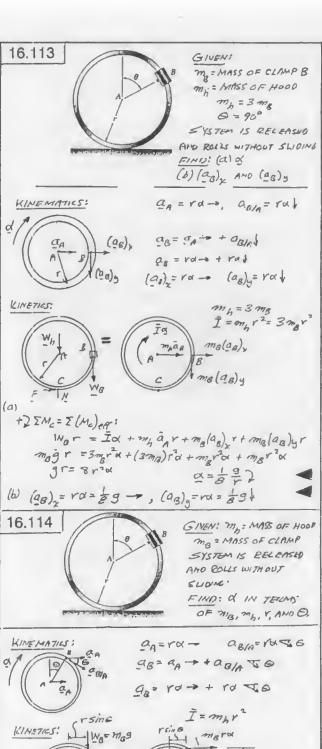
d_a= 0.1825 $\frac{9}{2}$: \bar{a}_{A} = V_{A} = 0.1825 9

 $\bar{a}_{A} = 0.18259 \text{ } 730$ $\bar{a}_{B} = 0.90852 \bar{a}_{A} = (0.90852)(0.18269) = 0.16599$ $\bar{a}_{B} = 0.16699$









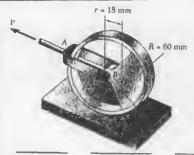
16.114 continued

magrains = mnr2+mn(rd)r+mara(1+cose)(r+rcose)
+marasine(rsine)

 $mgrsin6 = 2m_{h}r^{2}\alpha + m_{g}r^{2}\alpha [(1+cos6)^{2} + sin^{2}6]$ $= 2m_{h}r^{2}\alpha + m_{g}r^{2}\alpha [1+2cos6 + cos^{2}6 + sin^{2}6]$ $mgrsin6 = r^{2}\alpha [2m_{h} + m_{g}(2+2cos6)]$

 $x = \frac{g}{2r} \frac{m_B \sin \theta}{m_h + m_B (1 + \cos \theta)}$

16.115 and 16.116



GIVEN: m = 1.5 Rg

PROBLEM 16.115:

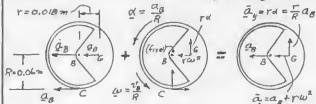
FIND: P WHEN

VB = 0.35m/s
a = 1.2 m/s
PROBLEM 16.116:

FIND: P WHEN

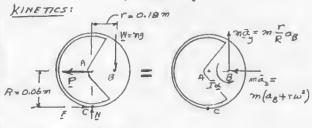
VB = 0.35 m/s
a = 1.2 m/s -

KINEMATICS: CHOOSE POSITIVE VE AND & TO LEFT



TRANS. WITH B + ROTATION ABT B = ROLLING MOTHON

$$\bar{a} = \left[a_B + r\omega^2 \right] + \left[\frac{r}{R} a_B \right] \uparrow$$



+) $\Sigma M_c = \Sigma(M_c)_{eff}$: $PR - Wr = (m\bar{a}_g)r + (m\bar{a}_{\chi})R + \bar{L} \propto$ $PR - mgr = m(\bar{R}a_g)r + m(a_g + rw^2)R + m\bar{k}^2 \frac{a_g}{R}$ $= ma_g(\frac{r^2}{R} + R + \frac{\bar{R}^2}{R}) + mr(\frac{\sqrt{g}}{r})^2R$ $P = mg(\frac{r}{R}) + ma_g(1 + \frac{r^2 + \bar{R}^2}{Q^2}) + m\frac{r}{Q^2}V_R^2$ (1)

(CONTINUED)

16.115 and 16.116 continued

\$= 0.044m AND g=9.81 m/s in EQ(1)

P=1,5(9,81) 0,06 + 1,5(08)(1+ 0,018+0,044) +15 0,06 V8

P= 4.4145 + 2.4417 aB + 7.5 25 2 (2)

PROBLEM 16.115: \(\sigma = \alpha 35 \overline{s} = \big ; \quad \qquad \qquad \qquad \qquad \qquad \qqqq \qqq \qqqq \qqq \qqqq \qqq \qqqq \qqq \qqqq \qqqqq \qqqq \qqqqq \qqqq \q

SUBSTITUTE IN EQ(Z).

P= 4.4145 + 2.4417(+1.2)+7.5(+0.35)²
= 4.4145 + 7.9300 + 0.9188 = 42.263 N

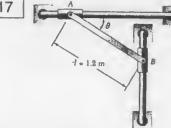
P= 8.26 N +

PROBLEM 16.111: RESALL WE ASSUMED POSITIVE TO LEFT $\frac{\partial g}{\partial s} = 0.35 \, \text{m/s} \longrightarrow ; \quad v_g = -0.35 \, \text{m/s}$ $a_g = 1.2 \, \text{m/s} \longrightarrow ; \quad a_g = -1.2 \, \text{m/s}^2$ SUBSTITUTE INTO 4-4(2):

P = 4.4145 + 2.4417(-1.2) + 7.5(-0.35)² = 4.4145 - 2.9300 + 0.9188 = + 2.403 N

P= 2.40 N -





GIVEN;

m = 10 ftg

G = 25°

RELEASE

FINO!

(a) A

(b) B

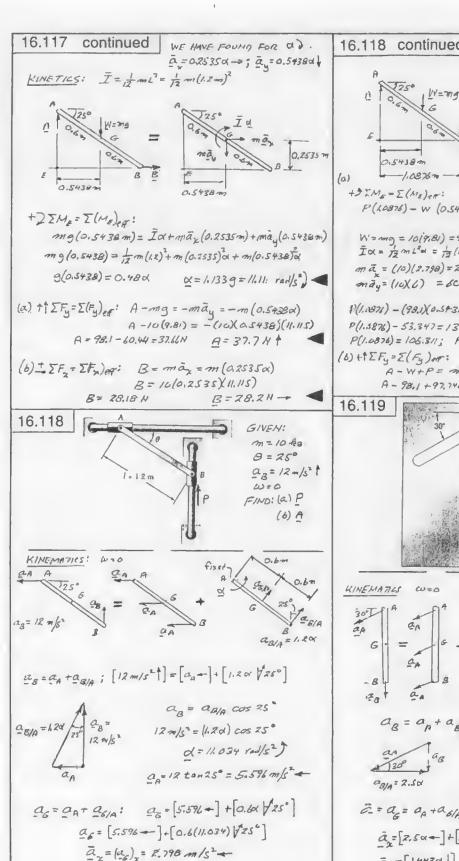
KINEMATICS: ASSUMG &)

an A Solar Color

an A S

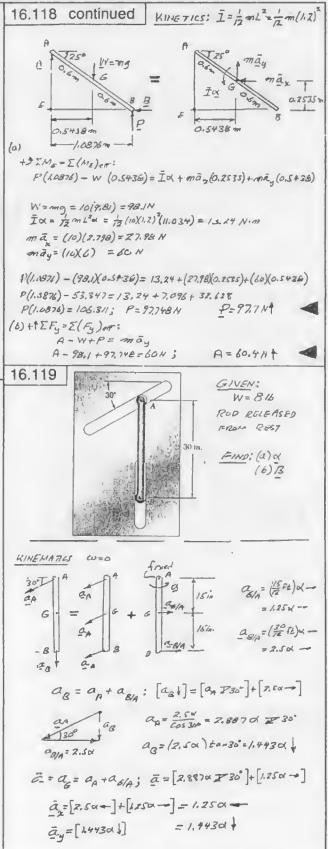
 $a_{B} = a_{A} + a_{C/A} = [a_{A} -] + [1.2 \times 1.25^{\circ}]$ $a_{B} = (1.2 \times) \cos 25^{\circ} - 1.08 \times 4$ $a_{B} = (1.2 \times) \sin 25^{\circ} = 0.507/4$

 $a_{6} = a_{A} + a_{8/A} = [a_{A} -] + [a_{6}\alpha A 25^{\circ}]$ $a_{6} = [0.507]\alpha -] + [0.6\alpha A 25^{\circ}]$ $\bar{a}_{2} = [a_{6}]_{2} = [0.507]\alpha -] + [0.2536\alpha -]$ $\bar{a}_{2} = 0.2535\alpha -$ $\bar{a}_{3} = [0.6\alpha \cos 25^{\circ} -] = 0.5438\alpha \downarrow$ (CONTINUED)

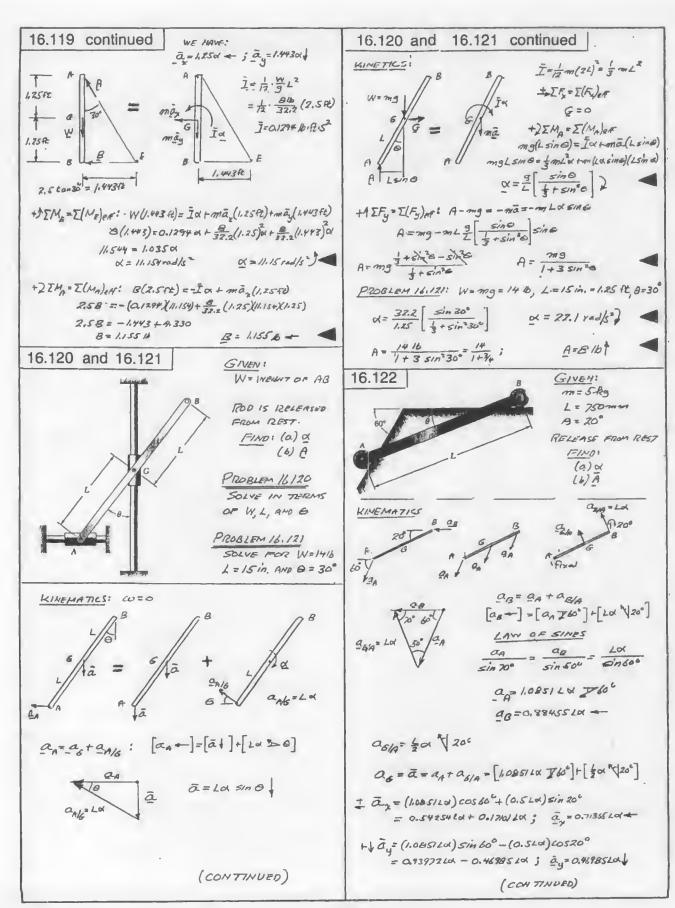


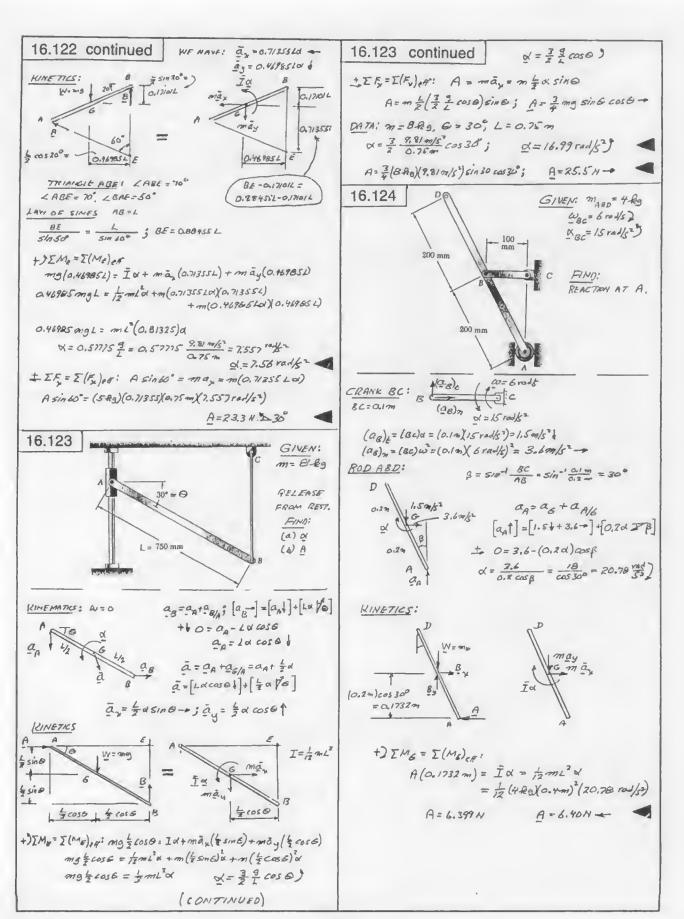
a = (a6) = 6.00 m/s2 +

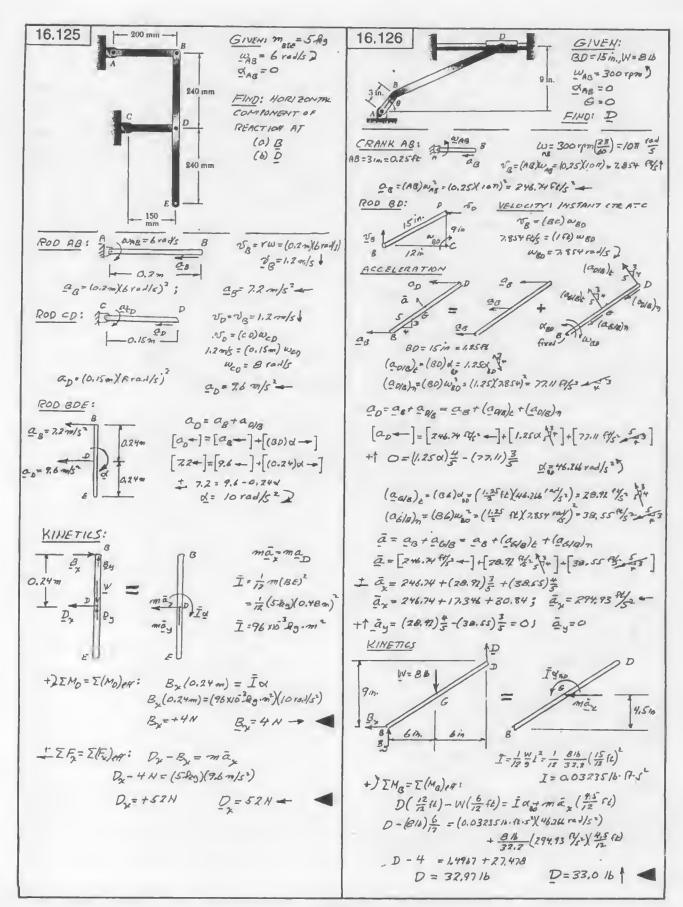
(CONTINUED)

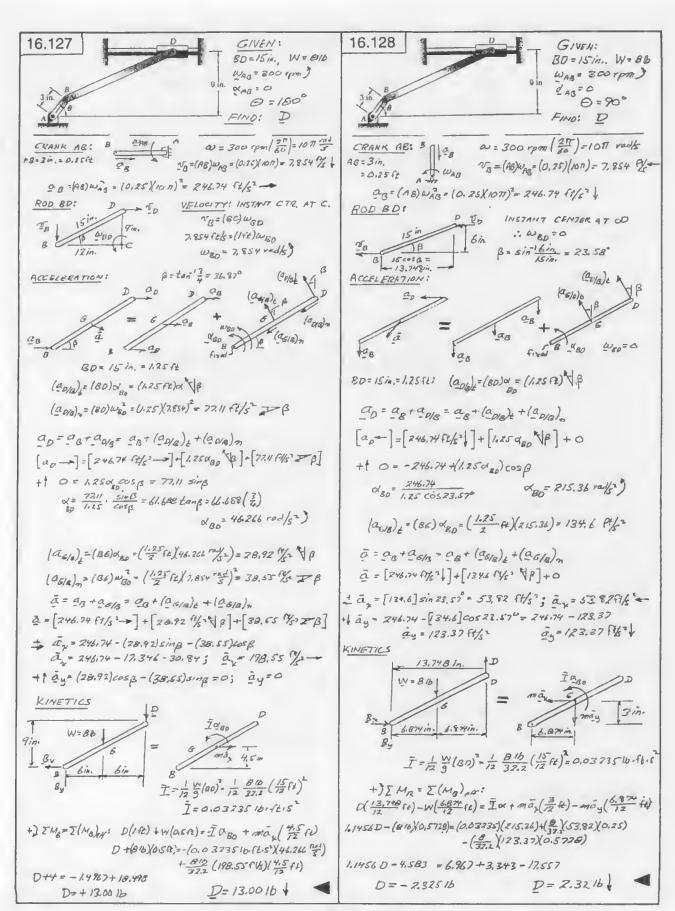


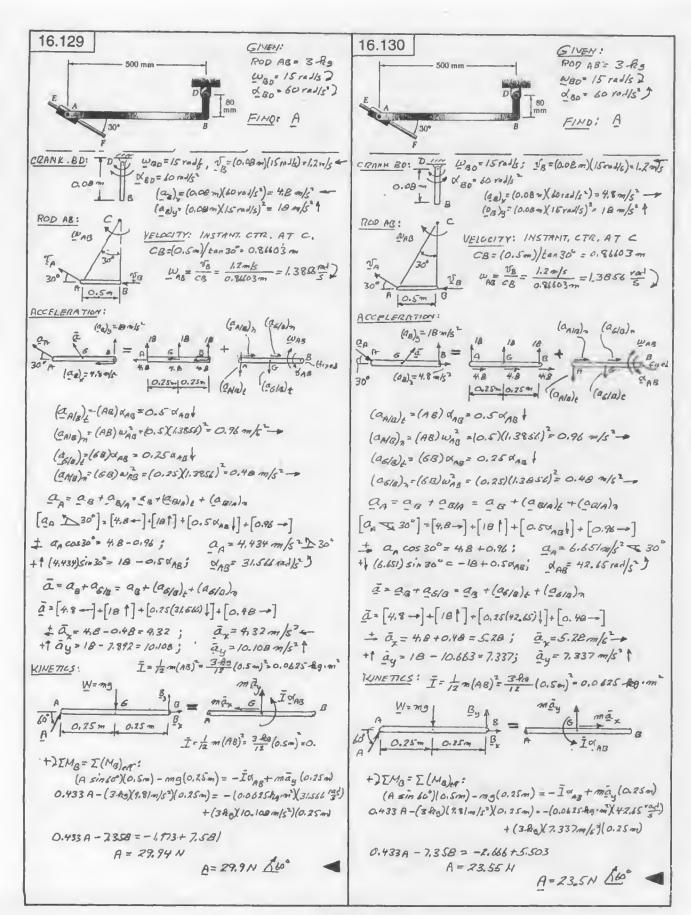
(CONTINUED)

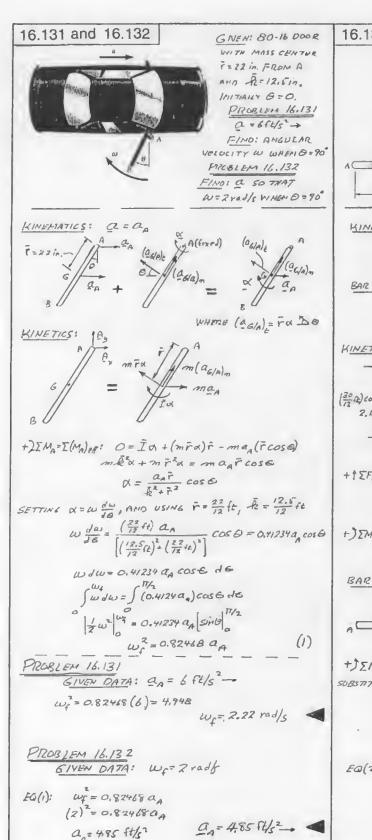


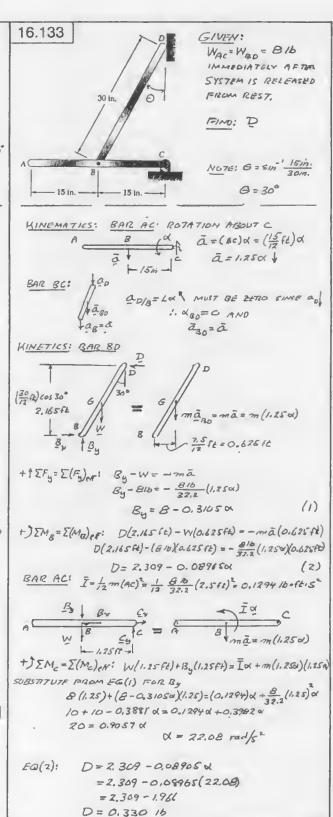




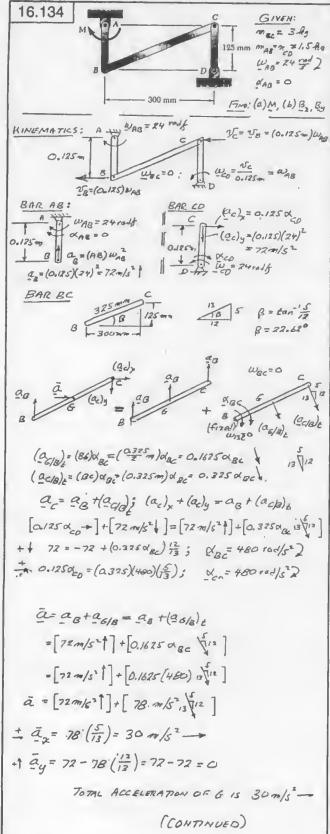


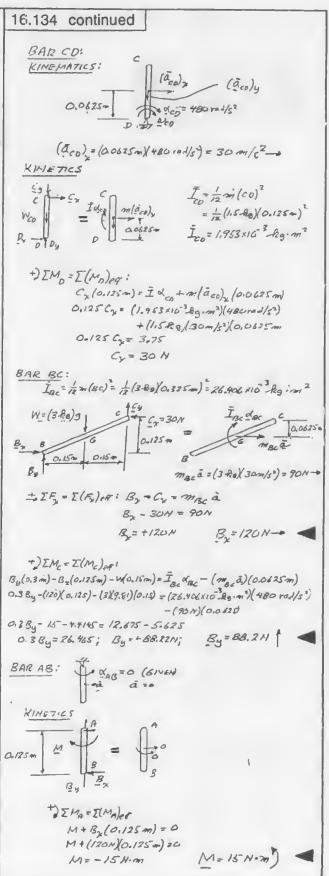


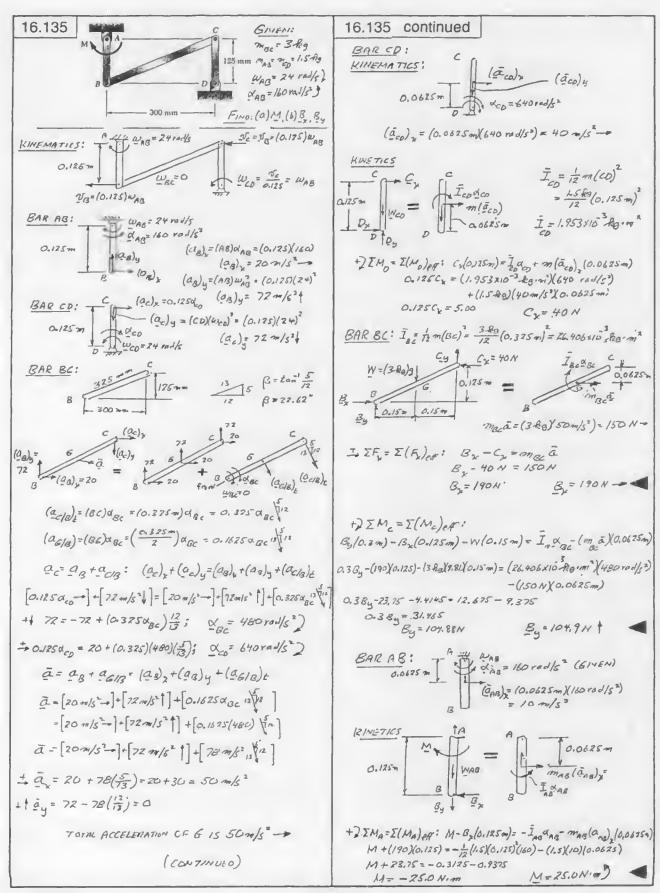


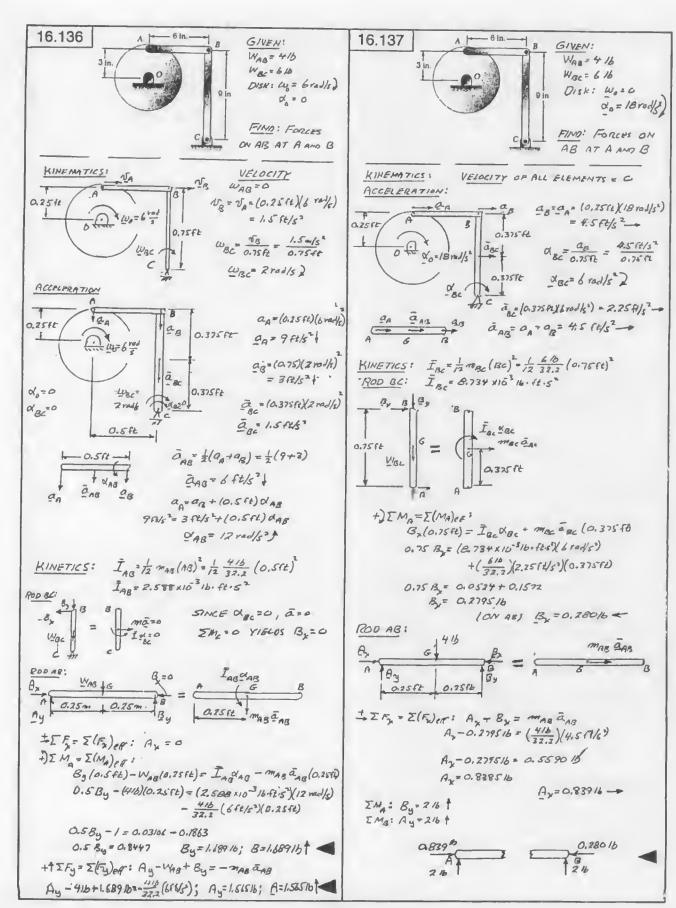


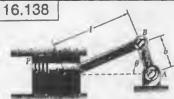
D=0.33016 +











GIVEN: WAS = 600 TP)

R= 250 mm, b= 100 mm

MBO = 1,2 lg, mp = 1,8 kg

0= 180°

FINDI FORCES ON

BD AT B AND D

KINEMATICS: CRANK AB:

QAB QANT-600 rpm (2.17) = 62.832 rad/s &

A = 0.1m B ag = (AB) WAN = (0.1m)(62.632 rad/s)²

Ag = 394.78 m/s²

ALSO: VB = (AB)WAB = (0.1 m)(62,822 rad/s) = 6.2832 m/s + CONNECTING ROD BD:

WED = 6.2832 = 15 25.133 rad/s)

ACCELERATION:

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}{D} = \frac{a_{B}}{B} + \frac{a_{D/S}}{D} = \frac{(fnel)}{D}$$

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} + \frac{a_{D/S}}{D} = \frac{(ge)}{D} = \frac{1}{2} \left(\frac{ge}{B}\right) = \frac{1}{2} \left(\frac{ge$$

KINETICS OF PISTON

$$\frac{p}{p} = \frac{p}{p} = \frac{m_p \, a_p = (1.8 \, \text{Rg})(236.86 \, \text{m/s}^2)}{D = 476.35 \, \text{N}}$$

FORCE EXERTED ON COMMECTING ROO AT D is:

KINETICS OF CONNECTING ROD: (NEGLECT WEIGHT)



IF, = I(F,) ef :

B - D = map and

B - 426.35N = (1.2.Rg)(3)5.82 m/s')

B = 426.85N + 378.98N = 805.33N

FORCES EXERTED ON CONNECTING ROD

B=805N
D=426N ->

16.139

GIVEN: WAG=600 rpm)

1=250 mm, b=150 mm

mgo=1.2 flag, mp=1.8 flag

B=0

ENDI FORCES ON

BR AT BAND D

KINEMATICS: CIZANK AB:

QB WAB WAB = 600 rpm, (21T) = 62.832 val/s)

B -0.1m A QB = (AB) WAB = (0.1 m) (62.832 val/s)

QB = 394.78 m/s² -

ALSO: Ng= (AB) WAB = (0.1 m) (62,882 red/s) = 6.2832 m/s \$
CONNECTING ROO BD:

DELOCITY INSTANT CENTER AT D: WED WED

ACCELERATION

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}{D} = \frac{a_{O/B}}{B} + \frac{a_{O/B}}{D} + \frac{a_{O/B}}{D} + \frac{a_{O/B}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}{B} + \frac{a_{O/B}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}$$

KINETICS OF PISTON

FORCE EXERTED ON COMMECTING ROD AT D is:

D=994.86N -

KINETICS OF CONNECTING ROD (NEGLECT WEIGHT)

$$D = 994.86N$$

$$B = D$$

$$D = D$$

$$E = D$$

$$B = D$$

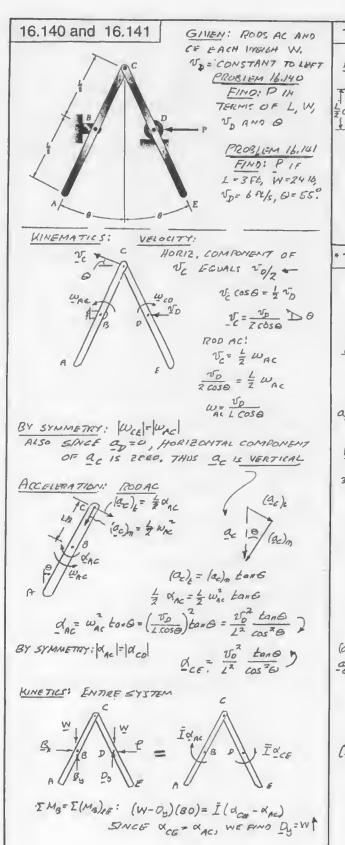
$$B = D$$

$$B = 0$$

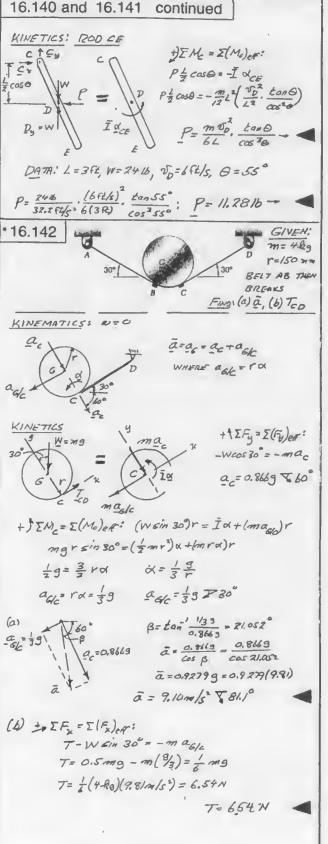
$$B = 994.86N = (1.240)(473.74m/s^{2})$$

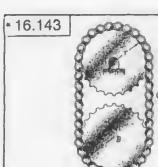
$$B = 994.86N + 568.44N = 1583.3N$$

FORCES ACTING ON CONNECTING ROD



(CONTINUED)





GIVEN:

DISK OF MASS ON AND RADIUS Y PILL AT C IS REMOVED FIND:

(a) da ano da

(b) TENSION IN CHAIN

(a) aB

KINGMATICSI WA=WQ=0 ASSUME DA) AND DB]



$$a_{B} = rd_{A} + a_{B} = a_{B} = rd_{A} + a_{B} = (rd_{A} + rd_{B}) + a_{B} = r(d_{A} + d_{B}) + a_{$$

KINETICS: DISKA:

+)
$$EM_{A} = E(M_{A})_{eff}$$
:
 $Tr = \overline{L}d_{A}$
 $Tr = \frac{1}{2}mr^{2}d_{A}$
 $d_{A} = \frac{2T}{2}$ (1)

DISK B:

$$\begin{array}{c}
T \\
E \\
E
\end{array}$$

$$\begin{array}{c}
T \\
E
\end{array}$$

+)
$$\Sigma M_2 = \Sigma (M_B) e R^2$$

 $Tr = \bar{I} \propto_B$
 $Tr = \frac{1}{2} m r^2 d_B$
 $d_B = \frac{2T}{mr} 2$ (2)

From (1) AND (2) WE NOTE THAT $d_A = d_B$ +) $EM_E = E(M_E)_{\Phi}F$: $Wr = \tilde{I}\alpha_S + (m\tilde{\alpha}_S)r$ $Wr = \frac{1}{2}mrd_S + mr(d_A + d_S)r$

d_B = = = = = = 1

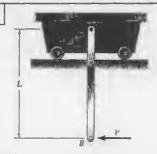
SUBSTITUTE FOR dA INTO (1):

$$\frac{2}{5}\frac{3}{r} = \frac{27}{24r}$$

$$a_{g} = r(a_{A} + a_{g}) = r(2a_{A}) = 2r(\frac{2}{5}\frac{9}{7})$$

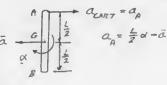
$$a_{g} = \frac{4}{5}9$$

*16.144



CART OF MASS M ROD OF MASS M CART AT REST WHEN P IS APPLIED FIND: QA QB

KINEMATICS:

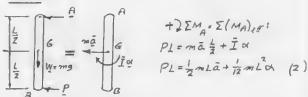


KINETICS: CART

$$T = m_{A} = m(\frac{1}{2}d - \bar{a})$$

$$T = \sum_{k=1}^{n} \sum_{k=1}^{n} (F_{k})_{pq} = A = m(\frac{1}{2}d - \bar{a}) \qquad (1)$$

ROD AB:



 $+ \sum_{k=1}^{n} \sum_{k=1}^{n} \sum_{k=1}^{n} P + A = m\bar{a}$ From (1) $P + m(\frac{1}{2}\alpha - \bar{a}) = m\bar{a}$

 $P = 2m\bar{a} - \frac{1}{2}mL\bar{a} \tag{3}$

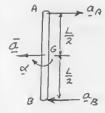
 $\frac{1}{4} \frac{1}{12} \frac{$

 $\bar{a} = \frac{7}{5} \frac{P}{m} +$

SUBSTITUTE (5) IN 70 (3):

$$P = 2m\left(\frac{7}{5}\frac{P}{m}\right) - \frac{1}{2}mL\alpha$$

$$P = \frac{14}{5}P - \frac{1}{2}mLd$$
 $Q = \frac{18}{5}P - \frac{1}{2}mLd$



$$1 \quad a_{A} = \frac{1}{2} \times -\bar{a}$$

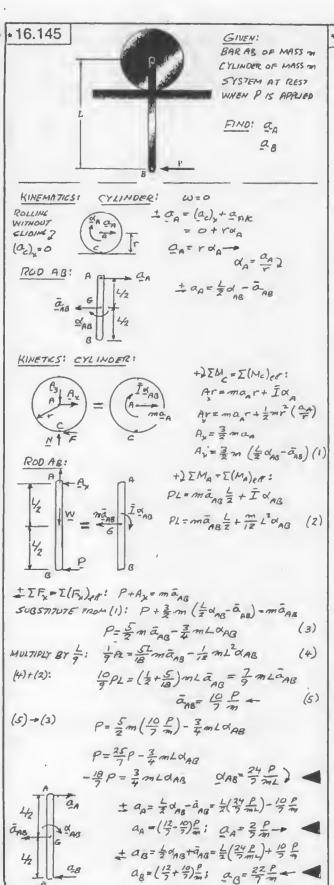
$$= \frac{1}{2} \left(\frac{18}{5} \frac{P}{mL} \right) - \frac{7}{5} \frac{P}{m}$$

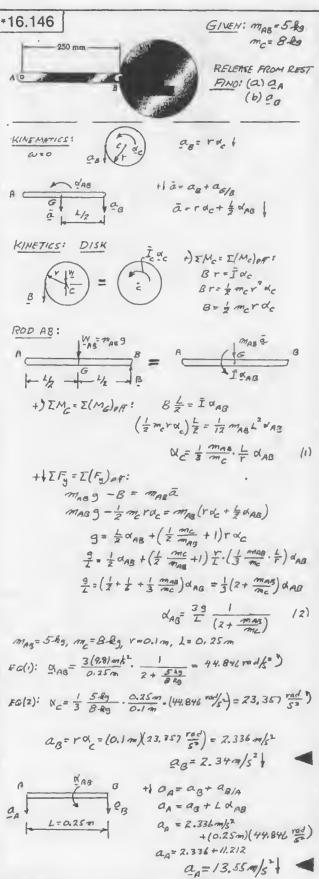
$$a_{A} = \frac{2}{5} \frac{P}{m}$$

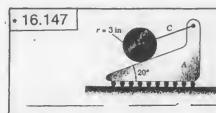
$$= \frac{1}{2} \left(\frac{1}{5} \cdot \frac{\rho}{mL} \right) + \frac{7}{5} \cdot \frac{\rho}{m}$$

$$= \frac{1}{2} \left(\frac{1}{5} \cdot \frac{\rho}{mL} \right) + \frac{7}{5} \cdot \frac{\rho}{m}$$

$$\alpha_R = \frac{16}{5} \cdot \frac{\rho}{m}$$







(4) X (4) X (5) X (6) X (7) X

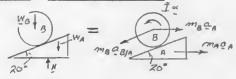
KINEMATICS! WE RESOLVE QB 11.76 QA AND A COMPONENT

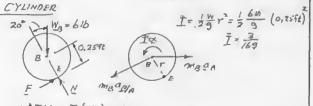


CYLINDER FOLIS OF WEDLE A.

ABJA = (0.25 ft) &

KINETICS: CYLINDER AND INFOLE





+) [M= [(ME)ex: (616) sin 20° (0.2512) = I ox + (MgO Bla 0.25ft) - mgOa cos 20 (0.25ft)

1.5 sin 20° = 3/16/323) A + 6/16/250) (0.25) - 6/16/32,2 A cas 20° (0.25)

0.51303 = 0.00582x + 0.01165x - 0.04378 an

SUBSTITUTE FROM (1):

0.51303 = 0.01747 x - 0.04378 (0.15 cas 20) ol

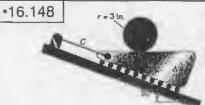
0,51303 = (0,01747 -0,00617) d

01 = 45.41 rad/s2

d = 45.4 124/5")

Ea(1): $a_{p} = (0.15\cos 20^{\circ}) d$ = $(0.15\cos 20^{\circ})(45.41)^{\circ}$ $a_{p} = 6.401 \text{ ft/s}^{\circ}$

a = 6.40 st/s ->



CIVEN: WB = 6 15

WA = 4 16

AFTER CORD IS CUT

CYLINDER ROLLS

FINO: (a) AA

111 A

KINEMATICS: WE RESOLVE an LITO AA AND A HORI-

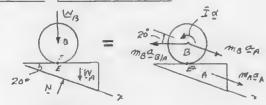
QUA B QA

NHERE CARA THE CYLHOER B ROLLS ON WEDGE A.

By = 6.2512)

ag= ag + asia

KINETICS: CYLINDER AND WEOGE:



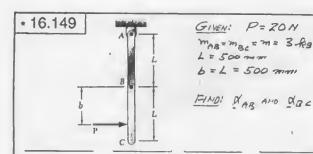
 $\begin{array}{l}
+ \sum F_{\mu} = \sum (F_{\mu})_{\mu} P^{2} \\
(W_{A} + W_{B}) \sin 20^{\circ} = (m_{A} + m_{B}) \Delta_{A} - m_{B} \Delta_{B/A} \cos 20^{\circ} \\
(10.16) \sin 20^{\circ} = (\frac{10}{3}) \Delta_{A} - (\frac{6}{3})(0.25 \alpha) \cos 20^{\circ} \\
\Delta_{A} = g \sin 20^{\circ} + \frac{6}{10}(0.25) \cos 20^{\circ} d \\
\Delta_{A} = g \sin 20^{\circ} + 0.15 \cos 20^{\circ} d
\end{array} \tag{1}$

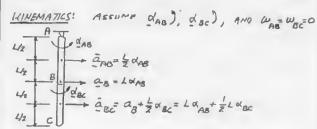
 $(YLINDER: +) IM_b = E(ME)ex$ $O = \overline{I} \propto + (m_B \alpha_{B/A})(0.254t) - (m_B \alpha_A \cos 2t)^2 (0.254t)$ $O = \frac{1}{2} \frac{6/6}{3} (0.254t) + \frac{6B}{3} (0.25a)(0.25) - \frac{6}{9} \alpha_A \cos 2t^2 (0.25)$ $O = \frac{1}{3} \left[0.1875 + 0.325 + 0.4995 \alpha_A \right]$ $O = 0.5625 - 1.4095 \alpha_A; \quad d = 2.506 \alpha_A \quad (2)$

SUBSTITUTE FOR & FROM (2) INTO (1): $a_{A} = g \sin 20^{\circ} + 0.15 \cos 20^{\circ} (2.506 a_{A})$ $a_{A} = 11.013 + 0.3532 a_{A}$ $(1-0.3532) a_{A} = 11.013$

 $a_A = 17.027 \text{ ft/s}^2$ $a_A = 17.027 \text{ ft/s}^2 = 20$

FO(2) $\forall = 2.506 \, O_{R}$ = 2.506 (17.627) $\forall = 42.7 \, nod/6^{2}$ $(= 42.7 \, nod/6^{2})$





$$\frac{1}{2} \sum F_{\chi} = \sum (F_{\chi})_{\mathcal{A}}; \qquad P - B_{\chi} = m \tilde{a}_{Bc}$$

$$P - B_{\chi} = m \left(L \alpha_{AB} + \frac{1}{2} L \alpha_{BC} \right) \qquad (2)$$

ADD (2) AND(2):
$$P = \frac{4}{3} m L d_{AB} + \frac{1}{2} m L d_{BC}$$
 (4)
SUBTRACT (1) PROM (4)

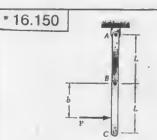
$$AB = -\frac{6}{7} \frac{P}{mL} \qquad (6)$$

$$AB = -5\left(-\frac{4}{7} \frac{P}{mL}\right) \qquad AB = \frac{20}{7} \frac{P}{mL} \qquad (7)$$

DATA: L=0.5m, m=3kg, P=204

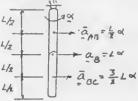
EG(6):
$$\alpha_{AB} = -\frac{4}{7} \frac{20 \text{ N}}{(3 \text{ Re})(0.5 \text{ m})} = -11.43 \text{ rod/s}^2$$

$$\alpha_{AB} = 11.43 \text{ rad/s}^2$$



GIVEN: P=20N MAB=MBC=M=3R9 L=500mm

FIND: (a) DISTANCE B FOR WHICH BAILS MOVE AS A SINGLE RIGHT BOOY (b) & OF BARS



KINETICS: BARS AB AND BC (ACTING AS RIGIN BODY)

$$P = \frac{A_{S} \wedge A_{A}}{\sum_{i=1}^{N} A_{S}} = \frac{A_{ABC}}{\sum_{i=1}^{N} A_{ABC}} = \frac{A_{ABC}}{\sum_{i=1}^{N} A_{ABC}$$

+)
$$\Sigma M_A = \Sigma (M_A)_{eA}$$
: $P(L+b) = \overline{L}_{ABC} \propto + m_{ABC} \alpha_B L$
 $P(L+b) = \frac{2}{3} mL^2 \alpha + (2m)(L\alpha)L$
 $P(L+b) = \frac{8}{3} mL^2 \alpha$ (1)

SUBSTITUTE FOR & INTO (1)

$$P(1+b) = \frac{8}{3}mL^{2}\left(\frac{b}{5}\frac{\rho b}{mL^{2}}\right)$$

$$PL+Pb=\frac{16}{5}Pb$$
 ; $L=\left(\frac{16}{5}-1\right)b=\frac{11}{5}b$

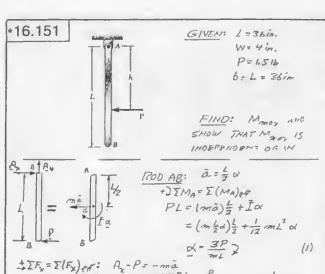
$$EQ(2)$$
 $\alpha = \frac{6}{5} \cdot \frac{P}{mL^2} \left(\frac{5}{11} L \right)$ $\alpha = \frac{6}{11} \cdot \frac{P}{mL}$

DATA! L= 0.5m, m = 3 kg, P= 20 N

(a)
$$b = \frac{5}{11} L = \frac{5}{11} (500 \text{ mm}); \quad b = 227 \text{ mm}$$

(6)
$$\forall = \frac{6}{11} \frac{P}{mL} = \frac{6}{11} \frac{ZON}{(3R_0 X o. 5m)} = 7.273 \text{ val} k^2$$

$$\forall = 7.27 \frac{\text{val}}{\text{s}^2}$$



PORTION AS OF ROD:

EXTERNAL FORCES: A. W.N., AXIAL FORCE F.,

SHEAR V.J., AND BENDING MUMENT M.

EFFECTIVE FORCES! SINCE ACCELERATION AT MY

POINT IS PROPORTIONAL TO DISTANCE FROM A. EFFECTAR

FORCES ARE UNEARLY DISTRIBUTED. SINCE WASS PER

UNIT LENGTH IS MIL, AT POINT V WE FIND

Ax= P-m= d= P-m= (3P)=-P: Ax=1P=-

$$A = \frac{A}{2}$$

+)
$$ZM_{J} = \Sigma(M_{J})_{PH}! M_{J} - A_{\gamma} \chi = -\frac{1}{2} \left(\frac{3P\chi}{L^{2}} \right) \chi \left(\frac{2\gamma}{\epsilon} \right)$$

$$M_{J} = \frac{1}{2} P\chi - \frac{1}{2} \frac{P}{L^{2}} \chi^{3} \qquad (2)$$

For
$$M_{max}$$
:
$$\frac{dM_{J}}{d\chi} = \frac{P}{Z} - \frac{3}{2} \frac{P}{L^{3}} \chi^{2} = 0$$

$$\chi = \frac{L}{\sqrt{3}}$$
 (3)

SUBSTITUTING INTG (2)

$$(M_{J})_{ma_{\chi}} = \frac{1}{2} \frac{PL}{V_{3}} - \frac{1}{2} \frac{P}{L^{2}} \left(\frac{L}{V_{3}}\right)^{3} = \frac{1}{2} \frac{PL}{V_{3}} \left(\frac{2}{3}\right)$$

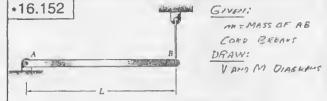
$$(M_{J})_{ma_{\chi}} = \frac{PL}{3V_{3}}$$

$$(4)$$

NOTE: EOS. (3) AND (4) ARE INDEPENDENT OF W

$$EG(3)$$
: $\chi = \frac{L}{V_3} = \frac{36 \, in.}{V_3} = 20.28 \, in.$

Mmax = 10.37 16-in. 20.8 in. BELOWI A.



FROM ANSWELS TO PRINTS 16.84: $a_{B} = \frac{3}{2}9$ $A = \frac{1}{4}m91$ $A = \frac{3}{4}$ $A = \frac{3}{4}$

PORTION AJ OF RODS

EXTERNAL FORCES: REACTION A, DISTRIBUTED LOAD
PER UNIT LENGTH M3/1, SHEAR VI, BENDING MEMBERT MJ.

EFFECTIVE FORCES: SINCE a ~ 2, THE EFFECTIVE
FORCES ARE LINEARLY DISTRIBUTED. THE EFFECTIVE
FORCE PER UNIT LENGTH AT J IS:

$$\frac{2n}{L} \alpha_{J} = \frac{m}{L} \cdot \frac{39}{2L} \chi = \frac{3mg}{2L^{2}} \chi$$

$$A = \frac{mg}{4} \qquad \qquad \chi \qquad \qquad M_{J} \qquad \qquad \chi \qquad \qquad M_{J} \qquad \qquad \chi \qquad \qquad M_{J} \qquad M$$

 $\begin{aligned} & + \sqrt{2} F_{g} = \sum \left(F_{g} \right)_{eff} : \frac{m \, 9}{L} \, \chi - \frac{m \, 9}{4} + V_{J} = \frac{1}{2} \left(\frac{3 \, m \, 9}{2 \, L^{2}} \, \chi \right) \, \chi \\ & V_{J} = \frac{m \, 9}{4} \, - \frac{m \, 9}{L} \, \chi + \frac{3}{4} \, \frac{m \, 9}{L^{2}} \, \chi^{2} \\ & + \mathcal{I} \sum M_{J} = \sum \left(M_{J} \right)_{eff} : \left(\frac{m \, 9}{L} \, \chi \right) \frac{\chi}{2} - \frac{m \, 9}{4} \, \chi + M_{J} = \frac{1}{2} \left(\frac{3 \, m \, 9}{2 \, L^{2}} \, \chi \right) \, \chi \left(\frac{\nu}{3} \right) \\ & M_{J} = \frac{m \, 9}{4} \, \chi - \frac{1}{2} \, \frac{m \, 9}{L^{2}} \, \chi^{2} + \frac{1}{4} \, \frac{m \, 9}{L^{2}} \, \chi^{3} \end{aligned}$

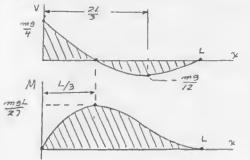
FINO
$$V_{min} = \frac{dV_{i}}{dx} = -\frac{mg}{4} + \frac{3}{2} \frac{mg}{L^{2}} \times = 0 \quad ; \quad \chi = \frac{7}{3} L$$

$$V_{min} = \frac{mg}{4} - \frac{mg}{L} \left(\frac{2}{3}L\right) + \frac{3}{4} \frac{mg}{L^{2}} \left(\frac{2}{3}L\right)^{2}; \quad V_{noin} = -\frac{mg}{12}$$

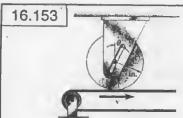
FIND M_{max} WHERE $V_j = 0$: $V_j = \frac{mg}{4} - \frac{mg}{L}x + \frac{3}{4}\frac{mg}{L}x^2 = 0$ $(3x^2 - 4Lx + L^2 = 0)$ (3x - L)(x - L) = 0 $x = \frac{L}{3}$ AND x = L

$$M_{may} = \frac{mq}{4} \left(\frac{L}{3} \right) - \frac{1}{2} \frac{mq}{L} \left(\frac{L}{3} \right)^2 + \frac{1}{4} \left(\frac{mq}{L} \right)^2 \frac{1}{3} = \frac{mq}{27}$$

$$M_{min} = \frac{mq}{4} L - \frac{1}{2} \frac{mq}{L} L^2 + \frac{1}{4} \frac{m^2}{12} L^3 = 0$$



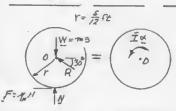
May 37 AT 1 FROM A



GIVEN: ©= 30° M_k= 0.20

FIND: & WINILE

SLIPPING OCCURS



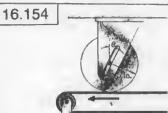
 $\begin{array}{l} \stackrel{+}{\rightarrow} \Sigma F_{\mu} = \Sigma (F_{\mu}) e \gamma \\ M_{\mu} N - R \cos \Theta = 0 \\ R \cos \Theta = M_{\mu} N & (I) \\ + \stackrel{+}{\uparrow} \Sigma F_{\mu} = Z (F_{\mu})_{eN} \\ R \sin \Theta + N - mg = 0 \\ R \sin \Theta = mg - N & (2) \end{array}$

0.1155N= mg-N; N= mg-N ; 0.5774= mg-N 0.1155N= mg-N; N= 1.1155 = 0.8965 mg

+) [M = [(M))eq: -1/2 H =] a (0.2)(0.8985mg) r = 1/2 mr2 a

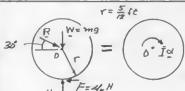
0 = 0.35858 \frac{9}{r} = 0.35858 \frac{32.2545}{(5/m ft)} = 27.71 \frac{10-1/2}{51}

X = 27.7 rad/53)



GIVEN: G = 30° Mx = 0.20

FINDI & WHILE SLIPPING OCCURS



 $\begin{array}{l}
\pm \sum F_y = \sum (F_x)_{MF}; \\
R\cos\theta - M_y N = 0 \\
R\cos\theta = -M_y N \quad (1) \\
\pm \sum F_y = \sum (F_y)_{GF}; \\
R\sin\theta + mg - N = 0 \\
R\sin\theta = N - mg \quad (2)
\end{array}$

DIVIDE (2) 87(1)1 Land = N-m9 : 0.5174 = N-m9

0.1155N= N-mg: N= mg = 1.1306 mg

+) IMo = E(Mo)eff: MENT = IX.

(0.2 X1.1306mg) v = 1 mr a

X = 0.4522 \frac{9}{r} = 0.4522 \frac{32.7(4/52)}{(5/12 ft)}

0 = 34.948 rad/s2

X=34.9 rad/s2)

16.155



GIVEN: CYLIMPERS
FIND: (a) MAXIMUM a
FOR ROLLIM WITH
IYO SLIDING
(b) MINIMUM a

FOR CYLINDER TO MOVE - WITH NO ROTHTING

(a) CYLINDER ROLLS WITHOUT SLIDING: a=Fa or Q= 2

$$B_{y} = HP$$

$$B_{y} = P$$

PIS HORZ.

COM POWENT OF

FORCE ARM

EXERTS ON CYLINGR

+) $\sum M_{\rho} = \sum (M_{\rho})_{e}q$: $Pr - (N_{\kappa}P)_{r} = \sum \alpha + (m\bar{\alpha})_{r}$ $P(1-4)_{r} = \sum mr(\frac{\bar{\alpha}}{r}) + (m\bar{\alpha})_{r}$

 $\rho = \frac{3}{2} \frac{m\ddot{a}}{(I-4')} \tag{1}$

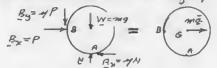
SOLVE (2) FOR N AND SUBSTITUTE FOR N INTO (2). P-. 42P-4mg=ma

SUBSTITUTE P FROM (1): (1-42) \frac{3}{2} \frac{m\alpha}{(1-4)} - 4 mg = m\alpha
3(1+4)\alpha - 2 4 g = 2\alpha

ā(1+34)-243=0 ā= 24.

(b) CYLINDER TRAVISLATES: X=0

ASSUME SCIOING IMPENIX AT A: By= of P



+) $EM_A = E(M_A) \cdot R$: Pr - gPr = (ma)rP(1-q)r = mar

 $P = \frac{m\bar{\alpha}}{1-4} \tag{4}$ $-4N = m\bar{\alpha} \tag{5}$

(4)

+ EF= E(Fx)ex: P-yN=ma + 1 EFy= E(Fy)ex: N-yP-mg=0

Solve (5) FOR N AND SUBSTITUTE FOR N INTO (6).
P-Py2-4mg=ma

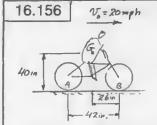
SUBSTITUTE FOR P FROM (4):

a(1+4)-49=a

a4-49 30

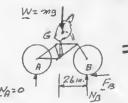
SUMMARY: a < 24 9; ROLLING

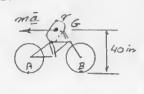
1+34 g < a < 9 : ROTATING AND GLIOING



FIND I SHORTEST STOPPING DISTANCE IF CYCLIST IS NOT TO BE THROWN OVER FRONT, WHEEL

WHEN CYCLIST IS ABOUT TO BE THROWN OVER THE FRONT WHEEL, HA = 0



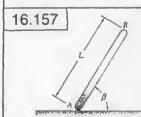


+) $IM_{a} = I(M_{s})_{eff}$: $mg(26in) = m\bar{a}(40in)$ $a = \frac{26}{40}g = \frac{76}{46}(32.28t/s^{2}) = 20.93 ft/s^{2}$

UNIFORMLY ACCEGERATED MOTION:

76 = 20 mph = 29.333 Ft/s2 52-16 = 2as: 0-(29.333 Ft/s) = 2(-20.93/3)5

S=20.555 12 S=20.612



GIVENI Q=70° UIVIFORM ROD

RELEASED FROM REST.

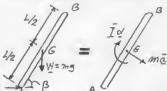
FRICTION IS EUFFICIENT

TO PREVEND SLIPNS AT A.

FIND: (a) OX

(b) NA, (c) FA

WE NOTE ROO POTATES ABOUT A. W=0



 β +) $\Sigma M_A = \Sigma (M_A) e \beta$: $mg(\frac{1}{2} \cos \beta) = \tilde{I} \propto + (m\tilde{a}) \frac{1}{2}$ $\frac{1}{2} mg L \cos \beta = \frac{1}{12} mL^2 \alpha + (m\frac{1}{2} \alpha) \frac{1}{2}$

 $= \frac{1}{3}mL^{2}\alpha$ $\alpha = \frac{3}{2}\frac{9\cos\beta}{1} \qquad (1)$

I = 1/2 ml2

 $\frac{d}{d} = \frac{1}{2} \frac{1}{L} \qquad (1)$ $\frac{1}{L} \sum F_{2} = \sum (F_{2})_{eq} : \quad F_{a} = m \vec{a} \sin \beta$ $F_{a} = m \frac{1}{2} d \sin \beta = m \frac{1}{2} \left(\frac{3}{2} \frac{g \cos \beta}{L} \right) \sin \beta$ $F_{a} = \frac{3}{4} mg \sin \beta \cos \beta \qquad (2)$

(CONTINUED)

16.157 continued

 $+ \uparrow I F_y = I(F_y) \cdot A : N_{-mg} = -m \bar{a} \cos \beta = -m \left(\frac{1}{2}\alpha\right) \cos \beta$ $N_{-mg} = -m \frac{1}{2} \left(\frac{3}{2} \frac{g \cos \beta}{L}\right) \cos \beta$

 $N_{\beta} = mg(1 - \frac{3}{4}\cos^2\beta)$ (3)

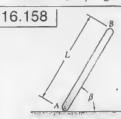
FOR β = 70°;

(a) Eq.(i): α = \(\frac{3}{2} \frac{9 \cos 20°}{2} \frac{1}{2} \frac{1} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \f

X=0.513 =)

(b) EG(3): NA=mg(1-3/200370); NA=0.912mg+

(c) Eq.(2) F= 3 mg sin70 cos 10; F= 0,241 ag-



GIVEN: B=70°. WHEARM
ROD RELEASED FROM REST.
FRICTION AT SURFACE
EQUALS ZERO,
FIND: (a) K, (b) ā,
(c) REACTION AT A.

+ $\Sigma F_y = \Sigma (F_y)_{e,q}$: $mg - A = m\bar{\alpha}_y = m(\frac{L}{2} \alpha \cos \beta)$ (1) $\Sigma M_G = \Sigma (M_G)_{e,q}$:

 $A\left(\frac{1}{2}\cos\beta\right) = \bar{I}\alpha = \frac{1}{12}mL^{2}\alpha$ $A = \frac{mL}{6}\frac{\alpha}{\cos\beta}$ (2)

SUBSTITUTE (2) INTO (1): $mg - \frac{mL}{6} \frac{\alpha}{\cos \beta} = m \frac{L}{2} \alpha \cos \beta$

$$g = \left(\frac{L}{2}\cos\beta + \frac{L}{6\cos\beta}\right) \propto$$

$$g = \frac{L}{6} \left(\frac{3\cos^2\beta + 1}{\cos\beta} \right) \propto$$

 $\alpha = \frac{69}{L} \left(\frac{\cos \beta}{1 + 3\cos^2 \beta} \right)$

 $\frac{a}{2} = \frac{1}{2} \alpha \cos \beta = \frac{1}{2} \left(\frac{69}{2} \cdot \frac{\cos \beta}{1 + 3\cos^2 \beta} \right) \cos \beta = 39 \left(\frac{\cos^2 \beta}{1 + 3\cos^2 \beta} \right) = \frac{1}{2} \cos \beta$

A= ml. casp = ml. (69. cosp) 1 mg 1+3casp 1

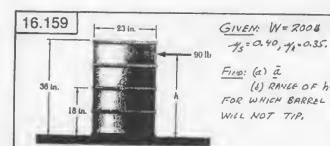
FOR B=70:

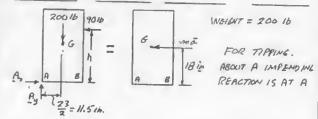
(a) $d = \frac{69}{4} \frac{\cos 26}{1 + 3\cos^2 70}$;

d=1.519 =)

(1) \(\bar{a} = 39 \) \(\frac{\cos^270}{1 + 3\cos^220}\); \(\bar{a} = 0.260\) g\

(a) A=mg 1 /+ 3cos = 700; A=0:740 mg t





$$\frac{+\sum_{k=1}^{n}\sum_{(k=1)}^{n}\sum_$$

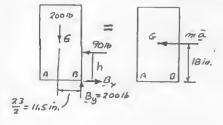
$$+ \int \sum M_{A} = \sum (M_{A})_{eA} + (9016)h - (20016)(\frac{11.5}{12}f_{c}) = m\bar{a}(\frac{16}{12}f_{c})$$

$$90h - 191.67 = \frac{70016}{32.2}(3.72f_{c})(1.5f_{c})$$

$$90h - 191.67 = 30; 90h = 221.67$$

$$h = 2.463f_{c} h = 29.6 in.$$

FOR TIPPING IMPENDING ABOUT B, REACTION IS AT B



$$B_{\chi} = \eta_{K} B_{g} = 0.35(200) = 7016$$

$$\stackrel{+}{=} \Sigma F_{\chi} = \Sigma (F_{\chi})_{eff} \quad SAME \quad AS \quad ABOVE: \quad \bar{a} = 3.22 \Gamma_{g}^{V} = 4$$

$$+) \Sigma M_{g} = \Sigma (M_{g})_{eff} : (90h) + (2001h) \frac{11.5}{12} F_{g} = m \bar{a} \left(\frac{18}{12} f_{g} \right)$$

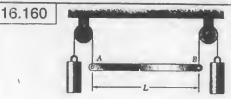
$$90h + 191.67 = \frac{20016}{32.2} \left(3.22 F_{g} \right) \times (1.5 f_{g})$$

$$90h + 191.67 = 30$$

$$90h = -161.67$$

h < 0 IMPOSSIBLE

THUS RANGE FOR NO TIPPING IS h < 29.6 in

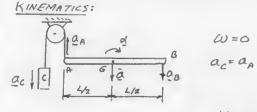


GIVEN: WELDTS
BAR AB: W
COUNTERWEIDTS
= \frac{1}{2}W.
IMMEDIATELY

AT BIS CUT.

(2)

FINO: (a) ap, (b) 03.



KINETICS: COUNTERWEIGHT M= MASS OF BAR AB

$$\frac{1}{2} \underbrace{W} = \begin{bmatrix}
1 & 1 & 1 \\
1 & 1 & 1 \\
1 & 1 & 1
\end{bmatrix}$$

$$\frac{1}{2} \underbrace{W} = T = \underbrace{M_{c} a_{c}}_{c} = \underbrace{\frac{1}{2} m a_{A}}_{c}$$

$$\frac{1}{2} m_{g} = T = \underbrace{\frac{1}{2} m a_{A}}_{c}$$

$$T = \underbrace{\frac{1}{2} m (g - a_{A})}_{c} \qquad (1)$$

KINETICS BAR AB

$$T \downarrow L/2 \qquad G \qquad B = A \qquad L/2 \qquad B$$

$$+ \downarrow IF_y = I(F_y)_{ef} : mg - T = m\bar{a}$$

$$mg - \frac{1}{2}m(g - a_A) = m(\frac{1}{2}Ld - a_A)$$

29-9+a= L= -za, 9+3a= La

+)
$$\Sigma M_6 = \Sigma (M_6)_{pp}$$
: $T_{2}^{1/2} = \overline{I}_{2}^{1/2}$

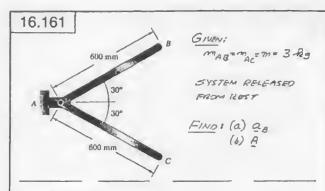
$$= \frac{1}{12} mL^{2} \alpha$$

$$39 - 3a_{p} = L \alpha$$
 (3)

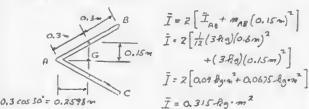
ADD EGS. (2) AND (3):
$$4g = 2L\alpha$$
 $\alpha = \frac{2g}{L}$

$$\bar{a} = (\frac{1}{2} L \alpha - a_n) = \frac{1}{2} L(\frac{29}{2}) - \frac{1}{5}9; \qquad \bar{a} = \frac{2}{3}9$$

$$a_{\beta} = (L \times -a_{\beta}) \cdot L \left(\frac{29}{L} - \frac{1}{3}9\right);$$
 $a_{\beta} = \frac{5}{3}9$

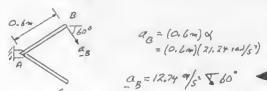


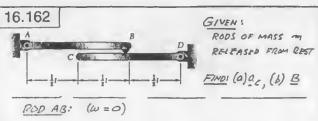
CENTER OF MASS AND I'

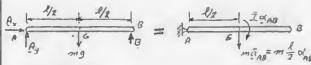


X = 21,24 rad/52] \$\alpha = 0.2598(21.24) = 5.518 \alpha \beta^2\$

SINCE A,=0, A=25.8 NT







+)
$$\sum M_{A} = \sum (M_{A})e^{A}e^{2}$$

 $mg(\frac{1}{2}) - 13 \cdot l = \sum \alpha + m \cdot \alpha_{AB}(\frac{1}{2})$
 $\frac{1}{2} mgl - B \cdot l = \frac{1}{2} m \cdot l^{2} \alpha + m \cdot (\frac{1}{2} \alpha_{AB}) \cdot \frac{1}{2}$
 $\frac{1}{2} mgl - B \cdot l = \frac{1}{3} m \cdot l^{2} \alpha_{AB}$ (1)

ROU CO: $(\omega = 0)$ $C = \frac{B}{\sqrt{2}} \frac{l/2}{D} = C = \frac{\overline{1} \times CD}{\overline{1} \times CD}$ $C = \frac{\overline{1} \times CD}{\overline{1} \times \overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times CD}{\overline{1} \times \overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$ $C = \frac{\overline{1} \times \overline{1}}{\overline{1} \times \overline{1}}$

+) $IM_D = \Sigma(M_0)eR$: $mg(\frac{1}{2}) + B(\frac{1}{2}) = \bar{1} \propto_0 + m \alpha_{cD} \frac{1}{2}$ $\frac{1}{2} mg \ell + \frac{1}{2} B\ell = \frac{1}{12} m\ell \propto_0 + m(\frac{1}{2} \alpha_{c0}) \frac{1}{2}$

MULTIPLY BY 2: $mgl + Bl = \frac{2}{3} ml \alpha_{c0} \qquad (2)$

ADD (1) AND (2):
$$\frac{3}{2}$$
 mg $\ell = m\ell^2 \left(\frac{1}{3} \alpha_{A3} + \frac{2}{3} \alpha_{CO} \right)$ (3)

MULTPLY BY 3: $\alpha_{AB} + 2\alpha_{CD} = \frac{9}{2} \frac{3}{D}$ (4)

KINEMATICS: WE MUST HAVE

$$A_{ab} = A_{ab} = A_$$

SURSTITUTE FUN (A_{AB}) FROM (S) INTO (4) $\frac{1}{2} \propto_{co} + 2 \propto_{co} = \frac{9}{2} \frac{9}{R}$ $\frac{5}{2} \propto_{co} = \frac{9}{2} \frac{3}{8} ; \qquad (x_{co} = 1.8) \frac{9}{8}$ (6)

(a) ACCELERATION OF C: ac=loco=l(1.8 \frac{3}{e}); ac=1.89\

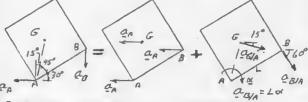
(b) FORCE ON KNOB B: SURSTITUTE FOR d_{CD} FROM (6) IN TO (2) $mgl + Bl = \frac{2}{3} ml^2 (1.8 \frac{9}{8})$ B = 1.2 mg - mg(ON ROO AB): B = 0.2 mg 16.163



GIVEN: SQUARE PLATE
OF SIDE L= 150 0000
AND M= 2.5 Rg IS
RELEASED FROM REST

FIND (a) & (b) A

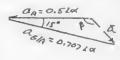
KINEMATICS:



PLINE MOTION = TRAISLATION + ROTATION

an = 1 of sin 30"

an = 1 of cos 30"



1AW OF COSINES

\$\bar{a}^2 = a_A^2 + a_{6/A}^2 - 2a_A a_{6/A} \cos /5' \\
\$\bar{a}^2 = (0.5 \text{Let})^2 + (0.707 \text{Let})^2 \\
-2(0.5 \text{Let})(0.707 \text{Let}) \cos /5' \\
\$\bar{a}^2 = \text{Let}(0.25 + 0.5 - 0.6830)

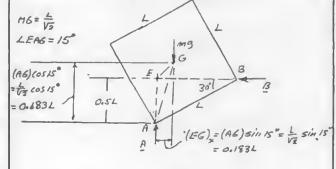
a=12 (0.06699); a=0.258821d

LAW OF SINES

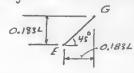
 $\frac{\bar{\alpha}}{\sin s} = \frac{\alpha_{SIA}}{\sin \beta}; \sin \beta = \frac{\alpha_{SIA}}{\bar{\alpha}} \sin \beta = \frac{0.7071 \times 10^{-10}}{0.2595210} \sin \beta$ $\sin \beta = 0.707; \beta = 135$ $\bar{\alpha} = 0.2588 Ld \ \ \ \ 45^{\circ}$

KINETICS (W=0)

OF ACTION OF A AND B INTERSECT.



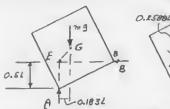
(EG) = 0.6829L - 0.5L = 0.183L



E6=(0.1834) V2 = 0.25884

(CONTINUED)

16.163 continued



J = 1 m L²
0.2588L
Jd

E - 7 45°
45°
m2

+) $[M_A = \Sigma(M_A)_{\theta\theta}: mg(0.183L) = \hat{I}\alpha + (ma)(0.2588L)$ 0.183 mg $L = \frac{1}{2}mL^2\alpha + m(0.2586La)(0.2588L)$ 0.183 g $L = L^2\alpha(\frac{1}{2} + 0.06198)$

0.183 $\frac{9}{4} = 0.2336 \, \text{d} \; ; \quad d = 0.7834 \, \frac{9}{4}$ $d = 0.7834 \, \frac{9.84 - 15^{3}}{0.15 - 9} \qquad \qquad \alpha = 51.2^{-6} \, \frac{1}{5^{2}}$

+ (EFy= [(Fy)eff: A-mg=-mā sin45" =-m(0.2588 Ld) sin45" =-m(0.2588 Ld) (0.7824] sin45"

A-mg = 0.1+34= 9 A= 0.8566 mg= 0.8566(2.540)(9.81=1/5") = 21.01N A= 21.0Nt

16.164



GIVEN: SQUARE PLATE OF SIDE L=150 mm AND M=2.5 ALG IS RELEASED FROM REST.

FINO: (a) & (b) A

me = 6 mg

SINCE BOTH A AND MY ARE VERTICAL, ax=0 AND a IS &

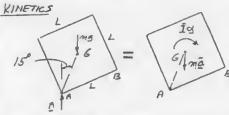
I= + m12

KINEMATICS

15° G Q L

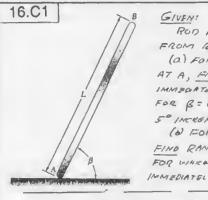
15° H5°

 $A6 = \frac{1}{1/3} V_{16}^{0}, \quad a_{G/A} = (A6) d = 15^{\circ}$ $a + = a_{A} + a_{G/A} = 15^{\circ}$ $a_{G/A} = \frac{1}{1/3} d = \frac{1}{1/2} a_{S/A} = 0.1831 d = 15^{\circ}$ $a_{G/A} = \frac{1}{1/2} d = 0.1831 d = 15^{\circ}$



* DIMA = Z(MA) EA: mg (A6) SIN 15° = I a + mā (A6) SIN 15° mg (\frac{1}{\sigma}) SIN 15° = \frac{1}{\sigma} + m(\O.183 La)(\frac{1}{\sigma}) SIN 15° \\
\text{C183} \frac{3}{3} = (\frac{1}{2} + 0.033494)

0.183 \frac{9}{L} = 0.2002 \d ; \d = 0.943 \frac{9}{L} = 0.9143 \frac{9.81 \sim 15^2}{0.150}



GIVEN: W= 516 RUD AB RELEASED FROM REST. (a) FOR NO SLIFFING AT A, FIND NA AND FA IMMBOATOLY AFTER RELEASE FOR B = 0 7085° DSING 50 INCKEMENTS. (B FOR 45 = 0.50, FIND RANGE OF VALUES OF B FOR WHER POD WILL SLIP IMMEDIATELY AFTEL ". GIERE.

WE NOTE THAT ROD RUTATES ABOUT AND THAT IMMEDIATELY AFTER RELEASE W.O.

+)
$$I(M_A) = I(M_A)_{AB}$$
; $mg(\frac{1}{2}\cos\beta) = I\alpha + m\bar{\alpha}(\frac{1}{2})$
 $\frac{1}{2}mgL\cos\beta = \frac{1}{2}mL^2\alpha + m(\frac{1}{2}\alpha)\frac{1}{2}$
 $= \frac{1}{3}mL^2\alpha$
 $\alpha = \frac{3}{2}\cdot\frac{9}{L}\cos\beta$ (1)

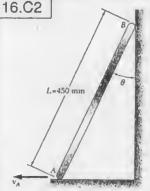
$$\begin{array}{ccc}
\pm \sum_{k=1}^{n} \sum_{j=1}^{n} \sum_{k=1}^{n} \sum_{k=1}^{n} \sum_{j=1}^{n} \sum_{k=1$$

$$+1$$
 $\Gamma_{y} = \Sigma(F_{y})eH: N-mg = m\bar{\alpha} \cos\beta$
= $m = \frac{1}{2} \cos\beta \cos\beta = m = \frac{1}{2} \left(\frac{3}{4} + \frac{3}{4} \cos\beta\right) \cos\beta$
 $N = mg(1 - \frac{3}{4} \cos^{2}\beta)$ (3)

GUTLINE OF PROGRAM:

- (a) FOR B = 0 TO 85° AT S INCREMENTS, DETERMINE F (from EQ(2)) AND N (from EQ. (3), ALSO DETERMINE REQUIRED VALUE OF ME FIN
- (b) USE SMALLER INCREMENTS TO FIND TWO VALUES OFB COERESPONDING TO M,= 0.50.

beta	F	N	WN	7s11p7	
0.000	0.000	1.250	0.000	no slip	
5.000	0.328	1.278	0.255	no slip	
10.000	0.641	1.363	0.470	no slip	
15.000	0.938	1.501	0.624	slip	
20.000	1.205	1.689	0.714	s11p	
25,000	1.436	1.920	0.748	slip	
30,000	1.824	2.186	0.742	slip	
35.000	1.762	2,484	0.709	slip	
40.000	1.847	2.799	0.860	slip	
45,000	1.675	3.125	0.800	s11p	
60,000	1.647	3.451	0.535	s11p	
55.000	1.782	3.766	0.468	no slip	
80.000	1.624	4.063	0.400	no slip	
85.000	1.436	4.330	0.332	no alip	
70.000	1.205	4.561	0.264	no alip	
75.000	0.936	4.749	0.197	no slip	
80.000	0.641	4,887	0.131	no slip	
85.000	0.326	4.972	0.065	no s11p	
Seek	sisri of	range			
10.810	0.891	1,362			
10.620	0.691	1.382	0.500	s11p	
Seek	end of re	nge			
52.620	1.809	3.618	0.500	s11p	
52.830	1.609	3.618	0.500	no slip	
52.840	1.809	3.819	0.500	no slip	



GIVEN: M = 5Ag NA= 1.5 m/s 4 an=0,

EIND: NORMAL RECTIONS AT A AND B FOR B=07050° USING 5° INCREMENTS. VALUE OF & AT WHICH

END BLOSES CONTACT WITH WYALL

KINEMATICS:

ACCELERATION [ast] = a+ [as/No = 0] + [(as/A) + [0]

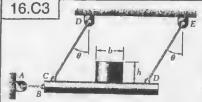
== 1/2 (a,+as)= 1/2 as; == 2 cose (2) (28/N) = L W

KINETICS: (3) A=m(q-a)

+) IM= I(MA) pt : B(LCOS6)-mg(=sin6) = - IX - ma(=sin6) B= m(9-2) + sin + + mL2

OUTUNE OF PROBLEMM: DATA M=529, L=0.45 m. FOR EACH VALUE OF & EVALUATE (U) AND a. THEN USE WAND a TO EVALUATE A AND B. USING SMALLER INCREMENTS FIND VALUE OF & FOR WHICH B=0.

theta	omega	alpha	a	A	8
dog.	rad/s	rad/s2	m/B2	N	N
			,		**
0.000	3.333	0.000	2.500	36.550	0.000
5.000	3.346	0.980	2.529	36,406	1,408
10.000	3.385	2.020	2.617	35.963	2.786
15.000	3.451	3.191	2.774	35.180	4.094
20.000	3.547	4.580	3.013	33.986	5 271
25.000	3.678	6.308	3.358	32.259	6.218
30.000	3.849	8.553	3.849	29.805	6.752
35.000	4.089	11.595	4.548	26.309	6.557
40.000	4.351	15.888	5.561	21.243	5.024
45.000	4.714	22.222	7.071	13.695	
50.000	5.186	32.049	9.413		0.955
00.000	3.100	32.040	0.413	1.984	-8.168
	Find theta	for 8 = 0			
45.747	4.777	23.420	7.357	12.265	0.000
45.748	4.777	23.422	7.367		0.002
45.749	4.777	23.423		12.264	0.001
40,140	4.111	23.423	7.358	12.262	-0.001



GIVEN: b=8 in, h=6 in.

30-16 CYLINDER

10-16 PLATFORM

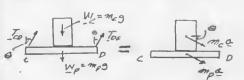
AFTER AB IS CUT,

FIND: MAN FOR WHEN

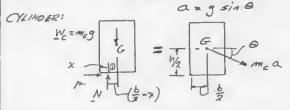
CYLINDED DOES NOT SUP

FOR 6=0 TO30° USING

5" INCIDENTENTS. THEN FOIL MS = 0.80, FIND & FOR WHICH SLIPPING IMPENDS. IN ALL CASES, CHECK WHETHER CYLINDER TIPS.



+ > IF = IF ext: (metmp) gsinG = (metmp) a



RESULTANT OF FORCES EXERTED BY PLATFORM ON TO CYUNOSIC ACTS AT DISTANCE & FROM LORNER.

$$\begin{array}{lll}
+\sum F_{y} + \sum F_{y} | q : & F = me \ a \cos \theta \\
+\sum F_{y} = \sum (F_{y})q : & N - me g = -me \ a \sin \theta \\
N = me (g - a \sin \theta) \\
(4)_{s} = \frac{F}{N} = \frac{Me \ a \cos \theta}{Me \ (g - a \sin \theta)} = \frac{(g \sin \theta)\cos \theta}{g - (g \sin \theta)\sin \theta} \\
4/_{s} = \frac{\sin \theta \cos \theta}{I - \sin^{2}\theta}
\end{array}$$

+) $ZM_0 = Z(M_0)_{2}$: $h_{2}g(\frac{b}{2}-x) = h_{2}a\cos(\frac{b}{2}) + h_{2}a\sin(\frac{b}{2}-x)$ $g(\frac{b}{2}-x) = (\frac{1}{2}\sin a)\cos(\frac{b}{2}+(\frac{1}{2}\sin a)\sin(\frac{b}{2}-x)$

 $(\frac{b}{4}-x)(1-5\ln^26)=\frac{1}{4}\sin \cos 3$; $(\frac{b}{2}-x)\cos^2\theta = \frac{1}{4}\sin 6\cos 6$ $\frac{1}{4}-x=\frac{1}{4}\frac{\sin 6}{\cos 6}$; $x=\frac{1}{4}(b-h \tan 6)$

CYLINDER TIPS IF X <0; ton 0 > b = din ; 0> 53.10

EVALUATE ME AND & FOR EACH VALUE OF Q.

PILINT ME AS MINIMUM VALUE OF ME FOR FOR FOR FOR FOR FOR FOR FOR FOR MUSICIPAL FOR MINIMUM VALUE OF ME FOR MUSICIPAL FOR MUSICIPAL

theta	x	mu req.	?slip?	?t1p?
0.000	4.000	0.000	no slip slips	no tip
5.000	3.738	0.087		no tip
10.000	3.471	0.176		no tip
15.000	3.196	0.268		no tip
20.000	2.908	0.364		no tip
25.000	2.601	0.466		no tip
30.000	2.268	0.577		no tip
35.000	1.899	0.700		no tip

-- Find theta for mu = 0.60 -----

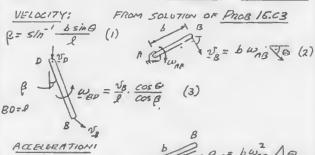
30.960 2.200 0.5999 30.980 2.199 0.6004



GIVEN: ENGINE SYSTEM OF PHOB 15.03.

WAS = 100010), AAR = 0 l = 160 mm, b = 60 mm mp = 2549, MBD = 349

FIND: COMPONENTS OF
DYNAMIC REACTIONS ON BD
AT PONTS B AND D FOR
B= 0 TO 180° USING 10°
INCREMENTS.



$$D = D$$

$$BD=R \qquad \alpha = \frac{\text{Rumo sing -ag caso}}{\text{R cos g}} \qquad (5)$$

$$W_{BD} = R \qquad (5)$$

$$\frac{1}{a} = a_{g} \cos \theta + lu_{n0} \cos \beta + l\alpha_{go} \sin \beta$$

KINETICS! WE FIRST FIND On AND ONE
(02) = -a sin & AND + (02) = a cost

SINCE G IS AT THE MIGOLE OF BD $\pm \bar{a}_{\chi} = \frac{1}{2} \langle x_{g} \rangle_{\chi} \tag{7}$

$$+i \bar{a}_y = \frac{1}{2} \left[\left(a_B \right)_y + a_D \right] \tag{8}$$

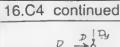
$$\frac{PSTON}{Py} \rightarrow \begin{bmatrix} = & 1 \\ \uparrow D_{y} & \uparrow m_{p}a_{p} \end{bmatrix}$$

$$+ \downarrow \Sigma F_{y} = \Sigma (F_{y})eF$$

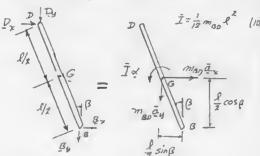
$$D_{y} = m_{p}a_{p} \tag{9}$$

NOTE: SINCE WE SEEK THE DYNAMIK REACTIONS, WE OMIT THE WEIGHT OF THE PISTON AND CONNECTINE ROD

(CONTINUED)



KINETICS: COHNECTING ROD



+) $\sum M_g = \sum (M_g)_{e,g}$: $D_x \cdot l \cos \beta \cdot D_y \cdot l \sin \beta =$ $-\frac{1}{2}\alpha + m_{g,p} a_x (\frac{1}{2} \cos \beta) - p n_{g,p} (\frac{1}{2} \sin \beta)$

Pivior BY & Save Fore
$$D_x$$

$$D_x = D_3 \frac{\sin \beta}{\cos \beta} - \frac{\vec{I} d}{A \cos \beta} + \frac{m_{B^0} \vec{a}_x}{\vec{z}} - \frac{m_{B^0} \vec{a}_y}{\vec{z}} \frac{\sin \beta}{\cos \beta}$$

$$D_x = Q_3 \tan \beta - \frac{\vec{I} d}{A \cos \beta} + \frac{m_{B^0} \vec{a}_x}{\vec{z}} - \frac{m_{B^0} \vec{a}_y \tan \beta}{\vec{z}} \frac{\sin \beta}{\cos \beta}$$

$$\pm \sum F_{\chi} = \sum (F_{\gamma})_{ef}:$$

$$B_{\chi} + D_{\chi} = m_{BD} \dot{a}_{\chi}$$

$$B_{\chi} = m_{BD} \dot{a}_{\chi} - D_{\chi} \qquad (12)$$

$$+\psi \cdot \Sigma F_{y} = \Sigma (F_{y})e^{x};$$

$$E_{y} + D_{y} = m_{0}D_{0} = 0$$

$$B_{y} = m_{0}D_{0} = 0$$
(13)

CUTLINE OF PROGRAM:

PROLUBIA, IN SEQUENCE, EQS. (1) THROUGH (13).
EVALUATE AND PUNT By, By, By, AND Dy FOR
VIALUES OF & FROM O TO 180° AT 5°
INCREMENTS:

Positive directions of force components sre: DOWN and TO THE RIGHT

theta	Bx	Ву	Dx	Dy
deg	N	N	N	N
0	0.00	4605.82	0.00	-2261.78
10	108.19	4497.37	-279.57	-2203.38
20	182.74	4177.66	-520.30	-2031.39
30	194.59	3663.87	-688.07	-1755.71
40	124.19	2985.61	-758.59	-1393.47
50	-33.57	2185.52	-722.48	-969.45
60	-265.48	1318.43	-589.25	-515.59
70	-539.76	447.34	-387.68	-68.61
80	-811.62	-364.52	-160.35	334.94
90	-1034.81	-1064.65	47.85	665.41
100	-1174.86	-1621.34	202.89	906.22
110	-1217.65	-2028.11	290.21	1056.59
120	-1169.20	-2300.43	314.47	1129.34
130	-1048.31	-2466.79	292.25	1145.24
140	-877.31	-2558.80	242.90	1126.72
150	-675.57	-2604.17	182.09	1093.40
160	-456.95	-2623.56	119.39	1060.08
170	-230.09	-2630.38	58.71	1036.51
180	-0.00	-2631.89	0.00	1028.08

16.C5



GIVEN: UNIFORM

BAR OF MASS AN

SUPPORTED BY

SPRINGS OF

CONSTANT AS.

IMMEDIATELY AFTER

BELLUNG AC BREAKS

FOR VALUES OF & FROM () TO 90, USING 10° PICKEMENTS

STATICS: INITIAL SPECIAL TENSIONS

* TEFF=0

TEFF=0

TEFF=0

TEFF=0

TEFF=0

TEFF=0

TEFF=0

TEFF=0

TEFF=0

KINETICS! JUST AFTER AC BREAKS

 $\frac{\pm \sum F_{x} = \sum F_{x|eA'}}{\frac{ma}{2\cos \theta}} = \frac{ma}{x}$ $\frac{a}{2\cos \theta} = \frac{1}{2}g \tan \theta = (1)$

+ $V\Sigma E_{y} = \Sigma (E_{y})_{ep}$: mg = Tcas E = may $mg = \frac{mg}{7cas} cos6 = mag$ $\ddot{a}_{y} = \frac{1}{2}g \downarrow \qquad (2)$

+) $\sum M_{\varepsilon} = \sum (M_{\varepsilon})_{\varepsilon} q : (T \cos \varepsilon) \frac{1}{2} = \sum d$ $\frac{m9}{2\cos \varepsilon} \cos \varepsilon = \frac{1}{2} = \frac{1}{12}mL^{2}d \qquad (3)$

 $(a_{B})_{y} \stackrel{\triangle}{=} a_{x} \stackrel{\triangle}{=} a_{y} \stackrel{\triangle}{=} a_{y} \stackrel{\triangle}{=} a_{y} \stackrel{\triangle}{=} a_{z} \stackrel{$

 $+ \sqrt{(a_A)_y} = \bar{a_y} + \frac{1}{2}\alpha = \frac{1}{2}g + \frac{1}{2}(\frac{3g}{2}) = 2g\sqrt{(3g)} + \frac{1}{2}(a_A)_y = -\bar{a_y} + \frac{1}{2}\alpha = -\frac{1}{2} + \frac{1}{2}(\frac{3g}{2}) = g$ (7)

END A: $\beta = \tan^{-1} \frac{(a_A)_y}{(a_B)_x}$; $a_A = \frac{(a_A)_{-1}}{\cos \beta}$ (8,9)

END B: $\chi = ton^{-1} \frac{(a_0)y}{(a_0)x}$, $a_8 = \frac{(a_8)x}{\cos x}$ (10,11)

OUTLINE OF PROGRAM:

PROBRAM, IN SEQUENCE, EGS. (1) THROUGH (11).

EVALUATE AND PRINT and, B, and FOR
VALUES OF & FROM O TO 900, USINIC
100 INCREMENTS.

theta	[AA	bets }	[aB	gsmms)	
0.000	2.000	90.000	1.000	90.000	
10.000	2.002	87.476	1.004	84.962	
20.000	2.008	84.801	1.016	79.686	
30.000	2.021	81.787	1.041	73.898	
40.000	2.044	78.153	1.084	67.240	
50.000	2.087	73.409	1.164	59.210	
60.000	2.179	66.587	1.323	49.107	
70.000	2.426	55.516	1.699	36.052	
80.000	3.470	35.196	3.007	19.425	
90.000	infinite	0.000	infinite	0.000	

17.1 GIVEN: 6000-16 FLYWHEEL, R=36in.,

FIND: MAGNITUDE OF COUPLE DUE TO FRICTION KNOWING FLYIMMERL ROTATES ISOU REVOLUTIONS WHILE CONSTING TO REST.

Wo = 300 rpm (20) = 10 17 rad

I = m R = 6000 16
32.2 Ft/s (3 ft) = 1677 16 ft s

T, = 1/2 I at = 1/2 (1677)(10 71) = 827,600 ft 16, T2 = 0

U1-2 = -M = -M (1500 rev)(271 red) = -9424,7 M

T,+U,-2=T2: 827,600-9424.7 M = 0

M=87.8/6.60 M=87.8/6.ft

17.2 GIVEN: 50-Rg ROTOR, R= 180 mm

W= 3600 rpm, M= 3.5 N.m

FIND: NUMBER OF REVOLUTIONS AS

ROTOR COASTS TO KEST

Cuo = 3600 pm (20) = 120 To rods

Î = m R = (50 Rg)(0.180m) = 1.620 Rg·m?

T, = \frac{1}{2} \bar{1} w_0^2 = \frac{1}{2} (1.620)(120 Ti) = 115.12 RJ, T2 = 0

U102 = -M6 = -(3.5 N·m)6

T+U-2=72: 115.12-83 - (3.5 H.m)6 = 0 G= 32.891×103 red G= 5230 rev

GNEN: 8-16 DISK OF 9-in, DIAMETER ROD AB WERNS 316/A M=416-fl

FIND: LENGTH L IF
W IS 300 YPM A FTER
2 REVOLUTIONS

 $Y = 4.5 in. = \frac{3}{8} \text{ ft}$: $\omega = 300 \text{ pm} \left(\frac{2\pi}{60}\right) = 10\pi \text{ ray}$

$$\begin{split} & W_{ROD} = 8/b, \quad W_{ROD} = (3/b/t^2)L \\ & \bar{I} = \frac{1}{2} m_{OS2} r^2 + \frac{1}{12} m_{ROD} L^2 \\ & = \frac{1}{2} \frac{8}{3} (\frac{3}{8})^2 + \frac{1}{12} \frac{3L}{3} L^2 = \frac{1}{3} (\frac{9}{16} + \frac{L^3}{4}) \\ & T_1 = 0, \quad T_2 = \frac{1}{2} \bar{I} W^2 = \frac{1}{29} (\frac{9}{16} + \frac{L^3}{4}) (10\pi)^2 \\ & U_{1-2} = MD = (9/b \cdot f^2) (2rev) (\frac{2\pi rad}{7er}) = 16\pi \\ & T_1 + U_{1-2} = T_2; \quad O + 16\pi = \frac{1}{29} (\frac{9}{16} + \frac{L^3}{4}) (10\pi)^2 \\ & \frac{16\pi r_{12}}{(10\pi)^2} = \frac{9}{16} + \frac{L^3}{4} \\ & 3.2779 = \frac{9}{16} + \frac{L^3}{4}; \quad \frac{L^3}{4} = 2.7/7 \end{split}$$

7 = 10.869 ft3.

GIVEN: WO=0

IDISK = IO

W= WEIGHT / UNIT LENGTH

OF ROD

EINO: LENGTH L FOR
MAXIMUM SA AFTER
COURSE M IS APPLIED FOR
ONE REVOLUTION

 $T_{z} = C$ $T_{z} = \frac{1}{2} \left(I_{o} + \frac{1}{12} \frac{\omega L}{9} L^{2} \right) \omega_{z}^{2}$

 $U_{1-2} = M 9 = M(2\pi rad)$ $T_1 + U_{1-2} = T_2 : O + 2\pi M = \frac{1}{2} \left(I_0 + \frac{\omega L^3}{12g} \right) \omega_2^2$

 $w_{2}^{2} = \frac{4\pi M}{I_{a} + \frac{w_{L}^{2}}{129}}$ $V_{A} = \frac{L}{2} w_{z} : V_{A}^{2} = \frac{L^{2}}{4} w_{z}^{2} = \frac{77 ML^{2}}{I_{o} + \frac{w_{L}^{2}}{129}}$

DIFFERENTIATING WITH RESIDET TO L,

 $2\sqrt{4}\left(\frac{d\sqrt{4}}{dL}\right) = \left[2L\left(I_{0} + \frac{\omega_{L}^{3}}{129}\right) - L^{2}\left(\frac{3\omega_{L}^{2}}{129}\right)\right] \frac{\pi M}{\left(I_{0} + \frac{\omega_{L}^{3}}{129}\right)^{2}}$

 $\frac{c^{1}\sqrt{\lambda_{A}}}{c^{1}Z} = 0: \quad 2L\left(I_{a} + \frac{\omega r^{2}}{r^{2}9}\right) - L^{2}\left(\frac{3\omega L^{2}}{r^{2}9}\right) = 0$ $2I_{a}L - \frac{\omega L^{4}}{r^{2}9}: \qquad \qquad L^{3} = \frac{249}{\omega}I_{a}$

17.5 GIVEN: 300-Re PUNCHING MACHINE FLYVIAELL, REQUIRES 2500 J.

FIND: (0) W, IMMEDIATELY AFTER A FUNCTIONS
(i) IF M= 25 Hom, FILD REVOLUTIONS
BEFORE W IS ABAM 200 Y, M.

I=mik = (300 ks)(0.6m) = 108 kg·m2

(0.) $\omega_{i} = 300 \text{ rpm} \left(\frac{2\pi}{60} \right) = 10\pi \text{ mod/s}$ $T_{i} = \frac{1}{2} m \omega_{i}^{2} = \frac{1}{2} (108 \text{ fig. m}^{2}) (10 \text{ Trod/s})^{2}$ $T_{i} = 53.2\% \text{ kJ}$ $U_{i-2} = -2500 \text{ J} = 2.5 \text{ kJ}$ $T_{2} = \frac{1}{2} m \omega_{i}^{2} = \frac{1}{2} (108 \text{ fig. m}^{2}) \omega_{i}^{2}$

 $T_1 + U_{1-2} = T_2$: 53.296 & -2.5 & = $\frac{1}{2} (108 \text{ Rg} \cdot \text{m}^2) w_2^2$ $w_2 = 30.67 \text{ md/s} \left(\frac{60}{2\pi}\right) = 292.9 \text{ pm}$

W= 293 rpm

(b) Uz=1 = M6 25001 = (25 N·m) 0 0 = 100 vad (\frac{vev}{2n.rad}) = 15.9155 vev

L = 2.22 ft

17.6 GIVETIL (U, = 360 PPM OF PULICHING PARCHINE
FLYWHEEL, EACH PURICH REQUIRES 1500 Ft.16,
AFTER EACH PUNCH (U) = 0.95 W,
FINO: (a) I OF FLYWREEL
(b) REVOLUTIONS REQUIRED FOR
ANGULAL VELOCITY TO AGAIN BE 360 PPM IF

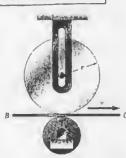
CONSTANT 18 16. FE COUPLE IS APPLEO

(a) $(\omega_{r} = 360 \, r_{pm} \left(\frac{2\pi}{40}\right) = 12 \, \pi \, rod f$ $(\omega_{z} = 0.95 \, \omega_{r} = 0.94 \left(12 \, \pi \, rod f\right) = 11.4 \, \pi \, rod f$ $T_{r} = \frac{1}{2} \bar{I} \, \omega_{r}^{2} = \frac{1}{2} \bar{I} \left(12 \, \pi\right)^{2}$ $T_{2} = \frac{1}{2} \bar{I} \, \omega_{r}^{2} = \frac{1}{2} \bar{I} \left(11.4 \, \pi\right)^{2}$ $U_{1-2} = -1500 \, (6-16)$

 $\frac{T_1 + U_{1 \to 2} = T_2 : \frac{1}{2} \bar{I} (12\pi)^2 - 1500 = \frac{1}{8} \bar{I} (11.4\pi)^2}{\bar{I} = \frac{2(1500)}{\pi^2 (12^2 - 11.4^2)} = \frac{3000}{130.57} = 21.649 \cdot 16.52 \cdot 5^2}$ $\bar{I} = 21.6 \cdot 16.52 \cdot 5^2.$

(b) $U_{Z=0} = M\Theta$: 1500 ib. it = (18 it. ib) ib $G = \text{ ib} = 2.32 \text{ rod} \left(\frac{\text{rev}}{2 \text{ it rod}}\right) = 13.263 \text{ rev}$ $\Theta = 13.26 \text{ rev}$

17.7 and 17.8



GIVEN: DISH PLACED ON EST.

THEN WED. COEFFICIENT OF

KINGTIC FRICTION = 4/5.

FIND: REVOLUTIONS BEFORE

WE CONSTANT,

PROBLEM 17.71

IN TERMS OF U, Y, AND H,

PROBLEM 17.8:

FOR Y = 6/17, 5 = 40 ft/s,

AND 4/5 = 0.20.

CNLY FORCE DOWN WORK IS F. SINCE ITS MOMENT ABOUT A IS METF, WE HAVE



U,-2=MO= rFO= r(4,209)0

AMBURAN NELUCITY BECOMES CONSTANT INHEM $\omega_2 = \frac{\sqrt{r}}{r}$

7 = 0 $7_2 = \frac{1}{2} \hat{I} w_3^2 = \frac{1}{2} \left(\frac{1}{2} m r^2 \right) \left(\frac{V}{r} \right)^2 = \frac{m V^2}{4}$

T,+U,-= T2: 0+r-4,mg0= == +

 $\Theta = \frac{\pi^2}{4rH_59} \text{ rad} \qquad \Theta = \frac{5}{8\pi r A_59} \text{ rev}$

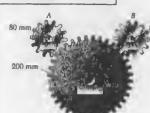
PRUBLEM 17.8: r=0.556, 45=0.20, 2=40 fl/s

6 = (40 Pe/s) 2 871(0.5ft)(0.70)(32.2 (1/52)

6=19.77 rev

NOTE: RESULT IS IMPERIORET OF W.

17.9 and 17.10



GIVEN: ME = 12Ag, RE 150mm

NO = ME = 8.4Ag

IM = ARE = 60 MM

M = 10 N·M

(=1MO: (a) REVOLUTION;

OF C AS WE INCREASES

I=ROM 100 YPM TO 450 YPD

(b) TANGENTIAL FORCE ON A

PROBLEM 17.9

M IS APPRILED TO GEAR C

PROBLEM 17.10

MIS APPLIED TO GEAR B

BOTO THE A COUNTY OF THE A COU

KINEMATICS:

 $\omega_A = \omega_B = \frac{200 \text{ mm}}{80 \text{ mm}} \omega_c = 2.5 \omega_c$

 $(\omega_{c})_{z} = 100 \text{ rpm} \left(\frac{2\pi}{60}\right) = 10.472 \text{ rad/s}$ $(\omega_{c})_{z} = 450 \text{ rpm} \left(\frac{2\pi}{60}\right) = 47.124 \text{ rad/s}$

WORK AND ENERGY $\bar{I}_{a} = \bar{I}_{a} = m \hat{k}^{2} = (2.4 \cdot k_{a})(0.06 \text{ m})^{2} = 8.64 \times 10^{-3} \text{kg} \cdot m^{2}$ $\bar{I}_{c} = m \hat{k}^{2} = (12 \cdot k_{a})(0.150 \text{ m})^{2} = 0.270 \cdot k_{a} \cdot m^{2}$

 $\frac{P_{OSI7704V} \int_{0}^{\infty} (\omega_{c})_{r} = 10.472 \text{ rad/s};}{(\omega_{A})_{r} = (\omega_{B})_{r} = 2.5 (\omega_{c})_{r} = 26.18 \text{ rad/s}}$ $T_{r} = Z \left[\frac{1}{2} \hat{I}_{A} (\omega_{A})_{r}^{2} \right] + \frac{1}{2} \hat{I}_{c} (\omega_{c})_{r}^{2}$ $= Z \left[\frac{1}{2} (8.64 \times 10^{3}) (26.18)^{3} \right] + \frac{1}{2} (0.276) (10.472)^{2} \right] = 20.726 \text{ J}$

 $\frac{FOSTRON 2: (u_{t})_{2} = 47.124 \text{ rad/s}}{(\omega_{h})_{2} = (u_{f})_{2} = 2.5(\omega_{o})_{q} = 117.81 \text{ rad/s}}$ $T_{2} = 2\left[\frac{1}{2}(8.64 \times 10^{-8} \text{X})^{17.81}\right] + \frac{1}{2}(0.2 \times 10^{124})^{3} = 419.71 \text{ J}$

PROBLEM 17.9: M=10N. M APPLIED TO GEAR C

T,+U,-2=T2: 20.7% 5+100=419.715

GEAR A: O = 2.5 G = 2.5(39.90) = 99.75 rod

F= 7.14 N F= 7.14 N 1

PROBLEM 17.10: M=10 N·m APPLIED TO BEAR B

NOTE I ANGULAR SPECOS AKE SAME
AS IN PROBLEM, THUS T, AND TZ ARE ALSO THE SAME
T,=20.726 J Te=419.71 J

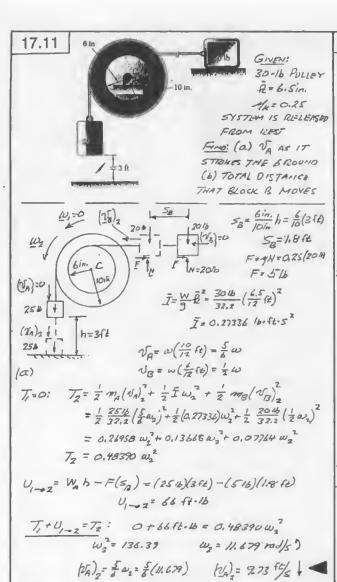
WE HAVE $U_{1-2} = M \in_B = 1000$ $T_1 + U_{1-2} = T_2$: 20.726] + 1000 = 419.71] $E_B = 39.90 \text{ rad}$

6=2.50; 39.90 rad= 2.56; 6=15.96 rad 6=2.54 rev

 $GFARRA: G_{A} = G_{B} = 39.90 \text{ rad}$ $T_{1} + U_{1-2} = T_{2}: \frac{1}{2} m_{A}(\omega_{A})^{2} + F(0.08)G_{A} = \frac{1}{2} m_{A}(\omega_{A})^{2}$ $\frac{1}{2} (B.64 \times 10^{3})(26.18)^{2} + F(0.08)(39.90) + \frac{1}{2} (B.64 \times 10^{3})(17.9)^{2}$ 2.961 + 3.192 F = 59.94

F=12861

F=17.86N/



(b) BLOCK B COASTS TO KEST

TOTAL ENERGY OF BLOCKS WID PULLEY JUST BEFORE IMPACT = 66 St. 16

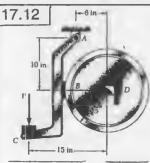
KINETIC ENERGY OF BLOCK A JUST BEFORE IMMIT $T_A = \frac{1}{2} \frac{v_{IA}}{9} (\hat{\tau}_{I})_2^2 = \frac{1}{2} \frac{251b}{32.2} (9.73 \text{ ft/s})^2 = 36.75 \text{ ft-b}$

AFTER BLOCK A STRIKES THE GROWD, WE FIND THAT THE KINETIC ENGREY OF THE PULLBY C AND BLOCK B IS T = 66 ft. 16 - 3675 ft. 16 = 29.25 ft-16

FOR SYSTEM TO STOP, 29.25 ft. 16 OF ENERGY MUST BG DISSIPATED BY THE FRICTION FORCE, F= SB.

29.25.ft.16 = (516)d d=5.85ft

TO FIND TOTAL DISTANCE MICNED BY B, WE ASD SB=11896. TOTAL DISTANCE= 1.8+5.85 = 7.65 ft



GIVEN: I= 14/6.ft.52 W,= 360 mm) 4-035

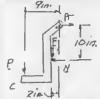
FIND: P SO THAT FLYVINEEL STOPS IN 100 REVOLUTIONS.

Bir. IW

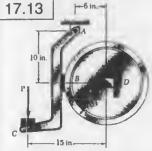
W= 360 rpm (27) = 127 rod/s) T = 1 Iw = + (14 16. St. 5) 12 Trad/s) TI = 9948.6 92.16 6= 100 rev (271 val) = 628.32 red U1=2=-MO=-FrO=-F(3 ft)(628,32 rad)

9948.6 - F(= 1628.32)=0 T, + U, = 7= T2: F=23.75 16 F= 0/5 N: 23.25 /b= (0.35) N; N= 67.86 16

FREE BODY. BRAKE AC;



+) [M=0 P(gin.) + F(zin.) - N(10 in,) = 0 9 p+(23,75)(2) - (67.86)(10) = 0 98-631,1=0 P=70.116 9 = 70.1216



GIVEN: I=1416.55.5 W, = ?60 yr] 1/20.35

FIND: P SO THAT FLY WHEEL STOPS IN 100 REVOLUTIONS

F= MEN W 7 8in.

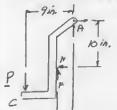
W,= 360 YPM (277)= 12 TT rad/s) T, = - 1 w, = - (14 16. ft s XIZ 17 rod/s) T,= 99426 ft. 16

Q=100 Yev (211 rad) = 628.32 rad

U1-1=-MO=-FrO=-F(2/5/6/0320)

9918.6-F(=)(6243=)=0 T;+U,-2= T2: F= 23,7516

F=4, N: 23.75/6=0,35N; N=628616



FREE BODY: BRAKE AC +) EM= C P(9in) - F(2in) - N(10in) =0 9P-123,25/2)-(67.86)(10)=C

9P-726.1=0 P=80.716 P= 80.1816

17.14 and 17.15



GIVEN: FIRKTION

DISKS A, B, AND C

ARE MADE OF SAME

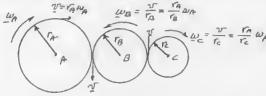
MATERIAL AND

HAVE SAME

THICKNESS

PROBLEM 17.14: FIND: EXPRESSION FOR WA AFTER
THE COUPLE M IS APPLIED FOR ONE REVOLUTION
PROBLEM 17.15: FIND: REVOLUTIONS OF A REGUINED
FOR WA = 150 rpm WHEN M = 6016-in,
YA=0in, YB=6in, YB=4in, AND WA=1210.

DENOTE VELOCITY OF PENIMETER. EY of:



DENOTE WASS DENSITY OF MATERIAL BY P AND THICKNESS OF DISKS BY t.

THEN MASS OF A DISK IS $m = (volume) p = (\pi r^2 t) p$ AND $\bar{I} = \frac{1}{2} m r^2 = \frac{\pi r t}{r} r^4$

KINETIC ENERBY: T= Z 1 IN2

$$T = \frac{1}{2} \left(\frac{\pi \rho c}{2} \right) \left[r_A^{\gamma} \omega_A^2 + r_B^{\gamma} \omega_B^2 + r_C^{\gamma} \omega_C^2 \right]$$

$$= \frac{1}{2} \left(\frac{\pi \rho c}{2} \right) \left[r_A^{\gamma} \omega_A^2 + r_B^{\gamma} \left(\frac{r_A}{r_B} \right)^2 \omega_A^2 + r_C^{\gamma} \left(\frac{r_A}{r_C} \right)^2 \omega_A^2 \right]$$

$$T = \frac{1}{2} \left(\frac{\pi \rho c}{2} \omega_A^2 \right) r_A^{\lambda} \left[r_A^2 + r_B^2 + r_C^2 \right]$$

$$WORK; \quad U = MG \qquad (\omega_i = 0; \ \omega_2 = \omega_A.$$

 $\frac{T_{1}+U_{1-2}=T_{2}}{T_{1}+U_{1-2}=T_{2}} = 0 + M\Theta = \frac{\pi \rho t}{4} w_{A}^{2} v_{A}^{2} \left[r_{A}^{2} + r_{B}^{2} + r_{C}^{2} \right]$

PROBLEM 17.19 FOR 0 = 287; M(28) = 700 MA TA [1+(10)2+(10)2)

$$\omega_{A}^{2} = \frac{8 M_{o}}{\ell + r_{A}^{2} \left[1 + \left(\frac{r_{B}}{r_{A}}\right)^{2} + \left(\frac{r_{C}}{r_{A}}\right)^{2}\right]}$$

PROBLEM 17.15: RECALL THAT MA=TIVALP AND WRITE
EG.(1) AS:

T= \frac{1}{4} (\pi r_A^2 t \rho) (r_A^2 + r_B^2 + r_c^2)

 $T = \frac{1}{4} \left(\frac{W_A}{9} \right) r_A^2 \left[1 + \left(\frac{r_B}{r_A} \right)^2 + \left(\frac{r_C}{r_A} \right)^2 \right] \omega_A^2$

DATA: WA = 150 YPM (21) = 517 rad/s

WA= 1216, rA=811, rB=611, rE=411. M=6016.in. = 5 ft.16

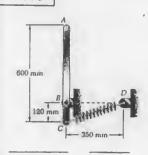
U1-2= MO= (5 Fl-16) G

$$\frac{T_1 + U_{1-2} = T_2}{O + 5\Theta} = \frac{1}{4} \left(\frac{12.16}{32.2} \right) \left(\frac{8}{12} + \frac{1}{4} \right) \left[1 + \left(\frac{6in}{6in} \right)^2 + \left(\frac{4in}{6in} \right)^2 \right] \left(5^- \pi \right)^2$$

$$5\Theta = 0.04/408 \left[1 + \frac{4}{7} + \frac{1}{7} \right] \left(5\pi \right)^2$$

$$5\Theta = 18.518 ; \Theta = 3.704 \text{ rad} \left(\frac{\text{rev}}{2\pi \text{ rad}} \right) = 0.52894 \text{ rev}$$

17.16



GIVEH! 4-Ag RODAC

SPRUIG: &= 400H/m

WISTRETURE, LEVETH

= 150 MM.

ROD IS KELEGRED FILCH

1255.

FIND: U) AFTER ROD

HAS ROTATED 90°

POSITION 1: UNSTRETCHED

SPANNS: X,= CD - (150 mm) = 370 - 150 = 250 mm = 0.20 m V= \frac{1}{2} \frac{1}{2} \tau \tau^2 = \frac{1}{2} \frac{1}{2} \tau^2 = \frac{1}{2} \frac{1}{2} \tau^2 = \frac{1}{2} \frac{1}{2} \tau^2 = \frac{1}{2} \tau^2 = \frac{1}{2} \frac{1}{2} \tau

51:A4177 Va = Whe more (4 Ro)(9.81 m/s)(0.60 m) = 7.063 J V=Ve Hy = 9.68 J +7.063 J = 16.743 J

KINETE ENERGYS TIES

POSITION 2: SPRING: Vz=230mm - 150mm = 60mm = 0,08 m, Ve= 2 R x, = 2/400N/m/0.08m) = 1.22 J

GRAVITY: Vg=Wh=0 V=Ve+Vg=1.281

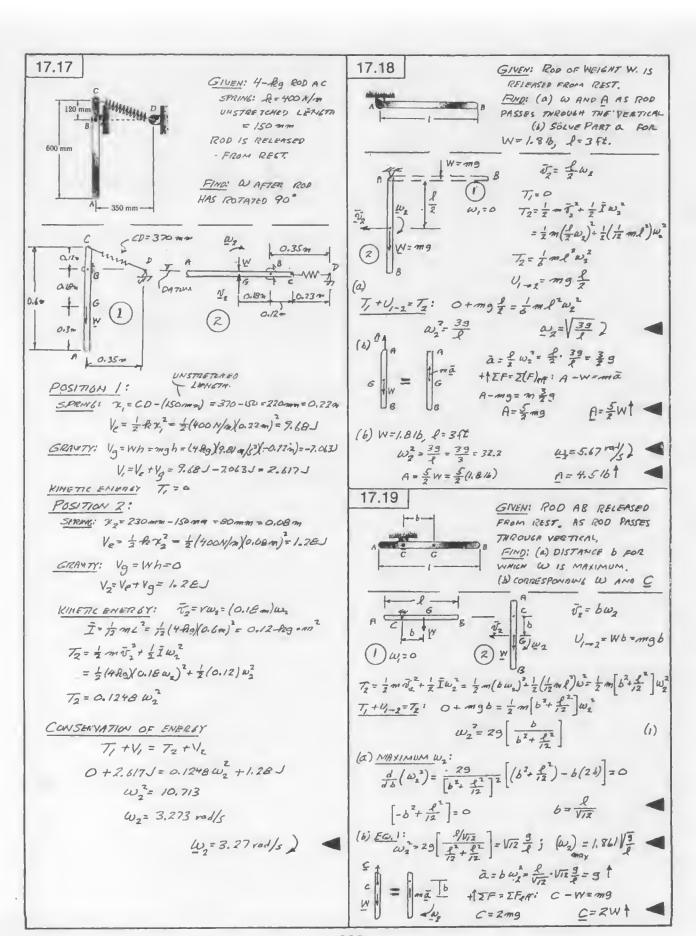
KINETIC EMERIETI $\bar{V}_Z = r\omega_\chi = (0.18m)\omega_1$ $\bar{I} = \frac{1}{12}mL^2 = \frac{1}{12}(4R_0\chi_0.6m)^2 = 0.12-R_0 \cdot m^2$ $\bar{I}_Z = \frac{1}{2}m\bar{V}_2^2 + \frac{1}{2}\bar{I}\omega_2^2$ $= \frac{1}{2}(4R_0\chi_0.18\omega_2)^2 + \frac{1}{2}(0.12)\omega_2^2$ $\bar{I}_Z = 0.1246\omega_2^2$

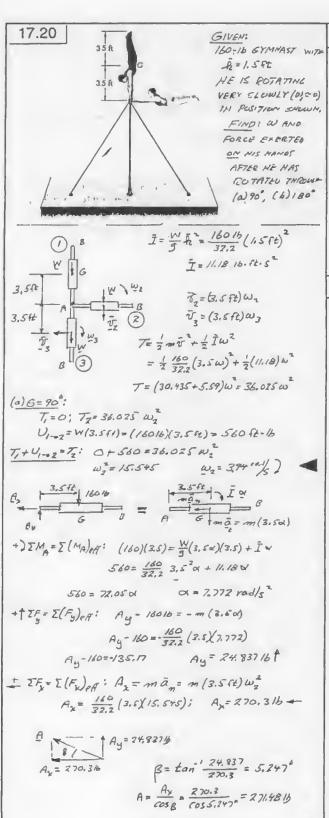
CONSERVATION OF ENERGY:

 $T_1 + V_1 = T_2 + V_2$ $O + 16.743J = 0.1248 \omega_e^2 + 1.28J$ $\omega_2^2 = 123.9$ $\omega_3 = 11.131 \text{ rad/s}$

W2=11.13 red/5)

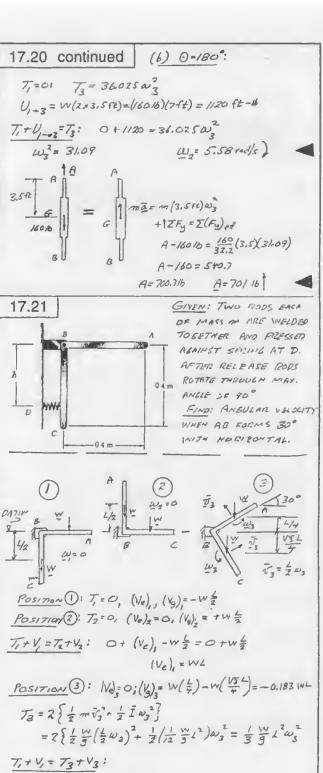
0=0,589 rey

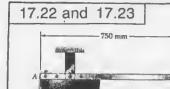




A= 27/16 1 5.2°

(CONTINUED)





GIVEN: MAB = 6.89

1.8-89 SEMICIRCULAR

DISK.

SPRING OF PE-160 M/A

BUNSTRETCHED WHEY

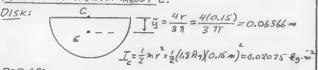
AB IS HORIZONTAL

IF SYSTEM IS

REGERSED FROM REST,
FIND: W AFTER 90° NOTATION
PROBLEM 1222: WITH SPRING ATTACHED
PROBLEM 1223: SHUNG REMOVED

MOMENT OF INDRITA ABOUT C.

150 mm



ROD AB:

| 0,375a | 0,371a | B

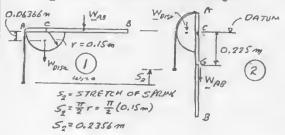
| C | G

| C | G

| C | C | S m

Ic= I+mix = 1/2 (6. Re)(0.75m) + (6. Re)(0.225m) = 0.28125 + 0.30375 = 0.585 Rg.m2

TOTAL I OF ASSEMBLY: I = 0.02025+0.585 = 0.60525 Ag. m2



POSITION 1: T, =0, $V_1 = |W_{DISK}(-0.06366 m)$ $V_2 = (1.8.89)(9.81)(-0.06336) = -1.1188 J$ POSITION 2: $(V_2)_2 = \frac{1}{2} - 25 S_2^2 = \frac{1}{2} (160 \mu/m)(0.2356 m)^2 = 4.44 J$ $(V_3)_2 = W_{AB}(-0.225 m) = (6.89)(9.51)(-0.225) = -13.24J$

FOR MON CENTROIDAL PLOTATION INE USE EQ.(17.10) $T_2 = \frac{1}{2} I_c w_2^2 \cdot \frac{1}{2} (0.60525) w_2^2 = 0.3026 w_2^2$

 $\frac{PROBLEM 17.22}{O - 1.1168J = 0.3026 w_{2}^{2} + 4.44J - 13.24J}$ $7.681 = 0.3026 w_{2}^{2}$ $w_{2}^{2} = 25.38 \qquad w_{3} = 5.04 \text{ radf}$

PROBLEM 17.23: SPRING IS REPLOYED, THUS $(V_e)_2 = 4.44 \text{ J}$ IS REPLOYED FROM POTENTIAL

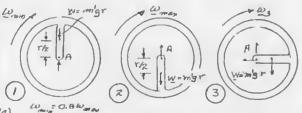
ENERGY IN POSITION 2. WE NOW WRITE $T_1 + V_1 = T_2 + V_2$ $0 - 1.1182J = 0.3026 \omega_2^2 - 13.24J$ $12.121 = 0.3026 \omega_2^2$ $\omega_2^2 = 40.05$ $\omega_2 = 6.33 \text{ Iod/s}$

17.24 8 in.

GIVEN: ASSEMBLY MADE OF CV = 0.25 16/A ROD, KNOWING THAT WORLD = 0.80 WMAX FIND: (a) WMAX (b) W WHEN G=90°.

DENOTE MASS PER UNIT LENGTH BY m' AND RADIUS BY r $I_{A} = \frac{1}{1_{ROD}} + I_{RING} = \frac{1}{3} (m' \dot{\tau}) r^2 + (2\pi r m') r^2 = 6.6165 m' r^3$

FOR HON CENTROIDAL ROTATION THE KINETIC ENERGY OF THE ASSEMBLY IS \$ 1 A W2



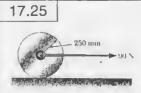
 $\frac{7_{1} + V_{1} = 7_{2} + V_{2}: \frac{1}{2}I_{A}\omega_{min}^{2} + nigr\frac{r}{2} = \frac{1}{2}I_{A}\omega_{max}^{2} - nigr\frac{r}{2}}{\frac{1}{2}I_{A}(\omega_{max}^{2} - \omega_{min}^{2}) = migr^{2}}$ $\frac{1}{2}6.6/65 mir^{3}(1 - 0.8^{2})\omega_{max}^{2} = migr^{2}$

 $W_{max}^{2} = 0.83\%5 \frac{3}{7} = 0.83965 \frac{32.2 \text{ Fils}^{2}}{(9/2 \text{ st})} = 40.555$

(b) $T_2 + V_2 = T_2 + V_3$: $\frac{1}{2} I_{A} w_{mox}^2 - m_g^2 t (\frac{V}{2}) = \frac{1}{2} I_{A} w_g^2$ $\frac{1}{2} (6.6165 \text{ m}^2 \text{ V}^3) (0.82765 \frac{3}{V}) = \frac{m_2 r^2}{2} = \frac{1}{2} (6.6165 \text{ m}^2 \text{ V}^3) w_g^2$ $Z.1736 m_g^2 r^2 - 0.5 m_g^2 r^2 = 3.3162 m_r^2 w_g^2$ $W_3^2 = \frac{7.27736}{3.3162} = \frac{9}{r} = 0.6885 \frac{9}{r}$ $W_3^2 = 0.6885 \frac{37.2575}{(9/2)^4 + 1} = 33.26$

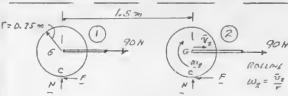
W3 = 5.77 " ad/s }

NOTEL RESULTS ARE INDEPENDENT OF WEILHT PER UNIT LENGTH OF THE ROD USED TO MAKE THE ASSEMBLY



GIVEN: 20-Rg ROLLER ROLLS WATHOUT SLIPFING FINO: (a) & AFTER 1.60

(b) FRICTION FORCE KIGOINED TO FILEVENT SLIPPING.



INSTANT CENTER ATC: THUS F DOES NO INDICE T,=0 U, == (90N)(1.5 =) = 135 J 72= 2m3 + 1 Iw2 = 1 mis + 1 (2mv2) 1/2 2 T2: 3 min = 3 (20 to (12) = 15 %

T,+ U, == T2: 0+ 1351 = 15 02 v=3m/5->

(B) CONDOER MUTION ABOUT MASS CENTER. $T_1 = 0$ $T_2 = \frac{1}{2} \left[\frac{1}{2} w_2^2 = \frac{1}{2} \left(\frac{1}{2} m r^2 \right) \left(\frac{\bar{V}_2}{r} \right)^2 = \frac{1}{4} m \bar{N}_1^2$ U1-02= F (1.5-20)

T,+U,-2=T2: 0+1.5F= +mv; 1, SF = 1/20.kg/(3a/s): F= 30 B -

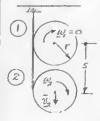
17.26 and 17.27



GIVENT: OBJECT SHOWN IS AFLEASED FROM REST FINO: I AFTER DOWNWARD MOVENIENT 5

PROBLEM 17.26 FOR A CYLINDER PILOBLEM 17.27

FOR A THII- WALLED PIPE



R = RADIUS OF STRATION F=rw W= T 万= シャガンナラエルア

= $\frac{1}{2}m\bar{v}_{2}^{2} + \frac{1}{2}(m\bar{R}^{2})(\bar{V})$ Ta= 1 m (1+ 1/2) N2 V= mgs

T,+U,-2+T2: 0+mg5= = = (1+ +) v2 N= 295 (1)

PROBLEM 17.25: CYLINDER Po= 17 $\sqrt{1/2} \sqrt{\frac{495}{3}}$

PROBLEM 17.27: THIN-MAUEO PIPE hisv V= 295 - 95 ~= V95 + 17.28



GIVEN: HOOP OF MASS TH ROLLS TO RIGHT . WITH COLLAR BOE MASS on AT TUP CU = QUIT AND AT BOTTOM W = 3 W,). FIND: W, IN TERMS OF 9 AND T.

U1-2= W(2r) = mg/20) = 2 mgr T,==m NA+ = IN,+ + = m VR $= \frac{1}{2}m(r\omega_i)^2 + \frac{1}{2}(mr^2)\omega_i^2 + \frac{1}{2}m(2r\omega_i)^2 = 3mr^2\omega_i^2$ T2= = = mNA2 + = IW2 + = mNB = 1 m(rw) + 1 mr w2 + 0 = mr w2 T,+ U,-2=T2: 3mrw, +2mgr=mrw2

GIVEN! W1=3W, 3mr2W, +2mgr=mr2(3W,) 2mgr = 6mr2w,2; W,= 3; W= 19/3r

17.29

GIVEN: HALE SECTION OF PIFE OF MASS M. RELEASED FROM RUST. AFTER ROLLING THROUGH FIND: (a) w

(6) REACTION

ω, = V, = 0 V, = (AG) W2 = r(1-2) W, U1-2= W(06)

I=mr=m(06)=mr=m(21)=mr2(1-42) 72= 1 m v2 + 1 I w2 = 1 m(1-2) 2 2 + 2 mr2 (1-41) w2 $(a) = \frac{1}{2} m r \left[\left(1 - \frac{4}{17} + \frac{4}{12} \right) + \left(1 - \frac{4}{172} \right) \right] = \frac{1}{2} m r \left(2 - \frac{4}{12} \right)$ Ti+U1-1= Tz: 0+mg 2r = 2mr2 (2-4)w2 $\omega_2^2 = \frac{2}{\pi (1 - \frac{2}{\pi})} \cdot \frac{9}{r} = 1.75/9 \cdot \frac{9}{r}$ $\omega_2 = 1.32 + \sqrt{9/r}$

(6) KINEMATICSI SINCE O MOVES HURIZOTTALLY, (QO)4=6 0.1 0.1 a=(06)w== 2 (1.7519 3)=1.11539

KINETICS: ma = 1.1153mg

+ [Fy = I(Fy), q: A-mg=1.1153mg; A=2.12mg]

17.30 and 17.31



GIVEN: 14-16 CYLINDERS

OF 5-IM. PRODUS.

PRUBLEM 1730:

[Wab = 30 rod/s]

FIND: (e) OISTANCE A

WILL HISE REFORE WB = 5 m/s.)

(1) TENSION IN CORD A - 8

HICLELAN 1731: SYSTEM IS

RELEASED FROM UEST

FIND: (b) YA AFTER 3 SE OF

MATTER. (b) T IN CORD A - B



$$\begin{split} \mathcal{V}_{O} &= r \, \omega_{g} \qquad \omega_{A} = \frac{\mathcal{V}_{O}}{2 \, r} = \frac{r \, \omega_{B}}{2 \, r} = \frac{1}{2} \, \omega_{g} \\ \bar{\mathcal{V}}_{A} &= r \, \omega_{A} = \frac{1}{2} \, r \, \omega_{g} \end{split} \tag{1}$$

KIMENC ENFECT: $T = \frac{1}{2}m\vec{n}_A^2 + \frac{1}{2}\vec{1}\omega_A^2 + \frac{1}{2}\vec{1}\omega_B^2$ $T = \frac{1}{2}m\left(\frac{1}{2}r\omega_B^2 + \frac{1}{2}\left(\frac{1}{2}mr^2\right)\left(\frac{1}{2}\omega_B^2\right) + \frac{1}{2}\left(\frac{1}{2}mr^2\right)\omega_B^2 = \frac{7}{16}mr^2\omega_B^2$



SINCE COILD IS INEXTENSIALE, WOUNT IS DONE DALY BY
THE WEIGHT OF CYLINDER B

Up-2 -Wh = -mgh

r= 5/12

PROBLEM 17.20: $(\omega_{R})_{z} = 30 \text{ vol/s}$); $(\omega_{R})_{z} = 5 \text{ vol/s}$) $T_{1} + U_{1-Z} = T_{2} : \frac{7}{16} mr^{2} (\omega_{R})_{1}^{2} - mg h = \frac{7}{16} mr^{2} (\omega_{R})_{2}^{2}$ $h = \frac{7}{16} \frac{r^{2}}{3} \left[(\omega_{R})_{1}^{2} - (\omega_{R})_{2}^{2} \right]$ (2)

h=\frac{7}{16}\left(\frac{5}{12}\text{ft}\right)\frac{(30)^2-(5)^2}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\left(\frac{5}{12}\text{ft}\right)\frac{1}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\right)\frac{7}{32.279/6}=2.064\text{ft}\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\text{ft}\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\text{ft}\text{ft}\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft}\text{ft}\text{ft}\text{ft}\text{h=\frac{7}{16}\text{ft

THE DISTANCE THAT A MOVES $V_{1-\frac{\pi}{2}} - T_{AB}(2h)$ FOR OWLY CYLINGER B, $T = \frac{1}{2} \tilde{I} \omega_{B}^{2}$

 $T_{1}+U_{1-2}=T_{2}: \frac{1}{2}\bar{I}(\omega_{B})_{1}^{2}-2hT_{AB}=\frac{1}{2}\bar{I}(\omega_{B})_{2}^{2}$ $T_{AB}=\frac{1}{4}\bar{I}\left[(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}\right]\frac{1}{h}=\frac{1}{4}\left(\frac{1}{2}mr^{2}\right)\frac{(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}}{\frac{1}{16}\frac{r^{2}}{5}\left[(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}\right]}$ $T_{AB}=\frac{1}{2}\frac{1}{2}mg=\frac{2}{3}W=\frac{2}{3}(1416) \qquad T_{AB}=\frac{416}{3}$

NOTE: T_{AB} IS INDEPENDENT OF (ω_B) , AND (ω_B) ?

PROOLEM 17.3! $(\omega_B) = 0$, h = 3 ft, $r = \frac{5}{72}$ ft

Since h and \tilde{t}_A are now downinard, U = +Wh = +mgh and Ea. 2 Ist' $h = -\frac{7}{16} \frac{v^2}{9} \left[(\omega_B)^2 - (\omega_B)^2 \right]$

 $3f2 = -\frac{7}{16} \left[\frac{5}{12} ft \right]_{32,2fl/2}^{2} \left[0 - (\alpha_R)_2^2 \right]$ $\left[(\omega_R)_2^2 \cdot 1271.8 \quad (\omega_R)_2 = 35.66 \text{ rad/s} \right]$

FO(1): $\vec{V}_{A^{\frac{1}{2}}} \frac{1}{2} \cdot r(\omega_{a}) = \frac{1}{2} \left(\frac{5}{12} \epsilon_{E} \right) 35.11 \text{ rad/s} = 7.430 \text{ FMs}$ $\vec{V}_{A^{\frac{1}{2}}} \frac{1}{2} \cdot r(\omega_{a}) = \frac{1}{2} \left(\frac{5}{12} \epsilon_{E} \right) 35.11 \text{ rad/s} = 7.430 \text{ FMs}$

TENSON TAB: SINCE TAB IS INDEPUNDENT OF VELCOTY,
INE AGAIN HAVE
TAB = 416

17.32 GIVEN

GIVEN: THE SASS

MA = 6 R9

MB = 1.5 R9

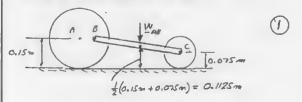
SYSTEM IS DELEMBED

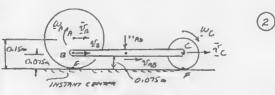
FROM IZET

TS mm FIND! VAS AFTER

DISK A HAS

ROTATED 90°,





 $\tilde{V}_{R} = \tilde{V}_{AB}$ $\tilde{V}_{A} = \frac{\tilde{V}_{B}}{BE} = \frac{\tilde{V}_{AB}}{0.075m}$ $\tilde{V}_{A} = 2\tilde{V}_{B} = 2\tilde{V}_{AB}$ $\tilde{V}_{C} = \tilde{V}_{AB}$ $\tilde{V}_{C} = \tilde{V}_{AB}$ $\tilde{V}_{C} = \tilde{V}_{AB}$ $\tilde{V}_{C} = \tilde{V}_{AB}$

U1-2= W(0.1125 m - 0.075 m)= (5 kg)(9.8)(0.0375 m) U1-2= 1,8394 J

 $= \frac{1}{2} \left[24 + 12 + 5 + 1.5 + 0.75 \right] \sqrt{48}$

T2 = 21.625 VAB

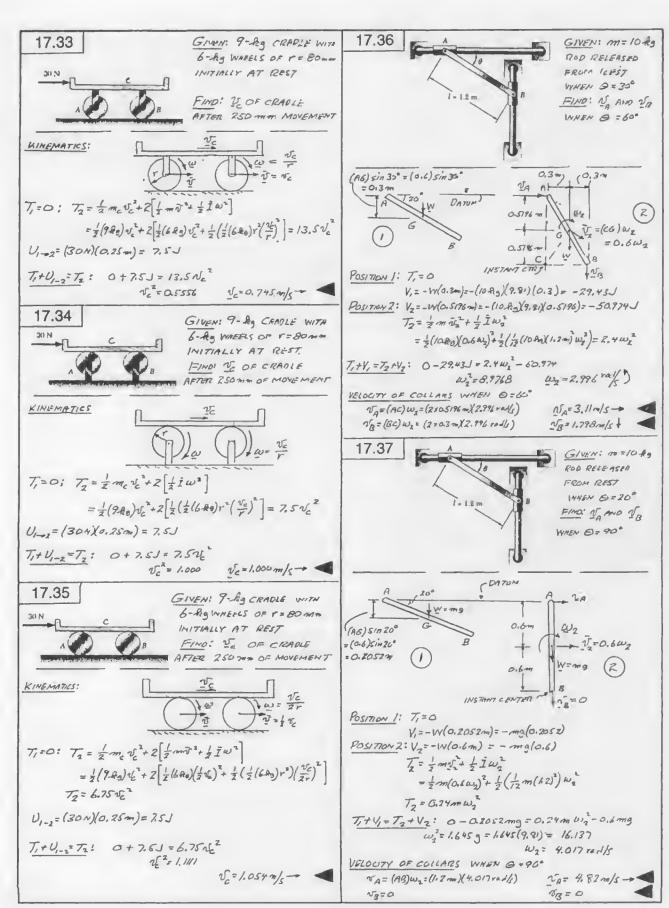
MORH ENERGY

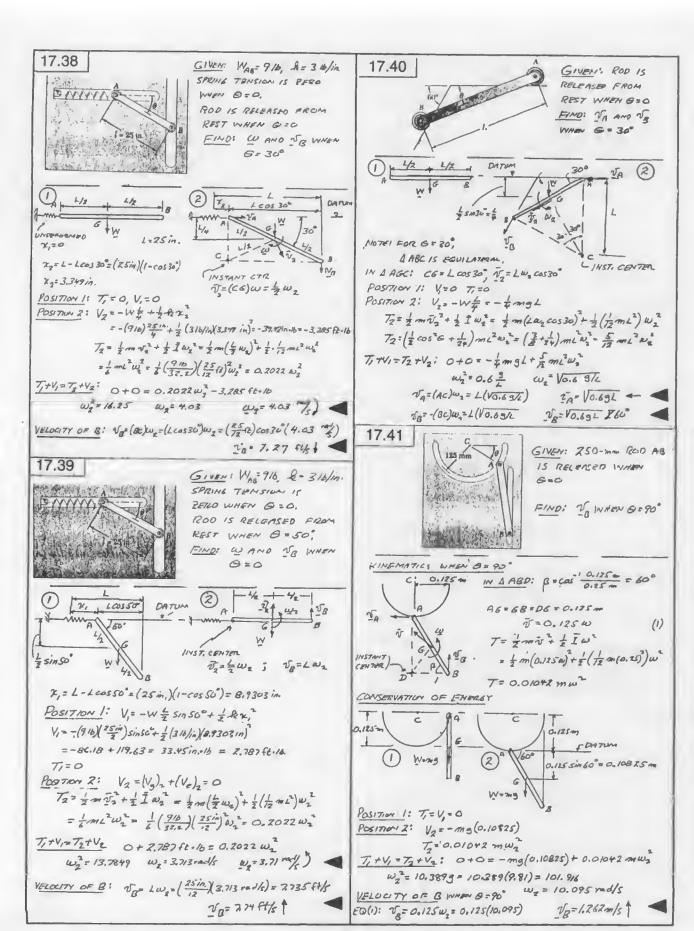
Ti+ U1-2 = Te

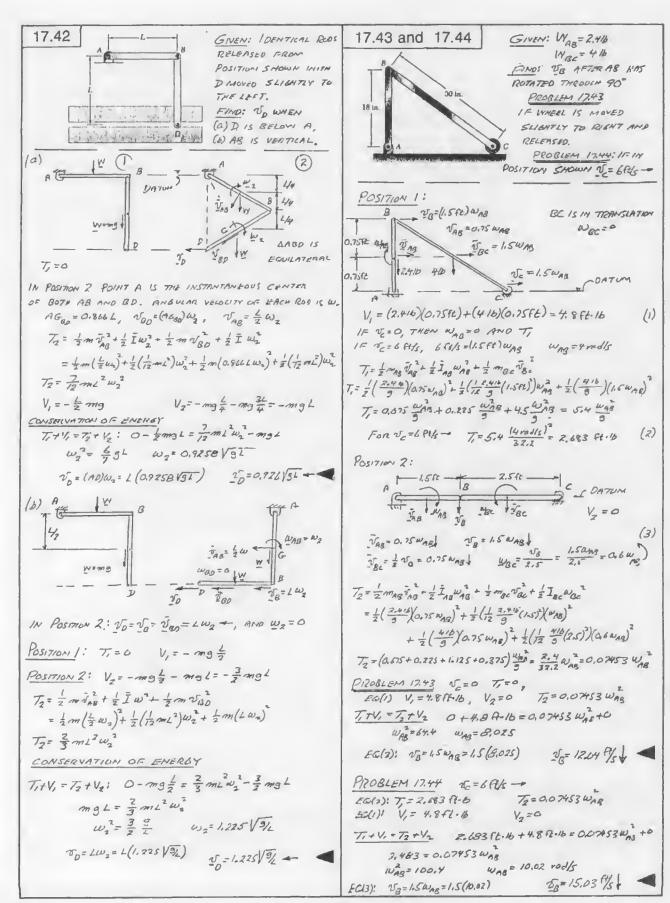
0+1.8394J = 21.625 NAR

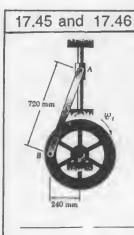
VAB = 0.08506 VAB = 0.2916 mg

1/18 = 292 mm/s ->









GIVEN: MAB = 4 fes

Marge = 16 fes

- Fe = 100 mm

PROBLEM 17.45?

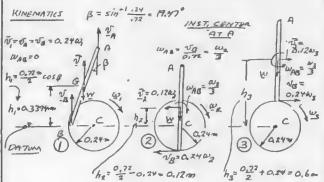
IF W = 60 rpm),

FIND: W WHEN B IS

DIRECTLY BELOW C

PROBLEM 17.46:

FIND: W, SO ANGLIAR
VALUTY IS THE SAME IN
FOSTIAN SHOWN AND WHEN
B IS DIRECTLY ABOVE C.



POTENTIAL ENERGY:

V=Wh,=mgh,=(4-20)(9.81)(0.337+nn)=13.318 -1 V2=Wh2=mgh2=(4-20)(9.81)(0.12nn)=4.709 -1 V3=Wh3=mgh3=(4-20)(9.81)(0.6nn)=23.5+4 -1

KINGTIC EMERGY

1 = I FLYWAREL = (16-Ra)(0.18 m) = 0.5/84 kg·m²

1 AB = 1 mabl = 12 (4-kg)(0.72 m) = 0.1718 kg·m²

T | 1 = 2 | = 25 = 16 Figure = 1 (4)(0.72 m)

 $T_{i} = \frac{1}{2} \bar{L} \omega_{i}^{2} + \frac{1}{2} m_{AB} \bar{V}_{i}^{2} = \frac{1}{2} (0.5184) \omega_{i}^{2} + \frac{1}{2} (4) (0.24\omega_{i}^{2})$ $= 0.2592 \omega_{i}^{2} + 0.1152 \omega_{i}^{2} \qquad T_{i} = 0.3744 \omega_{i}^{2}$

 $\begin{aligned} T_2 &= \frac{1}{2} \int_{\mathcal{L}} \omega_2^2 + \frac{1}{2} m_{AB} \tilde{v}_2^2 + \frac{1}{2} \tilde{I}_{AB} \omega_{AB}^2 \\ &= \frac{1}{2} (0.6784) \omega_2^2 + \frac{1}{2} (4) (0.12 \omega_2)^2 + \frac{1}{2} (0.1728) (\frac{\omega_2}{3})^2 \end{aligned}$

 $= 0.2592 \omega_{2}^{2} + 0.0288 \omega_{2}^{2} + 0.0088 \omega_{2}^{2} \qquad 7_{2} = 0.27% \omega_{2}$

 $T_3 = SAME$ COEFFICIENT AS T_2 : $T_3 = 0.2976 \, \omega_3$

PROBLEM 17.45: POSTION 1 TO POSITION 2

W, = 60 rpm (30) = 217 rad/s

 $7, + V_1 = 72 + V_2$: 0.374 + $\omega_1^2 + 13.318J = 0.2976 \omega_2^2 + 4.769J$ 0.3744(21) + 13.318 = 0.2916 $\omega_2^2 + 4.709$

0.2976 $w_{2}^{2} = 2339$ $w_{2}^{2} = 78.60$ rad t

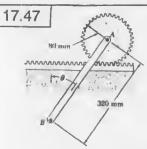
W2 = 8,865 rods (() W2 = 847 rpm)

PROBLEM 17.46: POSITION 1 TO POSITION 3 WITH W, = W,

 $T_1+V_1=T_3+V_3$: 0.3744 $\omega_1^2+13.318J=0.2978$ $\omega_2^2+23.544J$ 0.3744 $\omega_1^2+13.318=0.2978$ $\omega_2^2+23.544$

0.0768 w, = 10,226 w, = 133.2

w,= 11.54 rads (60) W,= 110.272)



GIVEN:

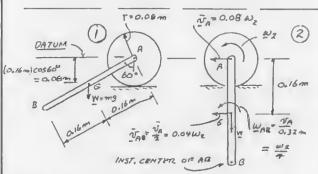
5-Ag GEAR, R=60 mm

4-Ag ROD AB

SYSTEM IS RELEASED

FROM REST WHEN G=60°

FIND:



POSITION 1: T,= 0

V,=-W(0.08m)=-(429)(9.81)(0.08)=-3.139.

POSITION 2: $T_2 = \frac{1}{2} m_A \tilde{\tau}_A^2 + \frac{1}{2} \tilde{I}_A w_2^2 + \frac{1}{2} m_{AB} \tilde{v}_A^2 + \frac{1}{2} \tilde{I}_{AB} w_{AB}$ $T_2 = \frac{1}{2} (5 \pi_3 (0.08 w_2)^2 + \frac{1}{2} ((5 \pi_3 (0.08)^2) w_2^2)$

+ 1/2 (4 feg) (0,040) + 1/2 (1/2 (4 ft g) (0,22 m)) (\(\omega 2 \)

 $7_2 = 0.06 \omega_2^2 + 0.009 \omega_2^2 + 0.0032 \omega_2^2 + 0.00107 \omega_2^2 = 0.02927 \omega_2^2$ $V_2 = -W(0.16m) = -(4.8g)(9.8)(0.16) = -6.278 J$

7,+V,=72+V2: 0-21391 =0.02927 w2-6.2781

W2= 107.26 W2= 10.357 rad/s VELOUTY OF A1 π= 0.08 W2= 0.06(10.357) = 0.829 m/s 1 = 829 mm/s 4

17.48 30 mm

GIVEN:

WA = 72.5 HZ

MOTOR

DEVELOPS 3kW

EIND:

(a) MA

(b) MA

WA WA = YBWB: (0.03m) (45 Trad/s) = (0.180m) WB

WB = 7.5 Trad/s

(a) PULLEY A: POWER = MA WA

3000 IN = MA (45 T rad/s)

Ma = 21.2 N.m

(6) PULLEY B: POWER = MB WB 3000 W = MB (7.577 rad/s) MB = 127.3 Nom 17.49

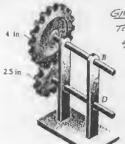
GIVEIL: MAXIMUM COUPLE THAT CAN BE APPLIED TO A SHAFT IS 15.5 hip in. FIND: MAXIMUM HOUSEPOWER THAT CAN BE TRANSMITTED AT (a) ISUTA, (b) 480WF.

M= 15.5 kg.in. = 1.2917 kip. ft = 1291.7 16. ft (a) w= 180 rpm (27) = 671 rod/s POWER = MW = (1291,7 16. Ft)(6# rox 6) = 24,348 12.13 Hurse Power = $\frac{24,348}{550} = 44.3 bp$

(b) W=480 rpm (2x)=1671 rad/s POWER = MW = (1291.716.12)(1671 YEN/s) = 64930 1116

HORSEPOWER = 64930 = 1/8.1 hp

17.50



GIVEN! MOTOR ATTACKED TO CHAFT AB DEVELOPES 4.5 hp WHEN WAR 720 you

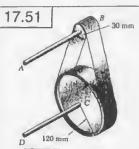
> FWO: MACHITURE OF COUNT EXTENSED ON (A SIMPT AB (d) SHALT CO

(a) SHAFT AB: WAS = 720 pm (211) = 75.398 rad/s POWER = 4.5 hp (550 R-B/E) = 2475 12.10/s

POWER = MAD WAR; 2475 FE-16/5 . MAR (75.398 mals) MAB= 32.826 16.50 MA= 32.8 16.50

(1) SMART CS: ALG = 50 WAS = 410. (75.395) = 120.64 rod/s

POWER = MEDWED: 2475 St. 4/5 = Med (120,64 rad/s) M= 20.5 16. 15



GIVEN: 2.4 FW TO BE TRANSMITTED FROM A TO D ALLOWABLE COUPLES ARE MAB = 25 N.m Mc0 = 80 N.71 FIND: RESURGE MINIMYM SPEED OF SMART AB

SHAFT AB: POWER = MAS WAS 2400 VY = (25 N. m) WAR

WARE 96 radts

SMAFT CO: POWER = MED WED 4 = 30 rodf 2400 W = (80 N.m) Wm

For up= 30rolls, Way= To up= (20 == |30rolls) WAR = 120 rod/s

WE CHOOSE INE LARLER WAS: WAS (12000 45) (60) WAB = 1146 rpm

17.52

GIVEN: 30-Rg ROTOR WITH h= 200 mm CORST TO REST IN S.3 AVA FROM INITIAL ANGULAR VELOCITY OF 3600 PPM. FIND: MAGNITUGE OF COURT DUE TO FRICTION

I=mh = (30 to)(0.2m)= 12 fig. m2 W = 3600 pm (27)=377 ralls

SYST. MUMENTA, + SYST. EXT. IMP = SYST. MOMENTA, +) MOMENTS ABOUT A: IW, -ME = 0 (1.2 kg-m3 × 377 ran/s) - M (5.3 min x 505) = 0

M= 1.423 N.m

17.53

GIVENI 4000-16 PLYWHEEL WITH & 2711 COASTS TO REST FROM ANGULAR VELOCITY OF 450 YPM. FRICTION COUPLE IS GF MAGNITUDE 125 W. NO. FIND: TIME REQUIRED TO CONST TO REST

I=m-R= (4000/6) (27 in) = 628.88 16.ft.5 W,= 450 -pm (21) = 47.125 rad/s M=12516.in. = 10.417 16. 8

SYST. MOVENTA, + SYST. FAT. IMP, -2 SYST. MOMENTA, +) MOMENTS ABOUT A: IW, - ME = 0 t: m = (612,00 10.05) (41.125 rolls) = 2845 s 10 417 16. St

£=20455 (min

£= 47.4 mm.



GIVEN: WA = 816, FA = 3in., FB = 4.5 in. DISKS OF SAME MATERIAL AND THICKNESS. M=2010.in., W,=0 FIND: TIME UNTIL WO = 960 YPM

W8 = (10) 2WA = (451) (810) = 1816 I = I + I = = 1 816 (2 R) + 1 1816 (45 ft) = 00470716. Ft.5 (W2= 960 pm (2m) = 100.53 rod/s, M=2016.in = 1.667 10. Ft

$$\begin{array}{c}
\overbrace{j\omega_{i}=0} \\
\overbrace{c}
\end{array} +
\begin{array}{c}
\overbrace{c}
\end{array} =
\begin{array}{c}
\overbrace{j\omega_{2}}
\end{array}$$

SYST. MOMENTA, + SYST. EXT. IMP, = SYST. MOMENTA,

+) MOMENTS ABOUT C: O+ ME = IW, t= Iw = (0.04707 16- 9.5 / 100.53 rod/s) 1.667 16.5t

t = 2.839 s

£=2.84 5



GIVEN: ma. 3. R. Ya = 100 mm, VE = 125 mm. DISKS OF SAME MATERIAL AND THICKNESS. W, = 200 rpm, W2 = 800 rpm L1-2= 35. FIND: MAGNITUDE OF COURF M

mg= (To)2 mg = (125 mm 2 3 Ag = 4.6875 Ag

I=I+I== 1/3 Rg/(0.1 m)+ 1/2 (4.6875 Rg/(0.125m)=0.05/62 Rg.m2 w, = 200 pm (20) = 20.944 rolls; wz=800 pm (20) = 83.76 rolf

$$\begin{pmatrix}
\hat{I}_{\omega_1} \\
\hat{G}
\end{pmatrix} + \begin{pmatrix}
\hat{G} \\
\hat{G}
\end{pmatrix} = \begin{pmatrix}
\hat{I}_{\omega_2} \\
\hat{G}
\end{pmatrix}$$

SYST. MONONIM, + SYST. EXT. IMP = SYST. MONANTAZ +) MOMENTS ABOUT G: IN, +ME = IW.

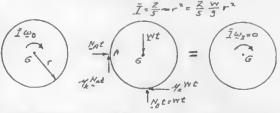
M = 1 (W2-W1) = 0.05/62 fg.m2 (83.776 rolls - 20.944 rolls)

M=1.081 N. 7

17.56



GIVEN: SPHERE OF WEIGHT W 1/4 " COEF. OF KINETIC FRICTION FIND: EXPRESSION FUR TIME REGULETO FOR STUBLE TO



SYST. MOMENTA, + SYST. EAT. IMP, - + SYST. MOMENTA (1) +4 COMPONENTS: O + Nat + 4, Nat - WE = 0 (2)

+x components. O+NAT -4 NGT =0

(3) FROM EG(2): NA= 4/ NB SUBSTITUTE INTO GOLD: Not +1/2 (4/2 No)t -WE

NB = 1 /+ 4/2 IN

EG(2):

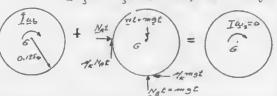
+) MONENTS ABOUT 6: I W. - (7, Mat) r - (4, Nat) r = 0 2 W rab - 42 rwt - 42 rwt = 0 2 V W - 4x+4/2 t=0

17.57



GIVEN: m = 3Ag, r=125mm W. = 90 rad/s 4x = 0.10 FIND. TIME RECVINED FUR STARRE TO COME TO REST

1=2 mr2= = (240(0.125m)= 18.75 110 Agon2



=YST. MONENTA, + SYST. EXT. IMP, = 575T. MCMENTA2

tyl components: O+Not -MANOR - met = c 13 COMPONENTS: U-MAL - 4/2 NEL = 0

=0(2): NA = 4x NB

FO(1): Not - 4/2 (42Ng)t - mgt = 0

NA = -1, NA = 0.1(29.129N) = 2.9/59 N

+) MOMENT AROUT 6: IN- - M. Natir- (MINBELT = C

E = Im, (18.75 x 10-3 kg.m. (90 m 1/s)
(0.10/0.175 m) (29.139 N + 2.9139 N) 2 = 4.212 5 1 = 4.215

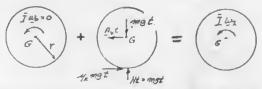
17.58 and 17.59



GIVENI DISK AT REST PLACED IN CONTACT WITH BELT. COFF. OF KIMETH FRICTIONS ME.

FIND: TIME REQUIRED FOR DISE TO REACH CONSTANT W. PROGLEM 1758: IN TERMS OF 2, 9, AND Th PROBLEM 17.59: FOR T= 3in. .W= 616, V= 50 (4/5, 4/k = 2.20.

I= +mr2 Nomg



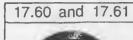
SYST. MONENA, + SYST. EXT. IMP, 2 = SYST. MOMENTA,

+) MOMENTS ABOUT 6: 0+ (-1/2 mgt)r = Iw, (1) FINAL AMEWAR VELOCITY: 5= (W, ; W2= 7/r

PROBLEM 17.58:

HOTE: RESULT IS INCOMMOUNT OF IN AND T. PROBLEM 17.59: DATA: 5=50 ft/s, 4/x = 0.20

> 2 = 50 Fels 20.2013 £=3.885



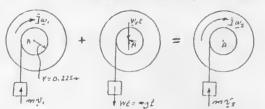
GIVENI 350- & FLYWHEEL OF £= 600 mm. W, = 100 Ypm] WHEN

PODIN IS CHOOLF MIND SYSTEM

FRORIEM 17.80: FINDS TIME REQUEED FOR SYSTEM TO COPUS

PROFIEM 17.11: FIND TIME ב יורין יוני ב בעני בוניים

7=mk=(350ke/0.6m)= 126. From



SYST, MOMENTA, + SYST. EXT. IMP = SYST. MOMENTA;

+ 2 MOMENTS ABOUT A: mare + IN-mate = mage + Ing EUSSDYUTE: TIFFW, ANDTERUS. (mr2+1) w, -matr = (mr2+1) wa

 $t = \frac{6.075 \cdot 1/26}{244.67} / \omega_1, \quad \omega_2) \qquad t = 0.42844 (\omega_1 - \omega_2) \qquad (1)$ PROBLEM 17.60:

(U=1001/m(20)=10.472 10 1/s EG(1): t=0.49864(10.472 rod/e)

t=5.225

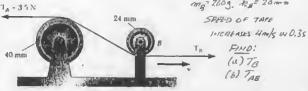
PROGLEM 17.61: (0) = 100 per (20) = 10.472 to 1/3 Wg = 407F= (=)= 41.189 rad/

EG(1): t=0.49864(10.472-4.189)

t= 3135

17.62

GIVEN: MA=6009, -RA=32 mm mg= 260g, kg= 20mm SPET-D OF TAPE



In=" = (0.6 Bo (0.052m) = 614.4 x10 6 82g-m2 IB= m3 - R8 = (0.26 ha/0.020m)= 104 × 10-6 fog m DRUM A: ASSUME W, TO INEN WE TO GOVE = 100 rolls



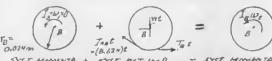
+] MOMENTS ABOUT A: C) -3,5 t(0,04m) + TAB t(0,04m) = In W2 t=0.351 -3.5(0.75×0.04)+Tm (0.35×0.04)=(614.4110 (100rods) -0.042 + 0.012 TAB = 0.06144 TAB - 8.62 H

(CONTINUED)

17.62 continued

DRUM B WE RECALL: TAS 8.62N

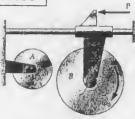
W2 = AV - 4 11/5 = 166.67 may/5



SYST. MOMENTA, + SYST. PAT. IMP 1-2 = SYST. MOMENTA, +) MOMENTS ABOUT B: 0 - Tatr - Tastr = I W. 2=0,35: To(0.55 Y0.054m)-(8.624)(0.35 X0.024m) (104510 fg-m) X166.67 m)

TR=11.03 N

17.63



GIVEN: DISK A IS AT REST WHEN DICKS A AND 3 ARE BROUGHT INTO CONTACT

SHOW THAT FINAL WO DEPENOS ON ONLY Ub AND min

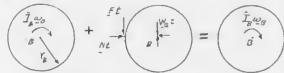
DISK A. (WA)000

$$\left(\begin{array}{c}
\tilde{I}_{a}(\omega_{a}) & \tau & \bullet \\
\uparrow & \uparrow_{a}
\end{array}\right) + \left(\begin{array}{c}
\downarrow & \downarrow \\
\uparrow & \uparrow \\
\uparrow & \uparrow
\end{array}\right) \frac{N\epsilon}{F\epsilon} = \left(\begin{array}{c}
\tilde{I}_{a} & \omega_{a} \\
\uparrow & \bullet
\end{array}\right)$$

SYST. MONENTA, + EYST. MOMENTA

+) MOMBYTS ABOUT A: O+(FE) " = IA WA Ft = Inwa (1)

DYSK B: (0) 8 = NO



+) MONENTS ALCOUT 2: ING (FE) " = IONS SUBSTITUTE FOR FE FROM EG(1)

I wo - I wa to = I was (2)

FOR FINAL ANGULAR VELOCITIES: VAR = TBW : Wa TO WA

$$\mathcal{L}_{\mathcal{C}(2)} = \tilde{I}_{\mathcal{B}} \omega_{0} - \tilde{I}_{A} \omega_{\mathcal{B}} \left(\frac{v_{\mathcal{B}}}{v_{A}}\right)^{2} = \tilde{I}_{\mathcal{B}} \omega_{\mathcal{B}}$$

$$\omega_{\mathcal{B}}^{2} = \frac{\omega_{0}}{1 + \frac{\tilde{I}_{A}}{\tilde{I}_{\mathcal{B}}} \left(\frac{r_{0}}{r_{A}}\right)^{2}}$$
(3)

BUT FOR UNIFORM DISUS: \\ \frac{\frac{1}{r_A}}{\tau_B} = \frac{\frac{1}{r_B}r_A^2}{\tau_Br_B^2} = \frac{m_A}{m_B} \left(\frac{r_A}{r_B} \right)^2

SUBTITUE INTO 16(1):

THUS, WB DEVENOS ON ONLY WE AND MA

17.64 P

GIVEN: WE 7.50, 1 = 6 in WE 104, 18 = 8 in We = 900 1pm

FIND: (a) FINAL WA AND WB

(b) I WRUSE OF FRICTION
FORES EXERTED ON DISK A.

$$\begin{split} \tilde{I}_{a} &= \frac{1}{2} \frac{W_{a}}{9} r_{a}^{-2} \cdot \frac{1}{2} \frac{7.50}{9} \left(\frac{2}{12} r_{c}\right)^{2} \cdot \frac{7.5}{89} \\ \tilde{I}_{a} &= \frac{1}{2} \frac{W_{a}}{9} r_{a}^{-2} \cdot \frac{1}{2} \frac{1016}{9} \left(\frac{8}{12} r_{c}\right)^{2} \cdot \frac{20}{74} \end{split}$$

DISK A:

In(an) = c

A



= (1/4/2)

4) MOMENTS ABOUT A: $0+(Ft)r_0=\tilde{I}u_0$ $(Ft)(0.5 ft)=\frac{7.5}{4.9}u_0$ $Ft=\frac{7.5}{4.9}u_0$ (1)

DISK B: (Wg) = W = 900 YPM (2T) = 30TT nad/s







SYST. MOMENTA, + SYST. EXT. IMP_2 = SYST MOMENTA_2 +] MOMENTS ABOUT E: $\overline{I}_{g}(w_{g})_{o}$ - $(FE)_{g}$ = \overline{I}_{g} a_{g} $\frac{20}{99} \cdot 30T$ - $(FE)_{13}^{2}(e)$ = $\frac{20}{15}$ a_{g}

SURSTITUTE FROM EQ(1):

 $\frac{20}{99} \cdot 20\pi - \left(\frac{25}{49}\omega_{p}\right)\left(\frac{e}{12}\right) = \frac{20}{99}\omega_{g}$ $30\pi - 0.5625\omega_{p} = \omega_{g} \tag{2}$

FIMIL VELOCITIES OCCUR WHEN:

 $V_A \omega_A = V_B \omega_Q$; $\omega_Q = \frac{V_A}{V_Q} \omega_A = \frac{\zeta_{in}}{\theta_{in}} \omega_A = 0.75 \omega_A$ (5)

SUBSTITUTE FOR WO FROM (2) INTO (3)

3071 - 0.5625 NA = 0.75 WA

3017 = 1.3125 WA WA = 71.807 mody WA = 71.807 mod/s (60) WA = 685.7 MM

EQ(3). WB = 0.75 NA = 0.75 (185.744) WB = 574.3 YPM

IMPULSE OF FL EXERTED ON CLER A.

EQ(1) $Ft = \frac{7.5}{49} \omega_A = \frac{35}{4(32.7)} (71.807 \text{ rad/s})$ Ft = 4.18 lb.5

W= 686 4PM 5)
WB= 574 412]

17.65



SHOW THAT SYSTEM

OF MOMENTA IS

EQUINALENT TO A

SHOLE VECTOR AND

EVILUES THE DISTANCE FROM

OF TO THE LINE OF ACTION

OF THE VECTOR IN TERMS

OF R. T. AND W.

+) MOMENTS ABOUT 6 $\overline{I} w = from \overline{w} d$ $d = \overline{I} w = \frac{mR}{mV}$ $d = \frac{R^2}{v}$

17.66 mro

SHOW THAT SYSTEM OF MOMENTA IS EQUIVALENT TO MIT W LOCATED AT P WASTE EPE \$ P = \$ 1/7

 $\frac{1}{2}\omega = \frac{m\tilde{v} \cdot m\tilde{v} \cdot \omega}{\int_{0}^{\infty} \frac{1}{2}\omega \cdot (m\tilde{v} \cdot \omega) \cdot \delta} = \frac{m\tilde{v} \cdot m\tilde{v} \cdot \omega}{\int_{0}^{\infty} \frac{1}{2}\omega \cdot (m\tilde{v} \cdot \omega) \cdot \delta} = \frac{m\tilde{v} \cdot m\tilde{v} \cdot \omega}{\int_{0}^{\infty} \frac{1}{2}\omega \cdot \omega} = \frac{m\tilde{k}^{2}}{\int_{0}^{\infty} \frac{1}{2}$

17.67 FOR A RIGID SLAS IN PLANE MOTION,

SHOW THAT HO IS ESUAL TO IAW, IF

AND OHIT IA (a) A IS THE MASS CENTRE, (b) A IS THE

INSTANTANEOUS CENTER SE ROTATION, (c) IN IS

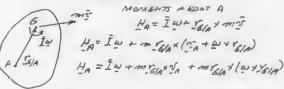
DIRECTED ALONG SINE AB.

A TEIN

FOR GENERAL PLANE MOTION

T = VA + JA = VA + WX VGIA

SYSTEM OF MONEHTA-



SINCE WITCH THE TRIPLE VECTOR PRODUCT

CAN BE WRITTEN: YOUN (W+YOIN) = YOIN W

TAOS HA = I w + myoIn + M oin w

BY PARALLEL-BAIS THE WEAR! $I_a = I + m r_{dia}^2$ WE NOW HAVE $H_a = I_a w + m r_{dia} y = I_a$.: $H_a = I_a w$, ONLY WHEN $V_{dia} + v_{dia}^2 = 0$ (a) $r_{dia} = c$: A connider with $G_{dia} = 0$ (b) $V_a = 0$: A is instant. Center

(c) Soly and of All COUNTY: VA IS DIRECTED ALONG AE.

17.68



GIVEN: MAPUSING FONCE F IS

RYPLIED TO SLAG.

SHOW THAT: (0) INST, CENTER

IS AT C AND GC = A GF.

16) IF F WENT MANED AT C

THEN P IS THE INST, CENTER

At TIME OF MELICATION OF F AT THE CONTER







SYST. MONEYITA, + SYST. EXT., $Imp_{rel} = SYST$, $MONEYITA_2$ +/COMPONENTS: $FOt = m\tau_0$; $\tau_0 = \frac{F\Delta t}{m}$ (1)

+) MOMENTS ABOUT G: $(FAt)(GP) = \bar{I}\omega_0$ $\omega_2 = \frac{FBt}{L}(GP) = \frac{FBt}{mR^2}(GF) \qquad (2)$



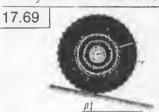
KINEMATICS: THE INSTANTANEOUS CONTRI MUST BE LOCATED ON A LINE L TO \tilde{V}_2 , THAT IS, ON GP. ALSO, $\tilde{V}_2 = (66) \omega_2$ GC= $\frac{\tilde{V}_2}{\omega_2}$

 $SC = \frac{FAL}{\frac{FAL}{M}} GC = \frac{R^2}{6P}$



WE NOW ASSUME THAT F IS
ARVED SO THAT THE NEW CENTER
OF PETCUESION P' IS LOCATED AT C.
FROM PART Q, WE NOTE
THAT NEW INST, CENTER WILL BE $LOCATED AT C' WHERE
GC' = \frac{k^2}{6P^2} = \frac{R^2}{GC} = \frac{k^2}{k^2/2} = 6P$

THUS, NEW INSTANTANTOUS CENTER IS LOCATED AT P



GIVEN; & = RADIUS OF GYRATION V, = 0

FINO: (a) \$\vartheta_{2} AT TIME \$\vartheta\$

(b) Af REGURED

TO PREVENT SUPANI







SYST. MOMENTA, + SYST. BYT. INP = SYST. MOMENTA:
+) MOMENTS AROUT C: $(Wt sin\beta)r = \tilde{I}W_1 + m\tilde{V}_2r$ $mgt sin\beta = m-\tilde{v}_1w_2 + mr^2w_1$ $W_2 = \frac{vgt sin\beta}{r^2 + \tilde{h}_2^2}$ (1)

(CONTINUED)

17.69 continued

(a) \$\bar{\sigma}_2 = r \omega_2:

D= T2+A2 gt sing Tp

(W+ COMPONENTS: 1

Nt-mgt cosp

+) MOMENTS ABOUT 6: (FE) = 1 a2

 $F = \frac{I}{rt} \omega_2 = \frac{\pi k^2}{rt} \cdot \frac{r^2 gt}{r^2 + k^2} \sin \beta = \frac{k^2}{r^2 + k^2} mg \sin \beta$

45= F = 12 mg sing; 45= 12+12

17.70



GIVEN Y = 1.5 in.

WHEEL STARTS FROM REST
AND ROLLS WITHOUT SCIONS. $\bar{N}_2 = 6$ in. $f_2 = 7$ and $f_3 = 7$ and $f_4 = 7$ and $f_5 = 7$ and $f_6 = 7$ a

√1 = 4√2 0 F = √3 √2 = 5 √2 1/3°





SYST. MOMENTA, + SYST. EXT. IMP, = = SYST MOMENTA, +) MOMENTS ABOUT C: $mgt(rsing) = \hat{1}\omega_0 + m\bar{x}_2r$ $mgtrsing = dr. \hat{h}^2\omega_2 + dr. r^2\omega_2$ $gtrsing = (-\hat{k}^2+r^2)\omega_2$ (1)

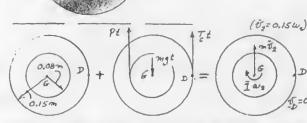
DATA: $r = \frac{1}{8}n$, $\bar{v}_2 = 6$ in s = 0.5 fels, t = 30s $\omega_2 = \frac{5}{r} = \frac{0.5}{\sqrt{6}} \frac{1}{6} \frac{1}{6} = 4 rod6$

 $-\frac{1}{2}$ = 2.79% $-\frac{1}{2}$ = 2.79% $-\frac{1}{2}$ = 2.79% $-\frac{1}{2}$ = 2.79 ft

17.71 P

GIVEN: m=3.kg, &=100 mm PULLEY IS 47 REST WHEN P=244 IS APPLIED TO B

FIND: (A) & AFTER 1.55 (U) TENSION IN COED (



SYST. MOMENTA, + $SYST EXT INP, _2 = SYST.$ MOMENTA 2 +) MOMENTS ABOUT D: Pt (0.08+0.15) - $mgt(0.15) = \overline{1}w_2 + m\overline{v}_1(0.15)$ $(2411)(1.55)(0.21) - (3.4g)(9.81)(1.55)(0.15) = (3.4g)(0.16)[w_2 + (3.4g)(0.15)[w_3]$ $1.6583 = (0.03+0.0615)(w_2; w_2: 17.008)$ rodt $\overline{v}_3 = (0.15)(17.008) = 2.551 m/s$ $\overline{v}_3 = 2.55 n/s$

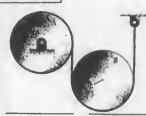
17.71 continued

WE NAVE FOUND J. - 2.55/m/st

the components: Pt + Tet - mgt = miz (24NX1.55) + Te(1.55) - (3kg)(9.81)U.55) = (3kg)(2.551mols) 36+1.57c - 44.195= 7.653 1.57 = 15.798

TE= 10.53 N

17.72



GIVEN: Two 14116 CTUNDERS OF RADIUS T= 5in. SYSTEM IS RELEASED FROM REST WHEN t=0. FIND: (a) No AT to 35. () TENSON IN BELT COMNECTING CYLINDERS



KINEMATICS CYCINDER R INSTANT. CENTER OF B IS AT C. Va=rwa Vo=V=zrws CHINOWI A

CYLINDER A:

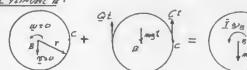


SYST, WOMENTA, + SYST, BIT. IMP, = SYST. MOMEN TA.

+) MOMBUTS ABOUT A: (QE) v = I WA (GDr= +mr2(2 mg)

at= mrwg 10

CYLINDER R:



t) MOMENTS ABOUT C: (mgt)r-(Ot)2r= Ing+migr mgtr-(Qt)(2+) = 1 mr2 +m(rwg)r 121 mgt-2(01) = 3-mrwg

SUBSTITUTE ECO (Ot) FROM(1) mgt-2(mray) = 3 mrug WR = 2 8#

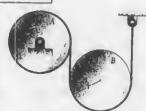
> BR= TNB; 7R = 39t

Ot=mrug; Qt=mr(2 9t) FG(1): Q= = = = = W

DATA: W= 1416 t= 35

(a) NB = = 391 = = (32.214/5-)(35) De= 27.6 1/5 1

(1) Q= = = = = = (14/b) = 4/b TENSION IN CONNECTING BELT = 4/6 17.73



GIVEN: TWO 14-16 CYUMBERS OF RADIUS Sin. HAMALLY WA= 30 rod/s) FIND: (a) TIME REGULED FOR WA TO SE REDUCED To Was 5 rods)

(6) TENSION IN BELT CONNECTING CYLINGERS

KINEMPTICS: CYLINGER B INSTATE CENTER OF B IS AT C. v= rws Vp = 5 = 25 WB

CYLINDER A:



SYST, MOMENTA, + SYST. FAT. IMP, -2 + I MAMENTS ABOUT A: I(Wal, - /GE) = I/Wal. (at) = 1 mr2 [(w), -(wi)] (QL) = 1 mr [2/WB), -2(WB), Qt= mr (ws), - (ws), (1)

CYLINDER E:



5'57. MOMENTA, + 57.57. EST. IMP, -2 = 5757. MOMENTA. +) MOMENTS ABOUT C:

 $\bar{I}(\omega_{e}) + m(\hat{r}_{e}) + \varphi t(x) - mgt) r = \bar{I}(\omega_{e}) + m(\hat{r}_{e}) r$ Suasmore I = + mr, (v.) = r(v.), AIN (v.) = r(wa);

Qt(2r)-(mgt)r=mr2/(wn)=(at), + + mr2/(wol_-(an))

2Qt -mgt = 5 mr ((NB)2-(WB)) (2) SUBSTITUTE FOR OF FROM (1):

2mr (wg), -(wa), -mgt = 3mr (wg), -(w4),

t= 7 / ((w8), -(w8))] (3)

SUBSTITUTE FOR & FROM (3) INTO (1)

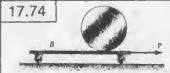
 $Q\left\{\frac{7}{2}\frac{r}{5}\left[\left(\omega_{B}\right),-\left(\omega_{B}\right)\right]=mr\left[\left(\omega_{B}\right),-\left(\omega_{B}\right)\right]$ Q= = = = = = = W 14)

DATA: (WA) = 30 rad/s - (WB) = 2(WA) = 15 rads (wa) = 5 rails - (wa) = 1 (wa) = 2.5 rails W= 1416, r= Sin. = 5/2 FL

(a) EG(3): 2. 7. (S/12 (2)) [15-od/ - 2.5rod/

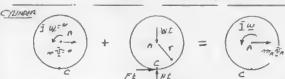
Z=0.566 5 t=0.5661 s

(b) PG(4): Q= = = (1410)= 4/15 TENSKA IN CONNECTED BELT = 4 16



GIVENI CYLINDERS ME BRO, Y= 240000 CARRIAGE 1 ma= 3-29 SYSTEM AT REST WHEN P = 10 N APPLIED FOR 1.25

Fino: (a) VB, (b) VA



SYST MOMENTA, + SYST. FAT. IND = SYST, HONONTA ? + JANUENTS ABOUT C: O=] W-MA VAY W= 22A (1)

I COMPONEITS: Ft = My T

CARRIAGE:

Pt-Ft= mg V8 + COMPONENTS:

$$Pt = m_A \tilde{v}_A = m_B v_B$$

$$Pt = m_A \tilde{v}_A + m_B v_B \qquad (2)$$

KWEMADES! ASSUME RULLING

$$\begin{array}{ccccc}
\overline{V}_{A} & \overline{V}_{B} = \overline{V}_{A} + ru & (3) \\
& \overline{V}_{B} & SUBSTITUTE FOLION FO (i) \\
& V_{B} = \overline{V}_{A} + r\left(\frac{2\overline{V}_{A}}{r}\right); & V_{B} = 3\overline{V}_{A} & (4) \\
& \overline{V}_{B} & Folion & pt = m_{A}\overline{V}_{A} + m_{B}(3\overline{V}_{A})
\end{array}$$

$$\hat{n}_{A} = \frac{Pt}{N_A + 3 m_B} \tag{S}$$

DATA: m= 8-kg, mg= 3-kg P=10N, E=1.25

$$V_A = \frac{(NN)(1.75)}{8R_0 + 3(3-R_0)} = \frac{12}{17} \text{ m/s}$$

$$\frac{FE(14)}{\sqrt{g}} = 3\sqrt{\frac{12}{17}} \, m/s = \frac{36}{17} \, m/s$$

$$\sqrt{g} = 2.12 \, m/s \rightarrow 0$$

17.75

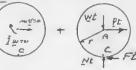


GIVEN: CYLINDER: M= Blog F= 240 mm CARRING: mg = 3 lo

EYSTEN AT REST LAWEN PERON APPLIED FOR 1.25

FIND: (a) 23, (b) 7A

CYLINDER



SYST. MOMENTA, + SYST. FYT. IMP, = SYST MOMENTA? +2 MOMENTS ABOUT A: (FE) FE IW (Ft) r= = mr2 w

$$Ft = \frac{1}{2} m_T u \qquad (1)$$

$$PL = Et = m_s \tilde{v}. \qquad (2)$$

- COMPONENTS: PE-FE- MAGA (3) +) MUMBITS ABOUT C: PL= IW+ MA TAT

CARRIAG:

ASSUME ROLLING KINEMATICS

SURSHINZ: FOI) - FOI): Pt - + mrw = maya

$$\bar{V}_{A} = \frac{PC}{m_{A}} - \frac{1}{2}r\omega \tag{6}$$

SURSTITUTE PC(1) - EC(5): (7) +marwomg(VA-ru)

SUBSTITUTE EG(6) - EG(7):

$$\frac{1}{2}m_{A}r\omega = m_{B}\left(\frac{Pt}{m_{A}} - \frac{1}{2}\gamma\omega - r\omega\right)$$

$$\left(\frac{1}{2}m_{A}r + \frac{2}{2}m_{B}r\right)\omega = \frac{m_{B}}{m_{A}}Pt$$

$$\omega = \frac{2Pt}{r}\left(\frac{m_{B}}{m_{A}}\right)\frac{1}{m_{A}+3m_{B}}$$
(8)

DATA: ma = & 29, ma = 3 kg P=10N, E=1.25, Y=0.24m

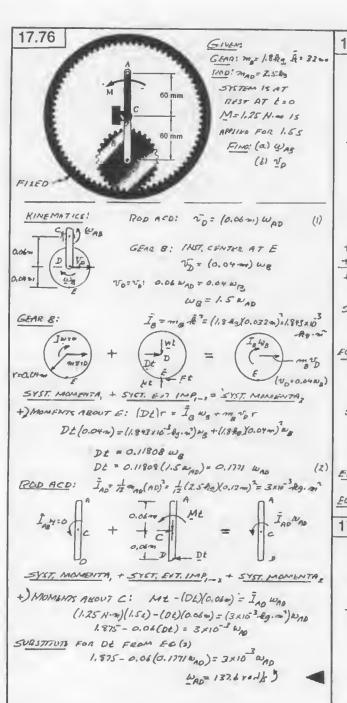
$$EG(\theta)$$
: $\omega = \frac{2(10NX1.25)}{0.124m} \cdot \frac{2.89}{8.89} \cdot \frac{1}{8.89 + 3(2.80)}$

$$\omega = \frac{37.5}{100} \text{ rad/s}$$

$$EO(6)$$
: $\tilde{V}_{A} = \frac{Pt}{m_{A}} - \frac{1}{2} r \omega = \frac{(ION)(L25)}{6 Rg} - \frac{1}{2} (0.24 m) \frac{37.5}{17} redt$

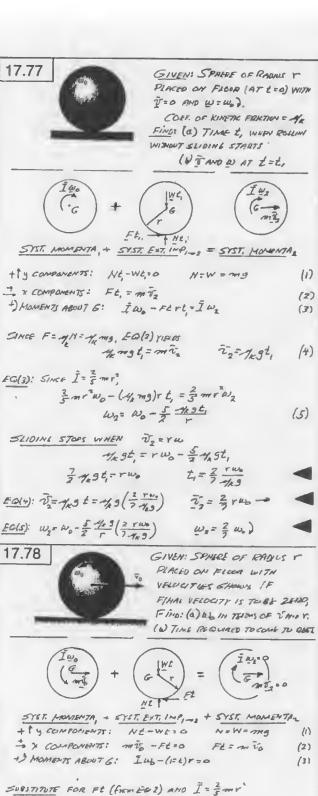
$$\tilde{V}_{A} = \frac{Pt}{m_{A}} - \frac{1}{2} r \omega = \frac{(ION)(L25)}{6 Rg} - \frac{1}{2} (0.24 m) \frac{37.5}{17} redt$$

UP 0,706 m/s ->



EG.(1): Vp= (0.06m) w = (0.06m)(137.6 radk)

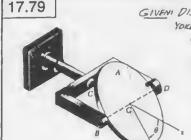
7 = 8.26 m/s -



SUBSTITUTE FOR Ft (fram Es 2) AND I = = mr

$$\frac{EG.(2)}{5}$$
: $\frac{2}{5}mr^2w_0 - (mR_0)v = 0$
 $w_0 = \frac{5}{2} \cdot \frac{\tilde{V}_0}{r}$

$$\frac{FG(1)}{F}: t = \frac{m\vec{v}_0}{F} = \frac{m\vec{v}_0}{\eta_{K}mg}; t = \frac{\vec{v}_0}{\eta_{K}g}$$



GIVENI DISK: W= Z.SB, r=4in. YOKE: Wy=1.510, R= 3 in. WHEN 6=0, W= 120173

FIND: WY WHEN 5 = 90°

ROTATION ABOUT & ASIS:



SYST. MOMENTA, + SYST ENT. IMP = SYST. MUMBITAZ WE HAVE CONSERVATION OF AMERICAS MOMBRITURA ABOUT THE 2 ANG. Iw= Iw, (1)

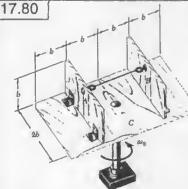
$$\begin{split} \tilde{I}_{j} &= \tilde{I}_{j000} + \tilde{I}_{DSS_{j},6=0} \pm m_{\gamma} \tilde{R} + \frac{1}{4} m_{\beta} r^{2} \\ &= \frac{h^{5/16}}{9} \left(\frac{3}{12} f_{2} \right)^{2} + \frac{1}{4} \frac{2.50}{5} \left(\frac{4}{12} in \right)^{2} = \\ &= \frac{0.07375}{5} + \frac{0.06944}{9} = 0.16319 \frac{1}{9} \end{split}$$

$$\begin{split} \bar{I}_{2} &= \bar{I}_{YOUg} + \bar{I}_{DSU,6+90} = m_{Y} + \bar{f}_{L} + \frac{1}{2} m_{D}^{T} \\ &= \frac{1.516}{3} \left(\frac{3}{12} f_{L}\right) + \frac{1}{2} \frac{2.516}{3} \left(\frac{4}{12} m_{L}\right) = 0.23244 \frac{1}{3} \end{split}$$

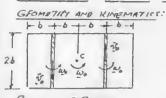
EG(1): 0.16319 + (120 pm) = 0.2324 & 100

W2=84.17 -pm

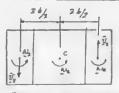
W= 84.2 ym



GIVEN: PANELS AND PLATE ARE MADE OF SAME MAZERIAL AND ARE OF SAME THICKNESS. IN THE POSITION SHOWN ANGUAL VELOCITY = WE FIND: AFTEK WIRE BREAKS ANGVLAR VECOCITY KINEN PANELS HAVE COME TO REST



PAMELS IN UP PUSITION 150 = b w.



ABAINST PLATE

PANELS IN DOWN POSITION できるい。

LET P= MASS DENSITY, &= THICKNESS DIATE: mplok= et (20X40) = 8pt 8 I plate = 1/2 (8/26) (26)2+(46) = 1/2 PZ6 = 3026

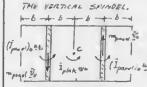
(CONTINUED)

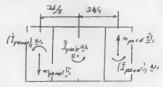
17.80 continued

EACH PANEL: OFFIRE Pt (6) SAJ = 2Pt 6" PHINEL IN UP POSTIONS (Ipone!) = 1/2 (2pt &)(2b) = 8/2 pt b = 3/2 pt 64

PRINTE IN DOWN POSITION (] proof) = 1/2 (2pebs) (62+(20) = 1/2 peb = 6 peb

WE HAVE CONSERVATION OF AMERICA, MULLIENTUM ALOUT





INITIAL MONIGHTA

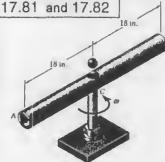
+) MONGATE ABOUT C:

FINAL MOMENTA

 $\bar{I}_{plain} \omega_{o} + 2 \left[\left(\bar{I}_{parello} \omega_{o} t^{-sp}_{proof} U_{o}(b) \right) = \bar{I}_{plain} \omega_{o} + 2 \left[\left(\bar{I}_{parello} \omega_{o}^{*} t^{*} + m_{f}^{*} \omega_{o} \sigma^{*} v_{o}^{*} \left(\frac{3b}{a} \right) \right]$ 40 pt 6 No + 2 [3 pt 6 No + 2 pt 6 (6 m) 6] = $\frac{40}{3}$ pt 6^{4} w₀ + $2\left[\frac{5}{6}(26^{4}\omega_{i} + 2i26^{2})^{\frac{2}{3}}bw_{0})(\frac{3}{2}b)\right]$

[40 + 3 + 4 | ped No = [40 + 9 + 9 | ped W,

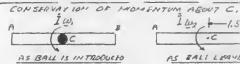
 $\frac{56}{3}\omega_{o} = 2+\omega_{i}$; $\omega_{i} = \frac{56}{3(24)}\omega_{o}$ $\omega_{i} = \frac{7}{9}\omega_{o}$

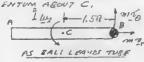


GIVEN: 4-16 TUBE AB INITIALLY W= grad/s BALLS INTROLUCED TO TUGE PROBLEM 17.82:

Fim: (0) W AS A 0.8-16 BALL LEAVES TUBE (b) as AS A SECOND O.S-15 GALL LEAVES TUSS. PRC81FM 17.83.

FIND: Q AS A SINGLE 1.6-16 BALL LEAVES TUBE.





) MUMENTS ABOUT C: IN = IN + m, 26 (1.5 ft) (1) I= /2 4/6 (3FU)= 3 No= (1.512)w.

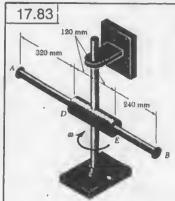
 $\frac{P_{ROBIEM}/7.82^{\circ}}{5}(a) \frac{F_{IRST}}{5} \frac{0.8 \cdot 10 \frac{F_{ALL}}{F_{ALL}}}{9}(b, Sw_{o})(l.SSF)$ W= 5 rads

AS FIRST BALL LEAVES TUBE: WESTERS

(b) SECOND O.B-16 BALL W, = 5 YANK Ea(1): 3(5 rods) = 3 W2+ 0.89 (1.5012 ×1.5(2) Wa = 3.125 rodb 15= (3+1.8) W2

AS SECONO BALL LEAVES TUBE: W=3,175121/3

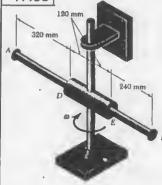
PROBLEM 17.83: A 1.6-16 BALL IS INTRODUCED, W. = 8 rolt 3 (Eradk) = 3 w + + 1.610 (1.54x) (+5) W2 = 3.636 rods 24 = (3+3.6) 42 AS 1.6.16 BALL LEAVES THE TUBE: W= 3.64 rod/s



GIVEN: 3-Ag ROD AB FOR CYLINDER DE: I = 0.025 Ra.m. IN POSITION SHOWN: w= 40 rod/s A10

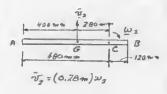
END B OF ROD IS MOUNTS TOWARD E AT 76 mm/s.

FIND: ANGULAR YELOCITY OF ASSEMBLY AS FNO B STRUKES CYLINIDERI AT E.



KINEMATICS AID GEOMETRY 15,= (0.04 m) W,= (0.04 m) 40 radk

V = 1.6 mg

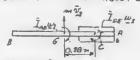


INMAL POSITION

FINAL POSITION

WE HAVE CONSTITUTION OF ANGULAR MOMENTUM ABOUT C.





+) MOMENTS ALOUT C: Ins 13 (3 Ray 0.9 m) = 0.16 kg. m Ina w, + m v, (0.0+0)+ Inaw, + Inaw, + m v, (0.780)+ I wa (allkg.m) (101016)+(3 20 X/16 m/s)0.04 m)+(0.025 29 m) (40 tolls)

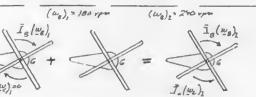
= (0.16 ftg.m2) wg r (3 Rg)(0.28 Wg)(0.28) + (0.025 ftg.m2) w. (6.4+0.192+100) = (0.16+0.2352+0.075) az

7.592= 0.4202 Wz : Wz=18.068 rad/s: W=18.07 rad/s

17.84

GIVEN: I = I = 650 ID. PEIS FACH BLADE WEIGHS 55 16 INITIAL ANGULAR VELOCITY OF CAR = ZERG

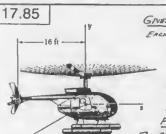
FINDS WE AS WOLADES IS INCREASED FROM 1801pm 70 240 rm



SYST. MOMENTA, + SYST. EXT. IMP, = SYST. MOMENTA. In 4 1 3516 (1412) = 446. + 10.18.5° (Oz) = (No) = ENO opm; [Uz] = (Uz) = 240 opm [NOTE: W IS OF SIADUR RELATIVE] +) MOMENTS ABOUT GI IB(W), + O= IB(WB) + Ic(Wc)2 (446.4 10.17 5) (150 = m) = (446.416.66.5) (65) + 240 =) + (650 10.17.5) (46),

(wc) = 26784 1096.4

(WCAB) = 24.4 rpm



GIVEN I = 650 16. FE. 5 EACH BLADE WEING 5516 WCAB = 1250 B

TAIL PROPERLER PREVENTS ROTATION OF CAB AS NO OF BLAOUS IS INCREASED FROM 180 Tpm TO 240 Tpm IN 125. FINO: FORCE EXERTED ST TAIL PROPERLOR AND FINAL TO.

 $\bar{I}_{a}(\underline{\psi}_{a}),$



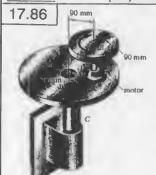
SYST. MONEYMA, + SYST. BUT. IMP, = = SYST. MOMBAITAS +) MOMENTS ABOUT 6: $\tilde{I}_a(\omega_a)$, + $Ft(165) = \tilde{I}(\omega_B)_2$ (1)

+ COMPLUIENTS: O+ Ft + m V2 (2) In=4 1 351/0 (14/2) = 446.4 16.56.52 m= mc+mg= 1/250 0+4(5516) = 45.65 16.5/A

(W2)=180 rpm 27 = 18.85 rad/s 3 (Ng)= 240 rpm 27 = 25.13 rad/s

ECULISE (446416.DSX10.05.0016)+(FIXICR)=(446910.A.5 × 25.13 radk) FE= 175.3 16.5

175.316.5 = (45.65 16.57/A) J; J=3.84 A/5 EG(2): Fon t= 125: Ft = F(120) = 175.3 16 5; F=14.61 1b



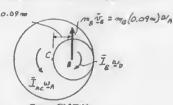
GIVEN: M= 4-29 In- 0.20 Rg. == 2 SYSTEM IS INITIALLY AT REST

FIND: WE AND WA WHEN SPEED OF MUTUI REACHE 360 1pm

I = = (4.A3)(0.08 m) I = 16.2 × 10 - Rg·m2

WE HAVE CONSERVATION OF ANGULAR MOMENTUM ABOUT SHAFT C





INITIAL POSITION

In w + mg v (000 -) - I wy -) MCMENTS ABOUT C: (0.20 29 -) W, + (4 29) (0.09 0, (0.09) - (16.2×10 29 m) W 0.2324 NA - 0.0162 WB + D WR = 14,346 WA (1)

WMOTOR = WATWA

360 rpm = WA + 14.346 WA

WA = 23, 46 rpm

Na= 23.5 rpm

EG(.) WR = 14.346(23.46)= 336.55 700

WB = 337 1/2



GIVEN: FOR 200- ~ RADIUS

ma: 5-Rg, R= 175 mm

INITIAL ANGULAR VELOCITY! Was 500ps

OISK: Mg=2-fig, Yg= ECIMM

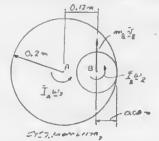
DISK PLACED, WITE NO VELOCITY,

OIL PLATFORM

FIND. FINAL ANGULAN VELOCITY

INE HAVE CONSERVATION OF AMELIAN MONIGATION AGINT SHAFF





+) MOMENTS ACOUT D' Iqui = Iqui + Iqui + mig(0.12+0) (1)

Iq = My R2 = (5-69/0.175-)2 = 0.153125 Ag. mi

Ig = Jame Iq2 = \$1369/0.0000) = 9.6×10 Ag. mi

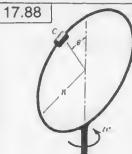
Ug = (0.172, 0.9)

EGG: (0.158175 Ag. 2) 0; = (0.153175 Agai) w, 1 (9.6810 hg.ni) w, + (3.89)0,17 m) w,

0.153125 01 = 0.20573 02

N= 0.7436 W= = 0.7428 (50 1pm)

102= 37.2 rpm



CIVEN 2-Reg COLLAR C

RING: Mm = 3-Reg

R = 250 mm

WHEN G=0, W,= 35 rod/s

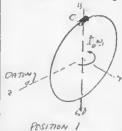
AND V_= 0

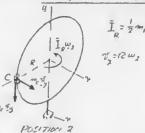
(4) VELOCITY OF

COLLAN QUATTUR TO RIVEL

WHEN 6 = 90°

INT HAVE CONSERVATION OF MIBULAR MOMENTUMA
PRONT VERTICAL Y AXIS AND CONSERVATION OF WHILEST





CONCLANATION OF ANGUAR MONTHERTUIN

MOMENTS AREALT & ALIS: $\bar{I}_{R}w_{1} = \bar{I}_{R}w_{2} + m_{e}V_{1}$ $\frac{1}{2}m_{e}R^{2}w_{1} = \frac{1}{2}m_{e}R^{2}w_{2} + m_{e}R^{2}w_{2}$ $m_{H}R^{2}w_{1} = (m_{R} + 2m_{e})R^{2}w_{2}$

 $\omega_2 = \frac{m_R}{m_R + 2m_c} \omega, \qquad (1)$

(CCN TIMUED)

17.88 continued

DATA: me=280, me= 280, 12=0,25m, W,=35 rods
(a)

EO(1) W2= 380 (35 rods) W2= 15 rods

(b) $\frac{1}{4}(3R_0)(0.75m)(35rod(s)^2 + (7.80)(9.81m)c^2)(0.75m)$ = $\left[\frac{1}{4}(3R_0) + \frac{1}{4}(2k_0)(0.25m)(15mb)^2 + \frac{1}{4}(2k_0)s_y^2\right]$

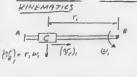
57.422 + 4.905 = 24.609 + 5 3 Vy = 37.76; Vy = 6.14 m/s

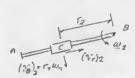


GIVEN: IN POSITION. SHOWN W, = 1.5 rad/s, (8,)=1.5 m/s

mc=8/2, FOR 100 AND SHOWS I8= 1.2 fig.m.

FIND: (a) MILLIAM DISTANCE
BETWEEN CAIDB, (b) COLRWPULLMAR ANGULAR VELLEITY





KINETKS: SINCE MONTH, OF ALL FORCET MEDUT & ARE ZERO, WE HAVE: $(H_0)_1 (H_0)_2$: $I_g w_1 + m_1(\tau_0)_1 r_1 = I_g w_2 + m_1(\tau_0)_2 r_2$ $(I_g + m_1 r_1^2) w_1 = (I_g + m_1 r_2^2) w_2 \qquad (1)$

[17AB.m.+ (BROXO.6m)] (1.5 ods) = [1.2 Ag.m.+ (6An)T]] W2 (2)

CONSERVATION OF ENERGY SINCE V.= Va. WE NAVE T, = T.

T, = \frac{1}{2} Ig \omega_1^2 + \frac{1}{2} mc (\frac{1}{6})^2 + \frac{1}{2} mc (-r)^2

= \frac{1}{2} (1.2 Ag.m.) (1.5 rod/s)^2 + \frac{1}{2} (8 Aglosm) (1.5 rod/s)^2 + \frac{1}{2} (8 Aglosm) (1.5 rod/s)^2

T, = 13.57 J

$$\begin{split} T_2 &= \frac{1}{2} I_B \, \omega_x^2 + \frac{1}{2} \, m_c (\gamma_B)_2^2 + \frac{1}{2} \, m_c |\gamma_1|_2^2 \\ &= \frac{1}{2} \left(i \cdot 2 \, fig \cdot m^2 \right) \, \omega_a^2 + \frac{1}{2} (6 \, fig) \gamma_a^2 \, \omega_a^2 + \frac{1}{2} (6 \, fig) |\gamma_1|_2^2 \\ T_2 &= 0.6 \, \omega_a^2 + 4 \, r_0^2 \, \omega_a^2 + 7 \, (2r_1)_2^2 \end{split}$$

 $T_r = T_2$: $13.69 = \{0.6 + 4r_7^2\} \omega_z^2 + 4/r_7^2\}_2^2$ (3)

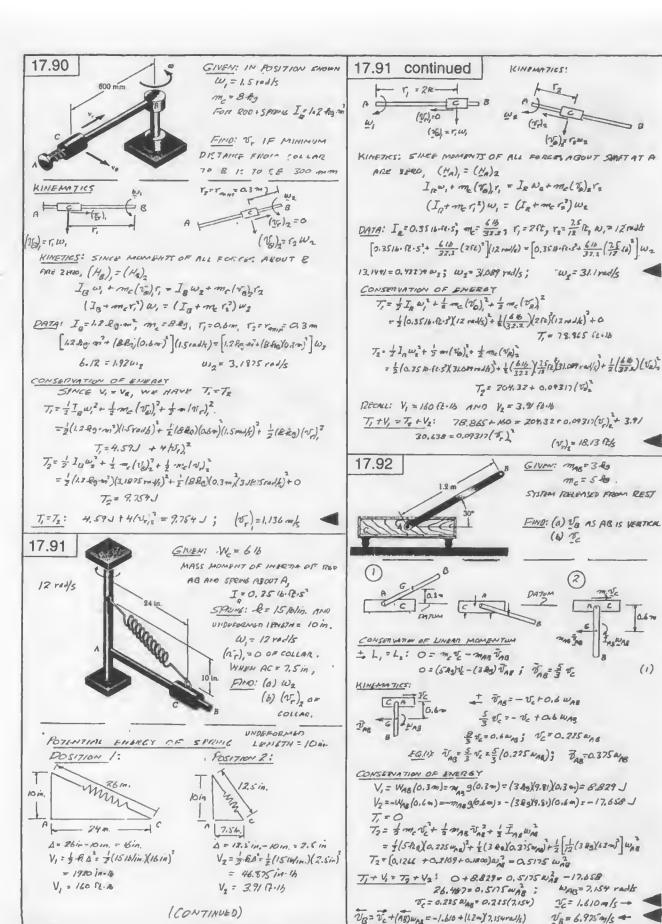
 $EQ(1): \qquad 6.12 = (1.2 + 8 r_1^2) \omega_{\ell} \quad ; \qquad \omega_{\ell} = \frac{6.12}{1.2 + 8 r_1^2} = \frac{3.06}{0.6 + 4 r_2^2} \tag{4}$ $EQ(2) \qquad 13.59 = \left(0.6 + 4 r_2^2\right) \left(\frac{3.06}{0.6 + 4 r_2^2}\right)^2 + 4 \left(\gamma_r\right)^2$

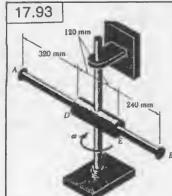
FOR THEOREM WE HAVE (Vr) = 0

13.59 = (3.06)2; 8.154 + 54.36 r2 = 9.

 $13.59 = \frac{(3.06)^2}{0.6 + 4 r_3^2} ; \quad \theta.154 + 54.36 r_2^2 = 9.364$ $Y_2^2 = 22.25 \times 10^{-3} r_3^2$ $Y_2 = 0.1492 m$ $Y_2 = 149.2 mm$

EG(4): $W_2 = \frac{2.06}{0.6 + 4 \Gamma_2^4} = \frac{3.06}{0.6 + 4(22.25 \times 10^{-3})} = 4.441 \text{ yelf}$ $W_2 = 4.441 \text{ yelf}$





FUR CYLINDER DE: 1=0.025. Rg al

IN POSITION SHOWA

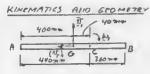
W = 40 valls AND

W = 40 valls AND

FND E OF ROD IS MOTING

TOWARD E AT 75 mals

FIND VELOCITY OF AR L'ELATIVE TO DE AS END & STAIRES END E CF THE CYLINDER



15,= (0.04 m) W, = (0.04 m) 40 rock

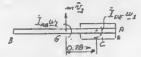
N = 1.6 m/s

INITIAL POSITION

FINAL POSITION

WE HAVE CONSTITUTION OF ANGULAR MIGHTUM ABOUT C.





+) MOMENT AROS = C: Ine 1/2 (3 he (0.9 m) = 0.16-kg·m = Ing w, + m v, (0.0+a) + Îor w, = Îng w = m = (0.28 m) + Îre wx

(0.16 kg·m (401016) + (280) (6.616) 0.04 m) + (0.025 kg·m) (401016)

= (0.16 fig. m. 2) wg + (3. Ra)(0.28 wg)(0.28) + (0.025 fig. m) wg

(6.4 + 0.192 + 1.00) = (0.16 + 0.7352+0.025) N2 7.592=0.4802 N2; W= 18.068 red/s: W= 18.07 red/s

CONSERVATION OF ENERGY $\{V_r\}_r = 0.075 \text{ m/s}$ $V_r = V_2 = 0$ $T_r = \frac{1}{2} \vec{I}_{OE} \omega_r^2 + \frac{1}{2} \vec{I}_{AS} \omega_r^2 + \frac{1}{2} m_{AS} \vec{v}_r^2 + \frac{1}{2} m_{AS} (\vec{v}_r)_r^2$ $= \frac{1}{2} (0.075 \cdot \hat{k}_S \cdot m^2) + 4 (0.16 \cdot \hat{k}_S \cdot m^2) \times 40 \text{ radk}^2$ $+ \frac{1}{2} (3 \cdot \hat{k}_S V_1 \cdot \hat{b}_S V_2^2 + \frac{1}{3} (3 \cdot \hat{k}_S V_2 \cdot 0.075 \cdot m_K^2)^2$

$$\begin{split} &\mathcal{T}_{1} = 80 \, \text{J} + 128 \, \text{J} + 3.84 \, \text{J} + 6.608 \, \text{J} = 151.85 \, \text{J} \\ &\tilde{\mathcal{D}}_{2}^{-1} = (0.28 \, \text{m}) \omega_{0} + (0.78 \, \text{m}) (18.068 \, \text{red/s}) = 5.059 \, \text{m/s} \\ &\mathcal{T}_{2} = \frac{1}{2} \, \tilde{\mathbf{I}}_{08} \, \omega_{0}^{2} + \frac{1}{2} \, \tilde{\mathbf{I}}_{RB} \, \omega_{0}^{2} + \frac{1}{2} \, m_{RB} \, \tilde{\mathcal{D}}_{2}^{2} + \frac{1}{2} \, m_{RB} \, (\mathcal{N}_{V})_{2}^{2} \\ &= \frac{1}{2} \left(0.025 \, \log_{10} \, \mathcal{N}_{1} \right) \left(18.068 \, \text{red/s} \right)^{2} + \frac{1}{3} \left(3.68 \, \log_{10} \, \mathcal{N}_{1} \right) \left(18.068 \, \log_{10} \, \mathcal{N}_{2} \right)^{2} \\ &+ \frac{1}{3} \left(3.83 \, \mathcal{N}_{2} \right) \left(5.059 \, m/s \right)^{2} + \frac{1}{3} \left(3.83 \, \mathcal{N}_{2} \right) \left(\mathcal{N}_{V} \right)_{1}^{2} \end{split}$$

 $T_2 = 4.081 J + 26.116 J + 38.371 J + 1.5 (V_1)_2^2$ $T_2 = 68.587 J + 1.5 (V_1)_2^2$

 $7, +V_1 = 7_2 + V_3$: $|51.85J + 0 = 40.582J + 1.5 (Y_1)_{1}^{3}$ $= 23.263 = 1.5 (7_1)_{1}^{3}$ = 2.45 m/s 17.94

GIVEN: 4-16 TUBE AB
INITIALLY W= 8 modys

AN OR-16 BALL IS
INTRODUCED INTO TUBE
AND LEAVE TUBE AT R.
A SECOND DIS-16 CALL
IS THEN PUT INTO TUBE
FIND: VELOCITY OF
RACH GALL CELATING TO
TUBE AS IT LEAVES THE
TUBE

AS BALL IS INTRODUCED AS SALL LEAVES TUPE

) MUMENTS ABOUT C: $\bar{I}N_{r} = \bar{I}N_{2} + m_{r} \mathcal{L}_{6}(1.5 \text{ ft})$ (1) $N_{6} = (1.5 \text{ ft}) \omega_{s}$ $\bar{I} = \frac{7}{12} \frac{410}{3} (3 \text{ ft})^{3} = \frac{3}{3}$

FIRST O.R. 16 RALL, $\omega_1 = \frac{1}{9} \text{ vall}$ FG(1): $\frac{3}{5} (8 \text{ rad/s}) = \frac{3}{9} \omega_2 + \frac{0.8 \text{ to}}{9} (1.5 \omega_2)(1.5 \text{ ft})$ $24 = (3 + 1.8) \omega_2$ $\omega_2 = 5 \text{ rad/s}$ As FIRST BALL LEAVES TUBE: $\omega_2 = 5^{\circ} \text{ rad/s}$ SECCIO 0.8-16 RALL. $\omega_1 = 5^{\circ} \text{ val}$ FG(1): $\frac{3}{9} (5 \text{ rad/s}) = \frac{3}{9} \omega_2 + \frac{0.8 \text{ to}}{9} (1.5 \omega_2)(1.5 \text{ to})$ $15 = (3+1.8) \omega_2$ $\omega_2 = 3.125 \text{ rad/s}$ As SECOND BALL LEAVES TUBE: $\omega_2 = 3.125 \text{ rad/s}$

CONSTRATION OF ENERGY

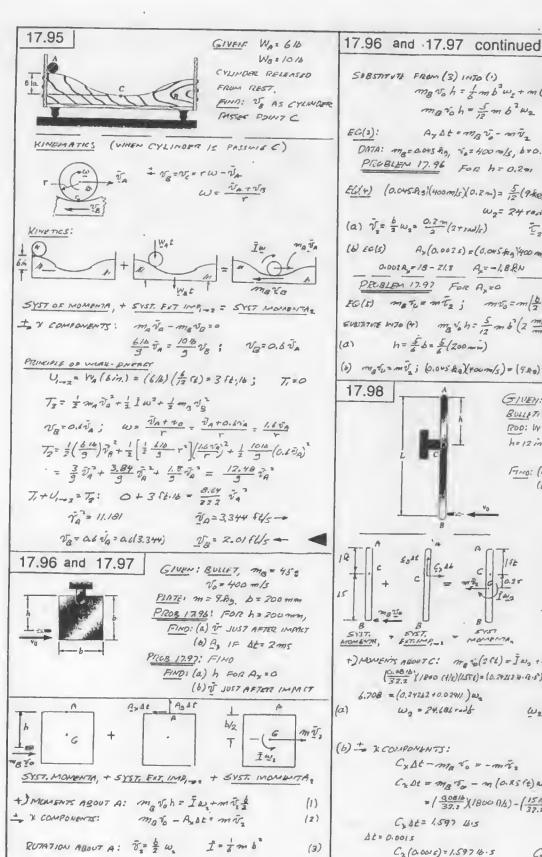
FIRST BAY: $w_1 = 8 \operatorname{radk}$, $w_2 = 5 \operatorname{radk}$ $V_1 = 0$, $T_1 = \frac{1}{2} \tilde{I} w_1^2 = \frac{1}{2} \left(\frac{3}{3} (8 \operatorname{radk})^2 = \frac{96}{3} \right)$ $V_2 = 0$, $T_3 = \frac{1}{2} \tilde{I} w_2^2 + \frac{1}{2} \operatorname{m} v_0^2 + \frac{1}{2} \operatorname{m} v_1^2$ $= \frac{1}{2} \left(\frac{3}{3} (5 \operatorname{radk})^2 + \frac{1}{2} \left(\frac{0.318}{3} (1.5)^2 (5 \operatorname{radk})^2 + \frac{1}{2} \left(\frac{0.88}{3} \right) v_1^2 \right)$ $T_2 = \frac{32.5}{3} + \frac{27.5}{3} + \frac{0.5}{3} v_1^2$

 $T_r = T_r + V_r : \frac{86}{3} + 0 = \frac{1}{9} (325 + 22.5) + \frac{0.4}{5} v_r^2 + 0$ $V_r = 90 \qquad V_r = 9.49 \text{ Sels}$

SECOND BALL W, = Stadk, W. = 3.125rall

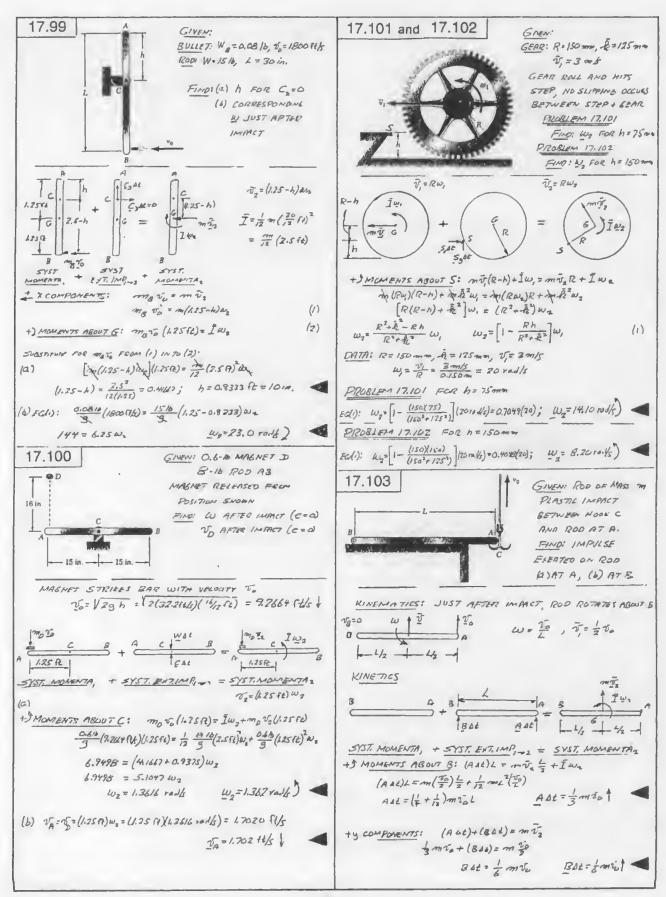
 $V_{1}=0, \ T_{1}=\frac{1}{2}\tilde{I}w_{1}^{2}=\frac{1}{2}\left(\frac{3}{2}\right)\left(srods\right)^{\frac{1}{2}}=\frac{32.5}{9}$ $V_{2}=0, \ T_{2}=\frac{1}{2}\tilde{I}w_{1}+\frac{1}{2}\log\delta_{0}^{2}+\frac{1}{2}m^{2}\delta_{v}^{2}$ $=\frac{1}{2}\left(\frac{3}{2}\right)\left(3.12srods\right)+\frac{1}{2}\left(\frac{0.8}{9}\right)\left(3.12srods\right)+\frac{1}{2}\left(\frac{0.8}{9}\right)\left(\frac{1}{2}\right)$ $=\frac{14.68v}{3}+\frac{0.789}{3}+\frac{0.4}{9}\tilde{V}_{v}^{2}$

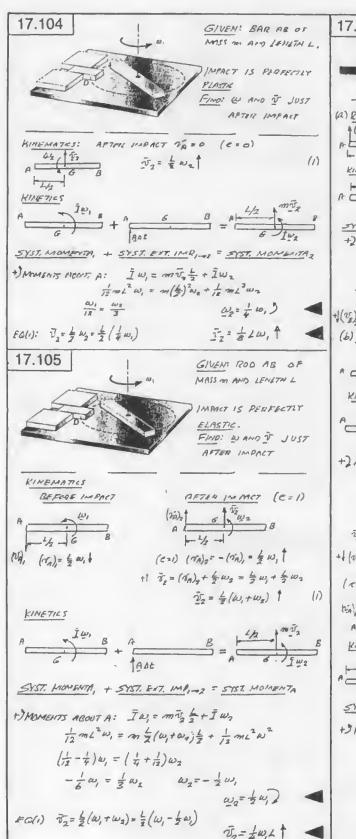
 $T_{r} + V_{r} = T_{2} + V_{2}$: $\frac{37.5}{3} = \frac{14.187}{3} + \frac{0.787}{9} + \frac{0.4}{9} v_{r}^{2}$ $V_{r}^{2} = 35.156 \qquad v_{r} = 5.93 \text{ ft/s}$

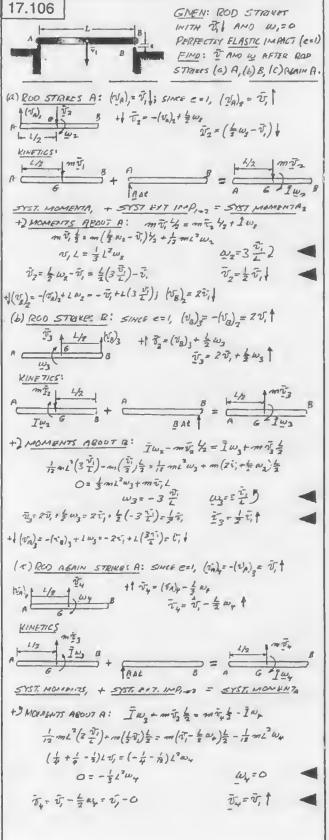


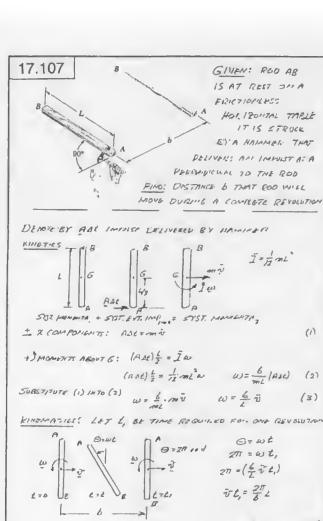
mg voh = /mb w + m (1 6 w) 1 6 march = 5 m 62 W2 AyAtomBr-mv, DATA: mg= 0.045 Ag, 5=400 m/s, b=0.2m, m= 9 Ag, St:0.0025 PROBLEM 17.96 FOR h= 0.201 E(14) (0.045 Rs)(400 m/s)(0.2 m)= 5 (9 ke)(0.20m) No W== 24 rad/s T= 2.4 m/s -Ax (0.0025) = (0.045 fra 400 m/s) - (9 fra X2.4 m/s) 0.001 R. = 18 - Z1.8 A. = -1.8 RN A. = 1.8 AN -FO(5) mato = mt2; mto=m/2 w2); w= 2 mo 20 EURIATURE M70 (4) mg No h = 5 m 6 (2 mg 10) h= 166.700m (b) mation wij; (0.005 kg) (+oum/s) = (9 kg) v; 3= 2m/5 GIVEN: BULLET, Wa= 0.0816, V,= 1800 P/5 ROO: W= 1516, L = 30 in. h= 12 in. FINO: (a) W JUST AFTER MARACT (b) C FOR At = 0.0015 I= 1 7142 = 1/2 (150) (304) 1=0.24262 1b. 12.52 == (0.25ft) an +) MANUENTS ABOUTC: ma is (2(1) =] w, + m x, (0.2592) (32.2) 1800 (1/5) (1556) = (0.24212 W.12.5) W2 + (1516) (0.25) W2 W= = 247 radk) Call = MR 5 - m (0.85 (t) Wa = (30.816) (1800 Ab) - (1516) (0.75 Q) (24.666 - 2/6) Cy= 159716-C2 (0.0015)=1.59716.5

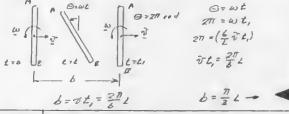
(CONTINUED)









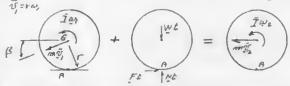




GINETTI SOTTERE ROLLS AND HITS HORIZONTAL SURFACE. AFTH. SLIPPING IT STADTS ROLLING AGAIN

FIND: 2 AMD W, AS IT 120115 70 7112 1487

> POSITION 2, SPHERE HAS RESUNTED ROLLING, TOTAL

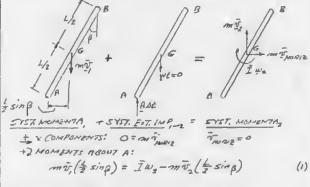


SYST. MOMENTA, + SYST. EVT. IMP, = SYST. MOMENTA +) MONNY OF ALLOW A! IN+ (MV, COSB) r= IW+ mozv 2mr2w,+(mrw, rosp)r=2mrw,+mrva)r (=+ cosp) w, = = w2 11== = 1/2+5cosBja,)

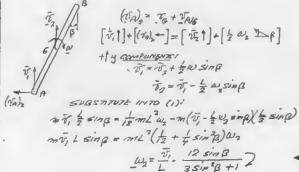
22 = rw = 3 (2+5 cusp) ru, v== = 1/2+500:pjn -- 17.109 and 17.110

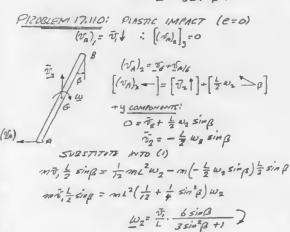
GIVEN: ROD AR STRIKES FRICHUNIES SUIFACE WITH THE VELOCITY SHOWN.

DERIVE AN EXPRESSION FOR W IMMEDIATELY AFTER IMPACT. PROBLEM 17.109 ASSUME PERFECTLY ELASTIC IMPACT, (e = 1) PROBLEM 17.110 ASSUMIL PERFECTLY PLASTIC IMMIT (e:0)



PROBLEM 12109: FLASTIC IMPACT AT A (e=1) : [(va)2] = v,1 (VA) = V,+ KINEMATICS!







GIVEH: UNIFORM CRATE
IS RELEASED FROM LIET.
IMPACT AT B IS PREFECTLY
PLASTIC.

FIND I SMALLEST VALUE OF & B FOR WHICH COENER A REMAINS IN CONTRCT WITH FLOOR

WIE CONSIDER THE LIMITING CASE WHEN THE CRATE IS JUST READY TO ROTATE ABOUT B. AT THAT INSTANT THE VELOCITES MUST BE TERM AND THE REACTION AT CORNER A MUST BE ZERO.



$$\begin{array}{c|c} & C & & \\ & &$$

SYST MOMENTA, + SYST. EXT. IMP, = 5YST. MOMENTA 2 +2 MOMENTS ABOUT B $\tilde{I}(\omega, + (m\tilde{r}), \frac{b}{2} - (m\tilde{r}), \frac{a}{2} + 0 = 0$ (1)

NOTE: $5in\phi = \frac{b}{\sqrt{a^2+b^2}}$, $\cos\phi = \frac{a}{\sqrt{a^2+b^2}}$

v = (96) ω = 1/2 / a2+ 43 ω.

THUS: $(m\bar{\nu}_i)_{\chi^{\pm}}(m\bar{\nu}_i)_{Sin} \dot{\phi} = \frac{m}{2} |a^i + b^{\pm} \omega_i|_{\sqrt{a^2+b^{\pm}}} = \frac{1}{2} mb\omega_i$ ALD, $(m\bar{\nu}_i)_{\omega} = (m\bar{\nu}_i)_{COS} \dot{\phi} = \frac{1}{2} m\alpha\omega_i$

 $\tilde{I} = \frac{1}{12} m(a^2 + b^2)$

 $\frac{2O(1)!}{12} \frac{1}{m(a^2+b^2)} \omega_1 + \frac{1}{2} (mb\omega_1) \frac{b}{2} - \frac{1}{2} (ma\omega_1) \frac{a}{2} = 0$ $\frac{1}{3} mb^2 \omega_1 - \frac{1}{6} ma^2 \omega_2 = 0$ $\frac{1}{3} mb^2 \omega_2 = 0$

17.112



GIVEN: ROD OF LENGTH L.
ASSUMING PERFECTLY
RASHE IMPACT,

FIMP: WAND & JUST AFTER CORD BECOMES TAUT.

KINE MATICS (JUST AFTER IMPACT) LET $\Theta = ton^{-1}/2$ $\{V_A\}_2 = V_S + T_{AK}$ $V_S =$

17.112 continued

SYST. MOVENTA, + SYST. FIT. IMF, -2 = SYST. MONERITA .

+7 MOMENTS ABOUT A

mis = Iw+miy =

mis = Baiw+mis =

 $\overline{V}_0 = \frac{1}{6} L \omega + \overline{V}_y \tag{2}$

mão coso = mão cino + mão coso

 $v_0 = v_1 + v_2$ (1)-(1) $v_0 = (\bar{v}_y - \frac{1}{2}\omega)t_{0n}^2 + v_2$

 $\tilde{v}_{\theta} = \tilde{\tau}_{y} \left(1 + \tan^{2} \theta \right) - \frac{1}{2} \omega \tan^{2} \theta$ $\omega = \frac{2}{L} \left(\tilde{\tau}_{y} \frac{1 + \tan^{2} \theta}{\tan^{2} \theta} - \frac{\tilde{\tau}_{\theta}}{\tan^{2} \theta} \right) \tag{4}$

 $(\psi) \rightarrow (2) \qquad \tilde{V}_0 = \frac{L}{6} \cdot \frac{2}{L} \left(\tilde{V}_y \frac{1 + L_0^{-1} \phi}{L_0 n^{-1} \phi} - \frac{\tilde{V}_0}{L_0 n^{-1} \phi} \right) + \tilde{V}_y$ $\tilde{V}_0 = V_y \left(1 + \frac{1}{3} \frac{1 + L_0 n^{-1} \phi}{L_0 n^{-1} \phi} \right) - \frac{\tilde{V}_0}{\tilde{V}_0} + \frac{\tilde{V}_0}{L_0 n^{-1} \phi}$

376 $ton^2 = v_y(1+4 ton^2 6) - v_0$ $v_y = \frac{1+3 ton^2 6}{1+4 ton^2 6} v_0^2$ (E.

 $(5) \rightarrow (2) \quad \bar{V}_{0} = \frac{L}{6} \cdot \omega + \frac{1+3 \tan^{2} 6}{1+4 \tan^{2} 6} \quad \bar{V}_{0}.$ $\omega = \frac{L}{L} \left[1 - \frac{1+3 \tan^{2} 6}{1+4 \tan^{2} 6} \right] \bar{V}_{0} = \frac{L}{L} \left[\frac{1+4 \tan^{2} 6}{1+4 \tan^{2} 6} - 1 - 3 \tan^{2} L \right]$ $\omega = \frac{L}{L} \frac{\tan^{2} 6}{1+4 \tan^{2} 6} \quad \bar{V}_{0}$ (6)

(6) No(5) -0(1) $\overline{v}_{\chi^{-}}\left(\overline{d}_{y}-\frac{1}{2}\omega\right)$ to ϵ $\overline{v}_{\chi^{-}}\left[\frac{1+3\tan^{2}\epsilon}{1+4\tan^{2}\epsilon}\,\overline{v}_{0}-\frac{1}{2}\cdot\frac{\epsilon}{L}\cdot\frac{\tan^{2}\epsilon}{1+4\tan^{2}\epsilon}\,\overline{v}_{0}\right]$ to ϵ $\overline{v}_{\chi^{-}}\frac{\tan\theta}{1+4\tan^{2}\epsilon}\,\overline{v}_{0}$

DATA: 4 5 8

ton8 = 1/2

Ea161: $\omega = \frac{6}{L} \frac{Lor^26}{1 + 4 Lor^26} v_0 = \frac{6}{L} \frac{as^2}{1 + 4(os)^3} v_0 = \frac{4s}{2} \frac{v_0}{L}$ $\omega = \frac{3}{4} \frac{v_0}{L} 2$

 $\frac{EG(7)}{\sqrt{1+4+ton6}} \vec{v}_0 = \frac{0.5}{1+4(0.5)^7} \vec{v}_0 = \frac{0.5}{2} \vec{v}_0$ $\vec{v}_0 = \frac{1}{4} \vec{v}_0 = \frac{$

 $\frac{FO(5)}{\sqrt{y}} = \frac{1+3\tan^2 6}{1+4\tan^3 6} v_0 = \frac{1+3/6 \cdot 5^2}{1+4/6 \cdot 5^2} v_0 = \frac{1.25}{2} v_0$ $\tilde{v}_y = \frac{7}{6} v_0 \downarrow$

CMER: EQ(1) $\vec{v}_{x} = (\vec{v}_{y} - \frac{1}{2}\omega) t_{0}c$ $= (\frac{7}{2}\vec{v}_{w} - \frac{1}{2} \cdot \frac{7}{4}\frac{\vec{v}_{w}}{c})(0.5)$ $= (\frac{7}{2} \cdot \frac{7}{2}v_{o}) a.5$ $\vec{v}_{y} = \frac{1}{4}v_{o}$



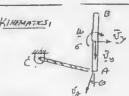
GIVEN: ROU AB OF LEHETH L.

ASSUMING PETERECTLY

PLASTIC IMPACT

FIND: W AMD I IMMEDIATELY

AFTER COTED RECOMES TAUT.



v= v, /0 + ½ω -.
v= (½ω - v, sine) -.
vy= v, cose |

KINETICSI

SYST. MOMENTA, +SYST. EXT. IMP, = SYST. MOMENTA.

+) MOMENTS ABOUT A: $0 = \tilde{I}\omega + m\tilde{\tau}_{1}\frac{1}{2}$ $0 = \frac{1}{12}mL^{2}\omega + m(\frac{1}{2}\omega - \tilde{\tau}_{1}\sin6)\frac{1}{2}$ $0 = \frac{1}{3}\tilde{\tau}_{1}L^{2}\omega - \tilde{\tau}_{1}\tilde{\tau}_{1}L^{2}\sin6$ $\omega = \frac{3}{2}\frac{\tilde{\tau}_{1}}{2}\sin6$ (1)

+ $\int components$: $\int v_{ij} cos\theta = \int v_{ij} c$

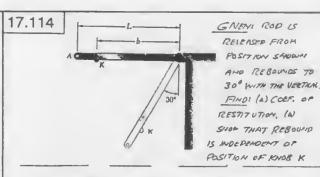
(2)-9(1) $\omega = \frac{3}{2L} \left(v_{\omega} \cos \theta + \frac{1}{2} \omega \sin \theta \right) \sin \theta$ $\omega = \frac{3}{2L} \cos \theta \sin \theta + \frac{3}{4} \omega \sin^2 \theta$ $\omega = \frac{3}{2} \frac{v_{\theta}}{L} \frac{\cos \theta \sin \theta}{1 - \frac{3}{2} \sin^2 \theta}$

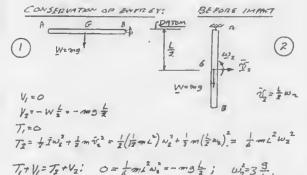
For $0 = ton^{-1}0.5$, $cas <math>0 = \frac{2}{\sqrt{5}}$ and $sin 6 = \frac{1}{\sqrt{5}}$ $\omega = \frac{3}{2} \frac{\tau_0}{L} \cdot \frac{(2\sqrt{3})^2/\sqrt{5}}{1 - \frac{3}{2}(\frac{1}{\sqrt{3}})^2} = \frac{3}{5} \cdot \frac{1}{0.85} \frac{v_0}{L}$ $\omega = 0.7059 \frac{v_0}{L}$ $\omega = 0.705 \frac{\tau_0}{L}$

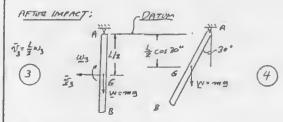
 $EQ(2) \cdot V_{A} = V_{0} \cos \theta + \frac{1}{2} \omega \sin \theta$ $= V_{0} \frac{2}{\sqrt{3}} + \frac{1}{2} (0.7059 \frac{V_{0}}{L}) \frac{1}{\sqrt{3}}$ $= (0.8144 + 0.1528) V_{0}$ $V_{A} = 1.0527 V_{0}$

 $\vec{V}_{Z} = \frac{1}{2} \omega - \vec{V}_{A} \sin e - \frac{1}{2} (0.259 \frac{\vec{V}_{O}}{2}) - (1.0572 \hat{V}_{O}) \frac{1}{15}$ $= (0.35295 - 0.47059) \hat{V}_{O} = -0.11764 \hat{V}_{O}$ $\vec{J}_{O} = 0.1176 \hat{V}_{O}$

vy= v4 cos6 = (1.0522 vo) = 0.94/1 vo vy= 0.94/1 vo







 $V_{3} = -W \frac{1}{2} = 0530^{0}$ $V_{4} = -W \frac{1}{2} = 0530^{0}$ $T_{3} = \frac{1}{2} \frac{1}{4} u_{3}^{2} + \frac{1}{2} m \tilde{V}_{3}^{2} = \frac{1}{2} \left(\frac{1}{2} m L^{2}\right) u_{3}^{2} + \frac{1}{2} m \left(\frac{1}{2} u_{4}\right)^{2} = \frac{1}{6} m L^{2} u_{3}^{2}$ $T_{4} = 0$ $T_{3} + V_{3} = T_{4} + V_{4}; \quad \int_{0}^{1} m L^{2} u_{3}^{2} - m_{6} \frac{1}{2} = 0 - m_{6} \frac{1}{2} \cos 30^{\circ}$

 $T_2 + V_3 = T_{yy} + V_{y}$: $\frac{1}{6} mL^2 \omega_3^2 - m_0 \frac{L}{2} = 0 - m_0 \frac{L}{2} \cos 30^\circ$ $\omega_3^2 = 3\frac{9}{L} (1 - \cos 30^\circ)$.

(Vx) = rw, = r /3 = (Vx) = rw = r /3 = (1-cos yo

 $C = \frac{(V_{K})_{3}}{(V_{K})_{2}} = \frac{\Gamma\sqrt{3\frac{9}{2}(1-\cos 36^{\circ})}}{\sqrt{\sqrt{3\frac{9}{2}}}} = -\sqrt{(1-\cos 36^{\circ})}$ $C = \sqrt{(1-\cos 36^{\circ})} = \sqrt{1-0.84603}$

e=0.366

WE NOTE THAT RESULT IS INDEPENDENT OF THE POSITION OF THE KNOB.



STRIFFS ORSTRUCTION
AT B.

FIND! V, FOR WHILE MASSIMUM VALUE OF 8-30°



SIST. NAMENTA, + SYST. EXT. IMP = = 575T, MOLLETTA :
+) MOMENTS ABOUT &: $m = \frac{\pi}{2} \frac{1}{2} = \frac{\pi}{2} w_0 + m \frac{\pi}{2} (86)$ $m = \frac{1}{2} \frac{1}{2} = \frac{\pi}{2} (86)^2 w_0$ $26 = (9/2)^2 + (40) = \frac{1}{2} (8^2 + 6^2)$

$$\tilde{J} = \frac{1}{13} \approx (a^{2} + b^{2})$$

$$\frac{1}{2} = \frac{1}{12} \approx (a^{2} + b^{2}) \omega_{2} + \frac{1}{2} (a^{2} + b^{2}) \omega_{2}$$

$$\tilde{J}_{1} = \frac{1}{2} = \frac{1}{12} \approx (a^{2} + b^{2}) \omega_{2}$$

$$\omega_{2} = \frac{3}{2} = \frac{b}{c^{2} + b^{2}} \tilde{V}_{1}$$

 $\frac{D^{ATA!}}{\omega_{2}} = \frac{10}{12} \int_{1}^{2} \int_$

CONSERVATION OF ENERGY:





 $\mathcal{T}_{i} = \frac{1}{2} \bar{I} w_{2}^{3} + \frac{1}{2} m \bar{V}_{1}^{2} = \frac{1}{2} \frac{1}{2} m (a^{2} - b^{2}) w_{3}^{2} + \frac{1}{2} m \frac{1}{4} (a^{2} + b^{2}) w_{1}^{2}$ $V_{i} = w \frac{b}{2} = mg \frac{b}{2}$ $V_{2} = w h = mg h$ $\mathcal{T}_{2} = G$

 $T_1 + V_1 = T_2 + V_2$: $\frac{1}{a} d_1(a^2 + b^2) u_2^2 + d_2 = \frac{b}{a} = d_2 h$ $\omega_2^2 = \frac{6(h - b/e)}{(a^2 + b^2)}$ (2)

Fon Omen = 30°

h Sin.

6 1

2.5in.

R.5in.

$$E_{0} = \frac{2.5m}{5.1m},$$

$$\beta = 26.565^{\circ}$$

$$B_{0} = \sqrt{2.5^{2} + 5^{2}} = 5.59021n$$

$$B_{0} = 0.46585 \text{ ft}$$

h=(B6)ton(30"+4)=(0.46585 f6) sin(20"+26.525) h=0.38876 ft

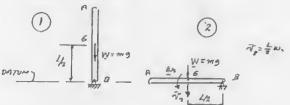
 $EG(2): \quad W_2^2 = \frac{6(h - \frac{b}{2})}{9(a^2 + b^2)} = \frac{6(0.188h - \frac{2.5}{11})}{\left[\left(\frac{10}{12}\right)^2 + \left(\frac{5}{12}\right)^2\right]} = 2.7 = 40.156$ $W_2 = 6.237 \text{ rod}$

FG(1): ω2 = 0720 v,; 6.337 = 0.720 v,
v= 8.80 ft/s

17.116

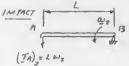
GIVEN: ROD AB IS GIVEN
SLIPT NUDGE AND ROTATES
COUNTERCLOCKINGE NITS
SWEFACE AND REBOUNDS
C=040
FIND: MAXIMUM & OF
REBOUND.

CONSERVATION OF ENERGYS



 $T_{1}=0 \quad V_{1}=m_{0}\frac{1}{2}$ $T_{2}=\frac{1}{2}\tilde{I}\omega_{1}^{2}+\frac{1}{2}m_{0}^{2}=\frac{1}{2}\cdot\frac{1}{12}m_{0}^{2}\omega_{2}^{2}+\frac{1}{2}m_{0}^{2}\omega_{2}^{2}+\frac{1}{2}m_{0}^{2}\omega_{2}^{2}$ $V_{2}=0$ $T_{1}V_{2}=T_{1}V_{2}; \quad O+m_{0}\frac{1}{2}=\frac{1}{2}m_{0}^{2}\omega_{2}^{2}+0$

 $\frac{77+V_1 = 72+V_2}{}$: $0 + m\theta \frac{1}{2} \circ \frac{1}{6} m_1^2 w_2^2 + 0$ $W_2^2 = 3 \frac{\theta}{L}$

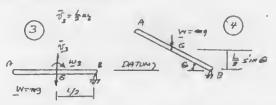




$$e = \frac{(v_A)_1}{(v_A)_2} * \frac{L\omega_3}{L\omega_2} : \qquad \omega_2 = e \omega_2$$

$$OR; \qquad \omega_3^2 = e^2 \omega_2^2$$

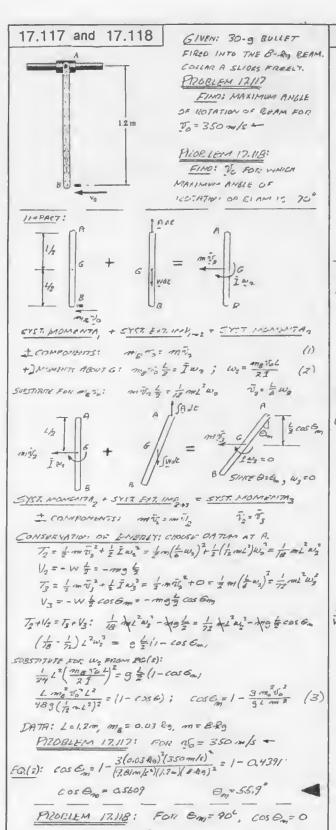
CONSERVATION OF ENERGY



$$\begin{split} V_3 &= 0, \quad T_3 = \frac{1}{2} \left[\overline{L} \omega_3^2 + \frac{1}{2} \cos \overline{t}_3^2 - \frac{1}{2} \cdot \frac{1}{12} \cos^2 \omega_3^2 + \frac{1}{2} \cos \left(\frac{1}{2} \omega_3^2 - \frac{1}{2} - \cot^2 \omega_3^2 \right) \right] &= \frac{1}{6} \cot^2 \omega_3^2 \\ V_4 &= \cos \frac{1}{2} \sin \theta \; , \quad T_4 = 0 \end{split}$$

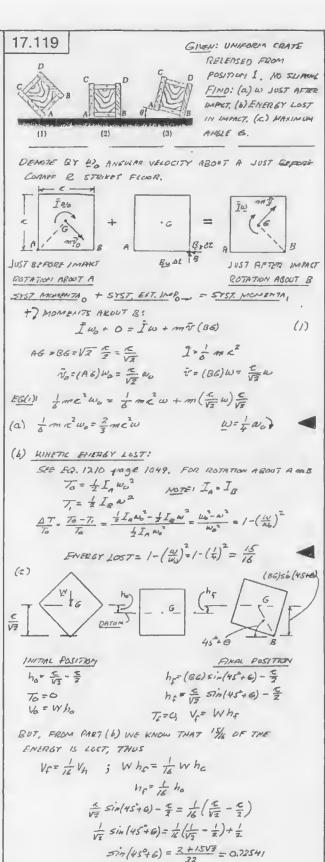
 $\frac{T_3 + V_4 : \frac{1}{6} mL^2 w_2^2 + 0 = m_0 \frac{1}{2} \sin 6}{\frac{1}{6} mL^2 (e^2 w_2^2) = m_0 \frac{1}{2} \sin 6}$ $\frac{1}{6} mL^2 (e^2 w_2^2) = m_0 \frac{1}{2} \sin 6$ $\frac{1}{6} mL^2 (e^2 w_2^2) = m_0 \frac{1}{2} \sin 6$ $\sin 6 = e^2$

For e= 0.40 61A6 = (0.40) = 0.16 0=9.21°



 $FO(2): O = /-\frac{3(0.03-ka)^{2}v_{0}^{2}}{(9.81 \text{ m/s}^{2})(1.2 \text{ m/s} \frac{2}{100})^{2}}; /-3.5837 \times 0.00 = 0$

Vu= 279.04 x103



4540= 46.5036

B=1.50°

20= 528 m/s +



FIRST IMPACT. (6) h2 AFTER

SECOND IMPACT.

CONSERVATION OF ENERGY

b/x

ho

CONSERVATION OF ENERGY

ho

CONSERVATION OF ENERGY

A

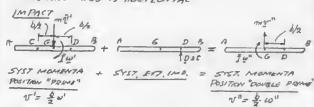
CONSERVATION O

POSITION O'

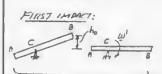
POSITION PRIMA $V = mga_0; T_0 = 0$ $V = 0; T' = \frac{1}{2} I \omega^2 + \frac{1}{2} m v^2$ $T' = \frac{1}{2} (\frac{1}{12} \pi L^2) \omega^2 + \frac{1}{4} m (\frac{1}{2} \omega)^2$ $T' = \frac{1}{2} (m \omega)^2 (L^2 + 3b^2)$

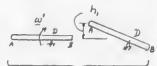
 $\frac{7}{6} + V_0 = 7^0 + V' : O + \frac{1}{24} g a_D = \frac{1}{24} \frac{1}{24} \frac{1}{24} g a_D^{1/2} (L^2 + 3b^2)$ $\frac{1}{24} g h_0 \frac{b}{1+b} = w^2 (L^2 + 3b^2)$ $(w)^2 = \frac{24 g b h_0}{(L+b)(L^2 + 3b^2)}$ (1)

NOTE THIS EXPRESSION ALSO RELATES THE HEIMT
THE EIRO OF THE ROO WISES WHEN ANGULAR VELOUTY W'
OCCURS WHEN ROU IS HOWIZONTAL



+) MOMENTS DEOUT D: $\bar{T}\omega' - m\bar{v}'\frac{b}{b} = \bar{I}\omega'' + m\bar{v}''\frac{b}{2}$ $\frac{1}{2}m^2 a^2 - m(\frac{b}{2})^2\omega' = \frac{1}{2}m^2 a^2 \omega'' + m(\frac{b}{2})^2\omega''$ $\omega'' = \frac{L^2/2 - b_A^2}{L^2/2 + b_A^2}\omega'$ $\omega'' = \frac{L^2 - 3b^2}{L^2 + 3b^2}\omega'$ (2)





Fa(1): $(\omega)^{\frac{1}{2}} \frac{243b}{(L+b)} \cdot \frac{h_0}{(L^2+3b^2)} \qquad (\omega'')^{\frac{1}{2}} \frac{243b}{(L+b)} \cdot \frac{h_1}{(L^2+3b^2)}$

 $(\omega'')^{2} = \frac{(L^{2} - 3b^{2})^{2}}{(L^{2} + 3b^{2})^{2}} (\omega')^{2}$ $\left[\frac{248b}{(L+b)} \cdot \frac{h_{1}}{(L^{2} + 3b^{2})}\right] = \frac{(L^{2} - 3b^{2})^{2}}{(L^{2} + 3b^{2})^{2}} \cdot \left[\frac{248b}{(L+d)} \cdot \frac{h_{0}}{(L^{2} + 3b^{2})}\right]$ $h_{1} = \left[\frac{L^{2} - 2b^{2}}{L^{2} + 7b^{2}}\right]^{2} h_{0}$ (3)

SECOND IMPACT: howh, h, he he

ha= [12-362] ho

DATA: ho= 4in., L=30 in., d=6in. L=36 = 30-3(5) = 0.84615

EQ(3): $h_1 = (0.8 + 445)^2 (4 in.) = 2.8639 in.$ EQ(4): $h_2 = (0.8 + 444)^2 (4 in.) = 2.050 Sin.$ h,=2.86 im hz=2.05 in. 17.121 and 17.122



GNEN: 3-16 COLINI A DILOFS

h=15 in.

B-16 DISK OF PAOJUS R=9in.

IMMEDIATELY AFTER IMPACT

FINO: (a) 1/A, (b) W

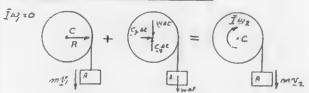
PROBLEM 12:121 ASSUME

PROPRETLY PLATIC IMPACT.

PROBLEM 17:172 ASSUME

PERFECTLY ELASTIC IMPACT.

CHAR A FALLS A DISTANCE h: V, = 129 h
PRINCIPLE OF IMPULSE - MOMENTUM



STOT MOMENTA, + SYSTERT, IMP, -2 = SYST, MOMENTA 2 +2 MOMENTS AROUT C:

 $m v_1 R = \hat{I} \omega_2 + m \hat{v_2} R$ (1)

PROBLEM 12121

PLASTIC IMPACT C=0 $V_2=RW_2$; $W_2=\frac{V_2}{R}$ M=IMASS OF DISK; $\tilde{I}=\frac{1}{2}MR^2$

 $FG.(i): \quad m \, V_i R = \frac{1}{2} M R^2 \left(\frac{\sqrt{2}}{R}\right) + m^{-1/2} R$ $m \, V_i = \frac{1}{2} M \, V_2 + m \, V_3$ $V_2 \cdot \frac{2m}{2m + M} \, V_i \qquad (3)$

D47A: $m = \frac{216}{5}$; $M = \frac{916}{5}$; h = 15 in. $V_1 = \sqrt{29} h = \sqrt{2(32.2 \text{ fU}_3^2)(\frac{14}{12}\text{ fe})} = 8.972 \text{ fU}_3$

FG(3) 7= 2(3/9) (8,977 (46) = 3,845 (46) 12=3.85 (2/6)

W2 = 12 = 3,845 (1/5 = 5.127 10) (4) (4) 5.13 rod/s)

PIZOBLEM 17.122:

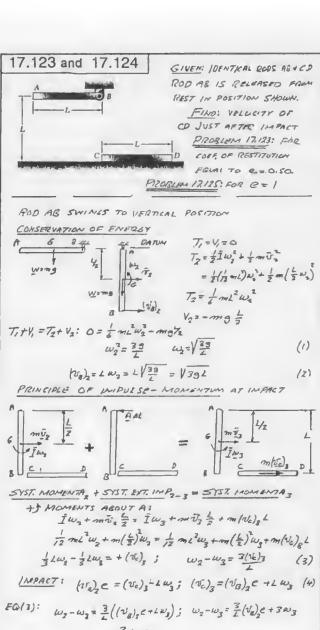
ELASTIC IMPACT e=1 $(v_{8})_{2}-(v_{A})_{3}=(v_{A}), -(v_{8}),$ $(v_{8})_{1}=0; (v_{5})_{2}=Rw_{8}:$ $Rw_{2}-v_{2}=v_{1}; w_{2}=(v_{1}+v_{2})/R$ (2)

 $EG(1): m N_1 R = \frac{1}{2} M R^2 \left(\frac{V_1 + V_2}{12} \right) + m V_2 R$ $m V_1 R = \frac{1}{2} M R V_1 + \frac{1}{2} M R V_2 + m V_2 R$ $V_2 = \frac{2m - M}{2m + M} V_1$ (4)

DATA: nn= 31/9; M= 84/9; h= 15in., V,= 8.972 ft/s

EO.(4) $\sqrt{s_2} = \frac{2(\frac{3}{3}) - \frac{9}{9}}{2(\frac{3}{9}) + \frac{9}{9}} (8.972 \, \text{ft/s}) = -1.2817 \, \text{ft/s}$

EQ(1): $W_2 = \frac{x_1 + y_2}{R} = \frac{8.977 (2/5 - 1.282) (2/5)}{(\frac{9}{12} + 6)}$ $w_2 = 10.25$ for d/5)



$$\begin{split} \mathcal{E}Q(3): & \quad \omega_2 - \omega_3 = \frac{3}{L} \left((\mathcal{V}_g)_2 e + \mathcal{I} \, \omega_3 \right); \quad \omega_2 - \omega_3 = \frac{3}{L} \left(\mathcal{V}_g \right)_2 e + 3 \, \omega_3 \\ & \quad \omega_2 - 4 \, \omega_3 = \frac{3}{L} \left(\mathcal{V}_g \right)_2 e \\ \mathcal{SURSTITUTE} & \quad \mathcal{FROM} (1) \quad A \sim 0 \, 2) \\ & \quad \sqrt{\frac{3 \, g}{L}} - 4 \, \omega_3 = \frac{3}{L} \sqrt{3 \, g} \, L \quad e \\ & \quad 4 \, \omega_3 = \sqrt{\frac{3 \, g}{L}} - \frac{3}{L} \sqrt{7 \, g} \, L \quad e \quad ; \quad 4 \, \omega_3 = \frac{1}{L} \sqrt{3 \, g} \, L - \frac{3}{L} \sqrt{3 \, g} \, L \, e \\ & \quad \omega_3 = \frac{1}{4L} \left((-3 \, e) \right) \sqrt{3 \, g} \, L \end{split}$$

 $EO(4); \ \langle V_{e} \rangle_{3} = \langle V_{e} \rangle_{2} e + L \omega_{3} = (\sqrt{39L}) e + L \left[\frac{1}{4L} (1 - 3e) \sqrt{39L} \right]$ $|V_{e}\rangle_{3} = \sqrt{39L} \left(e + \frac{1}{4} - \frac{3}{4} e \right) = \sqrt{39L} \frac{1}{7} (1 + e)$ $|V_{e}\rangle_{3} = \frac{1}{4} (1 + e) \sqrt{39L}$ (5)

PROBLEM 12/23: FOR e = 0.5FO(s): $(V_c)_3 = \frac{1}{4}(170.6)V_{391} = \frac{3}{6}V_{391}$ $V_{c0} = \frac{3}{6}V_{391} = \frac{3}{6}V_{391}$

20= = 1/39L -

 $\frac{PROBLEM 17.174}{\{t_{i}\}_{3}} = \frac{1}{4}(1+1)\sqrt{39}L$

17.125 and 17.126



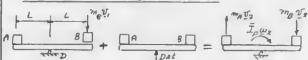
GIVEN: GYMMAST A IS
AT REST. SYMMAST B
JUMPS ON TO PLANK AT E.

h= 2.5 mi

MASS GF PLANK: mp= 15-Rg
ASSUMING PETIFECTLY
PLASTK IMPACT,
FIND: HEIGHT THAT

PROBLEM 17.125: USE ma=55-Rg AND mg=70-Rg
PROBLEM 17.126: USE ma=70-Rg AND mg=55-Rg

VELOCITY OF 8 AS IT STRIKES E: V, = VZgh
PRINCIPLE OF IMPOLSE - MOMENTUM



SYST. MOMENTA, + SYST. EYT. IMP₁₋₁ = SYST. MOMENTA₂
+) MOMENTS ABOUT DI $m_{\alpha} v_{i} = \overline{J}_{D} w_{2} + m_{\alpha} v_{2} l + m_{\beta} v_{1} l$ $m_{\beta} v_{i} = \overline{J}_{D} m_{\beta} (2l)^{2} u_{\alpha} + (m_{\alpha} + m_{\beta}) (l w_{\alpha}) l$ $m_{\beta} v_{i} l = \overline{J}_{\beta} m_{\beta} l^{2} u_{2} + (m_{\alpha} + m_{\beta}) l^{2} u_{2}$ $w_{2} = \frac{m_{\beta}}{\overline{J}_{m_{\beta}} + m_{\alpha} + m_{\beta}} \frac{v_{i}}{l}$ $v_{2} = l u_{2} = \frac{3m_{\beta}}{m_{\beta} + 3m_{\beta} + 3m_{\beta}} v_{i}$ (1)

FOR h=2.5m V,= VZgh = VZ(9.8/m/s= X2.5m) = 7.0036 m/s

PROBLEM 17.175

mp=15-Rg ma= 55-Rg ma= 75-Rg

 $F_{Q(j)}: V_2 = \frac{3(70)}{15^2 + 3(55) + 3(70)} (2003670/5) = \frac{210}{370} (20036)$ $V_2 = 3,77770/5$

 $V_2 = \sqrt{29} h_2$ $h_2 = \frac{V_2^2}{29} = \frac{(3.77/mls)^2}{2(9.81mk^2)} = 0.725 m$ GYMNAST & PISES 725 mm

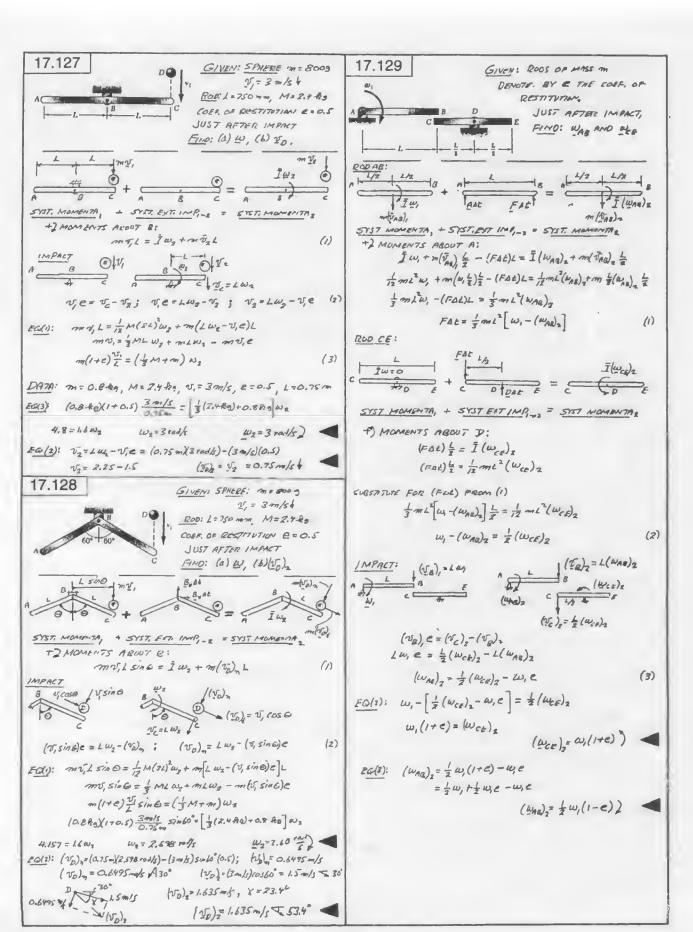
PROBLEM 17.126

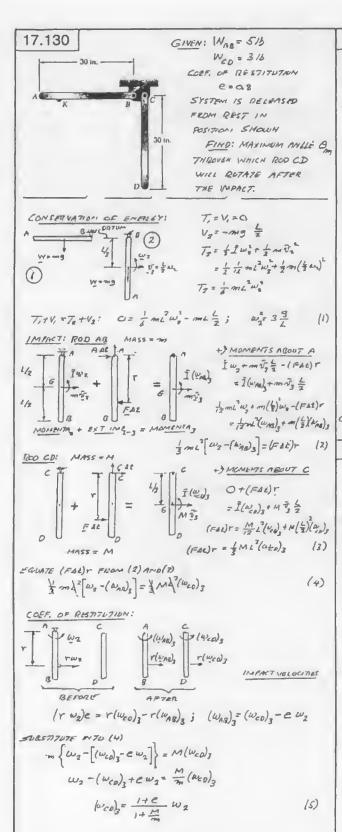
mp= 15 kg mn= 70-kg mg= 55kg

 $Z_{2}(1)$: $V_{2} = \frac{2(55)}{15 + 3(74) + 3(55)} (7.0036 m/c) = \frac{165}{390} (7.0036)$

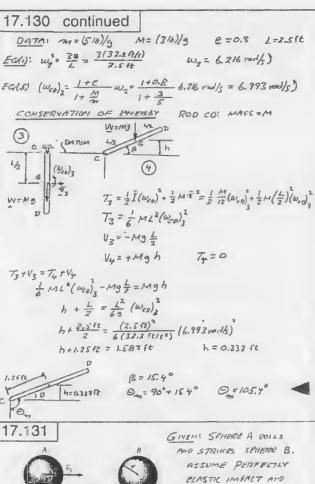
 $N_2 = \sqrt{29 h_2}$ $h_2 = \frac{\sqrt{2}}{29} = \frac{(2.913 \text{ m/s})^2}{2(9.91 \text{ m/s}^2)} = 0.447 \text{ m}$

GYM WAST A RISES 447 mm





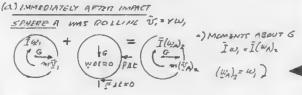
(CONTINUED)



FIM: JUST AFTER IMPACT, (a) WE AND IT OF EACH SPASSE.

DENOTE COEF, OF KINETIC

FICIETRA BY HA.



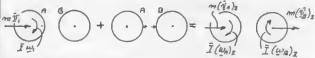
SPHERE 8: $G. \qquad \uparrow \qquad \downarrow_{G} \qquad = \qquad \tilde{I}(\omega_{g})_{2} \qquad m(\tilde{v}_{g})_{2}$ $Fat=C \qquad \uparrow$ $+) MOMENTS ABOUT G: \qquad \tilde{I}(\omega_{g})_{2} = C$

 $(\omega_3)_2 = 0$

(CONTIMUED)

17.131 continued

CONSIDER BOTH SPARRES AS A SYSTEM



+ SYST. EAT. IMP = SYST, MAMENTAZ I COMPONENTS: m 5, = m(2,) + m(1,)

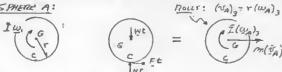
$$\bar{v}_{i} = (v_{0})_{2} + (\sigma_{8})_{2}$$
 (1)

RELATIVE VELOCITIES (C=1) V,= (VB) 2-(VA) 2 (v) - (v) = ev = v,

ADD EGS. (1) AND (2): 2 v,= 2(v,) ; (V,)= v, -> SUBTRAT EG(1) FROM FQ(1): 0 - Z(TA)2; (TA)2=0

(b) MOTION AFTER SPHERES START ROLLING UNIFORMLY NOTE: TIME INTERNAL IS NOT SMALL AND

IMPUSES OF FRICTION FORCES MUST BE INCLUDED



SYST, MOMENTA, + EYET, EXT. IMP = SYST. MOMENTA

+ 2 MOMENTS ABOUT C: IW = I(W) + m(VA), Y 3 mr2 w, = 3 mr2 (wA) 3+ mr2 (wA) 3 (WA) = = = W,)

SPHERE B:

Bolls: (VB)3 = r(W2)3

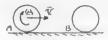
SYST MOMERTA + SYST EAT IMP = SYST, MOMENTA 3

+) MOMENTS ABOUT DI mv, r = I(wg)2+m(18)2 r mir= = = mr (we) + mr (we) 3

$$(\omega_B)_s = \frac{5}{7} \frac{\overline{v_1}}{r}$$
 But $\omega_i = \frac{\overline{v_1}}{r}$

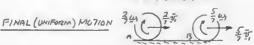
SUMMORY

INITIAL MOTION

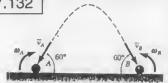


JUST AFER IMAGT





17.132



GIVEN: BALL BOUNCES AS SHOWN. ag= ab, VA= 26 FIMD: WO IN TERMS OF TO AND T

SINCE THE LINEAR NID ALIBUAN VELOCITIES ARE CHANGED DURING A SHORT INTERVAL AT, BOTH THE NORMAL AND FRICTION FORCES ARE IMPULSIVE WE ASSUME THAT NO SLIPPING OCCURS.

FOR THE VELOCITY OF THE BALL TO BE REVERSED AT EACH IMPACT, WE MUST HAVE AT POINT A.

AFTER IMPACT BEFULFIMPACT

INPACT AT A:



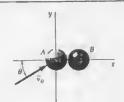
SYST. MOMENTA, + SYST. EXT. IMP = SYST. MOMENTAL

+ I MOMENTS AROUT C: Iw, - (m v. cos60) r = - Iw' + (mi cos60) r

SUBSTITUTE: WA = WA = WO VA = VA = 20 21 wa = 2(mv cos60°) r W= (mzocos60)r

Wo = m = (1/2)r

17.133 and 17.134



GIVEN: BALL A IS ROLLING WITHOUT SLIPPING YVERN IT HITS BALL B. COEF. OF KINE TIC FRICTION IS 4/4.
ASSUMING PERFECTLY ELASTIC IMMET,

FIRM: (a) \$\vec{x}\$ AND WOF

FACE GAIL, (b) \$\vec{x}\$ AFTOR

17 STANTS ROLLING

PROBLEM 17.134: FIND EQUATION OF PATH OF BALL A

PROBLEM 17.133

(1) MOTION IMMEDIATELY AFTER IMPACT

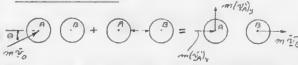
FRICTION PORCES ARE NON IMPULSIVE, THUS

AIGULAR MONIENTUM (AND THUS W) OF EACH BALL

IS UN CHANGED. WE HAVE W'S: O

AND SINCE BALL A WAS ROLLING:

LOOKING DOWNINARD



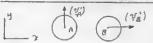
5/57. MOMENTA, + SYST. ELT. IMPO = = SYST. MOMENTA,

± COMPANENTS! mr. COSS = m(Va) + m V2 (1)

+1 COMPONENTS OF RAIL A: $m\vec{\tau}_{o}\sin 6 = m(\vec{\tau}_{o})_{u}$ $|\vec{\tau}_{o}|_{u} = \vec{\tau}_{o}\sin 6 \qquad (2)$

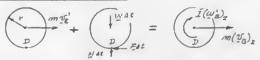
FOR FLASTIC IMPACT $e^{\pm i}$ $v_3^i - (v_4^i)_{\chi} = \bar{v}_0 \cos 6 \tag{3}$

MOTION IMMEDIATELY AFTER IMPACT:



 $v_A = v_b \sin \theta$; $v_B = v_b \cos \theta$ $\dot{\epsilon}$

(b) FINAL VELOUTY OF BALL B:



SYST. MOMENTA, + SYST. ETT. $IMP_{j-2} = SYST.$ MOMENTA 2+2 MOMENTS ABOUT D: $mT_g'r = \overline{1}(w_g')_2 + m(v_g)_2 r$ (5)

WE RECALL: V's=V, cos & AND I= 3 mr

RALL ROLLS: (NG) = r(WB).

 $\frac{EG(S):}{(w_B)_2} = \frac{5}{7} \frac{\vec{v}_0}{r} \cos \Theta = \frac{2}{5} mr^2 (w_B)_2 + mr(w_B)_2 r$ $(w_B)_2 = \frac{5}{7} \frac{\vec{v}_0}{r} \cos \Theta \qquad (\tilde{v}_B)_2 = r(w_B)_2 \rightarrow (v_B)_2 = \frac{5}{7} \tilde{v}_0 \cos \Theta \hat{i}$

(CONTINUED)

17.133 and 17.134 continued

PROBLEM 17.134 LOCKING DOWNWARD ON BALL A $\frac{\partial \hat{x}_{n}}{\partial x_{n}} = \frac{\partial \hat{x}_{n}}{\partial x_{n}} =$

WE ASSUME THAT BALL A ROLLS WITHAT SLIPPING IN Y DIRECTION (MA) X I'A = - (To Sing i) YI'A = + TO SING 1
ASSUMPTION IS CURRECT

THE Y COMPARED OF VECOUTY IS CONSTANT AND THUS THE Y COORDINATE AT ANY TIME & IS

 $y = (v_A)t = (\bar{v}_0 \circ s = 6)t$

BALL A ROLLS AND SLICES IN THE " DIRECTION

$$\tilde{I}\left(\frac{\tilde{\chi}_{g}^{2}}{r}\cos\theta\right)\left(-G\right) + \left(\frac{1}{G}\right)^{WL=ngt} = \left(\frac{\tilde{I}}{I}\omega_{g}\right)^{n} \left(\frac{1}{G}\right)^{n}$$

$$Nt = ngt \uparrow F_{g}$$

$$= cos \theta$$

+ x consusurs: 0 + 4, mg t = m (va), m(va), = 4.3t

 $(a_{A})_{\frac{1}{2}} = y_{A} = constant$ $y = \frac{1}{2}at^{2}$ $y = \frac{1}{2}y_{A} y t^{2}$ (2)

ELIMINATE & BETWEEN EGS (1) ANO (2)

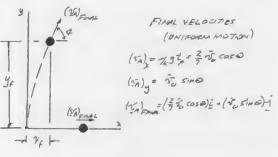
t= 3/(To sind): x= = = 1/2 (3/2 sin26);



 $\frac{1}{2} \text{ MOMENTS ABOUT GI } i \left(\frac{1}{r} \cos \theta \right) - \left(\frac{1}{4} - \log t \right) = \frac{1}{2} \log_2 \frac{1}{r} \log_2 \frac{$

ROLLING WITHOUT SLIDING BEAMS WARN $\xi_{+} = \frac{2}{7} \cdot \frac{\tilde{\chi}_{0} \cos \theta}{\tilde{\chi}_{0}^{2}}$ $EQ.(2): \quad T_{0}^{-\frac{1}{2}} \tilde{\chi}_{0}^{2} \tilde{\chi}_{0}^{\frac{1}{2}} = \frac{2}{4} \cdot \frac{\tilde{\chi}_{0}^{2} \cos^{2} \theta}{\tilde{\chi}_{0}^{2}}$

Ea(1): 45 = 10 sine to = 3 40 sinocore

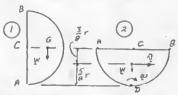


(VA) = (VB) = 5 V. COS & i

17,135



GIVEN: UNIFORM HEMISPHERE IS RELEASED FROM REST AND ROLLS WITHOUT SLIDING, AFTER HEMISPHERE ROLLS THROUGH 90, FINO: (a) W (b) NORMAL REPKTION .



T= Fra]= (2-9)mr

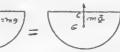
(a) Mary- FHEREY: U1 = 2 = W(1 + 1) = 3 mg "

$$T_{2} = \frac{1}{2} \tilde{I} \omega^{2} + \frac{1}{2} m \tilde{v}^{2} = \frac{1}{2} \left(\frac{2}{5} - \frac{9}{64} \right) m r^{2} \omega^{2} + \frac{1}{2} m \left(\frac{5}{6} r \right) \omega^{2}$$

$$T_{2} = \frac{13}{40} m r^{2} \omega^{2}$$

T,+ U,-2=77: 0+3 mgr = 13 mrw W= 15 8

(6) REACTION AT DS

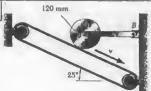


d=0 0 =0 a=a=+a===(=r)w=+ ma-m(3) wit

· I EF . EFAT: N-mg = m (3/8 r) w N= mg+m(音r)(音中)= 149mg

N=1.433 29 1

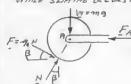
17.136



GIVEN: 4/2 0.15 2 = 25 m/s X CYLINDER IS AT REST WHEN PLACED ON BELT. FIND: (a) NUMBER OF REVIOLUTIONS

REFORE CYLINOET DEACHES CONSTANT VELOCITY. (b) TIME REQUIRED TO REACH CONSTAINT VELOUTY.

WHILE SLIPPING OCCURS:



+1 IFH = 0 NOSB-MAISINR-mg=0

SLIPPING GCCURS UNTIL:

WORK-ENERGY. M. Fr = MOMENT OF F ABOUT A. U== MD= Fr0= MAre T,=0: T2= = = [[= = = [/ m1]] = fmr2

T.+U,-2= Tz: 0+4,NrB = + mv2

$$\Theta = \frac{1}{4} \frac{mv^{2}}{\eta_{k}r} \frac{1}{N} = \frac{1}{4} \frac{mv^{2}}{\eta_{k}r} \cdot \frac{\cos\beta - \eta_{k} \sin\beta}{mg}$$

$$\Theta = \frac{1}{4} \frac{v^{2}}{\eta_{k}rg} \cdot (\cos\beta - \eta_{k} \sin\beta)$$
 [2]

(CONTINUED)

17.136 continued

PRINCIPLE OF IMPULSE - MOMENTUM



SYST. MOMENTA, + SYST EXT. IMP, -2 = SYST. MOMENTA +) MOMENTS ABOUT A: FLY = IN -yx Ner= = = mr2(X)

SUBSTANTE FOR HI 4 (cosp-4 sum) tr= imru

$$\dot{t} = \frac{1}{2} \frac{v}{7.9} \left(\cos \beta - 4 \sin \theta \right) \tag{3}$$

$$\frac{DA7A^{2} - y_{K} \circ 0.15}{8 = 25^{\circ}, \quad \pi = 25 \text{ m/s}, \quad r = 0.12 \text{ m}}$$

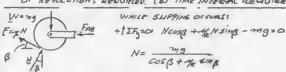
$$\frac{EC(2)!}{9 = \frac{(25 \text{ m/s})^{2}}{4(0.15)(0.12 \text{ m})(9.81 \text{ m/s})} \left[\cos 25^{\circ} - (0.15) \sin 25^{\circ}\right]}{\Theta = 745.86 \text{ rad} \left(\frac{RBV}{25^{\circ} + 4}\right); \qquad \Theta = 1/8.7 \text{ revolutions}}$$

17,137

GNEN: 4x . 0.15 25 = 25 m/5 CYLMORR AT REST PLAKED ON BELT. UNTRE MOTION BECOMES UNIAGEM FIND: (4) NUMBER

(1)

OF REVOLUTIONS REQUIRED. (b) TIME INTERNAL REQUIRED



FOR CYLHONES SLIPPING OCCURS UNTE WET WORK-ENERSY: Me Fr = MOMENT OF F ABOUT A. UI-Z= MIRO= Fro = 4, Nro

TITUING TE: OTHENOS TONES 0= 1 mr2 1 = 1 mv2 cosp+4, sino (Cosp + Mx sine) (2)

PRINCIPLE OF IMPULSE-MOMENTUM



Fir= Iw +] MON ENTS ABOUT A! -4x NET= = mr2(V)

SUBSTITUTE FUR M

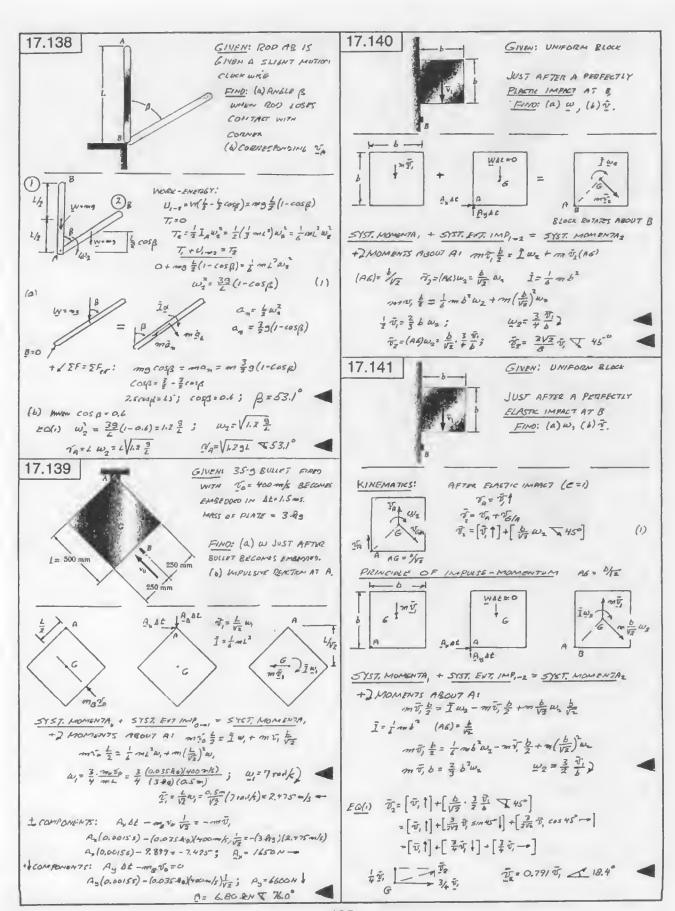
$$4/k \left(\frac{m_0}{\cos\beta + \gamma_k \sin\beta}\right) \pm r = \frac{1}{2} mrv$$

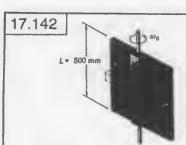
$$\pm = \frac{1}{2} \cdot \frac{v}{4/3} \cdot (\cos\beta + v_k \sin\beta) \tag{3}$$

DATA: 4/2 = 0.15, B = 25°, V= 25 m/s, r= 0.12m

$$E(d2)$$
: $\Theta = \frac{(25 \text{ m/s})^2}{4(0.15)(0.12 \text{ m/s}^2)} \left[\cos 25^2 + (0.15) \sin 25^2\right]$
 $\Theta = 858.05 \text{ rad} \left(\frac{784}{34744}\right)$ $\Theta = 136.6 \text{ resolutions}$

6= 1 T2 (cosp-1/2 sing) (2) Ea(s): t= 25m/s 2(a/s)(9.81 m/s) [cos25*1(a/s) sin 25*] t=8.245





GIVEN: 3-43 BAR AB 4-tig PLATE Wy = 120 TPM

AFTER BAR SWUNG TO HORIZONTAL,

FIND: (a) W, (b) EHERLY

LOST DURING GASTIC

IMPACT AT C

(a) LOOKING DOWNWARD

Conservation of arguar mumerity about shaft $I_0 u_0 = I_1 u_1$ (1)

I = I = 1/2 (4-85) L2

I,= Iplate+ Ism= 1/2 (4Aq)2+ 1/2 (3Aq) L= 1/2 (7Aq)L2

EQ(1): 12 (490) 12 (120 ypm) = 12 (749) 12 W

W= 3(120 pm) W= 68.6 rpm

(b) ENERGY: (WE MUST USE rad/s)

, Up = 120 rpm (27) = 47 rad/s = 12.516 rad/s

w = 4 rad = 4 (47 rad/s) = 7.12/ rad/s

To = 1/2 Io Wo2 = 1/2 (1/2 (4 & 0.5 m)²) (12.516 rad/s) = 6.580 J

 $T_i = \frac{1}{2} I \omega_i^2 = \frac{1}{2} \left[\frac{1}{12} (2Ag) (a.S.m.)^2 \right] (2.18) \operatorname{radk}^2 3.260 J$

ENERGY LOST = 6.580 J-5.7601 = 2.82 J

17.143



CIVEN: PIN B IS
REMOVED AND PLATE
SWAWS ABOUT A
FIND: (a) W AFTER
90° ROTATION,
(b) MAXIMUM W

7 = 1 I, w2

(a) B CDATUM a \overline{Z} B $I_{A} = \frac{1}{3} \operatorname{sn} (a^{2} + b^{2})$ 3 In. $I_{A} = \frac{1}{3} \operatorname{sn} (a^{2} + b^{2})$ $I_{A} = \frac{1}{3$

 $U_{i-j} = W(4in - 3in) = mg(\frac{1}{2}E)$ $T_i = 0: T_2 = \frac{1}{2}I_2\omega_1^2 = \frac{1}{2}\frac{2}{2}\sum_{i=1}^{n} m\omega_1^2$

 $7_1 + 4_{-2} = 7_2$: $\log(\frac{1}{7_2}) = \frac{1}{2} \frac{25}{432} \log u_2^2$ $w_1^2 = \frac{10}{75} 9$

DATUM

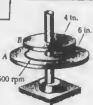
AG = (Bin)2+ (4in)2 = 5 in.

 $E_{Q(i)}: \quad I_{A} = \frac{25}{\sqrt{22}} \approx$

 $U_{l-3} = W(\sin - 3 \sin) = mg \left(\frac{2}{l^2} f_0^2\right)$ $T_l = G$ $T_2 = \frac{1}{2} I_0 w_0^2 = \frac{1}{2} \frac{25}{48} w_3^2$

 $\frac{7_1 + U_{1-3} = 7_3}{U_3^2 = \frac{36}{25} 9 = \frac{36}{25} (27,2) = \frac{46}{25} 25 \times W_3^2}$ $\frac{36}{25} 9 = \frac{36}{25} (27,2) = \frac{46}{25} 278 \times W_3^2 = \frac{6}{25} \frac{81}{25} rad/s$

17.144



GNEN: DISK OF SAME
THEKESS AND SAME MATERIAL
DISIL B IS AT REST WHEN
IT IS DRUPPED ON DISK A
KNOWING WA = 1816,
FIND: (a) FINAL W OF DISK?
(b) CHANCE IN KINETIC ENERGY

z)
FOR A DISK: mopuspart; Î= = mre= = |part)r= = patr*

Îw, 70 +

In any

SYST. MOMERIM, + STST. Ext. INP, =2 = SYST. MOMENTA,
+) MOMERITS ABOUT 6: $\hat{I}_{\alpha} \omega_{\alpha} = \hat{I}_{\alpha} \omega_{\alpha} + \hat{I}_{\alpha} \omega_{\beta}$ $\frac{1}{2} \rho n e r_{\alpha}^{\gamma} \omega_{\alpha} = \left(\frac{1}{2} \rho n e r_{\alpha}^{\alpha} + \frac{1}{2} \rho n e r_{\alpha}^{\alpha}\right) \omega_{\beta}$ $\omega_{\beta} = \frac{r_{\alpha}^{\gamma}}{\Gamma_{\alpha}^{\gamma} + \Gamma_{\alpha}^{\gamma}} \omega_{\beta}$

Wg = (6in)+(4in)+ (500 pm) = 417.5% rm Wg = 418 spm

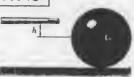
(b) ENERGY: $W_{A} = m_{A}g = \rho J \pi t r_{A}^{2}$ $W_{A} = \frac{r_{A}^{2}}{r_{B}^{2}}$ $W_{A} = \frac{r_{A}^{2}}{r_{A}^{2}}$ $W_{A} = \frac{r_{A}^{2}}{r_{A}^{2}}$

 $\frac{1 \text{MITTAL KIMETIC ENERGY}}{T_{i} = \frac{1}{2} \frac{1}{1}_{n} W_{A}^{n} = \frac{1}{2} \left[\frac{1}{2} \frac{10 \text{ To}}{22.2} \left(\frac{4}{12} \text{ Fz} \right)^{2} \right] \left(52.36 \text{ rod}_{A}^{2} \right) = 95.764 \text{ Fz.16}}$ $\omega_{i} = 412526 \text{ rom} \left(\frac{2\pi}{6\omega} \right) = 43.723 \text{ rod}_{S}^{2}$

 $\mathcal{T}_{2} = \frac{1}{2} \left(\hat{I}_{a} + \hat{I}_{b} \right) a_{b}^{2} = \frac{1}{2} \left[\frac{1}{2} \frac{18 \, b}{32.2} \left(\frac{6}{27} \, \Omega^{2} \right) + \frac{1}{2} \frac{816}{27.2} \left(\frac{4}{12} \, \Omega^{2} \right) \right] \left(43.773 \, \text{rad/s} \right)^{2}$ $\mathcal{T}_{2} = 79.985 \, \text{ft. 13}$

ENFACY LOSS: 17=7=7,=77,7859-16-95.2097-16

17.145



FIND: DISTANCE IN IF BALL IS TO START ROLLING WITHOUT SLIONL

SYST. MOMENTA, + SYST. EVT. IMP, = SYST. MOMENTA2

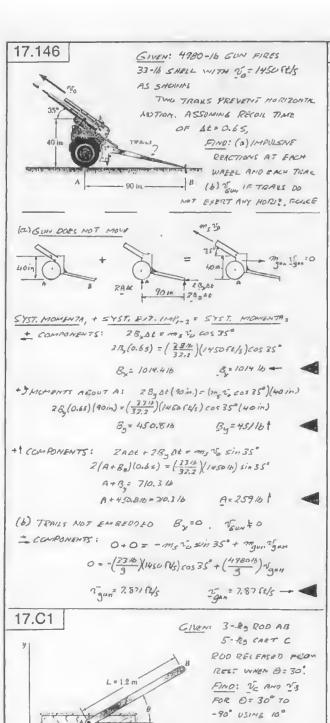
+] MOMENTS ABOUT 6: $(PAt)h = \tilde{I} \omega_2$ (1) + COMPUNENTS: $PAt = m \tilde{V}_2$ (2)

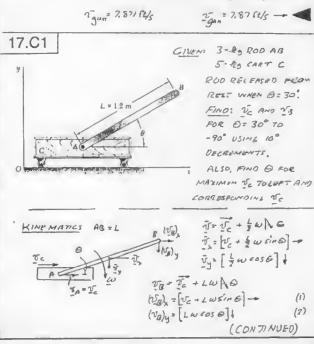
Divide EQ(i) BY EQ(2) MEMBER BY WEMBER $h = \frac{1}{2\pi} \cdot \frac{\alpha_0}{4\tau}$

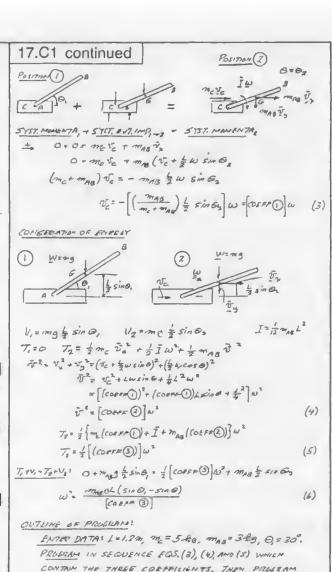
FOR ROLLING BOTOS

h= = - Las

h = 2 r







ENTER DATA! L=1.2m, $m_c=5.4a$, $m_{AB}=3.4a$, Θ , = 30° .

PROBLEM IN SECUENCE EQS.(3), (4), AND (5) WHICH
CONTAIN THE THREE COEFFICIENTS, THEN PRUSE AM
EGS.(1) AIM (2) THAT INVOLVE ($^{\circ}_{A}$), AND ($^{\circ}_{A}$),

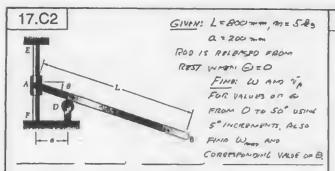
EVALUATE AND PRINT Θ , W, \overline{V} , \overline{V}_{C} , \overline{V}_{B}), (\overline{V}_{B}),

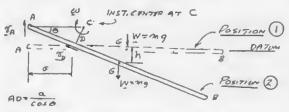
Linear velocities positive to the right and up Omega positive clockwise

theta	omega	VAB=0	٧C	vBx	vBy
deg.	rad/s	m/s	m/s	m/s	m/ss
30.00	0.000	0.000	0.000	0.000	0.000
20.00	2.002	1.157	-0.154	0.667	-2.257
10.00	2.841	1.689	-0.111	0.481	-3.358
0.00	3.502	2.101	0.000	0.000	-4.202
-10.00	4.082	2.427	0.159	-0.691	-4.824
-20.00	4.621	2.672	0.356	-1.541	-5.211
-30.00	5.136	2.837	0.578	-2.504	-5.338
-40.00	5.631	2.923	0.814	-3.529	-5.177
-50.00	6.098	2.933	1.051	-4.555	-4.704
-60.00	6.516	2.881	1.270	-5.502	-3.910
-70.00	6.854	2.795	1.449	-6.279	-2.813
-80.00	7.076	2.715	1.568	-6.795	-1.475
-90.00	7.154	2.683	1.610	-6.975	-0.000
					-

Pind max1mum velocity of cart to the 1eft. 19.70 2.0315 1.1760 -0.15408479 19.69 2.0325 1.1766 -0.15408483 19.68 2.0335 1.1772 -0.15408479

(VE) may TO LEFT = 0.1541 =/ WHEN 6 = 19.7°





CD= (A) to-G =
$$a \frac{\tan G}{\cos G}$$
; $AC = \frac{AD}{\cos G} = \frac{a}{\cos^2 G}$

MASS MOMENT OF INARTA ABOUT INST. CONTER

$$I_c = \frac{1}{12} + m \left[(cD)^2 + (0G)^2 \right]$$

$$I_c = \frac{1}{12} m L^2 + m \left[a^2 \frac{\tan^2 G}{\cos^2 G} - \left(\frac{L}{2} \sin G - a \tan G \right)^2 \right]$$
 (1)

CONSERVATION OF ENERGY

T= 1 I w" (SEE EO 17.10 page 1049)

$$\frac{7.+V_1 = \overline{I_2} + V_2}{\omega^2} = \frac{2mgh}{\overline{I_c}} \qquad \omega = \sqrt{\frac{2mgh}{\overline{I_c}}} \qquad (2)$$

VELOUTY OF A: VA= (Ac) W

OUTLING OF BROGRAM

PROGRAM, IN SEQUENCE, AD, DE, h, CD, AC, IC, W, VA.
EVALUATE AND PRINT &, h, W, AND NA FOR VALUES OF &
FROM O TO 50° AT 5° INTERVALS.

L = 800 mm	a = 200	mm m	= 5 kg
theta	h	omega	٧A
deg	200	rad / s	m/s
0.000	0.000	0.000	0.000
5.000	17.365	1.911	0.385
10.000	34.194	2.680	0.553
15.000	49.938	3.235	0.693
20.000	64.014	3.648	0.826
25.000	75.786	3.934	0.958
30.000	84.530	4.079	1.088
35.000	89.389	4.051	1.208
40.000	89.295	3.811	1.299
45.000	82.843	3.325	1.330
50.000	68.067	2.592	1.255
**********	*********	++++++++++	
	for max om		

	theta deg	h mm	onega _{Ma} , rad / s
	31.810	86.788	4.0907731056
	31.820	86.799	4.0907735825
	31.830	86.810	4.0907735825
	31.840	86.821	4.0907731056
÷	********	********	

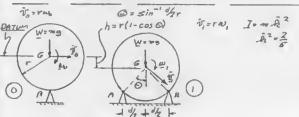
17.C3



GIVEN: 10-in. RADIUS STHERE POULS O'UTE BARS
WITHOUT SLIPPING. KNOWING THAT WO = 1. Erads?
AND ASSUMING PROFESTLY PLASTIC IMPACTS, FOR d = 1 in.
TO Gir. USING O.S-in. INCREMENTS,
FIRD: (A) W. A.S. G. PASSES PLOSECTLY AROUSE R.

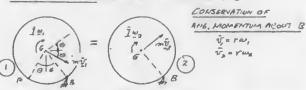
(b) NUMBER OF BARS THE SENERS WILL

POLL OVER AFTER LEAVING BAT A



CONSIDERTION OF INVERGY $V_0 = 0$; $T_0 = \frac{1}{2} \tilde{I} w_0^0 + \frac{1}{2} m \tilde{I}_0^0 = \frac{1}{2} m \tilde{K} w_0^0 + \frac{1}{2} m r^2 w_0^2 = \frac{1}{2} m (\tilde{K}^2 + r^2) w_0^2$ $V_1 = -mgh$; $T_1 = \frac{1}{2} m (\tilde{K}^2 + r^2) w_0^2 = \frac{1}{2} m (\tilde{K}^2 + r^2) u_1^2 - mgh$ $W_1 = W_0 + \frac{1}{2} m (\tilde{K}^2 + r^2) u_2^2 = \frac{1}{2} m (\tilde{K}^2 + r^2) u_1^2 - mgh$ (1)

AFTER IMPACT AT BY STAFFE ROTATES AROUT B

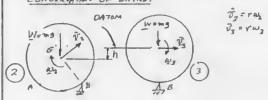


BETTER IMPARTAT B

AFTER IMPARTAT B

+2 MOMBYTT ABOUT B: $Iw, + (mv, coc 20)r = Iw_2 + mv_1r$ $mk^2u, + mv^2cos 20 u, = mk^2u, + mr^2w_2$ $w_2 = \frac{f^2 + r^2cos 20}{f^2 + r^2}w,$ (2)

SPHELE POTATES ABOUT B UNTIL 6 IS ABOVE B



$$V_{2} = -mgh; T_{2} = \frac{1}{2}m(-k^{2}+r^{2})\omega_{2}^{2}$$

$$V_{3} = 0; T_{3} = \frac{1}{2}m(-k^{2}+r^{2})\omega_{3}^{2}$$

$$\frac{T_{2}+V_{2}=T_{3}+V_{3}:}{\frac{1}{2}m(-k^{2}+r^{2})\omega_{2}-mgh} = \frac{1}{2}m(-k^{2}+r^{2})\omega_{3}^{2}$$

$$\omega_{3}^{2} = \omega_{2}^{2} - \frac{2gh}{k^{2}+r^{2}}$$
(3)

(CONTINUED)

17.C3 continued

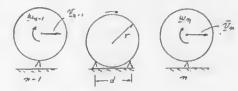
$$|NF| |MVE| |FOUND: |U|^2 = N_0^2 + \frac{73h}{b^2+r^2}$$
 (1)

$$\omega_{x} = \frac{\hat{f}_{x}^{2} + r^{2}\cos 2\theta}{\hat{f}_{x}^{2} + r^{2}} \omega, \qquad (2)$$

$$\omega_3^2 = \omega_2^2 - \frac{2gh}{\bar{R}^2 + \Gamma^2} \tag{3}$$

W3 IS ANGULAR VELOCITY OF SPHERE AS & PASSES
OVER B. (THIS IS SHOWN AS W, IN PROBLEM FILTER)

AS SPHERE ROLLS FROM THE (2-1) HEAR TO THE THE BAR.



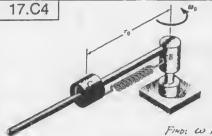
ENTER DATAL Y = 10 10, Wo = 1.5 VINIS & =0.4

FOR
$$d = \frac{1}{12} lt$$
 TO $\frac{L}{12} lt$ INGRESORNY $\frac{O.5}{12} lt$
 $\omega_n = \omega_0$
 $\omega_n = \omega_0$
 $\Theta = \sin^{-1}(\frac{d}{2}r)$
 $h = r(1 - \cos 0)$
 $\omega_i = \left\{ \frac{\omega^2}{2} + 2\frac{n}{2} h/(\frac{1}{n^2} + r^2) \right\}^{V_2}$
 $\omega_2 = \left\{ (\frac{2}{n^2} + r^2 \cos 76)/(\frac{1}{n^2} + r^2) \right\}^{V_2}$
 $\omega_3 = \left\{ \frac{\omega_i^2}{2} - 2\frac{n}{2} h/(\frac{1}{n^2} + r^2) \right\}^{V_2}$
 $1/F = 1/F = 1/F PRINT = 1/F U_3 = 1/F U_3$

r = 10.000 in. omega0 = 1.500 rad/s

Distance	omegs when	Number of bars
between bars	G is over B	sphere rolls over
ln.	rad/s	
1.0	1.494	491
1.5	1.487	169
2.0	1.476	76
2.5	1.460	40
3.0	1.438	23
3.5	1.409	14
4.0	1.370	9
4.5	1.319	6
5.0	1.252	4
5.5	1.164	3
6.0	1.047	2

NOTE: FOIL d=7im, SENDE: FAILS TO REACH
A POSITION WITH & ABOVE B



SPRINE: mc = 2.5 &g

SPRINE: & = 750 N/m

UN STRUTCHED

LUNLIF: C = 500 mm

ROD AND HUB:

IB = 0.3 &g = 2

INITIALLY: V = 500 mm

We 10 rells

(1)

FIND: W MID TO JAB FOR WALLEST OF I FROM 500 mm TO 700 mm AT 25-00 M JACKETUM TS. ALSO FIND THERE.

CONSERVATION OF ANGULAR MONTHRITUM ABOUT B

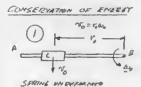
A TO B A TO B

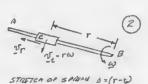
MOTE NO TO B A TO B

NOTE NO TO B

NOTE N

 $\omega = \frac{J_0 + mr^2}{J_0 + mr^2} \omega_0$



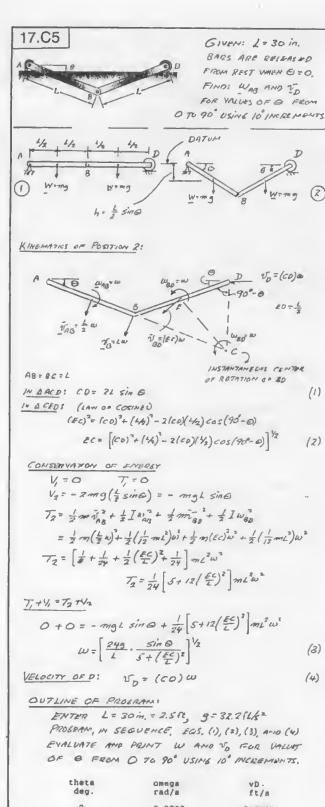


 $\begin{aligned} & \mathcal{T}_{i} = \frac{1}{2} \mathcal{I}_{B} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} - \frac{1}{2} \mathcal{I}_{B} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} w_{0}^{2} = \frac{1}{2} \left(\mathcal{I}_{0} + m v_{0}^{2} \right) w_{0}^{2} \qquad V_{i} = 0 \end{aligned}$ $\begin{aligned} & \mathcal{T}_{2} = \frac{1}{2} \mathcal{I}_{0} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} d d^{2} \end{aligned}$ $& = \frac{1}{2} \mathcal{I}_{B} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} m v_{0}^{2} \end{aligned}$ $& \mathcal{T}_{2} = \frac{1}{2} \left(\mathcal{I}_{0} + m v_{0}^{2} \right) w_{0}^{2} + \frac{1}{2} m v_{0}^{2} +$

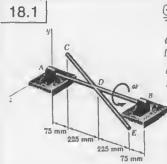
 $\mathcal{N}_{r} = \left\{ \frac{1}{m} \left[(I_{a} + mr_{s}^{a}) a_{b}^{a} - (I_{b} + mr_{s}^{a}) a_{s}^{2} - k(r - r_{s})^{2} \right] \right\}^{V_{2}}$ (2) $\frac{OUTLINE}{ENTER 0A7A} : m = 2.5 k_{0}, I_{a} = 0.3 k_{0} \cdot m_{s}^{a}, r_{s} = 0.5 m_{s} k_{s} = 75U \text{H/m} \text{ and}$ $w_{a} = 10 \text{ rad}$

PROGRAM EQ(1) AND THEN EQ(2). EVALUATE AND PRINT LU AND \mathcal{N}_{L} FOR VALUES OF Y FROM 0.5 m TO 0.7 m AT 0.005 m INCREMENTS. THEN SEEK V_{mag} where \mathcal{N}_{L} =0

г	omega	v radial
(M.HI	rad/s	m/s
500.00	10.000	0.000
525.00	9.352	1.486
550.00	8.757	1.962
575.00	8.211	2.221
600.00	7.708	2.341
625.00	7.246	2.346
650.00	6,820	2.239
675.00	6.428	2.007
700.00	6.066	1.599
Find	r max1mum	(where vr = 0)
r	omega	[v rsdla1]"2
m.m.	rsd/s	
731.75	5.645	0.0014211
731.76	5.645	0.0004968
731.77	5.645	-0.0004275
731.78	5.645	-0.0013555



theta	omega	vD.
deg.	rad/s	ft/s
0	0.0000	0.0000
10	2.4806	2.1537
20	3.1277	5.3487
30	3.3226	8,3066
40	3.3302	10.7031
50	3.2746	12.5423
60	3.2088	13.8945
70	3.1544	14.8210
80	3.1198	15.3622
90	3.1081	15.5403



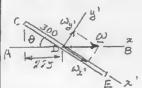
TWO UNIFORM RODS AB AND CE ARE WELDED AT MIDPOINTS D.

MASS OF EACH ROD = 1.5 Kg LENGTH = 600 mm

ASSEMBLY HAS CONSTANT ANG. VEL W= 12 rad/6. FIND:

ANG MOMENTUM HD.

SINCE ROD AB HAS MOM, OF INERTIA 20 ABOUT AXIS OF RUTHTION, ONLY ROD CE CONTRIBUTES TO ANGULAR MOMENTUM.



SINCE CD = 300 mm, Cas 0 = 225 $\theta = 41.41'$ USING THE PRINCIPAL CENTROIDAL AXES 2' 4' E, WE HAVE

$$\omega_{z} = \omega \cos \theta$$

$$\omega_{z} = \omega \sin \theta$$

$$\omega_{z} = 0$$

$$\bar{I}_{z} = 0$$

$$\bar{I}_{z} = \frac{1}{12} m \ell^{2}$$

$$\bar{I}_{z} = \frac{1}{12} m \ell^{3}$$

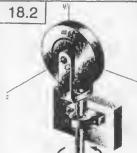
$$H_{i} = \overline{I}, \omega_{i} = 0$$

$$H_z = \overline{I}_2 \omega_z = 0$$

$$H_y$$
, = I_y , ω_y , = $\frac{1}{12}ml\omega\sin\theta$

= 1/2 (1.5 kg)(0.6 m)2 (12 rad/s)sin41.41°= 0.357

H = 0.357kg·m³/s; B= 48.6°, By = 41.4°, Oz = 90° ◀



GIVEN;

THIN . HOMOGENEOUS DISK OF HASS M AND RADIUS & SPINS AT CONSTANT RATE WI. FORK-ENDED ROD SPINS AT CONSTANT RATE W,

ANGULAR MOHENTUM HE OF DISK.

SINCE THE Zy, & AXES THE PRINCIPAL CENTROIDAL AXES, WE CAN USE EUS. (18.10) WITH $\bar{I}_{z} = \bar{T}_{y} = \frac{1}{4}m \, \xi^{2}, \quad \bar{I}_{z} = \frac{1}{2}m \, \xi^{2}$ Wx=0, Wy=Wz, Wz=W,

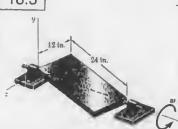
$$H_z = \overline{I} \omega_z = 0$$

$$H_y = \overline{I}_z^2 \omega_z = \frac{1}{4} m z^2 \omega_z$$

$$H_z = \overline{I}_z \omega_z = \frac{1}{2} m z^2 \omega_z$$

$$H_{6} = \frac{1}{4} m \dot{v} (\omega_{2} \dot{j} + 2\omega, K)$$

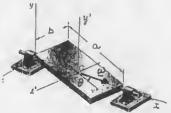
18.3



GIVEN!

RECTANGULAR PLATE SHOWN WEIGHS 18 16 AND ROTATES WITH CONSTANT W= 5 rad/s FIND:

ANGULAR MOMENTUH H ABOUT MASS



WE USE THE PRINCIPAL CENTROIDAL LXES Gz'y'z' WE HAVE W, = W COSA w= = - wsind

I, = 1/2 11 b, I, = 1/2 m(a+b), I, = 1/2 ma

USINE EUS. (18.10)

 $\begin{aligned} H_{x} &= \bar{I}_{x}, \, \omega_{x}, \, = \frac{1}{12} m b^{2} \omega \cos \theta \\ H_{y} &= \bar{I}_{y}, \, \omega_{y}, \, = 0 \end{aligned}$

 $H_{2'} = \vec{I}_{2}, \omega_{2'} = -\frac{1}{12} m \tilde{\alpha} \omega \sin \theta$

H = H2, E + H3, & + H2, &

WHERE L', I', K' ARE THE UNIT VEGIORS ALONG THE x, y, Z' AXES.

He= Im bwcosoi' - Imaw sing k'

TO RETURN TO THE ORIGINAL 2, Y. & AYES, WE NOTE THAT

k'=-Lsind + Kcos H L'= c cost + k sint

THERPFORE

 $H_{G} = \frac{1}{12} m b^{\epsilon} \omega \left(as^{\epsilon} 0 i + cososin \theta k \right) +$ $+ \frac{1}{12} m a^{2} \omega \left(sin^{2} \theta i - sin \theta \omega s \theta k \right)$

 $\underline{H}_{G} = \frac{1}{n} m \omega \left[(a^{2} \sin^{2} \theta + b^{2} \cos^{2} \theta) \dot{i} - (a^{2} - b^{2}) \sin \theta \cos \theta \dot{k} \right]$

GIVEN DATA:

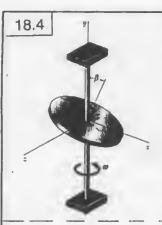
711 = (10/b) (32.2 ft/s') = 0.55901 b.s/ft b = 12 in. = 1 ft a = 24in. = 2 ft 0 = 26.565° tan 0 = = 0.5

71+051 11 = 12 (0.55901 /bs/ft) w [(4 sin 26.565"+ cos 26,565") i-(4-1) sin 26.565° cos 76.565° K7(ft)

= (0.046584 16.5/5E) W (1.600 ! 1.200 f)(ft) H₆ = [(0.074534 16.ft.s]) i - (0.055901 16.fts) K] W (1) LETTING W=5 rad/s.

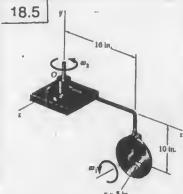
HG = (0.372716. ft.s) i - (0.274516. st.s) k (2)

H = (0.373 | b.ft.s) i - (0.280 | b.ft.s) K



GIVEN:
HOMOGENEOUS DISK OF
MASS on AND RAPIUS &
MOUNTED ON SHAFT AB
WITH \(\beta = 25^\circ\),
SHAFT ROTATES WITH
CONSTANT \(\omega \).

FIND:
ANGLE & FORMED BY
AB AND ANG. MOMENTUM
H OF DISK ABOUT G.



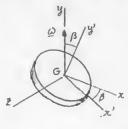
GIVEN:
HOMOGENEOUS DISK OF
WEIGHT W = 8 I.B
ROTATES AT CONSTANT
RATE W = 12 rad/s.
ARM DA ROTATES AT
CONSTANT RATE W = 4 rad/s

FIND:

ANGULAR MOMENTUM

HA OF DISK ABOUT ITS

CENTER A.



WE USE THE PRINCIPAL CENTROIDHL AXES GX'3'E.

WE HAVE:

I, = I = 1/4 m z²

I, = 1/2 m z²

U, = - wsin β

Wy, = W cos B

USING Eqs. (18.10): $H_{z} = \overline{I}_{z}, \, \omega_{z}, = -\frac{1}{4} \, m \, \tilde{\epsilon} \, \omega \, \sin \beta$ $H_{y} = \overline{I}_{y}, \, \omega_{y} = \frac{1}{2} \, m \, \tilde{\epsilon}^{2} \, \omega \, \cos \beta$ $H_{z} = \overline{I}_{z} \, \omega_{z} = 0$

WE HAVE

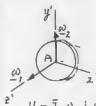
WHERE i', i', K ARE THE UNIT VECTORS ALONG THE 2'8' & AXES.

BUT $H_G \cdot \omega = \frac{1}{4} m z^2 \omega \left(-\sin \beta \dot{z} + 2\cos \beta \dot{z} \right) \cdot \omega \dot{z}$ OR, OBSERVING THAT $\dot{z} \cdot \dot{z} = -\sin \beta$ AND $\dot{z} \cdot \dot{z} = \cos \beta$, $H_G \cdot \omega = \frac{1}{4} \sin z^2 \omega^2 \left(\sin^2 \beta + 2\cos^2 \beta \right)$

 $A = \frac{1}{4} m t^{2} \omega^{2} (1 + \cos^{2} \beta)$ $= \frac{1}{4} m t^{2} \omega^{2} \sqrt{\sin^{2} \beta} + 4\cos^{2} \beta$ $= \frac{1}{4} m t^{2} \omega^{2} \sqrt{1 + 3 \cos^{2} \beta}$ (4)

SUBSTITUTING FROM (3) AND (4) INTO (2), $\cos \theta = \frac{1 + \cos^2 \beta}{\sqrt{1 + 3} \cos^2 \beta}$

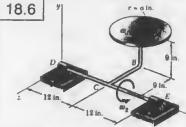
FOL p = 25°, cos 0 = 0,9786 0 = 11.88°



WE USE POINCIPAL CENTRUIDAL AXES AZ'y'z'. WE HAVE $\overline{I}_{z,1} = \overline{I}_{y,1} = \overline{I}_{y,1} = \overline{I}_{y,1} = \overline{I}_{z,1} = \overline{I}$

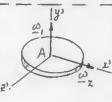
 $H = \frac{1}{2}, \omega_{1}, i + \frac{1}{3}, \omega_{1}, j + \frac{1}{4}, \omega_{2}, k = \frac{1}{4}mt^{2}(\omega_{2}j + 2\omega_{1}k)$ GIVEN DATA: $m = \frac{W}{g} = \frac{8}{32.776} = 0.24845$ lb·s°/ft t = 5 in. $= \frac{5}{12}$ ft, $\omega_{1} = 12$ rad/s, $\omega_{2} = 4$ rad/s $H = \frac{1}{2}(0.24845)(5)^{2}(4.342(12))t$

 $\frac{H_{A} = \frac{1}{4} (0.24845) (\frac{5}{12})^{2} [4j + 2(12)k]}{= (0.043133 /b.ft.s)j + (0.25880 /b.ft.s)k}$ $\underline{H_{A} = (0.0431 /b.ft.s)j + (0.259 /b.ft.s)k}$



GIVEN:
HOMOGENEOU: DISK
OF WEIGHT W= 616
ROTHES IT CONSIDET
RATE Q = 16 radg.
SHAFT DOE RIBLES
AT CONSTANT RATE
W= 8 radgs.

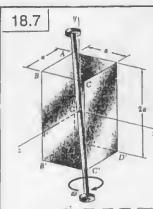
FIND: ANG, HIGH NATIFIE HA OF DISK ABOUT ITS CENTER A:



WE USE PRINCIPAL CENTROIDAL AXES Ax'y'z', WE HAVE $\overline{I}_{x}=\overline{I}_{x}=\frac{1}{4}mz', \overline{I}_{y}=\frac{1}{2}mz'$ $\omega_{x}=\omega_{x}, \omega_{y}=\omega_{1}, \omega_{z}=0$ FROM EUS. (18,10):

HA = (0.1656 16.91.3) i + (0.663 /6.94.5) }

$$\begin{split} & \stackrel{H}{=} \frac{7}{4} \cdot \omega_{x}, \stackrel{i}{=} \frac{1}{7} \cdot \omega_{y}, \stackrel{i}{=} \frac{1}{4} \cdot \omega_{y}, \stackrel{i}{=} \frac{1}{4} \cdot \omega_{z}, \stackrel{i}{=} \frac{1}{4}$$



GIVEN: SOLID RECTAL/GULL FIGHT-LELEPIPED SHOWING FIRESTING

AT AT LONSTAIT ITS DIAGONAL

FIND:

(a) MAGNITULE OF MIS OF

(b) AIRT THAT E FIRE

WE DENOTE BY \vec{I}_x , \vec{I}_y , \vec{I}_z THE PRINCIPAL CENTROIDAL MUMENTS OF INERTIA. WE HAVE $\omega = \omega - a \vec{i} + 2a \vec{j} - a \vec{k} = \frac{\omega}{\sqrt{6}} \left(-\vec{i} + 2\vec{j} - \vec{k} \right) \tag{1}$

 $\underline{H}_{G} = \overline{\underline{I}}_{2} \omega_{2} \underline{i} + \overline{\underline{I}}_{3} \omega_{3} \underline{j} + \overline{\underline{I}}_{2} \omega_{2} \underline{k} = \frac{\omega}{\sqrt{6}} \left(-\overline{\underline{I}}_{2} \underline{i} + 2\overline{\underline{I}}_{3} \underline{j} - \overline{\underline{I}}_{2} \underline{k} \right)$

COMPUTATION OF THE MOMENTS OF INEXTIA: $\vec{\overline{I}} = \vec{\overline{I}}_2 = \frac{1}{12} \ln (a^2 + 4a^2) = \frac{5}{12} ma^2$ $\vec{\overline{I}}_4 = \frac{1}{12} m (a^2 + a^2) = \frac{1}{5} ma^2$

SUBSTITUTE INTO (2):

 $\frac{H}{6} = \frac{\omega}{\sqrt{6}} \left(-\frac{5}{12} m a^2 \hat{i} + \frac{2}{6} m a^2 \hat{j} - \frac{5}{12} m a^2 \hat{k} \right)$ $\frac{H}{6} = \frac{m a^2 \omega}{12\sqrt{6}} \left(-5 \hat{i} + 4 \hat{j} - 5 \hat{k} \right)$ (3)

(a) $|H_G| = \frac{ma^2\omega}{12\sqrt{6}} \sqrt{25 + 16 + 25} = ma^2\omega \frac{\sqrt{11}}{12}$ (4)

14 = 0,276 +naw

(b) FRUM EQ. (3.24) WE HAVE H. W = |He|W COSB

$$\cos \theta = \frac{H_e \cdot \omega}{H_e \mid \omega} \tag{5}$$

RECALLING (1) AND (3):

 $\frac{H_{6} \cdot \omega = \frac{m\alpha^{2}\omega}{12\sqrt{6}} \left(-5\frac{1}{4} + 4\frac{1}{2} - 5\frac{1}{4}\right) \cdot \frac{\omega}{\sqrt{6}} \left(-\frac{1}{4} + 2\frac{1}{2} - \frac{1}{4}\right)}{72} (6)$ $= \frac{m\alpha^{2}\omega^{2}}{72} (5 + 8 + 5) = \frac{1}{4} m\alpha^{2}\omega^{2} \qquad (6)$

RECALLING (4): $|H_G|\omega = \sqrt{11} ma^2 \omega^2$ (7)

SUBSTITUTING FROM (6) A:12 (7) INTO (5):

 $\cos \theta = \frac{1/4}{\sqrt{11}/12} = \frac{3}{\sqrt{11}} = 0.90453$ $\theta = 25.239$

 $\theta = 25.2^{\circ}$

18.8 GIVEN: SOLID PARALLELEPIPED OF PRIE.
13.7 IS REPLIACED BY HOLLOW ONE MADE
OF 6 THIN METAL PLATES.

FIND: (a) MAGNITUDE OF ANG, MCHENTUM HG.

WE DENOTE BY I, I I THE PRINCIPAL CENTROICAL MOMENTS OF INERTIAL WE HAVE

 $\omega = \omega \frac{-a\underline{i} + 2a\underline{j} - a\underline{k}}{a\sqrt{6}} = \frac{\omega}{\sqrt{6}} \left(-\underline{i} + 2\underline{j} - \underline{k} \right) \tag{1}$

 $H_{G} = \bar{I}_{\chi} \omega_{\chi} \underline{i} + \bar{I}_{\chi} \omega_{\chi} \underline{i} + \bar{I}_{\chi} \omega_{\chi} \underline{k} = \frac{\omega}{\sqrt{6}} \left(-\bar{I}_{\chi} \underline{i} + 2\bar{I}_{\chi} \underline{j} - \bar{I}_{\chi} \underline{k} \right) (2)$

CCAPUTATION OF MOMENTS OF INERTIA:

EACH OF THE TWO SQUARE PLATES HAS A MASS EQUAL TO MYTO AND EACH OF THE RECTANGULAR PLATES HAS A MASS EQUAL TO MYS. USING THE PARALLEL (1) AXIS THEOLEII WHEN MEELES, WE CETAIN!

	Ī, Ī,	ī,	Ī
SQUARE PLATES	$\frac{2m(a^2+a^2)}{10(12+a^2)} = \frac{13}{60} ma^2$	$\frac{2\pi n}{10} \cdot \frac{a^2}{6} = \frac{m \cdot a^2}{30}$	13 ma
RECTANG. PLATES // yz PLANE	2m a2+4a2	$\frac{2\pi i}{5} \left[\frac{a^t}{12} + \left(\frac{a}{2} \right)^t \right]$ $= \frac{2}{15} ma^2$	$\frac{2m}{5} \left[\frac{\alpha^2}{3} + \left(\frac{\alpha}{2} \right)^2 \right]$ $= \frac{7}{30} m\alpha^2$
REC. PLV 24 PLANE		2 mat	1 max
sums	37 mai	9 m at	37 mai

(3) SUBSTITUTE THE VALUES OBTAINED FOR I, I, I, I, II, III

(4)
$$H = \frac{m \alpha^{2} \omega}{60 \sqrt{6}} \left(-37 \dot{\underline{i}} + 36 \dot{\underline{j}} - 5 \dot{\underline{k}}\right)$$
 (3)

(a)
$$|H_{6}| = \frac{ma^{t}\omega}{60\sqrt{6}}\sqrt{(37)^{2} + (36)^{2} + (37)^{2}} = \frac{ma^{t}\omega}{60\sqrt{6}}\sqrt{4034}$$
 (4)

(b) WE RECALL EQ. (5) IN SOLUTION OF PROB. 18.7:

$$CC \in \Theta = \frac{|H|^{2} |m|}{|H|^{2} |m|}$$
 (5)

RECALLING (1) AND(3) ABOVE :

$$\frac{H}{6} \cdot \omega = \frac{ma^2\omega}{60\sqrt{6}} \left(-37L + 36 \frac{1}{2} - 37 \frac{k}{b} \right) \cdot \frac{\omega}{\sqrt{6}} \left(-\frac{1}{2} + 2\frac{1}{2} - \frac{k}{b} \right)$$

$$= \frac{ma^2\omega^2}{360} \left(37 + 72 + 37 \right) = \frac{146}{360} ma^2\omega^2$$
 (6)

RECALLING (4) APOVE:

$$|H_{6}|\omega = \frac{\sqrt{4034}}{60\sqrt{6}} ma^{2}\omega^{2}$$
 (7)

SUBSTITUTING FROM (6) AND(7) INTO (5):

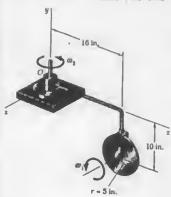
$$\cos \theta = \frac{146}{360} \frac{60\sqrt{6}}{\sqrt{4034}} = \frac{146}{\sqrt{6\times4034}} = 0.93845$$

$$\theta = 20.200^{\circ}$$

0=20.2°

18.9 GIVEN; DISK OF PROB. 18.5 WITH W=816, $\omega_1 = 12 \text{ rad/s}$, AND $\omega_2 = 4 \text{ rad/s}$.

FIND: ANGULAR MOMENTUM HO ABOUT POINT O.



WE USE EQ. (18.11): $H_0 = \overline{z} \times m\overline{v} + H_G$ (1)

WHERE $\overline{z} = \underline{z}_A = (\frac{16}{12}ft)\underline{i} - (\frac{10}{12}ft)\underline{j}$ $\overline{z} = \underline{z}_A = (\frac{4}{3}ft)\underline{l} - (\frac{5}{6}ft)\underline{j}$ $All = \frac{W}{g} = \frac{8}{32.776}$ $= 0.24845 |b.s^2/ft$ $\overline{v} = \underline{v}_A = \omega_e \times \underline{z}_A$ $= (4rad/s)\underline{i} \times (\frac{4}{3}\underline{i} - \frac{5}{6}\underline{i})$ $\overline{v} = -(\frac{16}{3}ft/s)\underline{k}$

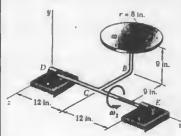
TROM THE SOLUTION OF PROB. 18.5, WE RECALL THAT $H_G = H_A = (0.0431 | b.ft.s) \dot{s} + (0.259 | b.ft.s) \dot{k}$ SUBSTITUTING INTO (1):

 $\frac{H}{0} = \left(\frac{4}{3} \cdot \frac{1}{6} - \frac{5}{6} \cdot \frac{1}{3}\right) \times 0.24845 \left(-\frac{16}{3} \cdot \frac{1}{6}\right) + 0.0431 \cdot \frac{1}{3} + 0.259 \cdot \frac{1}{6}$ $= 1.7668 \cdot \frac{1}{3} + 1.1042 \cdot \frac{1}{6} + 0.0431 \cdot \frac{1}{6} + 0.259 \cdot \frac{1}{6}$

Ho = (1.104 b.ff.s) i + (1.810 b.ft.s) j + (0.259 b.ft.s) k

18.10 GIVEN: DISK OF PROB. 18,6 WITH W = 61b, $\omega_1 = 16 \text{ rad/s}$, AND $\omega_2 = 8 \text{ rad/s}$.

FIND: ANGULAR MOMENTUM HD ABOUT POINT D.



WE USE EQ. (18.11) WITH RESPECT TO D: $H = \frac{1}{2} \times m \overline{V} + H_G$ (1) WHERE $\overline{z} = \frac{1}{2} = (1ft) \dot{i} + (0.75ft) \dot{j} - (0.75ft) \dot{k}$ $m = \frac{W}{g} = \frac{6B}{32.2ft/s^2}$ $= 0.186335 |b.5^2/5t$

 $\overline{\underline{y}} = \underline{y}_{A} = \underline{\omega}_{2} \times \underline{z}_{A} = (8 \operatorname{rad/s})\underline{i} \times (\underline{i} + 0.75 \underline{j} - 0.75 \underline{k})$ $\overline{\underline{y}} = (6 \operatorname{ft/s})\underline{j} + (6 + \frac{1}{5})\underline{k}$

FROM THE SOLUTION OF PROB. 18.6, WE RECALL THAT

HG = HA = (0.1656 16.ft.s) i + (0.663 16.ft.s) i

SUBSTITUTING INTO (1):

HD= (i+0.75j-0.75k) x 0.186335 (6j+6k)+
0.1656 + 0.663 j

= 1.1180 £ -1.1180 j +0.8385 i +0.8385 i +0.1656 i +0.663 j

H = (1.843 16.54.5) i - (0.455 16.51.5) j + (1.118 16.51.5) k

18.11



FROJECTILE WITH ME=30 kg

\$\bar{k}_{\infty} = 60 \text{mm}, \bar{k}_{\infty} = 250 \text{mm}, \\

\$ANGLE \text{0} = -5°, \text{ANG.MCM.} \\

\$\begin{array}{c} \text{H}_{\infty} = (320 g \cdot m^2/s) \dot \dot - (9g \cdot m^1/s) \dot \dot \end{array}

\$RESOLVE \text{20} NOTO COMPONENTS

(a) ALONG GX (RATE OF SPIN)
(b) ALONG GD (RATE OF PRECESSION)

BECHUSE OF AXISYMMETS! OF PROJECTILE, THE 2

AND Y ARES ARE PRINCIPAL CENTROIDAL ARES. $\vec{L} = m \vec{k}^2 = (30 \text{ kg})(0.060 \text{ m})^2 = 0.104 \text{ kg·m}^2$ $\vec{I}_g = m \vec{k}_g^2 = (30 \text{ kg})(0.250 \text{ m}) = 1.875 \text{ kg·m}^2$ GIVEN: $H_z = 0.320 \text{ kg·m/s}$, $H_g = -0.009 \text{ kg·m/s}$ FROM EUS. (18.10): $O = H_z = 0.320 \text{ kg·m/s} = 2.9630 \text{ rad/s}$

 $\omega_{z} = \frac{H_{z}}{\bar{I}_{z}} = \frac{0.320 \text{ kg·m/s}}{0.108 \text{ kg·m/s}} = 2.9630 \text{ rad/s}$ $\omega_{z} = \frac{H_{z}}{\bar{I}_{z}} = \frac{-0.009 \text{ kg·m/s}}{1.875 \text{ kg·m/s}} = -0.00480 \text{ rad/s}$

THUS: \(\omega = (2,9630 rad/s) \overline{i} - (0,00480 rad/s) \overline{j} \)
WE MUST NOW RESOLUTE OF INTO DELIQUE COMPONENTS
ALONG GZ AND GD.



WE NOTE THAT $-\omega_y = \omega_p \sin \theta$ $\omega_p = \frac{-\omega_y}{\sin \theta} = \frac{+0.00489}{\sin 5^{\circ}}$ $\omega_p = 0.055074 \text{ rad/s}$

ω_s = ω_z - ω_p cos 0 = 2.9630 - 0.055074 cos 5 = 2.408 rad/s

ANSWERS: (a) $\omega_s = 2.91 \text{ rad/s}$. (b) $\omega_p = 0.0551 \text{ rad/s}$

18.12 GIVEN: PROJECTILE OF PROF. 18.11.
ADDITIONAL DATA: \$\overline{v} = 650 m/s.

FIND: ANG. MOM. HA. (RESCLUE INTO X, y, 2 COMP.)

RESOLVE TINTO RECTANS. COMP. ALONG ZANDY AYES.

TO = (650 m/s)(0055° 1 - 5in5° 1) = (647.53 m/s) 1- (56.65 m/s) 1

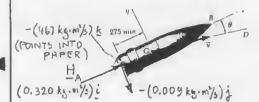
USING EQ. (18.11) AND RECALLING DATA FROM PRUB, 18.11,

H = E x m T + H

 $= (0.275 \text{ m}) i \times (30 \text{ kg}) [(647.58 \text{ m/s}) i - (56.65 \text{ m/s}) i + (0.320 \text{ kg·m/s}) i - (0.009 \text{ kg·m/s}) i$

= -(467.57 kg·m²/s)k +(0.320 kg·m²/s)i-(0.009 kg·m²/s)j

H = (0,320 kg·m/s) i - (0.009 kg·m/s) j - (467 kg·m/s) h



18.13

(a) Show that the ungular momentum H, of a rigid body about point B can be obtained by adding to the angular momentum HA of that body shout point A the vector product of the vector rAB drawn from B to A and the linear momentum mv of the body:

$$H_A = H_A + r_{A/B} \times m\overline{v}$$

(b) Further show that when a rigid body rotates about a fixed axis, its angular momentum is the same about any two points A and B located on the fixed axis (HA = HB) if, and only if, the mass center G of the body is located on the fixed axis.

(a) USING EQ. (18.11) TO DETERMINE H AND THEN

$$\underline{H}_{A} = \underline{F}_{G/A} \times m \overline{v} + \underline{H}_{G} \tag{1}$$

$$H_B = \underline{\iota}_{G/B} \times m \, \overline{\upsilon} + H_C \tag{2}$$

SUBTRACTING (1) FRUM(2)

A
$$\frac{2\epsilon_{A/A}}{B}$$
 G $\frac{H_{e} - H_{e}}{H_{e}} = (\frac{\epsilon_{G/E} - \epsilon_{G/E}}{\epsilon_{A/B}})^{\frac{1}{4} \cdot \frac{1}{4}}$
 $\frac{2}{4}$

B UT $\frac{2}{2}$
 $\frac{2}{6}$

B UT $\frac{2}{6}$
 $\frac{2}{6}$

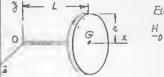
(b) IT FILLOWS FRIM EN. (3) THAT HA = HR IF AND CIVILY IF. TAIR XMV = U (4)

BUT, DENOTING BY 2 AB THE UNIT VECTOR ALUNG THE FIXED AXIS, WE HAVE TE WE AND X & GAS FO (4) YIELDS & MI (W] A/B X EG/A)

WE NOTE THAT END IS PERPENDICULAR TO WANX YOU AND, THUS, NOT PARALLEL TO IT. THEREPORE, THIS SECOND VECTOR MUST BE ZERD, WHICH WILL OCCUR IF EG/A IS PARALLEL TO PAB, THAT IS, IF, AND ONLY IF, G IS LUTHTED ON AB.

18.14 GIVEN: DISK OF SAMPLE PROB. 18.2 AND ANSWERS TO PART & OF THAT PROBLEM:

FIND: ANG. HOMENTUM HOUSING ER. (18.11), AND VEKIFY THAT RESULT IS SAME AS IN PAKT LOFS, P. 18.2



EQ. (18.11): H = E XMD + H, = Lixmrw, k+ 1 112 W (i- 2)

H = - mr L w j + 1 mr w i - 1 mr w i + 5 = 1 n1 2 w, i-n12 & w, j - 4 m 2 2 w, à $H = \frac{1}{2} m x^2 \omega_1 i - m (L^2 + \frac{1}{4} z^2) (i \omega_1 / L) j$

WHICH IS THE ANSWER OBTAINED IN PART L OF SAMILE PROP. 18.2.

18.15

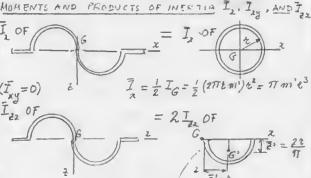
GIVEN:

SHAFT OF MASS M, MADE OF KOD OF UNIFORAL CROSS SECTION ROTATE WITH LONSTANT ANG. VEL. W.

(A) ANG. MOM. H. H AND AXIS AB.

MASS SER HAIT STORTH = MI' = AM E + TTZ+TZ+Z Z(T+1)Z

MOHENTS AND PRODUCTS OF INTERTIA I2. Izy , AND IZZ



THUS: I = 2 (AI'TE) Z' E' = 2 AI'TIL (2)(22) = 4 m'.43

(a) ANGULAR MOMENTUM HA

WE USE EQS. (18:1) SINCE PO_ED, Dy = D = D, WE HAVE H,= I, W = 170120 (1)

$$H_y = -\overline{1}_{xy} \omega = 0$$

$$H_z = -\bar{I}_{zz} \omega = -4 \, \text{m}^2 t^3 \omega \tag{2}$$

$$\frac{H_{G}}{H_{G}} = H_{\chi} i + H_{\chi} k = m' \epsilon^{3} \omega \left(\pi i - 4 k \right)$$

OR, RECALLING THE EXPRESSION OBTAINED FOR m';

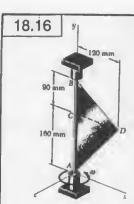
$$\frac{H}{G} = \frac{m_1 z^3 \omega}{2(\pi + i)^2} (\pi_1 - 4 \underline{k}) =
= m z^2 \omega \left[\frac{\pi}{2(\pi + i)} - \frac{2}{\pi + i} \underline{k} \right]
\underline{H} = m z^2 \omega \left(0.379 i - 0.483 \underline{k} \right)$$

(b) ANGLE FORMED EY H, AND AXIS AL.

DENOTING BY & THAT ANGLE, WE HAVE $tan \theta = \frac{|H_2|}{H_1}$

AND, RECALLING (1) AND (2):

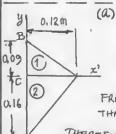
$$\tan\theta = \frac{4m^2 \pm 3\omega}{\pi n^2 n^3 \omega} = \frac{4}{\pi}$$
 $\theta = 51, 9^\circ$



TRIANGULAR PLATE SHOWN HAS MASS m = 7.5 kg AND IS WELDED TO SHAFT AB. PLATE RUTATES AT CONSTANT RATE W= 12 rad/s.

FINDI

(a) ANG HOMENTUM H (b) ANG. MOMENTUM HA (FIND I AND USE PROPERTY INDICATED IN PROB. 18.13a.)



(a) WE DIVIDE PLATE INTO TWO RIGHT TRIANGLES AND COMPUTE THER PRODUCTS OF INERTIA. 11 = 1 (7.5 kg) = 2.7 kg $m_z = \frac{16}{27} (7.5 \text{ kg}) = 4.8 \text{ kg}$

FROM SAMPLE PROB. 9.6, WE RECALL

Ixy, MASS = m (1 b'h) = mbh

TRIANGLE 1: (I2) = 12 (2.7 kg)(0.12 m)(0.04 m) = 2.43×10 TRIANGLE 2: (Izy) = 1/2 (4.8 kg)(0.12 m)(-0.16 m)=-7.68 ×10 THUS, FOR THE PLATE, Iz = 6.43-7.68) 10 = -5.25 × 10 /g m Izin = -5,25 g. 02 WE NOTE THAT INT = 0.

MOHENT OF INTETIA T OF ENTIRE PLATE In reen = 12 bh, Iy, Mas = m (12 bh) = 6 mh2

In = 1 (7.5 kg) (0.12m) = 0.018 kg·m = 18g·m

ANGULAR MOMENTUM HC

WE USE EQS. (18.13) TO DETAIN THE COMPONENTS Hx, Hy, Hz, OF H

 $H_{2'} = -I_{2'} \omega = -(-5.25 g.m^2)(12 rad/s) = +63.0 g.m^2/s$ $H_g = I_g \omega = (18 \text{ g·m}^2)(12 \text{ rad/s}) = 216 \text{ g·m}^2/\text{s}$ H_2 , = $-I_{22}$, $\omega = 0$

Hc = (63.0 g·m/s)i+(216 g·m/s)j

(b) ANGULAR MOHENTUM HA.

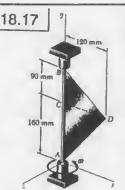
WE APPLY THE EQUATION GIVEN IN PARTA OF PROB. 18.13 TO POINTS A AND C.

H = Hc + 2c/A X m 5

WHERE 1/2/A- (0.16 m) j. NOTING THAT THE DISTANCE FROM THE AXIS OF ROTATION AB TO THE MASS CENTER G OF THE PLATE. 15 = 1 (0,12 m) = 0,04 M, WE HAVE m v = m (wx t) = (7,5 kg)(12 rad/s); x (0,04m) i =-(3.60 kg·m/s)k = -(3600 g·m/s)k

をc/A×mサ=(0.16m)jx(-3600g·m/s)k=-(576g·m/s)し SUBSTITUTING FOR HC AND ECIAXMID INTU (1):

HA = - (513g·m3/s): + (216g·m3/s)}



GIVEN'

TRIANGULAR PLAJE SHOWIN HAS MASS m = 7.5 kg AND IS WELDED TO SHAFT HE. PLATE, RATRITS AT CONSTANT K-TE W = 12 rad/s.

(a) ANG. MOHENTUM H (6) ANG. MOMENTUN HB (FIND I AND USE PROPERTY INDICATED IN PROP. 18.12 a.)

(a) SEE PART & OF SOLUTION OF TROB. 18.16. WE FIND

ANG. MOMENTUM H = (63.0g. mi/s) i + (216g. m/s) }

(6) ANG. MOHENTING H .

WE APPLY THE ENGATION SIVEN IN PORT & OF PRIB. 18.13 TO FOINTS B AND C .

(1) H= Hc++c/8×1115

WHERE 120/B = - (0.09 m) }.
NOTING THAT THE DISTANCE FLOOR THE AXIS OF FOTHTION AE TO THE HASS CENT LG OF THE PLATE IS

E= 1 (0.12m) = 0.04 m

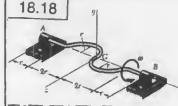
WE HAVE

mv = m (w x 3) = (7.5 kg)(12 rod/s) 2 x (0.0+m) 2 = - (3.60 kg·m/s) k = - (3600 g·m/s) k

2c/8 × mi = - (0.09m)j × (-3600 g. m/s) k = + (324 g. mb/s) =

SUBSTITUTING FUR H AND EC/exmit INTO (1);

HB = (387 gini/5) + (216 gini/5) }



GIVEN:

SHAFT OF PROJ. 18.15

FIND:

ANG. MOM. DF SHAF? (a) AROUT A

WE FIRST DETERMINE Ha. SEE SCLUTION OF FRIE. 18.15. WE FOUND

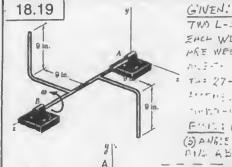
FROH EQ. (18.11) WE HAVE

HA = EGAXMV+H H= + 1/2 X 11 + + +

BUT W=O SINCE G IS LOCATED ON AXIS AB. THUS!

(a) AND (b): H=H=H=1112W (0.37) i-0.182 k)

NOTE . THE RESULT CETAINED VELIFIES THE PROPERTY INDICATED IN PEOP. 12.13 b, NAMELY, THAT IF THE THEM CENTER & OF A BODY ROTTING MELLI A FILE. NO A 13 LOCATED ON THE AVIS, THE ANGULAL MENTIONING IS THE SPITE AROUS ANY I'M POINTS OF THE LAYIS.



TWO L-SHATED PHIX,

ZHEL WEIGHING 5th.

PRE WEIGHEL AT THE

ALETTIC TO THE

TOTAL TO THE TOTAL

TO THE TOTAL

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MUMENTS AND PRODUCTS OF INTERTIA

FOR EACH NUMBERED ELEMENT: $a = 9 \ln = 0.75 ft$. $m = \frac{1}{2} (51b)/g = 2.5/g$

FOR 1 AND 4: I = I + md = 1 ma+m(a+a) = 4 ma

FOR 2 AND 3: I = 1 ma'

FOR ASSEMBLY: $I_2 = 2\left(\frac{4}{3} \min^2 + \frac{1}{3} \min^2\right)$ $I_3 = \frac{10}{3} \min^2$ PRODUCTS OF INERTIA OF ASSEMBLY:

$$\begin{split} I_{22} &= (I_{22})_1 + (I_{22})_2 + (I_{23})_3 + (I_{23})_4 \\ &= \pi i \left(-a/(2a) + \pi i \left(-\frac{a}{2}\right)(2a) + \pi i \left(\frac{1}{2}\right)a + \pi i \left(i\right)a = -\frac{g}{2}\pi i a^2 \\ I_{22} &= \pi i \left(\frac{a}{2}\right)(2a) + 0 + 0 + \pi i \left(-\frac{a}{2}\right)a = \frac{1}{2}\pi i a^2 \end{split}$$

(a) ANGULAR HOHENTUH ABOUT H

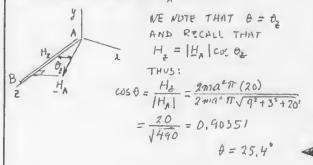
WE USE EQ. (18.13) TO OBTHIN THE COMPONENTS OF HA. WE HAVE WZ = W = 360 ppm = 6(217) = 12 17 rolls, WZ=WZ = 0.

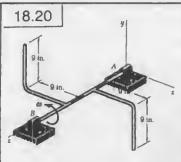
$$\begin{aligned} & H_{z} = -I_{zz} \, \omega_{z} + \frac{1}{2} \, ma^{z} (12\pi) = 18 \, ma^{z} \pi \\ & H_{y} = -I_{yz} \, \omega_{z} = -\frac{1}{2} \, ma^{z} (12\pi) = -6 \, ma^{z} \pi \\ & H_{z} = I_{z} \, \omega_{z} = \frac{10}{2} \, ma^{z} (12\pi) = 40 \, ma^{z} \, \pi \end{aligned}$$

THUS: $H_{A} = H_{2}i + H_{3}i + H_{2}k = 2ma^{2}\pi(9i - 3j + 20k)$ = $2(\frac{2.5}{32.7}/b.s^{3}/4t)(0.75ft)^{2}(\pi rad/s)(9i - 3j + 20k)$ = (0.2744 /b.ft.s)(9i - 3j + 20k)

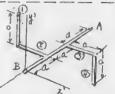
HA = (2.47 16. ft.s) i - (0.823 16. ft.s) j + (5,49 16. ft.s) k

(b) ANGLE & FORMED BY HA AND AB





GIVEN:
TWO L'SHAPED ARMS,
EACH WEIGHING 51b,
ARE WELDED AT THE
ONE-THIRD POINTS OF
THE 27-in. SHAFT AB.
ASSEMBLY RUTATES AT
CONSTANT 360-CPM RATE.
1-IND: (a) H
(b) ANGLE PORHED BY H
B
AND BA.



WE WILL USE AXES X, 3', 2 WITH

ORIGINAT B.

MOHENTS AND PRODUCTS OF INEKTITH

FOR EACH NUMBERED ELEMENT:

a = 9 in = 0,75 ft

m = \frac{1}{2}(516)/g = 2,5/g

FOR 1 AND 4; $I_2 = \overline{I} + md^2 = \frac{1}{Z}ma^2 + m\left(\frac{\alpha}{4} + a^2\right) = \frac{4}{3}ma^2$ FOR 2 AND 3: $I_2 = \frac{1}{3}ma^2$ FOR ASSENBLY: $I_2 = 2\left(\frac{4}{3}ma^2 + \frac{1}{3}ma^2\right)$ $I_2 = \frac{10}{3}ma^2$ PRODUCTS OF INTERTIA OF ASSEMBLY:

$$\begin{split} \mathbf{I}_{\mathbf{x}^{1}2} &= \left(\mathbf{I}_{\mathbf{x}^{1}2}\right)_{1} + \left(\mathbf{I}_{\mathbf{x}^{1}2}\right)_{L} + \left(\mathbf{I}_{\mathbf{x}^{1}2}\right)_{3} + \left(\mathbf{I}_{\mathbf{x}^{1}2}\right)_{4} \\ &= n(-a)(-a) + m\left(-\frac{a}{2}\right)(-a) + m\left(\frac{a}{2}\right)(-2a) + m(a)(-2a) = \frac{3}{2} ma^{2} \\ \mathbf{I}_{\mathbf{y}^{1}2} &= m\left(\frac{a}{2}\right)(-a) + 0 + 0 + m\left(-\frac{a}{2}\right)(-2a) = \frac{1}{2} ma^{2} \end{split}$$

(2) ANGULAR MOMENTUM ABOUT B

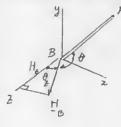
WE USE EWS. (18,13) TO OBTAIN THE COMPONENTS OF HB. WE HAVE $\omega = \omega = 360 \, \text{rps} = 6(2\, \text{N}) = 12 \, \text{Trad/s}, \ \omega_z = \omega_z = 0.$ H, $z = -I_{z_1 z_2} \omega_z = + \frac{2}{z_1} m \alpha^2 (12\, \text{T}) = 10 \, \text{ma}^2 \text{T}$ Hy, $z = -I_{y_1 z_2} \omega_z = \frac{10}{3} m \alpha^2 (12\, \text{T}) = -6 \, \text{ma}^2 \text{T}$ H₂ = $I_{z_1} \omega_z = \frac{10}{3} m \alpha^2 (12\, \text{T}) = 40 \, \text{ma}^2 \text{T}$

THUS: $H = H \cdot \dot{i} + H_y \cdot \dot{j} + H_z \dot{k} = 2 ma^{2} (9i - 3j + 20 \dot{k})$ = $2(\frac{2.5}{37.7} \cdot b \cdot s'/ft)(0.75ft)^{2} (\pi \cdot radb)(9i - 3j + 20 \dot{k})$ = $(0.2744/b \cdot ft \cdot s)(9i - 3j + 20 \dot{k})$

HB = (2.47 16.51.5) i - (0.823 16.4.5) j + (5.49 16.51.5)k

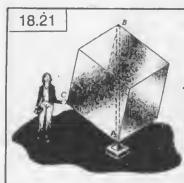
NOTE. THIS IS THE SAME ANSWERTHAT WAS DETAINED FOR HA
IN PRUB. 18, 19, THIS COULD HAVE BEEN ANTICIPATED, SINCE
THE MASS CENTER & OF THE ASSEMBLY LIES ON THE FIXED
AXIS AB (CF. PROB. 18.13 6).

(b) ANGLE & FORMED BY HB AND BA



AND RECALL THAT $H_2 = |H_2| \cos \theta_2$ THUS: $\cos \theta = \cos(\pi - \theta_2) = -\cos \theta_2$ $= -\frac{H_3}{|H_B|} = -\frac{2ma^3\pi}{2ma^3\pi\sqrt{q^3+2^3+20^2}}$ $= -\frac{20}{\sqrt{490}} = -0.90351$ $\theta = 154.6^\circ$

WE NOTE THAT & = TI - D,



HOLLDIN CUBE CONSISTS OF SIX 5x5 ft ALUMINUM SHEETS AND CAMPAGE 18 ABIDIT VENTICAL DIAGONAL AB. STUDENT PUSHES CORNER C FOR 1,25 IN DIRECTION PERPENDICULAR TO PLANE ABC WITH FORCE OF 12,5 lb, CAUSING CUBE TO COMPLETE 1 REK IN 55. FIND: WEIGHT OF CUBE.

HINT: PERP. PISTANCE FI " C TO AB IS O. 12/3. WHERE a 15 SIDE OF CUBE.



FOR CUBE, IAB = I SINCE THE ELLIBOID OF INTRIA AT G 15 A SPHERE (SEC. 9.17). FOR THE TWO HORIZONTAL FACES $(I_{DD'})_{H} = 2(\frac{m}{6})(\frac{\alpha^{2}}{6}) = \frac{m\alpha^{2}}{19}$

WHERE MI = MAS'. OF CUBE FOR THE FOUR VERTICAL PACES $\left(I_{pp}\right)_{y} = 4\left(\frac{m}{6}\right)\left[\frac{a^{2}}{12} + \left(\frac{a}{2}\right)^{2}\right] = \frac{2 ma}{a}$

FOR THE WHULE CUBE:

$$I_{AB} = I_{DD} = \left(I_{DD}\right)_{H} + \left(I_{DD}\right)_{Y} = \frac{ma^{3}}{1B} + \frac{2ma^{3}}{q} = \frac{5}{1B} ma^{2}$$

IMPULSE-MOMENTUN PRINCIPLE

ANG, IMPULSE ABOUT AB = FINAL ANG, MOMENTUM ABOUT AB (F Dt) a \2/3 = 5 ma2 W

GIVEN DATH: F= 12,5 16, Ot = 1,25, a=5 ft, w= 217 ad SUBSTITUTE DATA AND m= W/g INTO (1):

 $(12.5 \text{ lb})(1.2 \text{ s})(5 \text{ ft})\sqrt{\frac{2}{3}} = \frac{5}{18} \frac{W}{32.2 \text{ ft/s}} (5 \text{ ft})^{2} \left(\frac{2 \text{ ft rad}}{5 \text{ s}}\right)$

SOLVING FOR W: W = 225,96 16

W= 226 16

18.22 GILEN: ALUMINUM CUBE OF PROB. 18.21 16 REPLACED BY CUBE CONSISTING OF SIX

PLYWOOD SHEETS, WELGHING 20 ID ENCH. STUDENT PUSITES CORNER C AS IN PROB. 18.21 (FOR 1.25 WITH 12,5 - 15 FORCE).

FIND: TIME REQUIRED FOR CUBE TO COMPLETE ! REV.

SEE SOLUTION OF PROB. 18,21 FOR DERIVATION OF (Fat) a \(\frac{7}{3} = \frac{5}{10} ma' \omega)

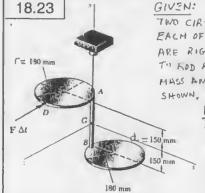
GIVEN DATA: F = 12.5 lb, DE = 1.25, a = 5ft $m = \frac{W}{3} = \frac{6(20\%)}{32,2\%} = 3.727 \% \cdot 5^{\circ}/fl^{-}$

SUBSTITUTE DATA INTO (1): $(12.5 \text{ b})(1.2 \text{ s})(5 \text{ ft})\sqrt{\frac{2}{3}} = \frac{5}{18}(3.727 \text{ b} \cdot \text{s}^{2}/\text{ft})(5 \text{ ft})^{2} \omega$

SOLVING FOR W: W = 2.366 5

$$\mathcal{C} = \frac{2\Pi}{\omega} = \frac{2\Pi}{2.3665^{-1}} = 2.65565$$

2 = 2.665



TWO CIRCULAR PLATES. EACH OF MASS IT = 4 Kg, ARE RIGIDLY CONNECTED TO LOD AB OF NEGLIGIBLE MASS AND SUSPENDED AS SHOWN, AN IMPULSE

FAt = - (2,4 N.S)k IS APPLIED AT D. FIND:

(a) VELOCITY V OF MASS CENTER G. (6) ANGULAR VELOCITY W OF ASSEMBLY.

COMPUTATION OF MOMENTS AND PRODUCTS OF INTERTIA FOR UPPER PLATE:

1 = 1, + md = m (1/4 t + d) = (4 kg)[1/4 (0.18 m) + (0.15 m)]

Iy = Iy, + me = + me + ne = = = ne = = = (4 hg)(0.18 m)] = 0,1944 Kg·m2

 $I_2 = \overline{I}_{2,1} + m(z^2 + d^2) = \frac{1}{L}mz^2 + m(z^2 + d^2) = m(\frac{5}{4}z^2 + d^2)$ = (4 kg) [2 (0,18 m) + (0,15 m)] = 0,252 kg·m

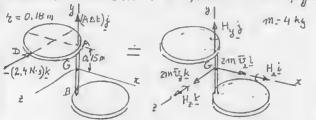
 $I_{xy} = m(-t)(d) = -mt d = -(4kg)(0.18m)(0.15m)$ $I_{xy} = -0.108 kg \cdot m^{2}, \quad I_{yz} = 0, \quad I_{zz} = 0$

FOR LOWER PLATE: WE OBTAIN THE SAME RESULTS THUS, FOR ASSEMBLY: WE DUVBLE RESULTS FOR UPPER

I = 0.2448 kg·m², I = 0.3880 kg·m², I = 0.504 kg·n² Izy=-0.216 kg·m2, Iy= 0, Izx=0

IMPULSE-MUMENTUM PRINCIPLE

WE NOTE THAT THE IMPULSIVE FORCES ARE F AND. PRINCIPLY, THE FORE AT A. ALSO, FRANCON STRAINTS, U.= 0.



(a) VELUCITY OF MASS CENTER . EQUATE SUMS OF - (2.4 N.5) k+ (A Dt) = 2(4 kg) (V2 i + V3 k) VECTURS:

THUS: Abt =0, V=0, V=-0.3 m/s

V=-(0.300 m/s) k (L) ANGULAR VELOCITY, EQUATE SUMS OF MONENTS ABUUT G:

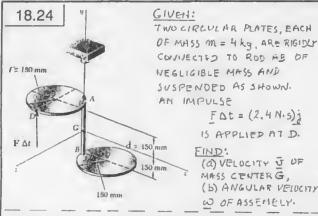
[(-0.1011)i+(0.15m)j]x(-2.4 N·s)k= Hi + H, j + H, b -(0.432 kg·m²)j-(0.360 kg·m²)¿ = Hx i + Hy j + H, k $H_{s} = -0.360$, $H_{s} = -0.432$, $H_{s} = 0$

SUBSTITUTE FROM (1) AND (2) INTO EUS. (18.7): $\begin{aligned} H_1 &= \bar{I}_1 \omega_1 - \bar{I}_2 \omega_1 : -0.360 = +0.2448 \omega_1 + 0.216 \omega_2 \\ H_2 &= -\bar{I}_1 \omega_1 + \bar{I}_1 \omega_1 - \bar{I}_2 \omega_2 : -0.432 = +0.216 \omega_2 + 0.3888 \omega_3 \\ H_3 &= -\bar{I}_1 \omega_1 - \bar{I}_2 \omega_2 + \bar{I}_3 \omega_2 : 0 = \omega_2 \end{aligned}$ (4)

SULVE (3) AND (49: W= -0.96154, W= -0.57692

w=-(0.962 rad/s)i-(0.577 rad/s)}

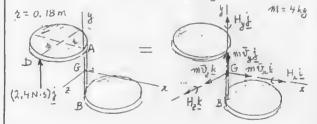
15)



COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA SEE SOLUTION OF PKOB. 18.23 WHERE WE FOUND I = 0.2448 kg·m, 1, = 0.3888 kg·m, I = 0.504 kg·nt In = -0.216 ky. 11, In = 0, In

IMPULSE- MUMENTUM PRINCIPLE

WE NOTE THAT THE CORD AT A WILL BECOME SLACK THUS, THE ONLY IMPULLIVE FORCE IS F.



(a) VELOCITY OF MASS CENTER

EQUATE SUMS OF VECTORSI

THUS: $V_{\chi} = 0$, $V_{\chi} = 0.300 \,\text{m/s}$, $V_{Z} = 0$ V = (0,300 m/s) }

(b) ANGULAG /ELOZITY

EQUATE SUMS OF MOHENTS ABOUT G:

[(-0.18 m)=+(0.18m)k|x(2.4 1/15);= Hz=+Hz=+Hz=+Hz=+ (0.432 kg.m2)(-k-i) = H2 (+ Hy i + H2 K

THUS: Hx = -0.432 kg·m², Hy = 0, Hz = -0.432 kg·m² (2) (a) VELOCITY OF HASS CENTER SUBSTITUTE TERM (1, AND (2) INTO ERS. (18.7):

H, = ID) - I, Wy - I, 02: -0.432=+0.2448W, +0.216Wy

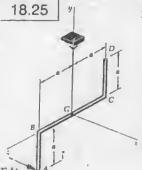
H3=- 1, Wx + 1, Wy - 1, Wy. 0 = + 6.216 0, +0.388803

H, = -I, W, -I, Wy + I, Wy: -0.432 = +0.504 Wz

SOLVING (3) AND (4) SIMULTANEOUSLY, Wz = -3.4616 rad/s, Wy = 1.9231 rails

SULVING (5) FOR Wz: Wz = -0.8571 rad/s THUS:

w = - (3.46 rad/s) i + (1.923 rad/s)j - (0.857 rad/s) k €



GIVEN:

UNIFORM BENT ROD OF MASS M IS SUSPENDED AS SHOWN. ROD IS ITHT AT A WITH IMPLIESE FAL IN DIRECTION PERPENDICIAR TO PLANE CUNTAINING ROD.

IMMEDIATELY AFTER IMPACT (a) VELOCITY TO OF MASS CENTER

(P) ANGULAS VELOCITY W

COMPUTATION OF MOHENTS AND PRODUCTS OF INERTIA

PORTION BC: (1,) = (1,) BC = 1/2 (20) = 1/6 ma, (7) = 0 (Izz) = (Jz) = (Izz) = 0

PORTIONS AB AND CD:

(12) = (12) = I + (1) d= 1 (1) a + 1 (1) a + 1 (1) = 3 ma $(I_z)_{AB} = (I_z)_{CD} = \frac{m}{4}a^z, \quad (I_z)_{AB} = (I_z)_{CD} = \frac{1}{3}\frac{m}{4}a^z = \frac{1}{12}ma^z$

THE MOMENTS AND PRODUCTS OF INERTIA OF THE ROD ARE OBTAINED BY ADDING THE ABOVE VALUES:

$$\vec{I}_{2} = \frac{1}{6} m \dot{a} + \frac{1}{3} m \dot{a} + \frac{1}{3} m \dot{a} = \frac{5}{6} m \dot{a}^{2}$$

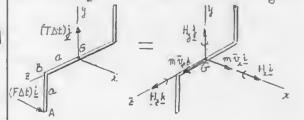
$$\vec{I}_{3} = \frac{1}{6} m \dot{a}^{2} + \frac{1}{4} m \dot{a}^{2} + \frac{1}{4} m \dot{a}^{2} = \frac{2}{3} m \dot{a}^{2}$$

$$\vec{I}_{2} = 0 + \frac{1}{12} m \dot{a}^{2} + \frac{1}{12} m \dot{a}^{2} = \frac{1}{6} m \dot{a}^{2} = -\frac{1}{4} m \dot{a}^{2}, \quad \vec{I}_{22} = 0$$

$$\vec{I}_{23} = 0, \quad \vec{I}_{32} = 0 - \frac{1}{6} m \dot{a}^{2} = -\frac{1}{4} m \dot{a}^{2}, \quad \vec{I}_{22} = 0$$

$$IMPULSE-MOMENTUM PRINCIPLE$$

THE IMPULSES CONSIST OF FAL = (FAL) AND, POSSIBLY, AN IMPULSE (TAH) AT G. BECAUSE OF CONSTRAINTS, Ty = 0.



EQUATE SUMS OF VECTORS: (FAL) L+ (TOE) = mvi + mvi k THUS: $\overline{V}_{x} = (F\Delta t)/m$, $\overline{v}_{z} = 0$, $T\Delta t = 0$ $\underline{\overline{V}} = (F\Delta t/m)\underline{i}$

(4) (b) ANGULAR VELOCITY

FOUNTE MOMENTS ABOUT G:

(-aj +ak) x (FDt) i = H, i + H, j + H, t a Fat (k+j) = Hz + Hy j + Hz k

THUS: H2 = 0, H4 = a F DE, H1 = a F DE (2)

SUBSTITUTE FROM (1) AND (2) INTO EQS. (18.7):

Hz= I, Wz-1, Wz-1, Wz: (3) 0= } 110 W (1)

H=-I, W,+I, W,-I, W: aFat= == maw ++ maw, Hz=-122 W2-1202+ 1W2: aFbt= 4 mawy + 6 maduz (5)

(CONTINUED)

18.25 continued

WE REPEAT THE FOLLOWING EQS. :

W2=0 (3)

(4)

(5)

aFot = 3 man + 4ma wz

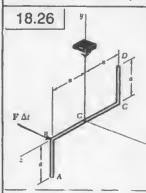
a Fot = ma wy + 2 ma wz

SOLVING (4) AND (5) SIMULTANEOUSLY, WE OBTAIN

$$\omega_{z} = -\frac{12}{7} \frac{F \Delta t}{ma}$$

THUS:

 $\omega = (12 F\Delta t / 7 ma)(-\frac{1}{2} + 5 K)$



GIVEN:

UNIFORM BENT ROD OF MASS M IS SUSPENDED AS SHOWN. ROD IS HIT AT B WITH IMPULSE FAL IN DIRECTION PERPENDIC-ULAR TO PLANE CONTAINING ROD.

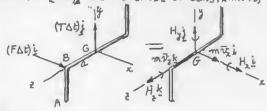
FIND:

IMMEDIATELY AFTER IMPACT (a) VELUCITY OF HASS CENTER (b) ANGULAR VELOCITY &

MOMENTS AND PRODUCTS OF INERTIA

SEE SOLUTION OF PROB. 18.25. WE OBTAINED I=zma, I=zma, 1=zma, I=-4ma, I=I=0 IMPULSE-MOMENTUM PRINCIPLE

THE IMPULSES CONSIST OF FAT = (FAT) + AND, POSSIBLY, AN IMPULSE (TDE); AT G. BECAUSE OF CONSTRAINTS, 1 = 0.



(a) VELOCITY OF HASS CENTER

EQUATE SUMS OF VECTORS: (FDE) i + (TDE) j = m v i + m v k

THUS: $\bar{v}_z = (F\Delta t)/m$, $\bar{v}_z = 0$, $T\Delta t = 0$

V= (FAt/m)i

(6) ANGULAR VELOCITY

EQUATE MOMENTS ABOUT G:

akx(FAt)i = Hi+Hj+H, k

(aFAb) = Hz 1 + Hy 1 + Hz K

THUS: H2 = 0, H4 = aFAt, H2 = 0

SUBSTITUTE FROM (1) AND(2) INTO EQS. (18.7):

(3) H= Iw,-I, w, - I, w: 0=5 maw

Hy=-I, w2+ 1, w3-1,2w2: aFDt = = maw, -4 11aw2 (4)

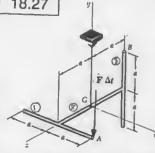
(5) $H_2 = -I_2, \omega_2 - I_1, \omega_2 + I_2, \omega_2$; D===mawy + + maw

SOLVING (4) AND (5) SIMULTANEOUSLY, WE OBTAIN

$$\omega_2 = -\frac{36}{7} \frac{FDt}{ma}$$

THUS:





GIVEN:

THREE RODS, EACH OF MASS M AND LENGTH 20 ARE WELDED TO FORM ASSEMBLY. ASSEMBLY IS HIT VERTICALLY AT A AS SHOWN.

FINDI

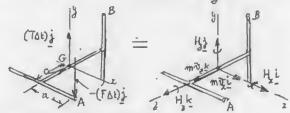
IMMEDIATELY APTER IMPACT (a) VELOCITY OF MASS CENTER (B) ANGULAR VELOCITY & OF ASSEMBLY.

COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA

 $I_{z} = (I_{z})_{1} + (I_{z})_{2} + (I_{z})_{3} = ma^{2} + \frac{1}{3}ma^{2} + m(a^{4} + \frac{a^{4}}{3}) = \frac{8}{3}ma^{3}$ Iy = (Iy), + (Iy), + (I1), = m (a+ + + + + + + + + ma+ ma = + ma" (1) $\bar{I}_{a} = (\bar{I}_{a})_{1} + (\bar{I}_{a})_{2} + (\bar{I}_{a})_{3} = \frac{1}{3} ma^{2} + 0 + \frac{1}{3} ma^{4} = \frac{2}{3} ma^{2}$ $I_{24} = 0$, $I_{32} = 0$, $I_{22} = 0$

IMPULSE- MOMENTUM PRINCIPLE

THE IMPULSES CONSIST OF - (FAT) A ARCHED AT A AND (TAT) APPLIED AT G. BECAUSE OF CONSTRAINTS, Dy = 0.



(a) VELOCITY OF MASS CENTER

EQUATE SUMS OF VECTORS: (TAH) - (FAt) = m TL + M V & THUS: TAL = FAL, Ux = 0, VJ = 0. SINCE U= 0 FROM ABOVE,

(b) ANEULAR VELOCITY

EQUATE HOMENTS ABOUT G:

(ai+ak)x (-Fat) = Hi + Hi + His - (aFAt) + (aFAt) = Hei+ Hy i+ Ho

THUS: H_= aFat, H_= 0, H_= - aFat (2)

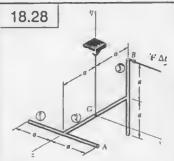
SINCE THE THREE PRODUCTS OF INTERTIA ARE ZERO, THE X, Y, AND & AXES ARE PRINCIPAL CENTROID AL AXES AND WE CAN USE EWS. (18. 10). SUBSTITUTING FROM (1) AND (2) INTO THESE EQUATIONS, WE HAVE

$$H_g = I_{\omega_g} : 0 = \frac{1}{2} \operatorname{ma}^* \omega_g = 0$$
 (4)

$$H_z = I_3 \omega_3$$
: $-a F \Delta t = \frac{2}{3} m d$ $\omega_3 = -3 F \Delta t / 2.71 a$ (5)

40

THEREFORE!



THREE RODS, EACH OF MASS ON AND LEUGTH 20 FAL ARE WELDED TO FORM ASSEMBLY, WHICH IS HIT AT B IN DIRECTION OPPOSITE TO X AYIS. FIND:

IMMEDIATELY AFTER IMIAKT (a) VELOCITY OF MASS CENTER, (b) ANGULAR VELOCITY W.

COMPUTATION OF MOMENTS AND PRODUCTS OF INTERTIA

$$\vec{J}_{z} = (\vec{I}_{z})_{1} + (\vec{I}_{z})_{2} + (\vec{I}_{z})_{3} = ma^{2} + \frac{1}{3}ma^{4} + m(a^{2} + \frac{a^{2}}{3}) = \frac{E}{3}ma^{4}$$

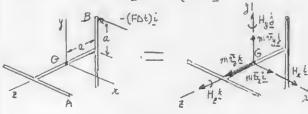
$$\vec{T}_{z} = (\vec{I}_{z})_{1} + (\vec{I}_{z})_{2} + (\vec{I}_{z})_{3} = m(a^{4} + \frac{a^{2}}{3}) + \frac{1}{3}ma^{4} + ma^{3} = \frac{\beta}{3}ma^{4}$$

$$\vec{I}_{z} = (\vec{I}_{z})_{1} + (\vec{I}_{z})_{\alpha} + (\vec{I}_{z})_{\alpha} = \frac{1}{3}ma^{4} + 0 + \frac{1}{3}ma^{6} = \frac{z}{2}ma^{6}$$

$$\vec{I}_{zy} = \vec{I}_{dz} = \vec{T}_{zx} = 0$$
(1)

IMPULSE-MONENTON PRINCIPLE

THE ONLY IMPULSE IS FAT = - (FAt) i.



5=0, 5=0 --- (FΔt/m)L

(6) ANGULAR TICTY

EQUATE MUMERIS - bour 6: (aj-ab) x (-Fot) = H, i + H, i + H, k (aFOt)k+(aFOt)= 11, i+ Hy i+ 12 !

THUS: H=0, Hy=aFat, H= aFat

SINCE THE THEFE INTO THE YOUR THAT FAR ZER THE Z, J, AND E AXES ARE PRIMOPAL CENTER IAL FIRES NIND WE CAN USE EUS. (18.10), SUBSTITUTING FLOM (1) AND (2) INTO THESE THUNTIONS, WE HAVE

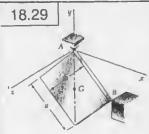
$$H_{\lambda} = \overline{I} \, \omega_{\lambda} : \qquad O = g \, m \, \alpha^2 \, \omega_{\lambda} \qquad \omega_{\lambda} = 0 \tag{3}$$

$$H_2 = \frac{\pi}{2} \omega_0$$
: $\alpha F \Delta r = \frac{\theta}{3} m \alpha^2 \omega_0$ $\omega_0 = 3 F \Delta t / 8 m \alpha$ (4)

$$H_1 = \bar{I}_1 \omega_2$$
: $\alpha F \Delta t = \frac{2}{3} m \alpha^2 \omega_2 = 3 F \Delta t / 2 m \alpha$ (5)

THEREFOREI

ω= (3FΔt/87. 2)(j+4k)



GIVEN:

SQUARE PLATE OF MASS M SUPPORTED BY BALL AND SUCKET WITH ANGU. AR VELOCITY W.S WHEN IT STRIKES DESTITION AT B IN XX PLAIR (2 = 0).

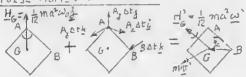
IMMEDIATELY AFTER IMPACT (a) AMP. VELOCITY (I PLATE. (6) VELOCITY OF G.

ANGULAR MOMENTUM

BELPIZE OF SYMPETRY OF SOUARE PLATE, I IS THE SAME ABOUT ANY AXIS THROUGH G WITHIN X & PEANE. (CF. SEC. 9.17), I = 1/2 ma. IT FOLLOWS THAT H = 1/2 ma. (1) FOR ANY W.

VELOCITIES AFTER IMPACT SINCE E = O, CONVER B REMAINS IN CONTRET WITTE BELL ALL 2BA = - cos 45° i + sin 45° j = (-i+i)/12 W'=W' 2BA = W'(-1+2)/12 $\overline{v}' = \omega' \times \underline{z} = \left[\omega' \left(-\underline{i} + \frac{1}{2} \pi v^{\dagger}\right) \times \left(\alpha / \sqrt{2}\right) (-\underline{i}) = \frac{1}{2} \omega' \alpha \underline{k}\right]$

IMPULSE- MORIENT JA: PRINCIPLE



EQUATING MUNERT: ABOUT LINE BAS H cos 42. + 0 = H, + 3 Bb. (Exmin,) RECALLING (1), (2), (3), AND VALUE OF 2BA. $\frac{1}{12}$ ma² ω_0 cos 45° = $\frac{1}{2}$ ma² $\omega^2 + [(-\frac{1}{2} + \frac{1}{2})/\sqrt{2}] \cdot [-\frac{\alpha}{2} + \frac{1}{2} \times \frac{1}{2} \times \omega + \frac{1}{2}]$

12V2 = W'(12+1)

Ca) ANGULAR VELUCITY FROM (2) ANE(4): W= -1+2 (3) 四=中の(-エ+ま)

(b) VELOCITY OF G. TROM (3) AND (4):

1 = 1 . 1 Wo ak = 0,08839 Wak 15'=0.0884 Wak

. 18.30 GIVEN: IMPACT DESCRIBED IN PROB. 18,29. FIND: IMPULSE ON PLATE AT (a) B, (b) A.

SEE SOLUTION OF PROB. 18.29 FOR IMPULSE - MUMENTUM DIAGRAM AND DETERMINATION OF W' AND T'. (a) EQUATING MOMENTS ABOUT A:

1/2 ma woj + a (i-j) x BAtk = 1/2 ma w' - a j x m v' SUBSTITUTING FOR W'AND PERFORMING PRISOCIS 1/2 maw 1 - a BAL(1+1) = 1/2 mat 1 w (-1+1)- 1/2 xm 10.4 k = ma wo (- 96 ! + 96 & - 16 !)

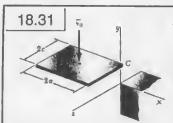
EQUATING THE COEFF. OF L: - a BAt = - 7 ma W

BAL = 0, 10312 mIWOR Bat= 0,1031 mw, ak

(b) EQUATING SUMS OF VECTURS:

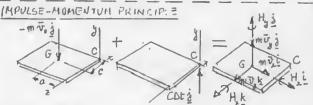
Ant + Bot = my' Abt = my'- Bbt = m(0.088390, ab) - 0.10312mw, ak

ADF=- 0.01473 MIN, a 1=



RECTANGULAR PLATE OF MASS IN FALLING WITH IT AND NO ANG. VELUCITY STRIKES OBSTRUCTION (0=0), FIND:

ANG, VELOCITY OF PLATE



EQUATING MOMENTS ABOUT C:

(-ai+ck) x (-m v j) = (-ai+ck) x m v j + Hzi+Hzi+Hzi

m v (ak+cl) = -m v (ak+ci) + Hzi+Hzi+Hzi+Hzi

since e = 0, Plate Rothtes ABOUT C IMMEDIATELY

AFTER IMPACT

 $\vec{y} = \omega \times \hat{z}_{GC}$: $\vec{q}_{j} = \begin{vmatrix} \hat{u} & \hat{u} & \hat{u} \\ -\hat{u} & \hat{u} & \hat{u} \end{vmatrix} = \hat{u}_{j} ci - (\hat{u}_{j} c + \hat{u}_{j} a) \hat{j} + \hat{u}_{j} a \hat{k}$

EWATE COEFF. OF UNIT VECTURS: $\omega_j = 0$, $\overline{v} = -(\omega_2 C + \omega_2 a)$ (4)

$$H_{x} = \overline{I}_{x} \omega_{x} = \frac{1}{12} m(2c)^{2} \omega_{x} = \frac{1}{3} m c^{2} \omega_{x}$$

$$H_{y} = \overline{I}_{x} \omega_{x} = 0 \quad [BECAUSE OF(4)]$$

$$H_{z} = \overline{I}_{x} \omega_{x} = \frac{1}{12} m(2a)^{2} \omega_{x} = \frac{1}{3} m a^{2} \omega_{x}$$
(5)

SUBSTITUTE PROM (4) AND (5) INTO (3); $m\vec{v}_0(a\underline{k}+c\underline{l}) = +m(\omega_0c+\omega_2a)(a\underline{k}+c\underline{l}) + \frac{1}{3}mc^2\omega_1i + \frac{1}{3}ma^2\omega_2\underline{k}$ $= (\frac{4}{3}mc^2\omega_2 + mac\omega_2)\underline{i} + (\frac{4}{3}ma^2\omega_2 + mac\omega_2)\underline{k}$

DIVIDE BY M AND ERVATE COEFF. OF UNIT VECTORS!

$$\frac{i_1}{3}c^2\omega_2 + ac\omega_1 = \overline{v_0}c$$

$$ac\omega_2 + \frac{i_1}{4}a^2\omega_2 = \overline{v_0}a$$
(6)

SOLVE (6) AND (7) SIMULTANTOUSLY:

$$\omega_{2} = 3\overline{V_{0}}/7c$$
, $\omega_{2} = 3\overline{V_{0}}/7a$ $\omega = \frac{3}{7}\overline{V_{0}}(\frac{1}{6}i + \frac{1}{a}k)$

18.32 GIVEN: IMPACT DESCRIBED IN PROB. 18.31

(a) VELUCITY OF G IMMEDIATELY AFTER IMPACT, (b) IMPULSE ON PLATE DURING IMPACT.

(R) FROM SOLUTION OF PRUB, 18.31: EQS.(1) AND(4): $\vec{U} = \vec{U}_3 \dot{i} = -(\omega_x c + \omega_z a)\dot{i}$ PROM ANSWER TO PROB. 18.31: $\omega_z = \frac{3\vec{V}_3}{7c}$, $\omega_z = \frac{3\vec{V}_3}{7a}$ THUS: $\vec{U} = -(\frac{3\vec{V}_3}{7} + \frac{3\vec{V}_3}{7})\dot{i}$ $\vec{V} = -\frac{6}{7}\vec{V}_3\dot{i}$

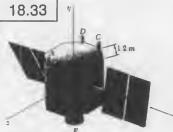
(b) PROM IMPULSE-MOMENTUM DIAGRAM OF PRUB. IB.31; EQUATING SUMS OF VECTORS:

$$-\mu \vec{v}_{0} \vec{j} + C\Delta t \vec{j} = m \vec{v}$$

$$C\Delta t \vec{j} = m \vec{v} + 4 \vec{v}_{0} \vec{j} = -\frac{2}{7} m \vec{v}_{0} \vec{j} + m \vec{v}_{0} \vec{j}$$

$$= \frac{7}{7} m \vec{v}_{0} \vec{j}$$

$$C\Delta t = \frac{1}{7} m \vec{v}_{0} \vec{j}$$



GIVEN: PROBE WITH

11 = 2500 kg, k = 0.98 ml

kg = 1.06 m, kz = 4.02 m;

500 - N HAIN THE PITT E;

20 - N THRUSTENS A,B,C,D

CAN EXPEL FUEL IN y DIR.

PROBE HAS ANG, VELOUTY

W=(0.040 ra4/s) k

FIN'D:

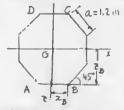
(a) WHICH TWO THRUSTING THE DE PROPER TO REDUCE ON TO ZERO,
(b) OPERATING TIME OF THESE THRUSTERS,

C) HOW LONG SHOULD E BE ACTIVATED IF TO 15 TO 1 E UNICHINA.

INITIAL ANGULAR MUNENTUM

$$\begin{split} &\frac{H_0 = \bar{I}_2 \omega_2 \, \underline{i} + \bar{I}_3 \omega_3 \, \underline{i} + \bar{I}_2 \omega_2 \, \underline{k} = m \left(k_2^2 \, \omega_2 \, \underline{i} + k_3^2 \omega_3 \, \underline{j} + k_2^2 \omega_3 \, \underline{j} \right) \\ &= (2500 \, \text{kg}) \left[0.98 \, \text{m}^3 \, \underline{i}^2 + (0.04 \, \text{rad/s}) \, \underline{i} + 0 + (1.02 \, \text{m})^2 \left(0.06 \, \text{m}^3 \, \underline{i} \right) \, \underline{k} \right] \\ &= (96.04 \, \text{kg} \cdot \underline{m}^3 \, \underline{k}) \, \underline{i} + (156.06 \, \text{kg} \cdot \underline{m}^3 \, \underline{j} + \underline{k}^2 \, \underline{m}^3 \, \underline{k} \right) \end{split}$$

ANGULAR IMPULSE OF TWO 20-N THINSTERS



LET US ASSUME THAT THE CT. C; A MID B WILL BE USED. FROM GEOMETRY OF OCTOGON, $X_{E} = \frac{1}{2}\alpha = \frac{1}{2}(1.2 \text{ m}) = 0.6 \text{ m}$ $R_{E} = \frac{1}{2}\alpha + a \sin 4S^{*} = 1.2011\alpha$ = 1.44853 m $R_{A} = -R_{B}$ $R_{A} = R_{B}$

ANG. IMPULSE ABOUT $G = \underline{\tau}_{A} \times (-F \Delta t_{A}) \dot{f} + \underline{\tau}_{B} \times (-F \Delta t_{B}) \dot{g}$ $= (-x_{B} \dot{i} + \overline{\tau}_{B} \underline{k}) \times (-F \Delta t_{A}) \dot{g} + (z_{B} \dot{i} + \overline{z}_{B} \underline{k}) \times (-F \Delta t_{B}) \dot{g}$ $= x_{B} (F \Delta t_{A} - F \Delta t_{B}) \underline{k} + z_{B} (F \Delta t_{A} + F \Delta t_{B}) \dot{i}$ $= (0.6 \text{ m}) (F \Delta t_{A} - F \Delta t_{B}) \underline{k} + (1.44 \text{ M} \times 3 \text{ m}) (F \Delta t_{A} + F \Delta t_{B}) \dot{i} \qquad (2)$

IMPULSE-MONTENTUM PLINICIPLE

SINCE THE PINAL ANGULAR VELOCITY AND, THUS, THE FINAL ANGULAR MUNICIPILITY MUST BE ZERU, THE SUM OF (1) AND (2) MUST BE ZERO, EQUITING THE COEFF. OF L AND K TO ZERO: (1.44 \$53 m)(FDta+Fbt_)+96.04 kg. m/s = 0 (0.6 m)(FBta-Fbt_) + 156.06 kg. m/s - 0

OR $F\Delta f_A + F\Delta f_B = -66.302 \text{ N·s}$ (3) $F\Delta f_A - F\Delta f_B = -260.1 \text{ N·s}$ (4)

SOLVING (3) AND (4) SIMULTANEOUSLY:

FOL = -163.20 N.S FALB = 76.90 N.S
THE FACT THAT FOLKO INDICATES THAT THE DIASONALLY
OPPOSITE THRUSTER SHOULD BE USED INSTRAD OF A THW

(a) THRUSTERS BAND C

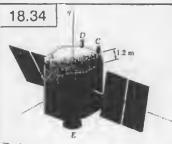
(b) $F\Delta t_B = 96.90 \cdot Ns$, $\Delta t_B = \frac{96.90 \cdot Ns}{20 \cdot N} = 4.84 s$ $F\Delta t_C = 163.20 \cdot Ns$, $\Delta t_C = \frac{163.20 \cdot Ns}{20 \cdot N} = 8.16 s$

(c) IF THE VELOCITY \$\vec{v}\$ OF THE MASS CENTER 10 TO BE UNCHANGED, THE RESULTANT OF THE LINEAR IMPUSES HUST BE ZERO.

 $-(F\Delta t_{B})_{\frac{1}{2}} - (F\Delta t_{C})_{\frac{1}{2}} + (500 \text{ N})\Delta t_{E}_{\frac{1}{2}} = 0$ $-96.90 \text{ N} \cdot \text{S} - 163.20 \text{ N} \cdot \text{S} + (500 \text{ N})\Delta t_{E} = 0$

 $\Delta t_E = \frac{260,18.5}{500N} = 0.52025$

 $\Delta t_E = 0.520 s$



GIVEN: PROBE WITH m = 2500 kg, K = 0.98 m, ky= 1.06m, kz= 1.02m; 500-N MAIN THRUSTER E; 20-N THRUSTERS A, B, C,D CAN EXPEL FUEL IN IL DIR. PROBE HAS ANG. VELOCITY W=(0.060 and/s) i -(0.040 rad/s k

PIND:

(a) WHICH TWO THRUSTERS SHOULD BE USED TO REDUCE CO TO IFE GIVEN:

(b) OPERATING TIME OF THESE THRUSTERS.

(c) HOW LUNG SHOULD E BE ACTIVATED IF IT IS TO HE UNCHANGED

INITIAL ANGULAR MOMENTUM

$$\begin{split} & \stackrel{H}{G} = \stackrel{T}{I} \omega_{2} \stackrel{i}{i} + \stackrel{T}{I}_{3} \omega_{3} \stackrel{i}{j} + \stackrel{T}{I}_{4} \omega_{2} \stackrel{k}{k} = \pi i \left(k_{1}^{2} \omega_{3}, i + k_{2}^{4} \omega_{3}, j + k_{2}^{2} \omega_{2} \stackrel{k}{k} \right) \\ & = \left(2500 \, \text{kg} \right) \left[(0.98 \, \text{n})^{4} (0.06 \, \text{trad/s}) i + 0 + \left(1.02 \, \text{m} \right)^{2} \left(-0.04 \, \text{rad/s} \right) \stackrel{k}{k} \right] \\ & = \left(144.06 \, \text{kg} \cdot \text{ni}^{2} / \text{s} \right) \stackrel{i}{i} - \left(104.04 \, \text{kg} \cdot \text{ni}^{2} / \text{s} \right) \stackrel{k}{k} \end{split}$$

ANGULIER IMPULSE OF TWO 20-N THRUSTER'S

SEE SOLUTION OF PROB. 18, 33. ASSUMING THAT THEISTERS A AND B ARE USED, WE FOUND

ANG. IMPULSE ABOUT G

= (0.6 m)(FOTA-FOTA)K+ (1.44853 m)(FOTA+FOTA)i (2)

IMPULSE-MOMENTUM PRINCIPLE

SINCE THE FINAL ANG. VELOCITY AND, THUS, THE FINAL ANG, MOMENTUM MUST BE ZERO, THE SUM OF (1) AND (2) MUST BE ZERO. ZOUATING THE COFFE OF L AND & TO ZERO, (1.44853 m)(FDtA+FDtB) + 144.06 kg·m3/5 = 0

(0,6m)(FAtA-FAtB)-104.04 kg. m3/5=0

FATA - FATA = 173.40 N.S (4)

SOLVING (3) AND (4) SIMPLITANEOUS . Y :

THE FACT THAT FATE TO INDICATES THAT THE THRUSTER D, WHICH IS DIAGONIALLY STITETTE TO 8 SHIVLD HE USED INSTEAD OF B. THUS!

(a) THRUSTERS A AND D

(6)
$$F\Delta t_{a} = 36.974 \text{ N.s.}, \quad \Delta t_{A} = \frac{36.974 \text{ N.s.}}{20 \text{ N}} = 1.8487s$$

$$F\Delta t_{D} = 136.43 \text{ N.s.}, \quad \Delta t_{D} = \frac{136.43 \text{ N.s.}}{20 \text{ N}} = 6.821s$$

$$\Delta t_{A} = 1.849s; \quad \Delta t_{D} = 6.82s$$

(c) IF THE VELUCITY TOF THE MASS LEWIER IS TO BE UNCHANGED, THE RESULTAINT OF THE LINEAR IMPULSES MUST BE ZERO.

$$-(f \Delta t_{A}) \underline{j} - (F \Delta t_{D}) \underline{j} + (500 N) \Delta t_{E} \underline{j} = 0$$

$$-36.974 N.S - 136.43 N.S + (500 N) \Delta t_{E} = 0$$

$$\Delta t_E = \frac{173.40 \text{ M.s}}{500 \text{ N}} = 0.34685$$
 $\Delta t_E = 0.3475$

18.35



PROBE WITH PRINCIPAL CENTRUIDAL AXES X, Y, E, AND W= 3000 16, k2= 1.375 ft, ky = 1.425 ft, k2 = 1.250 ft. PROBE HAS NO ANG. VELOCITY WHEN STRUCK AT A BY 5-02 METERRITE WITH VELOCITY RELATIVE TO PROBE 2 = (2400 ft/s)i - (3000 ft/s)j + (3200 ft/s)k

METEORITE EMERGES ON OTHER SIDE OF PANEL MOVING. IN SAME DIRECTION WITH SPEED REDUCED BY 20% FIND: FINAL ANGULAR VELOCITY OF PROBE.

ANBULAR HOHENTUM OF METEORITE ABOUT G.

$$(H_{G})_{M} = \frac{1}{2} \times m_{M} \underbrace{V}_{0}$$

$$= \left[(9.1t) \underbrace{i}_{1} + (0.75ft) \underbrace{k}_{1} \times \frac{(5/16) \text{ lb}}{32.2 \text{ fg}} \underbrace{(2400 \text{ fg}) \underbrace{i}_{2} - 3000 \underbrace{j}_{1} + 3200 \underbrace{k}_{2} \right]$$

$$= (9.705 \times 10^{3} \underbrace{b \cdot 5/1}_{1}) (-27 \underbrace{k}_{2} - 28.8 \underbrace{j}_{1} + 1.8 \underbrace{j}_{2} + 2.25 \underbrace{i}_{1} \times 10^{3} \underbrace{ft/s}_{2})$$

$$= (9.705 \underbrace{b}_{1} \cdot 5\underbrace{j}_{2} \cdot 27 \underbrace{j}_{2} - 27 \underbrace{k}_{2})$$

$$(H_{G})_{M} = (21.836 \underbrace{b}_{1} \cdot 5) (\underbrace{i}_{2} - 12 \underbrace{j}_{2} - 12 \underbrace{k}_{2})$$

$$(1)$$

FINAL ANGULAR MOMENTUM OF PROBE

$$(\underline{H}_{G})_{p} = \bar{I}_{\omega} \omega_{x} \underline{i} + \bar{I}_{y} \omega_{y} \underline{i} + \bar{I}_{z} \omega_{z} = m(k_{z}^{2} \omega_{x} \underline{i} + k_{y}^{2} \omega_{y} \underline{j} + k_{z}^{2} \omega_{z} \underline{k})$$

$$= \frac{3000 \text{ lb}}{32.2 \text{ ftb}_{z}} [(1.375 \text{ ft})^{2} \omega_{x} \underline{i} + (1.425 \text{ ft})^{2} \omega_{y} \underline{j} + (1.250 \text{ ft})^{2} \omega_{z} \underline{k}]$$

$$= (176.15 \text{ lb.ft.s}) \omega_{x} \underline{i} + (189.19 \text{ lb.ft.s}) \omega_{y} \underline{j} + (145.57 \text{ lb.ft.s})^{2} \omega_{z} \underline{k} \qquad (2)$$

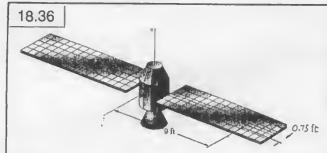
WE EXPRESS THAT (HG) = 020 (Hg)M RECALLING (1) AND (2):

176, 15
$$\omega_x$$
 \underline{i} + 189, 19 ω_y \underline{j} + 145, 57 ω_z \underline{k} = 0.20 (21.836) (\underline{i} - 12 \underline{j} - 12 \underline{k}) = 4.3672(\underline{i} - 12 \underline{j} - 12 - \underline{k})

EDUATING THE COEFF. (T THE UNIT VECTURS!

176.15
$$\omega_{x} = 4.367$$
 $\omega_{z} = 0.02479 \text{ rad/s}$
189.19 $\omega_{y} = -52.406$ $\omega_{y} = -0.2770 \text{ rad/s}$
145.57 $\omega_{z} = -52.406$ $\omega_{z} = -0.3600 \text{ rad/s}$

ω=(0.0248 rad/s)i-(0,277 rad/s)j-(0,360 rad/s)k



PRUBE WITH PRINCIPAL CENTROIDAL AXES Zy, E, AND W=3000 Ib, K=1.375 ft, Ky=1.425 ft, K, =1.250 ft.
PROLE HAS NO AHEOLIGE VELOCITY WHEN STREE AT A BY 5-02 METEOCITE WHICH EMERGES ON CTHER SIDE OF PRINCIPAL MOVING IN SHIPE DIRECTION WITH SPEED REDUCED BY 25% FINAL ANGULAR VELOCITY OF PROSE IS

ω=(0, C5 rod/s) v - (0.12 rod/s) j + ω h

AND X COMPONENT OF CHANGE IN V OF PROBE IS

Δ1/2 = - 0, 675 in./s.

FIND: (a) Wz .

(b) RELATIVE VELOCITY & OF METEULITE WITH WHICH
IT STRIKES PAYEL.

CONSERVATION OF LINEAR MOMENTUM IN X DIRECTION SINCE 25% OF LINEAR MOM, OF METEURITE 15 TRANS-FERRED TO PROF. E.

0.25 (5/16) 16 (Vo) = 3000 16 A V

 $(V_0)_2 = 38.4 \times 10^3 \Delta V_2 = 38.4 \times 10^3 (-0.675 in./s) = 25.92 \times 10^3 in./s$ $(V_0)_2 = -2160 \text{ ft/s}$

CONSERVATION OF ANEILINE MOMENTUM NEWS G-INITIAL ANGLIDM, OF METEURITE!

 $\frac{(H_{G})_{M}}{f} = \frac{1}{2} \times m_{M} U_{G} = \left[(94t) \frac{1}{2} + (0.75tt) \frac{1}{2} \right] \times \frac{(\frac{\pi}{2})(1/2)}{3} \left[(V_{0}) \frac{1}{2} + (V_{0})$

 $(H_{G})_{M} = \frac{(5/16) B}{32. \text{ ft/s}^{3}} \begin{vmatrix} \frac{1}{9} & \frac{1}{2} & \frac{1}{2} \\ -2160 \text{ ft/s} & (46) & (46) \end{vmatrix}$

 $(H_{q})_{H} = \frac{(5/16)/6}{32.2 \frac{17}{5}} \left[-0.75(V_{0}) \frac{1}{2} - (1620 + 9(V_{0})) \frac{1}{2} + 9(V_{0}) \frac{1}{2} \right]$

FINAL ANG, MOIL OF PROBE:

 $(H_G)_{p} = I_{\omega_2} \underbrace{i}_{\underline{i}} + I_{\underline{j}} \omega_{\underline{i}} \underbrace{i}_{\underline{j}} + I_{\underline{j}} \omega_{\underline{j}} \underbrace{k}_{\underline{j}} = m (k_{\underline{j}}^2 \omega_{\underline{j}} \underbrace{i}_{\underline{j}} + k_{\underline{j}}^2 \omega_{\underline{j}} \underbrace{k}_{\underline{j}} + k_{\underline{j}}^2 \omega_{\underline{j}} \underbrace{k}_{\underline{j}})$ $= \frac{3000 \ lb}{32.2 \ fl_{\underline{k}}} \underbrace{[(1.375 + 1)^2 (0.05 \ cod/\underline{j})] - (1.425 fl)^2 (0.12 \ cod/\underline{j})]}_{(1.250 ft)^2 \omega_{\underline{j}} \underbrace{k}_{\underline{j}}} + \underbrace{(2)}_{(2)}$

SINCE 25% OF AMEULIK MOM. OF HETEORITE IS TERMISFE. LED TO PROBE, (HG) = 0.25(H), OR, RECALLING (1) AND (2):

3000 $[(1.375)^2(0.05)_{\underline{i}} - (1.425)^2(0.12)_{\underline{j}} + (1.250)^2a)_{\underline{i}} = 0.25(5/16)[-0.73(50)_{\underline{i}} - (1620 + 9(5)_{\underline{i}})_{\underline{j}} + 9(50)_{\underline{k}}]$

FRUNTE THE COPFF OF UNIT VECTORS:

① $203.59 = -0.058394(V_0)_y$ (V_0)₂ = -4840 ft/s
② $-731.03 = -126.56 - 0.70313(V_0)_2$ (V_0)₂ = 859.7 ft/s
② $4687.5 \omega_2 = 0.70313(-4840)$ $\omega_2 = -0.726$ rod/s

ANSWERS:

(a) $\omega_z = -0.726 \text{ rad/s}$ (b) $\Psi_0 = -(2160 \text{ fHs})\underline{i} - (4840 \text{ fH/s})\underline{i} + (860 \text{ fH/s})\underline{k}$ 18,37 GIVENI

RIGID RUDY WITH FIXED POINT O, ANG. VELOCITY WARD ANGULAR MOMENTUM H_0 , AND KINETIC ENERGY T. 3HOW THAT: (a) $H \cdot \omega = 2T$,

(b) 0 < 90°, WHERE B IS ANGLE RETWEEN WAND HO

(a) USING PRINCIPAL AXES AS COORDINATE AXES, WE WRITE H.ω=(H, + H, + H, + H, +). (ω, +ω, +ω, κ)

 $= H_z \omega_n + H_y \omega_y + H_z \omega_z \tag{1}$

SINCE X 7 2 MOR PRINCIPAL AYES,

 $H_2 = I_2 \omega_1$ $H_2 = I_2 \omega_2$ $H_2 = I_2 \omega_3$

SUBSTITUTE INTO (1):

 $\underline{H}_{0} \cdot \underline{\omega} = \underline{I}_{2} \omega_{2}^{2} + \underline{I}_{3} \omega_{3}^{2} + \underline{I}_{2} \omega_{2}^{2} \tag{2}$

BUT, FROM FU. (18.20), $T = \frac{1}{2} \left(I_2 \omega_2^2 + I_3 \omega_3^2 + I_4 \omega_3^2 \right)$ WE CONCLUDE THAT

H. $\omega = 2T$ (U.E.D.)

(b) WE CAN EXPRESS THE SCALAR PRODUCT AS

H. $\omega = H_0 \omega \cos \theta$

THUS: $\cos \theta = \frac{H_0 \cdot \omega}{H_0 \omega} = \frac{2T}{H_0 \omega} > 0$, SINCE T>0

SINCE COS \$ >0, WE MUST HAVE B < 90 (Q.F.D)

18.38

GNEN:

RIGID BODY WITH FIXED PUNTO: $\omega = \text{INSTANTANEOUS}$ AND VELOCITY $I_{\text{CL}} = \text{MOMENT OF INTERTIA OF BODY}$ ABOUT LINE OF ACTION OLOF ω .

SHOW THAT T= 1/2 IOL W2
(a) USING EUS. (9,46) AND (18,19),

(b) CONSIDER ING T AS THE SUM OF THE K.E. OF PARTICLES P.

(a) EQ. (18.19):

 $T = \frac{1}{2} (I_{1} \omega_{1}^{2} + I_{2} \omega_{2}^{2} + I_{3} \omega_{2}^{2} + I_{4} \omega_{2}^{2} - 2I_{2} \omega_{2} \omega_{3} - 2I_{3} \omega_{3} \omega_{3} - 2I_{3} \omega_{2} \omega_{3})$

LET Wx = W cos 0x = W /2 !

Ly = W cos 0y = W /2 !

L2 = W cos 0y = W /2 !

SUBSTITUTE INTO EU.(18,14): T= \(\langle (I_2)^2 + \frac{1}{2} \langle \gamma_1 + I_2 \langle -2 \frac{1}{2} \langle \gamma_2 \rangle -2 \frac{1}{2} \langle \gamma_2 \rangle \

BUT, BY EN. (9.46) OF SEC. 9.16. EXPRESSION IN PAGENTHESES IS IOL. THUS:

 $T = \frac{1}{2} I_{OL} \omega' \qquad (Q.E.D.)$

(b) EACH PARTICLE P. DESCRIBES A CIRCLE OF PRODUCTOR

 $T = \frac{1}{L} \sum_{i} (\Delta m_i) v_i^2 = \frac{1}{L} \sum_{i} (\Delta m_i) e_i^2 \omega^2$ $= \frac{1}{L} (\sum_{i} e_i^2 \Delta m_i) \omega^2$

BUT E CLAM: = IOL

THEREFORE:

 $T = \frac{1}{2} I_{0L} \omega^2$

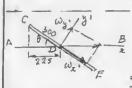
(Q, E.D.)

18.39 225 mm 225 mr

GIVEN: .. ASSEMBLY OF PRUB. 18.1. FUR EACH ROD:

> m=1.5kg LENGTH = GODMM ASSEMBLY RUTATES WITH W = 12 rod/s. KINETIC ENERGY

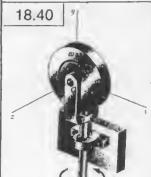
OF ASSEMBLY.



USING PRINCIPAL AXES 2'8' # : COSO = 125

 $\omega_{x} = \omega \cos 0$ $\omega_{x} = \omega \sin \theta$ $\omega_{z} = 0$ I,=0. I,= me, I,= 1/2 nie

 $EQ.(18.20): T = \frac{1}{2}(\bar{I}, \omega_{2}, +\bar{I}_{1}, \omega_{2}, +\bar{I}_{2}, \omega_{2})$ T= 1/2 (0+1/2 ml woint 0+0). = 1 (1.5 kg) (0.6 m) (12 rod/s) sin 41.41° T= 1.417 J



GNEN:

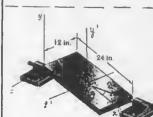
DISK OF PROB. 18,2 OF MASS AN AND RADIUS & ROTATING HS SHOWN

KINETIC ENERGY OF DISK.

EQ. (18,20): $T = \frac{1}{2} \left(\overline{I}_{1} \omega_{1}^{2} + I_{1} \omega_{2}^{2} + J_{2} \omega_{2}^{2} \right)$ =ナ(ロナ片州をいきナナ州をひう) T= = mE (W, + 2W,)

GIVEN: 18-16 RECTANGULAR PLATE M 18.41 PROB. 18,3 ROTHTING WITH W= 5 rod/s ABOUT X AXIS.

FIND: KINETIC ENERGY OF PLATE



WE USE PRINC, CENTROIDHL AKES GZYZ WITH

tand = 12 in = 0.5 0=26,565

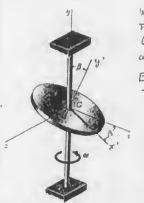
 \overline{I}_{2} ,= $\frac{1}{12}\frac{1816}{8}(111)^2 = \frac{1.5}{3}$ \bar{I}_{E} , = $\frac{1}{12} \frac{181b}{9} (211)^2 = \frac{6}{9}$

EQ. (18,20): $T = \frac{1}{2}(\bar{1}, \omega_x^2, +\bar{1}_y, \omega_y^2, +\bar{1}_z, \omega_z^2)$ T= \frac{1.5}{3} (5 rad/s) cos 26.5650 + 0 + \frac{6}{3} (5 rad/s) sin 26.5650] $=\frac{1}{9}\frac{1.5}{32.2}(5)^{2}(\cos^{2}24.565^{\circ}+4\sin^{2}26.565^{\circ})$ = (0.58230 ft.16)(0.8 + 4 x 0.2) = 0.9317 ft.16

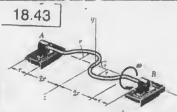
T=0,932 ft. 16

18.42

GIVEN: DISK OF PROB. 18.4. WITH 12=25. FIND: KINETIC ENERSY OF DISK.



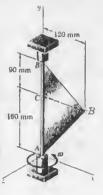
WE RESOLVE W= WE ALIMS THE PRINCIPAL CENTROIDA: AXES w,=-wsing, w,=wcosp, w=0 EU (18:10): $T = \frac{1}{2} \left(\bar{I}_{2}, \omega_{2}, + \bar{I}_{3}, \omega_{3}, + \bar{I}_{3}, 0 \right)$ =1(よいに必らいち+まれじかなど) = 1 n: 6"11 (sin / + 2 cos /) = = 112 02 (1+05/5) == 112" W (1+cos 25") T=0.228m2w



GIVEN: SHAFT OF FREE, 18, 15 OF MASS MI ROTATING WITH ANG. VEL. W. FIND: KINETIC ENERGY OF

MASS PEK UNIT LENGTH = m' = TH + 2TH SINCE WE = N = 0, FQ (18, 19) REDUCES TO T= + IW. BUT I, OF BOTH SEPTICIALLIENT POTTIONS OF KUD IS SHART AS SP FULL CIKELLINK KID. THAT IS, I, = 1/2 1/2 11/2 = 972 m1 = 11 8 11 = 11 = 11 T=0.1896712W

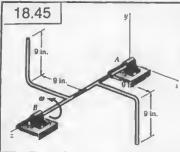
GIVEN: TRIAIKILLING PLATE OF PROB. 18,16 18.44 OF MASS M= 7.5 Kg WITH ANG, VEL, W= 12 rad/s FIND: KINETIC ENERGY OF PLATE



SINCE Wx = W, = 0, EQ, (18,19) REDUCES (1) AND Ig, MAIS = m/(1/2 6 h3) = 1/6 m h2 WHERE M = 7.5 kg, h = CB = (0.12 m) THUS Iy, MASS = 1 (7.5 kg) (0.12m) = 18.00 × 10 kg · m2

SUBSTITUTING THIS VALUE FOR I AND 12 rad/s POR W INTO (1), WE T== (18.00×103 kg. n12)(12 rad/s)

T= 1,296J



ASSEMBLY OF PROB. 18.19 WHICH ROTATES AT 360 Ppm. EACH L-SHAPED AKM WEIGHS 5 16. FIND!

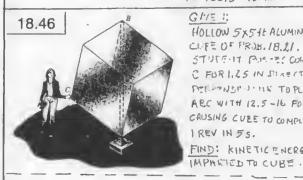
KINETICENERLY OF ASSEMBLY.

SINCE WZ = WJ = 0, EQ. (18,19) REDUCES TO T= 1 1, 102 FOR ONE ARM (OF MASS MI): ma + m (a2+ a1)+ 1 m a2 = 5 ma

FOR BOTH ARMS: I = 5 mat = 5 5 b 32.2 ft/s2 (3 ft). = 0.14557 1b.ft.5

AND W= 360 rev = 360 29 rad = 1297 rad/s

THUS: T= 1/2 1 02 = 1/2 (0,14557 16.41.5) (1291 rad/s) T= 103,5 ft. 16



GIVE 1:

HOLLOW 5x51t ALUMINUM CLIFE OF FRUE. 18.21. STUTE-IT PUR-E! CONTEN C FOR 1.25 IN DIXECTION PERSONSI JOHN TO PLANE ARC WITH 12.5 - 16 FOACE CAUSING CURE TO COMPLETE I REV IN 55. FIND: KINETIC ENERGY

DIRECT COFIPUTATION OF K.E. WE HAVE W=(21Trad)/55 = 1.2566 rod/s WE RECALL FROM PROB. 18.21 THAT AB IS A PRINCIPAL THUS, EQ. (18.19) YIELDS $T = \frac{1}{2} I_{AB} \omega^2 = \frac{1}{2} I_{B} ma^2 \omega^2 = \frac{5}{36} m (5 ft)^2 (1.2566 rad/s)^2$

BUT WE FOUND IN PROB. 18.21 THAT W= 126 16 T= 5 22616 32,2 ft/62 (541) (1,2566 rad/s) = 38,48 ft. 16 T= 38,5 ft.16

ATTERNIPTIVE SOL TIDIN

WE NOTE THAT THE K.E. IMPARTED TO THE CUBE IS EQUAL TO THE WORK U - DONE BY THE STICENT!

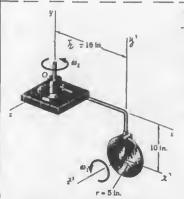
T=U102 = FAS

WHERE F= 12,5 IS AND DS = 1 UDt = 1 WEDE RECALLING THAT THE RADIUS & OF THE CIRCLE DESCRIBED EY C IS (SEE HINT IN PRUE. 18.21)

E= 1 1/2/3 = (5 11)1/2/3 = 4.0825 11 WE HAVE DS= = (1.2566md/s)(4.0825+1)(1.25) = 3.078 +1 AND T= (12.51b)(3,078 ft)= 38,48 ft.16, T=38.5 ft.16

GIVEN: 18.47

DISK OF PROB. IB.S WITH WEIGHT W = 8 lb. AND ANGULAR VELD CITIES WI= 12 rad/s AND WZ= 4 rolls. FIND: KINETIC ENERGY OF DISK.

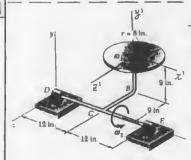


EQ. (18,17):

T= = mo+ = (I, of + I of + I of + = 10+ = m2 2 + = 11 60 +=(=ff)(4 mdk)++(=)(12) = (0,12422) 28.444 + + 0.6944 + 12.5] = 5.1724 ft.16 T= 5.17 A.Lb

18.48

DISK OF PROB. 18.6 WITH WEIGHT W= 6 16 AND ANGULAK VELOCITIES W; = 16 rad/s AND W,= 8 rad/s. FIND: KINETIC ENERGY OF DISK.

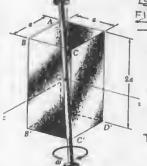


EW (18.17): T= = mD+ + (] a + 1 0 +] or WHERE U= W, (AC)2 WITH (AC) = (AB) + (BC) (AC)=2(急化)=1.125 12=中加を三十四(日出) 1 = 1 m 2 = 0.2222 m

THUS: T = 1/2 (1107+13, W2+13, W2+0) = $\frac{1}{2}$ m [1.125 ω_1^2 + 0.1111 ω_2^2 + 0.2222 ω_1^2] 32.2 ft/c [1.236 (8 rad/s) + 0.2212 (16 rats)] T= 12.67 Fib

18.49 and 18.50

GIVEN: PARALLELEPIPED OF 18.49: PROB. 18.7 (SOLID) 18,50: PROB. 18.8 (HOLLOW) FIND: KINETIC ENERGY



SINCE GIS FIXED AND I.Y. E ARE PRINCIPAL AXES, USE (18.20): T= = (I, N, + I, W; + I, W;)

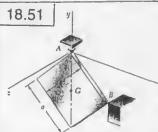
 $\frac{16.49}{16.49}$ WE HAVE $I = I = \frac{1}{12} m \left[a^2 + (2a)^2 \right] = \frac{5}{12} m a^2$, $I = \frac{1}{6} m a^2$ SUBSTITUTE IN (1): $T = \frac{1}{12} ma^2 \left(\frac{5}{12} + \frac{4}{6} + \frac{5}{12} \right) \omega^2 = \frac{1}{12} ma^2 \left(\frac{3}{2} \omega^2 \right)$ (CONTINUED)

18.49 and 18.50 continued

WE RECH . _ FROM THE PREVIOUS PAGE T=1-([,+4[,+1]) W2 (1)

18.50: SEE SOLUTION OF PROB. 18.8 FOR THE DETER-MINATION OF THE PRINCIPAL MUMERITE OF INERTIAS $I_{3} = \frac{37}{40} ma^{2}$ $I_{3} = \frac{9}{30} ma^{2}$ Iz = 37 ma SUBSTITUTE IN EQ. (1)

 $T = \frac{1}{12} ma^2 \left(\frac{37}{60} + \frac{4\times9}{30} + \frac{37}{60} \right) \omega^2 = \frac{146}{720} ma^2 \omega^2$ T=0,203maw GIVEN: 18.51

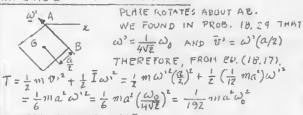


SQUARE PLATE OF PROB. 18.29 OF MASS M WITH MILLIEL Was STRIKES BWITH e = Q FINDI KINETIC ENERGY LUST IN MYF HCT.

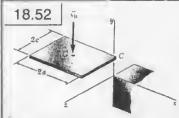
WE RELALL FROM PRUB. 18.29 THAT I = 1/2 THAT ABLI'T ANY ANY THROUGH GIN THE PLANE OF THE PLATE. KINETIC ENERGY BEFORE IMPACT

To = 1/2 I Wo = 1/2 (1/2 nia) wo = 1/2 mai wo

KINETIC ENERGY AFTER IMPACT



EINETIC ENERGY LOST $= \frac{1}{24} \operatorname{ma^2 \omega_0^2} - \frac{1}{192} \operatorname{ma^2 \omega_0^2} = \frac{7}{192} \operatorname{ma^2 \omega_0^2}$



GIVEN: RECTUNGULAR MLATE OF PRUBS. 18.31 AND 18.32 VELOCITY TO AND W=0 HITS OBSTRUCTION (E=0) FIND: KINETIC ENERGY LOST IN IMPACT.

BEFORE IMPACT

T = 1 m V

AFTER IMPACT

FROM PRUB. (18.31): $\omega_{\chi} = 3\bar{v}_{0}/7c$, $\omega_{y} = 0$, $\omega_{z} = 3\bar{v}_{0}/7a$ FROM PROB. (18.32): $\bar{\psi} = -(6\bar{v}_{0}/7)\dot{g}$

EQ.(18. 17): $T = \frac{1}{2} m \bar{v}^2 + \frac{1}{2} (\bar{L}, \omega^2 + \bar{L}, \omega^2 + \bar{L}, \omega^2)$ $= \frac{1}{2} m \left(\frac{6}{7} \vec{V_0} \right)^2 + \frac{1}{2} \left[\frac{1}{3} m c^2 \left(\frac{3 \vec{V_0}}{7c} \right)^2 + 0 + \frac{1}{3} m a^2 \left(\frac{3 \vec{V_0}}{7a} \right)^2 \right]$ $= \frac{1}{2} m \vec{\nabla}_{\nu}^{2} \left(\frac{1}{7} \right)^{L} \left[16 + 3 + 3 \right] = \frac{1}{2} m \vec{\nabla}_{\nu}^{2} \frac{42}{49} = \frac{1}{2} \frac{6}{7} n \vec{\nabla}_{\nu}^{2}$

KINETIC ENERGY LIST $T_0 - T = \frac{1}{2} m \vec{v}_0^2 \left(1 - \frac{6}{7} \right)$ $T_0 - T = \frac{1}{14} m \vec{v}_0$

GIVEN : 18.53

SPACE PROBE OF PROG. 18.35, WITH W= 3000 lb, K= 1,375 ft, K= 1.415 ft, K= 1.250 ft.

KINETIL ENERGY OF PROBE IN IT: METION APOUT ITS MASS CENTER AFTER ITS COLLISION WITH METEURITE

SEE SOLUTION OF PROB. 18.35 FOR DETERMINATION OF ω, = 0.0248 rad/s, ω, = -0.277 rad/s, ω, = -0.360 rad/s IN MOTION AROUT G, G IS A FIXED POINT AND THE x, 4. 2 AXES ARE PRINCIPAL AXES, WE USE EU (18,20); $T' = \frac{1}{2} (I_{\mu} \omega_{\lambda}^{2} + I_{\mu} \omega_{\beta}^{2} + I_{\mu} \omega_{\lambda}^{2}) = \frac{1}{2} m (k_{\lambda}^{2} \omega_{\lambda}^{2} + k_{\mu}^{2} \omega_{\beta}^{4} + k_{\lambda}^{2} \omega_{\lambda}^{2})$

 $= \frac{1}{E} \frac{3000 lb}{32.2 tt/s} \left[(1.375 ft \times 0.0248 rad/s)^{2} + (1.425 ft \times 0.277 rad/s)$ + (1.250 ft x 0.360 rad/s)]

 $=\frac{1}{2}\frac{3000 \text{ lb}}{52.2 \text{ ft/s}} (0.3595 \text{ ft}^3/\text{s}^4) = 16.747 \text{ ft} \cdot \text{lb}$ T' = 16.7T'= 16.75 ft.16

GIVEN: 18.54

SPACE PROBE OF PROB. 18.36, WITH W= 3000 16, K,= 1.375 ft, K4= 1.425 ft, K2= 1.250 ft. FIND:

KINETIC ENERGY OF PROBE IN ITS MOTION ABOUT ITS MASS CENTER AFTER ITS COLLISION WITH METEORITE.

SEE STATEMENT AND SOLUTION OF PROB. IP. 36 FOR THE VALUES OF W, W, W, AFTER COLLISION :

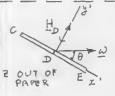
 $\omega_z = 0.05 \text{ rad/s}, \ \omega_y = -0.12 \text{ rad/s}, \ \omega_z = -0.726 \text{ rad/s}$ IN MOTION ABOUT G, G IS A FIXED POINT AND THE 2, 3, 2 AXES AFE PRINCIPAL NYES. WE USE EU. (18,20):

 $T' = \frac{1}{2} \left(I_{\lambda} \mathcal{Q}_{\lambda}^{2} + I_{\lambda} \mathcal{Q}_{\lambda}^{2} + I_{\lambda} \mathcal{Q}_{\lambda}^{2} \right) = \frac{1}{2} m \left(k_{\lambda} \mathcal{Q}_{\lambda}^{2} + k_{\lambda} \mathcal{Q}_{\lambda}^{2} + k_{\lambda}^{2} \mathcal{Q}_{\lambda}^{2} + k_{\lambda}^{2} \mathcal{Q}_{\lambda}^{2} \right)$ = $\frac{1}{2} \frac{3000 \text{ lb}}{32.2 \text{ kb}} \left[(1.375 \text{ ft} \times 0.05 \text{ rod/s})^2 + (1.425 \text{ ft} \times 0.12 \text{ rad/s})^2 + (1.425 \text{ ft} \times 0.12 \text{$ + (6250 ft x 0,726 rad/s)]

= 1 3000/b (0,8575 ft/ss) = 39,946 ft/b T'=39.9 H. 1b

GIVEN: ASSEMBLY OF PRUB 18.1 18.55 FOR EACH ROD: M= 1,5kg, &= 600 mm ASSEMBLY ROTATES WITH W= 12 rad/s.

FIND: RATE OF CHANGE H OF ANG. MOMENTUM H



FROM PROB. 18.1: 0 = 41.41° USING PRINCIPAL HXES X'Y'E: W = W (cos & l'+ sin & j') Hp = 1 mewsind ; EQ. (18.27') YIELDS HD = (HD) DXHD + TXHD

BUT (HD) DZ'J'E = O AND Q = W.

HD = W x HD = W (cos Oi'+ sin oj') x 1/2 ml'2 w sin oj' = $\frac{1}{12}$ mlw sinduso $\frac{1}{2}$ = $\frac{1}{24}$ mlw sin $\frac{20}{2}$

WITH GIVEN DATA. 11 = \frac{1}{24} (1.5 kg) (0.6 m) (12 rad/s) sin 82.82° K

H = (3.21 N·m) K

GIVEN: DISK OF PROB. 18.2. 18.56 18.60 GIVEN: DISK OF PROB. 18.6 WEIGHING GIL FIND: RATE OF CHANGE HE OF HE. WITH W,= 16 rod/s AND W,= Brad/6. FIND: RATE OF CHANGE H. OF H. FROM PROB. 18.2: W = W2 1 + W, K USING PRINCIPAL CENTROIDAL A FE AX'; 'e' . HG = 4 m2 (W2j + 201k) W= Wz i + Wij D= Dzi WE NOTE THAT THE ANGULAR VELUCITY OF THE Ha=1, Ni +1, 2, 2+1, w. == 1 me or + + modal PRAME GXY2 15 12 = 223 EQ. (18.22): HA = (HA)Axiy's. + 12 XHA = 0 + W2i X HA E &. (18.22): $\underline{H}_{G} = (\underline{H}_{G})_{GXYZ} + \underline{\Omega} \times \underline{H}_{G} = 0 + \underline{\Omega} \times \underline{H}_{G}$ = Wz ix (1 m 2 w i + 1 m 2 w)) = 7 m 2 w w x t THUS: H = Wzix Lm2 (Wi+2W, K) WITH GIVEN DATAL $\frac{H}{h} = \frac{1}{2} \frac{616}{32.2116} (\frac{\theta}{12} + t)^2 (16 \text{ rad/s}) (8.100 d/s) k$ 616 H = 1 mrw, w, i H = (5,30 16. ft) K GIVEN: PLATE OF PROB. 18,3 WEIGHING 18.57 18/b, WHICH ROTATES WITH W=5 rad/6. GIVEN: ASSEMBLY OF PRUB. 18.1. FIND: RATE OF CHANGE H. OF H. FOR EACH RUD: m=1.5 kg, l= 600 inin AT INSTAILT CONSIDENCED, W=(12 rad/s)i, Q=(96 rad/s1)i. WE HAVE w = (5 rad/5) i FIND: RATE OF CHANGE SEE SOLUTION OF PRUB. 18.3 FOR THE DERIVATION FROM PROB. 18.1: 0=41,41° $H_G = (0.3727 \text{ /b. ft. s}) i - (0.2795 \text{ /b. ft. s}) \underline{k}$ USIN: PRINCIPAL AXES X, X, Z: w=wi w=w(cosoi'+sinbj') EQ. (18.22): a = a (cosoi'+sinoj') H=(HG)x42+ UxHG=0+WxHA 2 OUT 08 Ho = 12 me wsind ; THUS: H = (5 rad/s) ix[0,3727 |b.ft.s) i - (0,2795/b.fl.s)k] H) Daig = 12 mliw sindj' = 12 mliasindj' H = (1.398 16.1t)j RPPLY EQ. (18,22), OBSTRVING THAT Q = W : GIVEN: DISK AND SHAFT OF PROB. IR 4. 18.58 $\underline{H}_{D} = (\underline{H}_{D})_{Dx'x'x} + \underline{\Omega} \times \underline{H}_{D} = (\underline{H}_{D})_{Dx'x'x} + \omega \times \underline{H}_{P}$ = 1 me'a sinoj' + w (cosoi'+sinoj') x 1 me'w sinoj' USING THE PRINCIPAL AXES GX'Y'Z. = 1 mle a sind j' + 1 mlew cost sind k WE FOUND IN PROB. 18.4 THAT w = w (-sinpi'+cospj') BUT j'= sindi + cosoj HG = 4 me w (-sins i'+205/j') H = 1 ml a sind (sinti + wsoj)+ En. (18.22): He = (He) Gx, A, + To x He = 0 + M x He ml'w'cososinok H₆ = ω(-sin β i'+ cosβ j') × + m n² ω(-sinβ i'+2ωςβ j') H = 17 ml sind (x sind it quest j + w wst k) = + mrw (-2sin Bcospk+cospsinph) = - 1 mt wsin2 pk = - 1 mt wsin50 WITH GNEN DATH! m= 1,5 kg, l= 0.6m, W= 12 rad/s, X= 96 rad/s. 0= 41.41. H=-0,0958m&iok H_ = 1/2 (1,5 kg) (0.6 m) sin 41,410 [(96 rad/s) sin 41,410 i+ GIVEN: DISK OF PAOB. 18.5 WEISHING 8 16. (96 rad/s) cos 41.41 2 + (12 rad/s) cus 41.41 k WITH W1 = 12 rad/s AND W2 = 4 rad/s. H = (1.890 Nom) + (2.14 Nom) j + (3.21 Nom) k FIND: RATE OF CHANGE H, OF HA USING PRINCIPAL CENTRADAL AXES AXY'E' GIVEN: ASSECTELY OF PROB. 18. 1. W= W, 1 + W, K IL=w,j FOR EACH ROD: MI = 1.5 kg, l = 600 mm. 日、ころ、いちこれで、からます」ではことがは、 AT INSTANT (INCIDENZO, W=(12rad/s), Q=-(96rad/c))i. FIND: RATE OF CHANGE HOF HO. EO. (18.22): HA = (HA) A2/4/2+ + QX HA = 0 + WZ X XHA SUBSTITUTE GIVEN DATA INTO EQ. (1) OF PROB. 18. 61. $\frac{H}{A} = \omega_2 j \times (\frac{1}{4} m \epsilon^i \omega_2 j + \frac{1}{2} m \epsilon^i \omega_1 k) = \frac{1}{2} m \epsilon^i \omega_1 \omega_2 i$ H= 12 m (sin B (a sin b i + a cos o j + w cos o k) H = 1/2 (1.5 kg) (0,6 m) sin 41.41° [- (96 rad/s) sin 41.41°+ VITH GNEV DATA: H= 1/2 816 (5 ft) (12 rad/s) (4 rad/s) = (-46 rolls) cos +1,41° j + (12 mid/s) cos 41.41° k) H = (1.035 16.1t) i H = - (1.890 N·m) = - (2.14 N·m) j + (3.21 N·m) K

18.63

GIVEN: AT INSTANT CONSIDERED, 18-16 PLATE OF PROB. 18.3 HAS W = (5 rad/s) L AND & = - (20 rad/s) i.

FIND: RATE OF CHANGE H OF H

SEE SOLUTION OF PRUE. 18. 1 FOR THE DERIVATION OF EQ. (1): H= [(0.074534 10. fl.5)i-(0.055901 16. H 5)] (1)

SINCE & = OF WE HAVE

(H) = (0.074534 i - 0.05591k) a

SINCE D=W, EU.(18.22) YIELDS

= (0,074534i-0.0559015)&

+Wix (0,074534 i-0.055701 k)W

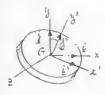
= 0.074534 Qi -0.055901 QK+0.055901 Qt LETTING & = - 20 rad/s. AND W= 5 rad/s,

H = 0.074534(-20)i + 0.055901(5)i - 0.055901(-20)k

H= - (1.471 16.41) i + (1.39816.41) j + (1.118 16.51) x

GIVEN: AT MISTHAIT CONSTITUTE, SHAT 18.64 OF PL J. 18.4 HAS ANGO OF TELXITY WEW & AND AMERICA ACE ENGLISH & = X ; FIND: RATE OF CHANGE H OF IT

SEE SOLUTION OF PROB. 18,4 FOR THE DETERMINATION OF HE USING THE PRINCIPHLOEITERINAL AND



GR'y'Z, HE OBTAINTE ZU.(1):

H= = 1 mew (-sinfil' + 2 cospj) TO REVERT TI TIE OF ISIMHL

AXES GXYZ, WE OKITHUE

i' = i cosp - j sinp j = i sinp + j cosp

SUBSTITUTING INTO (1):

He = + milw [-sinp (icos p-ising) +

2005 p(isin p + j we!)

= 4 112 (1) [5 11,3 cosp. i + (1+ cos 3) j.]

SINCE W=X

(HG)G212 + m22x [sings on pi+(HCO: P))

WE USE EQ. (18.27) WITH Q = W = W = .

) Gray + Qx H = + milk [sing con; + (1-cons)] + wjx + miw (sin Bus; i + (1+ws)) }

H= + m 1 a [simp co. 3 i + (1+003/3) i] - + 111 is inpace 1

LETTING B=2501 H = 1/4 1/2 x (0.38302 + 1.8214 =) - 1/2 + 1/3 (0.58307) k

 $H = mz^2(0.0958 \times i + 0.455 \times j - 0.0958 \times k)$

18.65 0. = 900 min

GIVEN:

ASSEMBLY CONSISTING OF TWO TRIANGULAR PLATES. EPCH OF MASS m= 5kg. WELDED TO VERTICAL SHAFT ASSEMBLY RUTATES WITH CONSTANT W= 8 rad/s.

DYNAMIC REACTIONS AT A

SINCE W= WS, FQS. (18.7) YIELD H=-I, w, H, = I, w, H=-I, w (1)

MOHENTS AND PRODUCTS OF INERTIA:

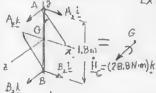
 $I_g = 2\left(\frac{m}{\Delta}I_g,_{AREA}\right) = 2\frac{m}{\frac{1}{3}ab}\left(\frac{1}{12}ab^3\right) = \frac{1}{3}mb^2$ [cf. front cover] Izy = 2 (m Izy, AREA) = 2 m (140 (140 b) = 1 mab (cf. Sample)

PROM EU. (1): H= - = - = mabwi + 1 mbaj

ER. (18.27): H= (Ho) = 1 + 1 × Ho= 0+ wx Ho= wj x mw(- 6 abi+ 3 bij)

 $\frac{H}{c} = \frac{1}{6} \text{ mabolik} = \frac{1}{6} (5 \text{ kg})(0.9 \text{ m})(0.6 \text{ m})(8 \text{ rad/s}) = (28.8 \text{ N·m}) \text{ k}$

EQUATIONS OF HOTION: WE EQUATE THE SYLTEMS (F EXTERNAL AND EFFECTIVE PLACES



ZMB = 2 (MB) ett: (1.8 m) j x (Azi+Azk) = (28.8 N.m)k -18A, K+1.8A, L = 28.8 K

Az=-16N, Az=0 A = -(16.00 N) i

B = (16,00 N)i

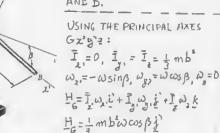
ZF=ZFif: A+B=D 18.66

GIVEN:

ROD AB OF MASS TO 15 WELDED TO SHAFT CD, OF LENGTH 25 WHICH ROTATES AT CONSTAUT RATE W.

FIND:

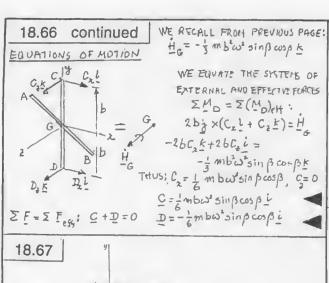
DYNAMIC REACTIONS AT C



OK, SINCT = isinp+icosp: H== mbwcosp(sinpi+cosp) (1) EQ.(18:2.):

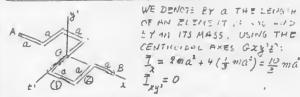
H=(H) + 5×H= =0+ 0×H = wj x 1 mba cocp (singli + wifij) = - = mbw sin, 1 cosps k

(CONTINUED)



GIVEN: 16-16 SHAFT WITH UNIFORM CROSS OF PROTATES AT CONSTANT SATE OF 12 mod/s.
FIND: DYNAMIC REALTHIN AT - MITE

MONEITS AND PRIOUS : DE 1. TOTIA



 $I_{z_{2}} = 2m\bar{z}_{1}^{2} + 2m\bar{z}_{2}^{2} + 2m\bar{z}_{2}^{2} = 2m(\frac{4}{2})a + 2ma(\frac{4}{2}) = 2ma^{2}$ $\frac{H}{G} = \bar{I}_{2}\omega i - \bar{I}_{2}\omega j - I_{2}\omega k = \frac{10}{3}ma^{2}\omega i - 2ma\omega k \qquad (1)$

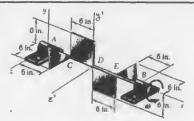
FQ.(18,22): Hg=(Hg)Gzy1,+QxHg=0+WixHg

Ho = wix (10mawi-2mawk) = 2mawi

AND AND TION

 $\Sigma F = \Sigma F_{eff}: \underline{A} + \underline{B} = 0 \quad \underline{A} = -\underline{B} = \frac{1}{2} m a \omega^2 k$ $\underline{DATA:} \ m = \frac{1}{8} \frac{W}{3} = \frac{1}{8} \frac{161b}{32,2 \text{ ft/s}} = 0.062112 \text{ lb.5} \text{ ft}$ $\alpha = 9 \text{ in.} = 0.75 \text{ ft} \qquad \omega = 12 \text{ radk}$

THUS: $A = \frac{1}{2} (0.062112/6.5^{3}/ft)(0.75ft)(12 rad/s)^{2} = 3.354 lb$ A = (3.35/b)k; B = -(3.351b)k 18.68



GIVEN:
ASSEMBLY WEIGHS 2.7 Ib AND RUTHTES AT CONSTANT
RATE $\omega = 240$ rpm
FIND: DYNAMIC REACTIONS PT. A AND B

COMPUTITION OF MUMERITY PROPUCTS OF INERTIA

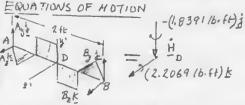
WE US = THE CENTROIDAL AXES DX3'Z'.

FOR EACH SQUESE: $MI = \frac{1}{3} \frac{2.71b}{3} = \frac{0.9}{9}$ $I_z = \frac{1}{3} ma' = \frac{1}{3} \frac{0.9}{9} \left(\frac{1}{2} \cdot f^2\right)^2 = 0.075/2$ $I_{zy} = m(\frac{a}{2})(-\frac{a}{2}) = -\frac{1}{4} \frac{0.9}{9} \left(\frac{1}{2} \cdot f^2\right)^2 = -0.05625/9$, $I_z = 0$ FOR EACH TKIANCE: $m = \frac{1}{6} \frac{2.71b}{3} = \frac{0.45}{9}$ $I_{zy} = \frac{1}{2} \frac{1}{$

 $I_{\chi} = (2 \times 0.075 + 2 \times 0.01875)/g = 0.1875/g$ $I_{\chi y} = 2(-0.05625)/g = -0.1125/g$ $I_{\chi z} = 2(-0.046875)/g = -0.09375/g$ $\frac{ANGULAL (1011E17:11M H_{\pm})}{H} = I_{\chi u} = I_{\chi y} \omega_{k}$ $H_{D} = (0.1875i + 0.1125j + 0.09375k)(\omega_{g}) \qquad (1)$

EO.(18.22): $\frac{\dot{H}}{D} = (\frac{\dot{H}}{D})_{DXY_{2}^{2}} + \Pi \times H_{D} = 0 + \omega L \times H_{G}$ SINCE $\omega = 240 \, \text{rpm} = 8\pi \, \text{rad/s, AND i x i = k, } L \times k = -i$

 $\frac{H}{H} = (0.1129 K - 0.09375 i)(8\pi)^{2}/g$ $\frac{H}{H} = (1.8391 (b.ft) i + (2.2069 (b.ft) k)$

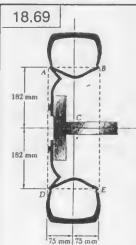


 $\sum_{n}^{M} = \sum_{k}^{M} (M_{n}) e_{k} : (2ft) i \times (B_{0} j + B_{0} t) = -1.8391 j + 2.2069 k$

THUS: $B_y = \frac{1}{2}(2.2069) = 1.1034 \, lb$ $B_z = \frac{1}{2}(1.8391) = 0.9196 \, lb$ $B = (1.103 \, lb) + (0.920 \, lb) k$

ZF = ZFes: A+8=0

A = -B $A = -(1.10316)\frac{1}{6} - (0.92016)\frac{1}{6}$

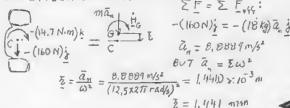


18- KY WHEEL IS ATTACHED TO BALANCINE MACHINE. WHEN MACHINE STINS AT THE RATE OF 12,5 rev/s, WHEEL IS FOUND TO EXERT ON MACHINE A FORCE-COUPLE - 11 -11 CONSISTING OF

F = (160N) & APPLIED ATC $M_{c} = (14.7 \, \text{N.m}) \, \text{K}$

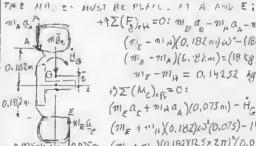
(a) DISTANCE & FROM 2 AXISTOG. AND IZY AND IZZ , (b) THE TWO CORRECTIVE MASS:S REQUIRED TO BALANCETHE WHEEL AND AT WHICH OF POINTS 4, B, C, D THEY SHOULD BE PLACED.

(a) THE FORCES EXERTED ON THE WHEEL MUST BE EQUIVALENT TO THE EFFECTIVE PORCES .



 $\Sigma M_c = \Sigma (M_G)_{eff} : -(14.7 \text{ N·m}) \underline{k} = \underline{H}_c$ BUT HG= I, Wi-I, Wi- I, WE AND Ho = WXH = WIX (INI - I, Wj - I WK) Ta = - I, w' k + I, w' f SUBSTITUTE IN (11: - 14.7K = - I,y w * K + I w 2 THUS: 14.7 N.M Iny = (14,7 N.M) = 2,3831 × 103 kg.m AND I = 0

 $I_{2,y} = 2.38 \text{ g·m}^2$, $I_{2,y} = 0$ (b) WITH COPRECTIVE MASSES THE FORCES EXERTED OIL THE WHEEL HRE EDUNVALENT TO SERD. FOR THE EFFECTIVE PIRCES TO ALSO PE EVENT TIZERO.

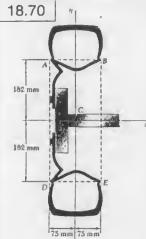


 $+12(F_0)_{\mu}=0: m_F a_F - m_{\mu} a_{\mu} - m_{\bar{a}} = 0$ (TIZ- 1) A)(0. 182m) W- (18/4) EW= (11/2 -711/A)(C. 82m)=(18 tg)(1441 x10ir) 711 = -11/4 = 0, 14252 kg

(ME a + MA a) (0,075 m) - HG = 0 (11) + +11/1 (0.182) (0.075) - 14,7 =0 $(m_E + m_H)(0.182\chi/2.5 \times 2\pi)^4(0.075) = 14.7$ m=+m= 0.17458 kg

SOLVING (2) AND (3) SIMULTANEOUSLY: MA = 16.034 x10 kg, m = 158.55 x10 kg

AT A AND E; MA=16.03g, ME=158.6g

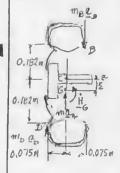


GIVEN: 18-Kg WHEEL IS ATTACHED

SPING AT THE RATE OF ISREVIS MECHANIC FINDS THAT A 170-9 MASS PLACED AT B AND A 56-5 MASS PLACED AT DARE NEEDED TO BALANCE WHEEL. BEFORE THE CURNET THE MASSES HAVE EFEN ATTACHED: (1) IMPANCE & FROM X -YIS TO GING IN ANT IEL. (b) रावह का स्टूडन्व प्राट ने प्राचित イナン まないいい ト ナマ ーララ デザスト INTENTED ENTHE WHEEL IN THE MIPCHINE.

TO PALANCING MACHINE AND

(a) APTER THE CURRECTIVE MASSES HAVE EBEN ADDED, THE SYSTEM OF THE EXTERNAL PORCES IS ZERU THEREFORE, THE STATEM OF THE EFFECTIVE FORCES MUST ALSO 8 - EQUINALENT TO BERON SINCE THE LARGED OF THE TWO MASSES IS PLACED ABOVE THE 2 AXIS, THE MASS CENTER G OF THE UNBALANCED VITEEL ALST PE PELOW THAT AXIS



+1 Z(F) =0: man - 111 2 + 110 a =0 (18 kg) 2 W - (0,170 kg)(0,182 m) W + + (0.056 kg) (0.182 m) W= U 18 = (0.170)(0.182) - (0.056)(0.182) 2=1.1527×10 m E=1.153 nm +) E(M) ess = 0: - m1 B a B (6.075m) - m D a (0.075in) =0

H = mp 200 (0.075) + mp 200 (0.079) = (0.170+0.056)(0.182)(0075)W H = 3.0849410-362K (1)

SINCE ma PASSES THRU C, He = H = 3,0849 x10 WK (2) BUT H = I, Wi-I, W; -I, WK

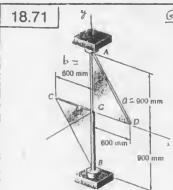
AND $H = \omega_i \times (I_i \omega_i - I_j \omega_i - I_j \omega_k) = -I_j \omega_k + I_i \omega_i$ EWATING (2) AND(3), WE HAVE - I = 3.0849 × 10 3, I2=0

Iz= -3.08 g. m2,

(b) THE FORCE-COURS STATEM EXERTED ON THE WHEEL REPORT THE CORATCHE MAY BY HAVE EEE IN A TRACKED IS EQUAL TO THE EFFECTIVE FURCES:

F= ma = mawi = (18kg/1.1527×10 m)(15×217 red/s) = (184.3 N) Mc=M=H=3.0849 x10 (15x21 rad/s) k = (27.4 N.m) k

THE FORCE COUPLE SYSTEM EXERTED BY THE WHEEL ON THE MACHINE BEFORE THE CORRECTIVE MASSER HAVE BEEN ATTACHED



ASSETT: , OF PROB. 18.6:
COVERCTING OF TWOTRIANGUIAN
PLATES EACH OF MASS
THE J KG , IS AT REST
WHEN A COUPLE OF
MOMENT MO = (36N·m);
IS APPLIED TO SHAFT AB.
FIND:
(a) ANGULAR ACCELERATION
OF ASSEMBLY,
(b) INITIAL DYNAMIC

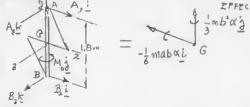
REACTIONS AT A AND B.

SEE SOLUTION OF THOS. 10,65 FOR DERIVATION OF ER. (2): $H_{G} = -\frac{1}{6} mab \omega L + \frac{1}{3} mb' \omega j \qquad (2)$

EU (18,72): $\frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{GXYZ} + \Omega \times \frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{GXYZ} + O$ SINCE $\Omega = 0$ WHEN COUPLE IC APPLIED, THUS $\frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{GXYZ} = -\frac{1}{G} \max b^{2} \alpha \frac{1}{2} + \frac{1}{3} m b^{2} \alpha \frac{1}{2}$ (3)

EQUATIONS OF MOTION: EQUIVATENCE OF A PILE D AND

EFFECTIVE FORCES.



 $\sum M_{B} = \sum (M_{B})eff;$ $(1.8m)\frac{1}{2} \times (A_{x}\frac{1}{2} + A_{x}\frac{k}{2}) + M_{0}\frac{1}{2} = -\frac{1}{6} mab \alpha \frac{1}{2} + \frac{1}{3} mb^{2} \alpha \frac{1}{6}$ $-1.8A_{x}\frac{k}{2} + 1.8A_{x}\frac{1}{2} + M_{0}\frac{1}{3} = -\frac{1}{6} mab \alpha \frac{1}{2} + \frac{1}{3} mb^{2} \alpha \frac{1}{3}$

EQUATING THE COEFF. OF L, 1, K:

- (4)
- (1) $H_0 = \frac{1}{3}\pi.b^2 \propto (5)$

(a) AHGIRAR ACCELERATION
SUBSTITUTING GIVEN DATA IN (5):

36 N·m = 1/3 (5 kg) (0,6 m) X

d = 60.0 rad/s

(1) PRINCE CONTRACTOR

EO (4): (1.8 m) Az = - = (5 kg)(0.9 m)(0.6 m) (60 rod/s)

Az = - 15.00 N

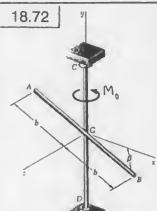
RICH' ING RO (1), Az=0,

A = - (15,00 N) k

 $\Sigma \underline{F} = \Sigma(\underline{F})_{eff}$:

A+B=0, B=-A.

P = (15.00 H)K



GIVEN:

ASSEMBLY OF PROB. 18.66,
CONSISTING OF ROD OF MASS MI
WELDED TO SHAFT CD OF
(ENGTH 2b. ASSEMBLY IS AT
REST WHEN COUPLE OF
MOMENT M = H. 15
APPLIED TO SHAFT CD
FIND:
(a) ANGULAR ACCELERATION
OF ASSEMBLY,
(b) IN ITIAL DYNAMIL
REACTIONS AT C AND D.

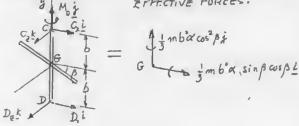
SEE SOLUTION OF PROB 18.66 FOR DERIVATION OF ED. (1):

HG= 1/3 mb2 cosp(singi+cospj) (1)

EQ. (18.22): $\frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{G2y2} + \Omega \times \frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{G2y3} + 0$ Since $\Omega = \omega = 0$ when couple is applied. Thus $\frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{G2y3} = \frac{1}{2} mb^2 \alpha \cos\beta \left(\sin\beta \frac{1}{2} + \cos\beta \frac{1}{2} \right)$ (2)

FOUNTIONS OF MOTION: FOUNTALENCE OF APPLIED AND

3 M & ZFFECTIVE FURCES.



- (1) $2bC_3 = \frac{1}{3}mb^4\alpha sinfcosfs$ (3)

- (a) ANGULHR ACCELERATION

PROM 20.(4): \(\alpha = 3 M_0 / m b cos \beta

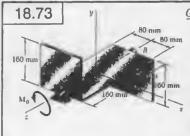
(b) INITIAL DYNAMIC REACTIONS

Priori =0 (3): $C_2 = \frac{1}{6} \text{ mb of sin } \beta \cos \beta = \frac{1}{6} \text{ mb sin } \beta \cos \beta (3M_0/mbas^2 \beta)$ $C_2 = (H_0/2b) \tan \beta$

RE(ALLING ER. (5), C, = 0,

C = (Mo/2b) tan pk

 $\Sigma F = \Sigma(F) \cdot \omega$: $C + D = 0, \quad D = -(M_0/2b) \cdot ton \beta k$



2.4-Kg COMPONENT SHOWN 15 AT REST WHEN COUPLE M = (0,8 N·m) K IS APPLIED 76 IT.

FIND!

(A) ANE. ACCELERATION (L) DYNHMIC REPCTIONS AT A AND & HYMEDIATELY AFTER COUPLE IS APPLIED

COMPUTATION OF HOMELTS HUD PRODUCTS OF INERTA



TUTAL MASS = 411 = 2.4 kg, a=160 mm PORT: 11: $I_2 = \frac{1}{12} \left(\frac{\partial M}{2} \right) a^2 = \frac{1}{24} \pi 1 a, \quad I_{3/2} = I_{3/2}$ PURTIONS 2 AND 3: $\underline{I} = 2\left(\frac{m}{4}\right)\left[\frac{\alpha}{6} + \left(\frac{\alpha}{2}\right)^2\right] = \frac{5}{14} \approx 10^{\circ}$ $\underline{I}_{yz} = \mathcal{R}(\frac{m!}{4})(\frac{a}{2})a = \frac{1}{4}ma$, $\underline{I}_{zz} = 0$

 $I_2 = \frac{1}{24} ma^2 + \frac{5}{14} ma^2 = \frac{1}{4} ma^2$, $I_{32} = \frac{1}{4} ma^2$, $I_{32} = 0$

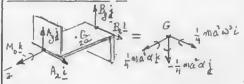
ANGULAR MOMENTUM

H = - Izz 2 - Iz w j + Iz w k = 0 - 1/4 maws + 1/4 maws H= + + 1 a w (- + + t)

RATE OF CHANGE.

PQ. (18.12) TIPLDS, SINCE Q= WK, He = (He) Gry + WK × HE = 1 maa (-j+k) + wkx 1 maw (-j+k) = 1 mad (-1+1)+1 mawi $H = \frac{1}{4} ma \left(\omega_{\perp} - \alpha_{j} + \alpha_{k} \right)$

EQUATIONS OF MUTICA



IM = E(M8) + 20kx (A21+Axi) + M6k = H6 20A, i-10A, i+M, K= 1 mawi - 4 max + + + mix x + (3)

(a) ANG. ACCELERATION

(a) ANG, ACCESTRUTE

EQUATE CUEFF, OF K IN (3): $M_0 = \frac{1}{2} m a^2 K \qquad 0 = \frac{4M_0}{ma^2} = \frac{4(0.8 \text{ M/n})}{(2.4 \text{ kg/D}, 16m)^2} = 52.0°3 \text{ rady}$ (4)

a = 52.1 rod/s'

(b) DYNAMIC REACTIONS

EQUATE COEFF. OF & IN (3): 2aAx = - 1 mdx = - 1 md 4Mu = - Mo

 $A_3 = -\frac{M_b}{2a} = -\frac{0.8 \text{ Nim}}{2(0.16\pi)} = -2.50 \text{ N}$

FRUATE CUEFF, OF i IN (3):

Ay = - Imaw (6) -2 a Ay = 1 n1a w

A =- (2.50N) 1 1 THUS; SINCE W= 0, Ay = 0; THUS:

EF = Z(F), 1: A+B=0, B=-A, B = (2,50N)i . 18.74

GIVEN: COMPONENT OF PROB. 40, 73. FIND: DYNAMIC REACTIONS AT A AND B AFTER ONE FULL REVOLUTION

SEE SOLUTION OF PRUB 12.73 FOR DERIVITION OF EUS. (4) (5), AND(6) FRON EQ. (4), 0 = 52.083 rad/s.

PUR ONE PULL REVOLUTION, 8 = 271 rad PROH EUS. (15.16):

ω=2 αθ= 2 (52,083 md/s)(271 md)=654,49 rad/s ER. (5): Az=-2,50H

EQ (6): Ay = - 18 (2.4 kg)(0.16 m)(654.41 rod/3)= -31.4N
THERE PORE:

 $A = -(2,50 \text{ n}) \cdot (31.4 \text{ n}) \cdot \beta \cdot B = -A = (2.50 \text{ n}) \cdot (31.4 \text{ n}) \cdot \beta$

18.75

GIVEN: 16-16 SHAFT OF PADE 18.67 IS ATREST WHEN A COUPLE M. IS APPLIED TO 17, CAUSING ANGULAR ACE: &= (20 rad/5)6. FIND:

(a) COUPLE Mo. (b) DYNHALL REACTIONS AT A AND & INMEDIATELY AFTER M. IS APPLIED.

SEE SOLUTION OF PRUB. 18.67 FOR DERIVATION OF H = 10 mawi - 2 maw k (1)

EQ. (18.22): HG = (HG)G242 + 1 × HG = (HG)G242 + 0

 $H_{G} = \frac{10}{3} ma^{2} x i - 2 ma^{2} x k$ WHERE a=9in =0.75ft AND m= 10(1616)=216

EDUATIONS OF MOTION

IM = Z(M) eff: 4aix(By j+B, E)+Mi= 10 max i-20 max 4a \$ K - 4a 8 j + M. i = 10 maa i - 2 maak (3)

(a) COUPLE M.
ENUATE CUTP. OF i IN ER (3): $M_{0} = \frac{10}{3} \text{ mn}^{2} \Omega = \frac{10}{3} \frac{2/b}{32.2H/s} \cdot (0.75H)^{4} (20 \text{ rad/s}) = 2.329 \text{ lb-ft}$ M=(2,33 1b.ft)i.

(b) DYNAMIC REACTIONS AT t=0

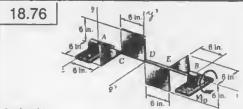
EQUATE COFFF OF 2 IN EW (3):

EQUATE COEFF. OF & IN EQ. (3): 4a By = -9 ma' x 9h

By = - 1 ma a = - 1 2/b (0,75 ft) (20 md/s) = - 0,466 lb THEREFORE: B = - (0.466 15) j

A =- B = (0,466 /b)j ZF = Z(F, 4): A+B=0

A = (0,466 16) j; E= - (0,466 16) j



GIVENI.
THE 2.7-16 ASSEMBLY OF PROB. 18.68 IS AT REST
WHEN A COUPLE M IS APPLIED TO AXLE AP, LA KINK
AN ANGULAR ACCELERATION Q = (150 rad/s):

FIND: (a) THE COUPLE M.

(b) THE DYNAMIC REACTIONS AT A AND B IMMEDIATELY AFTER MO IS APPLIED.

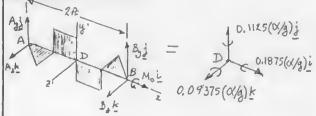
SEE SOLUTION OF PROB. 18.68 FOR DER HATION OF F.W.(1):

H = (0.1875 1 + 0.1125 1 + 0.0875 1/20/6) (1)

 $\frac{H}{D} = (0.1875 \stackrel{!}{L} + 0.1125 \stackrel{!}{d} + 0.09375 \stackrel{!}{K}) (\omega/g) \qquad (1)$ WHERE THE NOWERICAL WILLIES ARE EXPRESSIBLY 16.5t² $EQ.(18.22): \stackrel{!}{H} = (\stackrel{!}{H})_{Dzy^{12}}, + \Omega \times H_{D} = (\stackrel{!}{H})_{Dzy^{12}}, + 0$

SINCE à= a: H= (0,1875 i+0,1125 j+0.09375)(a/a) (2)

EQUATIONS OF MOTION



 $ZM_A = Z(M_A)_{eff}$: $(24)i \times (B_j \hat{j} + B_j \underline{t}) + M_0 \underline{1} = 0.1875(W_j)i + 0.1125(W_j)j + 0.01275(W_j)k$

(11) By K-(211) By + Mo = 0.1875() + 0.1125() +0.07375() +

(a) SOUPLE HO

ENURTE CORPT. OF ! IN ER (3): $M_0 = 0.1875 (\alpha/6) = (0.1875 /b.H^2) \frac{150 \text{ ranks}}{32.2 \text{ H/s}^2} = 0.873 /b.ft$ $M_0 = (0.873 /b.ft) !$

(b) DYNAMIC REACTIONS AT += 0.

(24) B=0.09375 (a/g)-(0.09375 16.4+) 150 radd - 0,43672 16.4+ (a) COUPLE MO
TOWNSTE COFFE OF

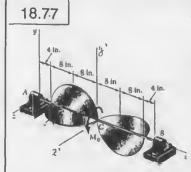
By = 0,218 lb

EQUATE COEFF. OF j IN EQ(3): $-(2ft)B_3 = 0.1125 (a/g) = (0.1125 |b.ft) \frac{150 \text{ red/s}^2}{32.2 \text{ f/s}^2} = 0.52407 \text{ lb.ft}$ $B_3 = -0.262 \text{ lb}$

THUS: B = (0,218 16) j - (0,262 6) k

ΣF = Σ(F) of A+B=0, A=-B

A = - (0.21816) j + (0.26216) k



GIVEN:

ASSEMBLY WEIGHS

12 IL AND CONSISTS OF

4 SEMICIRCIAN FRATES.

ASSISTLY IS WITHELL

AT t = D WHEN COUPLE

MO IS APPLIED FOR

OME FULL PENALUTION

WHICH LAST: 25.

FIND; (A) THE COUPLEM

(6) THE DYNALILL REACTION

AT A ANJ B AT t = 0

MASS OF ASSEMPLY = $mL = \frac{12 \text{ lb}}{32,2475} = 0.37267 \text{ lb.s}/4t$ RADINS OF SEMICIRCULAR PLATES = $t = 8 \text{ in.} = \frac{3}{3} \text{ ft}$ HOTENTS AND PROJUCTS OF INE IT IA

POR ASSEMBLY: $I_2 = 2(\frac{nt}{z})\frac{t^2}{7} = \frac{1}{7} \text{ ni.} 2$ FOR EACH VERTICAL PLATE: $I_{23} = \frac{m}{7} 2 \frac{\pi}{9} = -\frac{m}{4} t \left(\frac{nt}{37}\right)$

FOR EACH VERTICAL PLATE: $I_{23} = \frac{4\pi}{3} 2g = -\frac{4\pi}{4} 2 \left(\frac{4\pi}{5\pi}\right)$ $I_{23} = -\frac{m^2}{3\pi} \frac{2\pi}{3\pi} = 0$

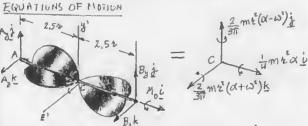
FOR ASSETTION IN THE FORM ASSETTION IN THE PROPERTY OF THE PR

ANGULAR MONTHNOWS

FROM EDS. (18.13) WITH $\omega_z = \omega$, $\omega_z = \omega_z = 0$; $H = I_z \omega_z^2 - I_z \omega_z^2 - I_z \omega_z^2 = m \epsilon^2 \omega_z^2 \left(\frac{1}{4} i + \frac{2}{3\pi} i + \frac{2}{3\pi} i + \frac{2}{3\pi} i \right)$ (1)

$$\begin{split} EQ.(18.22): & \stackrel{\circ}{H} = (\stackrel{\circ}{H}_{C})_{CZg'z}, + \stackrel{\circ}{D} \times \stackrel{H}{H}_{C} \\ & \stackrel{\circ}{H} = mt^{2}\alpha(\frac{1}{4}i + \frac{2}{3\pi}j + \frac{2}{3\pi}k) + \omega i \times mt^{2}\omega(\frac{1}{4}i + \frac{2}{3\pi}j + \frac{2}{3\pi}k) \\ & = mt^{2}\alpha(\frac{1}{4}i + \frac{2}{3\pi}j + \frac{2}{3\pi}k) + \frac{2}{3\pi}mt^{2}\omega(k - \frac{1}{4}i) \end{split}$$

 $\frac{\dot{H}}{C} = \frac{1}{4} m \dot{z}^2 \alpha \dot{i} + \frac{2}{3\Pi} m \dot{z}^2 (\alpha - \omega^2) \dot{j} + \frac{2}{3\Pi} m \dot{z}^2 (\alpha + \omega^2) \dot{k}$ (2)



 $\sum_{A}^{M} = \sum_{A} (M_{A})_{SS} : 52 i \times (B_{y}j + B_{z}k) + M_{0}i = H_{0}$ $52 B_{y}k - 52 B_{z}j + M_{0}i = \frac{1}{4} m 2^{2} \alpha i + \frac{2}{37} m 2^{2} (\alpha - \alpha')j + \frac{2}{37} m 2^{2} (\alpha + \alpha')k$ (3)

FOUNTE COEFF, OF i: $M = \frac{1}{4}m^{\frac{1}{2}}\alpha'$ SINCE ASSEMBLY ROTATES THROUGH $\theta = 2\pi r_0 A$ in $2 \le 1$ $\theta = \frac{1}{4}\alpha t^2$, $\alpha = 2\theta/t^2 = 4\pi/4 = \pi r_0 A/5^2$. THUS! $M_0 = \frac{1}{4}(0.37267 \text{ lb} \cdot \text{s}^2/\text{ft})(\frac{2}{3}\text{ft})^4(\pi r_0 A/5^2) = 0$; 1301 lb. ft $M_0 = \frac{1}{4}(0.37267 \text{ lb} \cdot \text{s}^2/\text{ft})(\frac{2}{3}\text{ft})^4(\pi r_0 A/5^2) = 0$; 1301 lb. ft $M_0 = \frac{1}{4}(0.37267 \text{ lb} \cdot \text{s}^2/\text{ft})(\frac{2}{3}\text{ft})^4(\pi r_0 A/5^2) = 0$; 1301 lb. ft

(b) DYNAMIC PEACHONS AT t=0

EWATING THE COEFF. OF JANUA IN (=) ALL FITTING W=0 AND X= Trod/St;

(a) $-5\pi B_z = \frac{2}{5\pi} m \, e^{t} (\pi \, rod/e^{t}), B_z = -\frac{2}{15} (0.37267)(\frac{2}{3}) = -0.0331 \, lb$ (b) $5\pi B_y = \frac{2}{3\pi} m \, e^{t} (\pi \, rod/e^{t}), B_z = +0.0331 \, lb$

 $B = (0.0331 | b)_{\frac{1}{2}} - (0.0331 | b)_{\frac{1}{2}}$ $\Sigma F = \Sigma F_{eff} : A + B = 0, A = -(0.0331 | b)_{\frac{1}{2}} + (0.0331 | b)_{\frac{1}{2}}$ 18.78 GIVEN: ASSEMBLY OF PROB. 18.77

FIND: DYNAMIC REACTIONS AT A AND B ATT = 25. SEE SOLUTION OF FROM, 18,77 FOR DERIVATION OF EQ. (3) 5284K-52B2 + Moi = 4 m2 a i + 2 m2 (a-w)j+2 m2 (a+w)k WHERE M = 0,37267 16.5/ft AND . E== ft SINCE ASSENCLY ROTHIES THROUGH &= 2 TT rad in 2 =: 0 = 1 xt, \ \alpha = 20/t = 417/4 = 17 rod/s AT t=25: W = Qt = (Trad/s)(23) = 27 rad/s EQUATING THE COEFF. OF & AND K IN EW (3) AND SUBSTITUTING THE ABOVE VALUES ! (1) -52 B2 = 2 m 2 (17-4Th) $B_{2} = -\frac{2}{15}(0.37267)(\frac{2}{3})(1-477) = +0.383 \text{ } B$ とっこ (1+47) B 52 B, = 2 m t (17+47) By = 2 (0.37267)(2/3)(1+47) = 0.449 B B=(0,447/6);+(0,383/b)k IF = I(F)eH: A+B=0 A = - (0,44919) - (0.38516) K.

18.79 GIVEN:

FLYWHEEL RIGIDLY ATTICHTS TO CRANICHIPT OF AUTOMOBILE ENGINE IS EQUIVALENT TO "CO-MIN-DIAM", 15-mai-thick Steel find (DENSITY = 7860 kg/m³). AUTUALOBILE TRAVELS ON UNBANKED CILVE OF 200-M RADIUS AT GORM/h WITH PLYWHEEL FIOTATING AT 2700 pm. FIND:

MAGNITUDE OF COUPLE EXERTED BY FLYNHEEL ON CHANKE HITT,

(a) REAR WHITEL DRIVE WITH ENGINE MOUNTED LIKE WITH END AND MENTED LIKE WITH END ME ME MOUNTED TRANSFER - 1.

(A) REAR-WHEEL DRIVE (LONGITUDINAL MOUNTINE)

ASSURIE SENSES SHOWN FOR W. D. T.

T = 70 km/n = 55 m/s

 $G = \omega_1 = \omega_2 =$

 $\omega_{2} = 2700 \text{ rpm} \left(\frac{277 \text{ rad/s}}{60.5} \right) = 281.74 \text{ rad/s}$ $\omega_{3} = \frac{\overline{\upsilon}}{P} = \frac{25.95}{2000} = 0.125 \text{ rad/s}$ $\overline{L}_{x} = \frac{1}{2} 10.2^{2} = \frac{1}{2} (P.72^{2} t) z^{2}$

== (() = (; 660 kg/m) ((0,2 m) (0.015 m) = 0.29622 kg m

ANGULAR MOMENTUM A BOUT G:

H= I, W, L+I, Wy

ER. (18.22): $\dot{H}_{G} = (\dot{H}_{G})_{GXYZ} + \dot{\Omega} \times \dot{H}_{G} = 0 + \dot{\omega}_{g} \dot{i} \times (I\dot{\omega}_{g} \dot{i} + I\dot{\omega}_{g} \dot{j})$ $\dot{H}_{G} = -\dot{I}_{x} \dot{\omega}_{x} \dot{\omega}_{g} \dot{k} = -(0.29632 \, kg.m)(282.14 \, rad/s)(0.125 \, rad/s) \dot{k}$ $\dot{H}_{G} = -\sqrt{(0.47 \, N \cdot m)} \dot{k}$

THE COUPLE EXERTED ON THE PLYWHEEL, THEREFORE, MUST BE M = HG = - (10.47 N·m) & AND THE COUPLE EXERTED BY THE FLYWHEEL IS - M = (10.47 N·m) &

ANSWER: 10,47 N.M

(CONTINUED)

18.79 continued

(b) FRONT-WHEEL DRIVE (TRINSVERSE MOINTING)

WE ASSUME THE SAME DIRECTION OF MOTION OF THE CAR AS INFARTA
REFERRME TO THE NUMBERICAL
VALUES FOUND IN PARTA:

Wa = 28274 rad/s

Wy = 0.125 rad/s

T. = 0.29632 kg/m²

KNEULHR MOHENTUM ABOUT 6:

$$\begin{split} & \underbrace{H}_{G} = \overline{J}_{g} \omega_{g} \underbrace{i} + \overline{I}_{g} \omega_{g} \underbrace{K} \\ & E0.(18.22): & \underbrace{H}_{G} = (\underbrace{H}_{G})_{GRY2} + \underbrace{\Omega}_{X} \underbrace{H} = 0 + \omega_{g} \underbrace{j}_{X} (\overline{J}_{g} \omega_{g} \underbrace{j}_{g} + \overline{J}_{g} \iota_{g} \underbrace{k}_{g}) \\ & \underbrace{H}_{G} = \overline{I}_{g} \omega_{g} \omega_{g} \underbrace{i} = (0.29631 \underbrace{kg.m}_{g} \underbrace{N.0.75 \text{ red/g}}_{g} \underbrace{282.7 \text{ y rad/g}}_{g}) \underbrace{i}_{g} \\ & \underbrace{H}_{G} = (10.47 \underbrace{N.m}_{g}) \underbrace{i}_{g} \end{split}$$

THE COUPLE EXECUTED ON THE FLYWHEEL, THEFEFORE,

MUST BE M = H = (10.47 N·m)1, AND THE COUPLE

EXERTED BY THE FLYWHEEL IS M = -(10.47 N·m)i

ANSWER: 10,47 N·m

18.80 | GIVEN:

FOUR-BLADED AIRPLANE PROPELLER
WITH AM = 160 kg AND E = 600 mm ROTATES AT 1600 pm.
AIRPLANE IS TRAVELING IN CIRCULAR PATH WITH
C = 600 m AT V = 540 km/h.
PIND:

MAGNITUDE OF COUPLE EXERTED BY PROPELLER ON

 $\Omega = \omega_1 = \omega_2$ $\Omega = \omega_2 = \omega_2$ $\omega_2 = \omega_2$

WE ASSUME SENSE SHOWN

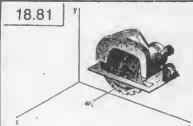
FOR ω_1 , ω_2 , AND \overline{v} $\overline{v} = 540 \text{ km/h} = 150 \text{ m/s}$ $\omega_1 = 1600 \text{ rpm} \left(\frac{277 \text{ rad}}{60 \text{ s}}\right)$ $\omega_2 = 167.55 \text{ rad/s}$ $\omega_3 = \frac{\overline{v}}{2} = \frac{750 \text{ m/s}}{600 \text{ m}} = 0.25 \text{ rad/s}$

J=mk2=(160kg)(0.8m)2=102.4 kg.m2
ANGULAR MUMENTUM ABOUT G:

 $\begin{array}{l}
\underline{H}_{G} = \bar{I}_{2} \omega_{2} \hat{i} + \bar{I}_{y} \omega_{y} \hat{a} \\
EQ (18.22); \ \underline{\dot{H}}_{G} = (\underline{\dot{H}}_{1})_{G2} + \underline{\Omega} \times \underline{H}_{G} = 0 + \omega_{y} \hat{j} \times (\bar{I}_{2} \omega_{2} \hat{i} + \bar{I}_{3} \omega_{y} \hat{j}) \\
\underline{\dot{H}}_{G} = -\bar{I}_{x} \omega_{x} \omega_{y} \hat{k} = -(102.4 \text{ kg·m}^{2})(167.55 \text{ rad/s})(0.25 \text{ rad/s}) \hat{k} \\
\underline{\dot{H}}_{G} = -(42.89 \text{ N·m}) \underline{K} = -(4.24 \text{ kN·m}) \underline{K}
\end{array}$

THE COUPLE EXECUTED ON THE PROPELLER, THEREFIXE, MUST BE M = H = - (4.29 kN·m)k, AND THE COUPLE EXERTED BY THE PROPELLER OIL ITS SHAFT IS - M = (4.29 kN·m)k.

ANSWER: 4.29 KN.M

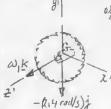


FOR BLADE AND ROTOR OF MOTOR OF PORTABLE SAW:

W=2,51b, k=1.5 in. BLADE ROTATESAS SHOWN AT RATE W, = 1500 rpm

FIND: COUPLE M THAT WORKER MUST EXERT ON HANDLE TO ROTHTE SAW WITH CONSTANT Q = - (2,4 rad/s) }.

USING AXES LENTERED AT MAYS CENTER G OF BLADE AND RUTOR AND ROTATING WITH CASINGS

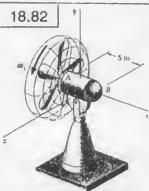


0)= ω, = 1500 rpm (211 rad/s 1= Wy = Wz = -2.4 rods $\overline{I}_{\underline{x}}^{2}, = \frac{W}{4} \overline{k}^{2} = \frac{2.5 / b}{32.2 \text{ f/s}} \left(\frac{1.5}{12} \text{ ft}\right)^{2}$ = 1.2/3/X/03/b.ft.5

> ANGULAE MONENTUN A EUST G: He=I,Wij+I, wak

EQ.(10,22); = I, w, w, i = (12131×10 16.fts')(-2.4rod/s)(507 rod/s)i

THE COUPLE THAT THE WORKER MUST AITPLY IS M = HG M =- (0.457 16.ft) i



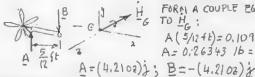
GIVEN:

FOR BLADE AND RUTAR OF MOTOR OF OSCILLATING FAN: W= 802, K = 3 in. BEARING SUPPORTS AT A AND B ARE Sin, APART. BLADE RUTHTES AT RATE

W.= 1800 rpm. FIND: DYNAMIC REACTIONS AT A AND B WHEN MUTOR CASING

HUS ANG. VEL. W= (0,6 rd/s)).

ANGULAR MOMENTUM PEOUT MASS CENTER! HG= IZWi+I, Wj = IWi+I, Wi EQ. (18,22): $\underline{H}_{G} = (\underline{H}_{G})_{G1Y2} + \Omega \times \underline{H}_{G} = O + \omega_{z} j \times (\underline{T}_{\omega}, \underline{i} + \underline{T}_{y}, \omega_{z} j)$ $H_G = -\frac{7}{2}\omega_1\omega_2 = -\frac{(B/16)^3b}{322755} (\frac{3}{12}ft) (1800 rpm) \frac{27008}{603} (0.6 red/s) \pm \frac{1}{12} (0.6 red/s) \frac{1}{12} (0.6 red/s) + \frac{1}{12$ = - (0, 10976 16.ft) k THE REACTIONS AT A AND B



FORM A COUPLE EQUIVALENT A (5/12+E)=0.10976 16.ft A= 0:26343 1b = 4.21 07

GNEN: 18.83

AUTOMOBILE TRAVELS AROUND UNBANKED CURVE WITH Q= 150 M AT STRED V = 95 km/n.

PUR EACH WHEEL: M = 27 kg , DIAM ,= 575 mm , K = 2.25 mm TRANSVERSE DISTANCE BETWEEN WHEELS = 1.5 m.

FIND! ADDITIONAL REACTION OF EXERTED BY GROUND ON EACH OUTSIDE WHEEL DUE TO MOTION OF CAR.

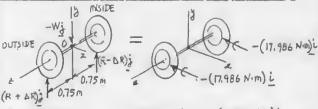


FOR EACH WHEEL! V = 95 Km/n = 26.589 m/s N = 26289 11/5 = 0.17593 rad/s 150 m *=- 26.389 111/5 =- 91.787 rod/s I = m k = (22 kg)(0,225m) = 1.1138 kg.ni

ANGULAR MOMENTUM OF EACH WHEELS H = 1, w, j + 1, w, k

H= I w wz L=(1.113B Kg. m²)(0,17593rad/s)(-91,787rad); i=-(17,986 N·m)i

EQUATIONS OF MUTION FOR TWO WHEELS ON SAME AYLE:



 $\Sigma M_0 = \Sigma (M_0)_{ey}$: $-2(\Delta R)(0.75m) \underline{i} = -2(17.986 N·m) \underline{i}$ AR = 2398 N DR = 24.0 NT

GIVEN:

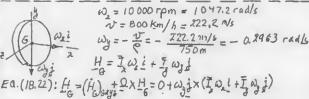
18.84



TYPE OF AIRCRAFT TURN INDICATOR. UNIFORH DISK: M = 2009, 2= 40 mm SPINS AT RATE OF 10000 FPM. EACH SPRING HAS 500-N/M CONSTANT. SPRINGS EXERT EGUAL FORCES ON YOKE AB IN STRAIGHT FLIGHT PATH.

FIND: INGLE OF ROTATION OF YOKE IN HORIZ-

ONTAL TURN OF 750-M RADIUS TO THE RIGHT WITH U= 800 km/h. DOES A MOVE UP OR DOWN?



-1 w, w, k =- = (0.2 kg)(0.04m)(1047,2 md/s)(-0.2963 od/s) k = + (0.049645 N·m) k



WEHAVE (0.1m)F= HG= 0.049645 N·M F= 0.49645 N F=kz: z= 0.49645N = 0.9929 mm 0= L = 0.9929 mm = 0.01906 rad= 1,14

SINCE SPRING AT A PULLS DUWN, A 15 MOVING UP

18.85 and 18.86

GIAEN:

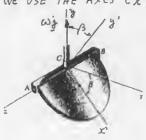
SEMICIRCULAR PLATE WITH t= 120 nm is hinged to CLEVIS WHICH ROTATES WITH CONSTANT W.



PROBLEM 18.851

FIND:
(a) B WHEN W= 15 rad/s,
(b) LARGEST W FOR WHICH
PLATE REMAINS VERTICAL (\$p = 90)
PROBLEM 18.86:
FIND W FORWHICH \$p = 50°.

MIMENTS AND PRODUCTS OF INERTIA.
WE USE THE AXES CX'8' + SHOWN.



WE NOTE THAT I_{34} , AND I_{34} .

ARE HALF THUSE FOR A CIRCULAN PLATE, AND SO IS THE MASS m, THUS $I_{24} = \frac{1}{4} mt^2$ $I_{37} = \frac{1}{2} mt$ BECAUSE OF SYMMETRY, ALL PRODUCTS OF INSERTING ARE EQUAL TO 25 RO! $I_{4/34} = I_{3/2} = I_{24} = 0$

ANGULAR MONENTUM ABOUT C

$$\begin{split} \frac{H}{c} &= I_{2}, \omega_{x}, \ \underline{i}' + I_{2}, \omega_{y}, \ \underline{i} \\ &= \frac{i}{4} m z^{2} (-\omega \sin \beta) \underline{i}' + \frac{1}{2} m z^{2} (\omega \cos \beta) \underline{j}' \\ &= \frac{1}{4} m z^{2} \omega \left(-\sin \beta \ \underline{i}' + z \cos \beta \ \underline{j}' \right) \end{split}$$

OR, SINCE 1 = -L' SINB + j'cos B:

 $\sum_{c} M_{c} = \omega(-i'\sin\beta + j'\cos\beta) \times \frac{1}{4}mz\omega(-\sin\beta i' + 2\cos\beta j')$ $= \frac{1}{4}mz^{2}\omega^{2}(-2\sin\beta\cos\beta k + \cos\beta\sin\beta k)$ $\sum_{c} M_{c} = -\frac{1}{4}mz^{2}\omega^{2}\sin\beta\cos\beta k \qquad (1)$

Z. G

BUT $\sum M_c = -mg x' \cos \beta k$ = $-mg \frac{42}{377} \cos \beta k$ (2)

EQUATING (1) AND(2): \[\frac{1}{4} m \frac{1}{2} \omega^2 \sin \beta \cos \beta = \frac{4 mg \pm }{3 \text{ if }} \cos \beta \]

 $\omega \sin \beta = \frac{16}{3\pi} \frac{g}{z} = \frac{16}{3\pi} \frac{9.81 \text{ m/s}^2}{0.12 \text{ m}} \quad \omega \sin \beta = 132.78 \text{ s}^2 (3)$

PROBLEM 18.85

(a) LET $\omega = 15 \text{ rad/s IN(3)}$: $\sin \beta = \frac{138,78}{(15)^2} = 0,61601$

B = 38.1°

(b) LET B=90° IN(3): W= 138.7852 W=11.78 rad/s

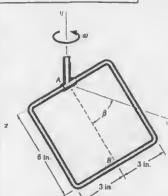
PROBLEM 18.86

LET B = 50° IN E0. (3):

 $\omega^2 = \frac{138.785^{-2}}{5in\,50^\circ} = 181,17\ s^*$

 $\omega = 13.46 \, \text{rad/s}$

18.87 and 18.88



GIVEN:

ROD BENT TO FORM 6-in.

SQUARE FRAME WHICH IS

ATTACHED BY COLLAR H. &

TO SHAFT ROTATION. WITH

CONSTANT W.

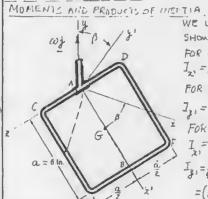
PROBLEM 18,67:

EIND: (a) B WHEN $\omega = 9.8 \text{ rad/s}$ (b) LARGEST ω FOR WHEH $\beta = 90^{\circ}$

PROBLEM 18.88:

FIND W FOR WHICH B= 48°.

(MASS OF FUAME = ATL)



WE USE THE AVES AX & 2 SHOWN FOR CD: $I_{2x} = \frac{1}{7}, = \frac{1}{12} \cdot \frac{10}{7} \cdot (\frac{1}{7} = \frac{1}{48} \text{ min}^2)$ FOR EF: $I_{2x} = \frac{1}{48} \text{ min}^2 + \frac{1}{7} \cdot \frac{10}{48} \cdot \frac{13}{7} \cdot \frac{1}{7}$

FOR CE QR DF: $I_{2} = \frac{m}{4} \left(\frac{\alpha^{2}}{2}\right)^{2} = \frac{1}{16} m a^{4}$ $I_{3} = \frac{1}{49} m a^{4} + \frac{m}{44} \left[\left(\frac{\alpha}{2}\right)^{2} + \left(\frac{\alpha}{2}\right)^{2}\right]$

 $= (\frac{1}{78} + \frac{1}{8}) \text{ pia} = \frac{7}{48} \text{ mat}$

FOR ENTIRE FRAME:

$$\begin{split} I_{2} = & \left[\frac{1}{48} + \frac{13}{48} + 2 \left(\frac{1}{16} \right) \right] n_{1} a^{2} = \frac{1}{6} n_{1} a^{2}; \\ I_{2} = & \left[\frac{1}{48} + \frac{13}{48} + 2 \left(\frac{7}{48} \right) \right] n_{1} a^{2} = \frac{7}{12} n_{1} a^{2} \\ \text{BECAUSE OF SYMMETRY: } I_{2} = I_{2} = I_{2} = 0 \end{split}$$

ANGULAR MOMENTUH A BOUT A

 $\frac{H_{A}=I_{\chi},\omega_{\chi},i'+I_{\chi},\omega_{\chi},j'=\frac{1}{6}ma'(-\omega sin\beta)i'+\frac{7}{12}ma'(\omega\omega s\beta)j'}{SINCE A 15 PIXED, WE USE EO. (18.28):}$

ZMa=(HA)Axy; + 1 × HA= O + wj x HA
OR, SINCE J = - i'sing+ i'co: f:

ΣM = ω (-i's in β + β'cosp.) y 1/2 maω (-2 sin β i' +7ωυβ j')

= $\frac{1}{12}$ maw (-7sin, scosp k + 2 cosps sings k) $\sum M_A = -\frac{5}{12}$ maw sin peosps k (1)

BUT $\sum M_{A} = -mg(\frac{a}{2})\cos\beta k$ (2) EQUATING (1) AND(2): $\frac{5}{12}mIa^{2}\omega^{3}\sin\beta\cos\beta = -\frac{1}{2}mga\cos\beta$

 $\omega^2 \sin \beta = \frac{6}{5} \frac{3}{6} = \frac{32.2 \text{ ft/s}^2}{(6/12) \text{ ft}}$ $\omega^2 \sin \beta = 77.28 \text{ s}^2$ (3)

PROBLEM 18,87
(a) LET $\omega = 9.8 \text{ rad/s}$ IN (3): $\sin \beta = \frac{77.18}{(9.8)^3} = 0.80466$ (b) $\beta = 53.6^\circ$

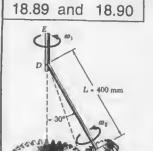
(b) LET B=90 IN (3): W=77.28 52

W= 8,79 rad/s

PROBLEM 18,88

LET B= 48° III 20 (3): \(\Omega^2 = \frac{77.285^2}{51945^2} = 103.995^2

W= 10.20 rod/5 €



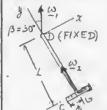
950-4 GEAR A CONSTRAINED TO ROLL ON FIYED GEAR B. BUT FREE TO ROTATE ABOUT AD. AXLE AD CONNECTED BY CLEVIS SHAFT DE WHICH ROTATES WITH CONSTANT W. (GEAR A CAN BE CONSIDERED AS THIN DISK.)

PROBLEM 18.89: FIND LARGEST ALDWAGLE IS, IF GENK A IS NOT TO LOST COURACT WITH GEAR B.

PROBLEM 18,90:

FIND FORCE F EXERTED BY GEAR B ONGEAR A WHEN W = 4 rad/s. (F 15 1 CD.)

ANGULHA VELOCITY OF GERR A



WE USE THE AXES GRY & SHOWN AND EXPRESS THAT Y = 0:

 $v_c = \omega \times D\hat{c} = 0$ ω= (ω, ω=β+ωz) j+ω, sing i

THUS:

 $U = -[(\omega, \cos\beta + \omega_z)j + \omega, \sin\beta i] \times (Lj + \epsilon i) = 0$ (W/cosp+W2)2 k- (W,sinB)(K = 0

DC = - (Lj + EL)

TIHUS: $\omega_1 \cos \beta + \omega_2 = (\omega_1 \sin \beta)(L/2)$ SUBSTITUTE INTO (1): ω = ω, sinβ(++ = 1) (2)

ANGULAR MOMENTUM ABOUT D

HD= INWithInW = m(12+ 1/4) WI sings i+ m 1/2 WI / sings H_=mw, sinp [(1+ + +) i + = 2 L)

SINCE D IS A FIXED POINT, WE USE EW. (18.28):

ZM = (HD) + 11 × HD = 0 + (W sinfi + W wsfa) × HD = mw2 sin [= ELsinp - (L + E) cos [] k

PROB. 10.89: WHEN FORCE EXERTED BY GEAR 8 ON GEAR A BECOMES ZEROS



(5) EMD = - my Lsinp K EQUITING (4) AND (5)1 m Wisin B [= 2 Lsinp - (2+ + 1) cosp] = - ny LsinB Wi [(L'+ 12) cosp - + 2 Lsinp] = gl WITH L= 0,4 m, t= 0.08 m, p=30°, 1=9.8/n/2 0. 13:45 N, = 3.924 W,= 5,45 rad/s €

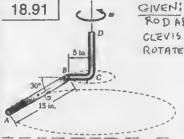
PROB. 18,90:

MOREM OF FIRE, TD = + FVL +22 K THUS, IN. (5) ABOVE MUST BE REPLACED BY ZM, = (FVL+22 - mg Lsinp)k EQUATING (4) AND (6):

mwising [themp- (L+t) cop) = FVL+2 - nig Lsinp WITH L=0.4 m, 2=0.0 am, 13=10, 2= 9.81 mb, m=0.95 kg, w= 4 rad/s:

0.95 (4) sin 30° (-0.13195) = F (0.1664-(0.95)(9.81)(0.4) sin 30° Q40792 F = 1.8637-1.0028 = 0.8611-F= 2,11 N

tan 0= = 0,2, 1= 11,31°, p-8= 18,7° F=2,11N & 18,7°



ROD AB IS ATTACHED BY A CLEVIS TO ARM BCD WHICH ROTATES WITH CONSTANT W.

> FIND: MAGNITUDE OF 00

LET L= 15 in. = 1,25 ft THEN: BC= 51n. = L/3

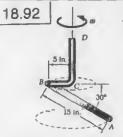
ANGULAR HOMENTUM ABOUT S

0= wsin30 i + a) cos301

H= Iwi+Iwj= 0+ +mlw ws303 = 0+ wxH = (wsin30 i+wws30 j) x 12 m [wcw30 j 12 mL w sin 30 cos 30 K

EWUATIONS OF HOTION Ma=mEw

+) ZM = Z(MB) ss: mg (= 10530°) = Hg + (ma) (= 5in30°). 1 mgle030= 17ml w31n30cos30+ m(1 cos30+ 1) w3(1 sin30) 1 1 cos30= (1 5m 30°co+30°+ 6 sin30°) w 1 32.2 ft/s cos30 = 0.2276702, W= 48,994 w=7.00 rad\$



GIVEN:

ROD AB IS ATTACHED BY A CLEVIS TO ARM BCD WHICH ROTATES WITH CONSTANT W).

FIND: MAGNITUDE OF S LET L= 15 in. = 1,25 ft THEN: BC =5 in. = 4/3

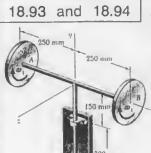
ANGULAR MOMENTUM ABOUT 6: ω=-ωsin30°i+ωω:30°j H=I, w=i+I, w=0+12m12was 30j H = (- wsin30 i+ wices30 j) x 1 m Lw cos30 f H = - 12711 W sin 30 cos 30 K



+) EM = Z(Me)+ : mig ((cos 30) = Hc + (mā) (= sin 30) 1 mg L cos 30 = 17 ml w's indoos 30+ m(= cos 30 - =) w (= sin 30) 2 = cos 30 = (3 sin 30 cos 30 - 2 sin 30) W

1 32,2 H/3 cos 30 = 0.061004 02, $\omega^2 = 182.85$ w= 13.52 rod/s

(4)



FOR ENCH DISK. M = 5 kg, Z = 100 mm W = 1500 rpm PROB 18.93:

FOR W2 = 45 CPM FIND DYNAMIC REHICTIONS HT C AND IF (a) BUTH DISKS ROTHTE AS SHOWN

(b) DIRECTION OF SAIN OF B 15 REVERSED PROB. 18,94:

FIND MAY, ALLOWAETE WZ
IF DYNAMIT PERCETONS
AT CAME DARE NOT TO
EXCEED 250 N EACH.

ANGULAR MONENTING OF EPP LISK ACCUT ITS HIMSS CENTER

$$\frac{H_{G} = \frac{1}{4} \omega_{2} \dot{i} + \frac{1}{4} \omega_{3} \dot{j} = \frac{1}{2} n_{1} \dot{i}^{2} \omega_{3} \dot{i} + \frac{1}{4} n_{1} \dot{i}^{2} \omega_{2} \dot{j}}{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})} \qquad (1)$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}$$

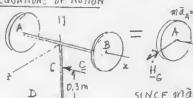
$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j})}$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j}^{2})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{2} \dot{j}^{2})}$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{3} \dot{j}^{2} + \omega_{3} \dot{j}^{2})}{H_{G} = 0 + \omega_{2} \dot{j}^{2} \times \frac{1}{4} n_{1} \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{3} \dot{j}^{2})}$$

$$\frac{H_{G} = \frac{1}{4} m \dot{i}^{2} (-2 \omega_{3} \dot{i}^{2} + \omega_{3} \dot{j}^{2} + \omega_{3} \dot{j}^{2} \dot{j}^{2} + \omega_{3} \dot{j}^{2} \dot{j}^{2} \dot{j}^{2} + \omega_{3} \dot{j}^{2} \dot{j}^{2}$$

EQUATIONS OF HUTION



 $\frac{m_{\bar{A}_{A}^{n}}}{m_{\bar{A}_{Z}^{n}}} = m_{\bar{A}_{A}^{n}} \frac{1}{m_{\bar{A}_{Z}^{n}}} = -m_{\bar{A}_{A}^{n}} \frac{1}{m_{\bar{A}_{Z}^{n}}} = -m_{\bar{A}_{Z}^{n}} = -m_{\bar{A}_{Z}^{n}} = -m_{\bar{A}_{Z}^{n}} = -m_{\bar{A}_$

SINCE MA AND MA CHUCEL OUT, EFFECTIVE FICES REDUCE TO COUPLE 2H = m & WIW. K

IT FOLLOWS THAT THE REACTIONS FORM AN EQUIVALENT COUPLE WITH $-C = D = (m \pm \omega_1 \omega_2/0.3 \text{ m}) \text{ i}$ (3)

PROBLEM 18,93

(0) WITH M=5 kg, 2=0.1m, 10=1500 rp=1-50 il rad/s, AND

W1=45 rpm=1.5 Trod/s, E0.(3) YIELDS

C=D=(5kg)(0.1m)*(50 Trad/s)(157 rad/s)(0.3m)=123,37 N

 $C = -(1/3.4 \text{ H}) \dot{L}; D = (123.4 \text{ N}) \dot{L}$

(b) WITH DIRECTION OF SPIN OF B REVERSED, ITS
ANGULAR FLOMENTUM WILL ALSO BE REVERSED AND
THE EFFECTIVE FORCES (AND, THUS, THE APPLIED
FORCES) REDUCE TO ZERO:

C = D = 0

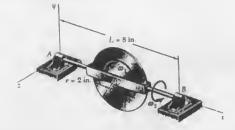
PROBLEM 18,94

MAKING C = D = 250 M IN EQ. (3) YIELDS $ME^2 \omega_1 \omega_2 = 250 \text{ M}$

WITH m1=5 kg, 12 = 0,1 m, W, = 1500 rpm = 50 Trad/s

WE HAVE $\omega_2 = \frac{(250 \text{ N/}(0.3 \text{ m}))}{(5 \text{ kg})(0.1 \text{ m})^2 (50 \text{ Trad/s})} = 9.5493 \text{ rad/s}.$ $\omega_2 = \frac{(250 \text{ N/}(0.3 \text{ m}))}{(5 \text{ kg})(0.1 \text{ m})^2 (50 \text{ Trad/s})} = 9.5493 \text{ rad/s}.$

18.95 and 18.96



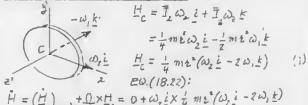
GIVEN: 10-02 DISK SPINS AT RATE W, = 750 pm PROBLEM 18.95:

FOR WZ = 6 ran/s FIND THE DYNAMIC REACTIONS AT A AND B.

PROBLEM 18,96:

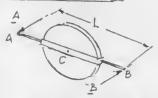
FIND MAX. ALLOWABLE W. IF DYNAMIC REACTIONS AT A AND B ARE NOT TO EXCEED 0,25 16 EACH

ANGULAR MOMENTUM PROUT C



 $\frac{\dot{H}}{\dot{H}} = (\frac{\dot{H}}{c})_{CXY'Z'} + \underbrace{\Omega} \times \underbrace{H}_{c} = 0 + \omega_{z} \underline{i} \times \frac{i}{4} mz^{2} (\omega_{z} \underline{i} - 2\omega_{i} \underline{k})$ $\frac{\dot{H}}{\dot{H}} = \frac{i}{2} mz^{2} \omega_{i}, \omega_{z} \underline{i} \qquad (2)$

EQUATIONS OF MOTION



 $\frac{\dot{H}}{dt} = \frac{1}{2} m t^3 \omega_1 \omega_2 \frac{\dot{s}}{\dot{s}}$

 $\sum M_A = \sum (M_A)_{eff} : BL = \frac{1}{L} \approx L^2 \omega_1 \omega_2 \qquad A = B = \frac{m_E \omega_1 \omega_2}{2L}$ (3)

PROBLEM 18.95
LETTING $m = \frac{W}{3} = \frac{(10/16)16}{32.2 \, \text{ff/s}} = 0.01941 \, \text{lb.s}/\text{ft, } \xi = \frac{1}{6} \, \text{ft.}, L = \frac{2}{3} \, \text{ft.}$ $\omega_1 = 750 \, \text{rpin} = 25 \, \text{ft.} \, \text{mal/s}, \, \omega_2 = 6 \, \text{rod/s} \quad \text{IN Eq. (3)};$ $A = B = \frac{(0.0194115.3/\text{ft}) \frac{1}{2} \, \text{ft.} \frac{1}{$

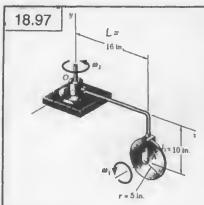
A = (0,1906/6)k; B=-(0.1906/6)k

PROBLEM 18.96

LETTING A = B = 0.25 lb, m = 0.01941 lb st/ft, $\xi = \frac{1}{6} st$, $L = \frac{2}{3} st$ AND $\omega_1 = 750 rpm = 95 \pi rat/s$ IN EU. (3) AND SOWING POR ω_2 :

$$\Omega_2 = \frac{2(\frac{2}{3}ft)(0.2516)}{(0.0194116.5/4t)(\frac{1}{6}ft)^2(2517 \text{ rad/s})} = 7.872 \text{ rad/s}$$

$$\omega_2 = 7.87 \text{ rad/s}$$



DISK OF WEIGHT

W=BIL ROTHES OF

CONSTANT W=12 mal/s.

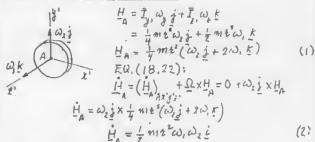
ARM OF RUTHTES HT

CONSTANT W2= 4 mal/s.

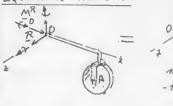
FIND:

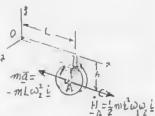
FINCE-COUNTY
SYSTEM REPRESENCE
DYNAMIC REACTION
AT SUPPORT O.

ANGULAR HOHENTUN ABOUT A



EQUATIONS OF MUTION





$$\Sigma F = \Sigma (F)_{eff} : R = -m L \omega_2^2 i$$
 (3)

 $\Sigma M = \Sigma (M_0)_{eff}$: $M^R = H_0 + (L_i - h_j) \times (-mL \omega_i^L i)$

 $=\frac{1}{2}m + \omega_1 \omega_2 i - m + L \omega_2 k$ (4)

WITH GIVEN DATA:

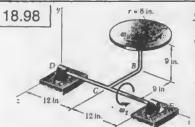
 $m = \frac{W}{g} = \frac{e \cdot b}{32.2 \, \text{M/s}} = 0.24845 \, \text{M-s} / \text{St}, \ L = \frac{4}{3} \text{ft}, \ h = \frac{5}{6} \text{ft}$ $\omega_1 = 12 \, \text{rad/s}, \ \omega_2 = 4 \, \text{rad/s}, \ \lambda = \frac{5}{12} \, \text{ft}.$

EO.(3): $R = -(0.24895 16.5)/5 + \sqrt{\frac{4}{3}} fi \sqrt{4 rad/5}$; = -(5.300 16)i

EQ.(4): MR = 1 (0.24845/b.s/4+)(5 ft) (12 rad/s)(4 rad/s) - - (0.24845/b.s/4+)(5 ft) (2 ft) (4 rad/s) E

=(1.0352 16.5t) 1 -(4.417 16.ft) k

FORCE-COUPLE AT 0: $R = -(5.30 \text{ lb}) \frac{i}{i} ; M_0^R = (1.035 \text{ lb.5t}) \frac{i}{i} - (4.42 \text{ lb.5t}) \frac{k}{k}$



GIVEN

DISK OF WEIGHT

W=616 ROTATES AT

CONSTAIT OF 16 FOR 1/8.

ARM ABOLIS WELDED

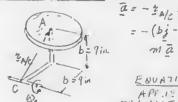
TO SHAFT DEE WHICH

ROTATES AT CONSTANT

W=811/4.

FIND: DYNAMIC REACTIONS AT DAILDE.

EFFECTIVE FORCE MIC



 $\bar{\alpha} = -\frac{y}{A/c} \, \hat{\mathcal{N}}_{i}^{t}$ $= -\left(b_{i} - b_{i} + b_{i}\right) \hat{\mathcal{N}}_{i}^{t}$ $= mb \, \hat{\mathcal{N}}_{i}^{t} \left(-\frac{1}{2} + \frac{k}{2}\right) \quad (3)$

ENVATIONS OF MODILINA
APPLIED PORTES ARE
ENVIOLEM TO EFFECTURE FORCES



EMD = Z(MD) Lix(E, i+ E, k) = H, +(Li) x ni?

RECHLING EUS (2) ACO(3):

Lix(E, i+ E, k) = \frac{1}{2}m^2\omega_1\omega_2\k + \frac{1}{2}ix mb\omega_2^2(-i+k)

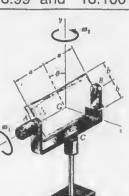
LE K-LE i = \frac{1}{2}m^2\omega_1\omega_2\k - \frac{1}{2}mbl\omega_2\k - \fra

 $E_{y} = \frac{1}{2} m \left[(\xi^{2}/L) \omega_{1} \omega_{2} - b \omega_{2}^{z} \right] \qquad E_{z} = \frac{1}{2} m b \omega_{z}^{z}$ $WITH 61/L 1 AnTA: m = \frac{6 b}{32.2 \text{ ft/s}}, = 0.18634 \text{ lb s}/\text{St}, \ \xi = \frac{2}{3} \text{ ft/s}$ $L = 2 \text{ ft/s}, b = 0.75 \text{ ft/s}, \ \omega_{1} = 16 \text{ rad/s}, \ \omega_{2} = 8 \text{ rad/s}$

ZF=ZFis; D+E=mā RECHLLING (3) AND-GIVEN DATAI D=Mā-E=mbw; (-j+k)-E-= (0.18634)(0.75)(8)*(-j+k)+(1.822 b)j-(4.472 lb)k = (1.821-8.944)j+(8.944-4.472)k

D=-(7.1216)j+(4.4716)k

18.99 and 18,100



GIVEN:

ADVERTISING PANEL m = 48 kg, 2a = 2.4m, 2b=1.6m. MOTOR AT A KEERS PANEL ROTATING ABOUT AB AT CONSTANT RATE WI. MOTOR AT C KEEPS FRAME RUTATING AT CONSTANT W2 . PANEL COMPLETES FUL. REVOLUTION IN 65. PRAME COMPLETES FULL REVOLUTION IN 125. PRABLEM 18.99: EXPRESS DYNAMIC REACTION AT DAS FUNCTION OF B.

PROBLEM 18, 100: SHOW THAT (A) DYNAMIC REACTION AT D IS INDEPENDENT OF LENGTH 2a.

(b) AT ARM INSTANT MI/MI = WE/ZWI, WHEFE M. AND MZ HARE THE MASSIFICATES OF THE COUPLES EXERTED BY THE MUTHIS AT A AND C, LE PECTIVELY.

USING AXES GZYZ WITH Z' PERPENLICULAR TO PANEL: Wz = Wz sin B, Wz = W, ws 0, Wz = W, $H_{G} = I_{\chi} \omega_{\chi} \omega + I_{\chi} \omega_{\chi} \omega + I_{\chi} \omega_{\chi} k$ $H = \frac{1}{3}m(a^2 + b^2)\omega_1 \sin \theta i^2 +$

+ 1 ma w cos + 1 + 5 mbw, k TO THE ORIGINAL FRAME GX42, WE NOTE:

i'= costi + s.noj i' = -sinti + cosoj

SUBSTITUTE IN (1)!

Ha = = = m(a2+b3) W, sind (cost i + sind 2)+ + 1 ma w coso (- sinti + cosoj) + 5 mbilik

H = = = m [b w, sin & coso i + (a + b sin o) w; + bw, k]

THE FIRST TERM IS OFTHINED BY DIFFERTURINTY IS E. WITH RESTER T T. ASSUMING FRAME GRYZ TO BE FIXED!

= = = = = = [6 w2 (wso-sinto) oi + x 6 w, sino wse 6,] DESTEVING THAT \$ = W, MAD SUBSTITUTION (2):

H= = mbw, w, (cox'+ -sin'b) i+ & sin + costig] +

Wilx + mb (w, sin & coli + w, k)

H = {mbuju, [(050 - sin') i + 2 sin decoso } +

I mit (WINZI - Wisin 0 cost K). $\underline{H}_{G} = \frac{1}{3} mb \left(2\omega_{1} \omega_{2} \cos^{2} \theta + \omega_{1} \omega_{2} \sin 2\theta - \frac{1}{2} \omega_{2}^{2} \sin 2\theta k \right)$ (4) LEATTER AL 2 MUST BE EQUAL TO WHILE HE

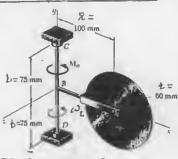
PROPLEM 18.79, WITH GIVEN CATA: 1 = H = 3(48 kg)(0.8 m)2 [2(2)/2) kcc Di+(21)/1) sinzej-2(2) sinzej

M = (11.23 Nom) cosoi + (5.61 Nom) sin 20 j - (1.404 Nom) sin 20 k

PROBLEM 18,100 (a) EU. (4) DOES NOT CONTAIN Q.

(b) FROM(4): M= = 1 216 W2 sin 20, M2= 5 mb WW2 sin 20 M,/Hz = Wz /2W,

18.101 and 18.102



PROBLEM 18.1015 GIVEN:

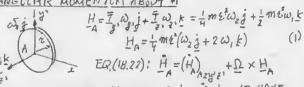
S. Ky DISK SPINS AT CONSTANT W, = 60 rad/s: ARM AB AND SHAFT ARE AT REST WHEN M = (0,40 N.m) } 15 APPLIED FOR 25.

DYNAMIC REACTIONS AT C AND D AFTER My IS REMOVED.

PROBLEM 18, 102

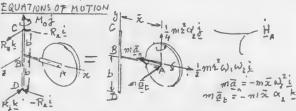
GIVEN: 3-Kg DISK SPINS AT CONSTANT W = 60 md/s. ARM AB AND SHAFT ARE AT REST WHEN M IS MEPLITE FOR 3 5, WITH ANG. VELOCITY OF SHAFT REACHING 18 rol/s. FIND: (a) Mo, (b) DYNAMICREACTIONS AT C AND D AFTER M IS REMOVED.

ANGULIAR MOMENTUM ABOUT A



SINCE DISK HAS AN ANG. ACCEL. OF = 021, WE HAVE ALSO, Q = W28 = + mE x, j + wz j x + m2 (W, j + 2WE)

(2)



FROM SYMMETRY AND INSPECTION OF EFFECTIVE FORES, WE FIND THAT THE COMPONENTS OF THE REACTIONS AT CAND DARE EQUAL IN MAGNITUDE AND DIRECTED AS SHOWN. $ZM_{3} = \sum (M_{3})_{eff}: M_{0} = \frac{1}{4}mz^{2}\alpha_{1} + \sum (m\bar{\lambda}\alpha_{2}) = m(\frac{1}{4}z^{2}\bar{n}z^{2})\alpha_{2}$ $M_{1} = (3kg)(\frac{1}{4}(0.06m)^{2} + (0.1m))\alpha_{2} \qquad M_{2} = 0.0327\alpha_{2} \qquad (3kg)(\frac{1}{4}(0.06m)^{2} + (0.1m))\alpha_{2} \qquad M_{3} = 0.0327\alpha_{2} \qquad (3kg)(\frac{1}{4}(0.06m)^{2} + (0.1m))\alpha_{3} \qquad M_{3} = 0.0327\alpha_{2} \qquad (3kg)(\frac{1}{4}(0.06m)^{2} + (0.1m)^{2})\alpha_{3} \qquad M_{3} = 0.0327\alpha_{3} \qquad (3kg)(\frac{1}{4}(0.06m)^{2} + (0.1m)^{2})\alpha_{3} \qquad (3kg)($ (3)ΣFx = Σ(Fx) ess: 2R = mxw2 = (3kg)(0.1 m) ω2, R=0.15 ω2 ΣM= Σ(M) (: 2bR, = 1 m & ω, ω,

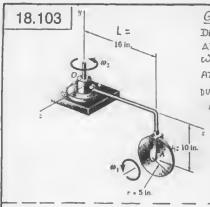
 $R_2 = (mz^2/46)\omega_1 \tilde{\omega}_2 = \frac{(3 \text{ kg/c.06 m)}(\text{horad/s})\omega_2}{4(0.075 \text{ m})}$ Rz=2.16W, (5)

PRUBLEM 18.101 $\frac{1}{LET}\frac{N}{n} = 0.40 \, \text{N·m} \, \text{IN (3)} : \alpha_{g} = \frac{0.40}{4.0327} = 12.232 \, \text{rad/s}^{6}$ FOR t=25: W= 02t=(12.232 rad/5 X25) = 24.464 rad/s EGS. (4) ANI(5): Rx = 0.15 W = 89.8 N; Rx = 2.16 W, = 52.8 N C=-(89,8N)i+(52.8N)k; D=-(89.8N)i-(52,8N)k

PROBLEM 18.102

(), = 0, t: 18 rod/s = 0, (35), $\alpha_2 = 6 \text{ rad/s}^2$ (a) E0.(3): M=0.0327(6)=0.1967 Hm M=(0.1962 Hm) } (b) Ec. (4): R = 0.15 (18 rod/s)2 = 48.6 N

ER(5)1R, = 2,16(18 rod/s) = 38,88 N C=-(48,64):+(38.9 H)k; D=-(48,64):-(38.9 N)k



GIVEN:

DISK OF WEIGHT W= 8 lb.

AT INSTANT SHOWN $\omega_1 = 12 \text{ rad/s} \text{ AND DERRASES}$ AT RATE OF 4 rad/s

DUE TO BEARING FRICTION,

ARM OA RUTHTES AT

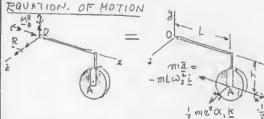
CONSTIANT $\omega_2 = 4 \text{ rad/s}$ FIND:

FORCE-COURE SYSTEM

FIND: FORCE-COURLESYSTEM REPRESENTING DYNAMIC REACTION AT ST PPORT 0.

ANTOLOGY ANT OF APART A

Thus: $\frac{\dot{H}_{A}}{\dot{H}_{A}} = \frac{1}{2} m \dot{z}^{2} X_{1} \dot{k} + \omega_{2} \dot{\underline{j}} \times \frac{1}{4} m \dot{z}^{2} (\omega_{2} \dot{\underline{j}} + 2\omega_{1} \dot{k})$ $\dot{H}_{A} = \frac{1}{2} m \dot{z}^{2} (\omega_{1} \omega_{2} \dot{\underline{c}} + \alpha_{1} \dot{k})$ (2)



 $\Sigma \vec{F} = \Sigma(\vec{F})_{ett}: \quad \vec{R} = -m L \omega_z^{\epsilon} \underline{i}$ $\Sigma M = \Sigma(M_0)_{ett}: \quad M_0^R = \frac{1}{2} M_1 + (L_1 - h_2^{\epsilon}) \times (-m L \omega_z^{\epsilon} \underline{i})$ $M_0^R = \frac{1}{2} m z^{\epsilon} \omega_1 \omega_2 \underline{i} + \frac{1}{2} m z^{\epsilon} \alpha_1 k - u_1 h L \omega_z^{\epsilon} k$ $M_0^R = \frac{1}{2} m z^{\epsilon} \omega_1 \omega_2 \underline{i} + m (\frac{1}{2} z^{\epsilon} \alpha_1 - h L \omega_z^{\epsilon}) \underline{K}$ (4)

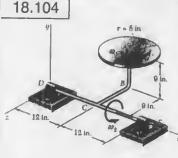
WITH GIVEN DATA: $m = \frac{W}{3} = \frac{0/b}{32.2 \text{ ft/s}} = 0.24845 \text{ lb.s}/\text{st.} L = \frac{4}{3} \text{ft.} h = \frac{5}{6} \text{ft.} \ \xi = \frac{5}{12} \text{ft}$

 $\omega_1 = 12 \text{ rod/s}, \ \alpha_1 = -4 \text{ rad/s}^2, \ \omega_2 = 4 \text{ rad/s}$

EQ.(3): $R = -(0.24845 /b.s^2/ft)(\frac{4}{3} ft)(4 rad k)^2 i$ = -(5.300 lb) i

 $E(0, 4) : M_{0}^{R} = \frac{1}{L}(0.24845/b.s/st)(\frac{5}{12}st)^{6}(12 \text{ rad/s})(4 \text{ rad/s}) + \\
+(0.24845/b.s^{3}/st)[\frac{1}{L}(\frac{5}{12}st)^{2}(-4 \text{ rad/s}) - (\frac{5}{2}st)(\frac{4}{3}st)(4 \text{ rad/s})^{2}] \frac{1}{L} \\
= (1.0352/b.st) \frac{1}{L} - (0.24845/0.34722 + 17.778) \frac{1}{L} \\
M_{0}^{R} = (1.0252/b.st)^{2} - (4.503/b.st) \frac{1}{L}$

FORCE-COUPLE AT 0: R=-(5,30/b)i; M_0 = (1.035/b)i-(4.50/b)fk



FIND:
(a) COUPLE APPLIED
TO SHAFT
(b) DYNAMIC REACTIONS
AT D AND E.

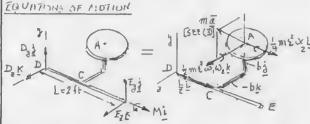
ANGULAR MOMENTUM ATHOR -

WHEN - THE FILLT TERM IS OF THIVE BY DIFFER PHILIPTINS HA ASSUMING THE PRAME DZ'g'E' TO AR TIKED!

 $(\stackrel{\leftarrow}{H}_{A})_{Ax'y'z} = \frac{1}{7} mz^2 \omega_z \stackrel{\leftarrow}{L} = \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L} = \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L} + \omega_z \stackrel{\leftarrow}{L} \times \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L} + \omega_z \stackrel{\leftarrow}{L} \times \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L}$ $(\stackrel{\leftarrow}{H}_{A})_{Ax'y'z} = \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L} + \omega_z \stackrel{\leftarrow}{L} \times \frac{1}{7} mz^2 \alpha_z \stackrel{\leftarrow}{L} = \frac{1}{7}$

 $H_{A} = \frac{1}{\mu} m_{2}^{2} \alpha_{2} + \frac{1}{2} m_{2}^{2} \omega_{2} \kappa \qquad (2)$

ACCELERATION OF MASS CENTER $\bar{a} = \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$ $= \frac{\alpha_z \times \alpha_{A/c}}{2} - \omega_z^2 \frac{\alpha_{A/c}}{2} + \omega_z^2 \frac{\alpha_{A/c}}{2}$



ΣMD= Σ(MD)++: Lix(E3+E2K)+Mi=(12i+bj-bK)xmb[(α,-ω²)j+(α,+ω²)K+HA LE3K-LE3j+Mi=1mbL[(α,-ω²)K-(α,+ω²)j+2mb°α, i+ -1mc²α, i+1mc²α, i+1mc²α, ω, k

EQUATING COUFF. OF UNIT VECTORS:

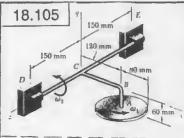
 $\sum_{i} M = m(2b^2 + \frac{1}{4}z^i) \alpha_2 \tag{4}$

 $E_{y} = \frac{1}{2} m \left[bL \left(\alpha_{z} - \omega_{z}^{2} \right) + \epsilon^{2} \omega_{1} \omega_{z} \right]$ $E_{y} = \frac{1}{2} m \left[b \left(\alpha_{z} - \omega_{z}^{4} \right) + \epsilon^{2} \omega_{1} \omega_{z} \right]$ (6)

 $\sum F_{y} = \sum (F_{y})_{e_{y}} : D_{y} + E_{y} = ma_{y} \qquad D_{y} = mb(\alpha_{2} - \omega_{2}^{*}) - E_{y}$ $D_{y} = \int_{1}^{1} m[b(\alpha_{2} - \omega_{2}^{*}) - \frac{1}{L}\omega_{1}\omega_{1}] \qquad (7)$

 $\sum F_{2} = \sum (F_{2})_{eff} : D_{2} + E_{\xi} = m \alpha_{\xi} \qquad D_{z} = m b(\alpha_{z} + \omega_{z}^{2}) - E_{z}$ $D_{z} = \frac{1}{2} m b(\alpha_{z} + \omega_{z}^{2})$

WITH GIVEN DATA: m = 6/32, z = 0.18634, L = 24F, $b = \frac{3}{4}4$, $z = \frac{2}{3}4F$, $\omega_1 = 16$, $\omega_2 = 8$, $\omega_2 = 6$ (a) M = 1.382 16.51 M = (1.382)6.4)! (b) $D_{y} = -6.70$ 16, $E_{y} = -1.403$ 16, $D_{z} = E_{z} = 4.89$ 16 $D = -(6.70)6)\frac{1}{2} + (4.89)6$ 16; $E_{z} = -(1.403)6$ 14, $E_{z} = -(1.403)6$



GIVEN:
2.5-kg DISK ROTATES WITH $\omega_1 = \omega_1 \frac{1}{2}, \alpha_2 = -(15 \text{ med/s}) \frac{1}{2}$ SHAFT DCE ROTATES WITH

SONCTANT $\omega_2 = (12 \text{ med/s}) \frac{1}{2}$ FIND:

DYNAMIC REACTIONS AT D
AND E WHEN WHEN BECREASED TO 50 rad/s.

ANGULAR MOMENTUM ABOUT A



HA = I , W, j + I W = I m 2 W, j + i m 2 W & EU, (18.22): H = (H) ALY 2, + I X HA

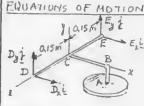
THE FIRST TERM IS THE RATE OF CHANGE OF HA WITH RESPECT TO THE
ROTATING FRAME AXY 2.

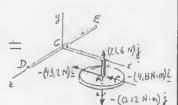
$$\begin{split} (\stackrel{\leftarrow}{H}_{A})_{A\lambda^{2}S^{1}2} &= \frac{1}{2} m z^{2} \mathring{\omega}_{1} \stackrel{\rightarrow}{I} = \frac{1}{2} m z^{2} \mathscr{V}_{1} \stackrel{\rightarrow}{J} ; \quad \text{ArLSO} : \Omega = \mathcal{W}_{2} \stackrel{\rightarrow}{K} \\ \text{THUS} : \stackrel{\rightarrow}{H}_{A} &= \frac{1}{2} m z^{2} \alpha_{1} \stackrel{\rightarrow}{J} + \mathcal{W}_{2} \stackrel{\rightarrow}{K} \times \left(\frac{1}{2} m z^{2} \mathring{\omega}_{1} \stackrel{\rightarrow}{J} + \frac{1}{2} n z^{2} \mathring{\omega}_{2} \stackrel{\rightarrow}{K} \right) \\ \stackrel{\rightarrow}{H}_{A} &= \frac{1}{2} m z^{2} \alpha_{1} \stackrel{\rightarrow}{J} - \frac{1}{2} n z^{2} \mathscr{\omega}_{1} \mathscr{\omega}_{2} \stackrel{\rightarrow}{L} = \frac{1}{2} m z^{2} \left(-\omega_{1} \mathring{\omega}_{2} \stackrel{\rightarrow}{L} + \alpha_{1} \stackrel{\rightarrow}{L} \right) \\ &= \frac{1}{2} (2.5 kg) (0.08 m)^{2} \left[- (50 radk) (12 radk) \stackrel{\rightarrow}{L} + (-15 radk)^{3} \right] \stackrel{\rightarrow}{H} = -(4.8 N \cdot m) \stackrel{\rightarrow}{L} - (0.120 N \cdot m) \stackrel{\rightarrow}{J} \end{split}$$

ACCELERATION OF MASS CENTER

UDING C AS THE FIXED ORIGIN, AND SINCE $\alpha_z = 0$: $\bar{\alpha} = -\frac{v}{A} \omega_z^2 = -\left[(0.12 \text{ m})i - (0.06 \text{ m})\right] \left(12 \text{ rad/s}\right)^4$ $\bar{\alpha} = -\left(17.28 \text{ m/s}^2\right)i + (8.64 \text{ m/s}^2)j$

THUS: mā = (2.5 kg) ā mā = - (43.2 N)i + (21.6 N)j (2)





 $\Sigma \stackrel{M}{=} \Sigma \stackrel{M}{=}_{D})_{eff}$: $-(0.3 \text{ m}) k \times (E_{\underline{i}} + E_{\underline{j}}) = -(4.8 \text{ N·n}) i - (0.12 \text{ N·m}) \frac{1}{2} + \frac{1}{2} A/D \times m\underline{a}$ $-0.3 E_{\underline{i}} + 0.3 E_{\underline{i}} = -4.8 i - 0.12 j +$

4(-0.15k+0.12i-0.06j)x(-43.2i+21.6j) -0.3Ezz+0.3Ezi=-48i-0.12j+6.48i+3.24i+2.54zk-2,54zk

-0.3E, j +0.3E, i = -1.56 i +6.36 j

ERNATING. THE CUEFF, OF THE UNIT VECTORS:

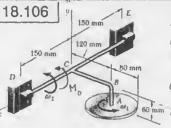
 $-0.3E_{g} = -0.36$ $E_{g} = -21.2 \text{ M}$ $0.3E_{g} = -1.56$ $E_{g} = -5.20 \text{ M}$

 $\Sigma F = \Sigma (F_1)_{ff}$: $D_2 - 21.2N = -43.2N$ $D_3 = -22.0N$

ZFg = \(\mathbb{E}(\mathbb{F}_g)_e\mathbb{H}: \mathbb{D}_g - 5.20N = 21.6N \\
\mathbb{D}_y = 26.8N \\

ANSWER D=-(21.0N) i+(26.8 N); E=-(21.2 N) i-(5.20 N) j

(ANSWER GIVEN WITH RESPECT TO RUTATING CRYP AXE)



GIVEN:

2.5-kg DISK ROTATE: WITH

CONSTANT W = (50 mdp) }.

AT INSTANT SHOWN, SHAFT

DCE ROTATES WITH

W=(12 rod/s):, \alpha = (8 m/k) k.

FIND:

(a) COUPLE M APPLIED

TO THE SHAFT.

(b) DYNAMIC REACTIONS AT D AND E

ANGULAR MUMENTHAN TENUT A

A Wik

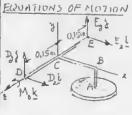
$$\begin{split} & \underbrace{H}_{A} = \underbrace{\bar{I}_{g}}_{,\omega_{g}}, \underbrace{\dot{\chi}}_{1} + \underbrace{\bar{I}_{g}}_{2}, \underbrace{\dot{\omega}_{g}}_{2}, \underbrace{\dot{k}}_{2} = \underbrace{\bar{I}_{g}}_{2}, \underbrace{\dot{\omega}_{g}}_{2}, \underbrace{\dot{I}_{g}}_{2} + \underbrace{\dot{I}_{g}}_{2}, \underbrace{\dot{\mu}_{g}}_{2} + \underbrace{\dot{\Omega}}_{2} \times \underbrace{\dot{H}_{g}}_{2} \\ & \underbrace{\bar{H}_{g}}_{2} = \underbrace{(\underbrace{\dot{H}_{g}}_{2})_{A \times g}}_{2}, \underbrace{\dot{I}_{g}}_{2} + \underbrace{\dot{\Omega}}_{2} \times \underbrace{\dot{H}_{g}}_{2} \\ & \underbrace{\bar{H}_{g}}_{2} = \underbrace{(\underbrace{\dot{H}_{g}}_{2})_{A \times g}}_{2}, \underbrace{\dot{H}_{g}}_{2} = \underbrace{(\underbrace{\dot{H}_{g}}_{2})_{A \times g}}_{2}, \underbrace{\dot{$$

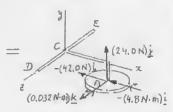
 $Az = \frac{1}{4} \frac{1}{4} \text{ which } Rv = 175 \text{ At } \Omega = \omega_z \text{ k}.$ $(\stackrel{\leftarrow}{H}_A)_{HX'Y'Z'} = \frac{1}{4} mz^2 \omega_z \text{ k} = \frac{1}{4} mz^2 \omega_z \text{ k}$ $THUS: \stackrel{\leftarrow}{H}_A = \frac{1}{4} mz^2 \omega_z \text{ k} + \omega_z \text{ k} \times (\frac{1}{2} mz^2 \omega_1) \frac{1}{4} + \frac{1}{4} \sin z \omega_z \text{ k})$ $\stackrel{\leftarrow}{H}_A = \frac{1}{4} mz^2 \omega_z \text{ k} - \frac{1}{2} mz^2 \omega_1 \omega_z \text{ i} = \frac{1}{4} mz^4 (-2\omega_1 \omega_2 \text{ i} + \omega_z \text{ k})$ $= \frac{1}{4} (2.5 \text{ kg}) (0.08 \text{ m})^2 [-2 (50 \text{ rod/s}) (12 \text{ rad/s}) \frac{1}{6} + (8 \text{ rad/s}) \frac{1}{8})$ $\stackrel{\leftarrow}{H}_A = -(4.8 \text{ N.m}) \frac{1}{6} + (0.032 \text{ N.m}) \frac{1}{8}$ (1)

ACCELERATION OF MASS CENTER USING C AS THE FIXED ARIGIN.

 $\bar{\underline{a}} = \underset{2}{\times} \underset{2}{\times} \underset{A/C}{} - \underset{A/C}{} \underset{2}{\times} \underset{2}{\times} = (8 \text{ rad/s}) \underset{1}{\times} \underset{2}{\times} (0.12 \text{ m}) \underset{1}{\downarrow} - (0.06 \text{ m}) \underset{2}{\downarrow}] - (0.06 \text{ m}) \underset{3}{\downarrow}] (12 \text{ rad/s})^{*} \\
= (0.96 \text{ m/s}) \underset{1}{\downarrow} + (0.48 \text{ m/s}) \underset{1}{\downarrow} - (17.28 \text{ m/s}) \underset{1}{\downarrow} + (8.64 \text{ m/s}) \underset{2}{\downarrow} \\
\bar{\underline{a}} = - (16.8 \text{ m/s})^{2} + (9.6 \text{ m/s})^{2} \underset{1}{\downarrow} + (9.6 \text{ m/s})^{2} \underset{1}{\downarrow}$

THUS: ma = (2,5 kg) & ma = - (42,0N) i + (24.0N)j (2)





$$\begin{split} & \sum M_{D} = \sum (M_{D})_{eff} : \\ & - (0.5 \text{ m})_{e} \times (E_{\lambda} L + E_{\lambda} \frac{i}{2})_{A} M_{0} \times = -(4.8 \text{ N·m})_{e} + (0.032 \text{ N·m})_{e} + \frac{2}{2} \times 70\overline{2} \\ & - 0.3 E_{\lambda} \frac{i}{2} + 0.9 E_{\lambda} \frac{i}{2} + M_{0} \frac{K}{2} = -4.8 \frac{i}{2} + 0.032 \frac{K}{2} + \\ & + (-0.15 \frac{K}{2} + 0.06 \frac{i}{2})_{e} \times (-42 \frac{i}{2} + 2.9 \frac{i}{2}) \end{split}$$

-0.3E, j+0.3E, j+Mok =-4.8 j+0.032k+6.30j+3.60i 12.80K-2.52 k

(a) (b) M_D = 0.032+2.88-2.52 = 0.392 N·111 M - (0.39

M=(0,392 N·m) K

(b) $\{1, 0.3\}$ = 6.30 = 2.10 N(c) $\{0.3\}$ = -4.8 + 3.6 = -1.20 = 2.400 N = 2.7 = 2.7 = 2.10 N = -420 N = -21.0 N= 2.7 = 2.7 = 2.0 N

D=-(21.0 N) i+(28.0 N) j; E=-(21.0 N) i-(4.00 N) j (ANS WER GIVEN WITH RESPECT TO ROTATIONS CZYZ AXES) 18.107 and 18.108

GIVEN:

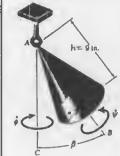
50 rpm 160 mm 8 80 mm

SOLID SPHERE WELDED TO END OF RUD AB OF NEOLIGIBLE MADS SUPPORTED BY BALL AND SUCKET AT IA, SPHERE PRECESSES AT CONSTANT RATE OF 50 MPM AS SHOWN.

DF 50 TPM AS SHOWN.
PROBLEM 18.107:
PIND RATE OF SPINY, KNOWING
THAT B= 25°.

PROBLEM 18. IDB: FIND B, KNOWING THAT RATE OF SPIN IS Y = BOO CPM. 18.109 and 18.110

CONE SUPPORTED BY BALL AND SUCKET AT A.



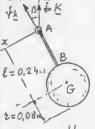
PROBLEM 18.109:

GIVEN: PRECESSES AS SHOWN AT CONSTANT

RATE OF 40 CPM WITH B= 40° FIND! RATE OF SPIN Y PROBLEM 18.11D

GIVEN: \$ = 3000 rpm, \$ = 60

TWO POSSIBLE VALUES OF \$



ANGULAR VELOCITIES:

SPHERE: $\omega = \phi K + \psi k$ $\omega = -\phi \sin \beta \dot{\iota} + (\sqrt{1} + \phi \omega s \beta) \dot{k}$ FRAME AXY: $\Omega = \phi K$ $\Omega = -\phi \sin \beta \dot{\iota} + \phi \omega s \beta \dot{k}$ ANGULAR MOMENIUM ABOUT A

H = I ω $\dot{\iota} + I \omega_{\dot{\iota}} \dot{k}$

H, = - m(=22+8") + singi+ = m2 (+ 1 + ws f)k

SINCE A 15 FIXED, WE USE = 0. (18.28): $\sum_{A}^{M} = (H_{A})_{AXY} + \Pi_{X}H_{A} = 0 + (-\frac{1}{2}s_{1}^{2}+c_{1}^{2}) + \frac{1}{2}s_{1}^{2}s_{1}^{2} + \frac{1}{2}s_{2}^{2}s_{1}^{2} + \frac{1}{2}s_{2}^{2}s_{1}^{$

BUT $\sum M_A = -lf \times (-mgK) = -mglsinfj$ (2)

 $n \sin \beta \left[\frac{2}{5} \epsilon^{2} (\dot{\gamma} + \dot{\phi} \cos \beta) - (\frac{2}{5} \epsilon^{2} + \dot{\epsilon}^{2}) \dot{\phi} \cos \beta \right] = -m \left[\sin \beta \right]$ $\frac{2}{5} \epsilon^{2} (\dot{\gamma} + \dot{\phi} \cos \beta) = \left(\frac{2}{5} \epsilon^{2} + \dot{\epsilon}^{2}\right) \dot{\phi} \cos \beta - \frac{2}{5} \epsilon^{2} \cos \beta$ (3)

GIVEN DATA: (NOTE THAT & IS MEGATIVE)
2=0.08 m, C=0.24 m, g= 9.81 m/s, d=-50 rpm = -5.236 rod/s
2=6 ×10⁻³ (\$\frac{1}{2} - 5.236 cos \beta\$) = 60.16 ×10⁻³ (-5.236 cos \beta\$) + 147.7×10⁻³

Ŷ=-117,81 cocβ + 175.65 (4)

PROBLEM 16.117

11.10 p=55, 50 (4) 115105

Y=-117.81 (4) 25"+ 175.65 = +68,875 rod/s
= 1657.7 rpm

Y=658 rpm

PROBLEM 18.198

B= 38.8°

PA A BANG

ANGULAR VELOCITIES

CONE: $\omega = \phi K + \dot{r} k$ $\omega = -\dot{\phi} \sin \beta l + (\ddot{V} + \dot{\phi} \cos \beta) \dot{k}$ FRAME AZY: $\Omega = -\dot{\phi} K$ $\Omega = -\dot{\phi} \sin \beta \dot{l} + \dot{\phi} \cos \beta \dot{k}$ ANGULAR MOMENTUM ABOUT A $H = I \omega_{\dot{l}} \dot{i} + I_{\dot{l}} \omega_{\dot{l}} \dot{k}$

$$\begin{split} & \underbrace{H}_{A} = -\frac{3}{5}m(\frac{k^{2}}{4} + h^{2}) + sin\beta \dot{i} + \frac{3}{10}mz^{*}(v + 4\cos\beta) \dot{k} \\ & \leq INCE A IS FIVED, & USE Ea. (18.28): \\ & \underbrace{Em}_{A} = (\underbrace{H}_{A})_{A,y} + \underbrace{\Omega}_{A} \times H_{A} = 0 + (-4sin\beta \dot{i} + 4\cos\beta \dot{k}) \times H_{A} \\ & = (-4sin\beta \dot{i} + 4\cos\beta \dot{k}) \times [-\frac{3}{3}m(\frac{k^{2}}{4} + h^{2}) + sin\beta \dot{i} + \frac{1}{10}mz^{*}(v + 4\cos\beta) \dot{k}] \\ & = \frac{3}{3}m + sin\beta \left[\frac{1}{2}z^{*}(v + 4\cos\beta) - (\frac{2k^{2}}{4} + h^{2}) + \cos\beta\right] \dot{j}. \end{split}$$

BUT EM = - 2 hk x (-mg K) = - 2 mg h sin p] (2) EQUATING (1) AND (2):

 $\frac{3}{5} m \dot{\phi} \sin \beta \left[\frac{1}{L} e^{2} (\dot{\psi} + \dot{\phi} \cos \beta) - (\frac{4}{5} + \dot{h}^{2}) \dot{\phi} \cos \beta \right] = -\frac{3}{4} mgh \sin \beta$ $\frac{1}{L} e^{2} (\dot{\psi} + \dot{\phi} \cos \beta) - (\frac{4}{5} + \dot{h}^{2}) \dot{\phi} \cos \beta = -\frac{5}{4} \frac{gh}{4}$ $\frac{1}{L} e^{2} \dot{\psi} - (\dot{h}^{2} - \frac{2^{2}}{4}) \dot{\phi} \cos \beta = -\frac{5}{4} \frac{gh}{4}$

WITH $4 = \frac{1}{4}$ ft, $h = \frac{2}{3}$ ft, g = 32.2 ft/s; ANDMULTIPLYING BY 32, $\psi - 17.5$ $\phi \cos \beta = -966/\phi$ (3)

PROBLEM 18,109 LETTING $\psi = -40 \text{ rpm} = -4.1888 \text{ rad/s}, \ p = 40^{\circ} \text{ IN (3)}, \ \psi = -17.5(-4.1888)\cos 40^{\circ} = -966/(-4.1888)$ $\dot{\psi} = -56.154 + 230.616 = 174.46 \text{ rad/s} = 1666.0 \text{ rpm}$ $\dot{\psi} = 1666 \text{ rpm}$

PRUBLEM 18,110

LETTING $\sqrt{-3000}$ rpul = 314.16 rad/s, $\beta = 60$ in (3), 314.16 - 17.5, $\dot{\phi}$ cas $60^{\circ} = -966/\dot{\phi}$ 1.75 $\dot{\phi}^{\circ} = 314.16 \dot{\phi} - 966 = 0$ $\dot{\phi}^{\circ} = 35.904 \dot{\phi} - 110.4 = 0$ $\dot{\phi} = \frac{1}{2}(35.904 \pm \sqrt{(35.904)^2 + 4(110.4)}) = \frac{1}{2}(35.904 \pm 41.602)$ $\dot{\phi} = +38.753$ rad/s = +370 rpm $\dot{\phi} = (370 \text{ rpm}) \text{ K}$ (Sense opposite To sense shown)

OR $\dot{\phi} = -2.849$ rad/s = -27.2 rpm $\dot{\phi} = -(27.2$ rpm) \dot{K} (SAME SENSE AS SHOWN)

18.111 and 18.112 TOP SUPPORTED AT FIXED POINT O.



PROBLEM 18.111: GIVEN:

m = 85g, k, = 21mm, k, = 45mm C = 37.5 mm. 8 = 30°, RATE OF SPIN ABOUT & AXIS =

= 1800 FPM.

TWO FOSSIBLE RATES & OF STEADY PRECESSION.

PRUBLEM 18.112

GIVEN: I, = I, I = I, W, = RECTANGULAR COMPONENT CPW ALUNG & AXIS

(a) SHOW THAT (IW= - I' + cos U) + = WC (b) SIFON THAT I + & WC IF +>> +

(C) FIND PERCENT EFFOR WHEN EXPRESSION UNDER b IS USED TO APPROXIMATE THE SLOWER & OF PROB. 18.111.

WE RECALL FROM PAGE 1150 THE FOLLOWING ERS. W=- +51001+W. K (19.40) Ha = - I'fsine + I w, k

= - + sinoi + + costk (18.42)

SINCE O IS A FIXED POPIT, WE USE ER (18.28): $\Sigma \overline{W}^{0} = (\overline{H}^{0})^{0 \times 43} + \overline{U} \times \overline{H}^{0} = O + \overline{U} \times H^{0}$

= (- \$\psin\text{vi} + \psin\text{vi} + \psin\text{vi} + \psin\text{vi} + \psin\text{vi} + \psin\text{vi} + \psin\text{vi} = (-\psin\text{vi}) \text{vi} = (IW, \$51.18 - I \$ cost 51118) }

= (Iw2 - 1 : 4 : 10) + sint ;

WHERE & IS I PLANE OZE AND GUITS AWAY

(2) BUT IMp = ckx (-WK) = We sindj

EQUATING EUS. (1) AND (2):

 $\langle (3) \rangle$ (IW, - I + cost) + = WC

PROF LEH 18,111

SINCE I=mk1, I'=mk1, W=mg, EG (3) YIELDS

(K, W - K + ws 8) + = gc

WHERE W, = Y + + cos 0 WITH GIVEN DATH AND V = 1800 pm = 60 Trod/s:

[(0.021)2(607+ + 50>30)-(0.045)++030)+=9.81(0.0375)| DIVIDE (2) BY (3): ran B= = $[(0.045)^{2},(0.021)^{3}] \cos 30^{3}\phi^{2} - (0.021)^{3}(0.074) + 9.81(0.0375) = 0$

\$ - 60.597 \$ + 268.17 = 0

SOLVING: \$ = 30.249 ± 25.492

\$ = 55.79 rods AND \$ = 4.807 mills

ANSWEF: 533 rpm AND 45.7 rpm

PROBLEM 18.112

(a) SEE DERIVATION OF Ed. (3) ABOVE

(b) FON +>> +; W, ≈ Y, AND EG. (3) RECVCES TO (IY- I' + asy) + = WC

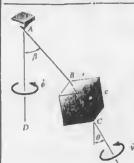
AND, WITH Y >> 4, TO

(Q.E.D.) IT+ = WC

(c) WITH DATE OF PROB. 18.11, ABOVE EQUATION VII-13 SOLVING (5) FOR C + V, $\dot{\varphi} = \frac{VC}{TV} = \frac{481}{MK^2 \dot{\psi}} = \frac{9.81}{(0.0375)} = 4.45570 \dot{\psi} = 3\sqrt{3} g \left(\frac{\tan \beta}{\tan \beta} - 1\right) = 3\sqrt{3} \left(\frac{9.81}{\tan \beta}\right) \left(\frac{\tan \beta}{\tan \beta} - 1\right) = 3\sqrt{3} \left(\frac{1}{2}\right) \left$ \$ = \frac{Wc}{IV} = \frac{Mgc}{Mk_2V} = \frac{9.81(0.0375)}{(0.021)^2(60)\tauralls)} = 4.455\tauralls = 42.2 E rpin

% ERRIN = 100 42.26-45.40 = -7.9 % 1 18.113 and 18.114

SOLIN C' PE ATTACHEDTO CURD AB



PROBLEM 18.113:

c = 80 mm, B = 30°

 $\psi = 40 \text{ rad/s}, \ \phi = 5 \text{ rad/s}.$ FIND: 8

PROBLEM 18.1141

GIVEN:

c = 120 mm, AB = 240 mm 8 = 25°, B = 40°

(a) Y, (b) \$

WE RECALL FROM SEC. 9,17 THAT, SINCE THE 3 PRINCIPI. MOMENTS OF INTERTIA OF A CUES ARE ERUAL ITS MOMENT OF HERETIA ABOUT AMY LINE THEONGY G IS ALSO 7= - THE USIN: GZPZ ATZS WIT- Z ALONG CB, X IN ABD PLAIR AND & I ABD AND FORMTING ANAY, WE HAVE

CUBE: W = + Sindi+ (Y+ + ws 0) } H = + mc [dsindi+(+++ cos) k]

FRAME GEHT:

1 = \$sindi+ + coso k

EQ.(18.22):

He = (He)exAF+ TX FR

H = 0+(+sin8i + \$ cos8 k) X Emc [dsindi+("++ costs)]

H = -mc + sino [-(V++ wso) + + cos + j] H = - Inc & Tsins j (1)

ERVATIONS OF MOTION 1 13 C COSO 13 csing mg

EF = E(F) = 1 HORIZ. COMPLI Tsing = ma VERTICAL CUMP: Toos A -mg = D Trosp=mg (3)

(4)

a=qtanp +5 ZM = E (MB) = 5: -- mg 13c sind = 6 mc + 15ind - (mg tan s) 13c coso

DNIDE BY mg 13c coso AND SOLVE FOR tand:

tan 0 = tor B 1+ (c + 1/3 1/3 g)

PROBLEM 18.113 LETTINS B=30, C=0.08m, += Stad/s, Y=40 radk, g=9.81m/s IN (5): tan B = 0.43942 $\theta = 23.7^{\circ}$

PROBLEM 18, 114 $\bar{a} = \bar{z} + \beta$ RECALLING (4): $\dot{\phi} = \frac{\bar{a}}{\bar{z}} = \frac{g \tan \beta}{(a \beta) \sin \beta} + \frac{g \sin \beta}{(a \beta) \sin \beta}$ LETTING 13=40", 0=25", AB=0.24 m, C=0.12 m, g=9.81 m/32:

P=6.4447 rad/s (b) = 6.44 rad/s

\$\frac{40.752}{(0.12)(6.4447)} = 52.694 rad/s V= 52,7 rad/s (a)

SOLID SPHERE ATTACHED 18.115 and 18.116 TO CORD AB. PROBLEM 18, 115: GIVEN: c = 31n., B=40, \$= 6 rad/s ANGLE B, KNOWING THAT (a) \(=0, (b) \(=50 \) rad/s, (c) \(=-50 \) rad/s. PROBLEM 18.116: GIVEN: C=3in., AB=15in., 0=20, B= 35° FIND: (a) V, (b) \$ USING GZYZ AXES WITH & ALONG CB, X IN ABD PLANE AND & I ABD AND POINTING AWAY: SPHERE: W= + sin 0 i+ (++ + coso) A H===mc[+sinoi+(+++cost)k] FRAME GZYZ: 1 = \$ sinti + \$ cost k EQ.(18,22): HG = (HG)Gzyz + QXHG (AB) sin B c sin B = O+ (& sindi+ + coult) XH H=(+sindi++cosbb) x 2 mc=[+sindi+(+++cosbb] H===mc+ sin+[-(++6050)++cosoi] Ha=-= mc+ Vsino; (1) EQUATIONS OF MOTION 3mc中Ysino ΣF = Σ(F) :: HORIZ, COMP.: TsinB=ma (2) VERTICAL COMP Tcosp-W=0 C shirt -T cos 1 = 419 (3) DIVIDE (2) BY (3): tan B = a a = q tans (4) + 9 2 MB = S(WB) + 1: -mgcsinb= = mc+ Vsino - (mgtans) c coso DIVIDE BY MIGC COSO AND SOLVE FOR Tand: tan 0 = 1+ (2c + 1/5g) (5) PROBLEM 18,115 LETTING (5=40°, C= 4 ft, p=6 rad/s, g = 32.2 4/5° IN (3)" tan 0 = tan 40 / (1 + 0.018 634 W) (a) POR V = 0; 0 = 40.0° tan 0 = tan 40° (b) FOR \= 50 ral/s: tand = 0.43438 (c) FOR \= -50 ral/s: tand = 12.285 $\theta = 23.5$ 0 = 85.30 PROBLEM 18, 116 $\bar{a} = \bar{z} \dot{\phi}^2$ RECALLING(4): $\dot{\phi} = \frac{\bar{a}}{\bar{z}} = \frac{g \tan \beta}{(AB) \sin B}$ (AB) sin B + c sin B WITH p = 35, 0 = 20, AB=1.25ft, c = 0.25ft, g = 32.2 fl/5 : \$ = 5.3006 red/s (b) \$= 5,30 rad/s SOLVING (5) FOR CAT,

c + V = 2.5 g (tan B -1) = 2.5(32.2) (tan 35° -1) = 74.366

(a) \ = 56.1 rad/s

V= 74.366 = 56.12 ead/s



(PRECESSI .. OF THE ENVINUEES)

GIVEN:

RATE OF PRESESSITY OF ENTIRE

ABOUT GA = Irev IN 25 800 yr
POR EHRTH: Pore = 5.51

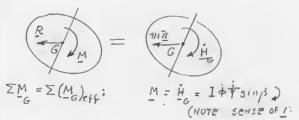
200= 6370 km, I = 2 m taxe

FIND:
AVERAGE VALUE OF COMES M
DUE TO GRAVITATIONAL
ATTRACTION OF SUN, MOON,
AND PLANEIS.

OF STATE OF

EQ.(18.7:): $\frac{H}{G} = (\frac{H}{G})_{GAyT} + \Omega \times \frac{H}{G} = O + (-\frac{1}{4}\sin\beta i + \frac{1}{4}\cos\beta k) \times H$ $\frac{H}{G} = (-\frac{1}{4}\sin\beta i + \frac{1}{4}\cos\beta k) \times [-\frac{1}{4}\sin\beta i + \frac{1}{4}(\frac{1}{4}i\cos\beta k)]$ $= \frac{1}{4}\sin\beta (\frac{1}{4}i\cos\beta - \frac{1}{4}\cos\beta k) \frac{1}{4}$ $= \frac{1}{4}i\sin\beta \frac{1}{4}i\cos\beta - \frac{1}{4}i\cos\beta \frac{1}{4}i\cos\beta k \frac$

EQUATIONS OF MOTION



WITH GIVEN DATH: $m = \frac{4}{3}\pi\epsilon^{2}\rho = \frac{4}{3}\pi(6.37 \times 10^{6} \text{m})^{3}(5.51 \times 10^{3} \text{kg/m}^{3})$ $= 5.9657 \times 10^{24} \text{kg}$ $\vec{I} = \frac{2}{5}\pi 1\epsilon^{2} = \frac{2}{5}(5.9657 \times 10^{24} \text{kg})(6.37 \times 10^{6} \text{m})^{24}$ $= 96.827 \times 10^{26} \text{kg} \cdot \text{m}^{6}$ $\dot{\phi} = \frac{2\pi \text{ rad}}{(2580037)(365.24 \text{ day})(3r)(24 \text{ h/day})(3600 \text{ s/h})} = 7.717$

 $\psi = \frac{2R \text{ rad}}{(23,93 \text{ h})(3600 \text{ s/h})} = 72.935 \times 10^{-6} \text{ rad/s}, \quad \beta = 13.45^{\circ}$ $M = \dot{H}_{S} = 1 + \dot{\psi} \sin \beta = (96.827 \times 10^{36} \text{kg·m}^{\circ})(7.717 \times 10^{12} \text{s}^{-1})(72.935 \times 10^{-6} \text{s}^{-1}) \sin 23.45^{\circ}$ $= 21.69 \times 10^{21} \text{ N·m}$

M=21.7×10 N.m

18.118



GIVEN:

PROJECTILE WITH M = 20 kg R, = 50 mm, R, = 200 mm \$ = 600 m/s (HORIZONTAL) 1 DRAG = D = 120 N (HORIZONTA) 15 = 3°, c = 150 mm V=6000 rpm

FIND :

(a) APPROYIMATE -LUE OF LATE OF PRECESSION , (b) EXT . VALUES OF TWO FOR BUT IN THE PERSONS.

SINCE THE DRAG D IS A FORCE CONSTANT IN MAGNITHDE AND DIRECTION (LIKE THE WEIGHT OF A TOP), IT WILL PRECESS, LIKE A TOP, " - W" AN AXIS GZ I PRIL EL TO THILL FOR .E.



UTIVE THE HEES -XHZ, Note & markets awas

He = I psing i + I (++ brosp)k w=+sinpi+(Y++cosp)k

FRAME GZYZ: IL = \$ sin si + \$ wspk

Ed (18.51): He = (HB) 347 + 1 ×He = 0 + 1 ×He H = (+ sinfi + + cosp k) x [] + sinpi +] (V+ + cosp) k] = \$\dots \inp\[-\bar{1}_2(\dark_+ \dark \cos\beta) + \bar{1}_1 \dark \cos\beta] j

THUS! ZM) = H = +simp (I - I) + cosp - I, +) à AN THE OTHER HAME.

ZM = ckx (-DK) = - cD sinfit (2)

 $Z_{G}^{M} = Z(\underline{M}_{G})_{eff}: -cD = \phi [(\underline{I}_{I} - \underline{I}_{I}) + \cos \beta - \underline{I}_{I} + \overline{I}_{I}]$ (3)

(a) APPROXIMATE VALUE OF &

SIMILE V>> b. WE MM; NECLECT THE FIRST TERM IN THE BURKET IN (3). WE DRITAIN

I. PT=cD

WITH GIVEN DATA: T= mk2 = (20kg)(0.05m)=0.05kg·m C=0.15m, D=120 N, V=6000 rpin= 200 Trad/c:

0.05 \$ (200 11) = (0.15)(120)

p= 0.5730 rad/s \$= 5,47 rpm

(b) EXACT VALVES OF &

USING EG. (3) WITH THE ABOVE DATA AND WITH B = 3° AND I = m K = (20 K) (0.2 m) = 0.8 kg·m*: - (0.15m)(120N) = \$ (0.8-0.05) \$ ccc 3-0.05(2007) $0.74897\dot{\phi}^2 - 31.416\dot{\phi} + 18 = 0$ \$ - 41.945 + 24.033 = 0

\$= \frac{1}{2} (41.945 \div \((41.945)^2 - 4(24.033))

 $=\frac{1}{2}(41.945 \pm 40.783)$ rad/s

= 41.364 rad/s n. \$= 0.58101 rad/s

\$= 395 rpm AND \$= 5,55 rpm

18.119

GIVEN:

AXISYMMETRICAL BODY UNLER NO FORCE I - MOMENT OF INERTIA ABOUT AXIS OF SYMPLETRY - TRANSVERSE AXIS THRU G. HG = ANG. MOM. ABOUT G.

SHOW THAT:

AND W = HG COS 8 (I'-I)

FROM E. O. (18,40), PAGE 1146:

$$\omega_{\lambda} = - + \sin \theta \tag{1}$$

FROM THE PIRST OF ERS. (18.41), PAGE 1147:

$$\omega_{\chi} = -\frac{H_{G} \sin \theta}{T'} \tag{2}$$

EQUATING THE R.H. MEMBERS OF (1) AND (2):

$$-\phi \sin \theta = -\frac{H_G \sin \theta}{I'} \qquad \dot{\phi} = \frac{H_G}{I'} \qquad (Q.E.D.)(3)$$

FROM FIG. 18.21: V = W, - \$ 5054 (4)

FRMI EUS. (16.48):
$$N_{c} = \frac{H_{c} \cos \theta}{\Gamma}$$
 (7)

FROM ER. (3) AHOVE: \$cos 8 = HE CIST (6)

SIBSTITUTE FROM (5) AND (6) INTO (4):

$$\dot{V} = H_G \cos\theta \left(\frac{1}{I} - \frac{1}{I'} \right)$$

$$\dot{V} = \frac{H_G \cos\theta \left(\mathbf{l'} - \mathbf{D} \right)}{I I'} \qquad (Q. E. D.)$$

18.120

AXISYMMETRICAL BUDY UNDER NO FORCE I = MUHENT OF INCRTIA ABOUT AXIS DESYMPETRY -TRANSVERSE. AXIS THRU G 6 = ANGLE METHTEN AXES OF PRECESSION & SPIN W, = COMPONENT OF W ALONG AXIS OF SYMMETRY

SHOW THAT:

$$\dot{q} = \frac{1}{100} \frac{1}{10$$

(b) EQ. (18.44) IS SATISFIED

(a) SEE SOLUTION OF PRUB. 18,119 FOR DETINITION (3)

of
$$\Xi \omega_{\bullet}(3)$$
: $\dot{\phi} = \frac{H_G}{I^3}$

FROM 205. (18.48): HG = INE

SUBSTITUTE FOR H_G IN (3): $\dot{\phi} = \frac{T \, \hat{\omega}_e}{I' \cos \theta}$ (Q.E.D.)

(b) FROM RELATION JUST OBTAINED, NE HAVE IW, - I' & cos 0 = 0

WHICH SHOWS THAT, FOR AN NXISYMMETRICAL BODY UNDER NO FORCE, THE R.H. MEMBER

IM = (IW2 - I + cost) + sint (18.14) IS EQUAL TO ZERO. BUT, SINCE THERE IS NO FORCE, WE ALSO HAVE ZM = 0 AND EQ. (18.44) IS SATISFIED. (Q.E.D.)

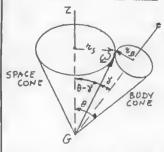
18.121

GIVEN:

AXISYMMETRICAL BUDY UNDER NO FORCE I = MUMENT OF INERTIA ABOUT AXIS OF SYMMETRY. - TRANSVERSE AXIS THRU G-W, = COMPONENT OF W ALONG AXIS OF SYMMETRY. SHOW THAT:

ANGULAR VELOCITY W IS OBSERVED FROM THE BODY TO ROTATE ABOUT THE AYIS OF STRINETRY AT THE RATE

$$m = \frac{\Gamma - I}{\Gamma'} \omega_{z}$$



ASSUMITIS DIRECT PRECESSION (I'> I) WE CONSIDED THE SPACE AND BODY CONES. THE PLANE ZGE ROTHTES ABOUT THE ZAXIE AT THE FATE &; SO WILL THE VECTOR W CONTAINED IN THAT PLANE THUS, THE TIP OF W

WILL DESCRIBE AN ARC OF CIRCLE OF LENGTH to A DE IN THE TIME At. BUT, ACCORDING TO THE DEFINITION OF N, THE VECTOR W IS OBSERVED TO ROTATE AT THE RATE IN WITH RESPECT TO THE BODY, THUS THE TIP OF O WILL DESCRIBE AN ARE OF CIRCLE OF LENGTH & n At IN THE TIME At, SINCE THE BODY CONE ROLLS ON THE SPACE CONE, WE HAVE ts + Dt = t, n Dl

BUT, FROM THE SKETCH ABOVE.

tg= W Sin (8-8) AND tg = W Sin &

SUBSTITUTING INTO (1).

$$\frac{\dot{\Phi}}{\sin(\theta-\delta)} = n \sin \delta
n = \dot{\Phi} \frac{\sin(\theta-\delta)}{\sin \delta}$$
(2)

WE RECALL THE RELATION DERIVED IN PROG 18, 120:

$$\dot{\phi} = \frac{\Gamma \omega_{\delta}}{1' \cos \theta}$$

$$n = \frac{I \omega_2}{I' \cos \theta} \frac{\sin \theta \cos \theta - \sin \theta \cos \theta}{\sin \theta}$$
$$= \frac{I \omega_E}{I'} \left(\frac{\tan \theta}{\tan \theta} - 1 \right)$$

RECALLING FROM EQ. (18.49) THAT $\frac{\tan \theta}{\tan x} = \frac{I}{I}$, WE HAVE

$$n = \frac{I}{I'} \left(\frac{I'}{I} - I \right) \omega_{z}$$

$$n = \frac{I' - I}{I'} \omega_{z} \left(Q, E, D, \right)$$

NOTE. FOR I > I' (RETRUCKALE PRECESSION) WE WOULD FIND $m = \frac{I - I'}{I'} \omega_z$

18.122

GIVEN:

AXISYMMETRICAL GODY UNDER NO PURCE AND IN RETROGRADE PRECESSION (I>I'), SHOW THAT:

(a) RATE OF RETROGRADE PRECESSION CANNOT BE LESS THAN TWICE THE RATE OF SPIN: 14/22/1/, (b) THE AXIS OF SYMMETRY IN FIG. 18:24 CHN NEVER LIE WITHIN THE SPACE CONE.

I' + coso = I We

SUBSTITUTING W=+++ ws 0, WE HAVE I' + cuso = I (+ + (uso)

SOLVING POR +,

$$\oint = -\frac{I}{I-I'} \frac{\dot{Y}}{\cos \theta} \quad \partial R \quad \dot{\phi} = -\frac{3ec \theta}{I-(I'/I)} \dot{V} \quad (2)$$

FOR RETRUGENCE PROTESSIONS I'/I < 1 ON THE OTHER HAND, THE SMALLEST POSSIBLE VALUE OF I/I IS 1/2 (WHICH CORRESPONDS TO THE CASE OF A FLAT DISK OR ANNILUS).

fus:

$$\frac{1}{2} \le \frac{\underline{\underline{I}}}{\underline{\underline{I}}} < 1 \quad OR \quad \frac{1}{2} \ge 1 - \frac{\underline{\underline{I}}}{\underline{\underline{I}}} > 0$$

$$OR \quad \frac{1}{1 - (\underline{\underline{I}}/\underline{\underline{I}}^2)} \ge 2$$

RECALLING THAT SEC & 21, WE MUST HAVE FROM (2)

(b) WE RECAL: EQ. (18.49): $\tan V = \frac{I}{r}, \tan \theta$

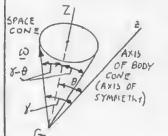
SINCE
$$\frac{I}{I} \ge \frac{1}{2}$$
 AS SHOWN ABOVE, $\frac{I}{I}$, ≤ 2 AND $\tan 3 \le 2 \tan \theta$ (3)

WE WRITE THE TRIGONUMETRIC IDENTITY tan (8-0) = tan 8 - tan 8

SUBSTITUTING INTO (2) AND EXPANDING Sin (8-8): SINCE & < T AND B < T, WE HAVE I+ tand tant >1

AND, PRUM (3): tan 8 - tan 0 5 tan 0

Titus
$$tan(8-8) \leq \frac{tan \theta}{\theta}$$



tan (1-1) < tan + X-8 5 A

THE Z AXIS CANNUT

LIE WITHIN THE S PACE CONE

(SEE SKETCH) (REDI)

(FREE PRECESSION OF THE EARTH) 18.123 GIVEN:

T = MOM. OF INERTIA OF EARTH ABOUT AXIS OF SYMMETRY. - - - - TRANSVEKSE 1XIS

I'- 0,9967 I

RELATION DE ZIVED IN PROB. 18.121:

$$n = \frac{I - I'}{I'} \omega_i \qquad (FOR \ I > I')$$

WHERE WZ = COMPONENT OF W OF EARTH ALONG AXIS OF SYMMETRY , AND M = RATE AT WHICH W IS OBSERVED FROM THE FARTH TO LOTHTE ABOUT ITS AXIS OF SYMMETRY.

FIND:

PERIOD OF PRELESSION OF NORTH POLE.

$$PERIOD OF PRECESSION = \frac{2\pi}{n} = \frac{I'}{I-I'} \frac{2\pi}{\omega_2} = \frac{I'}{I-I'} (1 day)$$

$$BUT \frac{I'}{I-I'} = \frac{0.9967 \ I}{0.0033 \ E} = 302$$

THUS: PERIOD OF PRECESSION = 302 days





GNEN:

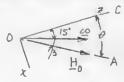
FOOTBALL KICKED WITH HORIZONTAL ANG. VEL. 12 OF HAGNITUDE 200 rpm. RATIO OF AXIAL AND

TRANSVERSE MOMENTS OF INERTIA 15 I/I'=1/3.

(A) ANGLE B BETWEEN WAND PRECESSION AXIS OA.

(b) RATES OF PRECESSION AND SPIN.

(a) USING REFERENCE FRAME OXY WITH I FOUNTING A PAY



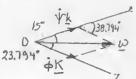
W, = W sin 150 = W cos 15 は,= I'w, = I'w sin 15" Hy = I'W4 = 0 H2 = I W1 = IW COS 15

H2 = I'w sin15" = 1 tan15= 3 tan15

0 = 38.794° tand = 0, 80385

B= 0-15° = 38.794°-15°=23.794° B=23,8"

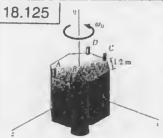
(b) USING THE OBLIGUE COMPONEIVTS OF W ALONG OA AND OC:



LAW OF SINES: w do Sin 15° Sin 23,794"

SETTING W= 200 ipm, WE FIND

RATE OF PRECESSION = 4 = 82.6 rpm RATE OF SPIN = 128.8 cpm



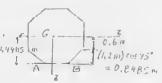
GIVEN: 2500 - kg Sti = . ITE, 2.4-m HIGH WITH MEMBER THE Z. kje kje oloomi, kje oloen. SIMPLE TEN 11 1 -KHIE OF BOARDS. IN 50 WHEN 20-H TIER TAL ALLINE ENTERISING FL . 25 FXPELLING FIRE IN FEITIVE I TI 30 MI

FIND: (a) PRECESSION AXIS, (b) \$, (c) \.

INITIM: ANG. VELOCITY: 14 (0.062 432 501/5) & = (0.062 432 501/5) &

(H) = mky w = (2500 kg)(0.9811)(0.062832 rad/s) &

= (150.86 kg.n1/5) } ANG. IMPUL 5: MGOL = (1.4485m) KX X2(-20N)1(251 M Dt = (115.88 kg.m/s)i



PRINCIPLE OF INTIME PAID MOMENTUIT

FINAL MONENTUN: He = (He) + Me Ot = (150.86 kg - m/s) j + (115.88 kg mi/) i

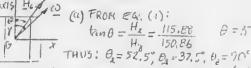
H = (115.88 kg·ni/s) i + (150,86 kg·ni/s) j (1)

WE RECALL THAT

H=Iwi+Iwij+Iwik H = (2500 kg)(0,90m) w i + (2500 kg)(0,98m) wy + I w k EQUATING THE COEFT OF E, j, K IN (1) HND (2):

2025 W, = 115.88 W2 = 57.225 × 10 3 rad/s Wy = 62.832 x 15 101/5 2401 Wy = 150.86 (3)

I2 W2 = 0 W== 0 SPINIJE AXIS HEA PRECESSION AXIS



PRIME 44(3): tay 8 = 10x = 57.225 8 = 42326 wy 62.8.32

W = 102 + Wy = 84.186 112 ranks PRECESSIUM LAW OF SINES! SPIN AXIS 5111(5-0) 84.986×10-3

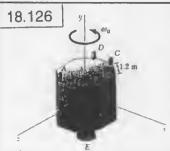
Sin 37,529 Sin 42,360 Sin 4,797 SOLVING FOR & AND Y:

(b; \$= 93.9+1×10 rad/s (c) = 11.667 × 10 rails

4=53,8 rev/h V= 6.68 rev/h

8=37,529

WE CHECK FROM DIAGRAM THAT PRECESSION IS RETROGRADE. (IT HAD TO BE, SINCE Ly > K AND, THUS, I > I')



GIVEN: 2500-18 SHTELLITE 2,4-m HIGH WITH OCTUGONAL BASE. K= K= U. Tom. Lu= 0.98m. SATELLITE SPINNING AT KATE OF 36 rev/h ABOUT Gy WHEN ZO-N THRUSTERS AT A AND D ARE ACTIVITED FOR 2 S, EXPELLING FUEL IN POSITIVE & DIRECTION

FIND: (a) PRECESSION AXIS, (b) + (c) T.

INITIAL ANG. YELUCITY: = (36 FeV) (2 Frad) (3600 s) = (0.062032 rod/s) j INITIAL ANG. MOMENTUM:

(HG) = 11 K, w = (2500 Kg)(0.98m) (0.062832 rack); = (150.86 kg·mi/s)j

ANG. IMPULSE:

Mc St=- (0.6m) i x2 (-20N) j (2s)

M Ot = (48.0 kg.mi/s)k



PRINCIPLE OF IMPULSE AND MOHENTUM

FINAL MOMERTUM:

HG=(HB)+MB Dt = (150.86 kg·m3/s)j+(48.0 kg·n1/5)k(1) WE RECAL THAT

HG= I20, 1+ I, W, 1+ I, W, 1

H = I, w, i + (2500 kg)(0.98m) w, j+(2500 kg)(0.90m) w, b

ERVATING THE COEFF. OF 1, 1. 1 IN(1) HAD (2):

 $I_x \omega_x = 0$

2401Wy=150.86

Wy = 62.832 x 10 3 rad/s W3 = 23.704 X10 TANK

2025 0, = 418.0 PREC H SPIN AXIS

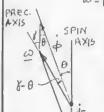
(a) FROM EU.(1):

 $tan \theta = \frac{hz}{H_y}$ $\frac{H_e}{H_1} = \frac{48.0}{150.86}$ B=17.650

THUS: 0 = 90°, 0 = 17.65°, 0 = 72.35°

FROM EQS. (3): $tan x = \frac{\omega_2}{\omega_3} = \frac{23.704}{62.832}$ 8 = 20.669° 62.832

w= 10, + W2 = 67.155 × 10 rad/s



(6)

LAW OF SINES!

Sin 20.669° sin 3.019° SCLVING FIL & AND Y:

\$ = 78.177 × 10 rad/s Y = 11.665 × 10 rad/s \$ = 44.8 rev/h Y= 6.68 rev/h

WE CHECK FROM DINGRAM THAT PRECESSION IS RETROFRADE (IT HAD TO BE, SINCE Ly>k, AND, THUS, I>I',)

18.127 and 18.128

GIVEN:



SPACE STATION CONSISTS OF TWO JECTIONS A AND B OF THE SAME WEIGHT WHICH ARE RIGIDLY CONNECTED. EACH SECTION IS DYNAMICALLY EQUI -VALENT TO A HUMO GENEOUS CYCINDER, STATION IS PRECESSING ABOUT 6.D AT THE CONSTANT RILTE OF 2 rev/h. PRUBLEM 18. 127:

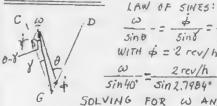
FIND THE RATE OF SPIN OF THE STATION ABOUT CC. PROBLEM 18, 128:

IF CONNECTION IS SEVERED BETWEEN A AND B, FIND FOR SECTION A:

(a) THE ANGLE CETWEEN CC' AND THE PRECESSION AXIS, (b) \$, (c) Y.

FOR ENTIRE STATION: $I' = \frac{1}{12} ni (3a^2 + L^2)$ EQ. (18.49): $\tan \delta = \frac{I}{I}$, $\tan \theta = \frac{6(9)^4}{3(9)^4}$ = 58.252 × 10 3 tan 40°, 8=2,7984°

PROBLEM 18.127



Sind WITH \$ = 2 rev/ h 1 2 rev/h Sin 2.7984" SOLVING FOR W AND Y'

W=26.332 rev/h W= 24.8 rev/h

PROBLEM 18.128 FOR SECTION H:

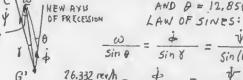
(3) (a) ANELE BETWEEN SPIN AXIS AND W IS STILL 8-2.7984" NOW: $\frac{I}{I^3} = \frac{6a^6}{3a^6 + L^2} = \frac{6(9)^6}{3(9)^2 + (45)^3} = 0.21429$

EQ.(18.49): $tan Y = \frac{I}{I}$ $tan \theta$ tan 1 = 0.21429 tan 0

tant = tant = tan 2.7984° = 0.22811

A=12.85 B= 12.850°

(b) AND(c) WE HAVE W= 26.332 RV/h, 8=2,7984° AND 8 = 12,850°



Sin (8-8) 26.332 rev/h _ ф Sin 12,850" sin 2,7984° sin 10.052°

SOLVING FOR + AND Y:

(b) \$ = 5.781 rev/h

\$ = 5.78 rev/h

V= 20.665 RU/h

V= 20,7 rev/h

18.129

GIVEN:

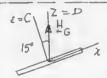
C 15* D

COIN SPINS AT THE RATE OF 600 pm ABOUT AMIS GC PERPENDICULAR TO COIN AND PRECESSES ABOUT VERTICAL DIRECTION GD.

FIND:

(a) ANGLE BETWEEN W AND GD.

(b) RATE OF PRECESSION ABOUT GD.



!T FULLOWS PROM THE ABOVE STATEMENT THAT H_G IS DIRECTED AS SHOWN AND THAT THE ANGLE BETWEEN THE AKES OF SPIN AND PRECESSION IS $\theta = 15^{\circ}$

$$I' = I_2 = \frac{1}{4} m t'$$

EQ. (18.49):

$$\tan \delta = \frac{I}{I} \tan \theta = 2 \tan 15^{\circ}$$
 $\delta = 20.187^{\circ}$

(a) ANGLE BETWEEN : AND GD



THE ANGLE & WE HAVE FOUND IS THE ANGLE BETWEEN & AND GC. THE ANGLE BETWEEN W AND GD 15

$$8-\theta = 28.187^{\circ}-15^{\circ}$$

= 13.187°
 $8-\theta = 13.19^{\circ}$

(b) RATE OF PRECESSION

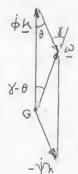
THE RATE OF SPIN 15 V = 600 PM

RESOLVING THE ANGULAR VELOCITY W INTO

ITS SPIN COMPONENT V & AND ITS PRECESSION

COMPONENT PK, WE DRAW THE FOLLOWING

DIAGRAM:



LAW OF SINES: $\frac{\dot{\phi}}{\dot{\phi}} = \frac{\dot{\psi}}{\sin(Y-\dot{\theta})}$ $\dot{\phi} = \dot{\psi} = \frac{\sin x}{\sin(Y-\dot{\theta})}$ $= (600 \text{ rpm}) \frac{\sin 28.187}{\sin 13.187^{\circ}}$ $\dot{\phi} = 1242 \text{ rpm}$

WE NOTE FROM DIAGRAM THAT THE PRECESSION IS RETROGRADE

THIS COULD HAVE BEEN ANTICI PATED, SINCE I/I' = 2 > 1.

18.130 SOLVE SAMPLE PROB. 18.6, ASSUMING
THAT THE METEURITE STRIKES THE
SATELLITE AT C WITH M. = (2000 m/s) L.

(a) ANGULAR VELOCITY AFTER IMPACT

FRUM SAMPLE PROB. 18.6:

$$I = I_2 = \frac{1}{2}ma^2$$
 $I' = I_2 = I_3 = \frac{5}{4}ma^2$

ANG, MOMENTUM AFTER IMPACT:

$$H_{c} = \mathcal{I}_{c} \times m_{o} \mathcal{Y}_{o} + I \omega_{o} k$$

$$= (-a \underline{i} - a \underline{k}) \times m_{o} \mathcal{V}_{o} \underline{i} + I \omega_{o} k$$

$$H_{c} = -m_{o} \mathcal{V}_{o} a \underline{i} + (I \omega_{o} + m_{o} \mathcal{V}_{o} a) \underline{k}$$
(1)

BUT $H_6 = I'\omega_2 L + I'\omega_3 J + I\omega_2 K$ (Z)

EQUATING THE WEFF, OF THE UNIT VECTORS IN (1) IND(2) $\omega_{z} = 0 \qquad \omega_{y} = -\frac{m_{0}v_{0}a}{I^{2}} = -\frac{H}{5}\frac{m_{0}v_{0}}{ma}$ $\omega_{z} = \omega_{0} + \frac{m_{0}v_{0}a}{I} = \omega_{0} + 2\frac{m_{0}v_{0}}{ma}$

EIVEN DATA: $\omega_s = 60 \, \text{rpm} = 6.283 \, \text{rad/s}$ $\frac{m_0}{m_1} = 0.001 \qquad a = 0.2 \, \text{m} \qquad c_0 = 2000 \, \text{m/s}$

WE FIND $\omega_{\lambda} = 0$ $\omega_{y} = -2 \text{ rad/s}$ $\omega_{z} = 11.283 \text{ rad/s}$ $\omega = \sqrt{\omega_{x}^{2} + \omega_{z}^{3}} = 11.459 \text{ rad/s}$ $\omega = 109.7 \text{ Fpm}$

 $\cos \delta_{\lambda} = 0 \qquad \cos \delta_{g} = \frac{\omega_{3}}{\omega} = -0.17453 \qquad \cos \delta_{z} = \frac{\omega_{3}}{\omega} = 0.48464$

 $\chi_2 = 90^\circ$, $\chi_y = 100.05^\circ$, $\chi_z = 10.05^\circ$

(6) PRECESSION AXIS

SINCE IT IS DIRECTED INCINE HE WIR USE EN (1) AND WRITE

 $\begin{aligned} H_{\chi} &= 0 \; , \quad |H_{\psi} &= -m_0 V_0 \; a = -\frac{m}{1000} \; (2000)(0.8) = -(1.6) m \\ H_{\chi} &= I \; \omega_0 + m_0 V_0 \; a = \frac{1}{2} \; m \; a^2 \omega_0 + m_0 \; V_0 \; a \\ &= \frac{1}{2} \; m \; (0.8)^4 (6.263) + (1.6) m \; = (3.6106) m \\ H_{\zeta} &= \sqrt{H_{\chi}^2 + H_{\chi}^2} \; = (3.71442) \; a_1 \end{aligned}$

 $\cos \theta_{g} = 0$, $\cos \theta_{g} = \frac{H_{f}}{H_{G}} = -0.40515$, $\cos \dot{\theta}_{g} = \frac{H_{f}}{H_{G}} = 0.91425$ DIRECTION OF PRICESSION AYIS IS

0 = 90°, 0 = 113,9°, 6 = 23.9°

(C) RATES OF PRECESSION AND SAIN



WE HAVE $\theta = \theta_2 = 23.9^{\circ}$ $\delta = 8_2 = 10.05^{\circ}$ $\theta - 8_2 = 13.85^{\circ}$

Sin (8-8)

51 n 13 850

 $\frac{\omega}{\sin \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \sin \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \sin \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \sin \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \cos \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \cos \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \cos \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$ $\frac{\partial}{\partial \cos \theta} = \frac{\psi}{\sin \theta} = \frac{\psi}{\sin \theta}$

RATE OF PRECESSION = $\dot{\phi}$ = 47.1 rpm RATE OF SPIN = \dot{V} = 64.6 rpm

18.131 and 18.132

GIVEN:



DISK OFMASS M IS FREE
TO ROTATE ABOUT A B
FURK-ENDED SHAFT OF NEGLIGIBLE MASS IS FREE TO ROTATE
IN BEARING C.
PROBLEM 18. 131:

PROBLEM 18.131:

INITIALLY, 0 = 90°, 0 = 0,

P = 8 rad/s.

IF DISK SLIGHTLY DISTURBED

FIND IN ENSUING MOTION

(a) MINIMUM VALUE OF \$.

PROBLEM 18.132: INITIALLY 0 = 30°, 0 = 0, + = Bradle.

FIND IN ENSUING MOTION (a) RANGE OF VALUES OF B, (b) MINIMUM &, (c) MAXIMUM &.



USING THE AXES GX42: \(\omega = \theta i + \theta sin \theta j + \theta cos \theta k
\)
CONSERVATION OF ANGULAP
MOMENTUM:

SINCE DISK IS FREE TO KOTATE
ABOUT THE Z AXIS, WE HAVE $H_Z = \text{constant} \qquad (1)$ BUT $H_Z = H_y \sin \theta + H_z \cos \theta$

 $H_{Z} = I_{g} \omega_{g} \sin \theta + I_{g} \omega_{g} \cos \theta = \frac{1}{4} m d + \sin^{2} \theta + \frac{1}{2} m d + \cos^{2} \theta$ $= \frac{1}{4} m d + (\sin^{2} \theta + 2\cos^{2} \theta) = \frac{1}{4} m d + (1+\cos^{2} \theta)$ $USING THE INITIAL CONDITIONS, EU. (1) YIELDS
<math display="block">+ (1+\cos^{2} \theta) = + (1+\cos^{2} \theta) \qquad (2)$

CONSERVATION OF EVERGY

SINCE NO WORK IS DONE, WE HAVE T = constant (3)
WHERE

 $T = \frac{1}{2} \left(I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} \right)$ $T = \frac{1}{2} \left(I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} \right)$ $T = \frac{1}{2} \left(I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} \right)$

 $T = \frac{1}{2} \left(\frac{1}{4} m \alpha^2 \dot{\theta}^2 + \frac{1}{4} m \alpha^2 \dot{\phi}^2 \sin^2 \theta + \frac{1}{2} m \alpha^2 \dot{\phi}^2 \cos^2 \theta \right)$ $= \frac{1}{6} m \alpha^2 \left[\dot{\theta}^2 + \dot{\phi}^2 \left(\sin^2 \theta + 2 \cos^2 \theta \right) - \frac{1}{6} m \alpha^2 \left[\dot{\theta}^2 + \dot{\phi}^2 \left(1 + \cos^2 \theta \right) \right] \right]$ $USING THE INITIAL CONDITIONS, INCLUDING <math>\theta_0 = 0$, EU. (3)

YIELDS $\theta^2 + \frac{1}{7}(1+\cos^2\theta) = \frac{1}{7}(1+\cos^2\theta_0)$

PROBLEM 18.131

(a) WITH $\theta_0 = 90^\circ$ AND $\phi_0 = 8 \, \text{rad/s}$, EQ.(1) YIELDS $\dot{\phi} = \frac{8}{1 + \cos^* \theta}$ $\dot{\phi}$ IS MINIMUM FOR $\theta = 0$: $\dot{\phi}_{\text{min}} = 4.00 \, \text{rad/s}$

(b) EQ.(4) YIELDS = 64 - + (1+cos't) = 64(1-11+cos't)

 $\dot{\theta}_{max}^2 = 64(1-\frac{1}{2}) = 32$ $\dot{\theta}_{max} = 5.66 \text{ rad/s}$

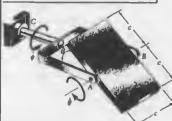
PROBLEM 18.132

(a) WITH 80=30°, \$= 8 rad/c IN (2) \$\display(1+\cos^2\theta) = 14 \display= 14/(1+\cos^2\theta) (5) SUBSTITUTE IN (4) AND SOLVE POR \$\display : \display= 112-\frac{196}{6}

SINCE \$ DO, WF MUST HAVE 1+cos'D > \frac{1945}{1125} - 30' \in \frac{1}{2} \\
(b) FROM (5), \$\phi\$ IS MINIMUM FOR \$\phi = 0\$: \$\phi_{min} = 7.00 \text{ rad}s\$

(c) FROM (6), \$\phi\$ is MAXIMUM FOR \$\phi = 0\$: \$\phi_{max} = 3.74 \text{ rad}s\$

18.133 and 18.134



GIVEN:

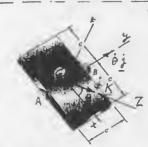
PLATE OF HASS M IS FREE TOROTH TE ABOUT AB.

FURK-ENDED SHAFT OF NEGLIGIBLE AIRSS IS FREE TO RUTHITE IN BEARING C. PROBLEM 18.133:

INITIALLY $\theta_0 = 30^\circ$, $\dot{\theta}_0 = 0$, $\dot{\phi}_0 = 6$ rad/s.

FIND IN ENSUING MOTION (a) RANGE OF VALUES OF & (b) MINIMUM VALUE OF &, (c) MAXIMUM VALUE OF &.
PROBLEM 18, 134:

INITIALLY & = 0, & = 0, + = 6 rad/s. IF PLATE
IS SLIGHTLY DISTURBED, FIND IN ENSUING MOTION
(A) MINIMUM VALUE OF +, (b) MAKIN'IM VALUE OF &.



USING THE AXES GZYZ

\(\Omega = \dagger \text{cos0} i + 0 \dagger + \dagger \dagger + \dagger \dagger

SINCE PLATE IS FREE TO

RUTATE ABOUT ZAXIS,

Hz = constant (1)

BUT Hz = H, cos 0 + Hz sin 0

 $H_{Z} = I_{Z} \omega_{Z} \cos \theta + I_{Z} \omega_{Z} \sin \theta = \frac{1}{12} mc^{2} + \cos^{2} \theta + \frac{5}{12} nic^{2} + \sin^{2} \theta$ $= \frac{1}{12} mc^{2} + (\cos^{2} \theta + 5 \sin^{2} \theta) = \frac{1}{12} mc^{2} + (1 + 4 \sin^{2} \theta)$ USING THE INITIAL CONDITIONS, EQ (1) YIELDS $+ (1 + 4 \sin^{2} \theta) = + (1 + 4 \sin^{2} \theta_{0}) \qquad (2)$

CONSERVATION OF ENERGY

SINCE NO WORK IS DONE, WE HAVE T = constant (3)

WHERE $T = \frac{1}{2} (I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2})$

 $T = \frac{1}{L} \left(\frac{1}{12} mc^2 \dot{\phi}^2 \cos \theta + \frac{1}{3} mc^2 \dot{\phi}^2 + \frac{5}{18} mc^2 \dot{\phi}^3 \sin^2 \theta \right)$

 $= \frac{1}{24} mc^{2} \left(4\dot{\theta}^{2} + \dot{\Phi}^{2} (\cos^{2}\theta + 5\sin^{2}\theta) \right) = \frac{1}{24} mc \left[4\dot{\theta}^{2} + \dot{\Phi}^{2} (1 + 4\sin^{2}\theta) \right]$ USING THE INITIAL CONDITIONS, INCLIDENCE $\dot{\theta} = 0$, Eq. (3) YIELDS $4\dot{\theta}^{2} + \dot{\Phi}^{2} (1 + 4\sin^{2}\theta) = \dot{\Phi}^{2} (1 + 4\sin^{2}\theta) \qquad (4)$

PROBLEM 18.133

(a) WITH 0=30 AND += 6 mils IN (2) AND (4):

 $\phi(1+4\sin^2\theta)=12$ $4\theta^2+\phi^2(1+4\sin^2\theta)=72$ (2,4) ELIMINATING & AND SOLVING FOR θ^2 : $\theta^2=18-\frac{36}{1+45}$ (5)

FOR $\dot{\theta}^2 \ge 0$: $1 + 4 \sin^3 \theta \ge 2$, $\sin^2 \theta \ge \frac{1}{4}$, $30^\circ \le \theta \le 150^\circ$

(6) FROM (2'), & ISHIN, FOR 0=90: \$\diangle_{ii_1} = 2.40 ral/s
(C) FROM (5), & ISHAX. FOR 0=90: \$\darkappa_{max} = 3.29 rad/s

PROBLEM 18.134

(a) WITH $\theta_0 = 0$, $\phi_0 = 6$ rad/s, EQ (2) YIELDS $\dot{\phi} = \frac{6}{1+4\sin^2\theta}$ $\dot{\phi}$ 15 MINIMUM FOR $\theta = 90^\circ$: $\dot{\phi}_{min} = 1.200$ rad/s

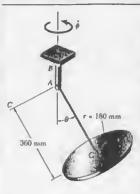
(b) Eq.(4) YIELDS: 40 = 36 - 4(1+4 sin'd) = 36(1 - 1+4 sin'd)

6 IS LANGEST FOR D = 10°:

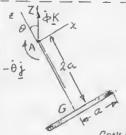
 $4b_{max}^2 = 36(1-\frac{1}{5})$ $b_{max}^2 = 7.20$ $b_{max} = 2.68 \text{ rad/s}$

18.135 and 18.136

GIVEN:



DISK WELDED TO RODA -OF NE -LIGIBLE HAY COMVECTED BY CLEVIS TO SHAFT AB. KOD AND DISK FREE TO KOTATE ABOUT AC; SHAFT FREE TO ROTATE ABOUT VERTICAL AXIS. INITIALLY, 8 = 90°, 8 = 0. PROBLEM 18.135: KNOWING THAT \$ = 2 \$0 FIND: (a) Omin, (b) 40 PROBLEM 18,136 KNOWING THAT Omin = 30°,



USING AXES AZYZ, WITH & POINTING INTO PAPER. W= + sin Di - Dj + + cost K 12 = 1y = 17 ma2, 12 = = = ma H= IW = 17 ma + sinb H2 = I, W2 = 1 ma + cost

FIND: (a) \$0, (b) +max

CONSERVATION OF ANG. HOM, ABOUT Z SINCE ONLY FURCES ARE REACTIONAT A AND W= - mg K, WE HAVE EM, = O AND H, = constant. THUS. $H_7 = H_2 \sin \theta + H_2 \cos \theta = \frac{1}{4} ma^2 \phi (17 \sin^2 \theta + 2 \cos^2 \theta)$ Hz= 1 ma + (2 + 15 sin 0) = constant USING THE INITIAL CONDITIONS, EQ. (1) YIELDS + (2+15 sin A) = 17 +

CONSERVATION OF ENERGY $T = \frac{1}{2} (I_2 \omega_1^2 + I_4 \omega_2^2 + I_4 \omega_2^2) = \frac{1}{2} \frac{m\alpha^2}{10} (174 \sin^2\theta + 176 + 24 \cos^2\theta)$ T= | ma [(2+15 sin' 0) +2+17 0] V=-21119acost USING THE INITIAL CONDITIONS, WE WRITE T+V= const. (R+15 sin20) ++ 17 02-16 \$ cas 0 = 17 %

PROBLEM 18.135 (a) LET += + may=2+ IN(2): 2+ (2+15 sin 0)=17+ 2+15 sin2 0 = 8.5, sin0 = V0.4333 0 = 41,169° Omin = 41,2°

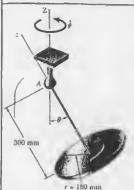
(b) LET 0=41,169°, 0=0, +=2+ IN (3): (2+15 sin 41.1690)(4+0) - 16 \$ cos 41.1690 = 1740 $17 \, \phi_0^2 = 12.044 \left(\frac{9.81}{0.118} \right) \quad \phi_0^2 = 38.63$ φ = 6.21 rad/s PROBLEM 18.136

(a) LET B=30° IN (2): \$\phi(2+3,75)=17\$\psi_0\$, \$\phi=\frac{17}{5.75}\$\phi_0\$ LET 0=30, 0=0 IN (3); $(2+3.75) \dot{\phi}^2 - 16 \left(\frac{9.81}{0.18}\right) \cos 30^\circ = 17 \phi_0$ 5.75 (17) \$ - 872 cos 30° = 179 17(17 -1) 4 = 872 cos30° 4. = 4.71 19 may. 4 = 4.76 raals

4 = 14.07 rank

(b) FROM (4): \$= 17 (9.7649)

GIVEN: *18.137 and *18.138



DISK WELDED TO ROD AS OF NEGLIGIBLE MASS SUPPORTED BY BALL AND SOCKET AT A. INITIALLY, 0=90, \$= 0=0. PROBLEM 18, 137: KNOWING THAT YO = 50 roals. FIND: (a) Dmin , (b) \$ AND IF FOR B = 0 min PROBLEM 18, 138: KNOWING THAT Dain = 30,

ARE THE REACTION AT A AND THE WEIGHT W=- TIGK AT 6.

USING AXES AZYZ, WITH Y AXIS POINTING INTO PAPER ω=+5in0i-0j+(+++cos0)k $I_{2} = I_{4} = \frac{17}{4} ma^{2}, \quad I_{2} = \frac{1}{7} ma^{2}$ H = ma[17 + sin 0 : - 17 + j +2 (++ + ms)) CONSERVATION OF ANG. MOMENTUM SINCE THE ONLY EXTERNAL FORCES

(b) & AND Y FOR B = Dain

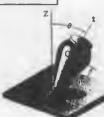
FIND: (a) y,

WE HAVE ZM_=0, ZM=0. SINCE Z IS PART OF A NEWTONIAN FRANCE, IT FOLLOW: THAT HZ=CONST.; PECHWE OF THE AXISMAMETRY OF THE DISK, IT ALSO FOLLOWE THAT H= CONST. (SEE PROB. 18.139). USING THE INITIAL CONDITIONS. WE WRITE TH + COSB = To H = const.: NOTING THAT HZ = H . K = MA [17 + 510 + 2 (+ + cost) cos e] AND SUBSTITUTING FROM (1) FOR THE INSIDE PARENTHESIS, Hy=wnst.: 17 \$ sin = 0 + 2 V cos θ = 0 CONSERVATION OF ENERGY T = \frac{1}{2} (I_2 \omega_2^2 + I_3 \omega_2^2 + I_4 \omega_2^2) = \frac{1}{2} \frac{1111}{4} [17 \phi \sin^2 \text{0} + 17 \theta + 2 (4 + 4 \omega \theta)^2] T= 1ma (17 + sin + 17 02+ 2 +0) 1=-2 mg a cost T+Y=const.: 17+ sin 10+170+210-162 cood = 110

\$ sin20 + 02 = 16 \$ cost (3) PROBLEM 18.137 (a) PROM (2): $\dot{\phi} = -\frac{2}{17} \dot{\psi}_0 \frac{\alpha \epsilon \theta}{\sin^2 \theta} = -\frac{2}{17} (50 \text{ rad/s}) \frac{\alpha s \theta}{\sin^2 \theta}$ CAKRY INTO (3) AND LET 0 = 0 FOR 0 = Dan : $\left(\frac{100}{17}\right)^2 \frac{\cos^2 \theta}{\sin^2 \theta} = \frac{16}{17} \frac{9.81 \, \text{ags}^4}{0.18 \, \text{m}}$ $\frac{10 \times 10^{3}}{16 \times 17} \frac{0.18}{7.81} \cos \theta = 1 - \cos^{3} \theta$ COSTO + 0,67458 COSB - 1 = 0

COSO= 1 (-0.67458 ± 2,1107) = 0.71806 8=44,105° OR - 1, 3926 (IMPOSSIBLE) 0 = 44,1" (6) SUBSTITUTING 70 = 50 rad/s AND 0 = 44. 105° IN (2) AND (1) EO.(2): \$= - \frac{2}{17} (50) \frac{\cos 44,105}{\cos 44,105} 4=-8,77 rad/s sin 441105° EQ(1): V= 50- (-8.72) cos 44.105° V= 56.3 radk PROBLEM 18.138 LET 0=310°, 0=0 IN (3): \$\display=\frac{16}{17} \frac{4.81 \text{ m/s}^2 \cos30^\circ}{5\text{in}^2 \frac{2}{30}}, \display=\frac{4}{5} = \frac{13}{330} FROM (2), WE NOTE THAT \$ <0 FOR \$>0. THUS: \$=-13.33 rad/s Eq.(2): $\psi = -\frac{17}{2}(-13.33)\frac{\sin^2 30}{\cos 30} = 32.708$, $\psi = 32.7744/5$

EU.(1): 7=32.700-(-13,33) cos30° 4 = 44.3 rad/s € 18.139



GIVEN:

TOP WITH FIXED POINT O P, O T = EULERIAN ANGLES I = MUM, OF INERTIA ABOUT & AXIS - - TRANSVERSE AXK, THRUUGH O. SHOW THAT!

 $I'\dot{\phi}\sin^2\theta + I(\dot{\psi} + \dot{\phi}\cos\theta)\cos\theta = \alpha$ (1)

 $I(\dot{\psi} + \dot{\phi}\cos\theta) = B$

(b) W= const. AND &= FUNCTION OF B



WEUSE PRAMEDXUZ WITH & AXIS POINTING INTO PAPER

ANG, VELOCITY OF TOP; w=-45in0 i+01+(4+4000)E(A) ANG. VELOCITY OF PRAHE! Q =- + sinti+ + j++ costk ANG, MOMENTUM ABOUT OF

Ho= I, wz i + I, wy j + Izwz k H =- I' + sin + i + i' + i (+ + coso) k

(a) WE RECALL IM = Ha (18.27)SINCE THE ONLY EXTERNAL FORCES ARE THE REACTION AT DAND THE WEIGHT WE- mg K AT G, WE HAVE ZMZ=0 AILD FRUM (18,27) H = D. SINCE THE Z AXIS IS PART OF A NEWTONIAN FRAME OF REFERENCE. IT FULLOWS THAT H, = constant. BUT H,= H.K. SUBSTITUTING FOR HO FROM (C) AND NOTING THAT i.K = - sind, j. K = O, k.K = cost, WE HAVE H, = H. K = - I' + sino (-sino) + 0 + I(++ + 650) cost THUS: WHERE ON IS A CONSTANT.

WE OBSERVE THAT WE ALSO HAVE IM, = 0, BUT WE CANNOT CONCLUDE THAT H = const. , FINCE THEZ AXIS IS NOT PART OF A NEWTONIAN PRAME OF REFERENCE USING ER. (1828), WE WRITE

(18,28) E MO = (H) 0x43+ 1x H0

SUBSTITUTING FROM (B) AND (C) INTO (18.28), IM =- I'd (+ sino) + I'o) + Id (++ + wso) + + (-+ sindi+0j++coods) x[-I' + sindi+I' + 1(++coods) CONSIDERING ONLY THE COEFFICIENTS OF K, WE OBTAIN EM, = I f (++cost)-I'+ & sind + I+ & sind =0 BUT THE SECOND AND THIRD TERMS CANCEL OUT, DUE TO THE AXISYMMETRY OF THE TOP.

IM = I & (++ + coso) = 0 I (++ cos 0) = B

WHERE A IS A CONSTANT (b) FROM ER. (A) WE HAVE W = Y++ COSO AND, IN VIEW OF (2):

Wz = P/I = constant SUBSTITUTING FOR I (++ + cost) FROM (2) INTO (1): I'+ sin'0 + B cos b = a \$ = 8-BC000 (FUNCTION OF θ) (5) I' sin'B

* 18.140

GIVEN: TOP OF PROB. 18.139

SHOW THAT:

(a) A TITIRD EQUATION OF MOTION CAN P.E OBTHINED FROM THE PRINCIPLE OF CONSERVATION OF ENERGY.

(b) BY ELIMINATING & AND Y FROM THAT EQUATION AND EQS(1) AND(2) OF PROB. 18,140 AN EQUATION &= f(8) CAN BE OBTHINED, WHERE

$$f(\theta) = \frac{1}{l'} \left(2E - \frac{\beta^a}{l} - 2mgc\cos\theta \right) - \left(\frac{\alpha - \beta\cos\theta}{l'\sin\theta} \right)^a$$
 (1)

(C) BY INTRODUCING THE VARIABLE 2 = cost, THE MAX. AND MIN. VALUES OF & CAN BE OBTAINED BY SOLVING THE CUBIC EQUATION

$$\left(2E - \frac{\beta^{2}}{l} - 2mgcx\right)(1 - x^{2}) - \frac{1}{l'}(\alpha - \beta x)^{2} = 0$$
 (2)

a) CONSERVATION OF ENERGY

 $T = \frac{1}{7} \left(I' \omega_{\mu}^{2} + I' \omega_{\mu}^{2} + I \omega_{\mu}^{2} \right)$ REFERRING TO EQ. (A) OF PROB. 18. 139:

T=+[I'+'sin'8+I'8+ I(+++cos 8))],

T+V = E: +[I'+ sin 0+ I'+ I(V++ cos 0)]+mgccos 0=E

(6) SUBSTITUTING IN (6) FOR + FROM EQ. (5) OF PROB. 18.3! AND FOR (4+4 COSD) FROM RW. (2) OF PROB. 18. 139, AND MULTIPLYING BY Z:

 $I'\left(\frac{\partial -\beta \cos \theta}{1'\sin^2 \theta}\right) \sin^2 \theta + J'\dot{\theta}' + I\left(\frac{\beta}{I}\right)' + I \operatorname{mg} c \cos \theta = 2E$

$$\frac{(\alpha - \beta \cos \theta)^{2}}{I' \sin^{2} \theta} + I' \dot{\theta}^{2} + \frac{\beta^{2}}{I} + 2 mig \cos \theta = 2E$$

SOLVING POR O, WE OBTAIN

WHERE

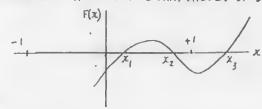
$$f(\theta) = \frac{1}{l'} \left(2E - \frac{\beta^2}{l} - 2mgc \cos \theta \right) - \left(\frac{\alpha - \beta \cos \theta}{l' \sin \theta} \right)^2$$

(C) SETTING COSO = Z $J(\theta) = \frac{1}{T_1} \left(2 E - \frac{\beta^2}{L} - \lambda \log c \lambda \right) - \frac{(\alpha - \beta z)^2}{L^{12} (1 - z^2)}$

LETTING 5(0)=0 AND MULTIPLYING BY I' (1-2), WE OBTAIN THE CUBIC EQUATION F(2)=0:

$$\left(2E - \frac{\beta^2}{l} - 2mgcx\right)(1 - x^2) - \frac{1}{l'}(\alpha - \beta x)^2 = 0$$

SOLVING THIS EQUATION WILL YIELD THREE VALUES OF X. THE TWO VALUES COMPRISED BETWEEN - I AND +1 CORRESPOND TO THE MAX. AND MIN. VALUES OF B.



18.141 and * 18.142

GIVEN: SOLID CONE. INITIALLY, 0 = 30, 0 = 0 1 = 300 rad/s. USING EQ.(2) OF PROB, 18. 140 AND PROBLEM 18.141: KNOWING THAT &= 20 rouls FIND: (a) Pray, (b) CORRESPONDING Y AND & PROBLEM 18, 142:

KNOWING THAT \$=-4 red/s FIND: (a) Oman, (b) CORRESPONDING WAND + (C) VALUE OF & FUR WHICH

SENSE OF 4 15 REVERSED WE FIRST DETERMINE THE FOLLOWING CONSTANTS: $I = \frac{3}{10} \text{ m b}^3 = \frac{3}{10} \text{ m} (0.25 \text{ ft})^3 = (18.75 \times 10^3 \text{ ft}^2)^{10}$ I' = 3 m (+22+ h2) = 3 [+ (0.25f)+(0.75f)] m = (346, 875 x 10 3 ft 3) m c = AG = 3 h = 3 (0,75 ft) = 562.5 x 10 ft

L'EXT, WE DETERMINE THE CONSTANTS B, Q, ALLE FRUM EUS. (2) AND (4) OF THE SOLUTION OF PROB. 18. 139 AND FROM EQ.(6) OF THE SOLUTION OF PROB. 18. 140, USING THE APPROPRIATE INITIAL CONDITIONS.

PROBLEM 18,141 B= I(Yo++0000)=(18.75×103)m(300+20 cos 30) = (5,94976)m a = I' + sin26 + 13 cos 0 = (346.875 x103)m(20) sin230° t +(5,94976)m cos 30 = (6,88702) m

E = 1 [1' + sin' 0 + I' + I (+ + + w + b)] + mg c cos 0 = \frac{1}{2} \left\{ (346, 875 \times 10^3) m (20) \sin 30 + 0 + \frac{(5, 94976 in)}{18.75 \times 10^3 m} \right\} + m (32,2)(562,5 × 10 3) cos 30 = (977, 020) m

SUBSTITUTE IN EQ. (2) OF PROB. 18,140: (2E-B'-2mgcz)(1-x')- 1, (a-Bz) = 0 (66.0593-36.225x)(1-23)-2.88288(6.88702-5.949762)=0 (a) SOLVING: Z = 0.743151 Balay = 42.0°

(b) ER.(5) OF PROB. 18.139: $\phi = \frac{\alpha - \beta \cos \theta}{L^2 \sin^2 \theta} = \frac{6.88702 - 5.94476 \cos 42.0}{(346.875 \times 10^2 3) \sin^2 42.0^2} = 15.8748 \text{ cod/s}$

FROM EQ.(2): 1 = 1 - + cos 0 = 5.44476 - (15.8748) cos 42.0°

#= 306 rad/s; \$= 15.87 rad/s PROBLEM 18,142

B= I(\$ + \$, cos 0,) = (18,75 × 10) m (300 - 4 cos 30) = (5,56005)m

E= = [1'40 sin' 00 + 1' 00 + 1 (16+40 coso)] + mgccus 00 = 1/ (346.875 x103) m (-4)25in30+0+ (5,56005m)2]-18.75×10-3m + m (32,2)(562.5 × 10-3) cos 30" = 8 +0.76 m

SUBSTITUTE IN EQ. (2) OF PROB. 18. 140: (2E-\$ -2mgcx)(1-x)-+ (a-12)=0 (32.765-36.225 x)(1-x1)-2,88288(4,46827-5,56005x)=0

(d) SOLVING: X = 0.37166, PAR = 68.18, PAR = 68.2° (b) FQ.(5): \$\phi = \frac{\alpha - \beta \cos\theta}{I'\sin'\theta} = \frac{\kappa + \beta \cos\theta}{346.875 \kappa \cos\theta} \frac{\kappa \cos\theta}{3} \frac{\kappa FO.(2): \P=(B/I) - \$ \cos = 296.536 - 8.0335 cos 68.18 = 293,55 64 V= 294 rad/s; += 8.03 rad/s

(c) + REVERSES FOR α-15 cos θ = 0, cos θ = 1.46827, θ = 36.5°

18.143



GIVEN:

RIGID BODY OF ARBITRARY SHAPE SUPPORTED AT ITS MASS CENTER OF AND SUBJECTED TO NO FURCE (EXCEPT AT SUFFIKT O).

SHOW THAT:

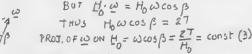
(a) Ho = constant (IN MAGNITIME A NIR") T = contant PROJ. OF W ALONG H = constant (b) TIP OF W DESCRIBES CHAVE ON FIXED PLANE (THE INVARIABLE FIRME") PEPP. TO HO AND AT DISTANCE ZT/HO FROM 0. (C) WITH RESPECT TO PRINCIPAL AXES Day & ATTACHED TO LODY, W

AFREMIS TO DESCRIBE A CURVE ON ELLIPSOID OF EQUATION I 4 + 1, 43 + I, 42 = 2T (POINSOT ELLIPSOID)

(a) FROM EG. (18. 27): EM = H SINCE EM = 0: . Ho = constant

T+V=const.; SINCE V= const, T=constant (2)

WERECALL FROM PROB. 18.37 THAT HOW = 2T

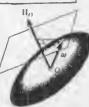


(b) IT FOLLOWS FROM (3) THAT THE TIP OF W MUST REMAIN IN A PLANE I HA ATIA DISTHIKE 2T/HA PROM O.

(C) FROM EW. (18.20): $T = \frac{1}{2} \left(I_{\lambda} \omega_{\kappa}^{2} + I_{\mu} \omega_{\mu}^{2} + I_{2} \omega_{2}^{2} \right)$ FROM (2) IT FOLLOWS THAT], W +], W + I, W = 2T = const.

EU. (4) IS THE EQUATION OF AN ELLIPSOLD ON WHICH THE TIP OF W MUST LIE. THIS IS POINSOT ELLIFSCID. COMPARING EQ. (4) WITH EW. (9,49) OF SEC. 9, 17, WE NOTE THAT POINSUT ELLIPSOID HAS THE SAME SHAPE AT THE ELLIPSOID OF INERTIA OF THE BODY BUT A DIFFERENT SIZE.

18.144



GIVEN:

POINSOT ELLIPSOID AND INVALIABLE PLANE DEFINED IN PROB. 18.143. SHOW THAT:

(4)

(a) THE ELLIPSOID IS TANKENT TO THE PLANE.

(b) AS THE BODY MOVES THE POINSOT ELLIPSOID ROLLS ON THE MYACIABLE PLANE.

(a) AT THE TIP OF W THE DIRECTION OF THE HONING TO THE ELLIPSOID IS THAT OF grad F(W, W, D), WHERE F DENOTES THE LEFT-HAND MEMBER OF EQ. (4) OF PROB. 18.143. FROM SEC. 13,7: grad F = 2F i + 2F j + 2F t

=2 I, w, L+2 I, w, J+2 I w, k = 2 (1, 0, 1+1, 0, 3+1, 0, 1) = 2 H

THUS, THE WORLD TO POINSUT ELLIPSOID IS PARALLEL TO HO, IT FULLOWS THAT POINSOT ELLIBOID IS TANGENT TO THE INVARIABLE PLANE

(CONTINUED)

* 18.144 continued

(b) THE POINSOT ELLIPSOID IS PART OF THE BODY WHOSE MOTION IS BEING ANALYZED, AND ITS POINT OF CONTACT WITH THE INVPRINELE PLANE IS THE TIP OF THE VECTOR W. SINCE W DEFINES THE INSTANTANEOUS AXIS OF ROTATION, THE POINT OF CONTACT HAS ZERO VELOCITY, THUS, THE POINSOT ELLIPSOID ROLLS ON THE INVARIABLE PLANE (WITH ITS CENTER O REMAINING FIXED).

* 18.145 GIVEN!

AXISYMMETRICAL RIGID BODY SUPPORTED AT ITS MASS CENTER O AND SUBJECTED TO NO FURCE (EXCEPT AT SIPPORT O).

USING THE RESULTS OBTAINED IN PRODS. 18.143-144. SHOWTHAT THE POINSOT ELLIPSUID IS AN ELLIPSUID OFREVOLUTION AND THE SPACE AND RODY CONES ARE BOTH CIRCULAR AND TANGENT TO EACH OTHER. FURTHER SHOW THAT

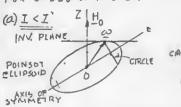
(a) THE TWO COMES ARE TANGENT EXTERNALLY AND THE PRECESSION IS DIRECT WHEN I (I' WHERE I = MOM OF INERTIA ABOUT AXIS OF SYMMETRY

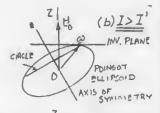
I' = - - - TRANSVERSE AXIS (b) THE SPACE CONE IS INSIDE THE ENDY CONE ALL THE PRECESSION IS RETROGRADE WHEN I > 1'.

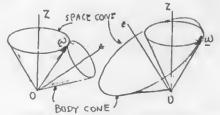
CHOOSING & ALONG THE AXIS OF SYMMETRY, WE HAVE I = I = I AND I = I . SUBSTITUTE INTO (4) OF PIB. 143: $I'(\omega_{\lambda}^2 + \omega_{y}^2) + I\omega_{\lambda}^2 = \text{const.}$

WHICH IS THE EQUATION OF AN ELLIPSUID OF REVOLUTION. IT FOLLOWS THAT THE TIP OF W DESCRIBES CIRCLES ON BOTH THE POINSOT ELLIPSOID AND THE INVARIABLE ELLIPSOIDS IN THE STANDARD FORM PLANE, AND THAT THE VECTOR W ITSELF DESCRIPTS CIRCULAR BODY ATTO SPECE CONES

THE POINSOT ELLIPSOID, THE INVARIABLE PLANE AND THE BODY AND SPACE CONES ARE SHOWN BELOW FUR CASES a AND b:







DIRECT PRECESSION

RETROGRADE PRECESSION

* 18.146



GIVEN:

RIGID BODY OF ARBITRARY SHAPE PHD ITS POINSOT ELLIPSOID (CF. PROBS. 18,143 AND 18,144. SHOW THAT:

(a) CURVE DESCRIBED BY TIP OF W ON POINSOT ELLIPSOID 15 DEFINED BY

 $I_x \omega_x^2 + I_y \omega_y^2 + I_4 \omega_x^2 = 2T = \text{constant}$ (1) $I_x^2 \omega_x^2 + I_y^2 \omega_y^2 + I_x^2 \omega_x^2 = H_O^2 = \text{constant}$ (2)

AND CAN THUS BE OBTAINED BY INTERSECTING THE POINSOT ELLIPSOID WITH THE ELLIP SOID BEFINED BY (2)

(b) ASSUMING I,> I, THE CURVES (CALLED POLHODES) OBTAINED FOR VARIOUS VALUES OF HO HAVE THE SHAPES INDICATED IN . FIGURE

(C) THE BODY CAN ROTATE ABOUT A FIXED AXIS ONLY IF THAT AXIS COINCIDES WITH ONE OF THE PRINCIPAL AYES, THIS MOTION BRING STABLE IP THE AXIS IS THE MAJOR OR MINER AXIS OF THE POINS OF ELLIPSOID (X OR & AXIS) AND UUSTABLE IF IT IS THE INTERMEDIATE AXIS (& AXIS).

(a) Eq.(1) IN STATEMENT EXPRESSES CONSERVATION OF ENERGY; THIS IS EQ. (4) OF PROB. 18. 143. WE NOW EXPRESS THAT THE MAGNITUDE OF A IS CONSTANT:

 $H_0^2 = H_X^2 + H_Y^2 + H_Z^2 = I_X^2 \omega_X^2 + I_Y^2 \omega_Y^2 + I_Z^2 \omega_Y^2 = \text{const.}$ WHICH IS EO (2) IN STATEMENT. SINCE THE COORDINATES WILLY, WI OF THE TIP OF W HUST SHTISFY BOTH EOS. (1) AND (2), THE CURVEDESCEIBED BY THE TIP OF W IS THE INTERSECTION OF THE TWO ELLI PSOIDS.

(6) WE NOW WRITE THE EQUATIONS OF THE TWO

$$\frac{z^2}{a^3} + \frac{b^2}{b^3} + \frac{z^3}{c^4} = 1$$

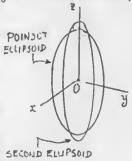
WHERE A, b, C ARE THE SEMPAXES OF THE ELLIPSOID.

WE HAVE POR POINSOT ELLIPSOID:

Wx + Dx + W/I, + Ho/I, + Ho/I, FOR SECOND ELLIPSOID:

SINCE WE ASSUMED THAT I, > I, , WE HAVE 27/1, < 27/1, < 27/1, AND Ho/1 < Ho/1, < Ho/1,

THUS, FOR BOTH ELLIPSOIDS, THE MINOR ANIS IS DIRECTED ALONG THE KAXIS, THE INTERMEDIATE AXIS ALONG THE. A AXIS, AND THE MAJOR, AXIS ALONG THE & AXIS.

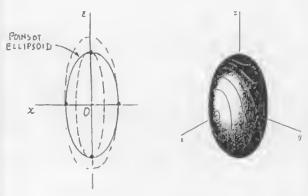


HOWEVER, BECAUSE THE RATIO OF THE MAJOR TO THE MINOR SEMIAXIS IS YI/I, POR THE POINSOT ELLIPSOID AND INTE FOR THE SECOND ELLIPSOID, THE SHAPE OF THE LATTER WILL BE MORE " PROHOUNCED".

(CONTINUED)

* 18.146 continued

THE LARGEST ELLIPSOID OF THE SECOND TYPE TO BE IN CONTACT WITH THE POINSOT ELLIPSOID WILL BE OUTSIDE THAT ELLIPSOID AND TOUCH IT AT ITS POINTS OF INTERSECTION WITH THE x AXIS, AND THE SMALLEST WILL BE INSIDE THE POINSOT ELLIPSOID AND TOUCH IT AT ITS POINTS OF INTERSECTION WITH THE z AXIS (SEE LEFTHAND SKETCH) ALL ELLIPSOIDS OF THE SECOND TYPE COMPRISED BETWEEN THESE TWO WILL INTERSECT THE POINSOT ELLIPSOID ALONG THE POLHODES AS SHOWN IN THE RIGHT-HAND FIGURE.

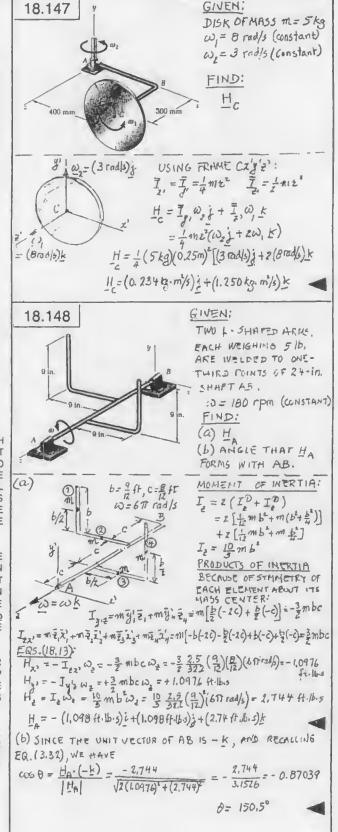


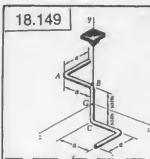
NOTE THAT THE ELLIPSOID OF THE SECOND TYPE WHICH HAS THE SAME INTERMEDIATE AXIS AS THE POINSOT ELLIPSOID INTERSECTS THAT ELLIPSOID ALONG TWO ELLIPSES WHOSE PLANES CONTAIN THE y AXIS. THESE CURVES ARE NOT POLHODES, SINCE THE TIP OF \$\omega\$ WILL NOT DESCRIBE THEM, BUT THEY SEPARATE THE POLHODES INTO FOUR GROUPS: TWO GROUPS LOOP AROUND THE MINOR AXIS (x AXIS) AND THE OTHER TWO AROUND THE MAJOR AXIS (z AXIS).

(c) IF THE BODY IS SET TO SPIN ABOUT ONE OF THE PRINCIPAL AXES, THE POINSOT ELLIPSOID WILL REMAIN IN CONTACT WITH THE INVARIABLE PLANE AT THE SAME POINT (ON THE x, y, OR z AXIS); THE ROTATION IS STEADY IN ANY OTHER CASE, THE POINT OF CONTACT WILL BE LOCATED ON ONE OF THE POLHODES AND THE TIP OF WILL START DESCRIBING THAT POLHODE, WHILE THE POINSOT ELLIPSOID ROLLS ON THE INVARIABLE PLANE.

A ROTATION ABOUT THE <u>MINOR</u> OR THE <u>MAJOR</u> AXIS (x OR z AXIS) IS <u>STABLE</u>: IF THAT MOTION IS DISTURBED, THE TIP OF $\underline{\omega}$ WILL MOVE TO A VERY SMALL POLHODE SURROUNDING THAT AXIS AND STAY CLOSE TO ITS ORIGINAL POSITION.

ON THE OTHER HAND, A ROTATION ABOUT THE INTERMEDIATE AXIS (z AXIS) IS UNSTABLE: IF THAT MOTION IS DISTURBED, THE TIP OF W WILL MOVE TO ONE OF THE POLHODES LOCATED NEAR THAT AXIS AND START DESCRIBING IT, DEPARTING COMPLETELY FROM ITS ORIGINAL POSITION.AND CAUSING THE BODY TO TUMBLE.





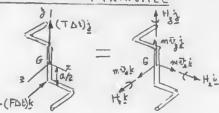
ROD OF MASS on 15 HIT AT C IN NEGATIVE & DIRECTION. IMPULSE = - (F Dt) k .

FIND!

IMMEDIATELY AFTER IMPACT (a) ANG, YELDCITY OF ROD,

(b) VEWCITY OF G.

MPULDE -



(WEIGHT IS OMITTED, SINCE HONIMPOLSIVE)

(a) ANGULAR VELOCITY

MOHENTS AND PRODUCTS OF INTERTIA:

$$I_{d2} = \frac{m}{5} \left(\frac{4}{2} \right) - \frac{a}{1} + \frac{m}{5} \left(-\frac{a}{2} \right) \frac{4}{3} = -0.1 \text{ ma}^{2}$$

$$I_{e2} = \frac{m}{5} \left(-\frac{a}{2} \right) - a + \frac{m}{5} \left(\frac{a}{2} \right) a = 0.2 \text{ ma}^{2}$$

EOS. (18,17) AND DIVIDING BY Ma"

$$\overline{H_{k}} = \overline{\underline{I}}_{k} \underline{\omega}_{k} - \overline{\underline{I}}_{k} \underline{\omega}_{k} - \overline{\underline{I}}_{k} \underline{\omega}_{k} : \quad \underline{\underline{F}\Delta t}_{2ma} = 0.35 \underline{\omega}_{k} + 0.3 \underline{\omega}_{3} - 0.2 \underline{\omega}_{k}$$
 (1)

$$H_{y} = -I_{y} \omega_{x} + I_{y} \omega_{y} - I_{y} \omega_{z}; \quad 0 = 0.5 \omega_{y} + \frac{2}{5} \omega_{y} + 0.1 \omega_{z}$$
 (2)

$$H_{s} = -\bar{\mathbf{I}}_{s} \cdot \hat{\mathbf{J}}_{s} - \bar{\mathbf{I}}_{y} \cdot \hat{\mathbf{J}}_{y} + \bar{\mathbf{I}}_{z} \cdot \hat{\mathbf{J}}_{z}; \qquad 0 = -\alpha \mathcal{E}_{s} + \alpha \mathbf{I} \cdot \hat{\mathbf{J}}_{y} + 0.75 \omega_{z}$$

SOLVING POS. (1), (2), (3) SIMULTHNEOUSLY:

$$\omega_2 = \frac{30}{6} \frac{F\Delta t}{TM}$$
 $\omega_1 = -\frac{15}{6} \frac{F\Delta t}{TM}$ $\omega_2 = \frac{10}{6} \frac{F\Delta t}{TM}$

THIS: 10 = FAT (301-15 + 10 K)

(b) VELOCITY OF G

WE PIRST NOTE THAT THE GIVEN CONSTRAINTS REQUIRE THAT U. - U. EQUATING THE COMPONENTS OF IMPULSE AND MOMENTUM!

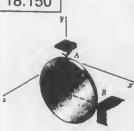
I COLIP: 0= mv,

I COMP.:
$$0 = \pi i \bar{v}_x$$
 $\bar{v}_y = 0$ $T\Delta t = 0$

$$\vec{v_2} = -\frac{F\Delta t}{4M}$$

THEREFORE

18.150



GIVEN:

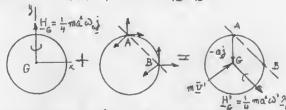
DISK OF MASS M SUPPORTED BY BALL AND SOCKET AT A ROTATES WITH CONSTANT W= 0, J WHEN OBSTRUCTION IS INTRODUCED AT B. IMPACT PERFECTLY PLASTIC (e=0).

FINDI

IMMEDIATELY AFTER IMPACT (a) ANGULAR VELOCITY OF DISK. (b) VELOCITY OF G.

IMPULSE-MOMENTUM PRINCIPLE

WE NOTE THAT IDIAM = 4 ma AND 3AB = TE (1-1)



WE NOTE THAT T'= W'X AG = W'AABX(-ag) = - 1 Wak

(a) EQUATE MOVENTS ABOUT AB OF ALL VECTORS AND COUPLES! 2 AD . H6 + 0 = 3 AB . (-aj x m v') + 3 AB . H6 1/2 (i-i) · 1 maw, i = 1/2 (1-i) · [-ai × (-1/2 mwak)] + 2 · H.

$$-\frac{1}{4V\bar{\epsilon}} ma^2 \omega_0 = \frac{1}{2} ma^2 \omega' + \frac{1}{4} ma^2 \omega' \\ \omega' = -\frac{1}{3V\bar{\epsilon}} \omega_0$$

$$\underline{\omega}' = \omega' \ \underline{\lambda}_{AB} = -\frac{1}{3\sqrt{2}} \omega_0 \sqrt{\frac{1}{2}} (\underline{i} - \underline{j}), \quad \underline{\omega}' = \frac{1}{6} \omega_0 (-\underline{i} + \underline{j})$$

(b) RECALLING THAT U'= W'X AG

$$\underline{\vec{v}}' = \frac{1}{6}\omega_0 (-i+j)\dot{x}(-aj)$$
 $\underline{\vec{v}}' = \frac{1}{6}\omega_0 ak$

GIVEN: 18.151

DISK OF PROB. 18, 150

FIND:

KINETIC ENERGY LOST WHEN DISK HITS OBSTRUCTION.

BEFORE IMPACT:

$$T_0 = \frac{1}{2} I_{\text{JIAM}} \omega_0^{\perp} = \frac{1}{2} \frac{1}{4} \text{ma} \omega_0^{\perp} = \frac{1}{2} \text{ma}^2 \omega_0^{\perp}$$

AFTER IMPACT:

BUT, FROM ANSWERS TO PROB. 18. 150:

$$v'^{2} = (\frac{1}{6}\omega_{0}a)^{2} = \frac{1}{36}\omega_{0}^{2}a^{3}$$

$$\omega'^{2} = \omega_{2}^{12} + \omega_{3}^{22} = \frac{\omega_{0}}{36}(1+1) = \frac{1}{18}\omega_{0}^{2}$$

THEREFORE:

$$T' = \frac{1}{7}m(\frac{1}{36}\omega_0^2c') + \frac{1}{7}\frac{md}{4}(\frac{1}{18}\omega_0^2) = \frac{1}{48}ma\omega_0^2$$

KINETIC ENERGY LOST

= \frac{5}{48} maw



TRIANGULAR PLATE OF MASS M WELDED TO SHAFT SUPPORTED BY REARINGS AT A AND B. PLATE RUTATES AT CONSTANT

DYNAMIC REACTIONS AT A AND B.

COMPUTATION OF MONEY AND PRODUCT OF INTESTA

FROM BACK COVER:

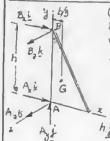
$$(I_g)_{AREA} = \frac{1}{12}b^3h, \quad A = \frac{1}{2}bh, \quad (I_g)_{ARSS} = \frac{1}{12}A_{ARSA} = \frac{11}{12}b^3h \left(\frac{m}{2bh}\right) = \frac{1}{6}mb^3$$

FROM SAMPLE PROC. 9.6 (PAGE 485 OF STATICS), $(I_{2y})_{AEEQ} = \frac{1}{24} b^2 h^2 (I_{2y})_{AEEQ} = (I_{3y})_{AEEQ} \frac{m_1}{R} = \frac{1}{24} b^2 h^2 (\frac{m}{Fbh}) = \frac{1}{12} mbh$

WE ALSO NOTE THAT I = 0

ANGULAR MOMENTUM HA SINCE W = 0, W = W, W = U, EW. (18.13) YIELD H = - I, wy = - 17 1 bhw, H = I, wy = 6 mbw, H = 0 H=-12 mbhwi + 5 mbwj

EQUATIONS OF MUTION



(WEIGHT INHITTED FOR DYNAMIC LEACTIONS) FRAME OF REPERENCE ARMY ROTATES WITH D= W= Wig.

EQ.(18,28):

RECALLING (1) AND COMPUTING EMA: $h_{j} \times (B_{x}i + B_{x}k) = \omega_{j} \times (-\frac{mbh}{(2}\omega_{i} + \frac{mb}{6}\omega_{d})$ - h & k + h & i = 1 mbhw k

EQUATING THE COLFT. OF THE UNIT NECTORS:

$$B_{\lambda} = -\frac{1}{12} \ln b \omega^{3} \qquad B_{\alpha} = 0$$

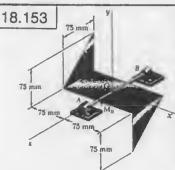
$$B = -\frac{1}{12} \ln b \omega^{2} \underline{i}$$

五Q. (18.1):

THUS:

$$\underline{A} + \underline{B} = -\frac{1}{3}b\omega^{2}i$$

$$\underline{A} = -\frac{1}{3}b\omega^{2}i - (-\frac{1}{12}mb\omega^{2}i)$$



GIVEN:

SHEET - METAL COMPONENT OF MASS m = 600g. LENGTH AB = 150 mm . COMPONENT AT REST WHEN M = (49,5 mN·m) k 15 A PPLIED.

IND: DYNAMIC REACTIONS AT A AND B (a) JUST AFTER COUPLE IS APPLIED (b) 0.65 LATER

b=0.075m MOMENT AND PRODUCTS OF INERTIA RECTANGLE Z: MASS = 2 m $I_2 = \frac{1}{12} \left(\frac{2}{3} m \right) (2b)^2 = \frac{2}{9} m b$ I2 = 7 = 0 TRIANGLE 1: MASS = 7 M FROM BACK COVER: $(\bar{I}_z)_{AECA} = \frac{1}{36}b', A = \frac{1}{2}b', (\bar{I}_z)_{MASS} = (\bar{I}_z)_{AECA} \frac{m/6}{A}$ $(\bar{I}_{2})_{MASS} = \frac{1}{36}b''(\frac{2m}{16}) = \frac{1}{100}mb'$

FROM SAMPLE PROB. 9.6 (PAGE 4858FSTATIG) (] 44 AKER = - 1/2 b", (] 47 MASS = - 1/2 b" (6 m) = - 1/2 l" (

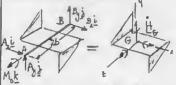
7HEREFORE: $I_2 = I_2 + \frac{m}{6}d^2 = \frac{1}{100}mb^4 + \frac{m}{6}[b^4 + (\frac{b}{3})^4] = (\frac{1}{100} + \frac{10}{54})mb^4 = \frac{7}{36}mb^4$ $I_{32} = \bar{I}_{12} + \frac{m}{6} \bar{y} \bar{z} = -\frac{1}{216} m b^2 + \frac{m}{6} (\frac{b}{3})(-\frac{b}{6}) = -\frac{1}{72} m b^3$ In= I+ + m I== 0+m (-b)(-b) = 1 mb

TRIANGLE 3: BY JYMHETRY, SAHE AS TRIANGLE 1.

FOR ENTIRE COMPONENT! I = = = n16+ 2 (7 mb2) = 11 mb2 $\vec{I}_{32} = 2(-\frac{1}{72}mb^2) = -\frac{1}{36}mb^2$ I = 2 (36 mb) = 1 mb

ANGULAR MOMENTUM Ho RUS.(18.7) WITH Wz=Wy=0, Wz=W: $H_{z} = -\frac{1}{12}\omega_{z} = -\frac{1}{18}mb^{2}\omega, H_{z} = -\frac{7}{15}\omega_{z} = \frac{1}{36}mb^{2}\omega, H_{z} = \frac{7}{18}mb^{2}\omega$ HG = 1/36 mbw (-2i+1+22k) (1)

EQUATIONS OF MOTION



ER. (18.22) AND USING (1): H = (H) + 12 xH = 1 mb a (-2i+j+22+)+ 0kx 1m + w(-zi+j+22k)

EQUATING MUMIENTS ABOUT B .:

2bkx(Ai+Aji)+Msk = 1 mb' (-2i+j+2k) (-(2j+i))) 26 P. j - 26 1 1 + M. K = - 10 mb2 [- (20x + 0.) + + (0x - 20) + 220x]

EDUATING THE COEFT OF THE UNIT VECTORS:

(1) $-2bA_{y} = -\frac{1}{36}mb^{2}(2\alpha+\omega^{2})$

A= 72 mb (20+02) (2)

(3)

(4)

1 26 Az = 36 mb (x-20)

Az= 1/2 1116 (x-20)

1) 10 = 11 mb x a = 16Mo/11mb (CONTINUED)

18.153 continued

WE RECALL THE RESULTS OBTAINED: $A_3 = \frac{1}{72} mb (2\alpha + \omega^2)$ $A_2 = \frac{1}{72} mb (\alpha - 2\omega^2)$

 $x = 18 \, \text{Mp/H mb}^2 \tag{4}$

WITH GIVEN DATA: M = 0.0495 N·m, 111 = 0.6 kg. b = 0.075m:
EQ (4): X = 18 (0.0495)/11(0.6)(0.675) = 24 rad/s2

 $\begin{array}{ll}
\mathcal{R}_{0}(3): & A_{\lambda} = \frac{1}{72} (0.6)(0.075)(24 - 210^{3}) = (15 - 1.25\omega^{2})10^{3}N & (3') \\
= 0.(2): & A_{\lambda} = \frac{1}{72} (0.6)(0.075)(2 \times 24 + \omega^{2}) = (30 + 0.625 \omega^{2})10^{3}N & (2')
\end{array}$

(a) JUST AFTER COUPLE IS A FTELED: LETTING (0=0 IN (3') HID (2'): Az = 15x10 N, Az = 30 x10 N

THUS:

A = (15,00 mN) i + (30.0 mN) i

ZF=m@: A+B=0, B=-(15.00mN)i-(30.0mN)j

(b) AFTER D.65:

IF=mā: A+B=0, B=(244mH)i-(159.6mN)j

18,154

GIVEN!

RING ATTACHES BY COLLAR AT A TO VERTICAL SUPFI ROTATING AT CONSTANT RATE ω .

FIND:

(a) CONSTANT AND R B THAT PLANE OF RING FORMS WITH VERTICAL WHEN W = 12 md/s, (b) MAX, VALUE OF AD FOR WHICH B = 0.



ANGULAR MOMENTUM HE
USING THE PRINCIPAL MYES GZYZ WITH

X PERPENSICILLIR TO PLANE OF RING:

H = I \(\int_{\int} \cdot + I \) \(\int_{\int} \cdot + I \) \(\int_{\int} \cdot \)

= mt \(\int_{\int} \sin \beta \cdot + I \) \(\int_{\int} \sin \beta \cdot \)

H = mt \(\int_{\int} \sin \beta \cdot + I \) \(\int_{\int} \sin \beta \cdot \)

H = mt \(\int_{\int} \sin \beta \cdot + I \) \(\int_{\int} \sin \beta \cdot \cdot \)

A PARTIONS OF MOTION

A PARTIE TO THE TOTAL PROPERTY OF THE PARTIE TO TH

Z= 2 SING Z= 2 COSP EQUATING MOMENTS ABOUT A:

*) $W\bar{z} = (m\bar{z}\omega^3)\bar{g} + H_0$ $mgz\sin\beta = m(z\sin\beta)\omega^2(z\cos\beta) + \frac{1}{2}mz^2\omega^3\sin\beta\cos\beta$ $g = \omega^2z\cos\beta + \frac{1}{2}\omega^3z\cos\beta$ $\cos\beta = \frac{2}{3}\omega^2$ (2)

(a) LETTING g = 32.2 ft/s, t=0.25 ft, w = 12 rod/s:

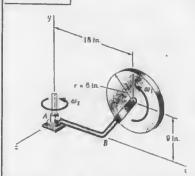
 $\cos \beta = \frac{2}{3} \frac{32.2}{(0.25)(12)^2} = 0.99630$ $\beta = 53.4$ °

(b) JOLUING EQ.(2) FOR ω^* AND LETTING $g = 32.24t/s^*$, z = 0.25 + t, $\beta = 0$: $\omega^* = \frac{2(32.2)}{32} = \frac{2(32.2)}{3(0.25)} = 85.87$ $\omega = 9.27 \text{ rad/s}$

18.155

(z)

(3)



GIVEN:

10-16 DISK ROTATES
AT CONSTANT RATE $\omega_1 = 15 \text{ rad/s}.$ ARM ABC ROTATES
AT CONSTANT RATE $\omega_2 = 5 \text{ rad/s}.$ FIND:

FURCE-COUPLESYSTEM

REPRESENTING THE

DYNAMIC REACTION
AT SUPPORT A.

ANGULFR HUNFHTUM OF DISK ABOUT C.



USING THE PRINCIPAL CENTROICHLAXES CX'y'e': $\omega = \omega_1 j + \omega_1 E$ ANG. VELOCITY OF FRAME CX'y'e':

 $H_{c} = \overline{I}_{2}\omega_{2}, \underline{i} + \overline{I}_{3}\omega_{3}, \underline{i} + \overline{I}_{4}\omega_{4}, \underline{k}$ $= 0 + \frac{1}{4}m z^{2}\omega_{2} \underline{i} + \frac{1}{2}mz^{2}\omega_{1} \underline{k}$ $H_{c} = \frac{1}{4}mz^{2}(\omega_{2}\underline{j} + 2\omega_{1}\underline{k}) \qquad (1)$

RATE OF CHANGE OF EC

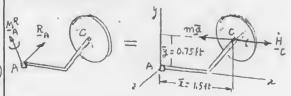
EQ.(18,22) AND USING (1): $H = (H_c)_{CX'g'z'} + \Omega \times H_c = 0 + \omega_z \dot{j} \times \dot{j} \text{ ant}(\omega_z \dot{j} + 2\omega_z \dot{k})$ $H = \dot{j} \text{ ant}(\omega_z \omega_z \dot{k})$

WITH GIVEN DATA:

 $\dot{H}_{c} = \frac{1}{2} \frac{101b}{32.217/s^{2}} (\frac{1}{2} ft)^{2} (15 rad/s) (5 rad/s) \dot{c} = (2.9115 /b.ft) \dot{c}$

 $\frac{COMPUTATION OF m2}{\bar{a} = -\bar{x}\omega_z^2 \bar{i} = -(1.5ft)(5 md/s)^2 i = -(37.5 ft/s^2) i$ $m\bar{a} = \frac{10 \text{ lb}}{32.7 \text{ ft/s}} \cdot (-37.5 \text{ ft/s}^2) i = -(11.646 \text{ lb}) i$

EQUATIONS OF MUTION



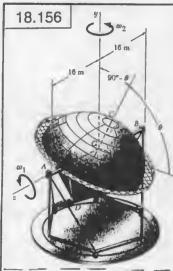
IF = I Feff: R = ma = - (11.646 16) i

 $\sum_{A} = \sum_{A} (M_{A})_{eff};$ $M^{R} = \frac{\dot{H}}{c} + (\bar{z}_{i} + \bar{y}_{j}) \times m\bar{a}$ $= (2.911516.ff) \dot{i} + [(1.5ff) \dot{i} + (0.75ff) \dot{j}] \times (-11.6ff)$

= (2,9115 16.ft) \(\bar{\text{L}} + \begin{bmatrix} (1.5ft) \(\bar{\text{L}} + \begin{bmatrix} (0.75ft) \(\bar{\text{L}} \begin{bmatrix} \begin{bmatrix} -11.6ft \(\beta \begin{bmatrix} \begi

FORCE-COURLE SYSTEM AT A:

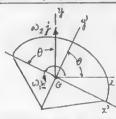
RA = - (11.65 16) i; MR = (2,91 16.ft) i+(8.73 16.ft) k



SOLAR-ENERGY CONCEN-TRATOR: m = 30 Mg RADII OF GYRATION ABOUT CD: K = 12 m ABOUT AB: R' = 10 m W, = 0.20 rad/s (constant) w. = 0.25 rad/s (constant)

FIND FOR B = 60°: (a) FURCES EXERTED ON CONCENTRATOR AT A AND B. (b) COUPLE M. E APPLIED TO CONCEN-TRATOR AT THAT INSTANT.

ANGULAR MOMENTUM ABOUT G



USING THE PRINCIPAL AFES w=-w, cost, a, -w, sind, w, = a, H = I.O. 2+ I.O. 3+ IO. 5 H = -IW, caso i + IW, sind) + IW, t WHERE I= mk AND I'= mk'

WE NOW RETURN TO THE REFERENCE FRAME GAYZ ATTACKED TO THE STEEL FRAMEWORK (1 = 2).

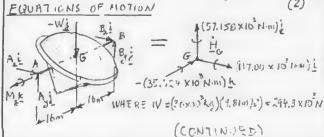
Ho=-I'W cos O(isino-j coso)+IW, sino(10050+jsino)+IW,K $H_{c} = (I-I)\omega_{l}\sin\theta\cos\theta i + (I'\cos\theta + I\sin^{2}\theta)\omega_{l}i + I'\omega_{l}k$ (1)

RATE OF CHANGE OF HA

WE NOTE THAT WI AND WI ARE CONSTANT, BUT THAT & IS A FUNCTION OF & WITH DERIVATIVE 8- WI EU. (18.22):

 $\vec{H}^{C} = (\vec{H}^{C})^{G^{2}M^{2}} + \vec{\nabla} \times \vec{H}^{C} = (\vec{I} - \vec{I}_{o}) \vec{n}^{s} (\cos_{o} \rho - \sin_{o} \sigma) \cdot \vec{\rho} \cdot \vec{r} \rightarrow \vec{r}$ + 2 (I-I') W, sind cost of + Wzjx ([I-I') w, sind cost it + (I cos'0 + I sin' 8) W 2 2 + I'W, E] = (I-I') W, W, (coszo + sinzo j)- + (I-I') w, sinzo k + Iw, w, 1 H = [I'+(I-I')cos20] www.i+(I-I')w, w sin20j-;(I-I')w, sin20 K WITH GIVEN DATA: I= m = 50×103kg)(12 m)= 4.32×106kg.m2 I'= m R' = (30× 103kg)(10m)=3.00 × 106kg·117 W, = 0,20 rad/s, Wz = 0,25 rad/s, 0 = 60°, 20 = 120° H = (3+1.32005120)106(0.20)(0.25)1+(1.32 sin 120)106(0.20)(425)j-- \$ (13251n 120")10" (0.25)" K

E = (117.00x103 N·m)i + (57.158x103 N·m)j - (35.724 x10 N·m)k



18.156 continued

 $\Sigma_{W}^{A} = \Sigma_{(W_{B})}^{A} = \Sigma_{(W_{B})}^{A}$ (52 m) K × (Az - Az) + (16 m) K × (-294.3 × 10 . 1+ Mx + = H (32 m) A = - (32 m) A = + (16 m) (294.2 x10 M) i + M = K= = (117 x10 N·m) i + (57. 158 v10 N·m) = -(=5.724 ×10 N·m) K

EQUATING THE COEFF. OF THE UNIT VECTORS: 1) -(32 m)A, + (16 m)(194.3×10 N) = 117×18 N.m

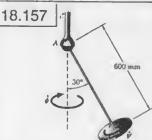
1) (32 m) Az = 57,158 × 10 N·m Az = 1.7862 × 103 N

K) M2=-35,724 X10 N.m

(a) FORCES AT A AND B

A = A 1 + A 1 A = (1,786 KN) + (143.5 KN) & IF = ma = 0: A+B-W=0 B=(294.3 EN)j-(1.786 EN)i-(143.5 +N)j B=-(1.786 KN) i+(150.8KN) }

(b) COUPLE MIK MK=-(35.7 EN.m)k N, k=- (35, 724 x103 Nin) k



GIVEN: 2 - K4 DISK OF 150-mm DIMMETER ATTACHED TO ROD SUPPORTED BY BALL AND SOCKET AT A. #=36 rpm AS SHOWN

RATE OF SPIN Y

USING THE FRAME AZYZ (WITH THE 4 AXIS POILITING TOWARD US), PIND NOTING THAT THE PRECESSION & STEADY EQ. (18 43) YIELDS



ZH = T × H (1) WHERE H=I,W, L+I, W, K H = I' (-+sin p) i + I (V++ osp)k

AND A = - + singi + + cospik THUS: ZM=(-fsinßi+twspk)x[I'(+sinß)i+I(V++aosp)k] IM=[I(++cosp)-I'+cosp]+sinpj

BUT IM = ABX-mgK = - (KX-mg (-sin be + wspk) = - mglsinßj

EQUATING THE R.H. MEMBERS OF (2) AND (2)1 [I(V++ cos B)- I'+cosp)] +sin B=-angl sin B

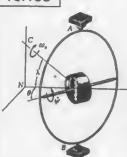
[IV+ (I-I') + cos p] +=-mgl ir= 1'-1 + cosp - 7

FROM GIVEN DATA: I = 1 m2 = 1 (2 kg)(0.075m) = 5.625 × 10 I'=m(l'+ b')= 2[(0.6)+ 1/4 (0.075)]=0,72281, I-1 = 127.5 \$=- 36 rpm=- 1,2 TT rad/s, B = 30

EQ.(4): \(-(127.5)(-1.27) \cos30 - \frac{2(9.81)(0.6)}{(5.625 \times 10^3)(-1.27)} = -416.27 + 555.13 = 138.86 rad/6

Y-1326 rpm

18.158



GIVEN:

GYROCOM PASS CONSISTING OF ROTOR SPINNING ATRATEY ABOUT AXIS MOUNTED IN GIMBAI RUTATING FREELY APOUT VERTICAL AB. O = ANGLE FORMED BY AXIS OF ROTOR AND HERIDIAN NS. A = LATITUDE = ANGLE PORHED BY NS AND LIKE OC PARALLEL TO GARTH AXIS WE = ANG. VELOCAT OF EARTH

SHOW THAT

(a) THE EQUATIONS OF MUTION OF THE EYROCOM FALS ARE

 $I'\ddot{\theta} + I\omega_z\omega_z\cos\lambda\sin\theta - I'\omega_z^2\cos^2\lambda\sin\theta\cos\theta = 0$ $I\omega_{\rm s}=0$

ABOUT ITS AXIS.

WHERE W, = RECTANGULAR COMPONENT OF FOTAL ANG. YELDCITY 6) ALONG AXIS OF ROTOR

(b) NEGLECTING TERMS IN W. AND FOR SMALL VALUES OF 8,

$$\ddot{\theta} + \frac{I\omega_s\omega_s\cos\lambda}{I'}\theta = 0$$

AND THAT AXIS OF RUTOR OSCILLATED ABOUT THE LINE NS.

(A) ANGULAR MOMENTUM ABOUT O.



WE SELECT A FRAME OF REFERENCE Day & ATTACHED TO THE GIMBAL.

THE ANG. VELOCITY OF OXYZ WITH RESPECT TO A NEWTONIAN FRAME IS D = No K + 0 }

R = - cost sint i + sind j + cost cost K THUS: Q = - 07 cos > Sinti + (0+62 Sinx) 3 + 62 cos > as Ok (1)

THE AUG. VELOCITY WO OF THE ROTOR IS OBTAINED BY ADI ING ITS SPIN YK TO Q. SETTING Y+W, COS ALOSO = W.

W= - We cos nem ti + (+ we sin) j + we k

THE ANE MOMENTUM HA OF THE ROTUR IS

WHERE I, = I, = I' AND I, = I . RECALLING(2) WE WRITE

$$H_0 = -1'\omega_c \cos \lambda \sin \theta \, i + 1'(\theta + \omega_c \sin \lambda) \, j + I \omega_c \, k$$
 (3)

EQUATIONS OF MOTION EQ.(10,28): ZM = (H) 0242 + Q XH OR, FROM (1) & (3): IM =- I'w, cosh coso 0 i + 1' 0 j + I w, k +

W 6057 6050

WE CHISCRYE THAT THE ROTOR IS FREE TO SPIN ABOUT THE E ANIS AND FREE TO ROTATE ABOUT THE ! AYIS. THESEFORE THE Y AND & COMPUNENTS OF EMO MUST BE ZERO. IT FOLLOWS THAT THE COEFFICIENTS OF J AND K IN THE R.H. MEMBER OF ER. (4) MUST ALSO BE ZERU.

(CONTINUED)

18.158 continued

SETTING THE COEFF. OF & IN THE R.H. MEMBER OF EQ. (4) EQUAL TO ZERO:

 $T' + (-I'\omega_{\rho} \cos \lambda \sin \theta)(\omega_{\rho} \cos \lambda \cos \theta) -$

- (- W, was sint) I w = 0

 $I'\theta' + I\omega_2 \Omega_c \cos \lambda \sin \theta - I'\omega_c^2 \cos \lambda \sin \theta \cos \theta = 0$ (5) (O.F.D.)

SPITING THE COFFE OF K EQUAL T ZERO!

$$I\dot{\omega}_2 + (-\omega_c \cos A \sin \theta) I'(\theta + \omega_c \sin \lambda) - (-I\dot{\omega}_c \cos \lambda \sin \theta) \dot{\theta} + \omega_c \sin \lambda) = 0$$

OBSERVING THAT THE LAST TWO TERMS CANCEL OUT, WE HAVE

(6)

(b) IT POLLOWS FROM EQ. (6) THAT W, E CONSTANT (7)

REWRITE EU. (5) AS FULLOWS :

I'b + (IW, -I'W, cos x cos 0) W, cos x sint=0

IT IS EVIDENT THAT Wa >>> No. WE CAN THEREFORE NEGLECT THE SECOND TERM IN THE PARENTHESIS AND WRITE

 $I'\theta + I\omega_*\omega_*\omega_s\lambda_sint = 0$ $\theta + \frac{\Gamma \omega_2 \omega_e \cos \lambda}{\Gamma^2} \sin \theta = 0$ (8)

WHERE THE COEFFICIENT OF SINB IS A CONSTANT. THE ROTOR, THEREFORE, OSCILLATES ABOUT THE LINE NS AS A SIMPLE PENDULUM.

FOR SMALL OSCILLATIONS, SIN 8 = 8, AND ED. (8)

YIELDS

$$\ddot{\theta} + \frac{I \omega_2 \omega_c \cos \alpha}{T} \theta = 0 \quad (0.E.D.)(9) \blacktriangleleft$$

ER. (9) IS THE EQUATION OF SIMPLE HARMONIC MOTION WITH PERIOD

$$\mathcal{Z} = 2\pi \sqrt{\frac{1}{I\omega_z\omega_c\cos\lambda}} \tag{10}$$

SINCE ITS ROTOR OSCILLATES ABOUT THE LINE NS, THE BYROCOMPASS CAN BE USED TO DETERMINE THE DIRECTION OF THAT LINE. WE SHOULD NOTE HOWEVER THAT POR VALUES OF & CLUSE TO 90°08-90. THE PERIOD OF OSCILLATION BEWILDS VERY LARGE AND THE LINE ABOUT WHICH THE ROTOR OSCILLATES CANNOT PE DETECHINED. THE GYRUGHPASS, THEREFORE, CANNOT BE USED IN THE POLAR REGIONS.

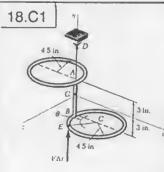


FIGURE SHOWN MADE OF WIRE WEIGHING \$ 02/TO IS SUSPENDED FROM LONG AB. (MPULSE FAL = (0.5 16.5)) IS APPLIED AT E FIND MIMEDIATELY AFTER IMPACT, FOR VALUES OF A FROM O TO INCLEMENTS

(a) VELOCITY OF G.
(b) ANGULINE VELOCITY

ANHLYSIS

LET m' = MASS PER UNIT LENGTH 2a = LENGTH OF ROD AB t = RADIVS OF BACH RING

COMPUTATION OF MASSES:

AB:
$$m_{AB} = 2a m'$$
 (1)
EACH RING: $m_{R} = 2\Pi L m'$ (2)

MOMENTS OF INERTIAL

AB:
$$(I_R)_{AB} = (I_2)_{AB} = \frac{1}{3} \pi_{AB} a^3, (I_3)_{AB} = 0$$
 (4)

EACH RING:
$$(I_{z})_{R} = \frac{1}{2} n I_{R} z^{2} + m_{R} a^{2} = m_{R} (\frac{1}{2} z^{2} + a^{2})$$
 (5)
 $(I_{y})_{R} = m_{R} z^{2} + m_{R} z^{2} = 2 m_{R} z^{2}$ (6)

$$(I_y)_R = m_R t^1 + m_R t^2 = 2m_R t^2$$

$$(L_z)_R = \frac{1}{2} m_R t^2 + m_R (t^2 + a^2) = m_R (\frac{3}{2} t^2 + a^2)$$
(6)

ENTIRE PIGURE :

$$I_{2} = (I_{2})_{AB} + 2(I_{2})_{R}, \quad I_{3} = 2(I_{3})_{R}, \quad I_{2} = (I_{2})_{A} + 2(I_{3})_{R} \quad (2)$$

PRODUCTS OF INTRAIA!

THE ONLY NON-ZERO PRODUCTS OF INERTIA ARE $(I_{29})_R$ $I_{29} = 2(I_{29})_R = -2 m_R 2a$ (9)

IMPULLE- MOMENTUM PRINCIPLE:

EQUATING IMPULSE AND MOMENTUM AFTER IMPACT

$$F\Delta t = m\bar{v}$$
: $(F\Delta t)_{j=m\bar{v}}$
 $\bar{v} = \frac{F\Delta t}{m} j (FOR ALL VALUES OF \theta) (10)$

ROUNTING MOTIENT OF INPULSE ABOUT G AND ANGULAR HEHENTUM HG APTER IMPACT (NOTE THAT THERE IS MI IMPALSIVE PILLE EXCENT F)

$$\underline{\mathcal{Z}}_{F} \times F\Delta t \underline{j} = \underline{H}_{6}$$

$$\underline{H}_{6} = \left[\frac{1}{2} (1 - \cos \theta) \underline{i} - a \underline{j} + 2 \sin \theta \underline{k} \right] \times F\Delta t \underline{j}$$

$$= -2 F\Delta t \sin \theta \underline{i} + 2 F\Delta t (1 - \cos \delta) \underline{k}$$

THUS: $H_{\chi} = -\frac{1}{2} F\Delta t \sin \theta$, $H_{\chi} = 0$, $H_{\chi} = \frac{1}{2} F\Delta t (1-\omega s\theta)$ (II) USING EQS. (18.7) AND RECALLING THAT $I_{\chi \chi} = I_{\chi \chi} = 0$;

$$I_{\lambda} \omega_{\lambda} - I_{\lambda \gamma} \omega_{\beta} = H_{\lambda}$$
 (12)

$$-I_{y}\omega_{x}+I_{y}\omega_{y}=0$$
 (13)

$$I_2 i \partial_2 = H_2 \tag{14}$$

(CONTINUED)

18.C1 continued

SOLVING FRS. (12) AND (13) SIMULTANEOUSLY FOR ω_{x} AND ω_{y} , AND FR. (14) FOR ω_{x} , WE OBTAIN

$$\omega_{z} = \frac{I_{1} H_{x}}{I_{z} I_{y} - I_{zy}^{2}}, \quad \omega_{y} = \frac{I_{2y} H_{z}}{I_{z} I_{y} - I_{zy}^{2}}, \quad \omega_{z} = \frac{H_{z}}{I_{z}}$$
(15)

ENTER m' = \(\frac{(5/8)/16)1b}{32,2 \text{ ft/s}^2}\), \(\alpha = \frac{3}{12}\)ft, \(\beta = \frac{45}{12}\)ft, \(\beta t = 0.5 \)bs

COMPUTE m_{AB} , m_R , AND m_l FROM EQS. (1), (2), AND (3) COMPUTE $(I_z)_{AB}$ AND $(I_d)_{AB}$ FROM EQS. (4) COMPUTE $(I_z)_R$, $(I_g)_R$, AND $(I_g)_R$ FROM EQS. (5), (6), AND (7) COMPUTE I_L , I_g , AND I_g FROM EQS. (8) AND I_{Zg} FROM EQ. (9) COMPUTE $\overline{U} = F\Delta t/m_l$ AND PRINT FOR $\theta = 0$ To $\theta = 180^\circ$ AND USING 10° INCREMENTS:

CALCULATE H, AND H, FROM POS. (11)
CALCULATE W, WY, AND W, FROM EWS. (15) AND TABULATE

PROGRAM OUTPUT

(b)

(A) Velocity of mass center vbar = 79.07 ft/s (directed upward)

	Angular	velocity	1
Theta	(Omega) x	(Omega) y	(Omega) z
degrees	rad/s	rad/s	
0.00	0.00	0.00	0.00
10.00	-54.88	18.29	1.81
20.00	-108.10	36.03	7.18
30.00	-158.03	52.68	15.94
40.00	-203.16	67.72	27.84
50.00	-242.12	80.71	42.50
60,00	-273.72	91.24	59.49
70.00	-297.00	99.00	78.29
80.00	-311.26	103.75	98.33
90.00	-316.06	105.35	118.99
100.00	-311.26	103.75	139.65
110.00	-297.00	99.00	159.68
120.00	-273.72	91.24	178.48
130.00	-242.12	80.71	195.47
140.00	-203.16	67.72	210.14
150.00	-158.03	52.68	222.03
160.00	-108.10	36.03	230.80
170.00	-54.88	18.29	236.17
180,00	0.00	-0.00	237.97



PROBE WITH $m=2500\,\mathrm{kg}$, $k_z=0.98\,\mathrm{m}$, $k_y=1.06\,\mathrm{m}$, $k_z=1.02\,\mathrm{m}$. 500-N MAIN THRUSTER E; 20-N THRUSTERS A, B,C,D CAN EXPEL FUEL IN Y DIRECTION. PROBE HAS ANG. VELOCITY $\omega=\omega_z\,\dot{b}+\omega_z\,\dot{k}$

FIND WHICH TWO OF THE 20-N THRUSTERS SHOULD BELSED TO REDUCE ANG. VECUCITY TO ZERO AND FOR HOW LUNG EACH OF THEM SHOULD BE ACTIVATED, ASSUMING

(a) $\omega = (0.040 \text{ rad/s})i + (0.080 \text{ rad/s})k$, AS IN PROB. 18.33,

(b) $\omega = (0.060 \text{ rad/s})i - (0.040 \text{ rad/s})k$, AS IN PROB. 18.34,

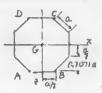
(c) $\omega = (0.060 \text{ rad/s})i + (0.020 \text{ rad/s})k$,

(d) w = - (0.060 rad/s) i - (0.020 rad/s) k.

ANALYSIS

INITIAL ANG, MOMENTUM:

 $H_{G} = I_{\chi} \omega_{\chi} \underline{i} + 0 + I_{\chi} \omega_{\chi} \underline{k} = m k_{\chi}^{1} \omega_{\chi} \underline{i} + m k_{\chi}^{2} \omega_{\chi} \underline{k}$ $Thus \quad H_{\chi} = m k_{\chi}^{2} \omega_{\chi} \quad H_{\gamma} = 0 \quad H_{\chi} = m k_{\chi}^{2} \omega_{\chi} \qquad (!)$ ANGULAR IMPULSE OF TWO 20-N THRUSTERS:



LET US ASSUME THAT A AND B ARE ACTIVATED.

ANE, IMPULSE ABOUT G = 2A × (-FDTA) & + 2B × (-FDTB) & = (-0.5ai + 1.2071 aK) × (-FDTB) & + + (0.5ai + 1.2071 aK) × (-FDTB) &

ANG, IMP. = 1.1071 a F (DtA+ DtB) i + 0.5 a F (DtA- DtB) k (2)
IMPULSE-HOMENTUM PRINCIPLE

WE MUST HAVE HE + ANE.IMP. = 0

OR, USING COMPONENTS:

 $H_1 + 1.2071aF(\Delta t_A + \Delta t_B) = 0$ $\Delta t_A + \Delta t_B = -\frac{f/k}{1.2071aF}$ $H_2 + \Delta 5aF(\Delta t_A - \Delta t_B) = 0$ $\Delta t_A - \Delta t_B = -\frac{f/k}{0.5aF}$ Solving these Equations Simultaneously:

 $\Delta t_{A} = -\frac{H_{e} + 0.41421 H_{x}}{0.F}, \quad \Delta t_{B} = \frac{H_{e} - 0.41421 H_{b}}{0.F}$ (3)

IF DtA > 0, ASSULTTION IS CORRECT, A SHOULD BE USED;
IF DtA < 0, ASSULTION IS WESTED; C SHOULD BE USED AND
ACTIVATED FOR Dto = [LtA].

SIMILARLY, IF Dt > 0, B SHOULD EF USES, AND IF GTE CO D SHOULD BE USED WITH DT = | Dt |.

OUTLINE OF PROGRAM

ENTER PART: a, b, c, or d

ENTER m = 2500 kg. K = 0.98 m, kz = 1,02 m

ENTER a = 1.2 m, F = 20 N

ENTER VALUES OF WE AND WZ

COMPUTE H, AND H, FIOM EDS. (1)

COMPUTE At AND At FROM ERS. (3)

IF At A > 0, PRINT At A; IF NOT, PRINT At = | At A |

IF Dt > 0, PRINT Dt ; IF NOT, PRINT Dt = | Dt |

PROGRAM OUTPUT

(a) CANDB; At = 8.1605; At = 4.8455

(b) A AND D; Dtn=1.8493; DtD=6.8215

(c) C AND D; At = 4.6543; At = 0.31883 (d) A AND B; Ata = 4.6545; Ata = 0.31885 AAAA

18.C3

8 in

6 in

6 in

8 in

6 in

GNEN:

A COUPLE MO=(0.03 16.4) [
15 APPLIED AT t = 0 TO
2.7-16 ASSEMBLY OF SMORT
ALUMINUM OF UNIFORM
THICKNESS

FIND:

(a) COMPONENTS ALONG THE RUTATING Y AND AYES OF THE DYNAMIC REAC-

TIONS ATA AND & FROM t=0 TO t=2s AT O.13 INTERVALS,

(b) THE TIME (WITH 3 SIGNIFICANT FIBURES) AT WHICH THE

FROMPONENTS OF THESE REACTIONS ARE BOUGHTO ZERO.

ANALYS 15

WE COMPUTE THE MOMENT AND PRODUCTS OF INERTIA OF THE ASSEMBLY WITH RESPECT TO THE CENTROLDAL AXES DIX'E', WE FIRST COMPUTE THE MITTERT AND PRODUCTS OF AREAS FOR EACH SOURCE: (Ix) AREA = \frac{1}{3}a', (Ix) AREA = \frac{1}{4}a', (Ix) AREA = 0

FOR EACH TRIANGLE: (TA) ATEA = 12 04, - (I A) ATEA = 0

 $(I_{22},)_{PPEA} = \frac{1}{2} a^2 \vec{x} \cdot \vec{E} + \vec{I}_{2'E'} = -\frac{1}{2} a^2 (\frac{1}{3} a)(\frac{1}{3} a) + \frac{1}{72} \delta^2 = -\frac{15}{72} a^2 [CF, SP9.6]$ FOR ENTIRE ASSEMBLY:

THE MASS MOMENT AND PRODUCTS OF INTERTIA ARE OBTAINED BY MULTIPLYING THESE EXPRESSIONS BY THE MASS M OF THE ASSEMBLY AND DIVIDING BY ITS AREA, WHICH IS EQUAL TO 30.2:

 $I_{\lambda} = \frac{5}{18}ma^{4}, \quad I_{3} = -\frac{1}{6}ma^{4}, \quad I_{\lambda + 1} = -\frac{5}{26}ma^{4}$ (1)

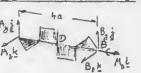
SETTING W= W, W= W= O IN EDS. (18.7), JE HAVE

 $H_{\lambda} = I_{\lambda}\omega$, $H_{y} = -I_{xy}$, ω , $H_{z} = -I_{zz}$, ω

HD=(I,i-I, , , , , , , , , , , ,) ...
EB (18.20): HD =(HD) D242+ + DX HD

$$\begin{split} & \stackrel{\cdot}{H} = (I_{\lambda} \stackrel{\cdot}{\cdot} - I_{\lambda y}, \stackrel{\cdot}{g} - I_{\lambda p}, \stackrel{\cdot}{E}) \stackrel{\cdot}{\omega} + \stackrel{\cdot}{\omega} \stackrel{\cdot}{\cdot} \stackrel{\cdot}{\cdot} (I_{\lambda} \stackrel{\cdot}{\cdot} - I_{\lambda y}, \stackrel{\cdot}{\sigma} - I_{\lambda p}, \stackrel{\cdot}{E}) \stackrel{\cdot}{\omega} \\ & = (I_{\lambda} \stackrel{\cdot}{\cdot} - I_{\lambda y}, \stackrel{\cdot}{g} - I_{\lambda p}, \stackrel{\cdot}{E}) \stackrel{\cdot}{\omega} - I_{\lambda y}, \stackrel{\cdot}{\omega} \stackrel{\cdot}{\cdot} + I_{\lambda q}, \stackrel{\cdot}{\omega} \stackrel{\cdot}{\tau} + I_{\lambda q}, \stackrel{\cdot}{\omega} \stackrel{\cdot}{\tau}) \stackrel{\cdot}{\omega} \end{split}$$

 $H_p = I_x \alpha \underline{i} + (I_{xx}, \alpha^2 - I_{xy}, \alpha) \underline{i} - (I_{xy}, \alpha^2 + I_{xy}, \alpha) \underline{k}$ EQUATIONS OF MOTION



 $\frac{\left(I_{x_{\xi}}, \omega^{\xi}, I_{x_{\xi}}, \kappa\right)_{\underline{i}}}{I_{x_{\xi}}} \\
-\left(I_{x_{\xi}}, \omega^{\xi}, I_{x_{\xi}}, \kappa\right)_{\underline{k}}$

 $\Sigma M_B = \Sigma [M_B]ess:$ $M_0 = 4ai \times (A_j j + A_k k) = H_0$ $M_0 = 4a_j k + 4a_k k = I_2 \times k + (I_{n_1} \Omega^2 - I_{n_1} X) j - (I_n \Omega^2 + I_n X) k$ EQUATING THE COSFF. OF THE ONIT VECTORS:

 $\begin{array}{cccc}
\textcircled{1} & M_o = I_z \alpha & \alpha = M_o/I_z \\
FROM WHICH WE OBTAIN & \omega = \alpha t
\end{array}$ (2)

FROM WHICH WE OBTAIN $\omega = \alpha \pm$ (3) $D A_{\pm} = (I_{\lambda \delta}, \omega^{\delta} - I_{\lambda \delta}, \alpha)/4\alpha \qquad (4)$

 $k = (I_{xy}, \omega^2 + I_{xy})/4a$ (5)

 $\sum_{F} = \sum_{F} (F)_{ess} = 0: \quad \underline{A} + \underline{B} = 0$ $T + vs: \quad \underline{B}_{g} = -\underline{A}_{g} \qquad \underline{B}_{g} = -\underline{A}_{g} \qquad (6)$

(CONTINUED)

18.C3 continued

OUTLINE OF PROGRAM

(a) ENTER Mo= 0.03 lb.ft, W= 2.7 lb., a= 0.5 ft COMPUTE m = W/32.2

COMPUTE Ix, Ixy, Ixe, FROM EQS. (1)

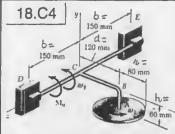
COMPUTE & FROM EQ. (2) FOR t = 0 TO t = 25 AT 0.1.5 INTERVALS: COMPUTE W FROM EQ (3)

COMPUTE Ay, Az, By, B, FROM EQS. (4), (5), AND (6)
AND THEOLOGY VS &

(b) DETERMINE BY IMPRECTION THE TIME INTERVAL IN WALCH AZ AND BZ CHA-15 5 SIGY AND RUN THE PROGENTY OVERTHAT INTERVAL, USING D.DI-S INCREMENTS. REPEAT THIS PROCEDURE, USING 0.001-5 INCREMENTS. THE DESIRED VALUE OF I 15 THAT POR WHICH AND AND BO ARE SMALLEST.

PRUGAMMI OUTPUT

(a)	t u	Ay 1b	A2 1b	Ву 16	Bz 1b
	0.00000 0.10000 0.20000 0.30000 0.40000 0.50000 0.70000 0.80000 0.90000 1.00000 1.10000	-0.00750 -0.00796 -0.00935 -0.01167 -0.01492 -0.01909 -0.02419 -0.03022 -0.03718 -0.04506 -0.05387 -0.06361	0.00900 0.00861 0.00745 0.00552 0.00282 -0.00066 -0.00491 -0.00993 -0.01573 -0.02230 -0.02964 -0.03775	0.00750 0.00796 0.00935 0.01167 0.01492 0.01909 0.02419 0.03718 0.04506 0.05387 0.06361	-0.00900 -0.00861 -0.00745 -0.00552 -0.00282 0.00066 0.00491 0.00993 0.01573 0.02230 0.02964 0.03775
	1.2000 1.3000 1.4000 1.5000 1.6000 1.7000 1.8000 1.9000 2.0000	-0.07427 -0.08586 -0.09838 -0.11183 -0.12620 -0.14150 -0.15773 -0.17489 -0.19297	-0.04664 -0.05630 -0.06673 -0.07794 -0.08992 -0.10267 -0.11619 -0.13049 -0.14556	0.07427 0.08586 0.09838 0.11183 0.12620 0.14150 0.15773 0.17489 0.19297	0.04664 0.05630 0.06673 0.07794 0.08992 0.10267 0.11619 0.13049 0.14556
(b)	t.	Ay 1b	Az 1b	Ву 1 b	Bz 1b
D	0.4000 0.4100 0.4200 0.4300 0.4400 0.4500 0.4500 0.4700 0.4800 0.4900 0.5000	-0.01492 -0.01529 -0.01568 -0.01607 -0.01648 -0.01689 -0.01774 -0.01818 -0.01863 -0.01909	0.00282 0.00250 0.00218 0.00186 0.00152 0.00118 0.00082 0.00046 0.00010 -0.00028 -0.00066	0.01492 0.01529 0.01568 0.01607 0.01648 0.01689 0.01774 0.01818 0.01863 0.01909	-0.00282 -0.00250 -0.00218 -0.00186 -0.00152 -0.00188 -0.00082 -0.00046 -0.00010 0.00028 0.00066
	t #	Ay 1b	Az 1b	8y 1b	Bz 1b
	0.48000 0.48100 0.48200 0.48300 0.48500 0.48500 0.48700 0.48700 0.48900 0.49000	-0.01818 -0.01823 -0.01823 -0.01832 -0.01836 -0.01845 -0.01845 -0.01850 -0.01850 -0.01859 -0.01859	0.00010 0.00006 0.00002 -0.00001 -0.00005 -0.00009 -0.00013 -0.00016 -0.00020 -0.00024 -0.00028	0.01818 0.01823 0.01827 0.01832 0.01836 0.01841 0.01845 0.01850 0.01854 0.01859 0.01863	-0.00010 -0.00006 -0.00002 0.00001 0.00005 0.00009 0.00013 0.00016 0.00020 0.00024 0.00028



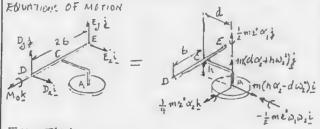
GIVEN: DISK: M=2.5 kg, t=80mm W. = 60 rea/s AT t=0 AND DECREASE AT RATE OF 15 rad/s. AT t=0, 0,=0 AMD COUPLE M= (0.5 N·m) k IS A PPLIED TO SHAFT DOL FIND: (a) COMPONENTS ALONG

THE ROTATING & AND & AXES OF THE DYNAMIC REACTIONS AT D AND E FROM t = 0 TO t=43 AT 0.2-5 INTERVALS, (b) THE TIMES to AND to (WITH 3 SIGNIFICANT FIGURES) AT WHICH E, AND E ARE RESPECTIVELY ENUMY TO ZERO.

ANALYSIS Ha= Iy, wi+ I, w. k = fmewij+ fmew, k EQ. (18,22): H = (H) + 1 × HA H = 1 m2 w1 + 1 m2 w2 k + w2 K x (1 :12 w1 + 1 m2 w,) = = = me'a, + + me'a, k - = me'a, will H = 1 mr (-w, w L + a, & + 0.5 a, b)

 $m\bar{a} = m(\alpha_1 \times \Delta_{N_c} - \omega_1^2 L_{N_c}) = m\alpha_1 k \times (di - hj) - m\omega_1^2(dl - hj)$ = # (dx, j + h x, 1-dw2 i + h w2 j)

mā = m (ha, -d.) i +n (da, + hw) j



ΣM = Z[M] ... -26k x (Ezi+Ezj)+Mor = - 1 m2in wi + 1 m2201 + 1 m20xx+ + (-bk+di-hj)x m [(hx-dw2)i+dx+hw]]

-26 Ej+26 Ezi+Mk = - + m12 WIWi i+ + m2 aj++ m2 a, k-- mb(ha,-dw) j +mb(dx+hw) +md(dx+hw)++mh(ha-dw)/k

EQUATE THE COEFF. OF THE UNIT VECTORS: B M=m(42+d+h)a, $\alpha_{z} = \frac{1}{m(\frac{1}{4}z^{2} + d^{2} + h^{2})}$ (3)

(4)

 $\bar{L}_{3} = \frac{m}{2h} \left(-\frac{1}{2} \dot{k}^{2} \omega_{1} \omega_{2} + bd \propto +bh \omega_{2}^{2} \right)$ (5)

 $\Sigma F = Z(F)_{eff}$: 2+E = maDi+Dj+Ej+Ej=m(hx,-dw2)i+m(dx+hw2)j

EQUATE THE COEFF. OF THE UNIT VECTORS: D, = m (hox, -dw, 1) - E, (6) Dy = m (doz+hwz) - Eg (7)

WE RECALL FROM THE GIVEN DATA THAT m = 2.5 kg, t= 0.08 m, b= 0.15m, d= 0.12 m, h= 0.06m (8)

wo = 60 mad/s 0, = - 15 mad/s NO NOTE THAT AT TIME t (9) Wz = CK, C WI= WO + Q, E

(CONTINUED)

M, = 0.5 N·m

18.C4 continued

OUTLINE OF PROGRAM

- (a) ENTER DATA SHOWN IN (B) ON PREVIOUS PAGE COMPUTE 02 FROM EU. (3) FOR \$ = 0 TO E = 4 5 AT 0.2-3 INTERVALS COMPUTE W, AND W, FRUM EWS. (9) COMPUTE Ex AND Eg FROM EGS. (4) AND (5) COMPUTE DE AND DE FROM EQS. (6) AND (7) AND TABULATE VS t.
- (b) TO FIND THE TIME & AT WHICH Ex = 0, DETERMINE BY INSPECTION THE TIME INTERVAL IN WHICH Ex CHANGES SIGN AND RUN THE PROGRAM OVER THAT INTERVAL, USING 0.01-5 INCREMENTS, REPEAT THIS PROCEDURE USING SPLECT FOR t, THE TIME 0.001-5 INCREMENTS AT WHICH EX IS SMALLEST. A SIMILAR PROCEDURE IS USED TO DETERMINE THE TIME to AT WHICH Ey = 0.

PROGRAM OUTPUT

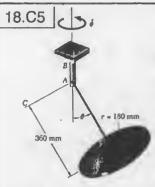
(ú)	(n)	Dx (N)	Dy (N)	Ex (N)	Ey (N)
	0.0000 0.2000 0.4000 0.6000 0.8000	0.3653 -0.2594 -2.1337 -5.2574 -9.6305	1.5306 4.9450 8.6576 12.6685 16.9775	1.1653 0.5406 -1.3337 -4.4574 -8.8305	1.5306 -1.2591 -3.0975 -3.9846 -3.9204
	1.0000 1.2000 1.4000 1.6000 1.8000 2.0000 2.2000 2.4000 2.6000	-15.2532 -22.1253 -30.2469 -39.6181 -50.2386 -62.1087 -75.2282 -89.5972 -105.2158	21.5848 26.4902 31.6939 37.1958 42.9958 49.0941 55.4906 62.1854 69.1783	-14.4532 -21.3253 -29.4469 -38.6180 -49.4386 -61.3087 -74.4282 -88.7973 -104.4158	-2.9050 -0.9384 1.9796 5.8488 10.6693 16.4411 23.1641 30.8384 39.4640
	2.8000 3.0000 3.2000 3.4000 3.6000 4.0000	-122.0838 -140.2012 -159.5681 -180.1846 -202.0505 -225.1659 -249.5307	76.4694 84.0588 91.9463 100.1321 108.6160 117.3982 126.4786	-121.2837 -139.4012 -158,7681 -179.3845 -201.2504 -224.3659 -248.7306	49.0409 59.5690 71.0483 83.4790 96.8609 111.1942 126.4786

(b) LAST STEP IN DETERMINATION OF t,

(s)	Dx (N)	Dy (N)	Ex (N)	Ey (N)
0.2700 0.2710 0.2720	-0.7733 -0.7817 -0.7902	6.2105 6.2289 6.2472	0.0267	-2.0107 -2.0206
0.2730 0.2740 0.2750	-0.7987 -0.8073 -0.8158	6.2656 6.2839 6.3023	0.0098 0.0013 -0.0073 -0.0158	-2.0305 -2.0403 -2.0501 -2.0599
0.2760 0.2770 0.2780	-0.8244 -0.8331 -0.8418	6.3207 6.3391 6.3575	-0.0244 -0.0331 -0.0418	-2.0697 -2.0795 -2.0892
0.2790	-0.8505 -0.8592	6.3759	-0.0505 -0.0592	-2.0989 -2.1086

LAST STEP IN DETERMINATION OF EZ

t;	Dx	Dy	Ex	Ey
(8)	(N)	(N)	(N)	(11)
1.2700	-24.8258	28.2776	-24.0258	-0.0253
1.2710	-24.8654	28.3034	-24.0655	-0.0114
1.2720	-24.9052	28.3292	-24.1052	0.0025
1.2730	-24.9449	20.3550	-24.1449	0.0165
1.2740	-24.9847	28.3808	-24.1847	0.0304
1.2750	-25.0245	28.4066	-24.2245	0.0444
1.2760	-25.0644	28.4325	-24.2644	0.0584
1.2770	-25.1042	20,4503	-24.3042	0.0724
1.2780	-25.1442	28.4842	-24.3441	0.0865
1.2790	-25.1841	28.5100	-24.3841	0.1006



GIVEN:

DISK WELDED TO ROD A 6 OF NEGLIGIBLE MASS CONVECTED BY CLEVIS TO SHAFT AB. ROD-AND DISK FREE TORUTATE ABOUT AC; SHAPT AB PREE TO ROTATE ABOUT VERTICAL AXIS. INITIALLY, 0=0, 0=0, 4=4 EIND:

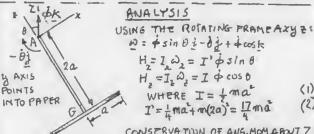
(a) MINIMUM VALUE OF B DURING ENSUING MOTION AND TIME REQUIRED FOR 8 TO RETURN TO BO (PERIOD) -

(1)

(2)

(5)

(b) ANG. VEL. & FOR VALUES OF & FROM & TOOM WING Z'INCREMENTS CONSIDER SUCCESSIVELY THE INITIAL CONDITIONS (i) θ = 90, φ = 5rod/s, (ii) θ = 90, φ = 10 rod/s, (iii) θ = 60, ф = 5 rod/s.



CONSERVATION OF ANG. MOM ABOUT Z SINCE THE PURCES CONSIST OF REACTION AT A AND WEIGHT W = - mg K AT G, WE HAVE ZM_= O AND H_ = CONSTANT SINCE Hy= H, sin0+ H, cosb = I'+ sin'0+ I + cos'0, WE HAVE

(I'sin'0+Ias'0)+=(I'sin'0+Ias'0)+ Q = I'sin't + Iw't (3) SETTING (4) AND Q = I'Sin't + I cos to

CONSERVATION OF ENERGY

AND SOLVING POR +:

T+V= E=CONSTANT: - (1, 0, + I, 0, + I, 0, + W(-20 cost)=E -) (I' + sin 0 + I' 0 + I + I + cus 0) - 2 mg a cos0 = E (I'sin'0+Icos'0)+I'0'-4mgacos0=ZERECALLING (3) AND SUBSTITUTING FOR + TROM (5):

中=(Q,/Q)か

(Q + 10/Q)+I'0 - 4mga cost= 2E

SOLVING FOR A: 62 = 1 (2 E + 4 mg a coso - 00 40 WHICH IS OF THE FORM 8 = f(0) (7) WHERE $f(\theta) = \frac{1}{T}(2E + 4mg a \cos \theta - \frac{Q_0^2 + \frac{1}{2}}{2})$ (8)

AND Q IS THE PUNCTION OF B DEPINED IN (3). CONSTANT 2E IS OBTAGNED DIMAKING 0=0,0=0, MD Q=Q, IN EO. (6): E= & & + - 2mg a coso, PROM (7) WE WRITE

t= 50 do (10) $\frac{d\theta}{dt} = \dot{\theta} = \sqrt{3(\theta)}$

(a) THE TIME & NE -DED FOR & TO DECREASE TO Om IS OBTAINED . THROUGH NUMERICAL INTEGRATION, BM SEING DEFINED BY THE PACT THAT \$ (On) = 0 (f CHANGES SIGN) (b) POR EACH DESIRED VALUE OF B, COMPUTE Q PRIMER(3) AND + FROM : 20. (5). (CONTINUED)

18.C5 continued

OUTLINE OF PROGRAM

ENTER a: 0.18 m, g = 9.81 m/s'. ASSUME m=1.

ENTER INITIAL CONDITIONS: 00 AND 9.

ENTER DECREMENT db You wish to use

COMPUTE I AND I' PROM (i) AND (2)

COMPUTE Q FROM (4) AND E PROM (9)

FOR 0=0, TO 0=0, (WHEN 5(0) CHANGES SIGN), AND

USING DECREMENTS db:

COMPUTE Q FROM (3)

COMPUTE \$(\theta) FROM (8)

CARRY OUT NUMERICALLY THE INTEGEN HOW

INDICATED' IN (10)

AT 2°INTERVALS, COMPUTE \$\frac{1}{2}\$ FROM (5) AND PRINT

THE VALUES OF \$\theta\$ AND \$\frac{1}{2}\$

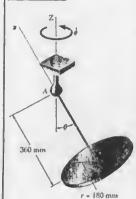
THE PERIOD OF THE OSCILLATION IN 8 15 OFTAINED BY DOUBLING THE VALUE OF £ WHEN O REACHES ITS MINIMUM VALUE On.

PROGRAM SHITEUT

113

(L)	(ii)			
THO= 90 PHIDO= 5 DTH= .1		TH0 = 90 PHID0 = 10 DTH = .1			
Theta	Precession Rate	Theta	Precession		
(degrees)	(rad/s)	(degrees)	Rate (rad/s)		
90.000 88.000 84.000 82.000 82.000 78.000 74.000 70.000 68.000 64.000 64.000 64.000 58.000 56.000	5.000 5.005 5.022 5.049 5.087 5.137 5.138 5.272 5.359 5.460 5.575 5.707 5.855 6.021 6.207 6.415 6.647 6.905 7.193	90.000 88.000 84.000 82.000 80.000 78.000 74.000 72.000 70.000 68.000 64.000 64.000 62.052 Theta min = Period = 0.5	10.000 10.011 10.043 10.097 10.174 10.273 10.397 10.545 10.719 10.920 11.151 11.413 11.709 12.042 12.404 62.1 degrees		
52.000 50.000 48.000 46.000	7.513 7.869 8.265 8.708	(iii)		
44.000 42.000 40.000 38.000 36.000	9.201 9.752 10.369 11.060 11.835	PHIDO= 5 DTH= .1 Theta (degrees)	Precession Rate (rad/s)		
34.000 32.000 31.959	12.705 13.683 13.704 32.0 degrees	60.000 58.000 56.000 54.000 52.000 50.000 48.000 46.000 42.000 40.000 38.000 36.894	5.000 5.181 5.382 5.606 5.855 6.133 6.787 7.171 7.601 8.082 8.620 8.945		
		Theta min = 3 Period = 0.72	6.9 degrees		

18.C6



GIVEN:

NEGLISTELE MASS SUPPLIED BY

EALL AND SUCKET AT A.

INITIALLY, $\theta = \theta_0$, $\dot{\theta} = 0$, $\dot{\gamma} = \dot{\gamma}_0$,

AND $\dot{\phi} = \dot{\phi}_0$.

FIND:

(a) Anni' in while θ_m of θ in

ENSUMES HOTION AND PERIOD (TIME

REQUIRED FOR θ To RETUCK TO $\dot{\phi}_0$).

(b) $\dot{\gamma}$ AND $\dot{\phi}$ FOR VALUES OF $\dot{\theta}$

DILK WELDED TO ROL HG OF

FROM θ_0 TU θ_m USING 2" INCREMENTS. CONSIDER SUCCESSIVELY THE INTIAL CONDITIONS (i) $\theta_0 = 40^\circ$, $\psi = 50$ rad/c, $\dot{\phi} = 0$ (ii) $\dot{\theta}_0 = 40^\circ$, $\dot{\psi} = 0$, $\dot{\phi} = 5$ rad/s

(iii) b = 90°, b = 50 rad/s, b = 5 rad/s (iv) b = 90°, b = 10 rad/s, d = 5 rad/s (v) b = 60°, b = 0, b = 5 rad/s (vi) 0 = 60°, b = 50 rad/s, b = 5 rad/s

ANALYSIS

VIL PARK 2

- 0j (A)

2a

VSING THE ROTATING PRAILS

Azyz WITH Y AND DIRITING

INTO THE PAPER: $\omega = \dot{\phi}$ Sind $\dot{\phi} = \dot{\theta} J + (\dot{Y} + \dot{\phi} \cos \theta) \dot{k}$ $H_a = I_z \omega_z \dot{i} + I_y \omega_y \dot{i} + I_z \omega_z \dot{k}$ $= I'\dot{\phi}$ Sind $\dot{i} - I'\dot{\theta} \dot{j} + I(\dot{Y} + \dot{\phi} \cos \theta) \dot{k}$ WHERE $I = \frac{1}{2} m a^a$ $I' = \frac{1}{4} m a^a + m(2a)^2 = \frac{17}{4} m a^a$ (2)

CONSERVATION OF ANGULAR MOMENTUM

SINCE THE ONLY EXPERIME FORCES ARE THE REACTION

HT A AND THE WEIGHT W=-11/K AT G, WE HAVE

ZMZ=0 AND ZMZ=D. SINCE Z IS PART OF A

NEWTONIAM FRAME OF REFERENCE, IT FULLIMS THAT

HZ = CONSTAIN. SECAUSE OF THE AXISYMMETRY OF

THE DISK, IT ALSO FOLLOWS THAT Hz = CONSTANT (SEE

P(NB. 18, 139). WE WRITE $H_{+} = const$: $I(\psi + \psi \cos \theta) = \beta$.

WHERE FROM INIT. COND.: $\beta = I(\psi + \psi \cos \theta)$ (4) $H_{2} = const$: $H_{2} \sin \theta + H_{2} \cos \theta = \alpha$ $I'\psi \sin^{2}\theta + I(\psi + \psi \cos \theta) \cos \theta = \alpha$

RECALLING (3) WE HAVE $I'\dot{\phi}\sin^2\theta + \beta\cos\theta = \alpha$ (5)
PROH INITIAL CONDITIONS: $\alpha = I'\dot{\phi}\sin^2\theta + \beta\cos\theta$ (6)
SOLVING (5) FOR $\dot{\phi}$: $\dot{\phi} = \alpha - \beta\cos\theta$ (7)
CONSERVATION OF ENERGY

 $T = \frac{1}{2} \left(I_{\chi} \omega_{\chi}^{2} + I_{\chi} \omega_{\chi} + I_{\chi} \omega_{\chi}^{2} \right)$ $= \frac{1}{2} \left[I^{2} \dot{\phi}_{S}^{2} \dot{\omega}^{2} \dot{\phi} + I^{2} \dot{\phi}_{\chi}^{2} + I \left(\dot{\psi}_{+} \dot{\phi}_{COS} \dot{\phi}_{\chi}^{2} \right) \right]$ SUBSTITUTE POR () FROM (3):

 $T = \frac{1}{2} \left(I' \dot{\phi}^{2} \sin^{2}\theta + I' \dot{\theta}^{2} + \frac{R^{2}}{I} \right) \qquad V = -mg \left(2a \right) \cos \theta - \frac{1}{2} \left(I' \dot{\phi}^{2} \sin^{2}\theta + I' \dot{\theta}^{2} + \frac{R^{2}}{I} \right) - 2mg \cos \theta = \frac{1}{2} \left(I' \dot{\phi}^{2} \sin^{2}\theta + \frac{R^{2}}{I} \right) - 2mg \cos \theta = \frac{1}{2} \left(I' \dot{\phi}^{2} \sin^{2}\theta + \frac{R^{2}}{I} \right) - 2mg \cos \theta \cos \theta$ (8)

From with the equation $E = \frac{1}{2} \left(I' \dot{\phi}^{2} \sin^{2}\theta + \frac{R^{2}}{I'} \right) - 2mg \cos \theta \cos \theta$ (9)

SOLVING (B) FOR \hat{O}^2 : $\hat{O}^2 = f(0)$ (10) WHERE $f(0) = \frac{1}{7!} (2\bar{E} - \frac{\hat{D}^2}{L} + 4mig \, a \cos \theta) - \dot{\phi} \sin^2 \theta \qquad (11)$

(CONTINUED)

18.C6 continued

Substituting FOR ϕ From (7) INTU (11), WE HAVE $f(\theta) = \frac{1}{I}, (2E - \frac{\beta^2}{I} + 4 \operatorname{mga} \cos \theta) - \left(\frac{\alpha - \beta \cos \theta}{I \sin \theta}\right)^L (12)$

FROM 20. (10) WE WRITE

$$\frac{d\theta}{dt} = \dot{\theta} = \sqrt{5(0)} \qquad \dot{t} = \int_{0}^{\theta} \frac{d\theta}{\sqrt{5(0)}} \qquad (13)$$

(a) THE TIME $\frac{1}{2}$ E NETDED POR B TO DECREASE TO θ_m is obtained through numberial integration, θ_m , Being defined by the Prat that $f(\theta_m) = 0$, that is, that $f(\theta)$ charkes sign for $\theta \pm \theta_m$ (b) For each desired value of θ , conjust ϕ Prom Eu. (7).

QUILINE OF PROGRAM

ENTER a = 0.18m, g = 9.81m/s. Assume m = 1.

ENTER INITIAL CONDITIONS: \$\theta_0, \forall_0, \text{AND} \forall_0.

ENTER DECREMENT AT YOU WISH TO USE

COMPUTE I AND I' FROM (1) AND (2)

COMPUTE IS FROM (4), A FRUM (6), AND E FROM (9)

FOR \$\theta = 0 and \$\theta = 0 m\$ (WHEN \$ 5(0) CHANGES \$ 516N),

AND USING DECREMENTS ATS

COMPUTE \$\theta FROM (7)

COMPUTE \$\theta FROM (7)

COMPUTE F(B) FROM (II)
CARRY OUT NUMERICALLY THE INTEGRATION DEFINED
IN EQ. (13)

AT 2° INTERVALS, PRINT THE VALVES OF θ , $\dot{\phi}$.

AND, FROM (3), OF $\dot{\psi} = \frac{A}{7} - \dot{\phi} \cos \theta$

THE PEULOD OF THE OSCILLATION IN θ IS DETAINED BY DOUBLING THE WILLIE OF E CORRESPONDING TO $\theta=\theta_{\rm m}$.

PROGRAM OUTPUT

(i)		(i	1)

TH0=90 PS DTH=0.10	1D0=50	PHIDO= 0	TH0=90 PS	1D0= 0	PHIDO= 5
Theta	Spin	Precese.	Theta	Spin	Precess.
degrees	rad/a	rad/s	degress		
			3	, -	
90.00	0.00	50.00	90.00	5.00	0.00
88.00	-0.21	50.01	88.00	5.01	
86.00	-0.41	50.03	86.00	5.02	
84.00	-0.62	50.06	84.00	5,06	
82.00	-0.83	50.12	82.00	5.10	-0.71
80.00	-1.05	50.18	80.00	5.16	-0.90
78.00	-1.28	50.27	78.00	5.23	-1.09
76.00	-1.51	50.37	76.00	5.31	-1.28
74,00	-1.75	50.48	74.00	5.41	-1.49
72.00	-2.01	50.62	72.00	5.53	-1.71
70.00	-2.28	50.78	70.00	5.66	
68.00	-2.56	50.96	68.00	5.82	-2.18
66.00	-2.87	51.17	66.00	5.99	
64.00	-3.19	51.40	64.00	6.19	
62.00	-3.54	51.66	62.00	6.41	
60.00	-3.92	51.96	60.00	6.67	
58.00	-4.33	52.30	58.00	6.95	-3.68
56.00	-4.79	52.68	56.00	7.27	
54.00	-5.28	53.11	54.00	7.64	
52.00	-5.83	53.59	52.00	8.05	
50.00	-6.44	54.14	50.00	8.52	-5.48
48.00	-7.13	54.77	48.00	9.05	
46.00	-7.90		46.00	9.66	
44.11	-8.72	56.26	44.00	10.36	
Theta min			42.00	11.17	
Period =			40.00	12.10	-9.27
		•	38,23	13.06	
			Theta min	= 38.2	degreea
			Period = 1	0.687 #	70.37770

	(LLL)			(TV)			
	SID0-50	PHIDO = 5	TH0=90 PS	ID0=10	PHIDO- 5		
DTH=0.10			DTH-0.10				
Theta	Spin	Precess.	Theta	Spin	Precess.		
degraes	red/s	red/s	dagress	rad/s	rad/s		
90.00	5.00	50.00	90.00	5.00	10.00		
88.00	4.80	49.83	88.00	4.96	9.83		
86.00	4.61	49.68	86.00	4.94	9.66		
84.00	4.43	49.54	84.00	4.93	9.48		
82.00	4.26	49.41	82.00	4.93	9.31		
80.00	4.10	49.29	80.00	4.94	9.14		
78.00	3.95	49.18	78.00	4.97	8.97		
76.00	3.80	49.08	76.00	5.01	8.79		
74.00	3.66	48.99	74.00	5.06	8.61		
72.00	3.52	48.91	72.00	5,13	8.42		
70.00	3.38	48.84	70.00	5.21	6.22		
68.00	3.25	48.78	68.00	5.30	8.01		
66.00	3.12	48.73	66.00	5.42	7.80		
64.00	3.00	48.69	64.00	5.55	7.57		
62.00	2.87	48.65	62.00	5.71	7.32		
60.00 58.00	2.75	48.63	60.00	5.88	- 7.06		
56.00	2.62	48.61	58.00	6.09	6.78		
54.00	2.49	48.61	56.00	6.32	6.47		
52.00	2.30	48.61	54.00	6.58	6.13		
50.00	2.08	48.66	52.00	6.89	5.76		
48.00	1.93	48.71	50.00	7.23	5.35		
46.00	1.77	48.77	48.00	7.63	4.90		
44.00	1.59	48.85	46.00	8.08	4.38		
42.00	1.40	48.96	44.00	8.61	3.81		
40.00	1.20	49.08	42.00	9.21	3.15		
38.00	0.96	49.24	38.00	10.75	1.53		
36.00	0.70	49.44	36.00	11.72	0.52		
34.00	0.39	49.67	34.00	12.87	-0.67		
32.00	0.04	49.97	32.00	14.25	-2.09		
30.00	-0.38	50.33	30.00	15.93	-3.79		
28.00	-0.88	50.78	28.23	17.71	-5.60		
26.00	-1.49	51.34	Thete min				
24.00	-2.26	52.06		0.655 8			
22.00	-3.24	53.00	101100 -	0.000			
20.00	-4.51	54.24					
18.00	-6.23	55.92					
16.00	-8.62	58.28					
14.00	-12.09	61.73					
12.00	-17.44	67.06					
10.00	-26.30	75.90					
8.00	-42.61	92.19					
6.00	-77.82	127.40					
5.62	-89.03	138.60					
	F 60	3					
Theta min Period =							

(v) (vi)

	- /				
TH0=60 PS DTH=0.10	1D0= 0	PHIDO = 5	TH0=60 PS DTH=0.10	ID0=50	PHIDO= 5
Theta	Spin	Precass.	Theta	Spin	Pracess.
degrees			degraes	rmd/m	red/s
60.00 58.00 56.00	5.20	-0.26	60.00 58.00 56.00	4.96	49.87
54.00	5.69		54.00	4.90	
52.00	5.98		52.00	4.89	
50.00	6.32		50.00	4.89	49.36
48.00	6.70	-1.98	48.00	4.90	49.22
46.00	7.14	-2.46	46.00	4.92	49.08
44.00	7.64	-2.99	44.00	4.96	48.93
42.00	8.22	-3.61	42.00	5 02	48.77
40.33	. 8 77	-4.18	40.00	5.10	48.59
		degrees	38.00	5.20	48.40
Period =	0.661 #	THE STATE OF	36.00	5.33	48.19
		•	34.00	5.49	47.95
			32.00	5.70	47.67
			30.00	5.96	
			28.00	6.28	46.95
			26.00	6.70	46.48
			24.00	7.23	45.90
			22.00	7.92	45.16
					44.19
			18.00		42.90
			16.00		41.10
			14.00		38.49
			12.00		34.47
			10.00		27.82
			8.00		
			6.01		
			Theta min		
			Period =	V.520 6	2

19.1 19.4 GIVEN: GIVEN: PARTICLE IN SIMPLE HARHONIC MOTION AMPLITUDE = 40 In. , PERIOD = 1.4 S. \$ 20 Ili/in EIND: HAXIMUM VELOCITY, UM EIND: MAIMUM ACCELERATION, QM SIMPLE HAP MONIC MOTION X=Xmsin(w,t+0) Wn= 211/tn = 2TT/(145) = 4.480 rAO/S (a) Ym= AMPLITUDE = 40 in = 3.333 ft X=(3.333) SIN (4.480++0) X= Xm Wn cos wn t+ b) Xm=Jn= KmWn 20 lb/in Um= (3.333 ft) (4 480 Hs Nm= 14.96 ft/5" $\mathring{x} = -\chi_m \omega_n^2 \sin(\omega_n t + \phi) \mathring{x}_m = \alpha_m = -\chi_m \omega_n^2$ Qm=(3,333 ft)(4,480r6) am= 67.1 ft/52 -19.2 GIVEN: PARTICLE IN SIMPLE HARMONIC MOTION MAXIMUM ACCELERATION 72 m/52 FREQUENCY f = 8 Hz. HAXIMUM VELOCITY FIND: AMPLITUDE, XM HAXIMUM VELOCITY, Nm SIMPLE HAPMONIC MOTION X= Xm sin(w,t+0) 19.5 GIVEN: Wn== = Tfn=(2TT)(8HZ.)=16 T rAD/S X= Xmwn cos(w, -+ 4) Nm= xmwn x=-xm(wn)251N(wn++++) am= xmwn am= 72 m/s2 = xm (16Ti rap/s)2 k = 12 kN/mKm=(7.2 m/s2)/(1617 rAD/s)2 Km=2.849x10m X-m= 2.85mm FIND; Um= xm wn= (2.849 mm)(16TT rAD/S) Um = 143.2 mm/s 19.3 GIVEN: (a) 15=0 PARTICLE IN SIMPLE HARMONIC HOTION AMPLITUDE = 300 MM MAXIMUM ACCELERATION = 5 m/s2 FIND: 1 U0=250 mm/s HAXIMUM VELOCITY, Um FREQUENCY F SIMPLE HALHONIC MOTION K=Km SIN(Wnt+0) K= 0.300 M X= (0.300) SIN (Wn+ 0) (M) THUS $\hat{x} = (0.3)(\omega_n)\cos(\omega_n t_1 \phi)$ (m/s) =-(0.3)(wni2sIN(wnt+0) (M/s) 1am1=(0.3 m/s)(Wn)2 am= 5m/s2 Wn= 1am1/(0.3m)=(5m/s2)/(0.3m)=16.667m0/s Wn= 4.082 rAD/S fin=Wn/2TT fn=(4.082 rAD/S)/(211 rAD/CYCLE)=0.649.7 HZ fn=0.650 HZ

BLOCK W=30 16 SPRING &= 2016/In. INITIAL DEFLECTION = Z.I IN . RELEASED FROM REST (a) PERIOD TO, AND FREQUENCY, In (b) HAXIMUM UELOCITY, UM AND ACCECERATION, am X= KM SIN (WIL+ 0) Wn= VR/m le=2016/in=340/6 Wn=V(24016/ft)/(3016)/(32.2/t) Wn= 16050 FAD/S Tn= Wn Tn= 217/16 050 = 0.3915 5 Tn=0.3915 fn= 1/tn= 1/0.391 = 2.55 HZ (b) 2m=2.1 in = 0.175 ft X= 0.175 SIN (16 USOT+0) Un = 2mW = (0.175 ft)(16 050 rails) Um= 2.81 ft/s am = Km Wn = (0.175 ft)(16.050 rAD/5)2 am=45.194/524 BLOCK M= 32 kg SPRING IR= 12 KN/m FINITIAL VELOCITY Un= 250 mm/s INITIAL DISPLACEMENT= 0 (a) PERIOD TO AND FREQ , FIN (D) ANDLITUDE Zm MAXIMUM ACCERTATION &M X= Xm SIN (Witt 0) MN=/18/IM =/15x103 N/M Wn= 19.365 FAD/S Tn= 211/wn Tn= 211/19.365 Tn= 0.3245 fn=1/cn=1/0.324=3081/24 (b) @t=0, x=0, &=v=20mm/s X= 0= Km SIN (W, (0)+ 0) AND Ф=O 2=0= xm wn cos (w,10)+0)= /mil, Un= 0.250 m/s= 7m (19.365 MD/s) 2m=(0.250m/s)/(19.365 rAD/s) Xm=12.91.x10-3 m 4m= 12.91 mm am= 4mwn = (12.91x103m)(19.365 +AD/5)2 am= 4.84 m/s2

Nm= 1.225 M/S

Um= KmWn=(0.3m)(4.08ZYAD/5)

19.6 GIVEN: PENDULUM IN SIMPLE HARHONIC HOTION PERIOD Tn= 1.35 HAXIMUM VELOCITY, UM= 15 In./S FIND: (a) AMPLITUDE OF THE HOTION, OM IN DEGREES (D) THE MAXIMUH TANGENTIAL ACCELERATION (04) M (0) SIMPLE HARHONIC HOTION 0=0m SIN (Wnt+0) WN= SH/CN=(SH)/(1135) Wn= 4.833 rAD/S B= Dmwncos (witte) Om= Omwn Um=lôm=lomWn Om= 1m/lwn (1) Um=1511/3 FOR A SIMPLE PENDULUM wn=/ 912 THUS 1= 3/w= 322 ft/s2 (4.833 MD/s)2 l= 1.378ft FROM (1) = Nm (EW,=(15/12 Pt/s)/(1.378 Pt) A.833 # @IN= 0.18769 TAD = 10.750 (b) at=10 HAX TAI GENTIAL ACCELERATION OCCURS WHEN & IS HAXIHUM, ==-OMWN SIN (WAT+O) BHAZ OMWAZ, (at) HAX= 10 mwnz (at inax = (1378 ft) (0.1879 rab) (4.833 m) (a+) m = 6.04 ft/s2 19.7 GIVEN: SIMPLE PENDULUM 2= 800 mm, ⊖HAY= 6° FIND: (a) FREQUENCY OF OSCILLATION. (b) HAXIMUH VELOCITY Nom OF THE BOB Wn= \ 9/R = \ (9.81 m/s2) (0.8 m) (a) Wn= 3.502 rAD/S fn= Wn/2TT=(3.502 ++1/5)/2TT fn= 0.557 Hz (b) == = sin(wnt+ b) = Omwncwa(wnt10) Dm= Om Wn N' m = (B m = 1 0 m W n = (0.8 m) (6) (1 rad) (3.52 m) (b) squaring (1) and (2) and Adding, x m cos o + x m s in o = A + 5 2 2 m (605 o + 5 in o) = x m = 41 m² Um= 293.4 x10 3m/s Um = 293 mm/s

19.8

GIVEN

PACKAGE A IN SIMPLE HAPHONIC HOTION AT A FREDUENCY WHICH IS THE SAKE AS THE HOTOR WHICH DeIVES IT. PEAL ACCEPATION = 150 ft/sz AHPLITUDE=2.310.

FIND: BEQUICED SPEED OF THE HOTOR IN PPH MAXIMUM VELOCITY OF THE TABLE (PACKAGE) IN SIMPLE HARMONIC HOTION

CHAX = XHAX WAZ 150ft/s==(2.3 ft) Wn2

W1= (782.6 rAD/3) Wn = 27.98 VAD/S fn=Wn = 27.98 = 4.452 Hz (CYCLE/S)

1 RPM = 1 CYCLE/(1min.) (605/min.) = 1/60 (HZ.)

(fHZ)/(1/60 HZ)= 4.45Z = Z67 RPM =

HAXIHUM VELOCITY NHAX=XMAXWN = (2.3 ft)(27.98 HADIS)

UHAX = 5.36 ft/s

19.9 GIVEN:

PARTICLE MOTION X= 55INZt+4coszt (m,s)

FIND:

(a) PERIOD, Tn

(b) AMPLITUDE, Y-M

(C) PHASE ANGLE, O

FOR SIMPLE HARMONIC HOTION X= Xm SIN (Wnt+0)

DOUBLE ANGLE FORMULA (TRIGONOMETRY) SIN (A+B)=(SINA) (COEB)+(SINB)(COSA)

LET A= wat , B= &

THEN X= XM SIN (WILL D)

X= Xm(DINWAt) (cos d) + Xm(SIND) (cos wat) X= (Xmcosd)(SINW, t)+(YmSIND)(COSW, t)

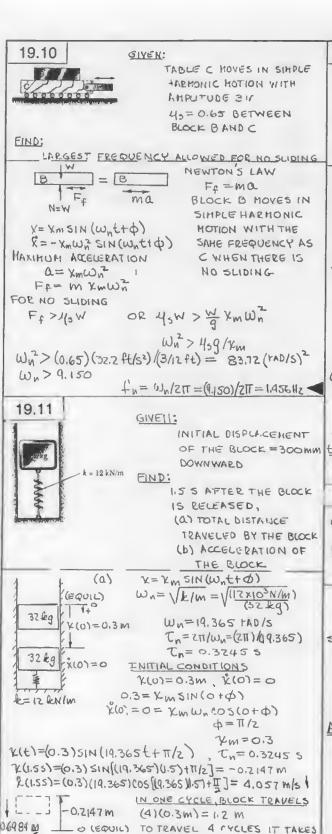
GIVEN X= 55INZt + 4coszt

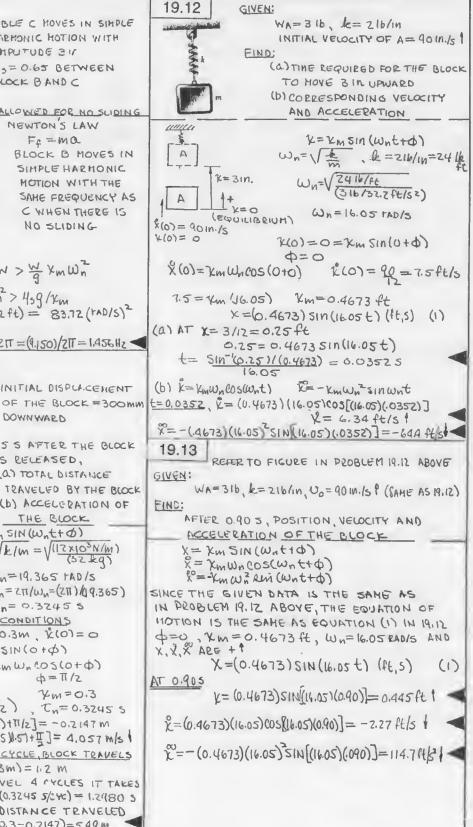
COMPARING Wn=2 Xmcosp = 5 YMSIND =4 (2)

a) Tn = 21 = 21 = TS

Xm= 6.40 m

(C) DIVIDE (2) BY (1) TAN $\phi = 4$ φ= 38.7°





(4 crc) (0,3245 5/24c) = 1.2980 5

THUS , TOTAL DISTANCE TRAVELED 15 4(1.2)+0.6+(0.3-0.2147)=5.49M

(1.5)=-(0.3)(19365)2SIN(19.365)(1.5)+= 80.5MB1

10 (b)



L= 800 mm AT t=0, 0=+5°, 0=0 ASSUME SIMPLE HARMONIC MOTION

FIND:

LGS AFTER RELEASE

(a) N AND a OF THE BOR.

 $\Theta = \Theta_m \sin(\omega_n t + \Phi)$ $\omega_n = \sqrt{\frac{9}{4}} = \sqrt{\frac{9.81 \text{ m/s}}{0.8 \text{ m}}}$ $\Theta(\omega) = S^0 = (S) (\Pi)/180 \text{ PAD}$ $\omega_n = 3.502 \text{ FAD/S}$ $\Theta(\omega) = 0$

0(0) = 517 = 0m 51N(0+4)

 $\hat{\Theta}(0) = O = \Theta_{M} \omega_{N} \cos(\omega + \hat{\phi})$ $\hat{\phi} = \pi/2$ $\Theta_{M} = \sin 2\Delta \theta$

0= 51 51N(3.502+ + 12)

 $\hat{\Theta} = \Theta_{\text{m}} \omega_{\text{n}} \cos(\omega_{\text{n}} t_{+} \phi) = \left| \frac{\sin(3.502)}{180} \right| (3.502) \cos(3.502)(1.6) + \frac{\pi}{2}$

U=10=(0.800 m)(0.19223 EAD/S)=0.19223 EAD/S

 $\vec{\Theta} = -\Theta_{m}\omega_{n}^{2} \sin(\omega_{n} t_{1} \phi) = -(\frac{5\pi}{160})(3.502)^{2} \sin[(3.502)(1.6) + \frac{\pi}{2}]$

0= -0.8319 PAD/52

a= V(Q+)2+(Qn)2

 $Q_{t} = 10^{\circ} = (0.8 \,\mathrm{m})(-0.8319 \,\mathrm{PAD/s^{2}}) = 0.6655 \,\mathrm{m}$ $Q_{t} = 10^{\circ} = (0.8 \,\mathrm{m})(0.0333 \,\mathrm{PAD/s^{2}}) = 0.6655 \,\mathrm{m}$

 $Q_{N} = \int \delta^{2} = (0.8 \, \text{m}) (0.19223 \, \text{PAD/s})^{2} = 0.02456 \, \frac{\text{m}}{52}$ $Q = \sqrt{(0.6655)^{2} + (0.02956)^{2}} = 0.6662 \, \text{m/s}^{2}$

a=0.666 m/s2



GIVEN:

M=5kg, UNATTACHED TO THE SPRING WHEN COLLAR IS PUSHED DOWN IBOMM OR HORE AND RELEASED IT LOSES CONTACT WITH THE SPRING

FIND:

(d) THE SPEING CONSTANT &

(b) POSITION, YELD CITY AND ACCELERATION

O. 165 AFTER IT IS PUSHED DOWN

180 MM AND RELEASED.

V=0 - 0=+g1 - 0=0,180 m V=0

 $k = k_{m} \sin(\omega_{n} + t + \phi)$ $k_{0} = k_{m} \sin(0 + \phi) = 0.180 \text{ m}$ $k_{0} = 0 = k_{m} \cos(0 + \phi)$ $k_{0} = 0 = k_{m} \cos(0 + \phi)$

X= 0.180 m X= 0.180 SIN (W, t+I)

WHEN THE COLLAR TUST LEAVES THE SPRING, ITS LOCKERPATION IS & LAND U=0

~=(0.180) m, cos (m,t+1)

U=0 0=(0.180)Wn cos (Wnt+11/2)

(Wnt+11/2) = 11/2

a=-g=(-0.180)(wn)2SIN(wn t+ T1/2)

 $-g = (-0.180)(\omega_n^2)$ $\omega_n = \sqrt{\frac{9.81 \text{ m/s}^2}{0.180 \text{ m}}}$ $\omega_n = 7.382 \text{ PAD/s}$

wn=Vle/m

k= mw= (5 kg)(7,332:40/5)=272.5 N/M

k=273 N/m

(b) $W_N = 7.382 \text{ RAD/S}$ $\chi = 0.180 \text{ SIN}[(7.382) + 17/2]$

At t= 0.16 s
POSITION

X=0.18051N[(7.382)(0.16)+ T/2]=0.06838M

X= 68.4 mm

BELOW EQUILIBEION FETTION

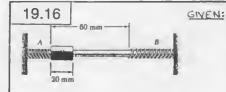
VELOCITY: \$ = (0.180) (7.382) COS[(7.382 X0.16)+ T/2]=-1.229W]

= 1.229 m/s 1

ACCELERATION

X= - (0.180X7.382) SIN[(7.382)(0.16)+ 1]]=-3.726 M

X= 3.73 m/52 1

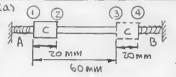


mc=8kg R = 600 N/M FOR EACH SPRING INITIAL DEFLECTION OF SPRING A = 20 mm. NO FRICTION

FIND:

(a) PERIOD

(b) POSITION OF C AFTER 1.5 5



FOR EITHER SPRING TN= ZIT V600 N/m/8kg Tn=0.72555

COMPLETE CYCLE 15 1234 4371

TIME FROM I TO 2 IS To/4 WHICH IS THE SAME AS TIME FROM 3 TO9, 4 TO3 AND 2 TO1 THUS THE TIME DURING WHICH THE SPRINGS ARE COMPRESSED IS 4 (Cn/4) = Tn = 0.72555 YELCCITY AT Z OR 3). U,=0 T,=0 V,=1 & x2=1 (600 N/m) (0.020m)2

$$T_2 = \frac{1}{2} m V_2^2 = \frac{1}{2} (8 \pm 8) (V_2)^2$$
 $T_2 = 4 V_2^2$

Ti+Vi=T2+1/2 0+0:20=402 0=0.1732 m/s

t = (0.020m) =0.115455 TIME FROM 2 TO 3 15

AND IS THE SAME AS THE TIME FROM 3 TO 2 THUS

> TOTAL TIME FOR A COMPLETE CYCLE IS T = Tn+2t2-3=0.7255+2(0.11545)=0.9564

> > Tc= 0.9565

(b) FROM (a), IN 0.9564 THE SPRING A IS AGAIN (a) DETERMINE THE CONSTANT & OF A SINGLE SPING FULLY PCHPRESSED. SPRING B IS COMPRESSED THE SECOND TIME IN IS CYCLES DE(1.5) (0.9564) = 1.4346 S. AT 1.5 S THE COLLAR IS STILL IN CONTACT WITH SPRING B HOVING TO THE LEFT IS AT A DISTANCE DX FROM THE MAXIMUM DEFLECTION OF B EQUAL TO AX = 20 - 20 cos [211_(1.5-1.4346)]

AX= 20 - 16.877 = 3.123 mm

THUS COLLAR C 15 60-3.123 = 56.877 MM FROM ITS INITIAL POSITION

> 569 MM FROH INITIAL POSITION

19.17 16 kN/m 8 kN/m 3 B kN/m

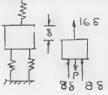
GIVEN: HASS AND SPRINGS AS SHOWN AFTER THE HASS IS PULLED DOWN AND RELEASED FROM REST THE AMPLITUDE OF THE RESULTING MOTION IS 45 mm

FIND:

(a) THE PERIOD AND FREQUENCY OF THE PUTION

(b) THE PHYILDH LEDCITY AND ACCELERATION OF THE BLOCK

(A) DETERMINE THE CONSTANT & OF A SINGLE SPRING EQUIVALENT TO THE THREE SPRINGS



k= 32 kN/m Wn=V-12/m = V32X103N/m 35kg

(IN=1/kg·Im/s2) Wn= 30.237 rAD/s

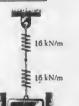
P= & S= 16 S+BS+BS

Tn= 211/Wn=211/3023 = 0.2085 fn=1/cn= 4.81HZ

(b) K= KmSIN(Watto) K0=0.045m= Xm Wn= 30.24 PAD/S

V= 0.0455IN (30.24++4) X=(0.045)(3024)COS (20.24+14) N=1.361 m/s 2=-(0.045)(30.257)374(30.24+60) QMAX=41.1.5

19.18



GIVEN: HASS AND SPRINGS AS SHOWN

AMPLITUDE OF MOTION 15 45 MM AFTER HASS IS FULLED DOWN AND RETERSED FROM REST

FIND:

(a) PERIOD AND PREQUENCY OF MOTION

(b) HAXIHUM YELOCITY AND ACCHERATION

EQUIVALENT TO THE TWO SPRINGS SHOWN S= 5,+82= P + P = 2,+8, le = 8 LN/m (2+82 CN= ZIT = ZIT

fu= Tn= 0.416= 2.41 HZ

(b) W=211fn=211(7.41)=15.12 KAD/S X= 0.045 SIN (15.12++0) &= (0.045) (15.12) cos (15.12+to)

NMAX= 0.680 m/s 1=-(0.042)(15.15), SIN(12.15f+0)

QHAX = 10.29 M/s2



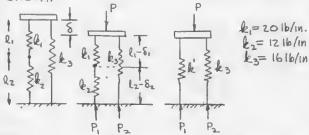
30 16 BLOCK AT t=0, %= 1.75 In. DOWN WARD, U= 0

FIND:

(a) PERIOD AND FREQUENCY OF MOTION

(b) MAXIMUM DELOCITY AND **LCCELERATION**

DETERMINE THE CONSTANT & OF A SINGLE SPRING EQUITALEST TO THE THREE SPRINGS CHOW!!



SPEINGS I AND Z (FORCE IN EACH SPRING IS THE SAME) 8= 81+82 $P_1 = P_1 + P_1$

& 15 THE SPRING CONSTANT OF A SINGLE SPEING EQUIVALENT TO SPEINGS I AND Z SPRINGS & AND 3 (DEFLECTION IN EACH SPRING IS THE SAME) P= RS P= RO B=R35 P=P,+P2

les= le' = + le = = k=k+l= k, l2+k3

R=(20)(12) + 16 = 23.5 lb/in = 282 lb/ft

(a) $C_n = \frac{zTT}{\sqrt{k/m}} = \frac{zTT}{\sqrt{k/m}}$ = 0.3615(2821/2/ft) (3016/32.2 ft/s2

fn= 1/0.361 = 2.77 Hz

(b) X=Xm SIN (Wnt+ 0)

Km=1.75 In = 0.1458 ft

K=(0.1458)(SIN 17 41++6)

 $\vec{X} = (0.1458)(17.40) SIN(17.41 + +6)$

NHAX= (0:458)(17.41) = 2.54 A/s

X= - (0,1458)(17.40)2 cos(0,1741++b) GHAX = 44.14 ALS

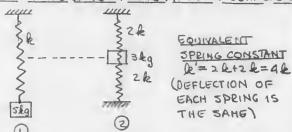
anx= 441 11/52

19.20 GIVEN:

J-LO BLOCK ATTACHED TO A SPRING FIXED AT THE OTHER END VIBRATES WITH A PERIOD TO = 685 SPEING CONSTANT & IS INVERSELY PROPORTIONAL TO THE SPRINGS LENGTH.

FINO:

THE PERLOD FOR A 3 kg BLOCK ATTACHED TO THE CENTER OF THE SAME SPRING FIXED AT BOTH ENDS



(TW-6.8=211/16/15/2) (Cn)=211/14/6/28)

k=(217) (5kg) (6.85)3 k= 4.269 N/M

(Tu)=211/V44.269 N/m X34) (Cn) = 2.63 5

19.21 GIVEN:

> SYSTEM AS SHOWN IS HOVED O. 8 IN. DOWNWARD AND RELEASED FROM PERIOD FOR RESULTING HOTION 15 Cn= 1.55 FIND:

(a) CONSTANT &

(b) MAXIMUM VELOCITY AND ACCELERATION OF THE BLOCK

SINCE THE FORCE IN EACH SPRING IS THE SAHE, THE CONSTANT & OF A SINGLE EQUIVALENT SPRING IS le= k/2.5 (SEE PROB 19.19) 九=立·七七

 $\omega_n = 2\pi f_n = (2\pi)(2.77) = 740 \frac{1}{3}$ (a) $C_n = 1.5 s = 2\pi / \sqrt{k'/m}$; $k = (2\pi) / \sqrt{k'/m}$ L= (311) (3016) (2.5) = 40.868 16/ft R= 40.9 16/ft

(b) Y= KmsIN(Wnt+\$) &= xmwnews(wnt+0) UHAX= Xmwn Wn = 21 = 211 = 4.189 eaols

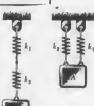
YM= 0.811.= 0.06667ft UHAX = (,06667 FE) (4189 RAD/S) WHAX=0.7279 \$H5◀

X=-xmwn2cos(wnt++)

(GHAX = 4mw) = (0.06667 FE) (4.189)

1 anax = 1.170 ft/s-





PERIOD FOR SPRINGS IN SERIES T3= 55 PERIOD FOR SPRINGS IN PARALLEL, Tp=25

FIND:

RATIO OF SPRING CONSTANTS RILL

19.23 CONTINUED

(0.6049)(k,+1.2)= k,

le,=1.338 leN/m

(b)
$$T = \frac{2\pi}{\sqrt{k_1/m}} (0.9 \text{ s})^{\frac{2}{3}} \frac{(2\pi)^3 m}{(1.838 \times 10^2 \text{ N/m})}$$

 $M = (0.95)^{2}(1.838 \times 10^{3} \text{ N/m})$

m = 37.7 kg

EQUIVALENT SPRINGS

$$T_{S} = \frac{2\Pi}{W_{S}} = \frac{2\Pi}{Vk_{S}/m} \qquad T_{P} = \frac{2\Pi}{W_{P}} = \frac{2\Pi}{Vk_{P}/m}$$

$$\left(\frac{T_3}{T_P}\right)^2 = \left(\frac{5}{2}\right)^2 = \frac{k_P}{k_S} = \frac{(k_1 + k_2)}{(k_1 k_2)/(k_1 k_2)} = \frac{(k_1 + k_2)}{k_1 k_2}$$

$$k_1 = (4.25)k_2 \mp \sqrt{(4.25)^2k_2^2 - 4k_2^2}$$

$$k_1 = 2.125 + \sqrt{3.516}$$
 k_2
 $k_1/k_2 = 4$

19.23 GIVEN:



PERICO = 0.75, = T AFTER & IS REHOVED PERIOD = 0.95. = T'

FIND:

(a) k,

(b) HASS OF A

 $k_z = 1.2 \,\mathrm{kN/m}$

EPLIVALENT SPRINGS

F,= &, 5 F2= &2 8

Fit $F_1+F_2=F=keS$ $k_1S+k_2S=k_2S$ $k_2=k_1+k_2$

$$\frac{T}{T'} = \frac{0.7}{0.9} = \sqrt{\frac{k_1}{k_2}} = \sqrt{\frac{k_1}{k_1 + k_2}}$$

$$\left(\frac{7}{9}\right)^{\frac{2}{3}} = 0.6049 = \frac{k_1}{k_1 + 1.2}$$

19.24 GIYEN:



PERIOD FOR SYSTEM SHOWN 15 T= 1.6 s PERIOD AFTER A 7- kg COLLAR IS PLACE DON A, IS T'= 2.15

FIND:

(a) HASS OF A (b) k

T= 211 - 1.62 INITIALLY

AFTER 1 & MASS IS ADDED TO A,

$$\sigma_i = \frac{2\tau}{\sqrt{k/(m_i + \tau)}} = 2.1 \text{ s}$$

$$\frac{T'}{T} = \sqrt{\frac{(m_{A}+7)}{m_{A}}}$$

$$\left(\frac{2.1}{1.6}\right)^2 = \frac{M_1 + 7}{M_A}$$

(1.7227) (MA) = MA+ 7

MA= 9.69 leg

(b)

$$k = (2\pi)^{2} (m_{h})/(\tau)^{2}$$

$$k = (2\pi)^{2} (9.69 \text{ kg})/(1.65)^{2}$$

k= 149.4 N/m



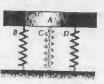
FOR SYSTEM SHOWN
PERIOD C = 0.2 S
AFTER & IS REMOVED
AND BLOCK A IS CONNECTED
TO & PERIOD C = 0.12 S



FIND:

(a) &1 (b) WEIGHT OF BLOCK A. 19.26

GIVEN:



WA=1001b

RB=RB=R=1701b/ft

FREQUENCY REHAINS

THE SAHE WHEN AN

BOILD BLOCK IS ADDED

TO A AND A SPRING OF

COMMETALITY & IS ADDED

TO THE SYSTEM

FIND: BC

EQUIVALENT SPRING CONSTANT FOR SPRINGS IN SERIES.

$$ke = \frac{k_1 k_2}{(k_1 + k_2)}$$

FOR
$$k$$
, AND k_2

$$T = \frac{2\pi}{\sqrt{k_e/m_A}} = \frac{2\pi}{\sqrt{(k_1 k_2)/(m_A)(k_1 + k_2)}}$$

FOR R, ALONE

(a)
$$\frac{T}{T_1} = \sqrt{\frac{(k_1 + k_2)(k_1)}{(k_1 k_2)}} = \sqrt{\frac{k_1 + k_2}{k_2}}$$
 $k_2 \left(\frac{T_1}{T_1}\right)^2 = k_1 + k_2$

(b)
$$T' = \frac{2\pi}{\sqrt{k_1/m_A}} \qquad m_A = W_A/g$$

$$W_{\Lambda} = \frac{(32.2 \text{ ft/s}^2)(0.12 \text{ s})^2(426.7 \text{ lb/ft})}{(2\pi)^2}$$

WA = 5.01 lb

FREQUENCY OF THE ORIGINAL SYSTEM

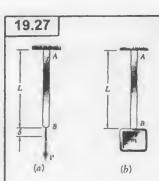
SPRINGS BANDO ARE IN PARALLEL

$$\omega_n^2 = \frac{ke}{mA} = \frac{2401r/ft}{(10016/32.2 ft/s^2)}$$

FREQUENCY OF NEW SYSTEM

SPRINGS A, BANDC ARE IN PARALLEL

&c= 197.0 16/ft



S= PL/AE L= 450 mm E= 200 GPa ROD DIAMETER

= Emm. m= 8kg

EIND:

(a) EQUIVALENT

SPRING CONSTANT

OF THE ROD, (kg)

(b) FREQUENCY OF

"LEET ICAL VIRENIONS

OF THE 8-22 MASS

(0)
$$P = ke S$$
 $S = PL/AF, P = (AE) S$
 $ke = \frac{1}{L}$ $A = \pi c^2/4 = \frac{\pi}{100} (8 \times 10^3 m)^2/4$
 $A = 5.027 \times 10^{-5} m^2$
 $E = 200 \times 10^9 \text{ N/m}^2$
 $ke = (5.027 \times 10^5 \text{ m}^2)(200 \times 10^9 \text{ N/m}^2)$
 (0.450 m)
 $ke = 22.34 \times 10^6 \text{ N/m}$

(b)
$$f_n = \frac{\sqrt{ke/m}}{2\pi} = \sqrt{22.3 \times 10^6/8} = 265.96 \text{ Hz}$$

fn= 266HZ

19.28

So = $PL^{3}/3EI$ L = 10 ft $E = 29 \times 10^{6} 16/10^{2}$ $I = 12.4 \times 10^{4}$

FIND:

(a) EQUIVALENT EPRING CONSTANT (22)

(b) FREQUENCY OF

(a) $P = k_e S_B = S_B = PL^3/3EI$, $P = (3EI) S_B$ $k_e = \frac{3EI}{L^3} = (3)(29 \times 10^6 lb/m^2)(12.4 ln^4)$ $(10 \times 12 ln)^3$

ke= 624.3 lb/in

ke=624.3 lb/in.

(b) $f_n = \sqrt{\frac{ke/m}{2\pi}}$ $k_2 = 624.3 \, lb/in$ $= 7.49 \, 2 \times 10^3 \, lb/ft$ $f_n = \sqrt{\frac{(7.492 \times 10^2 \, lb/ft)}{2\pi}}$ $= 7.49 \, 2 \times 10^3 \, lb/ft$ $f_n = 3.428 \, H_3$

fn= 2.43 Hz

19.29

GIYEN:

STATIC DEFLECTION OF THE FLOOR OF A BUILDING UNDER AN 8200-16 PIECE OF HACHINERY EQUALS &= 1.6 In.

FIND:

(A) EQUIVALENT SPRING CONSTANT & E (b) THE SPEED IN RPM OF THE HACHINGRY THAT SHOULD BE AVOIDED SO AS NOT TO COINCIDE WITH THE NATURAL FREQUENCY OF THE SYSTEM.

W= 8200 lb / \$= 1.6 in.

(a) W= ke So ke= W - 82001b 1.61n.

(b) $f_{N} = \sqrt{\frac{ke/m}{2\pi}} = \sqrt{\frac{5130 \times 12 \text{ lb/ft}}{(8200 \text{ lb/32.2 ft/s}^2)}}$

fn= 2.473 Hz 1 Hz= 1 CYCLE/5 = 60 RPM

SPEED=(2.473 HZ) (60 RPM) = 148.4 RPM

19.30

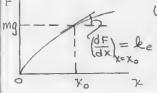
GIVEN:

FORCE-DEFLECTION EQUATION FOR A HON-LINEAR SPRING, $F = 5x^N$ (N,m)

FIND:

(a) STATIC DEFLECTION X. UNDER A
120-9 BLOCK

(b) FREQUENCY OF VIBRATION OF THE BLOCK FOR SHALL OSCILLATIONS



(a) $mg = (0.120 kg)(9.81 m/s^2)$ mg = 1.177 N $F = mg = 5 \times 6^{1/2}$ $V_0 = (1.177)^2 = 0.0554 m$

76= 55.4 mm

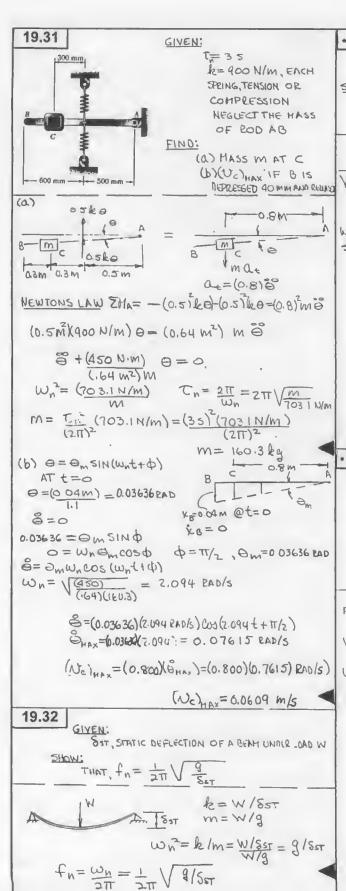
(b) AT x_0 , $\left(\frac{dF}{dx}\right)_{x_0} = \frac{5}{5}(x_0)^{\frac{1}{2}} = \frac{5}{2}(.0554)^{-\frac{1}{2}}$ $\left(\frac{dF}{dx}\right)_{x_0} = 10.618 \text{ N/m}$

he= 10.618 N/m

 $f_{N} = \sqrt{\frac{he/m}{2\pi}} = \sqrt{\frac{(0.618 \text{ N/m})(.120 \text{kg})}{2\pi}}$

fn= 1.4971 HZ

fn= 1.497 Hz.



19.33 GIVEN: $C_{N} = 4\sqrt{\frac{Q}{g}} \int_{0}^{\pi/2} \frac{d\phi}{\sqrt{sin^{3}\theta_{m}/2} - sin^{3}\phi}$ Show:

BY EXPANDING THE INTEGRAND OF THE ABOVE EQUATION, $C_{N} = 2\pi\sqrt{\frac{Q}{g}} \left(1 + \frac{1}{4} sin^{3}\frac{\theta_{m}}{2}\right)$ USING THE BINOHIAL THEOREM, WE WRITE $\frac{1}{\sqrt{1 - sin^{3}(\theta_{m})} sin^{3}\phi} = \left[1 - sin^{3}(\theta_{m}) sin\phi\right]^{\frac{1}{2}}$ $= 1 + \frac{1}{2} sin^{3}\theta_{m} sin^{3}\phi + \cdots$ WHERE WE NEGLECT TERMS OF ORDER HIGHER THAN 2 SETTING $sin^{3}\phi = \frac{1}{2} \left(1 - cos 2\phi\right)$, we have $C_{N} = 4\sqrt{\frac{Q}{g}} \int_{0}^{2\pi} \left[1 + \frac{1}{4} sin^{3}\frac{\theta_{m}}{2} \left[\frac{1}{4} (1 - cos 2\phi)\right]\right] d\phi$ $= 4\sqrt{\frac{Q}{g}} \int_{0}^{2\pi} \left[1 + \frac{1}{4} sin^{3}\frac{\theta_{m}}{2} - \frac{1}{4} sin^{3}\frac{\theta_{m}}{2} cos 2\phi\right] d\phi$ $= 4\sqrt{\frac{Q}{g}} \int_{0}^{2\pi} \left[1 + \frac{1}{4} sin^{3}\frac{\theta_{m}}{2} - \frac{1}{8} sin^{3}\frac{\theta_{m}}{2} sin^{3}\frac$

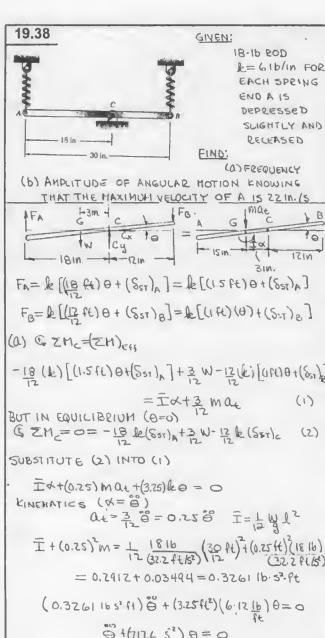
AMPLITUDE OM OF A PENDUUM FOR
WHICH THE PERIOD OF A SIMPLE
DENDULUM IS & PERCENT LONGER THAN
THE PERIOD OF THE SAME PENDULUM FOR
SMALL OSCICLATIONS

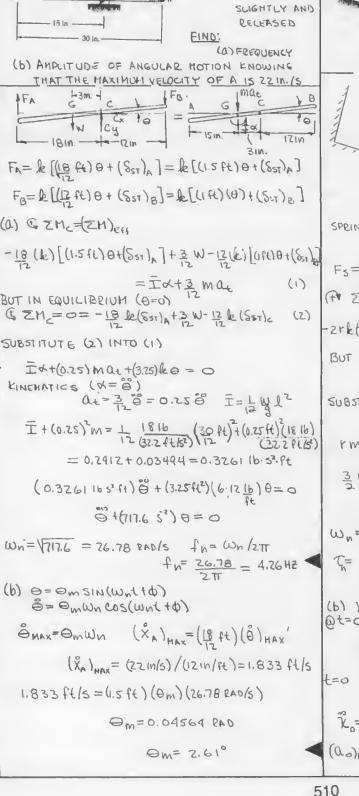
FOR SHALL Ø SCILLATIONS $(T_n)_0 = 2\pi \sqrt{\frac{9}{9}}$ WE WANT $T_n = 1.005(T_n)_0 = 1.0052\pi \sqrt{\frac{9}{9}}$ USING THE FORHULA OF PROB 19.33, WE WRITE $T_n = (T_n)_0 (1 + \frac{1}{4} \sin^2 \frac{9}{2} m) = 1.005(T_n)_0$ $51N^2 \frac{9}{2}m = 4[1.005-1] = 0.02$ $\frac{9}{2}m = 8.130^\circ$

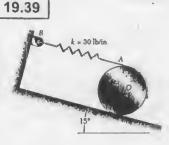
Om= 16.30

19.35 GIVEN 19.37 CONTINUED DATA OF TABLE 19.1 IN HOTION TO ZMA = (ZMA) ecc PENDULUM LENGTH, Q=750 mm (0.7) [kg[0.70+(ssr)d-mq]+1.4[kc(1.40+(ssr)c]= FIND: (A) PERIOD FOR SHALL OSCILLATIONS (b) PERIOD FOR AMPLITUDE O == 60° -IX-(0.7)(max) (C) DERIOD FOR AMPLITUDE OM= 900 BUT IN EQUILIBRIUH (8=0) (a) T= 211/ Q AZHA=0=0.7[kg(Ssr)&-mg]+1.4 kc(Ssr)c (2) (EQ. 19.18 FOR SHALL OSCILLATIONS) $T_n = 2\pi \sqrt{0.750 \, \text{m}} = 1.737 \, \text{s}$ SUBSTITUTING (2) INTO (1) Th=1.7375 IX+0.7 mat + (0.7) Le 0+(1.4) Le 0=0 (b) FOR LARGE OSCILLATIONS (EQ. 19.20) VINEHATICS (4=8)

Q=0.74 = 0.76° Cn=(2k)(211/ 0) = 2k (1.7375) FOR 0 = 60° (= 1.686 (TABLE 19.1) [I+ m(0.7)2] & +(0.7)2 kg+(14)2 kg) 0=0 Tu(60°)=2(1.686)(1.7375)=1.8645 I= 12 m l= 12 (5 lg) (14 m)= 0.8167 lg-m2 Tul60)=1.8645 (c) for 0 = 90°, K= 1.854 (.7)2m=(0.49 m2)(5 kg)=2.45 kg-m2 Th= 2 (1.854)(1.7375) = 2.055 (0.7)2 kg+ (1.4)2kc= (.49 m2)(500 N/m)+(196 m2)(6204) . 19.36 GIVEN: DATA OF TABLE 19.1 PERIOD = 2 S. AMPLITUDE = 90° = 245+1215.2=1460.2 N.M FINO: [0.8167+2.45] 0+1460.20 = 0 LENGTH LOF A SIMPLE PENDULUM (In.) FOR LARGE OSCILLATIONS (EQ 19.20) \$ + (460.7 N·m) = = 0 FOR Om=900 CN= (学)(5山/番 K=1.854 (TABLE 19.1) 0 + 4470 = 0 $(\frac{N}{kg} = 5^{-2})$ 2=(25)(322ft/s2) = 2,342 ft N= 28.1 IN. (4)(1.854)72 W= \4475 = 21.14 PAD/S 19.37 fn= Wn = 21.14 = 3.36Hz GIVEN: 5-kg ROD AC SPRING B, &= 500N/W (b) == = m SINWn++++) SPRINGC, R=620N/W 0=(0,)(wn) cos (wnt+0) (TENSION OR COMP.) FIND: HAXIHUH ANGULAR VELOCITY, Om= Om WIN WHEN END C IS DEPOESSED SLIGHTLY (a) FREQUENCY OF HAXIMUM VELOCITY AT C VIGRATION (1.4 Bm= (1.4 m) (0 m) (wn) (b) AMPLITUDE OF POINT C KNOWING $\theta_{m} = \frac{(0.9 \text{ m/s})}{(1.4 \text{ m})(21.14 \text{ PAD/s})} = 0.03041 \text{ RAD} \quad \omega_{n} = 21.14 \text{ PAD/s}$ THAT ITS HAXINUM VELOCITY IS 0,9 m/s HAVIHUM AMPLITUDE AT C $(X_c)_m = (1.4 \text{ m})(\Theta_m) = (1.4 \text{ m})(0.03041)$ FB= kg (28+(SET)0) = kg (0.70+(SET)B) (Xc) = 0.0426 m Fc=lec(xc+(SsT)c)=lec(1.40+(SsT)c)







GIVEN: 30-16 CYLINDER ROLLS WITHOUT SLIDING. DATA AS SHOWN. INITIAL DISDUCCHEU = ZIT DOWN

(0) FIND: (a) PERIOD MUHIXAH (d) ACCELEBATION OF C

EQUIL POSITION

SPRING DEFLECTION KA= YO+ KA/O KAIO= re 0= Yolr XA = ZXO F5 = le (XA+S5) = le (276+557)

(FY ZMc=(ZM)ORC

BUT IN EQUILIBRIUM, X0=0 ZM=0=-2r&SattrWSINIS

SUBSTITUTE (2) INTO (1) AND NOTING THAT @= Xo/r, 0 = Xo/r

rmx0+Ix0+4rhx0=0 I=1mr2 3 mrx +4rkx =0

\$\frac{8}{3} \frac{1}{100} \frac{1}{3} = 0 Wn= \ \ \frac{3}{8}\text{M} = \(\left(8)(30x12 \text{lb/st})\) = 32.1 5

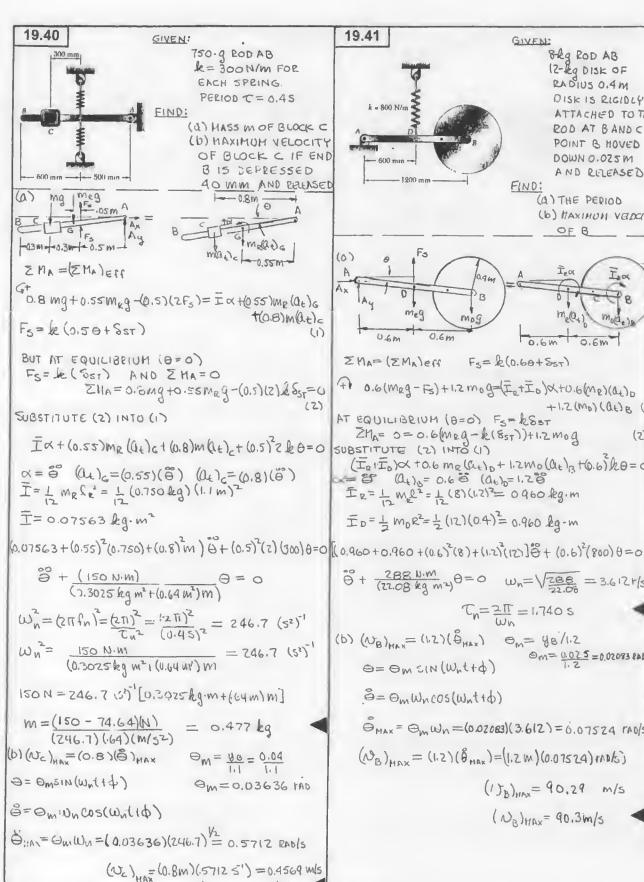
두 - 프 = - 0.1957 5

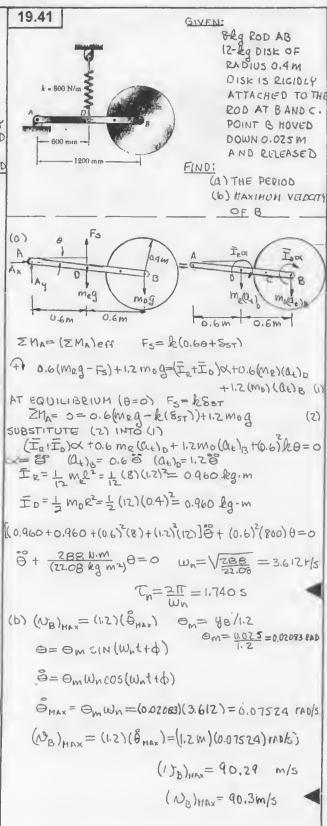
Tn=0.19575 (b) x0=(x0)m SIN (wnt+0) Qt=0 x0= 2 ft 20=0

20=(x0),w, cos(w,t+0) t=0,0=(x0),w,1050 THUS 4= 17/2 t=0 x0(0)=1 ft=(x0)m51Np=(x0)m(1)

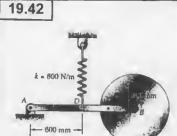
(Yolm= & ft 1 = - (Ko) ... W 2 SIN (W + 4)

(Qo) MAX= (Xo) MAX= -(XO) W W = -(fit) (32.15-1)= 171.7 ft/s2 (Qu) HAX= 171.7 (1/524



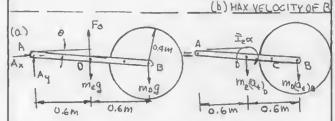


(Nc) HAX = 457 1941



8kg ROD AB 12 Kg DISK OF PADIUS 9.4 M PIN C REMOVED AND DISK CAN ROTATE FREELY ABOUT PIN B. POINT B HOVED DOWN O. UZS IM AND PELEASED

FIND: (a) PER100



NOTE: THIS PROBLEM IS THE SAME AS PROB 19,41 EXCEPT THAT THE DISK DOES NOT RUTATE. SO THAT THE EFFECTIVE HOHENT IN = 0. ZMA=(ZMWeff Fs=k(0,60+5=T)

(0.6) (meg-Fs)+1.2 mpg= Izx+(i.6)(me)(a)0 +1.2(Mo)(at)e (1)

AT EQUILIBRIUM (0=0) FSTEEST 2 ZIIA-0=0.6 (MEg-SST) T1.2 Mag (5)

SUBSTITUTE (2) INTO (1) = 0 (at) = 0.60 (at) = 1.20 (at) = 0.960 kg m

[0.960+(0.6)2(8)+(1.2)]12)]0+(0.6)2(800)0=0

 $\Theta + \frac{(288 \text{ N·m})}{21.12 \text{ kg·m}^2} \Theta = 0 \quad W_M = \sqrt{\frac{288}{21.17}} = 3.693 \text{ g}$

 $T_{N} = \frac{211}{W_{N}} = \frac{211}{3.693} = 1.7015$

(b) (UB) = (1.2) (B) MAX $\Theta_{M} = \frac{40}{1.2} = \frac{.025}{1.20} = 0.02003$ and

0=0m CIN (W, t+0)

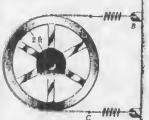
= 0 mWn cos(wnt+ 0)

ONAX = OMWn = (0.02083)(3.(93) = 0.07694 MADIS

(b) MAX = (1.2)(3,1x)=0.2 1.07694)=0.9233 m

(NB) = 92.3 mm/s

19,43

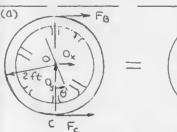


GIVEN:

600-16 FLYWHEEL OF PADIUS OF GYPATION=201 &=75 lb/in, FOR EACH SPRING. POINT C IS PULLED TO THE RIGHT I IM. AND RELEASED

EIND:

(a) PERIOD OF VIBRATION (b) HAXIMUM ANGULAE VERCETT OF THE FLYWAR



ZM=(ZKa), eff Fc=k((8st)-20) FB=k(20+(8st)B)

AT EQUILIBRIUM (0=0) Fo= & (65) , Fc= & (65) e

$$ZH_A=0=2k(\delta_{ST})_c-2k(\delta_{ST})_B$$
 (2)

SUBSTITUTE (2) INTO (1)

 $\overline{T} = m k^2 = \frac{(600 \text{ lb})(20/12 \text{ ft})^2}{(32.2 \text{ ft/s}^2)} = 51.76 \text{ lb ft·s}^2$

R=(84E)(75×12 1b/ft)= 7200 1b.ft

$$C_N = \frac{2\pi}{\omega_N} = \frac{2\pi}{\sqrt{139.1}} = 0.533 \text{ s}$$

(b) 0=0msin(watto)

&= Omwn cos(watto)

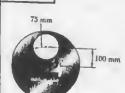
$$\Theta_{\text{MAX}} = \Theta_{\text{M}} W_{\text{N}}$$

$$\Theta_{\text{M}} = \frac{1/12}{2}$$

$$W_{\text{N}} = \frac{\sqrt{139.1}}{11.79 \text{ BAD/S}}$$

$$\Theta_{\text{M}} = 0.04167 \text{ PAD}$$

(1) = 0.491 TAD/S

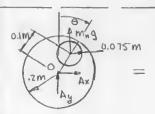


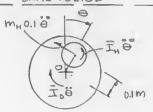
GIVEN:

DISK ATTACHED TO A
FRICTIONLESS PIN AT ITS
GEOMETRIC CENTER
AS SHOWN

FIND:

(a) PERIOD OF SHALL
OSCILLATIONS
(b) LENGTH OF A SIMPLE
PENDULUM OF THE
SAME PERIOD





ZMo=(ZMo)eff

 $(+ -m_{H}g(0.1)SIN\Theta = \bar{I}_{0} - \bar{I}_{H} - \bar{B} - (0.1)^{2}m_{H} - \bar{B}$ $m_{b} = S + \pi R^{2} = (S + \pi)(.2)^{2} + (0.04)\pi S + (0.05625)\pi S + (0.005625)\pi S$

In= 1 MH r2 = 1 (0.005625 TPL) (00)= 15.8240 TPL

SHALL ANGLES SINGRO

(0.005625) TEE (9.81)(1) 0.00 \$ (9.81)(1) 0 = 0

727.9 ×10 0+ 5.518×10 0=0

 $W_{N}^{2}: \frac{5.518 \times 10^{3}}{727.9 \times 10^{6}} = 7.581$

Wn= 2.753 RAD/S

 $T_{N} = \frac{2\pi}{\omega_{N}}$ $T_{n} = \frac{2\pi}{2.753} = 2.28 \text{ S}$

(b) PERIOD OF A SIMPLE PENDULUH

 $T_{n} = 2\pi \sqrt{l/g}$ $l = (T_{n}/2\pi)^{2}g$ $l = [(2.753)/2\pi)^{2}(9.81 \text{ m/s}^{2})$ l = 1.294 m

19.45



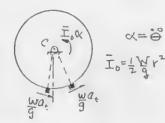
GIVEN:

WEIGHTS W AT A AND BIAND DISK W FOR B=0, PERIOD=To

EIND:

ANGLE & FOR A PERIOD OF 200

Ay CONA, B



ZMc = (\(Mc) eff

f wrsin(β-0)-wrsin(β+0)= zwrsin0 cosβ

SINDZO at= rô

(2w2+ 1 2) =+ (2wrcosp) == 0

 $\omega_{r} = \sqrt{\frac{2\omega_{9}\cos\beta}{2\omega + w/2}} = \sqrt{\frac{49\cos\beta}{4 + w/\omega}r}$ $\xi = 0 \quad C_{0} = \frac{2\pi}{\omega_{0}} = 2\pi / \sqrt{\frac{49}{4 + w/\omega}r}$

 $C_{n} = 2\pi / \sqrt{\frac{\cos \beta}{(4+\frac{1}{16})r}} = 2C_{0} = 4\pi / \sqrt{\frac{49}{4r^{1}\sqrt{\omega}}r}$ $\cos \beta = \left(\frac{1}{2}\right)^{2} = \frac{1}{4}$

B=75.50

19.46 REFER TO FIGURE IN PROB 19.45

GIVEN:

w= 0.116 , W= 316, r= 4 in. , B=60°

FIND:

FREQUENCY OF SHALL OSCILLT 1016

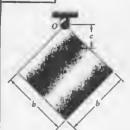
FROM DERIVATION IN PROB 19.45 (EQ. 1)

 $W_{N} = \sqrt{\frac{(A)(37.2 \text{ ft/s}^2)(0560^\circ}{(4+3/0.1)(4/12)}} = 2.384 \text{ f/s}$

fn= Wn/2TT= 2.384/2TT

1'n= 0.379 HZ

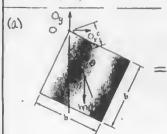
GIVE N:

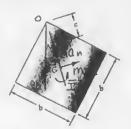


SQUARE PLATE, b=0.3m

FIND:

(a) PERIOD OF SMALL OSCILLATIONS ABOUT O (t) DISTANCE C FROM O TO A POINT A FROM 'N-ICH THE PLATE CHOULD BE SUSPENDED TO HINIHIZE THE PERIOD





Σho=(ZMo)eff α=0 α=(06)(α) OG= bY2/2 I=+mb2 Q=(b12/2) 0°

4) (DGKSINO)(mg) = -(DG) Mat-IX SINOXO brz/zm (brz/z) ++ m b + (brz/z)mg 0=0 (b)(=+6)m0+ 12/2 mg0=0 0+ (12/2) 9 0=0 b=0.3m

 $(T_n)_0 = \frac{2\pi}{(\omega_n)_0} = 2\pi \sqrt{\frac{(2/3)b}{(12b)q}} = 2\pi \sqrt{\frac{4(.3m)}{3(2)9.818}}$

(Tn)=1.0675

(b) SUSPENDED ABOUT A

ZMA=(EMA)ess at=(06-c)x

A (06-c)(sine)(mg)=-(06-c)mar-IX

((b 5/2-c) + + b2) m0 +, c (2-c) mg 0 = 0

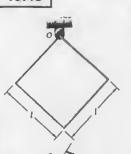
 $(C_N)_A^2 = \frac{(2\pi)^2}{\omega_N^2} = \frac{4\pi^2[(bE/2-c)^2+b^2/6]}{(bE-c)}$

FOR HINIHOM PERIOD dth la = 0 0= 2(b12/2-c)(-1)(b12-c)-(-1)[(b12/2-c)+6/6]

(bR/2-c)2+ b2/6=0 b=0.3m

briz-c= b c=03[12-1] = 0.08966m C= 89.7 MM

19.48

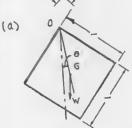


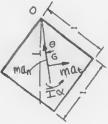
GIVEN:

THIN WIRE, L=1.2 ft

FIND:

(a) PERIOD ABOUT O (6) PERIOD ABOUT A POINT AT THE HIDPOINT OF ONE OF THE SIDES





M= MASS OF THE FRAHE Zno=(ZMo)eff x=0 4=06)(x) 06= 2 12/2. Qt=(1 12/2) 0

FOR ONE LEG (IG IG+ (MA) (R/Z)2 IC'=1 M 22 (Ic) = mol +2+(=) = m 22(3)

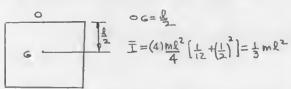
FOR COMPLETE WIRE FRAHE

I = 4 (I6),=(4) M 22(1) = 1 ml2

G-mglasino=IX+Malla SINOXO $(\frac{1}{3} + \frac{1}{2}) M L^2 \theta^2 + Mg L G/2 \theta = 0$

TN = 31 = 31 = 21 \ (312) (322) (4(52)

(b) FOR FRAME SUSPENDED FROM MIDPOINT



- mg 1 sino= IO+(m 0) 1 = (3+4) ml 0 7 ml20+ mg & 0=0

 $C_N = \frac{2\pi}{\omega_N} = \frac{2\pi}{\sqrt{6}} = 2\pi \sqrt{\frac{7}{6}(\frac{1.2 \text{ ft}}{32.2 \text{ ft/5}^2})} = 1.310 \text{ s}$

Tn= 1.310 S



UNIFORM EQUILATERAL
TRIANGLE OF SIDE L=0.3 M



EIND:

(a) PERIOD IF PLATE IS
SUSPENDED FROM ONE
OF ITS VERTICES
(b) PERIOD IF PLATE IS
SUSPENDED FROM THE
HIDPOINT OF ONE OF ITS

19.50

GIYEN:

POD AB OF NEGLIGIBLE
MASS ATTACHED TO A DISK
OF MASS M. AB=L=0,650m
F= 0:250 M

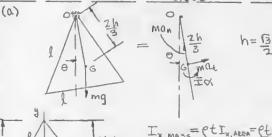
FIND:

THE PERIOD OF SMALL
OSCILLATIONS IF

(a) THE DIST IS FREE TO

ROTATE IN A BEARING AT A

(b) THE DIST IS RIVETED AT A



h e 2h/3

$$T_{V_1} MASS = \frac{M h^2}{18}$$

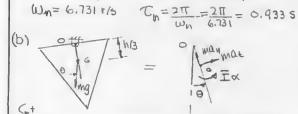
Iy, mass =
$$8tIy$$
, area Iy, area = $\frac{h\ell^3}{48}$
Iy, mass = $\frac{m\ell^2}{70}$

$$I = I = I \times Iy = \frac{Mh^2 + Ml^2}{18} \times \frac{Mh^2}{24}$$

$$h = 1.5/2 \quad \hat{I} = Ml^{2} \left[\frac{3/4}{18} + \frac{1}{24} \right] = Ml^{2}/12$$

$$\alpha = \hat{\Theta} \quad \alpha_{t} = \frac{13}{18} \hat{\Theta} \quad \text{SINO} \approx \Theta$$

$$\omega_n^2 = \frac{\sqrt{3}/3}{5/12} \frac{9}{2} = (\frac{\sqrt{3}/4}{5}) \frac{(9.81 \text{ m/s}^2)}{(0.3 \text{ m})} = 45.31 \text{ s}^2$$



ZMo=(ZHo)ess - mgh/3 sino = I 0+ m(h) 0 h= 1 3/2 I=wl/(2 (++(12)) ml0+ mgl = 0=0 20+ 12 9 =0

$$\omega_{N}^{2} = \sqrt{3} \frac{9}{2} = \sqrt{13} \frac{9.81 \, \text{m/s}^{2}}{0.3 \, \text{m}} = 56.63 \, \text{s}^{-1} \quad \omega_{N} = 7.5758 \, \text{Hs}$$

$$\tau_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{2756} = 0.835 \, \text{s}$$



R=0.650m 0H man 1=0.250m

$$I = \frac{1}{2}mr^2 = \frac{1}{2}(0.250)^2 M = \frac{M}{32}$$

$$Q = \frac{1}{2}mr^2 = \frac{1}{2}(0.250)^2 M = \frac{M}{32}$$

(A) THE DISK IS FREE TO ROTATE AND IS IN CURVICINEAR TRANSLATION
THUS IX=0

$$W_{N}^{2} = \frac{9}{1} = \frac{9.81 \text{ m/s}^{2}}{0.650 \text{ m}} = 15.092$$
 $W_{N} = 3.885$
 $C_{N} = \frac{2\pi}{W_{N}} = \frac{2\pi}{3.885} = 1.617.5$

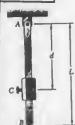
(b) WHEN THE DISK IS RIVETED AT A, IT POTATES AT AN ANGULAR ACCELERATION &

$$(\pm -mglsin0 = Ix + lmat I = \pm mr^2$$

 $(\pm mr^2 + ml^2) = + mglo = 0$

$$\omega_{N}^{2} = \frac{90}{(F^{2}/2 + 0^{2})} = \frac{(9.81 \text{ m/s}^{2})(0.650 \text{ m})}{[(0.250^{2})/2 + (0.650)^{2}]} = 14.053 \tilde{5}^{2}$$

$$T_{N} = \frac{2\pi}{W_{N}} = \frac{2\pi}{3.49} = 1.6765$$

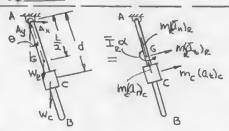


GIVEN:

COLLAR C WEIGHT, WE= 21b ROD AB WEIGHT, WE= 616, L= 3ft

FIND:

PERIOD OF SHALL OSCILLATIONS
WHEN,
(1) d= 3 ft
(b) d= 2 ft



EMA = (EMA)eff

$$\frac{1}{6} + \frac{(1/2 + \frac{1}{m_{e}}d)}{(1^{2}/3 + \frac{1}{m_{e}}d)} = 0$$

$$\Theta + (\frac{3}{3} + \frac{1}{3} d^2) = 0$$

$$T_n = 2\pi / \omega_n = 2\pi \sqrt{\frac{(3+d^2/3)}{(\frac{3}{2}+d/3)(9)}}$$

(a) d=3ft

$$T_n = 2\pi \sqrt{\frac{(3+3)}{(\frac{2}{2}+1)(32.2)}} = 1.715 \text{ s.}$$

(b) d= 2ft

$$T_N = 2\pi \sqrt{\frac{(3+4/3)}{(3/2+2/3)(32.2)}} = 1.5665$$

19.52



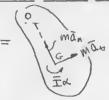
GIVEN:

COMPOUND PENDULUM WHICH OSCILLATES ABOUT O RECENTRODAL RADIUS OF GYRATION $GA = \overline{\mathbb{R}}^2/F$

SHOW THAT!

PERIOD EQUALS THE PERIOD OF A SIMPLE PENDULUM OF LENGTH OA.





$$\Theta + \frac{9F}{F^2 + \mathbb{Z}^2} SIN\Theta = 0 \qquad (1)$$

FOR A SIMPLE PENDULUM OF LENGTH OA= &



 $\Theta + Q \Theta = 0 \tag{5}$

COMPARING EQUATIONS (1)
AND (2)
= F2+12

GA=1-F= 12/F (QED)

19.53 GIVEN:

COMPOUND PENDULUH AS IN PLOB. 19.52 SHOWN ABOVE

SHOW THAT:

SHALLEST PERIOD OF COCILLATION OCCURS

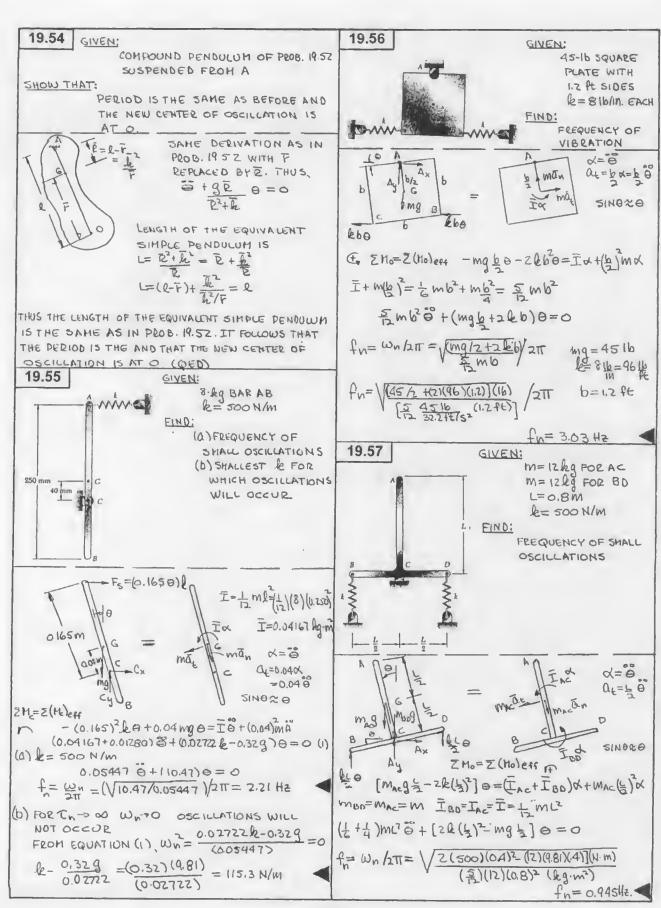
SEE SOLUTION TO PROBLESS FOR DERIVATION OF

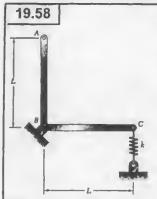
FOR SHALL OSCILLATIONS SINDED AND

The ST = 3T / F+ 12 = 2T / F+ 12 = 13 / F+ 12 / F+ 12 = 13 / F+ 12 / F

FOR SHALLEST TN WE HUST HAVE FILL

$$\frac{d(\bar{r}+\bar{k}^2)}{d\bar{r}} = -\frac{\bar{k}^2}{\bar{r}^2} = 0$$

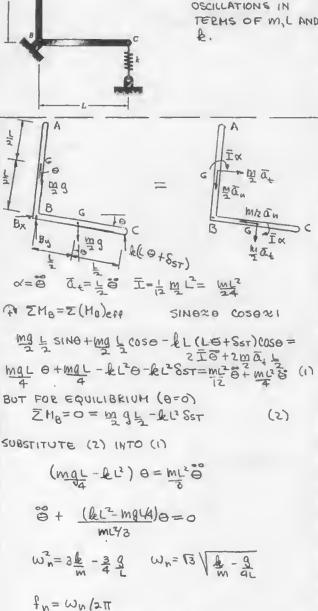




ROD ABC OF TOTAL MASS M

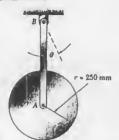
FIND:

FREQUENCY OF SHALL OSCILLATIONS IN TERMS OF MIL AND



fn= 13 / 1 - 9

19.59

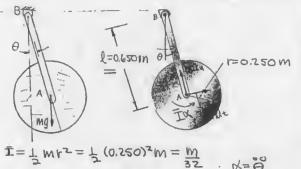


GIVEN:

ROB AB LENGTH R= 0.650 M HASS OF AB IS NEGUGIBLE AB IS DISPLACED 20 FROM THE POSITION SHOWN AND RELEASED

FIND:

MAXIMUM VELOCITY OF A IF THE DISK IS. (A) FREE TO POTATE ABOUT A (b) PIVETED TO AB AT A



(a) THE DISK IS FREE TO ROTATE AND IS IN CUEVILINEAR TRANSLATION. THUS IX=0 ZMB=Z(HB)eff

FF - mglsino = lmax

Q+= 2x=0.650 0

ml=9+mgl0=0 w= 9 FROM 19.17, THE SOLUTION TO THIS EQUATION IS e= em sin(watto)

At t=0, 0= 2. II = IT RAD, 8=0 8 = Omwncos(watto)

0 = 0 m wn cos p p = I II = OM SIN(0+II)
OM = II PAD

THUS O= I SINWALT]

(UA) HAX = lêmax = lemwn l=0.650 M Em= To wn= 13

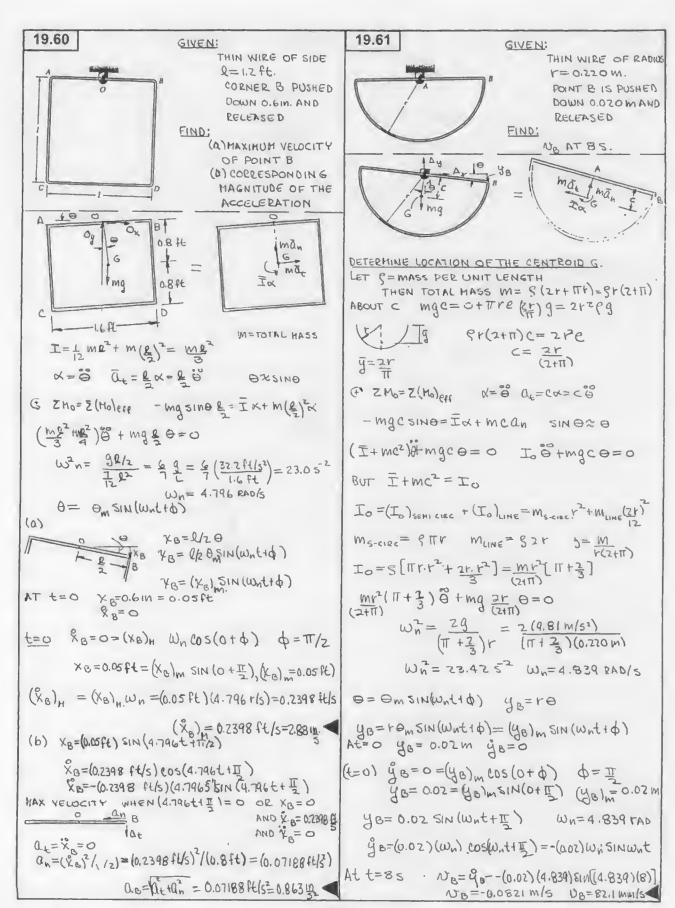
(NA) HAX = (0.650 M) (T) (V9.81M/52)

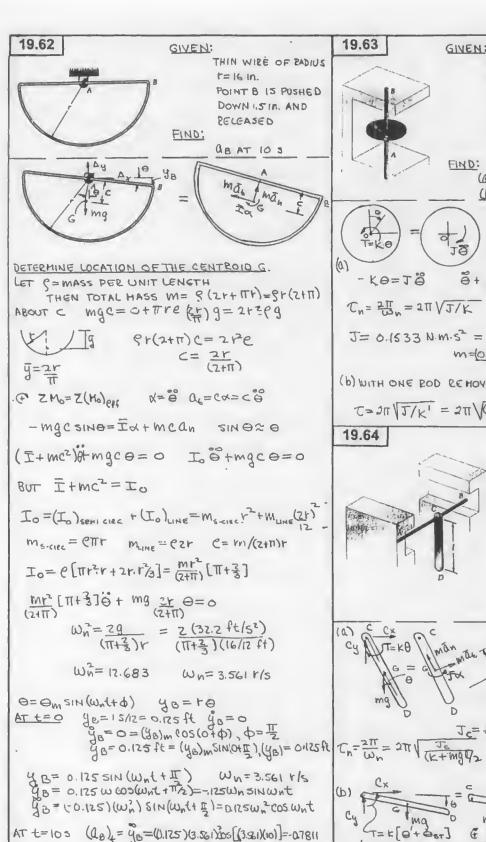
(NA) HAX = 0.08815 M/S (NY) HYX = 88" MM. b) FOR DISK RIVETED AT A (IX INCLUDED) (7 m hs+m s) Q + md s Q = 0 (HP) obt C - md s 21 MA - T ZHO=Z(Ho)eff G-mglsine=Ix+ PMa+

O= I SIN (WAT+ I) (SEE (O))

(NA) max = lomwn

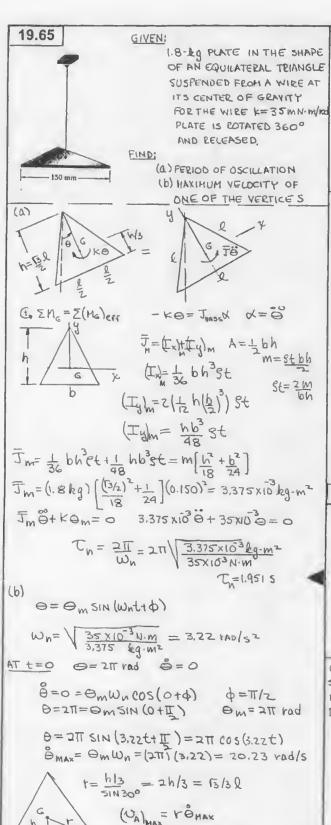
(UA) MAX = (0.650 M) ((9.81 M/s)(0.650 M) = 0.0851 M/s (0.250/2+0.650 M) (WA) MAX = 85.1 MM

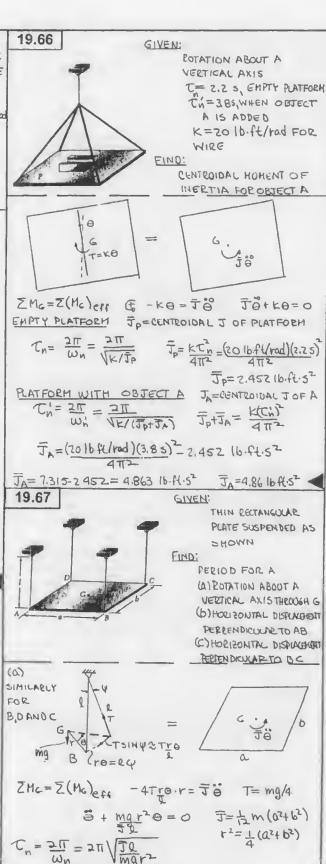




0.8 = [(00) 2+ 08] = [(0.7811) 3+ (0.3874)] = 0.789 FUS2.

19.63 GIVEN: DISK OF RADIUS 1=120 mm IS WELDED TO BOD AB WHICH IS FIXED AT A AND B. DISK POTATES B" WHEN A 500-MN·M IS APPLIED PERIOD TUE 1.35 WHEN THE COUPLE IS BEHOVED (A) THE MASS OF THE DISK (b) PERIOD IF ONE ROD IS REMOVED K= I = 0.5 NM (8)(11/180) K=3.581 N.M/tao ZMo= Z(Ho)eff 8+ K0=0 - K0=J8 (211)2 (1.35)2(35810 m/r) Cn= 恐n=211/J/K (211)2 J= 0.1533 N.m.s2 = 1mr=1m (0.120m)2 m=(0.1533 N.m.52)(2) = 21.3 leq (0.120 M) (b) WITH ONE BOD REMOVED K = K/2=3.581 =1791 UM T=211 VI/K1 = 211 VO.1533 N.m.52) = 1.8385 19.64 GIVEN: 10.16 200 CD OF LENGTH 1=2.2 ft WELDED ROD FIXED AT. A AND B WITH K= 18 lb.ft/rad FIND: PERIOD OF SHALL OSCILLATIONS IF THE EQUILIBRIUM POSITION (a) VERTICAL AS SHOWN LATHOSIPOH (d) A ZM = E(Mc)eff - KO - MG SIND= 24 HMT & x= & at=8x=8 & (J+m/2) 18 HK+mg2/2/8=0 J+W(\$/2=JC=1 MR2 Je= = (1016) (2.2 Pt) = 0.501 lbf(5) TN= 211 (0.501 16 H.52) Tn=0.8765 d=86 € 1JX T= [0+6s+] & ZMc= Z(Mc)en - K(0+0s+)+ mylk = Jo + m(l/2) = Jo (UB) = 4B=(0.125)(3561) 5(N/3 121)(10) = 0.3874 PEL BUT IN EQUILIBRIUM (0=0) ZH=0=-KO++Mg4z THUS JOB+KO=0 TN=211 = 211 \ - 1/K = 211 \ 0.501 16.4652 = 1.048 S

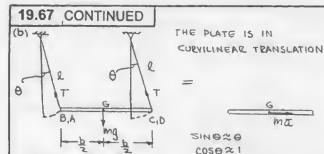




 $T_N = 2\pi \sqrt{\frac{1}{12} m (0^2 + b^2)} = 2\pi \sqrt{\frac{1}{m} g \frac{1}{4} (0^2 + b^2)} = 2\pi \sqrt{\frac{1}{m} g \frac{1}{4} (0^2 + b^2)}$

(NA)HAX=(13/3)(0.150m)(20.23 rod/s)

(NA) HAX = 1.752 M/S



Le cose ≈ Le = 7.

+ 12F=0=4 (TCOSO)-mg=0 T= mg/4

(C) SINCE THE OSCILLATION ABOUT AXES PARALLEL TO AB (AND CD) IS INDEPENDENT OF THE LENGTH OF THE SIDES OF THE DLATE, THE PERIOD OF VIBRATION ABOUT AXES PARALLEL TO BC (AND AD) IS THE SAME TO BE TO BE THE PARALLEL TO BC (AND AD) IS THE SAME

19.68

GIVEN:

2.2-kg CIECULAR DISK

r= 0.8 m

WIRE AB, K_= 10 N m/rad

WIRE BC, K_= 5 N m/rad

FIND:

PERIOD OF OSCILLATION ABOUT AXIS AC

EQUIVALENT TORSIONAL SPRING CONSTANT



 $T = k_{\theta} \theta , T = k_{2}(\theta - \theta_{1}), T = k_{1}\theta_{1}$ $k_{2}\theta = (k_{1} + k_{2})\theta_{1}$ $\theta_{1} = \underbrace{k_{2}}_{k_{1} + k_{2}} \theta$ $T = k_{\theta} \theta = k_{1}\theta$ $k_{\theta} = k_{1} \underbrace{k_{2}}_{k_{2}} \theta$

Ke = KIKZ

NEWTONS LAW



ZH=Z(H)eff (+ -ke=J0

J= = mrz

1 mr2 0 + Ke0 = 0

$$T_{N} = \frac{2\pi}{W_{N}} = \frac{2\pi}{\sqrt{\frac{2E_{e}}{Mr^{2}}}} = 2\pi \sqrt{\frac{(2.2 \log)(0.8 m)^{2}}{2[(0.15)](0+5)]N-M}}$$

Tn= 2.895

19.69



GIVEN:

PARTICLE WHICH HOVES WITHOUT FRICTION INSIDE A CURYED SURFACE

FIND:

PERIOD OF SHALL OSCILLATIONS

DATUM AT (2)

R(1-cosem)

POSITION D

V=WE(1-c030m)
3 SMALL OSCILLATIONS
(1-C050m)=ZSIN20m20/2

VI= WEBZ

POSITION (2)

 $V_{m} = R \Theta m \qquad T_{2} = \frac{1}{2} m V_{m}^{2} = \frac{1}{2} m R^{2} \Theta_{m}^{2}$ $V_{3} = 0$

CONSERVATION OF ENERGY THYI=TZ+VZ

O+ WREZM = IMPZ &Z + O & = W + O M

W= Mg

W= Mg

mg Reim = Im Riwisom

TN= 211 = 211 \ 2

19.70

GIVEN:

1402. SPHERE A

I ha

810

ROD AC OF NEGUGIBLE WEIGHT

EIND:

PERIOD OF SMALL OSCILLATIONS OF THE BOD

8 in.

Sin.

Sin.

B

DATUM AT (1)

POSITION(1)

Ti = 0

Vi = Who-Waha

hc = Bo (1-cosom)

ha = Bin (1-cosom)

SHALL ANGLES

SHALL ANGLES 1-COSOM & OM/2

Vi=(Wc)(BC)-(MARI

٨١= (الراب) (الراب) الراب الراب) الراب الراب (الراب) (ال

 $V_{1} = (0.4167 - 0.3646) \underbrace{0.00}_{m} = 0.05208 \underbrace{0.00}_{m}$ $V_{2} = 0 \quad T_{2} = \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m} \underbrace{0.00}_{m} + \underbrace{1}_{m$

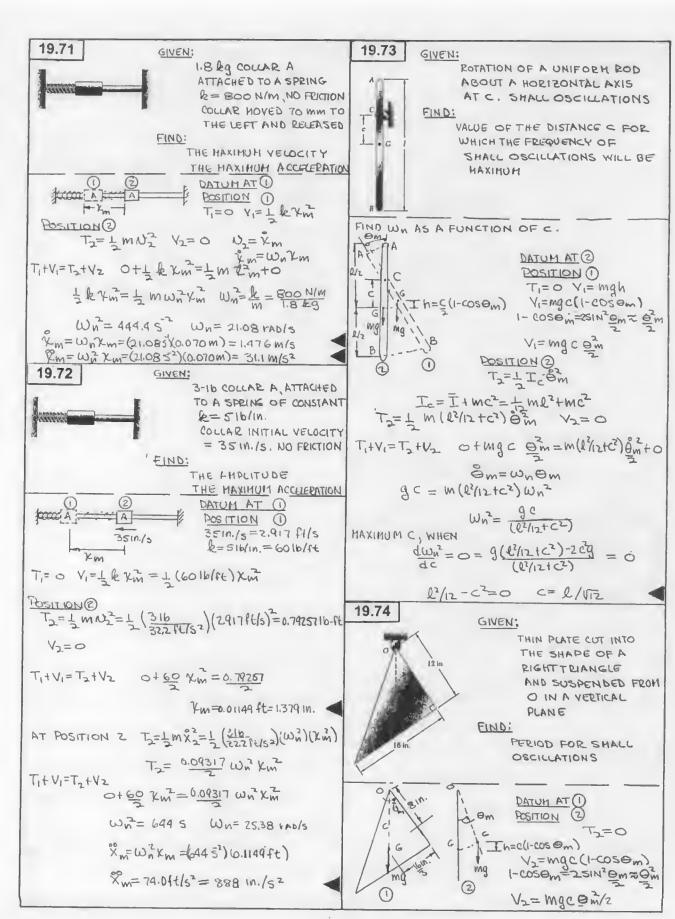
 $T_2 = \frac{1}{29} [0.2778 + 0.1519] \omega_n^2 \Theta_m^2 = \frac{1}{29} (0.4297) \omega_n^2 \Theta_n^2$

Tity= Taty2 0+0.05208 = 0.4297 Widom

 $(U_N^2 = \frac{(32.2)(.05208)}{(0.4297)} = 3.902$, $(V_N = \frac{211}{U_N} = \frac{211}{\sqrt{3.902}} = 3.185$

(m(3/2))

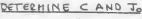
(2)

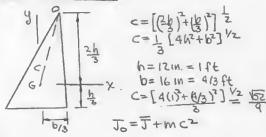


19.74 CONTINUED

Position () $T_{i} = \frac{1}{2} J_{i} \Theta_{m} \quad V_{i} = 0$ $Conservation of energy \quad T_{i} + V_{i} = T_{2} + V_{2}$ $1 J_{0} \Theta_{i}^{2} + 0 = 0 + mgc \Theta_{m}^{2}$ $2 \Theta_{m} = \omega_{n} \Theta_{m}$ $1 J_{0} \omega_{n}^{2} \Theta_{m}^{2} = mgc \Theta_{m}^{2}$

 $\omega_n = \frac{mgc}{J_0}$





$$\bar{J} = St[\bar{I}_x + \bar{I}_y] = St[\frac{1}{36}bh^3 + \frac{1}{36}hb^3]$$
 $m = St[\frac{1}{2}bh]$

$$2f = \frac{\rho N}{2M}$$
 $\underline{I} = \frac{18}{M} [N_3 + \rho_5] = \frac{18}{M} [1 + (6/3)] = \frac{195}{522} M$

$$W_{n}^{2} = \frac{mgc}{J_{0}} = \frac{m(322)(\frac{52}{9})}{m(129/162)} = 32.4 \, s^{2} \quad W_{n} = 5.592 \, \frac{1}{5}$$

TN= 211 = 1.1045

19.75



GIVEN:

85-16 FLYWHEEL PERIOD = 1.265 FOR SHALL OSCILLATIONS

FIND:

CENTROIDAL MOMENT OF



POSITION ()
T= 1 Jo 8 V = 0

POSITION (2)

T2=0 V2=mgh

h=r(1-cosem)=r2sin2em/2

**rem/2

CONSERVATION OF ENERGY

TITYI=T2+Y2 17.6 m+ 0=0+mgram/2

$$W_{n}^{2} = \frac{mgr}{J_{0}}$$
 $T_{n}^{2} = \frac{4\pi^{2}}{\omega_{n}^{2}} = \frac{(4\pi^{2})J_{0}}{mgr}$

$$J_0 = \overline{J} + Mr^2 = \overline{J} + Mr^2 = (C_n^2)(mgr)$$

J=(2)(mgr)-mr=(1.265)2(8516)(72tt)(8216)(45t)(6) FROM (1')

J=1.994-0.8983=1.09616.4652

19.76



GIVEN:

FOR SHALL OSCILLATIONS

PERIOD ABOUT A = T_A =0.8955

PERIOD ABOUT B= T_B =0.8055 T_A+T_b =0.270 M

FIND:

(a) LOCATION OF THE MASS CENTER G (b) CENTROIDAL PADIUS OF GYRATION E.

CONSIDER GENERAL PENDULUM OF CENTROIDAL PADIUS OF GYRATION &.



DATUM AT ()
POSITION

T₁ = 1 J₀ \(\theta_{m}\)

 $T_1+V_1=T_2+V_2$ $\frac{1}{2}J_0\hat{\Theta}_m+O=O+\frac{1}{2}mq^T\hat{\Theta}_m$ $\hat{\Theta}_m=\omega_n\Theta_m$

$$\int_{0}^{\infty} \omega_{n}^{2} = \omega_{n}^{2} = \omega_{n}^{2} = \omega_{n}^{2}$$

$$\omega_{n}^{2} = \frac{\omega_{n}^{2}}{J_{0}} = \sum_{m=1}^{\infty} \sqrt{\frac{J_{0}}{m_{0}^{2}}}$$

$$\omega_{n}^{2} = \frac{\omega_{n}^{2}}{J_{0}} = \sum_{m=1}^{\infty} \sqrt{\frac{J_{0}}{m_{0}^{2}}}$$

 $J_0 = \overline{J} + m\overline{F}^2 = m\overline{k}^2 + m\overline{F}^2$

(a) $C_{N^{2}}2\pi\sqrt{\frac{k^{2}+F^{2}}{gF}}$ FOR THE POO SUSPENDED AT A $C_{N^{2}}$ F= $C_{N^{2}}$ (1)

FOR THE ROOD SUSPENDED AT B $C_N = 0.805 S = \sqrt{\frac{12+12}{415}} F = r_0 \qquad (2)$

But $f_a + r_b = 0.270 \text{ m}$ (3) FROM (1) $f_a^2 + r_b^2 = g r_a (0.895)^2$ (1')

FROM (2) R+10= gro (0.805)2 (2')

SUBTRACING (2') FROM (1')

 $(a^2-r_0^2) = (g/4\pi^2)(0.801r_0 - 0.648r_0)$ (4)

DIVIDING (4) BY (3) HENGER BY HENGER

(3/4112) (0.8016-0.64816)

 $r_a - r_b = \frac{9.81/4\pi^2}{0.270}(0.801 r_b - 0.648 r_b) = 0.7372 r_b .963 r_b = 0.6510 r_a$ (5)

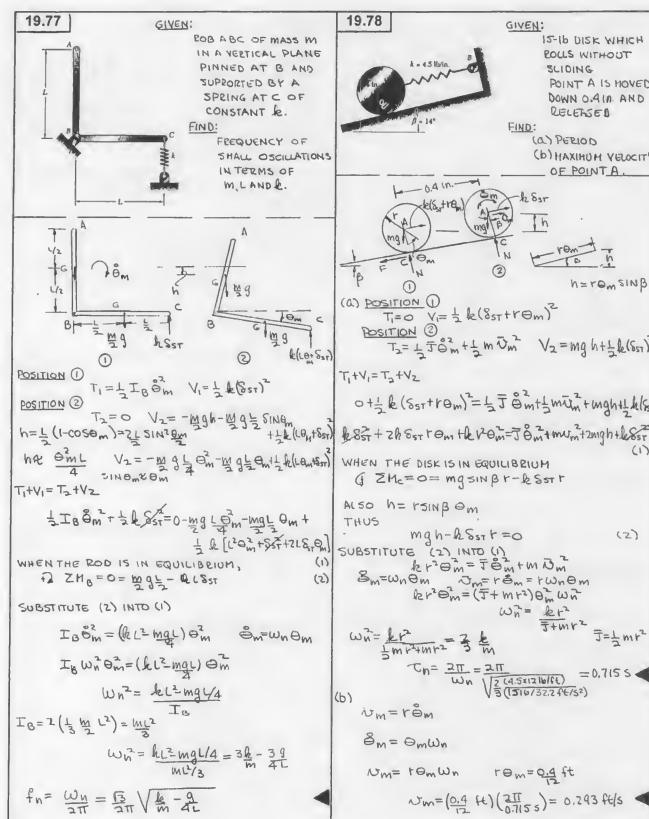
SUBSTITUTE FOR 16 FROM (5) INTO (3)
16+0.6510 16=0.270 16=0.1635 M

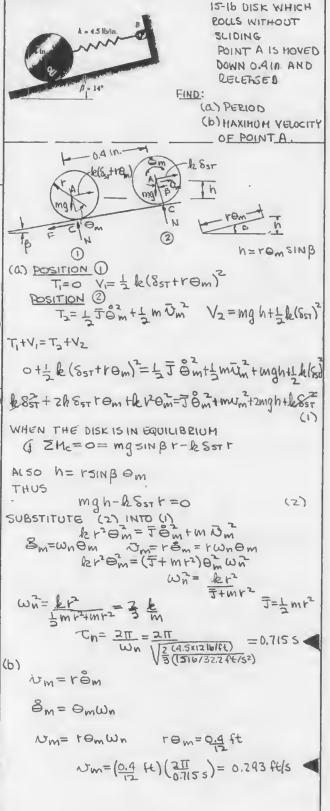
(b) FROM (1')

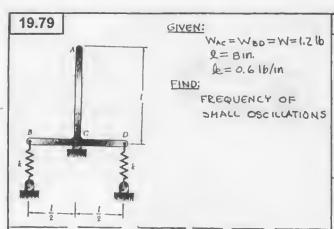
102=(9.81)(0.1635)(0.895/2π)2-(0.1635)2

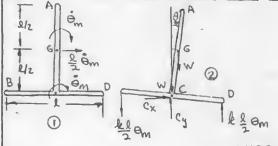
102=0.03254-0.02673=0.05812m2

102=0.03254-0.02673=0.05812m2









Position ()
$$1-\cos\varphi=25iN^2\varphi=\frac{1}{2}m$$

$$T_1=2\left(\frac{1}{2}\overline{J} \stackrel{\circ}{\varphi}_{M}\right)+\frac{1}{2}m\left(\frac{1}{2}\stackrel{\circ}{\varphi}_{M}\right)^2$$

$$V_1=0$$

POSITION 3

$$T_2 = 0$$
 $V_2 = -W_{\frac{1}{2}} (1 - \cos \frac{1}{2}) + \frac{1}{2} k (\frac{1}{2} \frac{1}{2} \frac{1}{2} \cos \frac{1}{2})$

$$V_2 = -\frac{W_{\frac{1}{2}}}{2} + \frac{1}{2} \cos \frac{$$

CONSERVATION OF ENERGY

$$T_{1}+V_{1}=T_{2}+Vz$$

$$\frac{1}{2}(2\overline{J})\overset{\circ}{\theta}_{m}^{2}+\frac{1}{2}M\overset{0}{\overset{\circ}{4}}\overset{\circ}{\theta}_{m}+0=0-\underbrace{\omega_{1}}^{2}\overset{\circ}{\theta}_{m}^{2}+\underbrace{\omega_{1}^{2}\overset{\circ}{\theta}_{m}^{2}}$$

$$\overset{\circ}{\theta}_{m}=\omega_{n}\theta_{m}\overset{\circ}{J}=\frac{1}{12}\underbrace{\omega_{1}}^{2}\overset{\circ}{\theta}_{m}^{2}$$

$$(\underbrace{\frac{\omega_{1}}{6g}+\frac{\omega_{1}}{4g})\overset{\circ}{Q}\overset{\circ}{\omega}_{n}^{2}=(-\frac{\omega_{1}}{2}+\underbrace{\omega_{1}^{2}}^{2})\overset{\circ}{\theta}_{m}^{2}}$$

$$(\underbrace{\frac{\omega_{1}}{6g}+\frac{\omega_{1}}{4g})\overset{\circ}{Q}\overset{\circ}{\omega}_{n}^{2}=(-\frac{\omega_{1}}{2}+\underbrace{\omega_{1}^{2}}^{2})\overset{\circ}{\theta}_{m}^{2}}$$

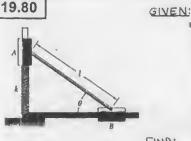
$$(\underbrace{\frac{\omega_{1}}{6g}+\underbrace{\omega_{1}}^{2})\overset{\circ}{\theta}_{m}^{2}=(-\frac{\omega_{1}}{2}+\underbrace{\omega_{1}^{2}}^{2})\overset{\circ}{\theta}_{m}^{2}$$

$$\underbrace{\omega_{1}}_{12}=\underbrace{\frac{\omega_{1}}{3}}^{2}(-\frac{32.2}{12}\underbrace{\text{ft/s}}^{2})+\underbrace{(0.6\times12}\underbrace{\text{1b/ft}}^{2})\overset{\circ}{\theta}_{m}^{2}$$

$$\underbrace{\omega_{1}}_{12}=\underbrace{\frac{\omega_{1}}{3}}^{2}(-48.3+i93.2)=173.9\overset{\circ}{s}^{-2}$$

$$\underbrace{\omega_{n}}_{13}=13.19 \text{ rad/s}$$

fn = \frac{\Omega_n}{2\T} = \frac{13.19}{2\T} = 2.10 Hz.



ELG POD AB

L= 0.6 M

COLLARS A AND B

OF NEGLIGIBLE

HASS

L= 1.7 LN/M

Ø= 40° AT

EQUILIBRIUM

FIND:

PERIOD OF VIBRATION

VERTICAL ROD

Y= $1 \sin \theta$ Sy= $1 \cos \theta \le \theta$ 8G = $1 \cos \theta \le \theta$ 8G = $1 \cos \theta \le \theta$ 8G = $1 \cos \theta \le \theta$ 8 X = $1 \cos \theta \le \theta$ 8 X = $1 \sin \theta \le \theta$ Y= $1 \sin \theta \le \theta$

POSITION (MAXIMUM VECOCITY SEM) $T_1 = \frac{1}{2} \frac{1}{6} \frac{80}{10} + \frac{1}{2} \frac{W(8)}{2} + \frac{1}{6} \frac{W}{10} + \frac{1}{2} \frac{W(8)}{2} + \frac{1}{2} \frac{W}{10} + \frac{1}{2} \frac{W}{10}$

POSITION (ZERO YELDCITY, HAXIMUM SOM)

T= 0

V2= 1 k (Syt6st)2+ mg (y-sqm)

 $V_2 = \frac{1}{3} k (\delta_{ij} + \delta_{ij}) + mq (\bar{y} - \delta_{ij})$ $T_i + V_i = T_2 + V_2$

 $\frac{1}{2}$ ml $\frac{1}{3}$ log $\frac{1}{3}$ + $\frac{1}{2}$ k Ss $\frac{1}{3}$ + mg $\frac{1}{3}$ = 0+ $\frac{1}{2}$ k Sy the string y=k(Sy+2SySst+Ss $\frac{1}{3}$) + mg(y-Sy)

But when the ROD is in Equilibrium,

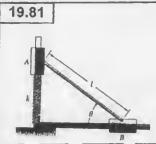
GZMB= mg $\frac{1}{2}$ - kSst x = 0 mg=2kSst (2)

SUBSTITUTE (2) INTO (1)

MEZ {SOM} = RSYM SYM=10050 SOM

MEZ {SOM} RLZOSO (SOM)Z

FOR SIMPLE ITACHONIC MOTION $S\Theta = S\Theta_{m} SIN(w, t+0)$ $S\Theta_{m} = S\Theta_{m} W_{N}$ $\frac{1}{3} M (SO_{m})^{2} W_{n}^{2} = \frac{1}{2} COS^{2}\Theta (SO_{m})^{2}$ $W_{n}^{2} = 3 \frac{1}{2} COS^{2}\Theta = 3 \frac{(1200 N/m)}{8 \log Q} COS^{2}40^{\circ}$ $W_{n}^{2} = 264.07$ $T_{n} = \frac{2\Pi}{W_{n}} = \frac{2\Pi}{\sqrt{264.07}} = 0.387S$

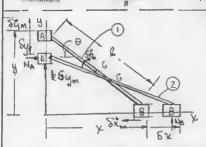


L=0.6m ma=ma=m=8kg k=1.2kn/m 0=40°

ROD AB OF NEGLIGIBLE HASS

EIND:

PERIOD OF VIBRATION



HOPIZONTAL ROD Y= LSIN 0 8y= LCOS 0 80 X= LCOS 0 8x=-LSIN 0 80 8x-LSIN 0 80

POSITION() (MAXIMUM VECOCITY, SOM) $T_{i} = \frac{1}{2} m [Sym)^{2} + \frac{1}{2} m (Sxm)^{2}$ $T_{i} = \frac{1}{2} m [(10050)^{2} + [15100)^{2}] (60m)^{2}$ $T_{i} = \frac{1}{2} m 1^{2} (60m)^{2}$ $V_{i} = 0$

POSITION (SEED VETOCITY, MAXIXUM SO)

V2 = 2 h Sym

 $T_1 + V_1 = T_2 + V_2$

1 ml2 (80m) + 0= 1 2 2 Sym

Sym= 1 cose Som

ml2(88m)2= kl2cos20(80m)2

SIMPLE HAPHONIC HOTION

SO = SOM SIN (WINE+d)

SOM= SOMW,

ml2(80m) Wn2 = Ll2cos20 (50m)2

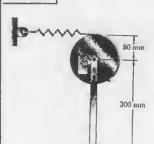
Wn= k cos20

Wn2 = 1200 N/m cos240 = 88.02 5-2

 $C_{N} = \frac{\omega_{N}}{2\pi} = \frac{\sqrt{8802}}{\sqrt{8802}} = 0.66915$

Cn=0.670 S. ◀

19.82



GIVEN;

MAB= 3&9

MOISE = 5&9

SPEING IS

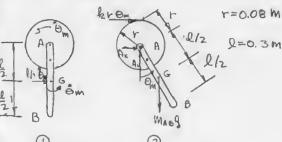
UNSTRETCHED IN

THE POSITION

SHOWN &= 2804

FIND:

PERIOD OF SHALL OSCILLATIONS



1

DOSITION ()

V1=0

Joise = 1 Mor2
(TA)200 - 1 MAG. 2

POSITION (2)

V2= 1 & (rem)2+mag = (1-cos em)

1-COSO m= 2 SIN2 OM & OM

V2= 3 & 120m + male/10m

T1+1=T2+12

= (= mor2+3mapl2) = m+0 = 0+1 & r262m+1 = maggen

= whom

(= mot2+ = mas (2) Wn 0 m= (k+2+ Maeg 2) 0 m

muz= lerz+ MABBELZ

 $W_n^2 = \frac{(280 \text{ N/m})(0.08 \text{ m})^2 + (3 \text{ kg})(0.81 \text{ m/s}^2)(0.3/2 \text{ m})}{1 (5 \text{ kg})(0.08 \text{ m})^2 + 1 (3 \text{ kg})(0.300 \text{ m})^2}$

 $W_{\rm N}^2 = \frac{6.207}{0.106} = 58.55$

 $T_{N} = \frac{2\pi}{w_{N}} = \frac{2\pi}{\sqrt{58.55}} = 0.821 \text{ S}$



WA=1402. Wc=1002. ROD AC WEIGHT = 2002 VERTICAL PLANE

EIND

PERIOD OF SMALL OSCILLATIONS

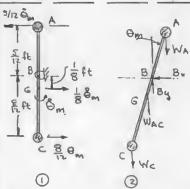
19.84

GIVEN:

SPHERES AND ROD ARE ALL OF HASS M $\beta = 40^{\circ}$ l = 0.5 m

FIND:

FREQUENCY OF SHALL OSCILLATIONS



POSITION ()

 $T_{1} = \frac{1}{2} \frac{W_{A}}{g} \left(\frac{5}{12} \hat{\Theta}_{M} \right)^{2} + \frac{1}{2} \frac{W_{C}}{g} \left(\frac{8}{12} \hat{\Theta}_{m} \right)^{2} + \frac{1}{2} \frac{W_{AC}}{g} \left(\frac{1}{8} \hat{\Theta}_{m} \right)^{2} + \frac{1}{2} \frac{1}{2} \sum_{A \in A} \hat{\Theta}_{M}^{2}$

IAC = 12 WAC (13)2

 $T_{1} = \frac{1}{2g} \left[\frac{14}{16} \left(\frac{5}{12} \right)^{2} + \frac{10}{16} \left(\frac{5}{12} \right)^{2} + \frac{20}{16} \left(\frac{1}{5} \right)^{2} + \frac{1}{12} \left(\frac{10}{16} \right) \left(\frac{13}{12} \right)^{2} \right] \stackrel{0}{\theta}_{m}^{2}$ $T_{1} = \frac{1}{2} \left(\frac{0.5715}{32.2} \frac{16.52}{5615^{2}} \right) \stackrel{0}{\theta}_{m}^{2} = \frac{1}{2} \left(0.01775 \right) \stackrel{0}{\theta}_{m}^{2} \left(16.56 \right) \stackrel{1}{\theta}_{m}^{2} \left(16.56 \right) \stackrel{1}{$

POSITION (2)

T2=0

V2=-WA = (1-COSOM) + Wc = (1-COSOM) +
WAC = (1-COSOM) +
WAC = (1-COSOM)

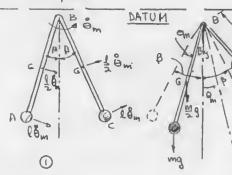
 $V_{2} = \left[-\frac{14}{16} \right) \left(\frac{5}{12} \right) + \left(\frac{16}{16} \right) \left(\frac{8}{12} \right) + \left(\frac{76}{16} \right) \left(\frac{8}{12} \right) \right] \stackrel{2}{\Theta}_{m} \quad \text{(Ib.ft)}$

V2=[-0.3646+0.4167+0.1563] 82m

Vz= 0.2084 8m/2

 $E_{m} = W_{n} \Theta_{m}$ $E_{m} = W_{n} \Theta_{m}$ $E_{m} = W_{n} \Theta_{m}$ $E_{m} = W_{n} \Theta_{m}$

 $W_{N}^{2} = \frac{0.7084}{0.01175} = 11.738$ $T_{N} = \frac{2\Pi}{\sqrt{W_{N}}} = \frac{2\Pi}{\sqrt{11.738}} = 1.8345$



POSITION ()

 $T_{1} = \frac{1}{2} m (V_{A})_{m}^{2} + \frac{1}{2} m (V_{C})_{m}^{2} + \frac{1}{2} (2I_{O}) (\mathring{\theta}_{m}^{2}) + \frac{1}{2} (2M_{O}^{2} \mathring{\theta}_{m}^{2})$ $T_{C} = \frac{1}{12} M_{O}^{2} (V_{A})_{m} = (V_{C})_{m} = \ell \mathring{\theta}_{m}$ $T_{1} = M \ell^{2} \mathring{\theta}_{m}^{2} + (\frac{M \ell^{2}}{24} + \frac{M \ell^{2}}{6}) \mathring{\theta}_{m} = \frac{7}{6} m \ell^{2} \mathring{\theta}_{m}$ $V_{1} = -2 m g (\cos \rho - m g (\cos \rho) = -\frac{\pi}{2} m g (\cos \rho)$

POSITION @

 $V_2 = - mg l cos (\beta - \Theta_m) - mg l cos (\beta - \Theta_m) - mg l cos (\beta + \Theta_m) -$

COSOM & 1-02m/2 (SMALL ANGLES)

V2=- 5 mgl cosp [1-02m/2]

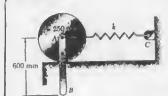
 $T_1 + V_1 = T_2 + Y_2$

7 m l2 om = = mglcoso = 0-5 mgleosp(1-€m)

Om=Wn8m

 $W_{N}^{2} = \frac{15}{14} \left(\frac{9.81 \text{ m/s}^{2}}{0.5 \text{ m}} \right) \cos 40^{\circ} = 16.10 \text{ s}^{-2}$

fn= Wn = V16.10 = 0.639 HZ

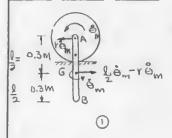


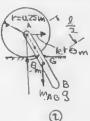
GIVEN:

0.8 kg ROD GOLTED TOA
1.2 kg DISK
L= 12 N/M
DISK ROLLS WITHOUT
SLIDING.

EIND:

PERIOD OF SHALL OSCILLATIONS





POSITION ()

 $T_{i} = \frac{1}{2} \overline{T_{c}} | \theta_{m}^{2} + \frac{1}{2} M_{0}(\frac{1}{2} - r)^{2} \theta_{m} + \frac{1}{2} \overline{T_{c}} | \theta_{0}^{2} + \frac{1}{2} M_{0} r^{2} \theta_{m}^{2}$ $(\overline{T_{c}})_{N,0} = \frac{1}{12} m l^{2} = \frac{1}{12} (0.8) (0.6)^{2} = 0.024 \text{ kg·m}^{2}$ $M_{N,0} (\frac{1}{2} - r)^{2} (0.8) (0.3 - 0.25)^{2} = 0.002 \text{ kg·m}^{2}$ $(\overline{T_{c}})_{0.5 k} = \frac{1}{2} M_{0.19 k} r^{2} = \frac{1}{2} (1.2) (0.25)^{2} = 0.0375 \text{ kg·m}^{2}$ $M_{0.05 k} = \frac{1}{2} M_{0.19 k} r^{2} = \frac{1}{2} (1.2) (0.25)^{2} = 0.0375 \text{ kg·m}^{2}$ $T_{i} = \frac{1}{2} [0.024 + 0.002 + 0.0375 + 0.0750] \theta_{M}^{2}$ $T_{i} = \frac{1}{2} [0.1385] \theta_{M}^{2}$

POSITION 2

V1=0

T1= 0

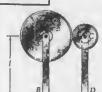
 $V_{2} = \frac{1}{2} k (r \Theta_{m})^{2} + M_{AB} g \frac{1}{2} (1 - \cos \Theta_{m})$ $1 - \cos \Theta_{m} = 2 \sin^{2} \Theta_{m} \approx \underline{\Theta}_{m}^{2} (shall Augusts)$ $V_{2} = \frac{1}{2} (12 \text{ N/m}) (0.25 \text{ m})^{2} \Theta_{m}^{2} + (8 \text{ kg}) (1.8 \text{ kg}) (0.6 \text{ m}) \frac{0.6 \text{ m}}{2}$ $V_{2} = \frac{1}{2} [0.750 + 2.354] \Theta_{m}^{2} = \frac{1}{2} (3.104) \theta_{m}^{2} \text{ U.m}$

 $T_1 + U_1 = T_2 + V_2$ $\mathring{S}_m^2 = \omega_n^2 \Theta_m^2$ $\frac{1}{2} (0.1385) \Theta_m^2 \omega_n^2 + 0 = 0 + \frac{1}{2} (3.104) \Theta_m^2$

 $W_N^2 = \frac{(3.104 \text{ N/m})}{(0.1385 \text{ kg·m}^2)} = 22.415^2$

$$T_N = \frac{2\pi}{W_N} = \frac{2\pi}{\sqrt{22.41}} = 1.3275$$

19.86

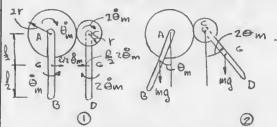


GIVEN:

PODS AB AND CD EACH OF MASS M AND LENGTH & ATTACHED TO GEARS A ANDOMASS OF GEAR A = 4 M MASS OF GEAR C= M

FIND

PERIOD OF SHALL OSCILLATIONS



CINEHATICS 2 POA= POC ZOA= OC 20A= OC LET OA= OM 20M= OC) M 20M= OC) M

TI = 1 IA OM + 1 Ic (20m) + 1 I IAB OM + 1 IC (20m) + 1 I IAB OM + 1 IC (20m) + 1 IMAB (10m) + 1 Mall (10m)

 $\begin{array}{l}
\bar{I}_{A} = \frac{1}{3} (4m) (2r)^{2} = 8mr^{2} \\
\bar{I}_{C} = \frac{1}{3} (m) (r)^{2} = \frac{1}{3} mr^{2} \\
\bar{I}_{AB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{12} m L^{2} \\
\bar{I}_{AB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{12} m L^{2} \\
\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{12} m L^{2} \\
\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{12} m L^{2} \\
\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{12} m \Omega^{2} \\
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\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \\
\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \\
\bar{I}_{CB} = \frac{1}{3} m \Omega^{2} \quad \bar{I}_{CB}$

POSITION 2

T=0

V1= mgg (1-cosom)+mge (1-cosom)

SHALL ANGLES 1-COSOM= ZSIN' OM = OM

1-cos40=251200 = 20m

Tit VI=T2+12 62m=W202m

= m [10+2+ \$ 02] W2 02 +0=0+ = mgl 502m

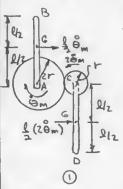
 $\omega_{N^{2}} = \frac{5 g \ell}{10 r^{2} + 5 \ell^{2}} = \frac{3g \ell}{12 r^{2} + 2\ell^{2}}$

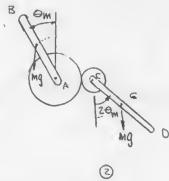
 $\mathcal{L}^{N} = \frac{\sqrt{m^{N}}}{3 \, \mu} = 3 \, \mu \sqrt{\frac{3 \, \delta \sigma}{15 \, k_{3} + 5 \, \delta_{3}}}$

GIVEN:

RODS AB AND BC EACH OF MASS M GEAR A OF HASS 4M GEAR C OF MASS M

FIND: PERIOD OF SHALL OSCILLATIONS





200=0c KINEHATICS 2rea=rec 70,=0c LET BA=Om 20 m=(02)m 28 m= (0.)m

POSITION ()

T= 1 In 8 m+ 1 I (10) 1 1 In 8 0 m + 1 Ico (20 m)2 + = mag(20m)+ = mo(20m)

IA= 12 (4m)(2t)= 8m+2 Ic= 7 (W)(ks) = 7 W Ls

IAB= 12 ml2 Ico= 12 ml2

T== 1 m[8 r2+ (12/2)4+ 22/12+ 22/3+ 22/4+ 22] = m

T_= 1 m[lor2+ 5 2] 0m

POSITION @ T2= 0 Y2= -mg & (1-cosem) +mg (1-cos 20m)

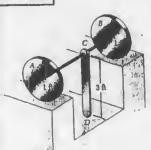
SMALL ANGLES 1-LOS OM= ZSINTOM & Om

1-005 20m=251 n2 0m2 20m2 N= - mgg =2m + mgg 20m= 1 mg/3 0m

TitVI=Tz+Vz Om=WNOM

1 m[1017+ 312] 02 W1+0=0+1 mg = 02

Wn = 392 = 992 107+502 6072+1012 Th= 211 /60 +2+10 12 19.88

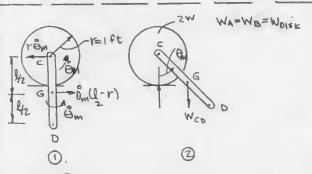


GIVEN:

10-16 ROD CD DISKS A AND B EACH WEIGH ZO 16 AC OF NEGLIGIBLE WEIGHT NO SLIDING

FIND:

PERIOD OF SMALL OSCILLATIONS



T,= = 2 2 1 10 m + 1 (2 mb, s=) (+ 6 m) + 1 Izo 8 m (IV) = 7 Moise 13 7 (50) (1)= 10 Ico = 12 Meo (2 = 1 (10) (3) = 15 T== 29 [20+40+15+5] 0m T= 1/29 (70) 8m

V1=0

POSITION 2

Vz= Wco & (1-cosom)

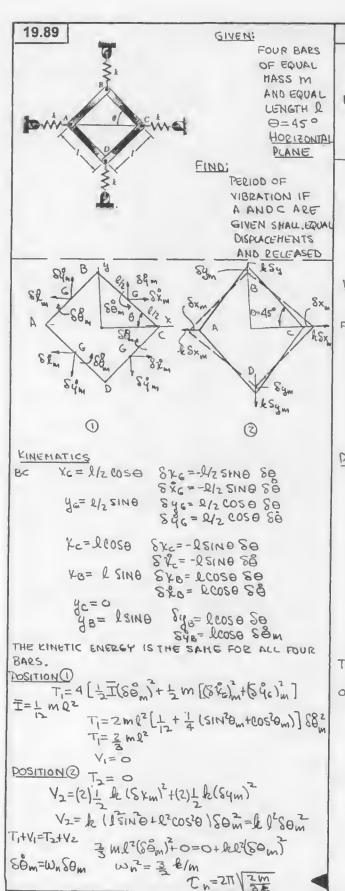
SHALL ANGLES 1-COSOM=25IN29 2 0m N= 7 Ncol Qm = 7(10)(1.2) gm = 7 12gm

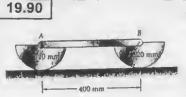
TI+VI=TZ+VZ Om=WnOm

79401m282+0=0+11802m

Wn= 159

 $T_N = \frac{2\pi}{\sqrt{\omega_N}} = 2\pi \sqrt{\frac{70}{(15)(32.2)}} = 2.395.$



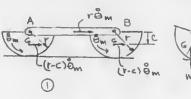


GIVEN:
DISKS OF HASS 3 lg
EACH
MASS OF ROD AB.
= 2 lg

FIND:

PERIOD FOR

NO SLIDING



Mog mog

POSITION (1)

 $T_{1} = 2\left(\frac{1}{2}\right) \overline{L}_{0} \stackrel{\circ}{\Theta}_{M}^{2} + 2\left(\frac{1}{2}\right) m_{0} \left(r - c\right)^{2} \stackrel{\circ}{\Theta}_{M}^{2} + \frac{1}{2} m_{r} \dot{r}^{2} \stackrel{\circ}{\Theta}_{M}^{2}$ FOR ONE DISK $\overline{L}_{0} = \left(\overline{L}_{0}\right)_{A} - m_{0} c^{2} = \frac{1}{2} m_{0} r^{2} - m_{0} \frac{1}{4} r^{2} = m_{0} \left[\frac{r^{2}}{2} - \frac{16r^{2}}{9\pi^{2}}\right]$ $\overline{L}_{0} = 0.3199 m_{0} r^{2}$ $M_{0}(r - c)^{2} = m_{0} r^{2} \left(1 - \frac{4}{3\pi}\right)^{2} = 0.3315 m_{0} r^{2}$ $T_{1} = \left[\left(0.3199 + 0.3313\right) m_{0} r^{2} + 0.5 m_{r} r^{2}\right] \stackrel{\circ}{\Theta}_{M}^{2}$ $T_{1} = \left[\left(0.6512 m_{0} + 0.5 m_{r}\right) r^{2}\right]$

 $\frac{Dosition}{T_2=0}$ $V_2=2m_0g\ C(1-cose_m)$ $C=\frac{4r}{3\pi}$ $1-cose_m=2sin^2e_m \approx \frac{e^2m}{2}$ shall shall

 $V_2 = 2 mog \frac{4r}{3\pi} \frac{G^2m}{2}$ $V_2 = mor \frac{9(4)}{3\pi} = mor (9.81)(4) = 4.164 mor$

 $T_1 + V_1 = T_2 + V_2$ $O + (0.6512 m_0 + 0.5 m_r) r^2 \theta_m^2 = 0 + 4.164 m_0 r$ $\theta_m = \omega_n \theta_m$

 $\omega_{N}^{2} = \frac{4.164 \, \text{M}_{D} + (4.164)(3)}{(0.6512 \, \text{M}_{D} + .5 \, \text{M}_{F}) + 7 \, \text{M}_{D}(6.6512)(3) + 0.5(2)]_{0.120}}{(0.3544)} = 35.24$

Wn = 5.936

 $T_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{5.936} = 1.0585$

GIVEN:

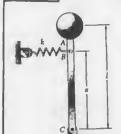
SPHERE OF WEIGHT W BAR ABC OF NEGUGIBLE WEIGHT



(a) FREQUENCY OF SHALL OSCILLATIONS

(b) SHALLEST VALUE OF a FOR WHICH OSCILLATIONS WILL OCCUR.

19.92

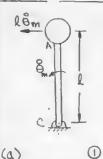


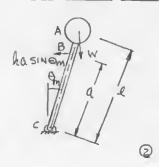
GIVEN:

SPHERE OF MEIGHT M E= 1.2 HF MHEN M=SIP f=0.8HZ WHEN W= 416

FIND:

FOR GIVEN &, a AND LITHE LARGEST VALUE OF W FOR WHICH OSCILLATIONS WILL OCCUP





POSITION O

$$T_{i} = \frac{1}{2} m (l \mathring{\Theta}_{m})^{2} = \frac{1}{2} m l^{2} \mathring{\Theta}_{m}^{m}$$

$$V_{i} = 0$$

DOSITION @

SHALL ANGLES SINOM & OM 1-COS OM = ZSINZOM & OM

V2= 1 6 62 02m - Wl 02m = 1 [kd2-W] 02m

T1+V1=T2+V2

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

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$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

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$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

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$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} + 0 = 0 + \frac{1}{2} [ka^{2} - Wl] \Theta_{m}^{2}$$

$$\frac{1}{2} m l^{2} \Theta_{m}^{2} +$$

SEE SOLUTION TO PROBIGGI FOR THE FRQUENCY IN TERMS OF W. L. Q AND I

fu=1.5 Hz W=216 1.5= \frac{1}{2\pi}\g/2\left(\frac{\alpha^2}{20}-1)

fn=0.8Hz W=41b 0.8== 1 \ g(1 (ka2-1) (2) DIVIDE (1) BY (2)

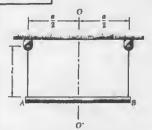
$$\binom{(.5)}{0.8}^2 = \left(\frac{k a^2}{2 a}\right) / \left(\frac{k a^2}{4 a}\right)$$

3.516 ka2-3.516 = ha2-1

$$\frac{ka^{2}\left(\frac{3.516}{4} - \frac{1}{2}\right) = 2.516}{ka^{2} - 6640}$$

fn= = 1 / 9/1 (6.640-1), fn=0, 6.640-1=0

19.93

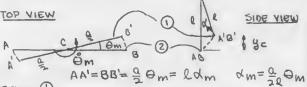


GIVEN:

PIPE SUSPENDED FROM TWO CABLES AT A AND B

FIND:

FREQ VENCY YIBRATION FOR A SHALL ROTATION ABOUT OO'



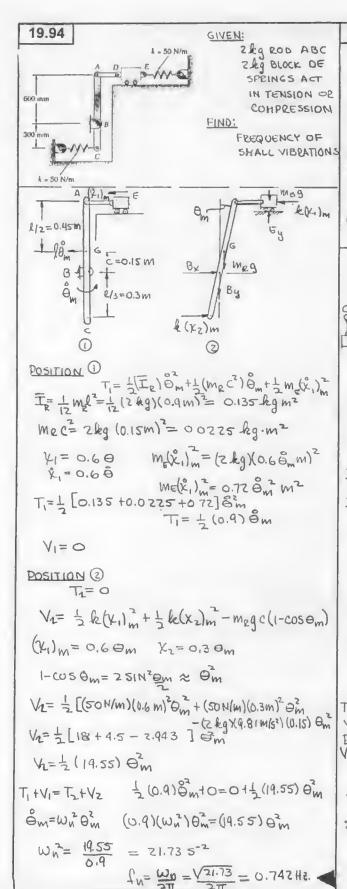
DADITION (1) T=0 V= mg y=mg l(1-cosx)

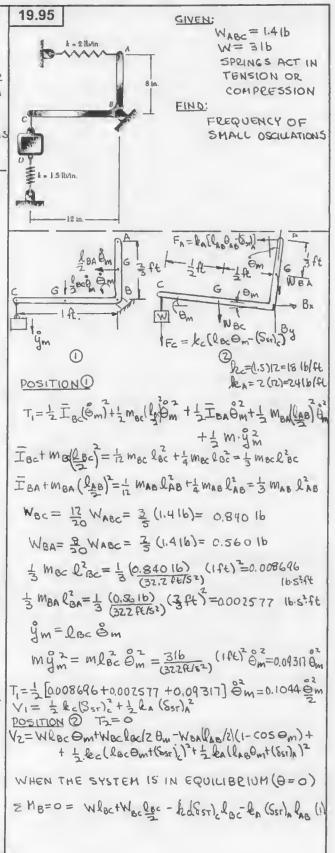
SHALL ANGLES 1-cosx = 2 sin x x x = 02 0 m

V= mg l (02/802) 02 m 2 802

POSITION (2) T2= 1 I & m = 1 (12 m a2) & m V2=0 T,+ V,= T2+ V2 mg l(a2/822)+ 0+ 1/24 ma2 Wn2 Om

 $w_{n}^{2} = 39/2$ $f_{n} = \frac{1}{2n} \sqrt{39/2}$





19.95 CONTINUED 1-COS @m=2 SIN2 @m = @2m V2=[Wloc+Weckec(2)] @m-[Weaklas/2/(@m/2)] + + ½ kcloc @m- kcloteleom + ½ kc(@sr)2 + + ½ ka 120 @m- kallteom + ½ ka(@sr)2 + 12 ka 120 @m- kallteom + ½ ka(@sr)2 + Taxing € EQUATION (1) INTO ACCOUNT

TAKING EQUATION (1) INTO ACCOUNT

\(\lambda_2 - \left[W_{\text{Ba}} \left[A_{\text{B}} \right] \right] \frac{2}{2} + \frac{1}{2} \left[A_{\text{B}} \right] \frac{2}{2} \right] \frac{1}{2} \left[A_{\text{B}} \right] \frac{2}{2} \left[A_{\text{B}} \right] \frac{2}{2} \left[A_{\text{B}} \right] \frac{2}{2} \right] \frac{2}{2} \left[A_{\text{B}} \right] \frac{2}{2} \right] \frac{

V2= = = [0.1867+18+10.67] 0m+ = lec (Sst) 2+ = la Sst) 2

V2= = [28.48] 0 + 1 2 ((Sst) = + 1 2 & (Sst))

 $T_1 + V_1 = T_2 + V_2$

1 (0.1044) 0 m + 1 holosi) = + 1 holosi) = = 0 11 (28.48) 0 m + 1 holosi) = 12 holo

=w, Om

 $0.1044 \text{ W}_{n}^{2} \Theta_{m}^{2} = 24.92 \text{ } \Theta_{m}^{2}$ $W_{n}^{2} = \frac{28.48}{0.1044} = 272.8 \text{ } S^{-2} \text{ } f_{n} = \frac{\sqrt{272.8}}{2.11} = 7.63 \text{ Hz.}$

·19.96

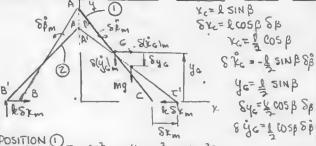


GIVEN:

RODS AB AND AC EACH OF MASS M AND LENGTH &

FIND:

PERIOD WHEN A IS GIVEN A SHALL DOWN DEFLECTION AND RELEASED



POSITION () T = 2 [] I (Sp) + 1 ((Sx) + (Sg)) = 1 M1 Sp., = (12 m 2 + m 2 5 1 m 2 + cos p) Epin = 1 M1 Sp., V = 0

POSITION (2) $T_1 = 0$ $Y_2 = \frac{1}{2} k (28 \text{ km})^2 = \frac{1}{2} (4 1 \cos \beta) = 2 1 \cos^2 \beta$ $8 \beta_m = \omega_m \delta \beta_m$

T1+V1=T2+V2 3 m 22ω, δ ε, +0=0+2 k 2 cos 2 β β, ων 3 = 4 cos 2 β

Ch Valendos's COSSI GR

19.97



GIVEN:

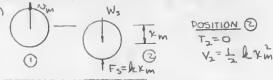
V= YOLUNG OF THE STYLE EINCTIC ENERGY = \$ 6 VU?
WHERE P= MASS DENSITY
AND N = URLOCITY OF
THE SPHERE
SPHERE HASS = 500-3
& = 500 N/M, HOLLOW OF
SPHERE RADIUS = 90MM

FIND:

(A) DERIOD WHEN DISPLACED YERTICALLY AND RECEASED

(b) PEZIOD WHEN THE TANK IS ACCELERATED UPWARD AT BM/52

THIS IS NOT A DAMPED YIBRATION. HOWEVER THE KINETIC ENERGY OF THE FWID MUST BE INCULED



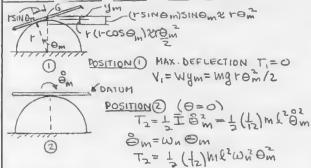
POSITION () $T_1 = T_{SPQSE} + T_{PUNO} = \frac{1}{2} M_S U_M^2 + \frac{1}{4} (V U_M^2)$ $V_1 = 0$ $V_2 = 0$ $V_3 = 0$ $V_4 = 0$ V_4

19.98

GIVEN:
PLATE ON A SEMI-CIRCULA
CYUNDER AS SHOWN

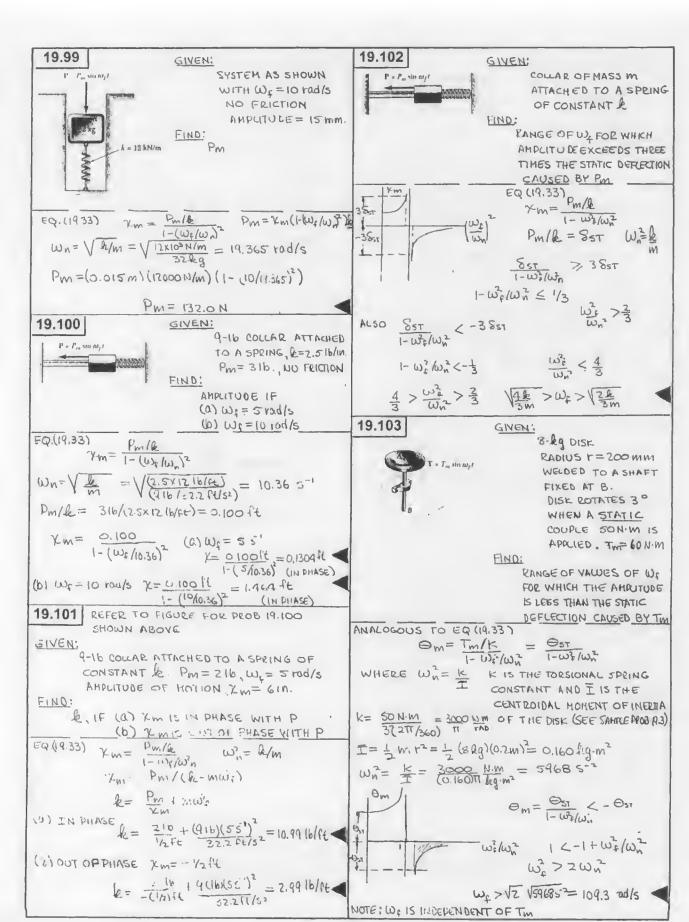


EPERIOD FOR SMALL OSCILLATIONS



T,+V,=T2+V2 0+1mg+0m=12(12)ml2w20m

 $W_n^2 = \frac{12gr}{\Omega^2}$ $T_n = \frac{2\pi}{W_n} = \frac{2\pi}{\sqrt{2}} \sqrt{\frac{12gr}{\sqrt{2}}}$ $T_n = \frac{\pi \Omega}{\sqrt{2}}$



19.104 T = Tm sin tort ANALOGOUS TO EG (19.33) (30) (11/360) TT YOU Dm= 2.5, = 0.06109 1nd Opr 3.50 19.105

GIVEN:

8kg DISK, r= 200 mm. WELDED TO A SHAFT FIXED AT B. DISK POTATES 3° WHEN A STATIC COUPLE OF SOUN IS APPLIED, TW= 60 N.M.

FIND:

PANGE OF VALUES FOR WHICH THE AMPLITUDE OF YIBRATION

13 LESS THAN 3.50 Om = Tim/k - OLT - 1-10); 1-10: 100%

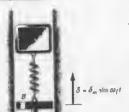
WI = K/T WHERE K IS THE TORSIONAL SPRING CONSTANT AND I IS THE CENTROIDAL HOHENTO: " ZITA OF THE LISK, (SEE SAMOCE LEDS 11.3)

I= 1 mr2= 1 (829)(0.2m)=0.160kg·m2 Wil= K/I = (3000/11)/(0.160) = 596835-2 Oct = 1 m/k = 60 N m/ (3000A) = 0.06263 Has



1-W=1w: 13 06283 >-0.06107 1-W1/5968)

.062834[3/(5968)-1] (0:0614) 1,028574(2)62,7681-1 WE >121075-2 WE> 110.0 PAD/S



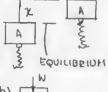
GIVEN:

4-16 BLOCK A SPRING &= 816/ft Sm= 1 in. : 1)5=5 rod/5

EIND:

(MINHPUTUDE "! I'OTION OF THE BLOCK (b) AMPLITUDE OF FLUCTUATING FORCE OF THE SPRING ON BLOCK

FROM EQ. (14.33') 7-m= 8m/(1-(W/W)))



(a) $\omega_n^2 = \frac{k}{m} = \frac{(7.16/12.2)(11/52)}{(4.16/32.2)(11/52)}$

Win= 64.4 5-2 $\gamma_{m} = \frac{(4/12 \text{ ft})}{1 - (25/64.4)} = 0.5448 \text{ ft}$

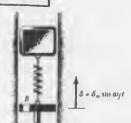
(b)

7-m=0.545ft SINCE WE ZWA, 1 AND S ARE IN PHASE AND NET SPRING DEFLECTION 15 X-8 AND F- 6(X-8)

Fm=(816/16) (05448 ft -0.333 ft)

Fm= 1.6921b

19.106



GIVEN:

8 leg BLOCK A SPRING R= 1.6 RN/m 8m= 150mm

FIND:

VALUES OF WE FOR WHEN THE FLUCTUATING FORCE OF THE SPRING ON THE BLOCK IS LESS THAN

FROIT EQ. (19.33') Km= Sm/(1- W/(W)) Win - E/M = 1.6 x 10 11/1 = 200 52

IN PHASE Fm=k(4m-8m)= &8m/(1-W2113m)-k8m < 120 N 1Fm 1/(1-43,1602)-1 < 120/(1600)(150)= 1/2 LI HOU 2/3 < 1- W2/W2 m² ~ 5 < 500/3 mt < 8.16 mg/2 GUT OF PHASE Fm = & (4m+Sm) = 1600 (4m+0.150)

= 1600 × m + Z40 N7120 N THERE IS NO YALVE FOR YM WHICH WILL HAKE

FM < 120 WHEN X AND & ARE OUT OF PHASE 19.107



(Sor) = ZIn. Sm = 0.5 in

PANGE OF WE FOR WHICH I'M B < 1 in.



MX=- k(x68) Migthe xo= Re Sm SINWit

THUS, FRON EQ (19.31) AND (19.33')

$$\omega_{n}^{2} = \frac{4}{5} = \frac{32.7 \text{ ft/s}^{2}}{2/12 \text{ ft}} = 193.2 \text{ s}^{-2}$$

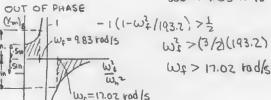
$$(x_m)_8 = \frac{0.5}{1-w_1^2/1027}$$

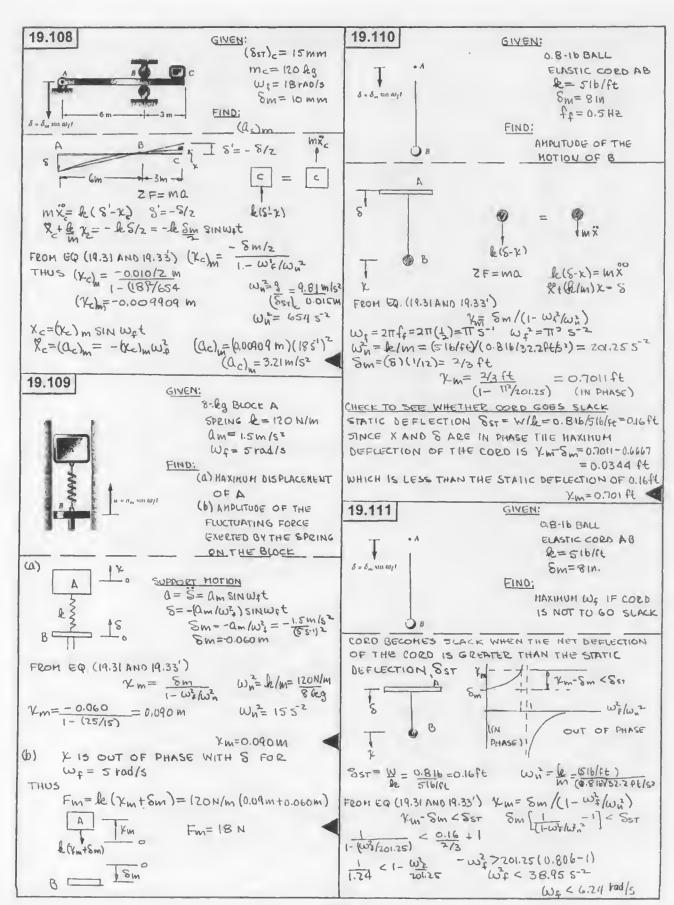
$$|1| < \frac{1/2}{1 - \omega^2 \epsilon / 193.7}$$

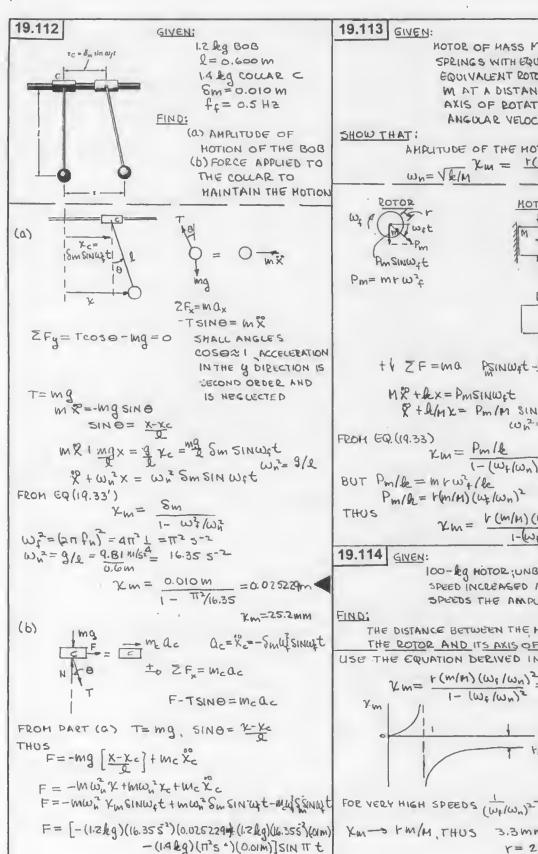
IN PHASE +1 (1-W3/1937) < 3

W? < 193.2

Ws < 9.83 tod/s







F= -0.437 SINTT(N)

19.113 GIVEN: HOTOR OF HASS M SUPPORTED BY SPRINGS WITH EQUIVALENT CONSTANT & EQUIVALENT POTOR HASS UNBALANCE M AT A DISTANCE I FROM THE AXIS OF POTATION. ANGULAR VELOCITY OF HOTOR, W. SHOW THAT: AMPUTUDE OF THE MOTION OF THE KODE Xm = r(m/m)(we/wis 1-(ws/wn) ROTOR MOTOR 1 Pm SINWIT y sinost Pm= mrw2 th ZF=ma Psinwit-lex=nx MX+RX= PMSINWEE & + SIMK = Pm/M SIN WET W2= le/n FROM EQ. (19.33) Km= Pm/le (- (W+(Wn)2 BUT Pm/le=mrwifle Pm/h= +(m/M) (wflwn)2 2m = 1-(m/M) (W+/Wn) 2 QED 19.114 GIVEN: 100-kg HOTOR; UNBNANCED 15-kg ROTOR. SPEED INCREASED AND AT VERY HIGH SPECOS THE AMPLITUDE NEADS 3.3 MM FIND: THE DISTANCE BETWEEN THE HAS S CENTER OF THE ROTOR AND ITS AXIS OF ROTATION USE THE EQUATION DERIVED IN PROB P.113 (ABOVE) $\frac{1 - (m^{2} (m^{2})^{2})}{1 - (m^{2} (m^{2})^{2})^{2}} = \frac{1}{(m^{2} (m^{2})^{2} - 1)}$ Y.m Wellen +(m/n)

Xm-s + m/M, THUS 3.3 mm= + (15/100) r = 22 mm

19.115 GIVEN:

SPRING SUPPORTED HOTOR WHOSE SPEED IS INCREASED FROM 200 TO 300 EPM AMPLITUDE DUE TO UNBALANCE INCREASES CONTINUOUSLY FROM 2.5 TO 8 MM

FIND:

SPEED AT RESONANCE

FROH PROB 19.113 2m= +(M/H)(W=/W)2 25= r(m/m)(200/wn)2 1-(200/Wn)2 8 = Km/W) (300/m")3 $\frac{2.5}{80} = \frac{1 - (300/\omega_n)^2}{1 - (200/\omega_n)^2} \left(\frac{200}{300}\right)^2$ 0.703-0.703(200/w,)= 1- (300/w,)2

1 [90x103 28.125 x103] = 0.2969

Wn= 208.4 Wn=457 rpm

RESONANCE WHEN WE=WW

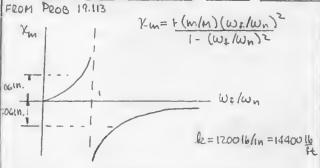
Wc= 457tpm ◀

19.116 GIVEN:

400-16 HOTOR SUPPORTED BY SPRINGS WITH TOTAL R=120016/in ROTOR UNBALANCE IS 102, 8 IN FROM THE AXIS OF ROTATION

FIND:

RANGE OF ALLOWABLE VALUES OF HOTOR SPEED IF THE AMPUTUDE OF VIBRATION IS NOT TO EXCEED 0.0611



Wn= k/M= (19400 16/ft)
(40016/32.2 ft/s2) = 1159.25-2

t m/n = (8/12 ft) (1/16 16) = 104.17 × 10-6 ft

X=0.06 in (0.06/12 ft) < 104.17 x10 ft (W=/41) 1-(W+/W)2 47.998-47.998 (W/Wn)2 (Wf/Wn)

 $(\omega_{\rm f}/\omega_{\rm h})^2 \leq \frac{47998}{98.988}$

W € 0.9897 (1159.2) 12 War < 0.9897

wf € 33.69 rad/s WG = (33.69 rad 160 \$)(211 RAD/REV) = 322 RPM

XM=-0001N (-000/12ff)> 104.17x100ff (m3/(m2) 1-w12/wn2 (W+/Wn)2 > 47.998=

Wf 7(1.0106)(1159.2)1/2 Wf 734.40 rod (5=329 rpm

19,117



GIVEN:

220-16 HOTOR UNBALANCE OF THE ROTOR= 202,411 FROM THE AXIS OF ROTATION RESONANCE AT 400 PM

FIND:

AMPUTUOE AT (a) 800 rpm (b) 200 rpm (c) 425 rpm

FROM PROB 19.113

Km= r(m/H) (W f/wn)2

RESONANCE AT 400 PM HEARS THAT Wn= 400 pm $V(M/M) = (4 \text{ in.})(2/16)/(220) = 2.2727 \times 10^{-3} \text{ in.}$

(a) $(\omega_f/\omega_n)^2 = (800/400)^2 = 4$

Km=2.27.7x1030.(4) =0.003031m

(b) (w, 16,)= (200/400)= 1/4

7-m= 2.2721(103(4)-0.00075811.

(c)(wf/wn)=(925/400)=1.1289

Km=2272x103(1.1289) _-0.0199011. 1-1.1289

19,118



GIVEN:

180-ly MOTOR UNBALANCE OF THE ROTOR = 289 ISO MM FROM AYIS OF POTATION STATIC DEPERTION 551=12mm

FIND:

HASS OF A PLATE ADDED TO THE BASE OF THE HOTOR SO THAT AHPUTUDG OF VIBRATION IS LESS THAN GOXIOGM FOR HOTOR SPEEDS ABOVE 300 Ppm.

FROM PROB 19.113

1- W= /Wn2

SINCE HWIZER Xm= (m+/b) W} /(1-W3/W2)

BEFORE THE PLATE IS ADDED, WAZ = 9 = 9.81 m/s2

k= HW= (180 kg) (817.553) W= 817.5 5-2 k= 147.15 ×103 N/m

mr/l= (28x10-3kg)(0.150m)/147.15x103N/m) = 28.542×10-9 m.52

AFTER THE PLATE IS ADDED THE NATURAL FREDUENCY OF THE SYSTEM CHANGES SINCE THE HASS CHANGES W" = JR/M'

SINCE THE VIBRATION IS TO BE LESS THAN GONO M FOR HUTOR SPEEDS ABOVE 300 FPM, WE HAVE Xm= -60x106 m= (28.542x109 m.52)(300.2115)

1- (300.211/60)

-2.1299+21299 (98696)=1

 $w_n^2 = \frac{2.1299 (986.96)}{1299} = 671.65^{-2} = \frac{1}{1299}$ M'=(147.15×103N/m) /(671.652)= 219.1 kg AM=M'-H=219-180= 39.1 &q

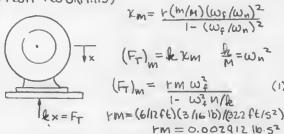


400-16 HOTOR UNBALANCE OF 302. 6 IN FROH AXIS OF ROTATION FORCE TRANSHITTED TO FOUNDATION LIMITED TO 0,210 WHEN MOTOR IS BUN AT 100 rpm AND ABOVE

FIND:

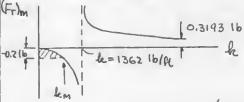
(a) HAXIMUH ALLOWABLE SPRING CONSTANT & OF A PAD PLACED BETWEEN THE HOTOR AND THE FOUNDATION (b) CORRESPONDING AMPLITUDE OF THE FLUCTUATING FORCE WHEN THE HOTOR IS RUN AT 200 rpm

(a) From PROB. (19.113)



AT We = 100 rpm = 100 (21/60) = 10.472 rad/s

(FT)m= 0.31928 1-1362/k



-. Z = 0.319 /(1-1362/fe) -0.2+0.2 (1362)/2=0.31928

$$k_{\rm m} = \frac{(0.2)(1362)}{0.51928} = 525 \, \text{lb/ft}$$

(b) AT 200 rpm , W, = (200)(211)/60=20.94 rad/s FROM (1), AND USING & FOUND IN PART (0)

$$(F_T)_{M} = \frac{(0.002912 \text{ lb.s}^2)(20.94 \text{ s}^{-1})^2}{(-(20.94 \text{ s}^{-1})^2(400 \text{ lb}/32.2 \text{ ft}/6^2)}$$
 $(F_T)_{M} = \frac{(525 \text{ lb/ft})}{(525 \text{ lb/ft})}$
 $(F_T)_{M} = -0.1361 \text{ lb}$
 $(F_T)_{M} = -0.1361 \text{ lb}$
 $(F_T)_{M} = -0.1361 \text{ ft}$

19.120 GIVEN:

180-LA HOTOR, SUPPORTED BY SPRINGS OF TOTAL CONSTANT &= 150 kN/m UNBALANCE OF THE ROTOR 15 28-9 AT 150 MM

FIND:

PANGE OF SPEEDS FOR WHICH THE FLUCTUATING FORCE (FT) IS LESS THAN 20 N

FROM PROB (19.113) E/M=Wn (Wx/Wn)2 xm= +(m/m) (w))

(Fr) m= lexm FT = +m w} /(1-(4/wn)2)

rm=(0.150 m)(0.028 lg)=0.0042 m·lg Wn= l/n=(150x103 N/m)/(180 lg)=833.352 (Ft= ((0.0042)(W3))/(1-W2/838.3)



 $(F_{\tau})(1-\omega_{\tau}^{2}/833.3) = 0.004 \text{ 2 m}_{\tau}^{2}$

W2 = (Fr) [(Fr) [(853.3)+0.0042] FOR (FT) m = 20 H Wif < 0.024 +0.00 42 -= 709.2 52

Wf 5 26.63 tod/s

FOR (Fr) = -20 N $W_{\rm f}^2 > \frac{-20}{0.004 + .0042} = 10105^{-1}$ Wg 731.78 rad/s

W>31.78 (60) = 303 rpm (JII)

19,121



GIVEN: fu= 120 Hz

ZIM = AMPLITUDE RELATIVE TO THE BOX IS USED AS A HEASURE OF SIM

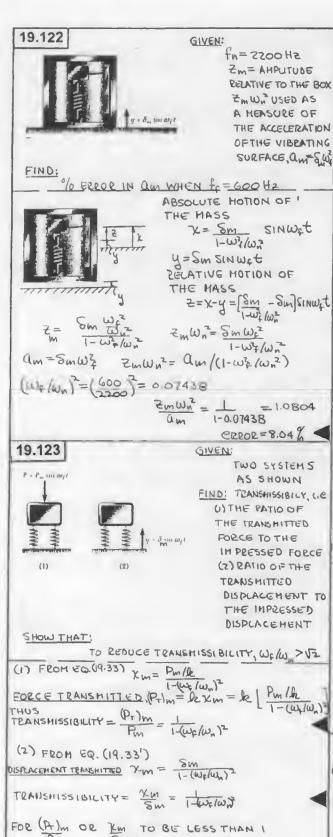
(a) % EREDE FOR fr= cook (b) for ZERO ERROR

7 = (3m 2) SINWet

y= Sm SINWet 2 - RELATINE HOTION

2 m = Sm[1-w/10,2] = Sm w//11 $\frac{\omega_{\rm f}^2/\omega_{\rm h}^2}{1-\omega_{\rm f}^2/\omega_{\rm h}^2} = \frac{(600/120)}{1-(600/120)^2} = \frac{25}{24} = 1.0417$ ERROR=4.17%

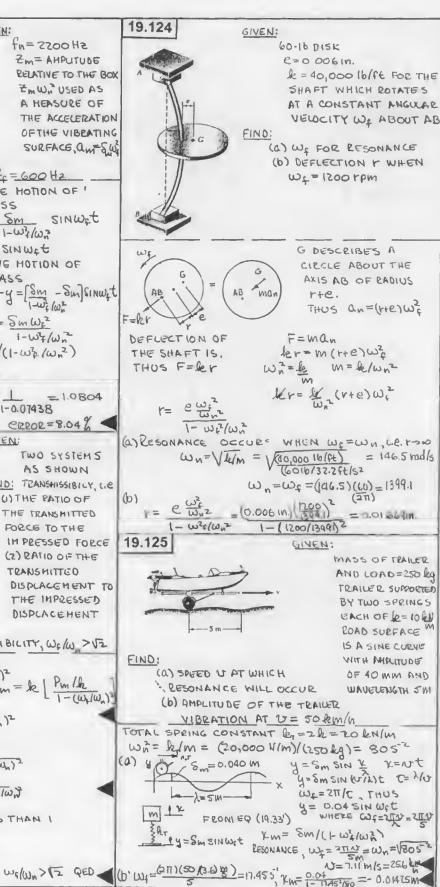
(b) == = = W= 1 - W= 1 1-62/6/2 1=2 m3/m3 ft= = 12 (150)= 84.8H5

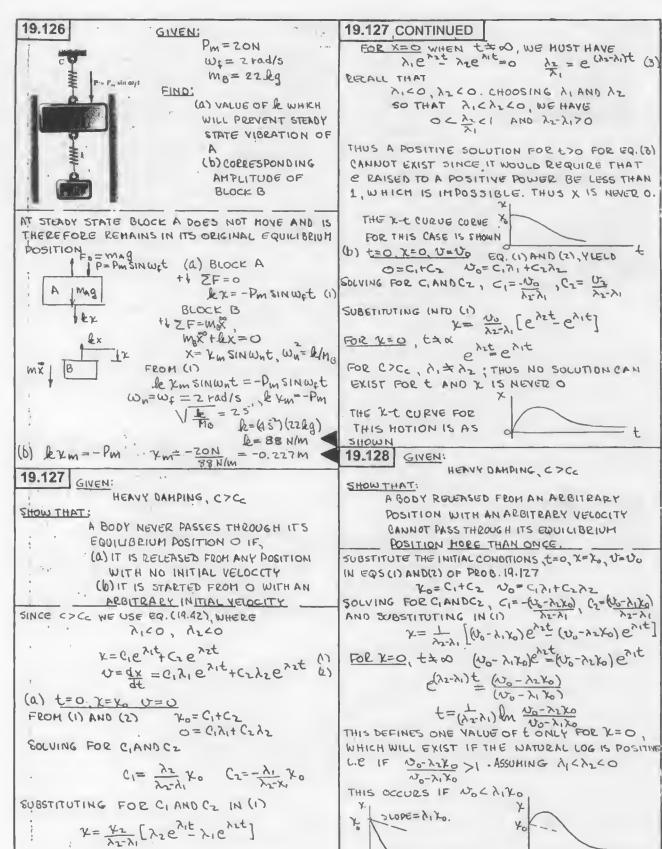


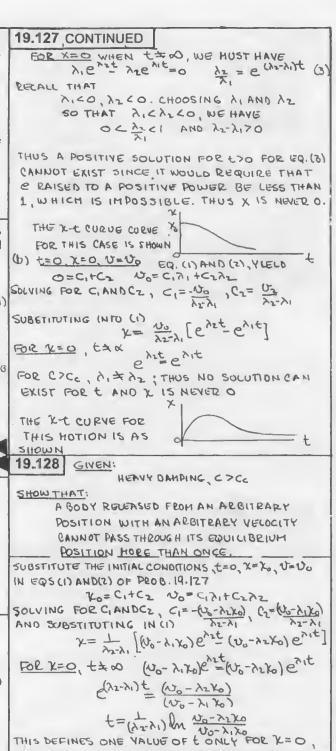
11-60,10,12/ < 1

1 < 1- (m+/m) 1

5 5 Thulsw)







19.129 GIVEN:

LIGHT DAMPING, C < CC

SHOW THAT:

THE RATIO OF ANY TWO SUCCESSIVE HAXIHUM DISPLACEMENTS Ky AND Ky IN FIG. 19.11 IS A CONSTANT AND THAT THE NATURAL LOGARITHM OF THIS PATIO CALLED THE LOGARITHHIC DECREHENT IS,

 $2n \frac{\chi_n}{\chi_{n+1}} = \frac{2\pi(c/c_c)}{\sqrt{1-(c/c_c)^2}}$

FOR LIGHT DAMPING, CCCE
EQ (19.46) Y=X0 E FIMILSIN (Wot+ \$\phi) AT GIVEN MAX. DISPLACEMENT, t= tn x= kn SIN (Wet+ +)=1, xn= x0 expension AT NEXT MAX. DISPLACEMENT, t=tn+1, X=Xn+1
SIN (Wotn+1to)=1 Xn+1= Xoe (C/2m)tn+1 BUT Wotner-Wotn=2TT

tn+-tn=211/WO

PATTO OF SUCCESSIVE DISPLACEMENTS: Kn = 100 = 5mtn = e = 5m(tu-tu-i) + 5m 200 YNTI YO C- Emthi

THUS In Kn = CTT mwo

FROM EQS. (1945) WD = WN VI- (5)2

AND (1941) WD= Cc

ZM VI-(5)2

In Ko = CTT 2M 1

ln Ku = 211 (c/c) 2

19.130 GIVEN:

LIGHT DAMPING C/C. CI

SHOW THAT:

SHOW THAT THE LOGARITHMIC DECREMENT CAN BE EXPRESSED AS 1/2 en(41/444). WHERE & IS THE NUMBER OF CYCLES BETWEEN READINGS OF THE HAVINUM DISMINIENT (b) ZERO DISPLACEMENTS OCCUP WHEN

AS IN PEOB. 19.129 FOR HAXIHUM DISPLACEMENTS Xn AND Xute AT to AND tute, SIN(wotato)=1 AND SIN (Wortnest 1) = 1. Ruth = 40 e- (zm)(tuth) Yn= Xoe-KAmten

RATIO OF MAXIMUM DISPLACEMENTS

Yn/Yn+&= Yo & C/2m)th = e = m/(tntmb) TAN(wpt+0)

BUT Wotned-wotn=leizTT) tn-tn-g=k 2TT

THUS Kn = + 5m (2km); ln kn = le CTT WWO (5)

BUT FROM PROB. 19.129 EQ.(1) LOG DECREMENT = ln kn = CTT Kuis mwo

COMPARING WITH EQ (2) LOS DEZERENENT = I In Kin Q.E.D. 19.131 GIVEN:

LIGHT DAMPING, CCCE TD= 211/WD

SHOW THAT

- (A) TIME BETWEEN A MAXIMUM POSITIVE DISPLACEMENT AND THE POLLOWING MAX NEGATIVE DISPLACEMENT IS COLZ
- (b) THE BETWEEN TWO SUCCESSIVE ZERO DISPLACEMENTS IS TO/2
- (C) TIME BETWEEN A HAXIMUM POSITIVE DISPLACEMENT AND THE FOLLOWING ZERO DISPLACEMENT IS GREATER THAN CO/4

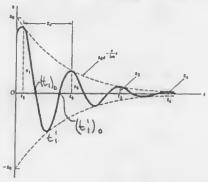


FIG. 19.11

x= to e (c/zm)t EQ. (19.46) SIN(Wotta) (a) MAXIMA (POSITIVE OF NEGATIVE) WHEN L=0

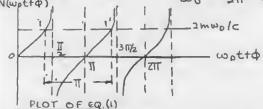
= 40 (-c/2m) = 51N (wot +0) + Vowo e (chamtecos wated)

THUS ZEED VELOCITIES OCCUR ATTIMES WHEN V=0, OR TANLWot+0)= 2mwo/c THE TIME TO THE FIRST ZERO YELOCITY & , , 15 t_= (TAN (ZMWO)C) - b] /wo (Z)

THE TIME TO THE HEXT ZERO VELOCITY WHERE THE DISPLACEMENT IS NEGATIVE, IS t, = [TAN" (2mwolc) - + 17) (W. (3)

SIN (Wotto) = 0 OR AT INTERVALS OF $\omega_0 t + \varphi = \pi, 2\pi$ NT

THUS, (t110=(1-0)/WO AND (t,) = (217-6)/WD THE BIWEEN 0'S = (ti) - (ti) = 211-11 = 11 to = TO O.E.O



(C) THE FIRST HAXIMA OCCUES AT 1, (Wotito) THE FIRST ZERO OCCURS AT (Wolfi) + 4)=TT FROM THE ABOVE PLOT (Wolfi) - (Wotito) > I OR (t,) 0-t, > 17/2000 (t,1,-t,> To/4 Q.E.D SINILAR PROOFS CAN BE HADE FOR SUBSEQUENT HAX AND HIN



GIVEN:

BLOCK IN EQUILIBRIUM AS SHOWN IS DEPRESSED IZ IN. AND RELEASED AFTER IO CYCLES THE HAXIMUM DISPLACEMENT OF THE BLOCK IS 0.5 IN.

FIND:

(a) THE DANDING FACTOR C/CC
(b) THE YALLE OF THE
CONFFICIENT OF VISCOUS DAMPING

FROM PROB 19.130 AND 19.129
(1/2) ln (4n/kn+e) = 2 TT c/ce

WHERE R= NUMBER OF CYCLES=10 VI-(C/C_)2
(a) FIRST HAXIMA IS, X = 1.2 IN.

THUS, N=1
$$\frac{Y_1}{Y_{1+10}} = \frac{1.2}{0.5} = 2.4$$

$$\frac{1}{10} \ln 2.4 = 0.08755 = \frac{2\pi c/c_c}{\sqrt{1 - (c/c_c)^2}}$$

$$(\frac{c}{c_c})^2 \left(\frac{2\pi}{0.08755}\right)^2 \left(\frac{c}{c_c}\right)^2$$

$$(\frac{c}{c_c})^2 \left(\frac{2\pi}{0.08755}\right)^2 + 1 = 1$$

$$\left(\frac{C}{C_c}\right)^2 = 1/(5150+1) = 0.0001941$$

(Eq. 19.41)

OR Cc= 2 VEM Cc= 2 VE10/ft) (9 16/32.2ft/s2) Cc= 2.991 16.5/ft

FROIT (a) &= 0.01393 C= (0.01393)(2.991)
C=0.0417 lbs/ft

19.133 GIVEN:

SUCCESSIVE MAXIMUM DISPLACEMENTS OF A SPRING - HASS-DASHPOT SYSTEM ARE 25, 15, AND 9 MM M= 18 Lg , R= 2100 N/M

EIND:

(a) THE DAMPING FACTOR C/Cc

(b) THE COFFEICIENT OF YICOUS DAHPING C.

(a) FROM PROB 19.29 In xn = 211(c/c)

FOR Ku= 25 mm AND Ruti ISMM

$$\lim_{t \to 0} \frac{25}{15} = 0.5108 = \frac{2\pi (c/cc)}{\sqrt{1 - (c/cc)^2}}$$

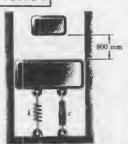
 $\left(\frac{C}{C}\right)^{2}\left[\left(\frac{10.5108}{10.5108}\right)^{2}+1\right]=1$

 $\left(\frac{C}{C_c}\right)^2 = \frac{1}{(151,3+1)} = 0.006566, \frac{C}{C_c} = 0.0810$

(b) Cc = 2m/ 2/m (EQ. 19.41)

Cc= 2 VEm = 2 /(2100 N/m)(18 lg) = 0.3888 N.S.

FROM (a) $\frac{c}{c_c} = 0.810$ c = (0.810)(.3888) = 31.5 N·S/m 19.134



GIVEN:

4-leg BLOCK A
9-leg BLOCK B
le= 1500 N/M
C= 730 N.S/M
BLOCK A IS DROPDED
FROM AN 800 MM
HEIGHT ONTO B WHICH
IS AT REST
NO REBOUND

EIND:

MAXIMUM DISTANCE BLOCKS MOVE AFTER IMPACT

VELOCITY OF BLOCK A JUST BEFORE IMPACT

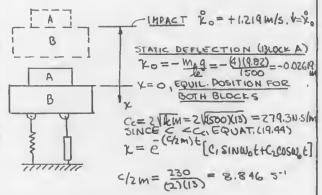
NA = VZgh = VZ(0.81)(0.8) = 3.962 m/s

VELOCITY OF BLOCK S AAND B IMMEDIATELY

AFTER IMPACT
CONSERVATION OF HOHENTUM

(4)(3,962)+0=(4+9)0

21=1,219 m/s



FRON TOP OF PAGE 1221 Wo = 1 - (Sm)

$$\omega_0 = \sqrt{\frac{1500}{13} - (\frac{230}{2243})^2} = 6.094 \text{ rad/s}$$

 $\chi = e^{-8.846t}$ (c, SIN 6.094 t+C2COS6.094t) INITIAL CONDITIONS $\chi_0 = -0.02619$ m. (t=0) $\chi_0 = +1.219$ m/s $\chi_0 = -0.02619 = e^{0}[C_1(0)+C_2(1)]$

\$(0) = -8.8462 + e-5.8463(-0.02619)+ (-.02619)(1)] 1.219 = (-8.846)(-0.02619)+ 6.084C1 -28846 (-0.02619)+ 6.084C1

X= e 8.896t (0.16202SING.094t-0.02619(056.094t)
HAXIMUM DEFLECTION OCCUPS WHEN &=0

2=0=-8.8462 (0.1620251Nb.094t-0.02619C056.094t) + e.89466(6.094)[0.1620 CQ56.094t+0.02619CNL09

0=[+8.846)(.16202)+(6.094)(002619)]SIN6.094tm +6-8.846)-0.02619)+(6.094)(0.1620))COS 6.099+m

19.134 CONTINUED

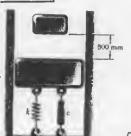
 $O = -1.274 \sin 6.094 \pm 1.219 \cos 6.094 \pm 1.219 \cos 6.094 \pm 1.219 = 0.957$

TIME AT HAX DEFLECTION = t = TAN 0.957 = 0.1753 S

7m= 68.846)(0.1253)[0.162051N(6.094)(.1253) -0.02619 cos(6094)(.1253)]

Y-m=(0.3301) (0.1120-0.0189)=0.307 in BLOCKS HOVE, STATIC DEFLEXTION + X in TOTAL DISTANCE=0.02619+0.307 =0.0569 m = 56.9 mm

19.135



GIVEN:

4. kg BLOCK A

9 kg BLOCK B

6 = 1500 N/m

C= 300 N·s/m

BLOCK A IS DROPP

FIND:

HAXI HUM DISTANCE BLOCKS MOVE AFTER IMPACT

VELOCITY OF BLOCK A JUST BEFORE IMPACT

NA = VZgh = VZ(9.81)(0.8) = 3.962 m/s

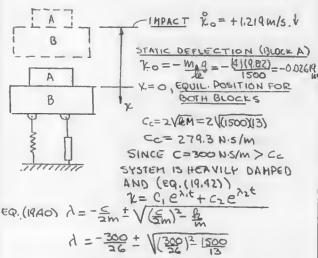
VELOCITY OF BLOCK S AND B IMMEDIATELY

AFTER IMPACT

Conservation of Hohentum

(9)(3A62)+0=(4+9)01

v=1.219 m/s=20



λ=41.538 ± 4.213

K= C, @15.751t C2 @7.325t

 $\lambda_1 = -15.751$ $\lambda_2 = -7.325$

19.135 CONTINUED

THITIAL CONDITIONS $\gamma_0 = -0.02619 \, \text{m}, \ \tilde{\chi}_0 = 1.219 \, \text{m/s}$ $\gamma_0 = \gamma_0 = -0.02619 = 0.0000 + 0.0000 \, \tilde{\chi}_0 = 1.219 \, \text{m/s}$ $\gamma_0 = \gamma_0 = -0.02619 = 0.0000 + 0.0000 \, \text{m}$ $\gamma_0 = \gamma_0 = -0.0000 \, \text{m/s}$ $\gamma_0 = \gamma_0 = -0.0000 \, \text{m/s}$ $\gamma_0 = \gamma_0 = -0.0000 \, \text{m/s}$ $\gamma_0 = \gamma_0 = -0.0000 \, \text{m/s}$

 $\chi(t) = -0.1219e^{-15.75t}$ $+0.09571e^{-7.325t}$ $\chi(t) = -0.1219e^{-15.75t}$ $\chi(t) = -0.1219e^{-15.75t}$ $\chi(t) = -7.325t$ $\chi(t) = -0.1219e^{-15.75t}$ $\chi(t) = -7.325t$ $\chi(t) =$

0=1.920e -0,701e -7.325 tm

 $\frac{1.920}{0.701} = e^{(-7.325 + 15.75)\xi_{m}}$ $2.739 = e^{(-7.325 + 15.75)\xi_{m}}$

2n2.739 = tm

tm=0.11965

- (15.75)(.1196) - (7.325)(.1196) xm = (-0.1219)e + (0.09571)e

Km= -0.01851+ 0.03986 = 0.02136 M

TOTAL DEFLECTION = STATIC REFLECTION + Km

TOTAL DEFLECTION = 0.02619 +0.02136

= 0.0475 m = 47.5 mm

19.136 GIVEN:

GUN BARREL WEIGHT = 1500 16 RECUPERATOR CONSTANT C = 1100 16-5 (ft

FIND:

(A) CONSTANT & FOR RECUPERATOR TO RETURN THE BARREL TO ITS FIRING POSITION IN THE SHORTEST TIME WITHOUT OSCILLATION (b) THE TIME NEEDED FOR THE BARREL TO HOVE TWO THIRDS OF THE WAY FROM ITS

MAXIMUM-RECOIL POSITION TO ITS FIRING POSITION

(a) A CRITICALLY DAMPED SYSTEM REGAINS

ITS EQUILIBRIUM POSITION IN THE SHOPTEST TIME

THUS C = C = 1100 = 2 m/ = 2 V & EQ (19.41)

le= (<</r>
(
100/2 16 5/ft)² = 6493.7
(1500 16 /32.2 ft/s²)

FOR A CRITICALLY DAMPEN SYSTEM EQ. (19.43)

χ= (C,+ C2t) e-ω, t

Two X

WE TAKE t= O AT HAXIHUH DEFLECTION. YO THUS \$(0)=0 \ X(0)= Xo

INITIAL CONDITIONS

X(0)=Y0=(C,+0)e° C,= K0

X=(Y0+C2+DeWnt)

k=-wn(xo+czt)e-wnt+cze-wnt

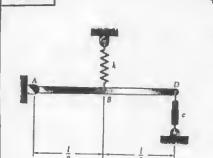
X(0)=0=-wnx0+C2 C2= wnx0

X= Y0(1+wnt)e-wnt

FOR X= Y0 1/3=(1+wnt)e-wnt wn=\fill

BY TRIAL wnt= 2.289 Un=\fill

Wn= 11.806-1 t= 2.289/11.806=0.19395



19.137

GIVEN:

PINNED AT A

FIND:

INTERMS OF

M, &, AND C

(A) DIFFERENTIAL

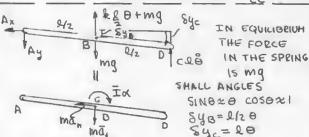
EQUATION OF

MOTION

(b) CRITICAL

DAMPING

COE FFICIENT



(a) NEWTONS LAW ZHA=(ZHA)eff

HINGRIZ-(RIZD+mg) LIZ-CLB 1 = IX+mq NZ KINEMATICS X= B Q1= R/ZX = R/ZB

 $[\bar{I} + m(\ell/2)^2] \ddot{\Theta} + C\ell^2 \ddot{\Theta} + k(\ell/2)^2 \Theta = 0$

I+m(e/2)2= = = m22

6+(3 C/M) +(3 L/AM) 0 = 0

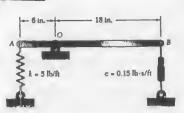
(b) SUBSTITUTING 0= ext INTO THE DIFFERENTIAL EQUATION OBTAINED IN (0), WE OBTAIN THE CHARACTERISTIC EQUATION.

2 +(3c/m)2+ 32/4m=0 AND OBTAIN THE ROOTS

$$\lambda = -3c/m \mp \sqrt{(3c/m)^2 - (3k/m)}$$

THE CRITICAL DAMPING COEFFICIENT CC, IS THE VALUE OF C IN THE RADICAL TO ZERO.
THUS

19.138



GIYEN:

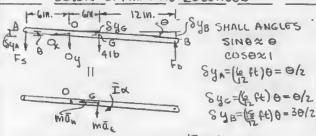
4-1b ROD AB
PINNED AT O.
AND SUPPORTED
BY A SPRING
AT A. DIMENSIONS
AND OTHER
CONSTANTS
AS SHOWN

FIND:

FOR SHALL OSCILLATIONS

(a) THE DIFFERENTIAL EQUATION OF HOTION (b) THE PORMED BY THE ROD WITH THE HORIZONTAL IS AFTER END B IS DUSHED

DOWN O.9 IN. AND RELEASED



(a) NEWTONS LAW ZMo=(ZMo)eff

1 - (= ft) Fs + (= ft) (4)- (= ft) FD = I x+ (= ft) male

 $F_{D} = k(\delta y_{A} + (\delta_{ST})_{A}) = k(\frac{\theta}{2} + (\delta_{ST})_{A})$ $F_{D} = c(\delta \dot{y}_{B}) = c(\delta/2) \delta$

I= 12 m 2= 12 m (24 ft)= 13 m

THUS FROM (I)

[m+m] + + (3/2)200 + (k/z)(g+6st))-z=0

BUT IN EQUILIBRIUM ZNO=0

(Ssr) (6)-4)(5)=0, 2(Ssr) = 2

EQ (2) BECONES (7/12) M = (9/4) C = 4/4 = 0 $\frac{7}{12}$ M $= (\frac{7}{12})(4/322) = 0.07246, 9/4 <math>= (9/4)(.15) = 0.3375$ 2/4 = 5/4 = 1.25

0.07246 θ +0.3375 θ + 1.25 θ = 0 (b) Substituting $e^{\lambda t}$ into the above differential equation 0.01246 λ^2 + 0.3375 λ + 1.25 = 0

 $\lambda = (-0.3375 \mp \sqrt{(.3375)^2 - 4(.07246)(1.25)})/2(004)$ $\lambda = (-0.3375 \mp \sqrt{-0.2484})/2(0.07246)$

λ =- 2.329 ± 3.439 L

SINCE THE KOOTS ARE COMMEX AND CONTUGATE (LIGHT DAMPING), THE SOCUTION TO THE DIFFERENTIAL EQUATION IS, (EQ. 19.46),

LEONTINDED)

19.138 CONTINUED

INITIAL CONDITIONS (SYB(O)) = 0,9 in. €(0) = (540) / 181 = 0.9 @(0)=0.05 rad A(0) = 0 FROM (3)

8(0)=0.05 = 80 51N \$

B(0) = 0 = -2.329 00 SIND +3.439 00 COS \$ TAN 0 = 3A39/2.329

Φ= 0.9755 rad

SIN (.9755) = 0.06039 rad

SUBSTITUTING INTO (3)

0=006039 = 2.329 t 5IN (3,439 t+ 0,9752)

AT t= 5 5 (-2.329/5) @(5)=0.06039@ SIN[3.439)(5)+0.9752]

9(5)=-0.333 x10 rad ⊕(5)=(0.019 09x10) ABBUE HOE 120NTAL

19.139 GIVEN:

1100-16 HACHINE SUPPORTED BY TWO SPRINGS EACH WITH &= 3000 Lb PERIODIC FORCE APPLIED OF 30-16 AT 2.8 HZ. C= 110 16/ft

FIND:

AMPUTUDE OF STEADY STATE VIBRATION

1 m = 1 ((fr - m (m)) = + (c (m)) EQ. (19.52)

TOTAL SPEING CONSTANT DE=(2)(3000 lb/ft) = 6000 lb/ft

 $W_f = 2\pi f_c = 2\pi (2.8) = 5.6\pi \text{ rad/s}$

m = w/g = 1100 1b/(32.7 ft/s2) = 34.161 16 52/ft

7 m = 3016 V((6000-(34.161)(5.611)2)2+(11045.611)2)[16]2

X= 0.00604 ft

Z = 0.0725 in.

19.140 GIVEN:

1100-16 HACHINE SUPPORTED BY TWO SPEINGS PERIODIC FORCE OF 30 16 APPLIED AT 2.8 HZ. C= 110 16/FL AMPCITUDE OF VIBRATION, 7m=0.05in

FIND:

SPRING CONSTANT OF EACH SPRING

2m= Pm V(b-mar)2+(cws)2

 $[(k-m\omega_1^2)^2+(c\omega_1^2)\chi_m^2=P_m^2$

 $R = \Lambda (s.8) = 2.91 \quad W = \frac{3}{M} = \frac{35.5 \, \text{Le}}{100 \, \text{lp}} = 34.161 \, \text{lp}.$ $R = \Lambda (b.4) - (cm^2)_2 + Mm^2$

 $k = \sqrt{\frac{3016}{105(12)}} + (34.161)(5.6\pi)^2$ R= V 51.84×106-3.745×106 +

R= 6935+10573=1750816/ft

b/2= 8750 1b/ft

19.141 GIVEN:

FORCED VIBRATING SYSTEM

FIND:

VALUES OF C/CC FOR WHICH THE MAGNIFICATION FACTOR WILL DECREASE AS WE WIN INCREASES

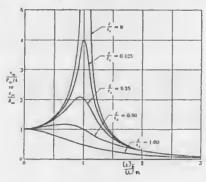


FIG. 19.12

EQ. (19.53)

MAG. FACTOR Km Pm/k V(1-(WE/WW)2]2+[2(C/C)(WE/Wn]2

FIND VALUE OF CICE FOR WHICH THERE IS NO HAXIMUM FOR KM AS WILWN INCREASES

 $\frac{d\left(\frac{V_{m}}{V_{m}}\right)^{2}}{d\left(\frac{W_{k}}{W_{m}}\right)^{2}} = \frac{-\left[2\left(1-\left(W_{k}/W_{m}\right)^{2}\right)\left(-1\right)+4\,c^{2}/c_{c}^{2}\right]=0}{\left\{\left[1-\left(W_{k}/W_{m}\right)^{2}\right]^{2}+\left[2\left(C/c_{c}\right)\left(W_{k}/W_{m}\right)^{2}\right]\right\}^{2}}$

 $-2 + 2(\omega_s/\omega_n)^2 + 4c^2/c_s^2 = 0$ $(\omega_s/\omega_n)^2 = 1 - 2c^2/c_s^2$ $c^2/c_s^2 \geqslant \frac{1}{2}$ There is no maximum for

AND THE HAGNIFICATION FACTOR WILL DECLEASE AS AS WIND INCREASES

C/C=>/1/VZ C/C=7,0,707

19.142 GIVEN:

FORCED YIBRATING SYSTEM SHALL C/CC

SHOW THAT:

HAXIHUM AMPLITUDE OCCURS WILEN WE SUM AND THAT THE CORRESPONDING VALUE OF THE HAGNIFICATION FACTOR 15 - C/cc.

EQ. (19.53')

HAG. FACTOR =
$$\frac{k_{m}}{p_{m}/k} = \frac{1}{\sqrt{(1-(\omega_{s}/\omega_{n})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{s})^{2})^{2}+(2\cdot(c/c_{s})(\omega_{s}/\omega_{$$

FIND YALUE OF WELL FOR WHICH EM 15 A HAXIHUM

$$0 = \frac{d \left(\frac{V_{mn}}{\rho_{mn}/k} \right)^{2}}{d \left(\omega_{k} / \omega_{m} \right)^{2}} = - \frac{\left[z \left(1 - \left(\omega_{k} / \omega_{m} \right)^{2} \right) \left(-1 \right) + 4 \left(c^{2} / c_{c^{2}} \right) \right]}{\left\{ \left[1 - \left(\omega_{k} / \omega_{m} \right)^{2} \right]^{2} + \left[z \left(c / c_{c} \right) \left(\omega_{k} / \omega_{m} \right) \right]^{2} \right\}^{2}}$$

-2+2(w/w) +4 (c/c) =0

FOR SHALL C/CC Wf/w. 21 Wz 2 Wn

(Pm/4) = 1 Cc

19.143 GIVEN:

15-leg MOTOR SUPPORTED BY FOUR SPEINGS EACH OF CONSTANT R= 45 RN/m

MOTOR UNBALANCE IS EQUIVALENT TO HASS OF 209 AT 125 MM FROM AXIS OF BOTATION

FIND:

AMPLITUDE OF STEADY STATE VIBRATION AT A SPEED OF 1500 PPM ASSUHING,

(a) NO DAMPING

(b) DAMPING FACTOR C/cc = 1.3

EQ. (19.52) ~m=V(k-mw})2+(cw)2 W,2=[1500)(211)/60]=24674 52 R=(4)(4500)=1800001 PmsINWst Pm= m'rw= (0.02lg)(0.125m)(246745) Pm=61.685 N 209=W1 (a) C=0 [180000-15(24674)](N/m) $V_{\rm m} = -0.324 \times 10^3 \, \text{m} = -0.324 \, \text{mm}$ (b) FOR C/C= 1.3

Eq. (19.41) Cc= 2 m/ = 2/m/e = 2 Vistg)(180000) C=(13)(3286)=4272 115 Cc= 3286 NS/M

61.685 N V(180000-15(24674)]2+(4272)2(24674)

Km=0.0884 XIU m=0.0884 mm

19.144



GIVEN:

18-kg motor BOLTED TO A BEAM HAS A STATIC DEFLECTION SST= 1.5MM UNBALANCE IS EQUIVALENT TO A HASS OF ZOG LOCATED 125 MM FROM A'MS OF ROTATION

FIND:

AMPUTUDE AT A HOTOR SPEED OF 900 PPM (a) FOR NO DAMPING

(b) FOR C/CC = 0,055 EQ. (19.52)

 $\chi_{\text{m}} = \sqrt{(k-m\omega_f^2)^2 + (c\omega_f)^2}$

m= 209 ROTOR rwet r= 125 mm sinuxt

wit=[(00)(21)/60]=8882.652 FIND SPRING CONSTANT & FOR THE BEAM R= M9 = (18kg)(9.81 52)

SST (1.5X10"3 M) le= 117720 N/m

Pm=m'rw=6.020kg)(0.125 m)(8882.652) Pm= 22,20 N

(a) C=0 22.20N 2m=((117720-(18)(8882.6 N/m) 12m=-0.527 x103 m=-0.527 mm

(b) FOR c/c= 0.055 EQ. (19.41) Cc= 2 V&m = 2 Vkm = 2 V(17770)(18) Cc= 2911 N.5/m C= 0.055 C= (0.055)(2911) = 160.12 N/M

22.21 V[117720-(18)(8883)] 34 (160.12) (160.12)

 $\sqrt{(1.779 \times 10^{9}) + (0.2278 \times 10^{9})} = 0.000496$ 26 0.496 mm

19.145

GIVEN

100-16 MOTOR BOLTED TO BEAM WHICH HAS A STATIC DEFLECTION SST= 0.25 In. UNBALANCE IS 4 02. AT 3 In. AMPLITUDE Km= 0.010 IN AT 300 rpm

FIND:

(a) DAMPING FACTOR C/Cc (b) COEFFICIENT OF DAHPING C

EQ. (19.53') V(1-(wf/wn)2)2+(2(c/c)(wf/wn)2 $\omega_n^2 = \frac{g}{\delta_{sr}} = \frac{32.2 \text{ ft/s}^2}{(6.25/2\text{ ft})}$ w= 1546 rad/s W= (300 XT/30)= 987.25 m'=402. $\left(\frac{\omega_{\rm f}}{\omega_{\rm h}}\right)^2 = \frac{987.2}{1506} = 0.63875^{-2}$

Pm=m'rwf==(416)/(32.2 ft/s2)(32ft)(987.252) Pm= 191616

R= W7 M=(1546)(100/32.2) = 4801 16/ft Pm/&= 1.916/4801 = 0 0003991 ft

$$\frac{0.01}{17} = \frac{0.0003991}{\sqrt{(1-.6387)^2 + (4)(0.6387)(c/c_c)^2}}$$

$$0.2293 = 0.1305 + 2.555(c/c_c)^2$$

$$(c/c_c)^2 = \frac{0.0988}{2.555} = 0.03867$$

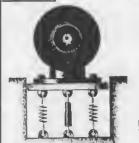
$$c/c_c = 0.1966$$

(b) EQ. (19.41) Cc= 2 MWn

Cc= 2 (10016 32.2 ft/s2) (1546) 1/2

Cc = 244.216.5/ft

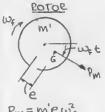
C= 0.1957 C= (244.2)(.1967) = 48.0 165 19.146



GIVEN:

100-leg MOTOR SUPPORTED BY FOUR SPRINGS EACH OF CONSTANT R= 90RN/m DASHPOT C = 6500 N.S/M AMPLITUDE Km=2.1 mm AT A SPEED OF 1200 FPM Mass of THE ROTOR m'=15 kg

DISTANCE BETWEEN THE MASS CENTER OF THE LOTOR AND THE AXIS OF SHAFT



EQ. (19.52)

m3 = (1500)(511)/60] W2=157915-2

Pm=mewi

R= 4 (90,000 N/m)=360,000 N/m

Pm=(15kg)(e)(1579152)

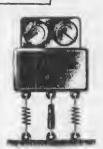
Pm= 236870.e

0.0021 = [[360,000-(100)(15791)]] (6500) (15791)]}

(1.4674 x 106) (0,0021) = (236870)e

C=0.1301 m e= (3.01 mm

19.147



GIVEN:

TWO 400-9 MASSES AT T= 150 MM POTATE AT THE SAME SPEED OF 1200 FPM IN OPPOSITE SENSES WHEN THE MASSES ARE EXACTLY BENEATH THEIR RESPECTIVE BOTATION AXES AMPLITUDE OF THE MOTION AT THIS SPEED EQUALS ISMM TOTAL MASS = 140 kg

FIND:

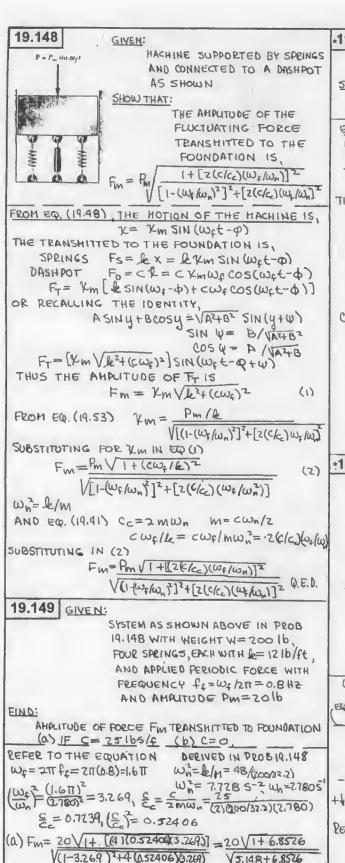
(a) THE COMBINED SPEING CONSTANT & (b) THE DAMPING FACTOR C/CE

(a) \$= 11/2, AT 1200 FPM EQ. (19.54) TAN Q = 2(C/Cc)/W/

SINCE \$= TT/2 TAND=00 THUS 1-(wo/w.)2=0

Wn=W= (1200X211)/60=40115" k= MW, = (140 kg)(40115")=2210 km

Wg/Wn=1 Pm=2mrw2f=(2)(0.4kg)(0.150m)(401151)2 Pm=1895# (0.015m) (1885/11/2210x13)] = 1 /(21c/cc)] C/Cc =0.0286 US



.19.150 GIVEN: STEADY STATE VIBRATION UNDER A HARHONIC FORCE SHOW THAT: HECHANICAL ENERGY DISSIPATED PER CYCLE IS E=TICKINUS -ENERGY IS DISSIPATED BY THE DASHPOT FROM EQ (19.48) THE DEFLECTION OF THE SYSTEM IS X= KmSIN(Wet-p) THE FORCE ON THE DASHPOT, FO = CX Fo = Cxmwf cos(wt-q) THE WORK DONE IN A COMPLETE CYCLE WITH C= ZIT/W. Foldx (L.E FORCEX DISTANCE) dx=xmwccos(wt-q)dt E= (cxmmicosmt-b) dt ws2wot-φ)=(1-2 cos(ω+t-φ))/2 E=Cxmw3, [1-2cos(w+t-0) dt E= Cxmwf [t-2 SIN(wff-4)] 24/mt E= Crimus [211 -2 (SIN(211-0)-SINO)] E= ITC KINWE Q. E. D. ***19.151** SPRING -DASHPOT SYSTEM AS SHOWN WITH HASS IN HOVING AT U, OVER A ROAD WITH A SINUSOIDAL CROSS SECTION OF AMPUTUDE Sm AND WAUELENGTH L. FIND: LANDIFFERENTIAL EQUATION OF VERTICAL DISOLACEMENT OF WASS IM (b) expression for the AHPLITUDE OF M (a) +1 ZF=ma: W- k(Ss++x-8)-c(dx-ds)=m dx PERALLING THAT W= & SST, WE WRITE m dr +cdx +lex=les+cds. (1)

(CONTINUED)

Fm= 16.18 lb

(b) (=0, Fm=(20)/15,148 = 8.81 1b

• 19.151 CONTINUED

HOTION OF WHEEL IS A SINE CURVE, S= Sm SINW, t THE INTERVAL OF TIME NEEDED TO TRAVEL A DISTANCE L AT A SPEED U, IS t= YU, which is the PERIOD OF THE ROAD SURFACE.

Thus $\omega_{s} = 2\pi/C_{s} = 2\pi \omega L$

AND 8= 8m sinwit

ds=Sm211 cos wet

THUS EQ. (1) 15.

mdik+cdx+lex=(lesinux++cu)cosux+) Su

(b) FROM THE IDENTITY

COS Q = A/VAZ+BZ

WE CAN WRITE THE DIFFERENTIAL EQUATION

M dix +C dx + lex= 8m / lot + (cut) * SIN(upt+w)

der

w = TAN-1 cut

THE SOLUTION TO THIS EQUATION & IS (ANALOGOUS TO EG'S 19.47 AND 19.48, WITH PM = Sm V&7(CW) 2)

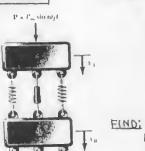
X = Xm SIN (Wft-Q+ 4) INTERE ANALOGOUS

TO EQ'S (19.52))

 $TAN \varphi = \frac{k - m \omega_t^2}{k - m \omega_t^2}$

TAN W= cws/R

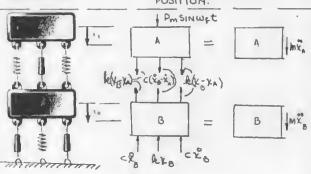
19.152 GIVEN:



BLOCKS A AND B HAVE
MASS M
THREE SPRINGS, EACH HAVE
CONSTANT &
THREE DASHPOTS, EACH
HAVE CONSTANT C.
BLOCK A ACTED UPON
BY A FORCE P. PM SINWET

DIFFERENTIAL EQUATIONS
DEFINING THE DISPLACEMENTS

YA AND YB OF THE BLOCKS
FROM THEIR EQUILIBRIUM
POSITION.



• 19.152 CONTINUED

SINCE THE ORIGINS OF COORDINATE ARE CHOSEN FROM THE EQUILIBRIUM POSITION, WE HAY OMIT THE INITIAL SPRING COMPRESSIONS AND THE EFFECT OF GRAVITY

FOR LOAD A

+ + ZF=man; Pinsinwet+Zle(xBKA)+c(xBXA)=m XA

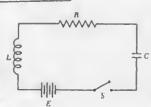
FOR LOAD B

+ TF= mas; - Ze (xg-xa) - c (xg-xa) - hxg-zcx = m x a rearranging Eqs (1) and (2), we find:

mx + c(2 - 20)+2h(4 - 20) = Pm sinust

mxg+3cxg-cha+3kxg-2kxa=0

19.153



R, L, C CIRCUIT

AS SHOWN WITH

SUDDENLY APPLIED

VOLTAGE E WHEN

THE SWITCH IS

CLOSED

WHICH OSCILLATIONS

VALUES OF R FOR

WILL TAKE PLACE
WHEN THE SWITCH S
IS CLOSED
OSCILLATIONS TAKE

FOR A HECHANICAL SYSTEM OSCILLATIONS TAKE PLACE IF C<C. (LIGHTLY DAMPED)
BUT FROM EQ. (19.41),

THEREFORE

FROM TABLE 19.2:

SUBSTITUTING IN (1) THE ANALOGOUS ELECTRICAL VALUES IN (2), WE FIND THAT OSCILLATIONS WILL TAKE PLACE IF,

R < 2 VL/C

19.154 GIVEN:

ELECTRICAL

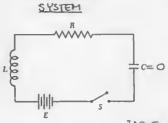
R,L,C CICCUIT OF FIG. PROB 19.153 WITH CAPACITOR C REMOYED

FIND;

IF SWITCH S IS CLOSED AT t=0

- (a) THE FINAL VALUE OF THE CURRENT IN THE CIRCUIT
- (b) THE TIME & AT WHICH THE CURRENT WILL HAVE REACHED(I- YE) TIMES IS FINAL YALUE.

 (L.C. THE TIME CONSTANT)



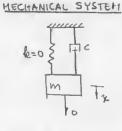


TABLE 19.2 FOR ANALOGUE CLOSING SWITCH S IS EQUIVALENT TO SUDDENLY APPLYING A CONSTANT FORCE OF MAGNITUDE P TO THE HASS

(a) FINAL VALUE OF THE CURRENT CORRESPONDS TO THE FINAL VELOCITY OF THE HASS. SINCE THE CAPACITANCE IS ZERO THE SPRING CONSTANT IS ALSO ZERO

$$P-c\frac{dx}{dt} = m\frac{d^2x}{dt^2}$$
 (1)

FINAL VEWCITY OCCURS WHEN $\frac{d^{2}x}{dt^{2}} = 0$ $P - c \frac{dx}{dt} = 0 \frac{dx}{dt} = 0$

FROM TABLE 19.2: U-OL, P-OE, C-OR
THUS

(b) READRANGING EQ. (1), WE HAVE

 $W \frac{d\xi_x}{d\zeta_x} + c\frac{d\xi}{d\zeta_x} = b$

 $m[-Ale^{-\lambda t}] + c[Ae^{-\lambda t}lp] = P$ $-m\lambda + c = 0$ $\lambda = c/m$

THUS $\frac{dx}{dt} = Ae^{\frac{1}{2}t/m} + \frac{P}{2}$ $Art=0 \frac{dx}{dt} = 0 = A + P/c \quad A = -P/c$ $w = dx = \frac{P}{1 - e^{\frac{1}{2}(1/m)t}}$

FROM TABLE 19.2. Noc, Pos, Cor More (= [1-e-(e/L)t]

FOR 1=(E/R)(1-1/e) (R/L)t=1

19.155

GIVEN:

MECHANICAL SYSTEM SHOWN

DRAW:

THE ELECTRICAL ANALOGUE



M TA TA TO THE MADE

WE NOTE THAT BOTH THE

SPRING AND THE DASHPOT

EFFECT THE HOTION OF

POINT A. THUS ONE LOOP

IN THE ELECTRICAL CIRCUIT

SHOULD CONSIST OF A

CAPACITOR (N=1/C) AND A

RESISTANCE (C=R)

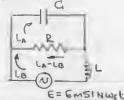
THE OTHER LOOP CONSISTS

OF (PMSINWFC->EMSINWFL), AN

INDUCTOR (M-> L) AND THE

RESISTOR (C->R)

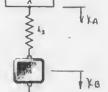
SINCE THE RESISTOR IS COMMON TO BOTH WOPS, THE CIRCUIT IS

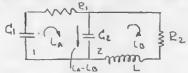


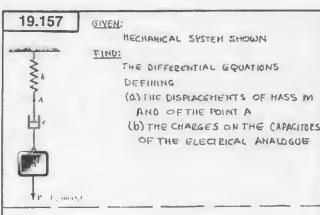
19.156 GIVEN:

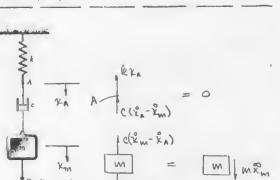
HECHANICAL SYSTEM SHOWN FIND:
THE ELECTRICAL ANALOGUE

CIECUIT









(a) HECHANICAL SYSTEM

(b) ELECTRICAL ANALOGUE

FROM TABLE 19.2

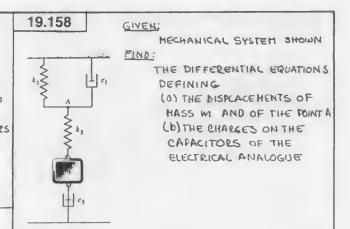
SUBSTITUTING INTO THE RESULTS FROM PART (0), THE ANALOGOUS ELECTRICAL CHARECTEISTICS.

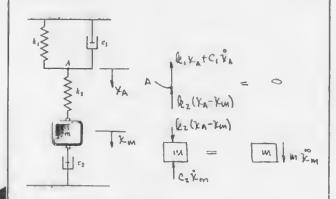
$$(1/c)G_{r} + R \frac{d}{dt}(G_{A} - G_{m}) = 0$$

$$L \frac{d^{2}G_{m}}{dt^{2}} + R \frac{d}{dt}(G_{m} - G_{A}) = E_{m} \sin \omega_{r}t$$

NOTE:

THESE EQUATIONS CAN ALSO BE OBTAINED BY SUMMING THE VOLTAGE DEOPS AROUND THE LOOPS IN THE CIRCUIT OF PROBIGISS





(0) HECHANICAL SYSTE!

MASS M

$$\xi F = MQ \cdot \frac{dx_m}{dt} = m\frac{dx_m}{dt^2}$$

(b) ELECTRICAL ANALOGUE

SUBSTITUTING INTO THE RESULTS FROM PART (0) USING THE ANALOGOUS ELECTRICAL CHARACTERISTICS FROM TABLE 19.2 (SEE LEFT).



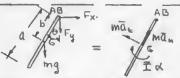


GIYEN!

THIN SQUARE PLATE
OF SIDE
OSCILLATIONS ABOUT
AB AT A DISTANCE
D FROM G

EIND:

(a) Region if b=a/z
(b) a sezond value of
b which gives the
SAHE PERIOD AS IN (a)



ZWB= (SIINB) ett

$$(\overline{1} + mb^2) \stackrel{\circ}{\theta} + mgb \stackrel{\circ}{\theta} = 0$$

$$\overline{1}_{\chi} = \frac{1}{12} ma^2 a \stackrel{\varsigma}{-1} - \chi$$

$$\stackrel{\circ}{\theta} + \frac{gb}{12} a^2 + b^2 = 0$$

$$C_{N} = \frac{2\pi}{\omega_{N}} = 2\pi \sqrt{\frac{\alpha^{2} + 12b^{2}}{129b}}$$
 (1)

(a) WHEN b= a/2

$$C_{N}=2\pi\sqrt{\frac{\Omega^{2}+12(\Omega^{3}/4)}{129(0/2)}}=2\pi\sqrt{\frac{2\Omega}{29}}$$

(b) EQUATING THE RESULT FROM PART (a) TO EQ. (1) AND SQUARING BOTH SIDES,

$$\frac{158p}{\sigma_s + 15p_s} = \frac{38}{50}$$

36b2-(24a)(b)+3a2=0

$$b = +\frac{2}{3}a \mp \sqrt{\frac{4}{9}a^2 - \frac{6}{3}} = \frac{a}{2}, \frac{a}{6}$$

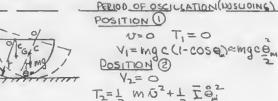
19.160



GIVEN:

HALF SECTION OF A SOCIO CYLINDER IS ROTATED THROUGH A SHALL ANGLE AND RELEASED

FIND:





~=(r-c) 0

INSTANTANGOUS

Io=I+mcz I=Io-mcz= 7 m Lz-mcz= M[+z-cz]

$$T_2 = \frac{1}{2} M \left[(r-c)^2 + (\frac{r^2}{2} - c^2) \right] \hat{\theta}_M^2$$

$$C = \frac{4r}{3\pi} \qquad T_2 = \frac{1}{3} \text{ m} \left[\frac{3}{3} r^2 - 2 \left(\frac{4r}{3\pi} \right) r \right] \theta_{\text{m}}^2$$

$$T_3 = \frac{1}{3} \text{ m} r^2 \left[\frac{3}{3} - \frac{9}{3\pi} \right] \theta_{\text{m}}^2 = 0.3256 \text{ m} r^2 \theta_{\text{m}}^2$$

T, + U, = T2 + V2

FOR SHALL OSCILLATIONS

 $g + 0.2122 \Theta_m^2 = 0.3276 \text{m} \text{m}^2 \Theta_m^2 \omega_n^2$ $\omega_n^2 = \frac{0.2122}{0.3276} \frac{q}{r} = 0.6518 \frac{q}{2} \text{ s}^2$

$$T_{n} = \frac{2\pi}{\omega_{n}} = \frac{2\pi}{0.8073} \sqrt{\frac{1}{9}}$$

$$T_{\text{N}} = 7.78 \sqrt{\frac{r}{g}} \text{ S}$$

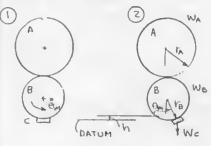
19.161

GIVEN:

WA=30 16, WB=1216 WC= 516, ATTACHED TO B NO SCIPPING

FIND:

PERIOD OF SHALL OSCILLATIONS



SHALL OSCILLATIONS h= 18 (1-COSOm) 2 18 03/2 POSITION (1) TOBB = TA OA

$$T_{1} = \frac{1}{2} m_{c} (r_{6} \delta_{m})^{2} + \frac{1}{2} \overline{L}_{8} \delta_{m}^{2} + \frac{1}{2} \overline{L}_{A} (\frac{r_{6}}{r_{6}} \delta_{m})^{2}$$

$$\overline{L}_{8} = \frac{m_{6} r_{6}^{2}}{2} \overline{L}_{A} = \frac{m_{A} r_{A}^{2}}{2}$$

$$T_{1} = \frac{1}{2} [m_{c} r_{6}^{2} + m_{6} r_{6}^{2} / 2 + (m_{A} r_{A}^{2} / 2) (r_{6} / r_{A})^{2}] \delta_{m}^{2}$$

$$T_{1} = \frac{1}{2} [(m_{c} + m_{6} / 2 + m_{A} / 2) r_{6}^{2} \delta_{m}^{2}$$

$$V_{1} = 0$$

POSITION (2)

V2= Mcg h= Mcg 0 m/2

 $T_1+V_1=T_2+V_2$ $S_m=W_n\Theta_m$

1 [(mc+ ma/2+ Ma/2] /3 wn on + 0 =

0'+ magra = 1/2

$$w_n^2 = \frac{5}{5 + (12+30)/2} \frac{(32.2 + 1/5^2)}{(6/12) + 1}$$

Wn= 12.39 5-2

$$C_n = \frac{2\pi}{\omega_n} = \frac{2\pi}{\sqrt{0.39}} = 1.785 \text{ s.}$$

19.162



GIVEN:

4.02 GYEDSCOPE BOTOR; TN= 6.00 S WHEN ROTOR IS SUSPENDED FROM A WIRE AS SHOWN WHEN 1.25 IN DIANGTER SPHERE IS SUSPENDED IN THE SAME FASHION THE PERIOD (Cn) = 3.805.

FIND:

RADIUS OF GYRATION & OF THE SOTOS

K= SPRING CONSTANT OF THE WIRE

FOR SPHERE OR ROTOR



ZMo=(EHO)eff

$$\omega_{N^{2}} = \frac{1}{4}$$
 $C_{N} = \frac{2\pi}{\omega_{N}} = 2\pi\sqrt{\frac{1}{2}}/k$

FROM (1)
$$6=2\pi\sqrt{2}$$

$$M = \frac{4}{3} \pi \frac{W = (Vol)(SPWE) = \frac{4}{3} \pi r^{3} | e}{[32.2 ft/s^{2}]} [490 lb/ft^{3}]$$

M= 9.006 x 10-3 16.53/ft

$$\bar{I}_{s} = \frac{2}{5} (9.006 \times 10^{3} \text{ lb·s}^{2}/\text{fe}) [(1.25/2)/(12)\text{fe}]^{2}$$

$$\bar{I}_{s} = 9.772 \times 10^{6} \text{ lb·s}^{2}.\text{ft}$$

FROM (1)
$$3.80 = 2\pi \sqrt{\frac{9.172 \times 10^{-6}}{}}$$
 (3)

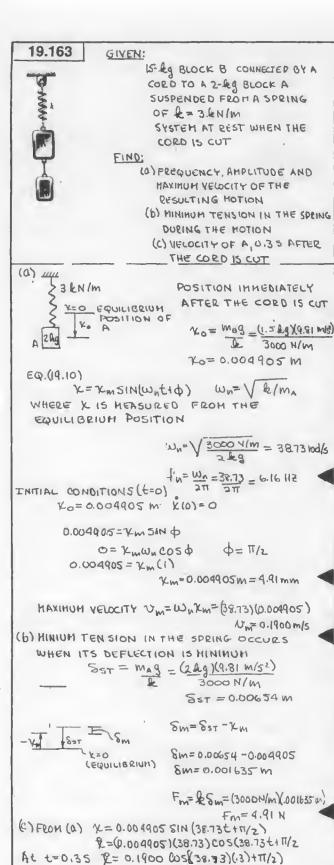
DIVIDE EQ. (2) BY EQ. (3) AND SQUARING .

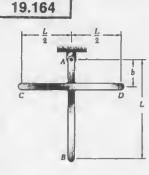
$$\left(\frac{6}{3.80}\right)^2 = \frac{\left(7.764\times10^3 \text{ lb s}^2/\text{fe}\right)^{\frac{1}{2}}}{\left(7.72\times10^6 \text{ lb s}^2\text{ ft}\right)}$$

$$\overline{k}^{2} = \frac{(9.712 \times 10^{6})}{(7.764 \times 10^{3})} \left(\frac{6}{3.80}\right)^{2} = 3.138 \text{ ft}^{2}$$

$$\overline{k}^{2} = 0.0560 \text{ ft}$$

E=0.672 in.





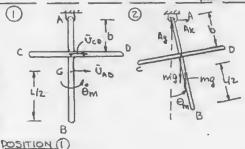
GIVEN:
TWO CODS EACH OF
MASS M AND LENGTH
L WELDED TOGGTHE

L WELDED TOGETHER TO FORM THE ASSEMBLY SHOWN

EIND:

(A)THE DISTANCE D
FOR WHICH THE
FREQUENCY OF SHALL
OSCILLATIONS IS
MAXIMUM
(b) THE CORRESPONDING

HAXIHUM FREDUENCY



 $T_{i} = \frac{1}{2} \left[m \tilde{\Omega}_{co}^{2} + m \tilde{\Omega}_{AB}^{2} + \tilde{\Gamma}_{co} \tilde{\theta}_{m}^{2} + \tilde{\Gamma}_{AB} \tilde{\theta}_{m}^{2} \right]$

Jos bom JAG=(LIZ)OM

Ico=Ino= 12 m 12

T1= = 1 m [b2+(L/2)2+ 12 L2++2 L2] == 1 [b2+51/2] 0 m

POSITION (2)

V= mg b (1-cosem) + mg yz(1-cosem)

SHALL ANGLES 1-cosem= 2 SINYEM/2)= @m/2

 $V_2 = Mg \frac{\Theta}{2}^m (b + L/2)$

T1=0

Ti+Vi=Ti+Vz

 $\omega_{n}^{2} = \frac{g(b+1/2)}{(b^{2}+5/(2^{12}))}$

MAXWIZ WHEN dwildb = 0

 $p = -\Gamma \pm \sqrt{\frac{13+(50/13)\Gamma_{3}}{15}} = 0.319\Gamma^{3}1911\Gamma$ $= \frac{qp}{qp} + \frac{(p_{3}+2\sqrt{13}\Gamma_{3})\Gamma_{3}}{(p_{4}+2\sqrt{13}\Gamma_{3})\Gamma_{3}} = 0$ $= \frac{qp}{qp} + \frac{(p_{4}+2\sqrt{13}\Gamma_{3})\sigma_{3}\sigma_{3}(p_{4}+\sqrt{15})(5p)}{(p_{4}+2\sqrt{13}\Gamma_{3})\sigma_{3}\sigma_{3}(p_{4}+\sqrt{15})(5p)} = 0$

p= 0.316 F

(1)

(b) FROM ED (1) AND THE ANSWER TO (0)

$$\psi_{N}^{2} = \underbrace{9[0.316+0.5]}_{(0.316)^{2}+5/12]L} = 1.5809/L$$

$$\psi_{N}^{2} = \underbrace{\sqrt{1.580}}_{2\pi} \sqrt{9}L = 0.200\sqrt{9}L + 1.580$$

& (0.3) = 0.1542 W/S V

19.165 GIVEN:

SPRING SUPPORTED HOTOR SPEED INCREASED FROM 700 FPM TO 500 FPM
AMOUTUDE OF VIGRATION DECREASE

CONTINUOUSLY FROM 8 MM TO 2.5 MM

FIND:

(a) RESONANT SPEED

(b) AMOUTUDE OF STENDY STATE VIBEATION AT ICOM

(1) FOR A HOTOR WITH A ROTOR UNBALANCE THE AMPLITUDE OF VIBRATION IS GIVEN BY (SEE SAMPLE PROB 19.5)

AT 200 FPM

$$-8 = \frac{(1 - (500)^{2})}{M \cdot (500)^{2}}$$
 (1)

AT 500 rpm $-2.5 = \frac{Mr(500)^2/k}{(1-(500)fm)^2}$ (2)

DIVIDING GO (1) BY EQ. (2) TERM BY TERM,

$$\frac{5.5}{8} = \frac{(-(500/t^{N})^{5})(500)^{5}}{(-(500/t^{N})^{5})(500)^{5}}$$

$$(3.2)(1-(200/f_n)^2) = 0.160(1-(500/f_n)^2)$$

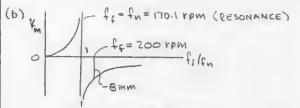
$$(3.2-0.160)(f_{N}^{2})=3.2(200)^{2}-(0.160)(500)^{2}$$

fn= 170.14 rpm

fn= 170.1 rpm

RESONANCE WHEN ff=fn

ff = 170.1 rpm



$$\chi_{m} = \frac{mr}{k} \omega_{\phi}^{2} \qquad \text{AT 200 rpm } \omega_{\phi} = 2\pi(200)$$

$$\omega_{\phi}^{2} = 20\pi r_{\phi} + 2\pi r_{\phi}^{2} + 2\pi r_{\phi}^{2} = 2\pi r_{\phi}$$

$$-8 = \frac{Mr}{1 - (200/170.14)^2} (EQ.1)$$

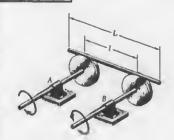
$$\frac{mr}{le} = \frac{(-8)(-0.3818)}{(20\pi/3)^2} = 0.006963$$

AT 100 PM

$$\chi_{\rm m} = (0.006963)(\frac{10\,{\rm ff}}{3})^2 = 1.1666\,{\rm mm}$$
 $1 - (100/170.14)^2$

2m=1.167 mm

19.166



GIVEN:

ROD OF HASS MAND
LENGTH L RESTS
ON TWO PULLEYS
WHICH ROTATE
IN OPPOSITE
DIRECTIONS AS
SHOWN
ME COEFFICIENT
OF KINETIC FRKING
BETWEEN THE ROD
AND THE PULLEYS

END:

FREQUENCY OF VIBRATION IF THE ROO IS GIVEN A SHALL DISPLACEMENT TO THE RIGHT

+PZFy=Z(Fy)eff:

$$N_B=(\frac{1}{2}+\frac{\chi}{2})mg$$
 - $mg=0$
 $N_A=(\frac{1}{2}-\frac{\chi}{2})mq$

FB= 4 NB = 1 = 1 = 1) mg

FB= 4 NB = 1 = 1 = 1 | mg

FA-FB = MX

$$T$$
 $ZF = Z(F_x/eff$

$$t = \frac{3\pi}{\Omega^*} = \frac{3\pi}{1} \sqrt{\frac{\delta}{5\pi^2 \delta}}$$

19.167

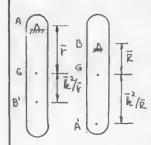
GIVEN:



COMPOUND PENDUCH WITH ENIFE EDGES AT A AND B A DISTANCE L APART COUNTERWEIGHT D IS ADJUSTED SO THAT THE PERIOD IS THE SAME WHEN EITHER KNIFE EDGE IS USED

SHOW THAT!

THE PERIOD IS THE SAME AS A SIMPLE DENDULUM OF LENGTH & (LE The 2TV RIG)



FROM PROB 19.52 THE LENGTH OF AN EQUIVALENT SIMPLE PENDULUM IS:

211 V 20 = 211 V 20

THAT IS, A=A' AND B=B'

$$\tau = 2\pi \sqrt{\frac{1}{9}}$$

19.168

GIVEN:

400-Lg MOTOR SUPPORTED BY
FOUR SPEINGS. EACH SPRING HAS
A CONSTANT OF 150 LN/M
UNBALANCE IS 23 Q AT 100 MM
FROM THE AXIS OF ROTATION

FIND:

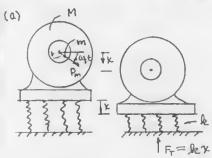
FOR A SPEED OF 800 PPM

(A) THE AMPUTUDE OF THE FLUCTUATING

FORCE TRANSMITTED TO THE FOUNDATION

(b) THE AMPUTUDE OF THE VERTICAL

MOTION OF THE MOTOR



FLOH EQ (19.33)

$$Y_{m} = \frac{P_{m}/\ell_{e}}{1 - (\omega_{e}/\omega_{n})^{2}} \tag{1}$$

THUS
$$F_T = 2 \kappa_M = \frac{P_m}{1 - w_t^2/w_n^2}$$
 (2)

$$m_s^t = (sut^t)_s = [(su)(800/60]_S = 1018 s_s]$$

Pm=mrw==(0.023 lg)(0.100 m)(7018 52) Pm=16.14 N

SUBSTITUTING THE ABOVE VALUES INTO 69.2

$$F_T = \frac{16.14}{1-(7018/1500)} = -4.388 N$$

(b)
$$\chi_{m} = Fr/k = \frac{(4.388 N)}{(600,000 N/m)}$$
 $\chi_{m} = 0.00731 \times 10^{-3} m$

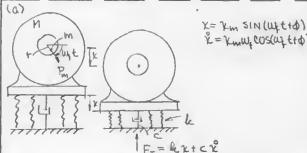
2m=0.00731 MM

19.169 GIVEN:

400- LA HOTOR SUPPORTED BY FOUR SPRINGS EACH WITH &= 150 &N/M. AND A DASHPOT WITH C= 6500 N.S/M UNBALANCE IS 23 9 AT 100 MM FROM THE AXIS OF ROTATION

FIND:

FOR A SPEED OF 800 rpm (a) ANDLITUDE OF THE FLUCTUATING FORCE TRANSHITTED TO THE FOUNDATION (b) AMPLITUDE OF THE YERTICAL HOTION OF THE HOTOR



X = Kmw, cos(uz t+0)

FT=lex+cx=lexm SINWet+ 0)+cxmwscoswft

le= 4 (150,000 N/m)= 600,000 N/m W= k/M= 600,000/400 = 15005-2 $m_s^t = (sut^t)_s = (su(800)/80)_s = 2018 e_s$ Pm=mrw== (0.023kg)(0.100m)(7018)=16.14 N

FROM (2)
$$\gamma_{m} = \frac{16.14}{\sqrt{(600,000 - 400)(7018)^{2} + (6500)^{2}(7018)}}$$

 $\gamma_{m} = 7.10 \times 10^{-6} \text{ m}$ (3)

7-m= 7.10 × 10-6 m

(FT) = 5.75 N

(b) FROM (3)

2m= 0.00710 mm

NOTE: COMPARING RESULTS WITH PROB. 19.168 IN WHICH THERE IS NO DASHPOT, THE AMPLITUDE OF THE FORCE HAS INCREASED WHILE THE AHDUTUDE OF VERTICAL HOTION DECREASES.

19.170



GIVEN!

SHALL HASS IN ATTACHED TO AN ELASTIC CORD OF LENGTH & IN A HORIZONTAL PLANE

TENSION IN THE CORD REMAINS CONSTANT AS THE BALL IS GIVEN A FHALL DISPLACE HENT PERPENDICULAR TO THE CORD AND RELEASED

FIND:

- (a) DIFFERENTIAL EQUATION OF MOTION OF THE BALL
- (b) THE PERIOD OF YIBRATION



+ P ZF= ma

FOR SHALL X SIND & TANO = X /(2/2)

$$MX + 2T X = 0$$

$$C_{n} = \frac{2\pi}{\omega_{n}} = \frac{2\pi}{\sqrt{4\pi/m}} = \pi \sqrt{\frac{m\ell}{T}}$$

19.C1 GIVEN:

PERIOD OF A SIMPLE PENDULUM OF LENGTH & 15, $\mathbb{C}^{N=3L}\sqrt{\frac{d}{d}}\left[1+\left(\frac{1}{2}\right)_{C_{3}}+\left(\frac{1\times3}{1\times3}\right)_{C_{4}}+\left(\frac{1\times3\times2}{1\times3\times2}\right)_{S_{6}}+\cdots\right]$

WHERE C = SIN 10m AND OM ISTHE AMPLITUDE

FIND:

THE SUM OF THE SERIES IN BRACKETS USING SUCCESSIVELY 1, 2, 4,8 AND 16 TERMS FOR VALUES OF & IM FROM 30°TO 120° USING 30° INCREMENTS. EXPLESS RESULTS WITH FIVE SIGNIFICANT FIGURES

REWRITE GIVEN SERIES INTERHS OF N= 1,2,3....

$$AMPLITUDE = \Theta_{m} \qquad C = \frac{1}{2} \sin \Theta_{m}.$$

$$LET \quad \gamma = 2\pi \sqrt{\frac{I}{3}} \left[B \right] \qquad \text{where} \quad B = \left[1 + \left(\frac{1}{2} \right)^{2} e^{2} + \left(\frac{113}{217} \right)^{2} e^{4} + \left(\frac{1\times3\times5}{217\times6} \right)^{2} e^{6} + \cdots \right]$$

WE MAY COMPUTE & AS FOLLOWS:

$$n = 1: \qquad B = \left[1 + \left(\frac{2n-1}{2n}c\right)^{2}\right]$$

$$n = 2: \qquad B = \left[1 + \left(\frac{2n-1}{2n}c\right)^{2}\right]$$

$$n = 3: \qquad B = \left[1 + \left(\frac{2n-1}{2n}c\right)^{2}\right]$$

AT EACH STEP 7.5 GUANITY ABOVE THE ____ IS THE CHANGE IN B AND IS DENOTED BY "DELTA

CUTLINE OF PROGRAM

CALCULATE C= \$ SIN Om FOR OM = 30° CALCULATE BUSING THE ALGORITHM ABOVE FOR N= 1,2,4,8,16 PRINT B FOR OM AND N REPEAT FOR OM = 60°, 90° AND 120°

PROGRAM OUTRIT

Amplitude =	36 degrees			98 degrae
H	Brecket		N	Bracket
1	1.01675		3	1.12500
'E	1.01738		3	1.16014
4	1.01741		4	1.17704
	1.01741			1.18422
16	1.01741		16	1.18834
Amplitude =	AB degrame		Amplitude =	120 degrame
N	Bracket		н	Bracket
1	1.06250		1	1.19758
2	1.67129		2	1.26660
4	1.07311		4	1.33144
0	1.07310			1.34448
14	1.07310		16	1.37248
		and the same		

19.C2

GIVEN:

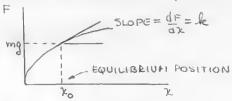
FORCE DEFLECTION EQUATION FOR A CLASS OF SPRING IS F=5X^{1/N} WHERE F IS IN NEWTONS AND X IS THE DEFLECTION IN HETERS

FIND:

FOR A BLOCK OF MASS M JUSPENDED FROM
THE SPRING AND IS GIVEN A SHALL DOWNWARD
DISPLACEMENT FROM ITS EQUILIBRIUM POSITION,
THE FREQUENCY OF VIBRATION OF THE
BLOCK FOR M= 0.2,0.6 AND 1.0 &g AND
FOR VALUES OF N FROM ITOZ USING
0.2 INCREMENTS

ANALYSIS

FORCE- DEFLECTION CURVE F=5x VA



$$k = \frac{dF}{dx} = \frac{5}{N} x^{N-1} = \frac{5}{N} x_0^{\frac{1-N}{N}}$$

$$w_N = \sqrt{\frac{1}{N}} = \sqrt{\frac{5}{N}} x_0^{\frac{1-N}{N}}$$

$$f_N = \frac{w_N}{2\pi} = \frac{1}{2\pi} \sqrt{\frac{5}{N}} x_0^{\frac{1-N}{2N}}$$
(1)

FOR ANY Mg, THE EQUILIBRIUM POINT IS F= Mg= 5 x."

$$K^{0} = \left(\frac{2}{md}\right)_{N} \tag{5}$$

OUTLINE OF PROGRAM

- 1. CALCULATE TO FROM EQ (2) FOIR M=0.2 kg
 AND N= Z
- 2. SUBSTITUTE YO FROM (2) INTO (1),
- 3. CALCULATE I'M, AND PRINT I'M, M AND M
- 4. REDEAT STEPS 1-3 FOR N=1.8,1.6,1.4, 1.2 AND 1.0
- 5. PEPENT STEPS 1-4 FOR M= 0.6 AND 1.0 kg

PROGRAM OUT PUT

ก	m (kg)	f (Hz)
2.0	0.20	0.898
1.8	0.20	0.862
1.6	0.20	0.833
1.4	0.20	0.811
1.2	0.20	0.798
1.0	0.20	0.796
2.0	0.60	0.299
1.8	0.60	0.321
1.6	0.60	0.346
1.4	0.60	0.376
1.2	0.60	0.413
1.0	0.60	0.459
2.0	1.00	0.180
1.8	1.00	0.203
1.6	1.00	0.230
1.4	1.00	0.263
1.2	1.00	0.304
1.0	1.00	0.356

19.C3

P. P. dia o., I

GIVEN:

HACHINE ELEHENT SUPPORTED
BY SPRINGS AND CONNECTED TO
A DASHPOT IS SUBJECTED
TO A PERIODIC FORCE AS
SHOWN

EIMD:

FOR FREQUENCY RATIOS WE/WAR EQUAL TO 0.8,1.4 AND 2.0 AND FOR DAMPING FACTORS C/C. EQUAL TO 0,1, AND 2, THE TRANSMISSIBILITY TIME FM /PM WHERE FM IS THE MAXIMUM FOR CE TRANSMITTED TO THE POUNDATION TO THE MAXIMUM VALUE PM.

ANALYSIS

FROIT PEOB. 19.148,

OUTLINE OF PROGRAM (USING THE ABOVE PROGRAM)

- I INPUT C/CC=0
- 8.0 = NOUT WATER
- 3. CALCULATE TM AND PRINT FOR C/CC AND WILW THE VALUE OF TM
- 4. REPEAT SIEPS 2 AND 3 FOR $\omega_t/\omega_n = 1.4$ AND THEN FOR $\omega_t/\omega_n = 7.0$
- 5. REPEAT STEPS ITHROUGH 4 FOR C/Cc=1.0
 AND THEN FOR C/Cc= 2.0

PEDGEAM CUTPUT

w f/wn		cke	Tim
	PREQ. RATIO	DAMPING FACTO	R TRAN. RATIO
	0.80	0.0	2,778
	1.40	0.0	1.042
	2.00	0.0	0.333
	0.80	1.0	1.150
	1.40	1.0	1.004
	2.00	1.0	0.825
	0.80	2.0	1.041
	1.40	2.0	1.001
	2.00	2.0	0.944

19.C4

GIVEN:

15- Lg HOTOR SUPPORTED BY FOUR SPRINGS EACH OF CONSTANT GO RN/M. UNBALANCE EQUALS 20 g AT 125 MM FROM AXIS OF ROTATION.

EIND:

AMPLITUDE AND ACCELERATION FOR MOTOR SPEEDS OF 1000 TO 2500 rpm using 100 rpm INCREMENTS

ANALYSIS

FROM EQ (19.33) $R_{m} = \frac{P_{m}/R_{c}}{1 - (\omega_{f}/\omega_{n})^{2}} \qquad (1)$

WHERE PM = MY WIT (SAMPLE PROB. 19.5)

 $k = 4 \times 60,000 \text{ N/m} = 240,000 \text{ N/m}$ $P_{\text{M}} = (0.020)(0.125) \omega_{\text{f}}^2 = 2500 \times 10^6 \omega_{\text{f}}^2$ $\omega_{\text{h}}^2 = \frac{1}{15} = \frac{240,000}{15} \text{ N/m} = 16000 5^{-2}$

SUBSTITUTE THE ABOYE VALUES INTO (1)

$$\gamma_{M} = \frac{(2500 \times 10^{6} M_{\odot}^{2})/(240,000)}{1 - M^{2}/16000} M (2)$$

 $0 m = m_5 km m/s_5$ (3

$$\omega_t = (\epsilon_{bh})(s_{\mu})/60 \tag{4}$$

OUTLINE OF PROGRAM

- 1. USING EQ.(2) AND NOTING EQ.(4) INPUT AN INFIAL VALUE OF HOTOR SPEED OF 1000 PPM
- 2. CALCULATE KM
- 3. CALCULATE FROM EQ. (3), QM
- a. PRINT FOM, KM AND am
- 5. REPEAT STEPS I THEOUGHA FOR HOTOR LPEEDS OF 1100 TO 2500 FPM IN STEPS OF 100 FPM

PROGRAM OUT PUT

TO OBTAIN THE UNITS CORRESPONDING TO THE ANSWERS, BELOW, HULTIPLY EQ. (2) BY (000, AND IF THE RESULT (INMM) IS USED IN EQ (3), DIVIDE IT BY 1000.

SPEED (RPM)	AMP. (mm)	ACCEL. (m/s * * 2)
1000	0.363	3,98
1100	0.810	10.75
1200	12.615	199.21
1300	-1.219	22.60
1400	-0.652	14.02
1500	-0.474	11.70
1600	-0.388	10.88
1700	-0.337	10.67
1800	-0.303	10.77
1900	-0.280	11.07
2000	-0.262	11.51
2100	-0.249	12.05
2200	-0.239	12,66
2300	-0.230	13.35
2400	-0.223	14.10
2500	-0.217	14.90

19.C5 GIVEN:

SAME AS 19.C4 AT LEFT WITH A
DASHPOT HAVING A COEFFICIENT
OF DAMPING C = 2.5 &N /5 IS
CONNECTED TO THE MOTOR BASE
AND THE GROUND

FIND:

AMPUTUDE AND ACCEPTATION FOR HUTOL SPEEDS OF 1000 TO 2500 FPM USING 100 FPM INCREMENTS

ANALYSIS

FROM EQ. (952)
$$V_{m} = \frac{P_{m}}{\sqrt{(k_{-}M\omega_{k}^{2})^{2} + (C\omega_{k})^{2}}}$$
 (1)

 $k = 4 \times 60,000 \text{ N/m} = 240,000 \text{ N/m}$ $p_m = m r \omega_t^2 = (020)(0.125) \omega_t^2 = 2500 \times 10^6 \omega_t^2$ SUBSTITUTE INTO (1)

$$GM = M_2 \times M$$
 (3)

$$\omega t = (8bu)(SLL)/60$$
 (4)

OUTLINE OF PROGRAM

- I USING & Q.(2) AND NOTING EQ.(4), INPUT AN INITIAL VALUE OF HOTOR SPEED OF 1000 FPM
- 2. CALCULATE Km (IN METERS)
- 3. CALCULATE PROM ED. (3) THE ACCELERATION
- 4. PRINT FPM, Km, am
- 5. REPEAT STEPS I THROUGH 4 FOR HOTOR SPEEDS OF 1100 TO 2500 FPM IN INCREMENTS OF 100 FPM

PROGRAM OUT PUT

SEE NOTE AT LEFT

SPEED (RPM)	AMP. (mm)	ACCEL. (m/s * * 2
1000	0.1006	1.103
1100	0.1140	1.513
1200	0.1257	1.984
1300	0.1353	2.507
1400	0.1430	3.074
1500	0.1491	3.679
1600	0.1538	4.318
1700	0.1574	4.987
1800	0.1601	5.688
1900	0.1621	6.419
2000	0.1637	7.180
2100	0.1649	7.972
2200	0.1657	5.796
2300	0.1664	9.653
2400	0.1669	10.542
2500	0.1673	11.464

19.C6

GIVEN:

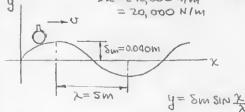
TRAILER AND LOAD MASS = 250 29 SUPPORTED BY TWO SPRINGS EACH WITH R= 10 RN/m POAD IS A SINE CURVE WITH AN AMPUTUDE OF 40 MM IND WAVE LENGTH OF 5 m.

EIND:

(6) ANDLITUDE OF VIBRATION AND MAXIMUM VERTICAL ACCELERATION OF THE TRAILER FOR SPEEDS OF 10 TO BORMIN USING 5 RMIN INCREMENTS

(b) USING APPROPRIATE SHALLER INCREMENTS DETERHINE THE RANGE OF VALUES OF THE SPEED FOR WHICH THE TEMLER WILL LEAVE THE GROUND.

MALYSIS (0) M y = Smsinwit 2 k= 2.10,000 N/M y = 20,000 N/M



THUS
$$\chi_{m} = \frac{40 \text{ mm}}{1 - (2\pi \pi U(1000))^{2}/80} \text{ mm}$$
 (1)

(b) WHEN 4 AND X. ARE IN PHASE THEY HAVE THE SAME SIGN (LE Km+)

GROUND WHEN THE FORCE IN THE SPRING IS ZERO. THIS OCCURS WHEN 7,>Sm+SsT WHERE

SST = 0.1226M = 122.6 MM THUS WHEN Km>122.6+40=162.6mm THE TRAILER WILL LEAVE THE GROUND WHEN Y AND & ARE OUT OF PHASE (7m-) THE TRAILER WILL LEAVE THE GROUND WHEN 74m<-122.6+40=-87.6mm

19.C6 CONTINUED

SF

OUTLINE OF PROGRAM

(4) INPUT TO EQ. I VALUES OF VELOCITY FROM 10 TO BO Rm/h IN 5 Em/h INTERVALS AND PRINT THE RESULTS PROGRAM OUTPUT

SPEED (km/h)	AMPLITUDE (mm)
10.0	47.19
15.0	60.85
20.0	102.36
25.0	832,11
30.0	-107.88
35.0	-46,20
40.0	-27.84
45.0	-19.19
50.0	-14.25
55.0	-11.09
60.0	-8.92
65.0	-7.36
	-6.19
70.0	-5.29
75.0	
80.0	-4.57

(b) FROM PART (b) OF THE ANALYSIS WE NOTE THAT IF Xm7167.6 mm or Xm <-826.mm THE TRAILER WILL LEAVE THE GROUND. FROM THE RESULTS OF PART (0) WE NOTE THAT THIS OCCURS BETWEEN THE VELOCITIES OF ZOEM/h AND 35ML/h REPUN EQ. (1) FOR VELOCITIES OF 20 km/h TO 35 Rm/h AT INTERVALS OF O.I Rm/h AND

	., .,, ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	On Acres	0 000
PRINT THE	ERESULTS		
PEED (RH/N)	AHPLITUDE (MM)	SPECD(LIM/h)	AMPLITUME (W)
22.2	160.41	26.8	-425.78
22.3	164.89	26.9	-391.68
22.4	169.65	27.0	-362.54
		27.1	-337.35
		27.2	-315.35
22.7	ONTACT160.13	27.3	-295.98
22.8	192.09	27.4	-278.79
22.9	198.73	27.5	-263.44
23.0	205.88	27.6	-249.64
23.1	213.60	27.7	-237.17
23.2	221.96	27.8	-225.85
23.3	231.04	27.9	-215.53
23.4		28.0	-206.08
	240.94	28.1	-197.39
23.5	251.77		-189.37
23.6	263.68	28.2	-181.96
23.7	276.82		-175.08
23.8	291.41	28.4	-168.68
23.9	307.70	28.5	-162.72
24.0	326.00	28.6	
24.1	346.70	28.7	-157.14
24.2	370.31	28.8	-151.91
24.3	397.49	28.9	-147.00
24.4	429.12	29.0	-142.39
24.5	466.39	29.1	-138.04
24.6	510.94	29.2	-133.94
24.7	565.14	29.3	-130.06
24.8	632.52	29.4	-126.38
24.9	718.53	29.5	-122.90
25.0	832.14	29.6	-119.59
25.1	989.16	29.7	-116.45
25.2	1220.36	29.8	-113.45
25.3	1594.54	29.9	-110.60
25.4	2303.69	30.0	-107.88
25.5	4161.94	30.1	-105.28
25.6	******	30.2	-102.80
25.7	*****	30.3	-100.42
		30.4	-96.14
25.8	*****	30.5	-95.96
25.9	*****	30.6	-93.86
26.0	*****	30.7	-91.85
26.1	• • • • • •	30.8	-89.91
26.2	-878.93	30 9 10455	-88.05
26.3	-747.58	30.9 LOSES	TACT -86.26
26.4	-650.06	31.1	-84.54
26.5	-574.79		-92.88
26.6	-514.95	31.2	-81.27
26.7	-466.22	31.3	-81.27
20.	100.22		A .

563 WHEELS LOSE CONTACT WHEN

22.3 km/h < U < 31.2 km/h